

The AMS02 Silicon Tracker: the detector and a first look to the on orbit data



G. Ambrosi

on behalf of the Tracker group

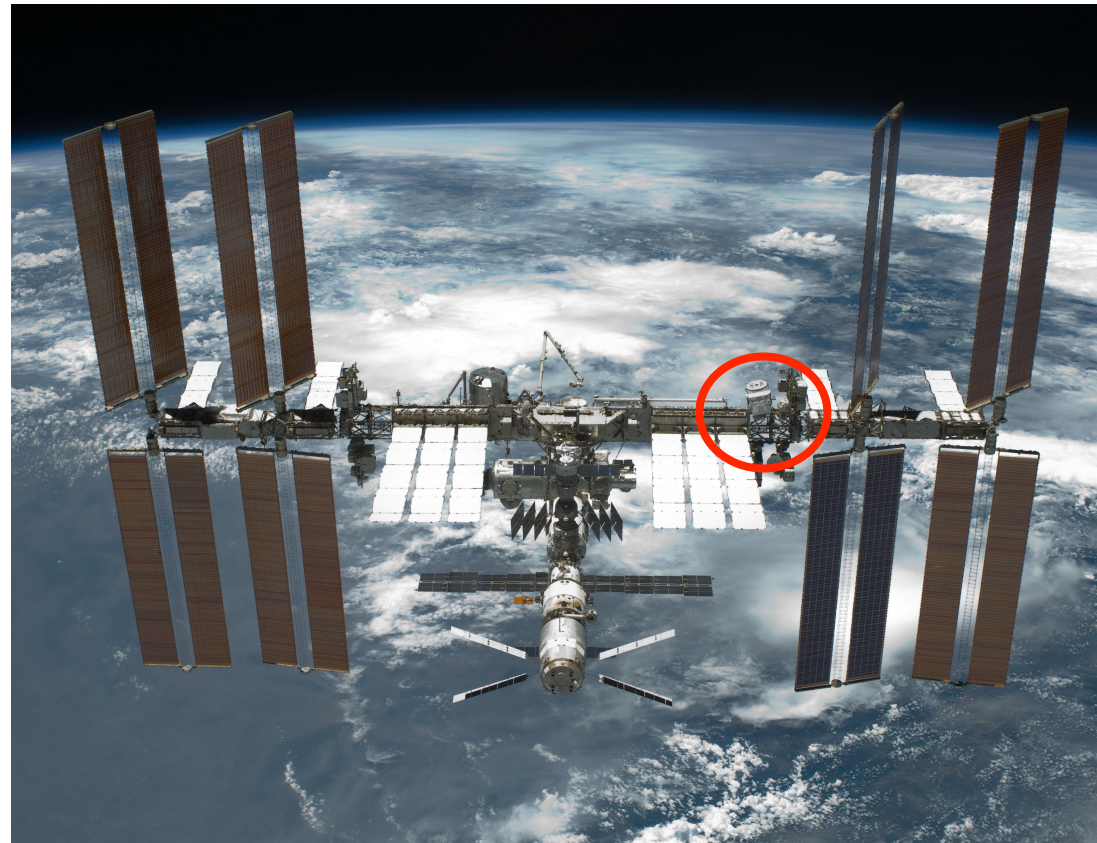
(Perugia INFN and University, Geneva University)



AMS on the International Space Station



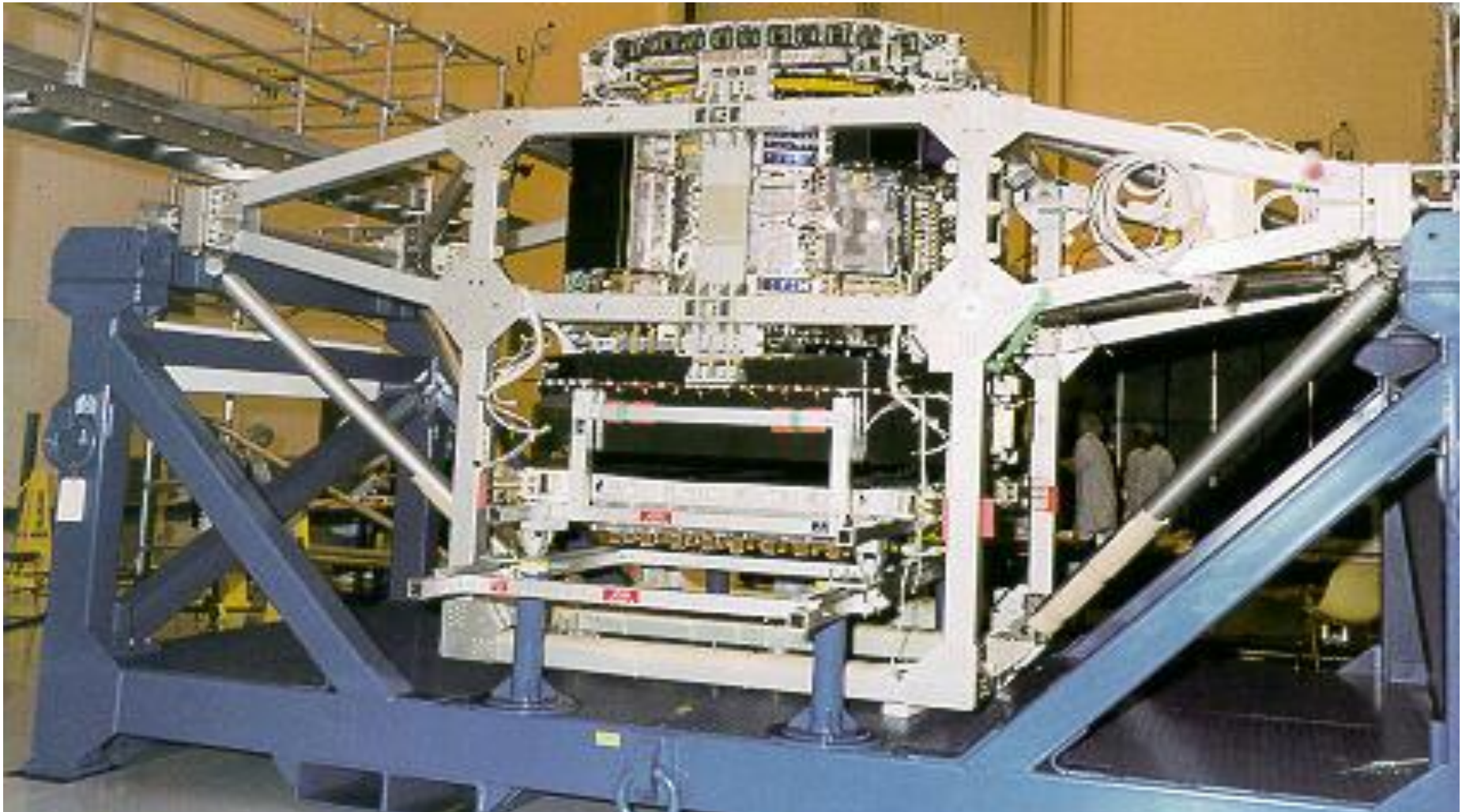
- Cosmic Antimatter search with 10^{-9} sensitivity
- Indirect Dark Matter search (e^+ , \bar{p} , γ)
- Relative abundance of nuclei and isotopes in primary cosmic rays
- γ ray astrophysics



The purpose of the AMS experiment is to perform accurate, high statistics, long measurements of charged cosmic rays (0.5 GV - 1 TV) and γ rays ($E > 1 \text{ GeV}$)



AMS01 at KSC (Florida) in 1998

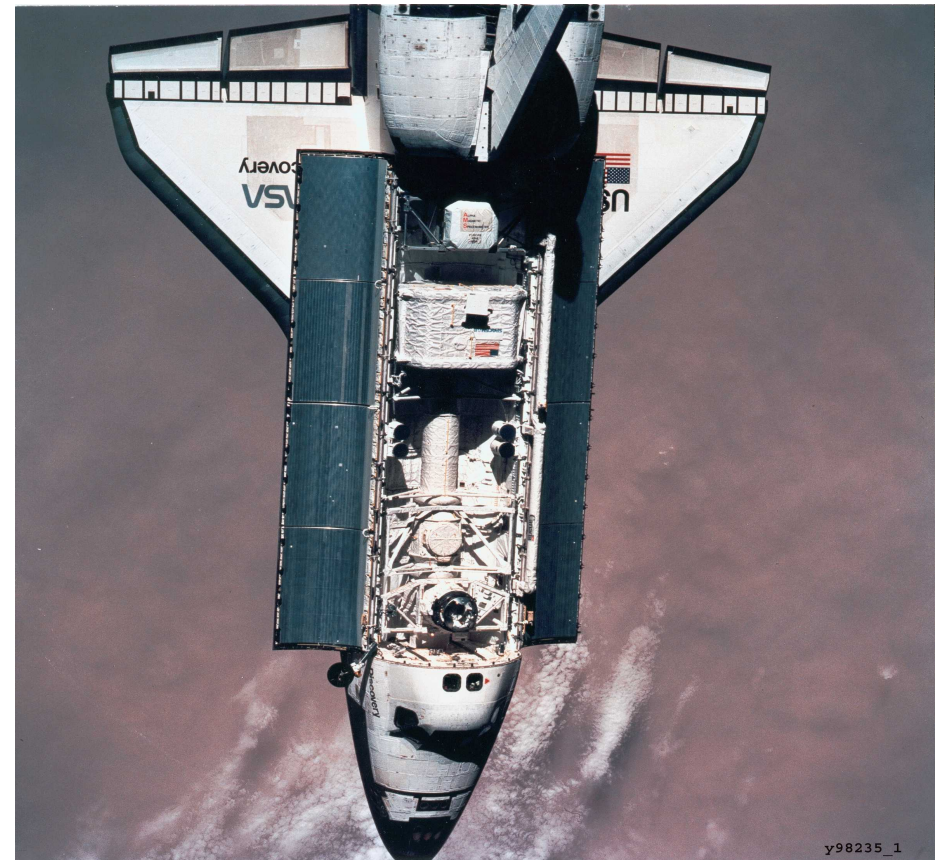




AMS-01 pilot experiment: STS91, June 2nd - 12th 1998



- 10 days of data taking in orbit:
 - 400 Km altitude
 - latitudes $+51.7^\circ$
 - all longitudes
- 10^8 events recorded
- Physics results
(Phys. Rep. 366 (2002) 331)
 - precise measurements of primary fluxes
 - detection of secondary fluxes (quasi trapped)
 - antimatter limit at 10^{-6}

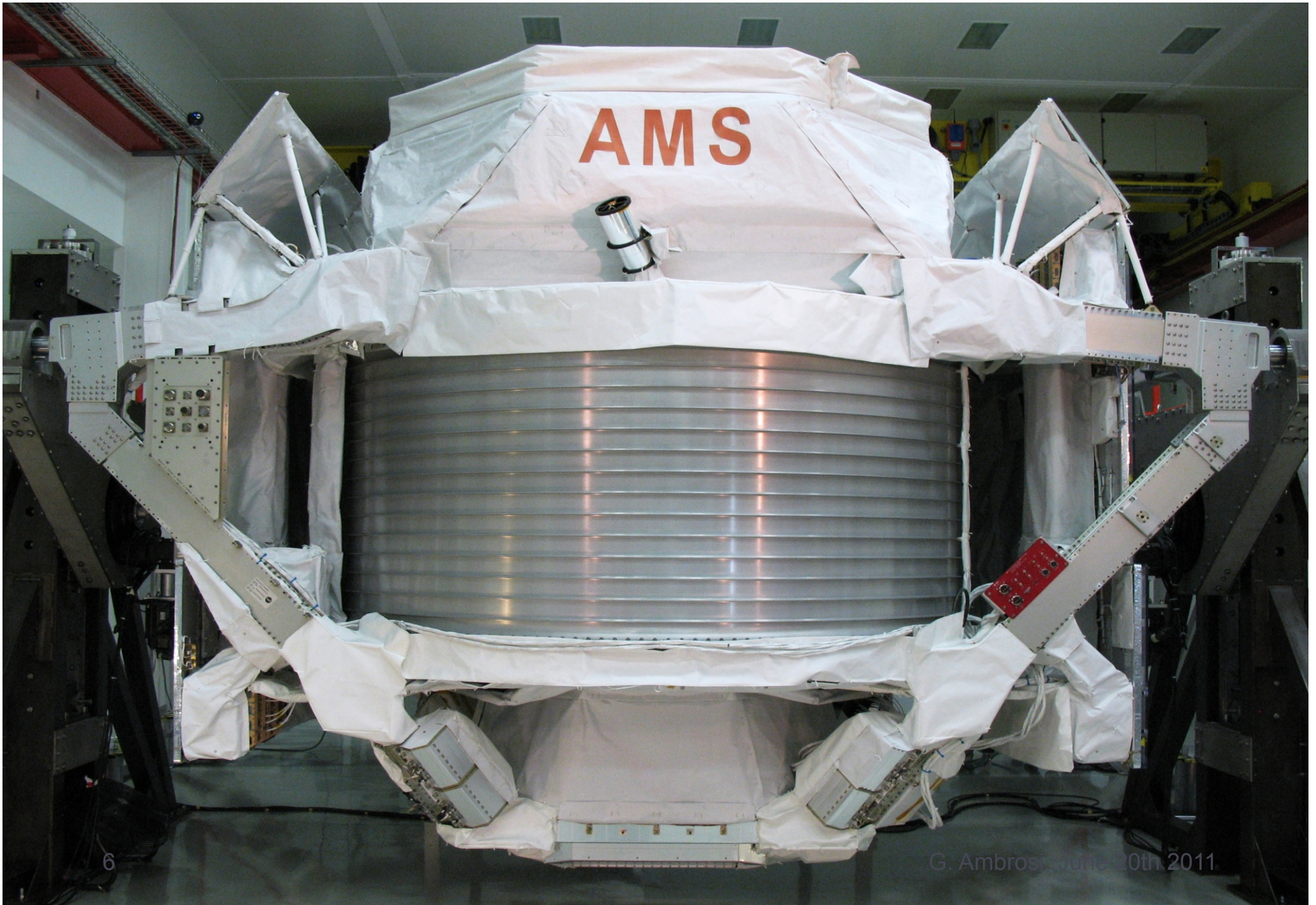


Tracker performance presented at Vertex 1998, Santorini



The instrument we need has ...

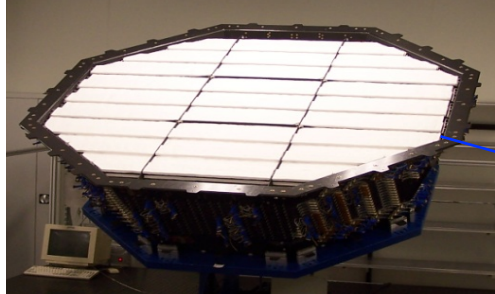
- performance a la 'particle physics':
 - high resolution measurements of momentum, velocity, charge and energy
- characteristics to properly work in the space environment:
 - Vibration (6.8 G rms) and acceleration (17 G)
 - Temperature variation (day/night $\Delta T = 100^{\circ}\text{C}$)
 - Vacuum (10^{-10} Torr)
 - Orbital debris and micrometeorites
 - Radiation (Single Event Effect)
- limitation in weight (15000 lb), power ($\sim 2\text{KW}$), bandwidth and maintenance
- Compliant with Electromagnetic Interference and Electromagnetic Compatibility specs



AMS

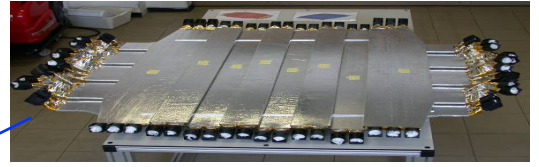
AMS: A TeV precision, multipurpose particle physics spectrometer in space.

TRD
Identify e^+ , e^-

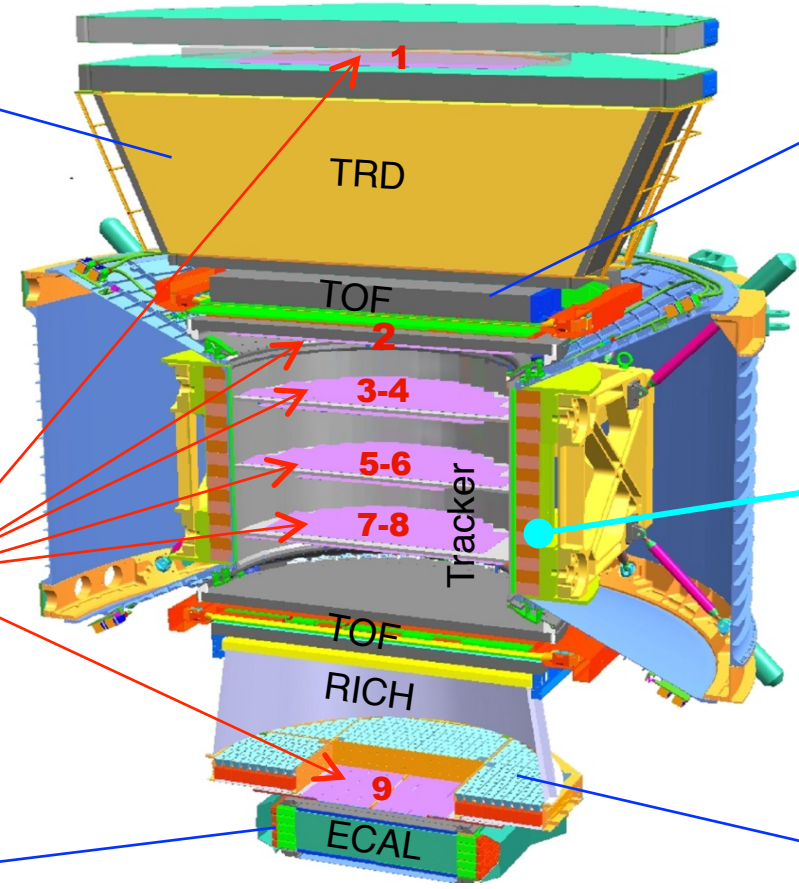
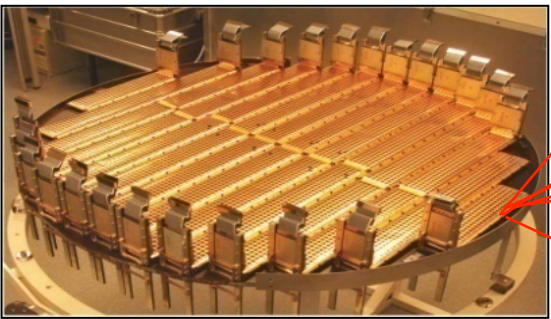


Particles and nuclei are defined by their charge (Z) and energy ($E \sim P$)

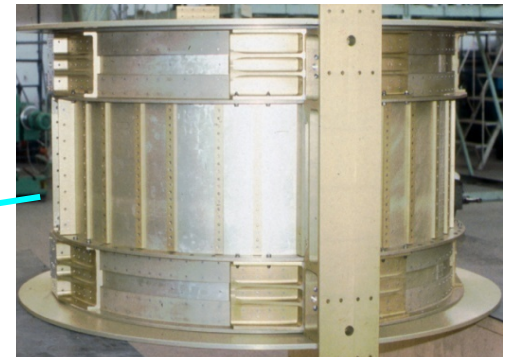
TOF
 Z, E



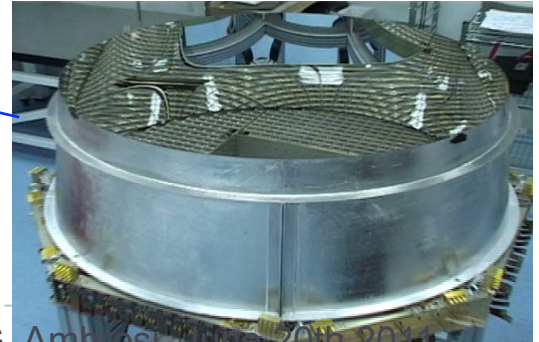
Silicon Tracker
 Z, P



Magnet
 $\pm Z$



RICH
 Z, E



ECAL
 E of e^+ , e^- , γ



Z, P are measured independently by the Tracker, RICH, TOF and ECAL



recent AMS02 history in short

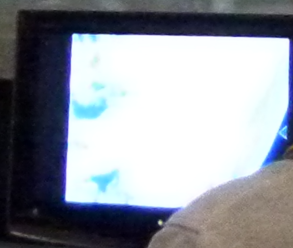
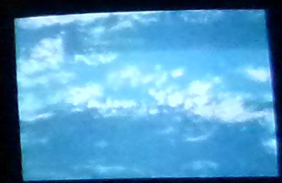
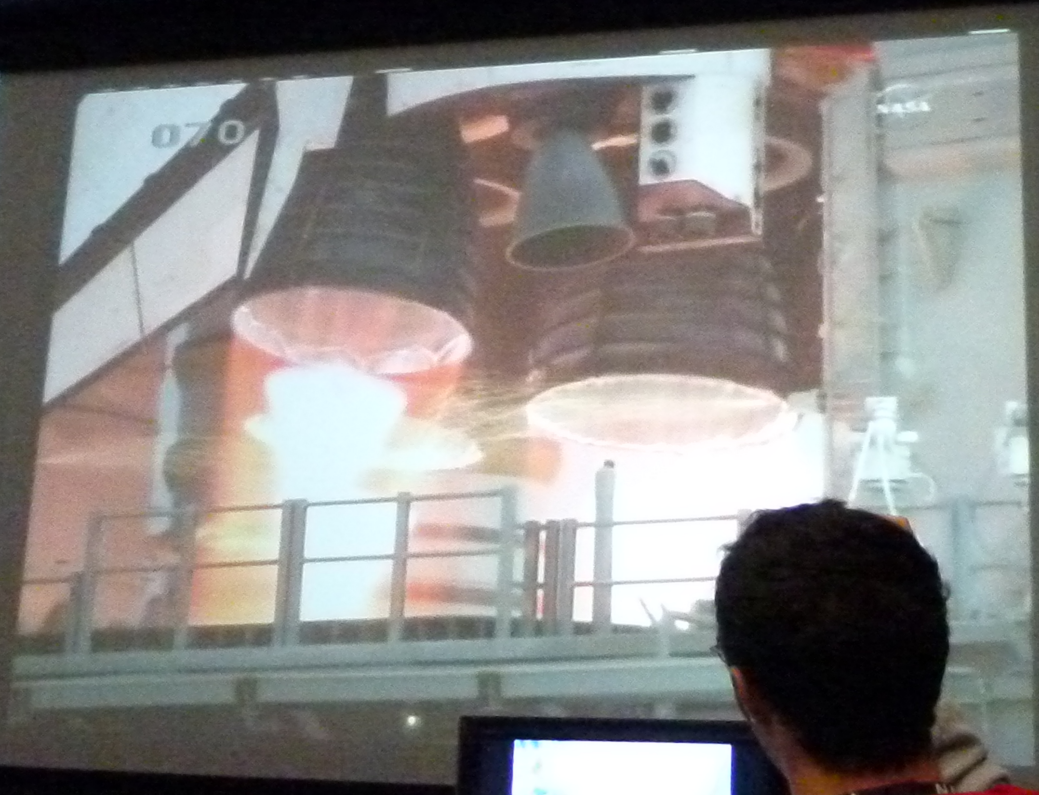


- fall 2009: integration at CERN
- February 2010: test beam at CERN
- spring 2010: EMI and TV test at ESTEC (ESA)
- late spring 2010: magnet replacement at CERN
- August 2010: test beam at CERN
- fall/winter 2010-2011 integration at KSC (Florida)
- May 16th 2011: launch!
- May 19th 2011: first activation in space:
everything is working!!



MET 000:00:00:01+
SBND MILA->POL 00:00:59-
SBND PDL ->MILA 00:02:29-
SBND MILA->TORS 00:07:29-

```
070  
[Log output showing system messages and error reports]
```



Launching G. Ambrosi, June 20th 2011



G. Ambrosi, June 20th 2011

May 16, 2011 @ 08:56 AM



G. Ambrosi, June 20th 2011

Monitoring Interface



Until now or

GTSN

JPD-A
JPD-B

CHECK

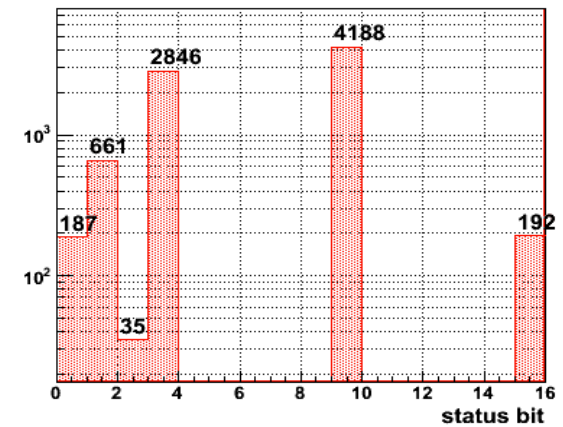
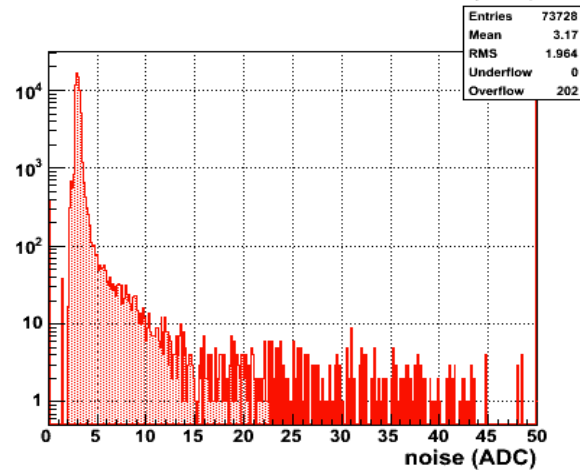
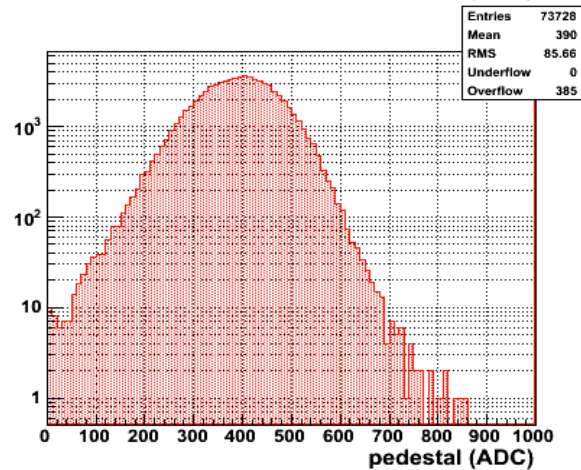
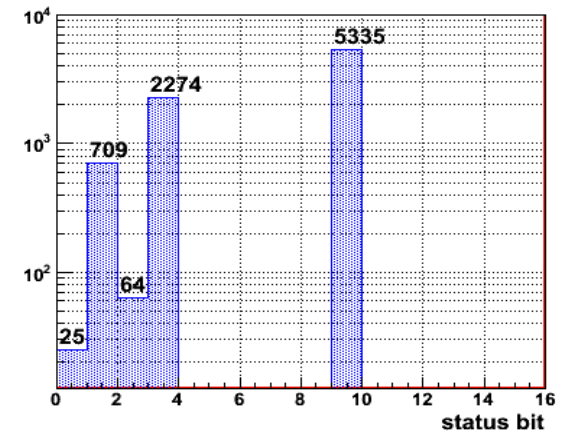
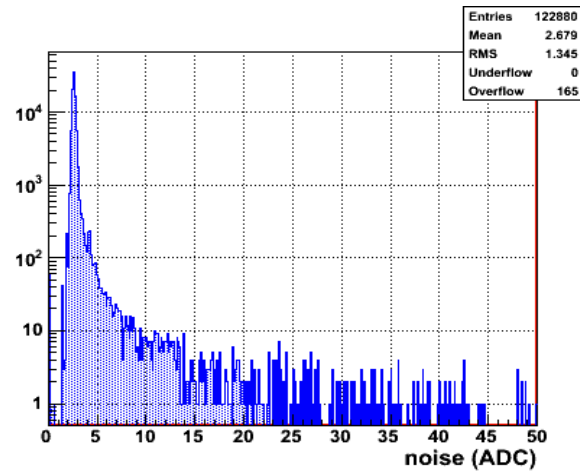
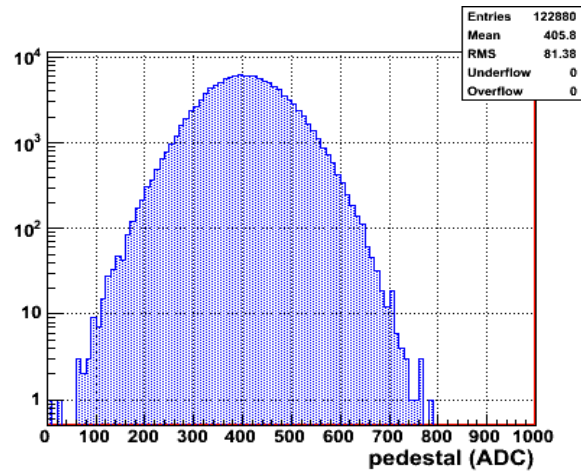
Everything OK

| | |
|------------------|------------|
| MPD @ TMPD2 | 13.875 °C |
| M | 11.9375 °C |
| GPS | 12.5 °C |
| TT | 14.0625 °C |
| TTCBP | 16.0625 °C |
| TTCBS | 16.0625 °C |
| UGPD | 13.75 °C |
| UG | 12.5 °C |
| CCEB Signal Side | 13.625 °C |
| CCEB Power Side | 13.5625 °C |
| UPD0 | 13.6875 °C |
| U0 | 12.1875 °C |
| UPD1 | 13.8125 °C |
| U1 | 11.625 °C |
| SPD0 @ TSPD1 | 13.6875 °C |
| S0 | 11.9375 °C |
| SHV0 | 13.8125 °C |
| SPD1 @ TSPD3 | 13.5625 °C |
| S1 | 12.0 °C |
| SHV1 | 13.0625 °C |
| SPD2 @ TSPD4 | 13.625 °C |
| S2 | 14.0625 °C |
| SHV2 | 13.3125 °C |
| SPD3 @ TSPD6 | 13.875 °C |
| S3 | 14.3125 °C |

2.5 h after the launch



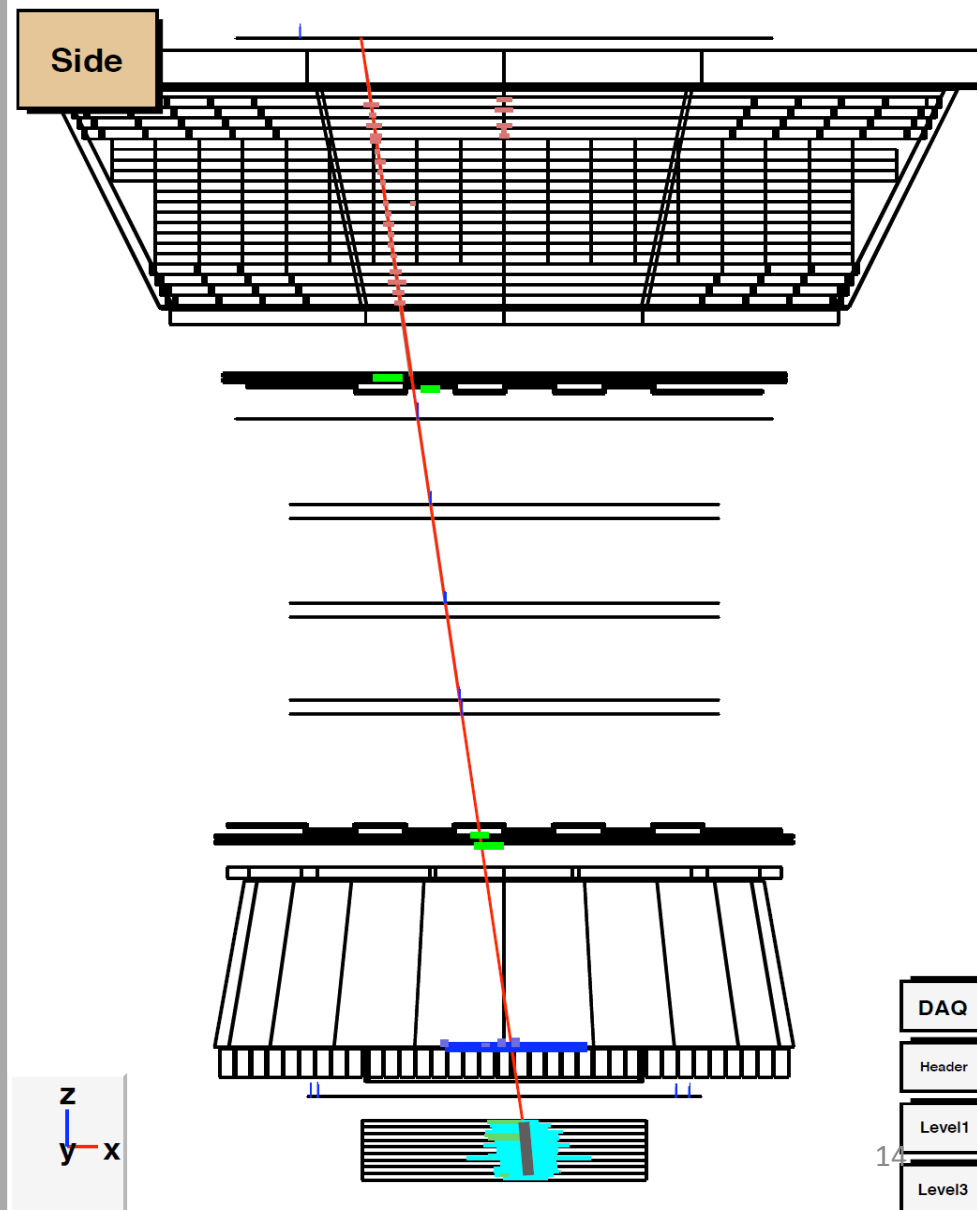
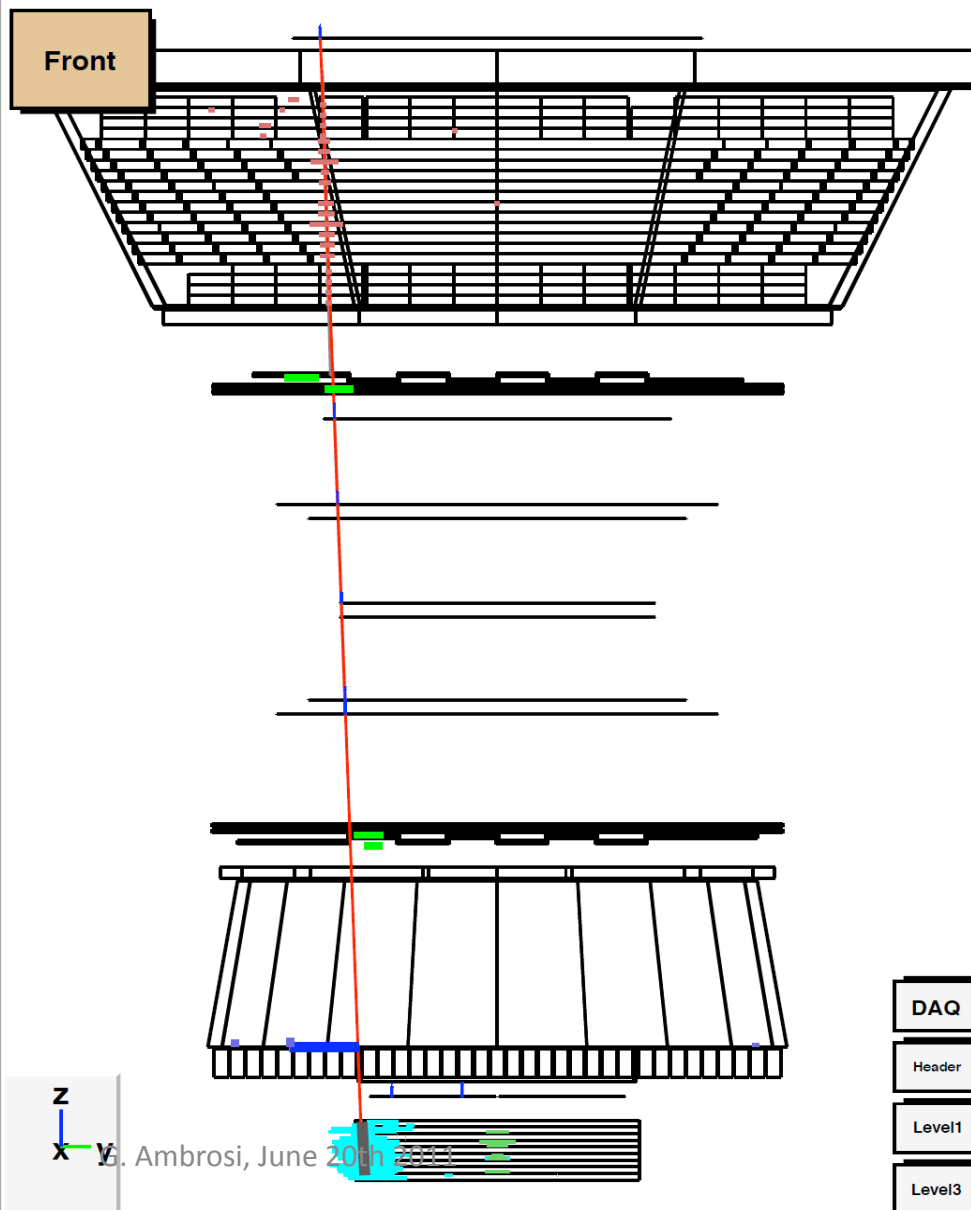
First Tracker calibration in space



Data from the 1st few minutes – 20 GeV Electron, 19 May 2011

AMS Event Display

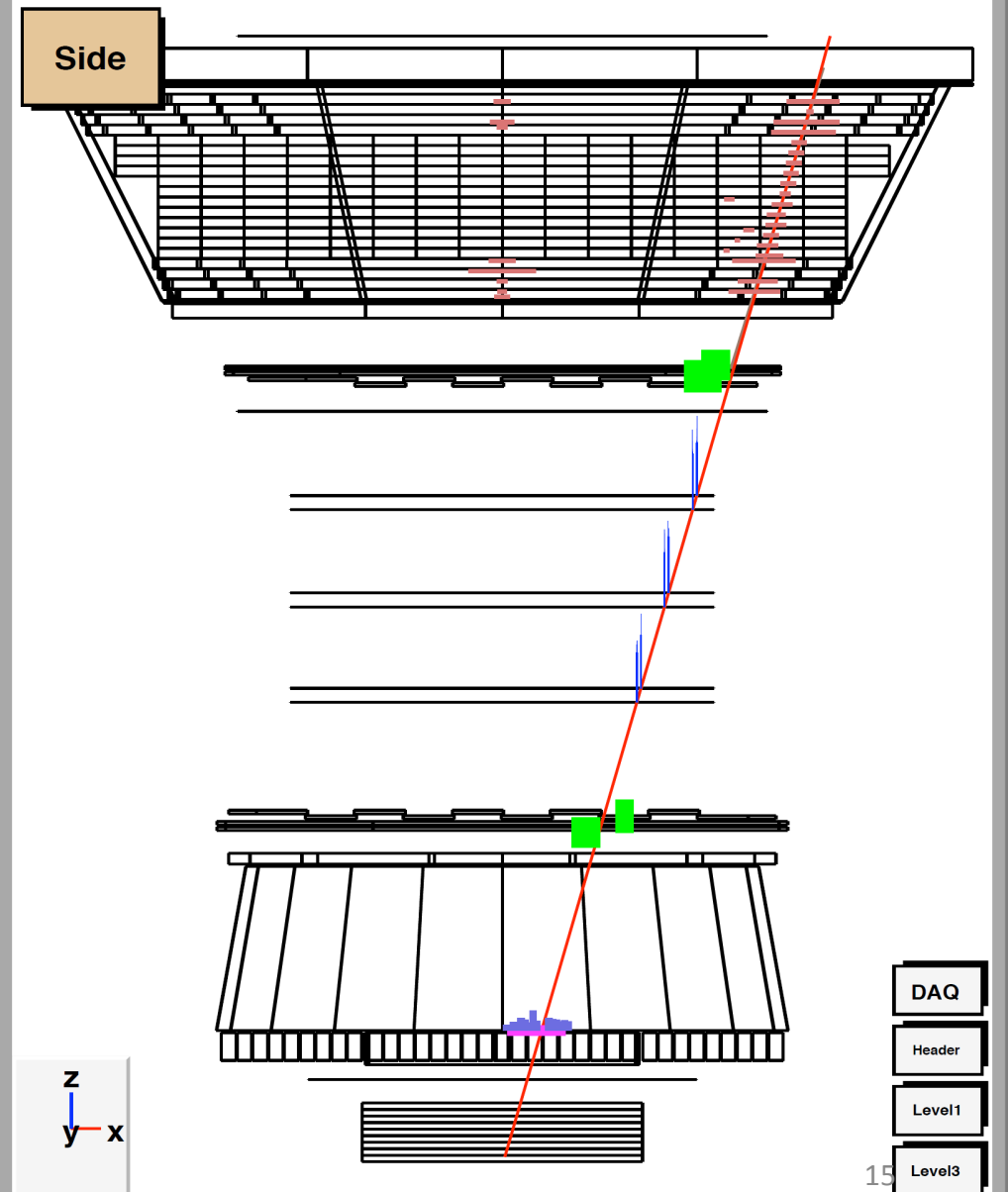
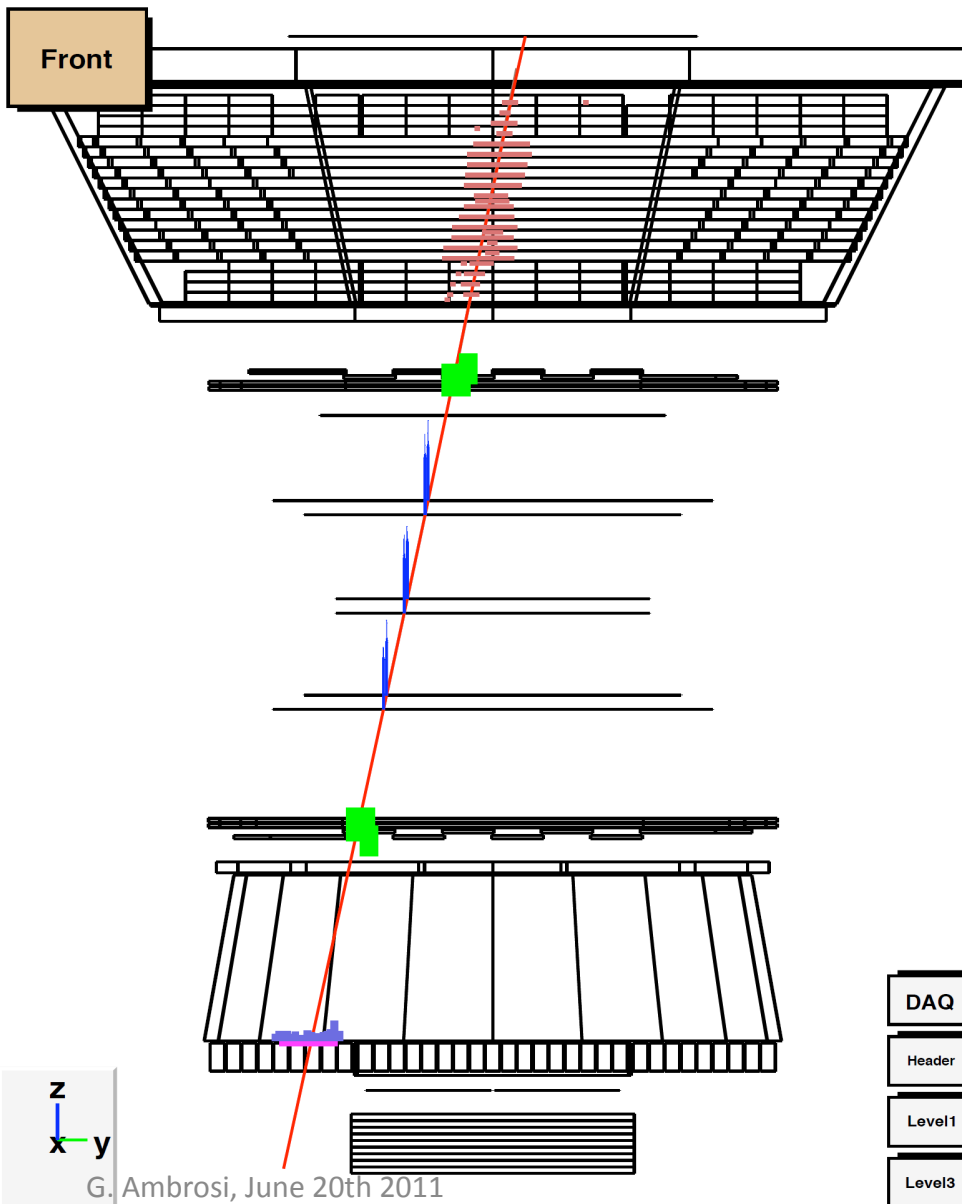
Run 1305815610/ 114493 Thu May 19 16:38:57 2011



Data from the 1st few minutes – 42 GeV/c Carbon, 19 May 2011

AMS Event Display

Run 1305815610/ 224169 Thu May 19 16:42:29 2011



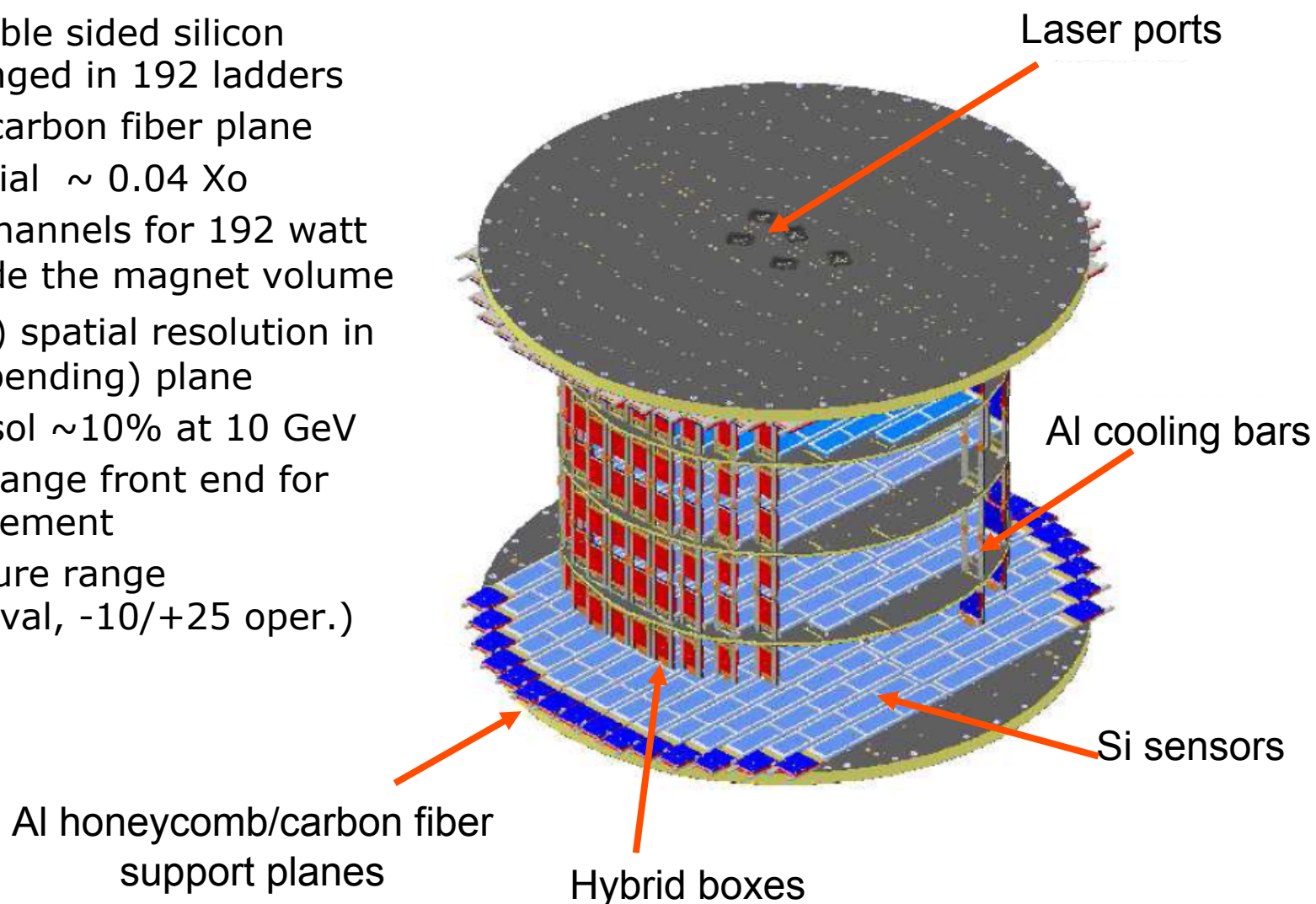
G. Ambrosi, June 20th 2011

15



Silicon Tracker

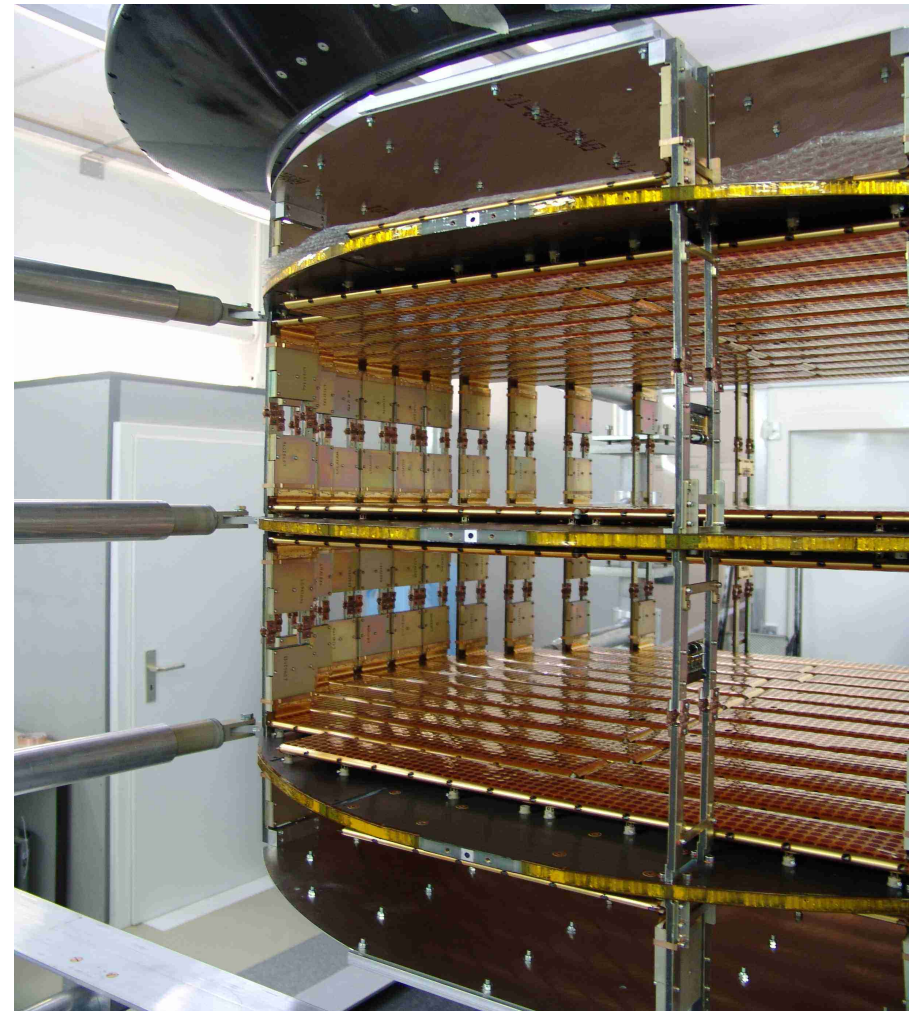
- 9 layers of double sided silicon detectors arranged in 192 ladders
- 6 honeycomb carbon fiber plane
- detector material $\sim 0.04 X_0$
- total of 200 kchannels for 192 watt dissipated inside the magnet volume
- $10 \mu\text{m}$ ($30 \mu\text{m}$) spatial resolution in bending (non bending) plane
- momentum resol $\sim 10\%$ at 10 GeV
- high dynamic range front end for charge measurement
- wide temperature range (-20/+40 survival, -10/+25 oper.)





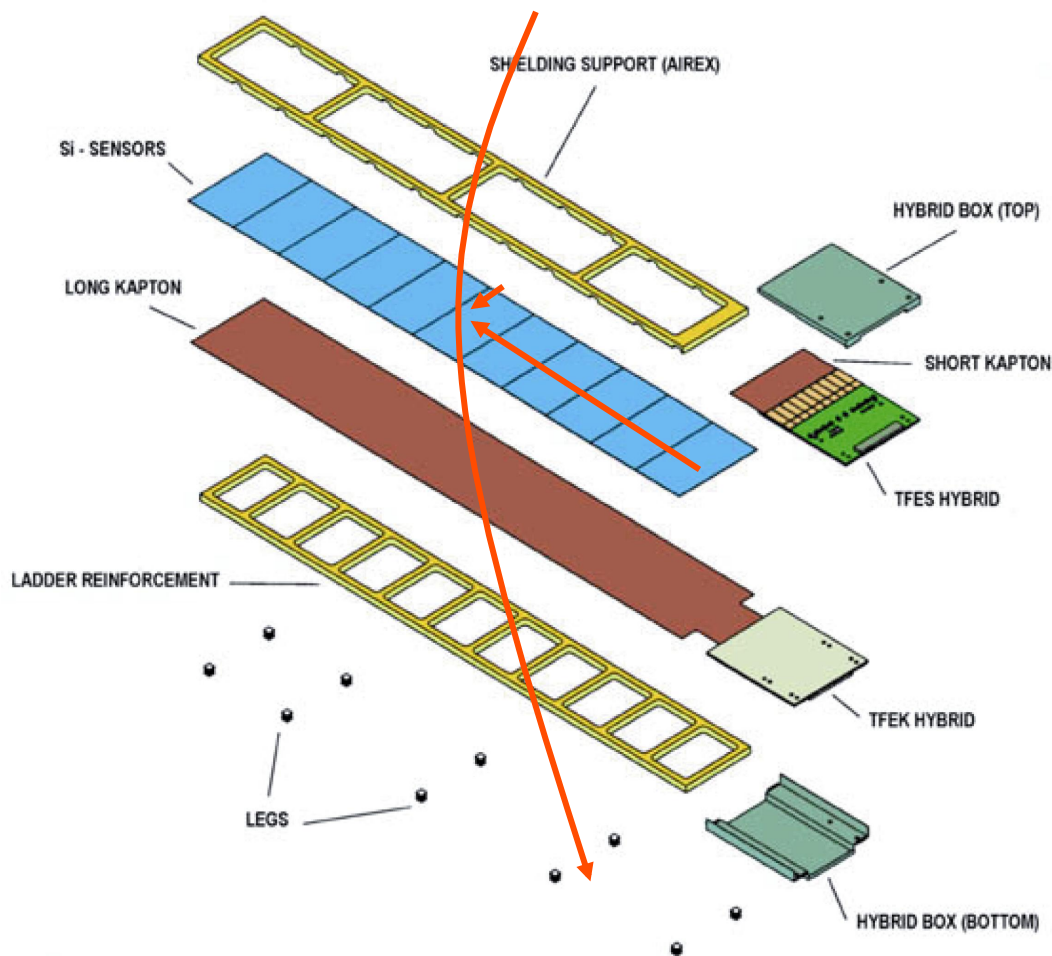
Silicon Tracker

- 9 layers of double sided silicon detectors arranged in 192 ladders
- 6 honeycomb carbon fiber plane
- detector material $\sim 0.04 X_0$
- total of 200 kchannels for 192 watt dissipated inside the magnet volume
- 10 μm (30 μm) spatial resolution in bending (non bending) plane
- momentum resol $\sim 10\%$ at 10 GeV
- high dynamic range front end for charge measurement
- wide temperature range (-20/+40 survival, -10/+25 oper.)





AMS silicon ladders

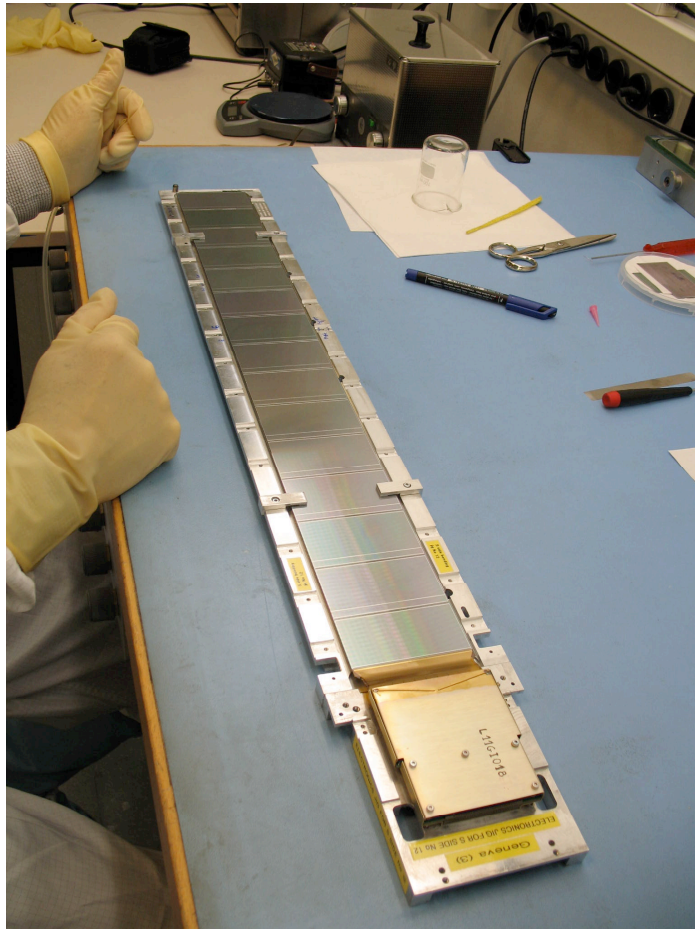


- 1024 high dynamic range, AC coupled readout channels:
 - 640 on junction (S) side
 - 384 on ohmic (K) side
- Impl/readout pitch:
 - 27.5/110 μm (S side)
 - 104/208 μm (K side)
- 7 - 15 wafers (28 - 60 cm)

192 flight units, 210 assembled in 3 lines:
Perugia (I), Geneva-ETHZ (CH), G&A (Carsoli, I)



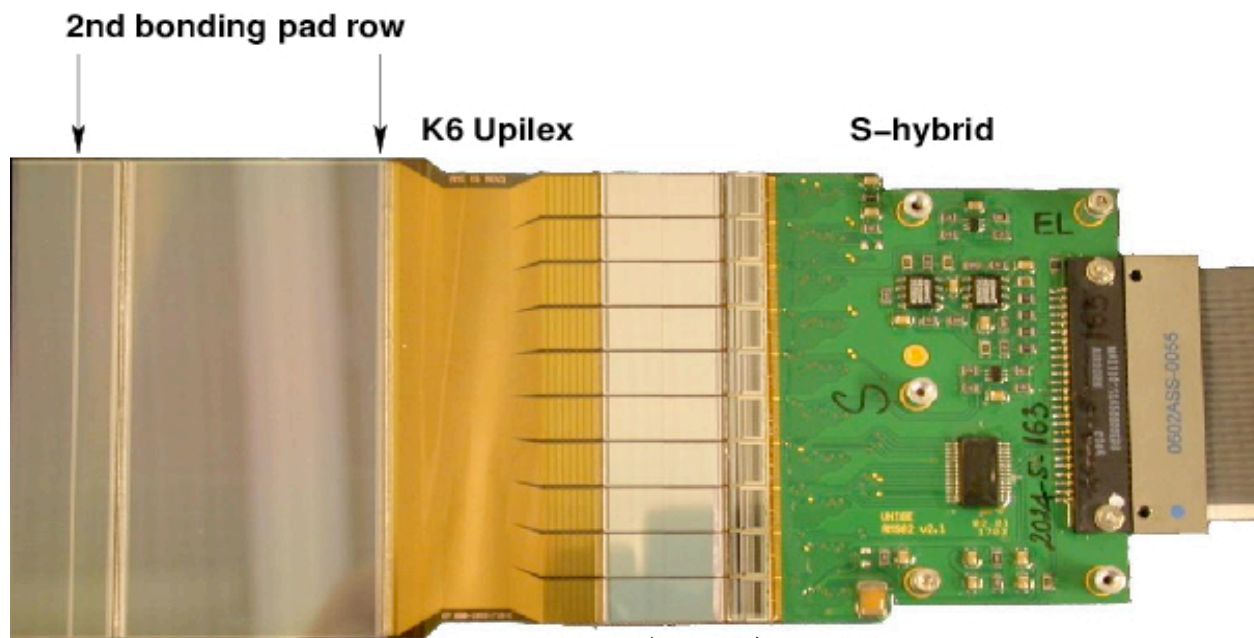
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192 flight units, 210 assembled in 3 lines:
Perugia (I), Geneva-ETHZ (CH), G&A (Carsoli, I)

Ladder components (p side)



double sided, DC coupled
 300 μm thickness
 7 - 15 sensors in a ladder
 produced at:
 - Colybris (CH)
 - IRST (IT)

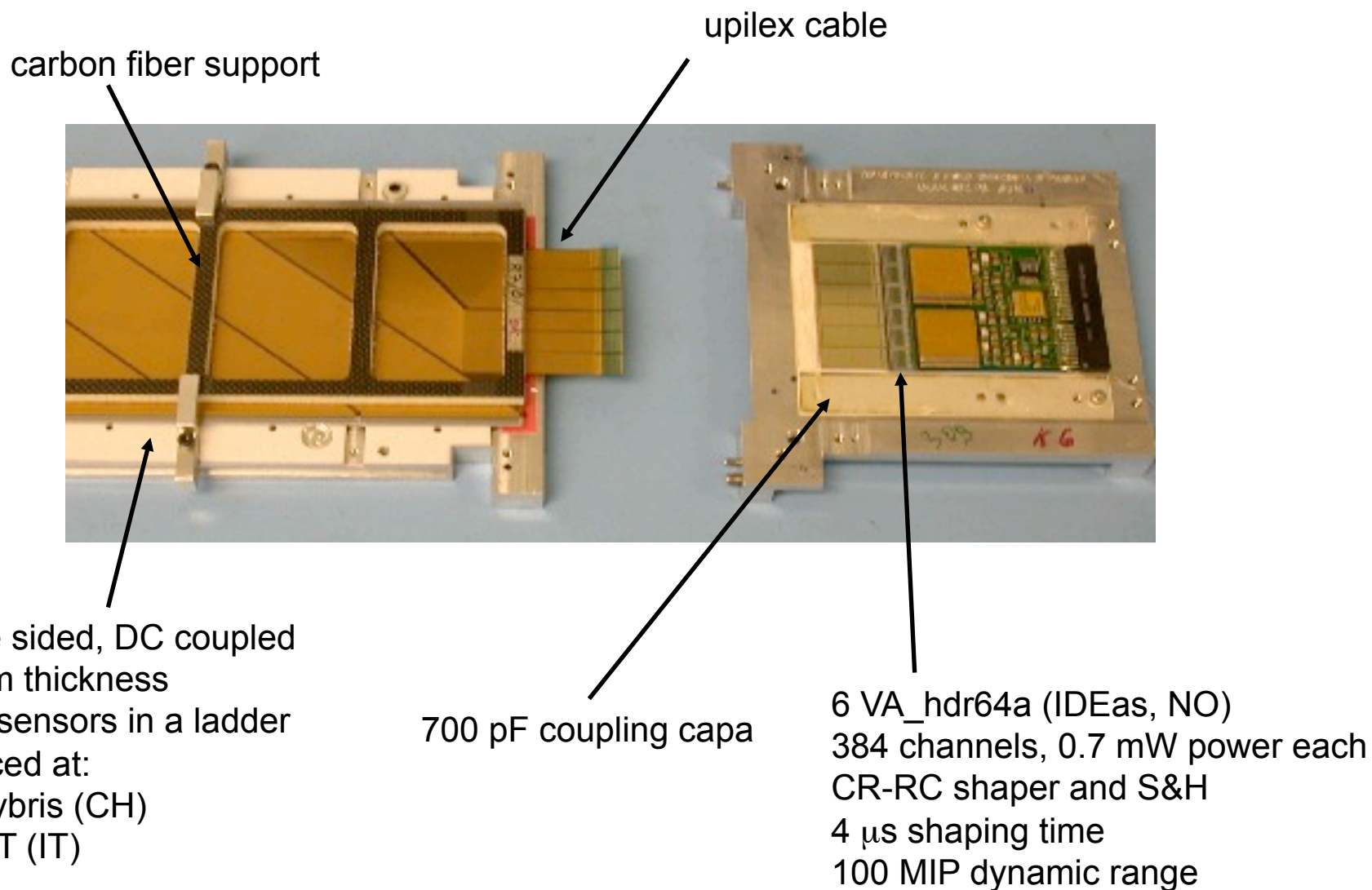
1st sensor edge

700 pF coupling capa

10 VA_hdr64a (IDEas, NO)
 640 channels, 0.7 mW power each
 CR-RC shaper and S&H
 4 μs shaping time
 100 MIP dynamic range



Ladder components (n side)





Radiation 'hard' electronics

The problem are the SEE (Single Event Effect)

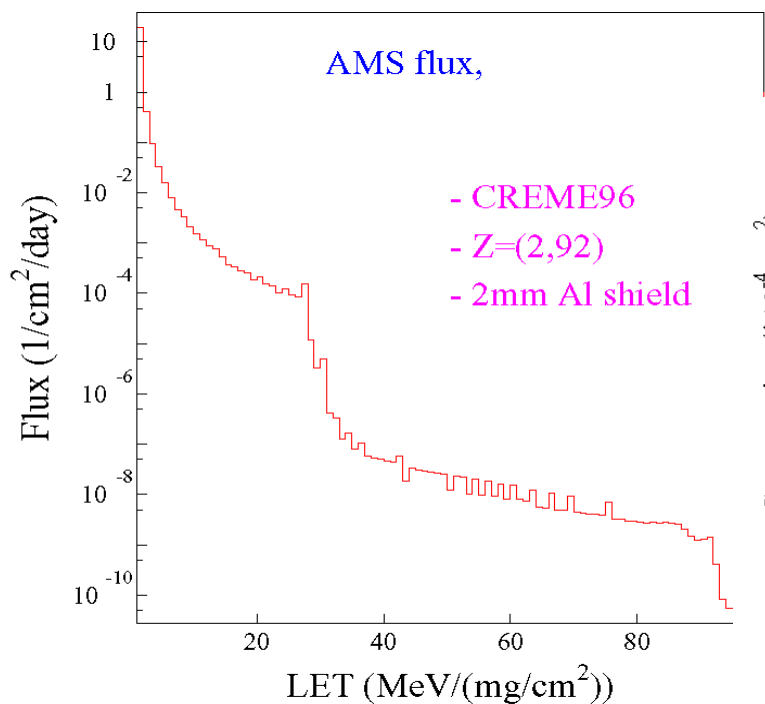


Figure 5: Expected fluxes on ISS in 2003.

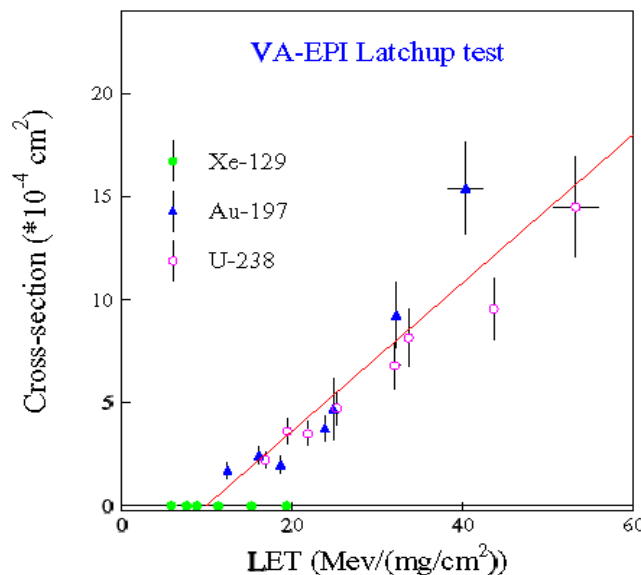
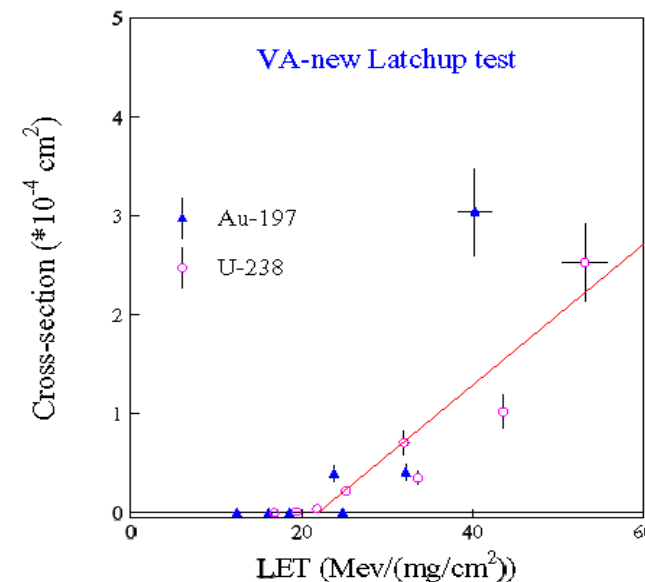


Figure 12: The new VA – SEL rates as measured in GSI



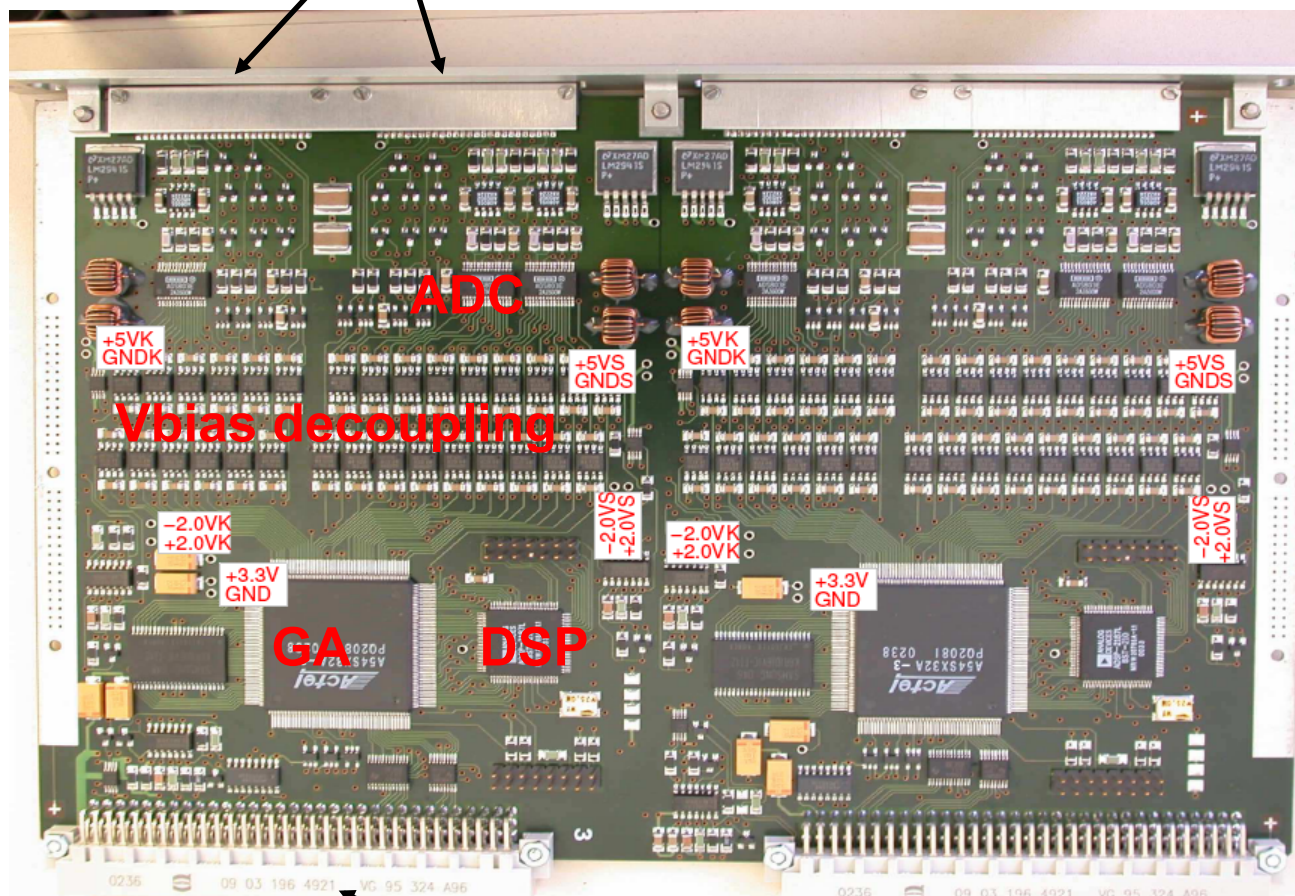
current limit protection is present for all active components



Data Reduction Board (TDR2)



analog signal in



compressed digital out

Collect analog data and digitize it (100 μ s irred. dead time)

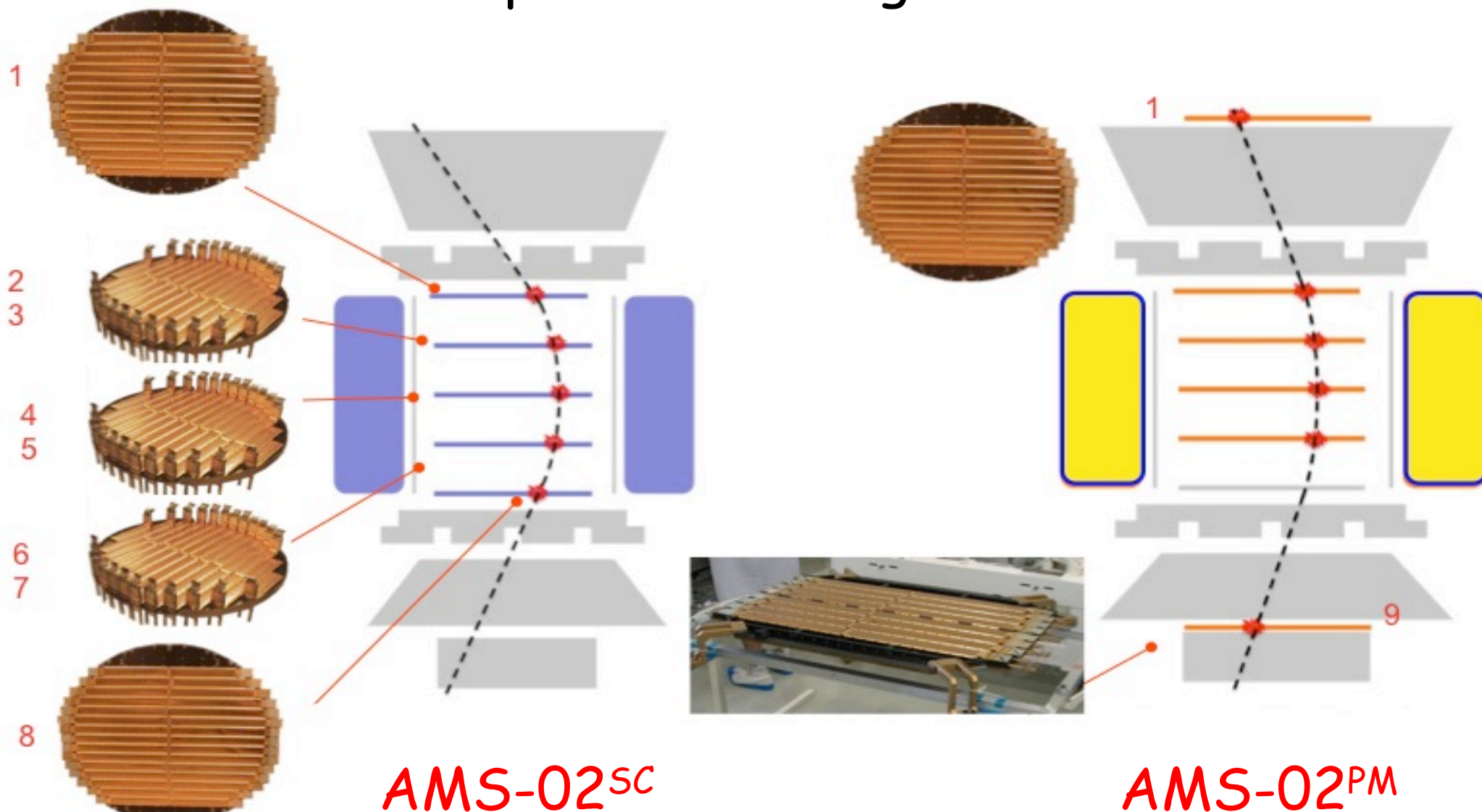
Perform online data compression

- Remove Pedestals
- Calculate and Remove Common Noise
- Search Clusters

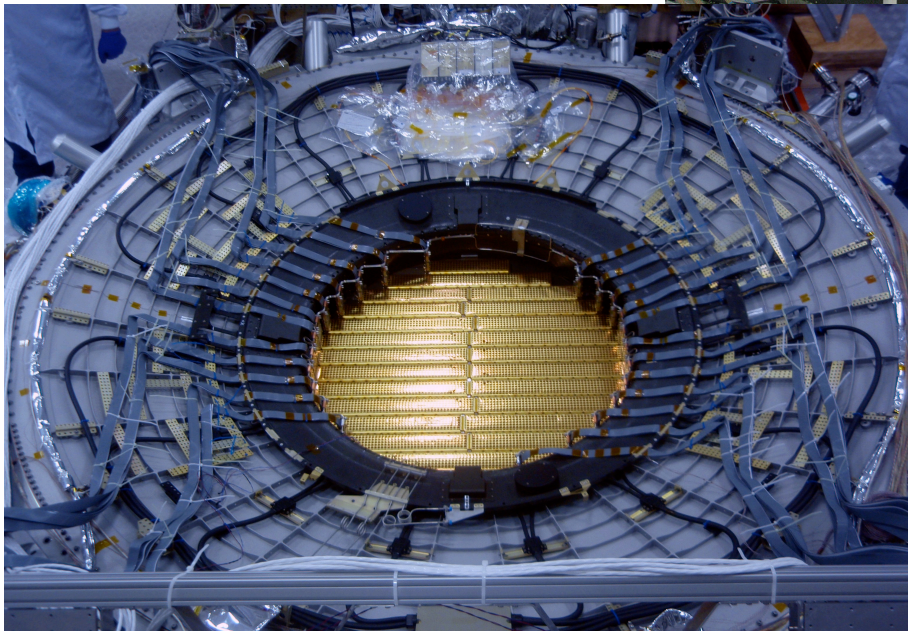
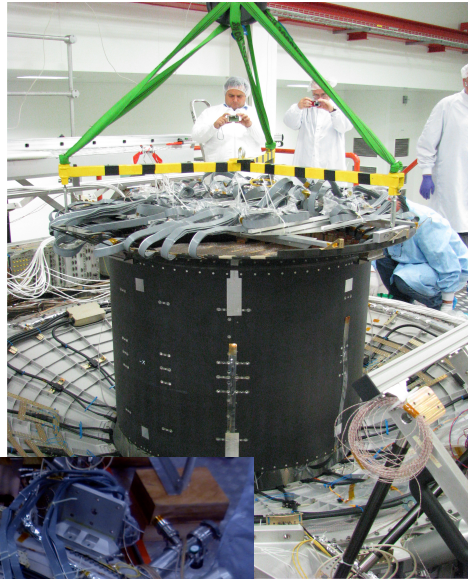
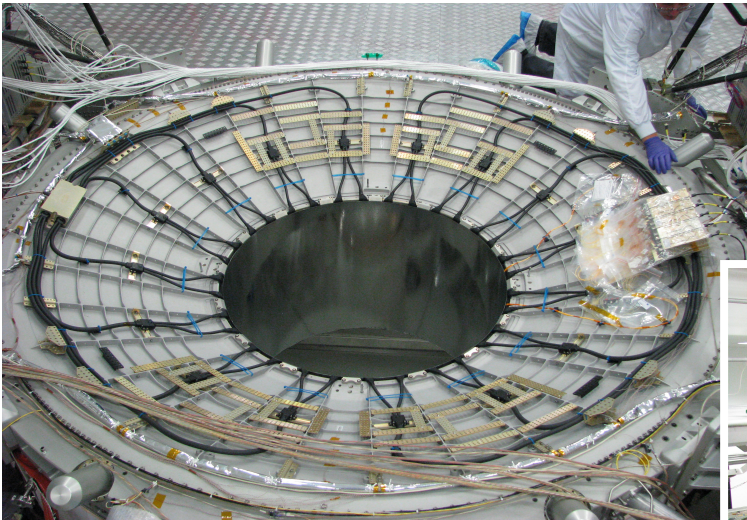
Up to 5 KHz trigger rate in compressed mode

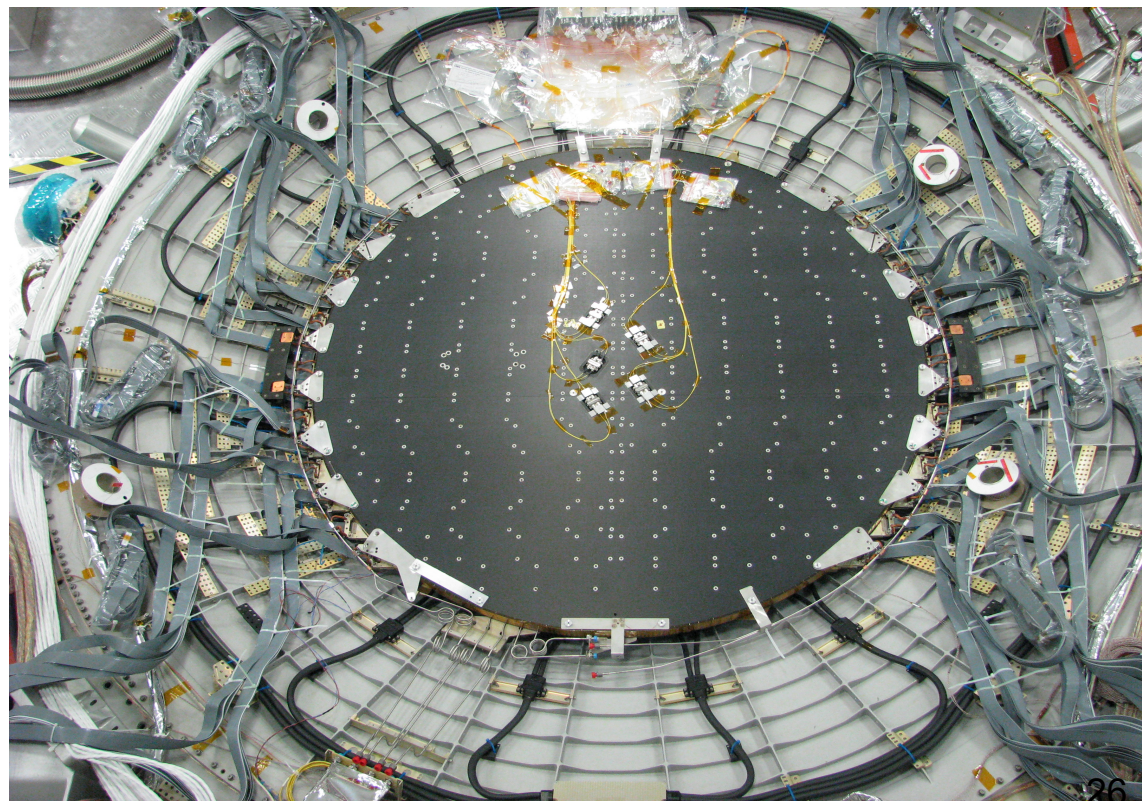
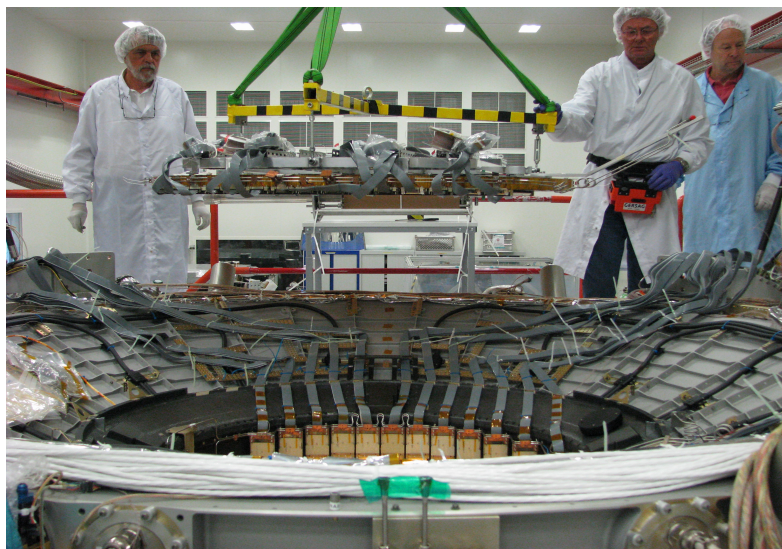
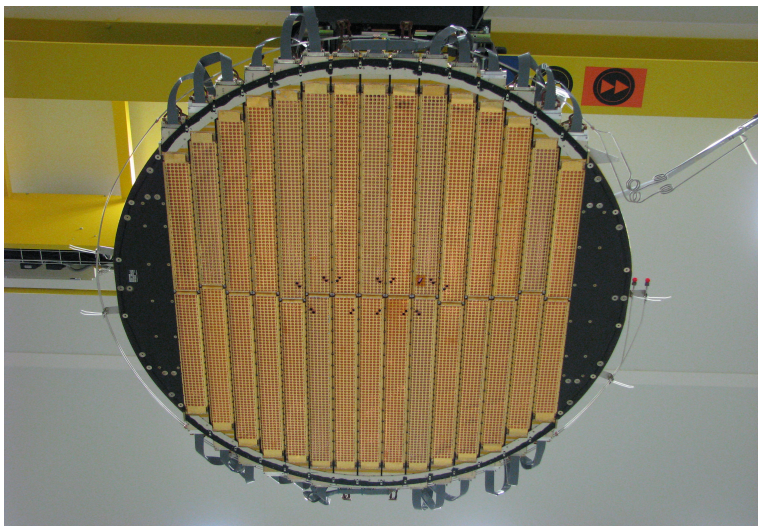


Layout modification to use the permanent magnet



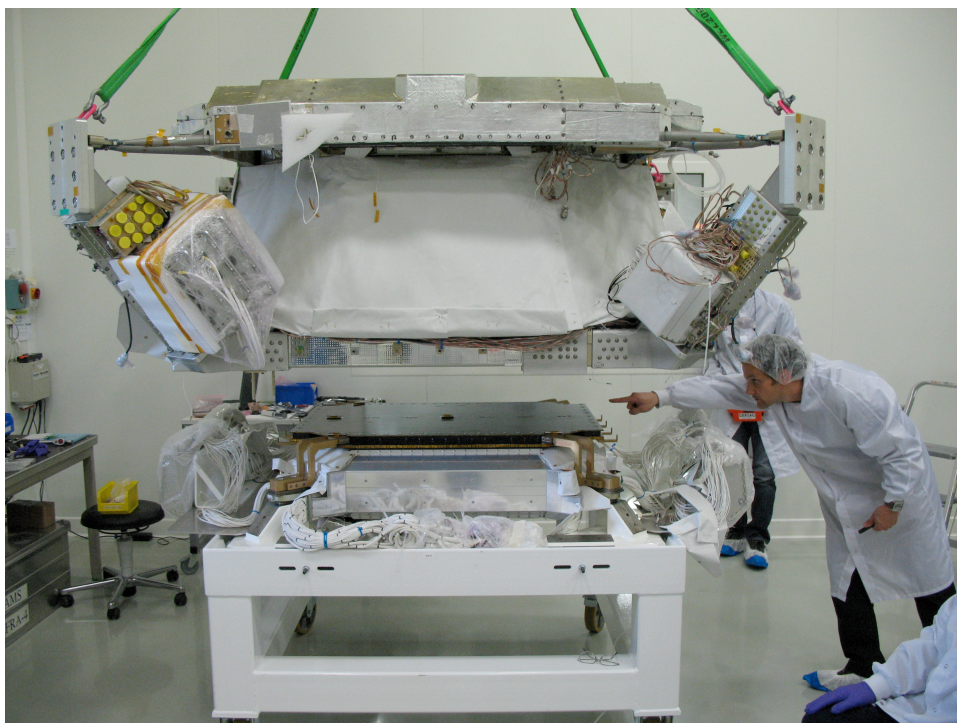
inner Tracker integration





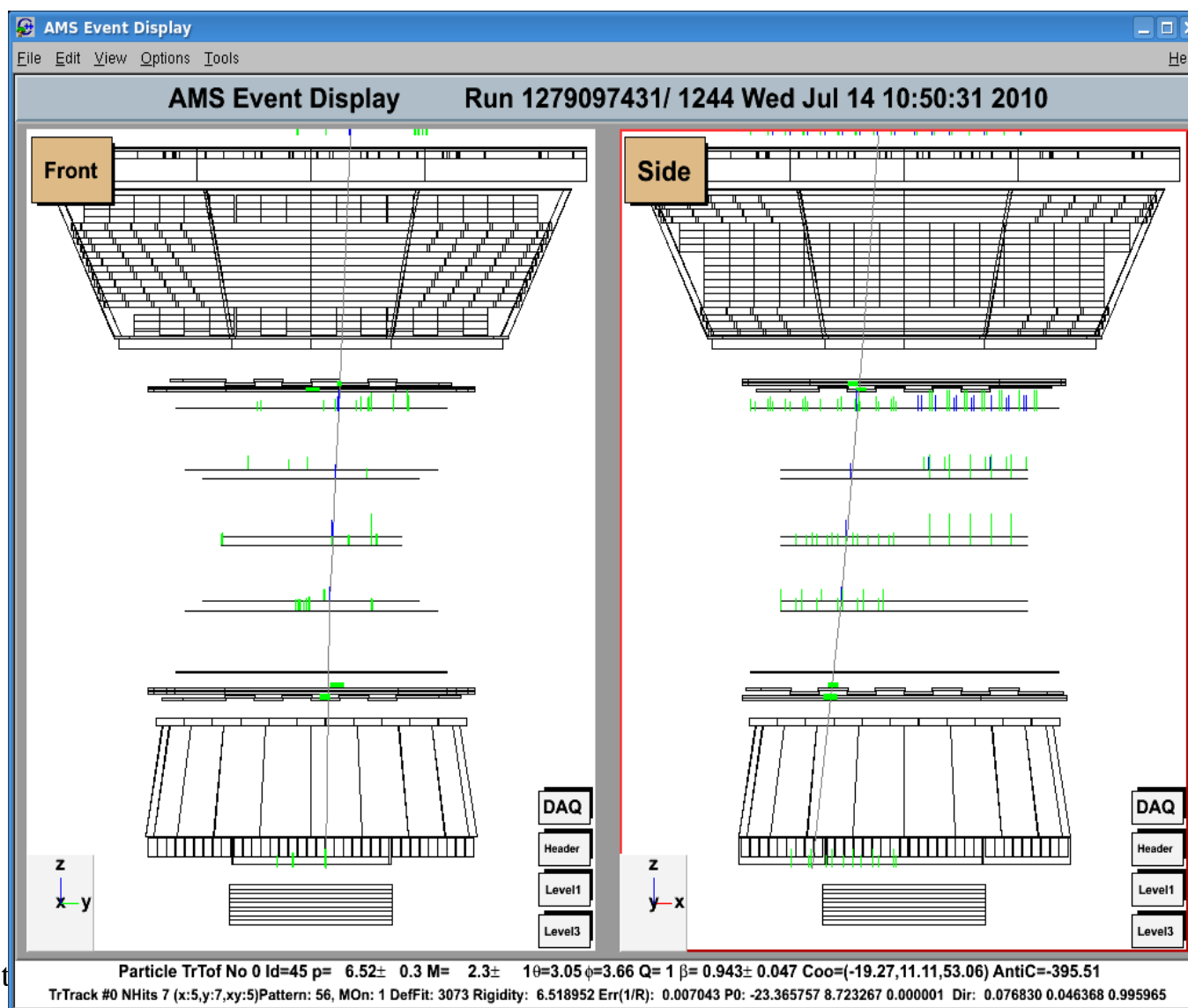


Tracker integration (2)



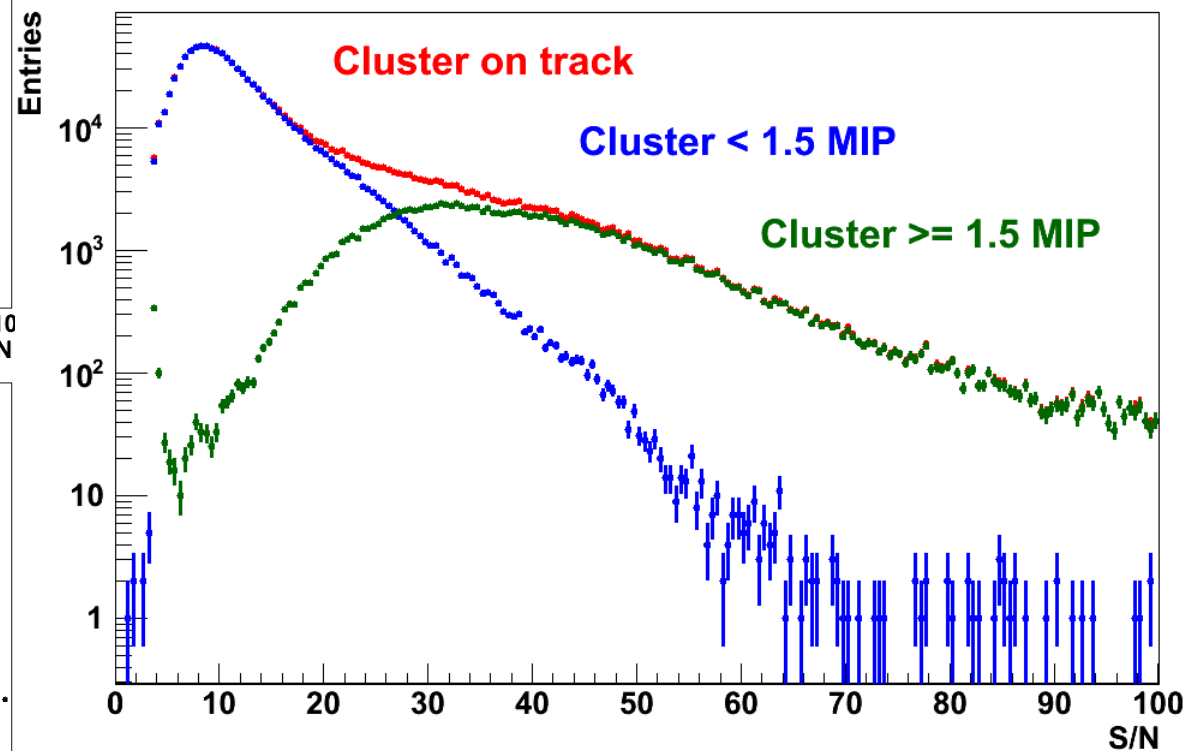
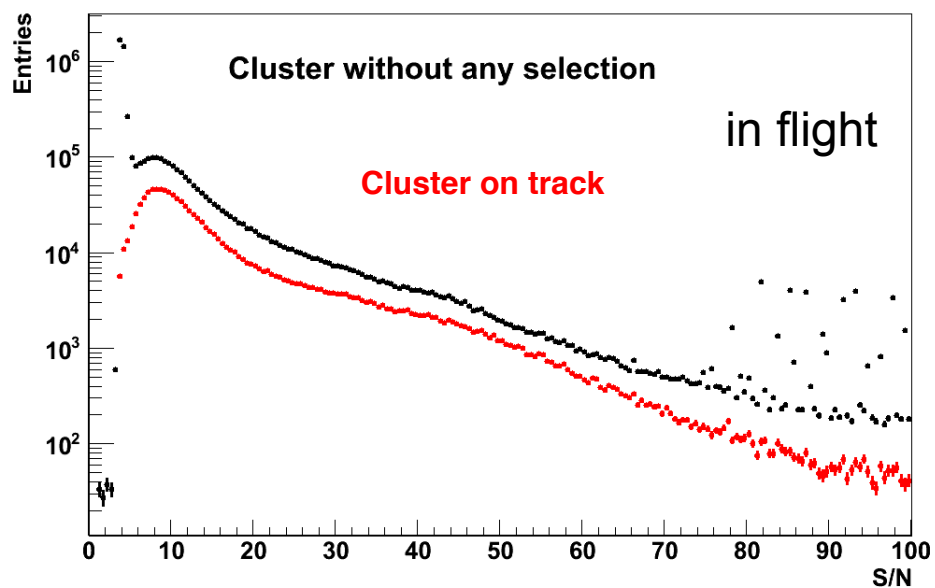
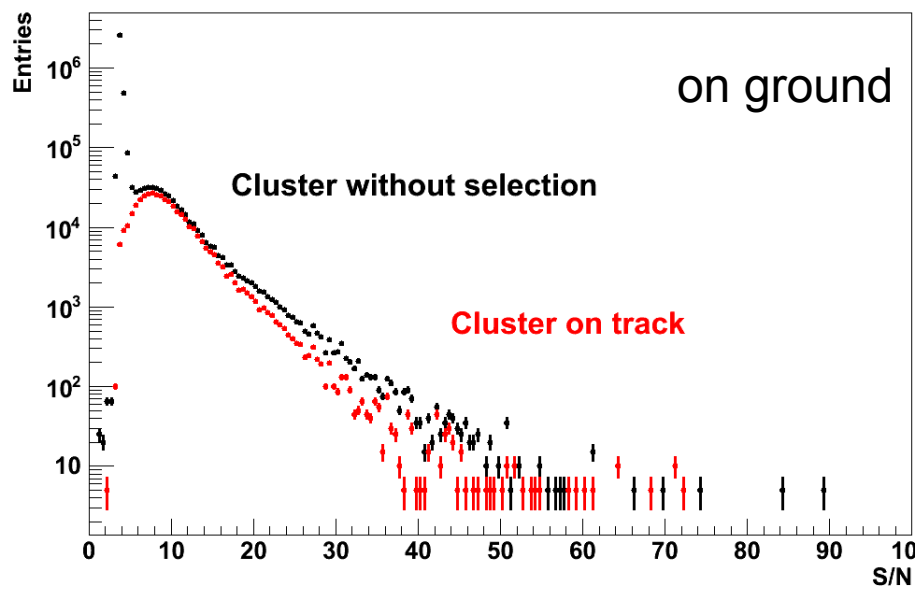


the first muon with the new Tracker



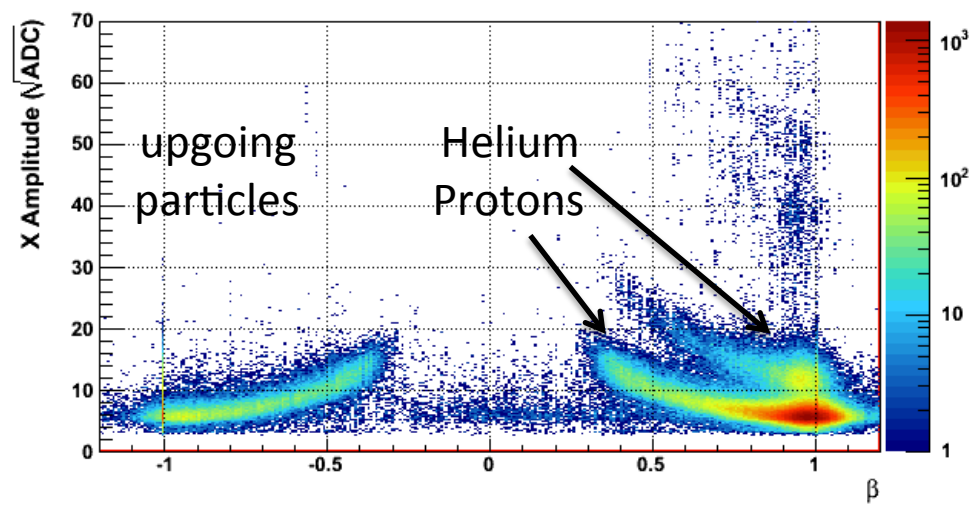
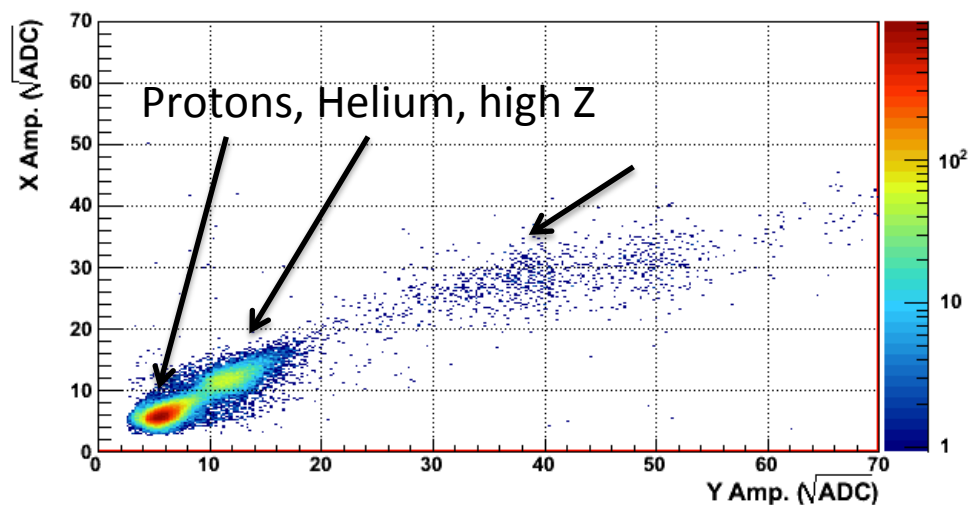
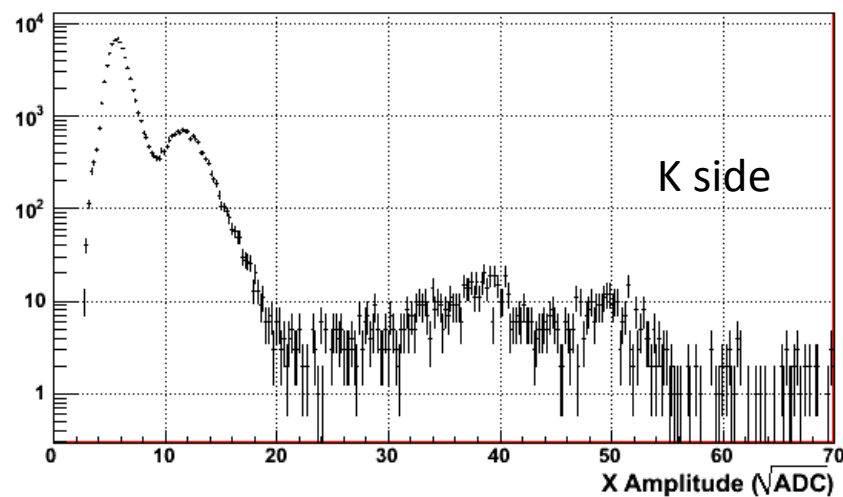
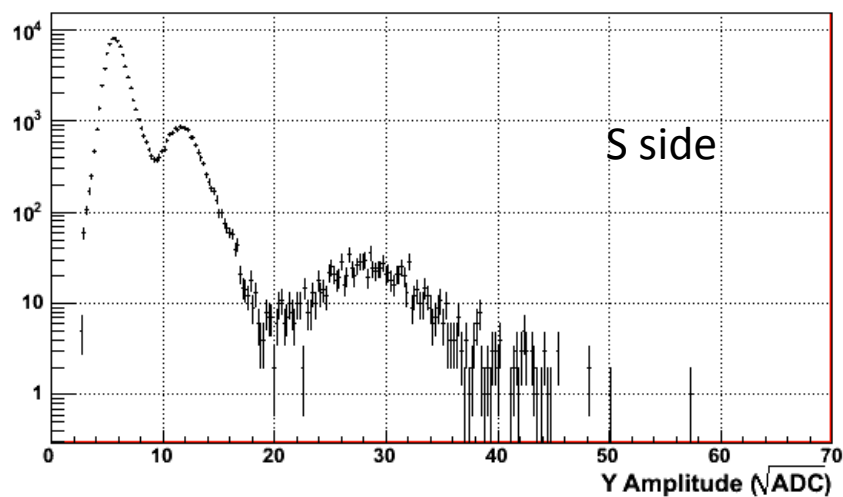


the Tracker signals



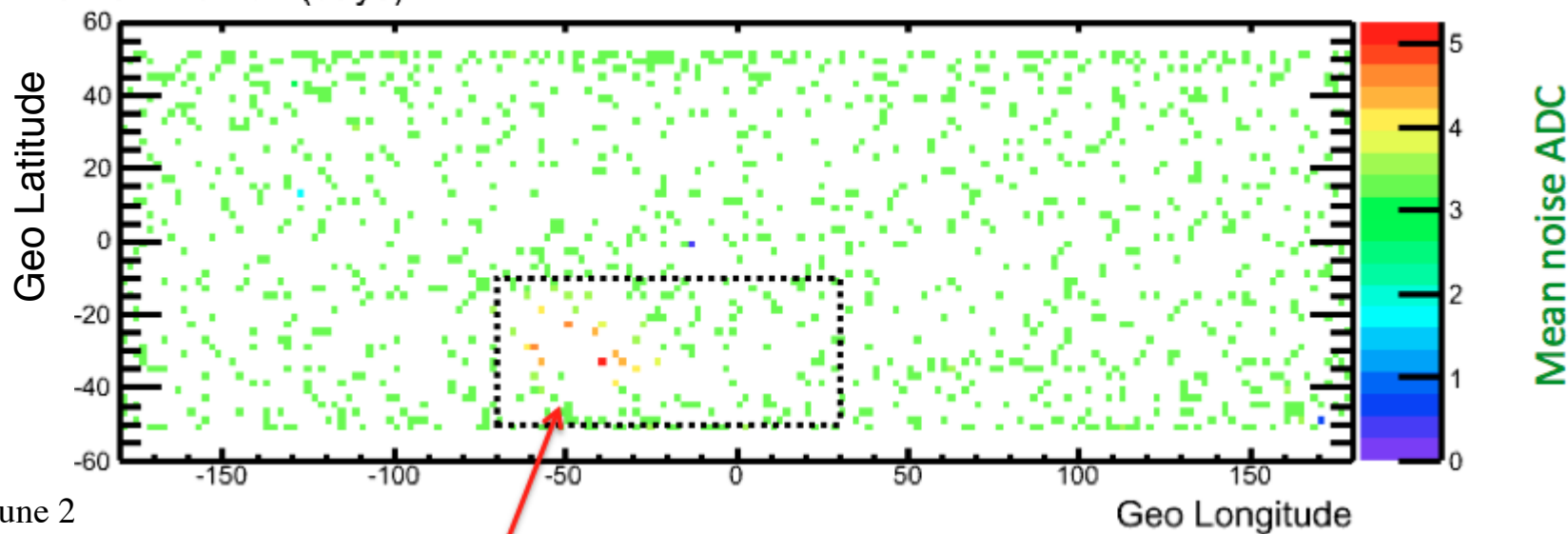
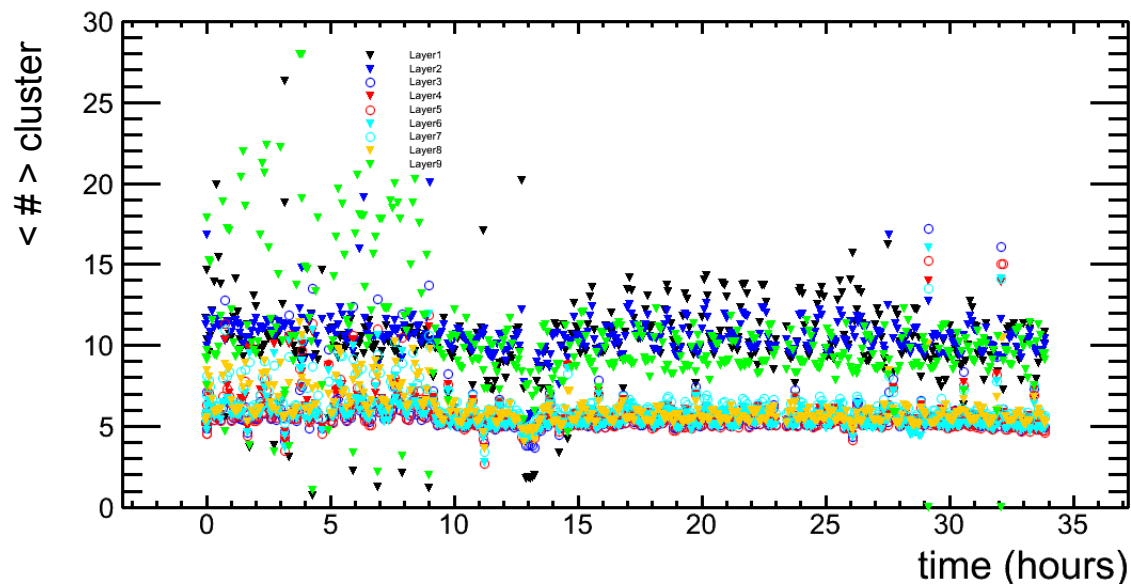
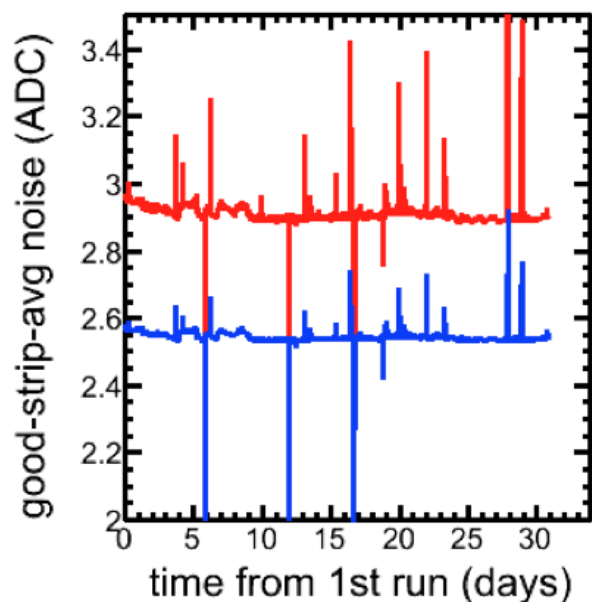


the particles we see



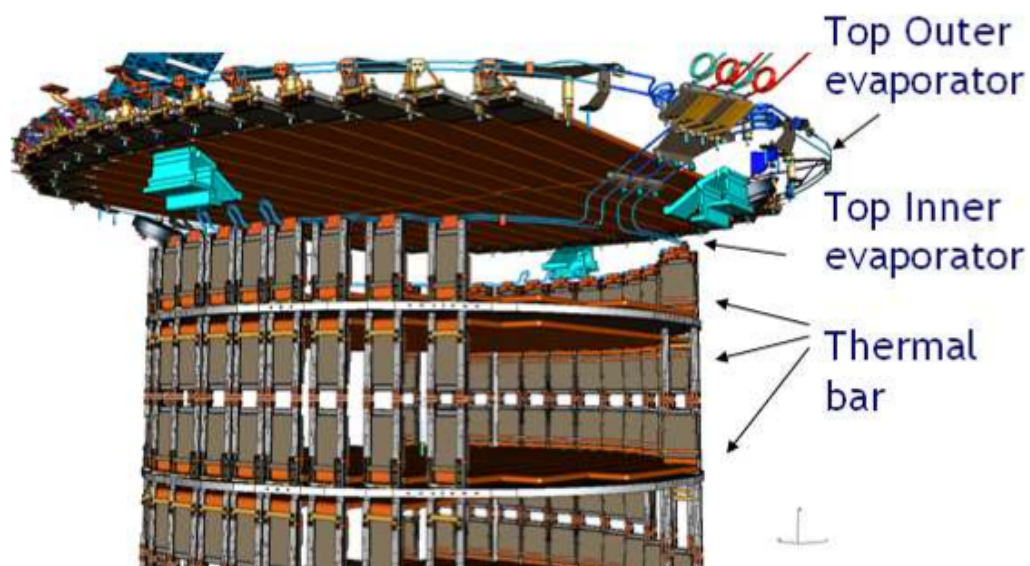
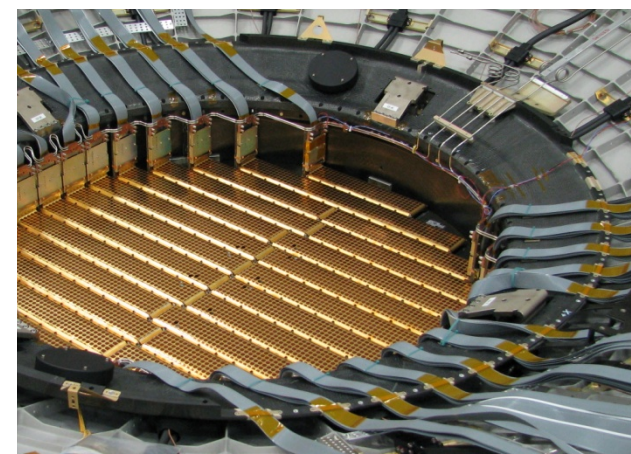
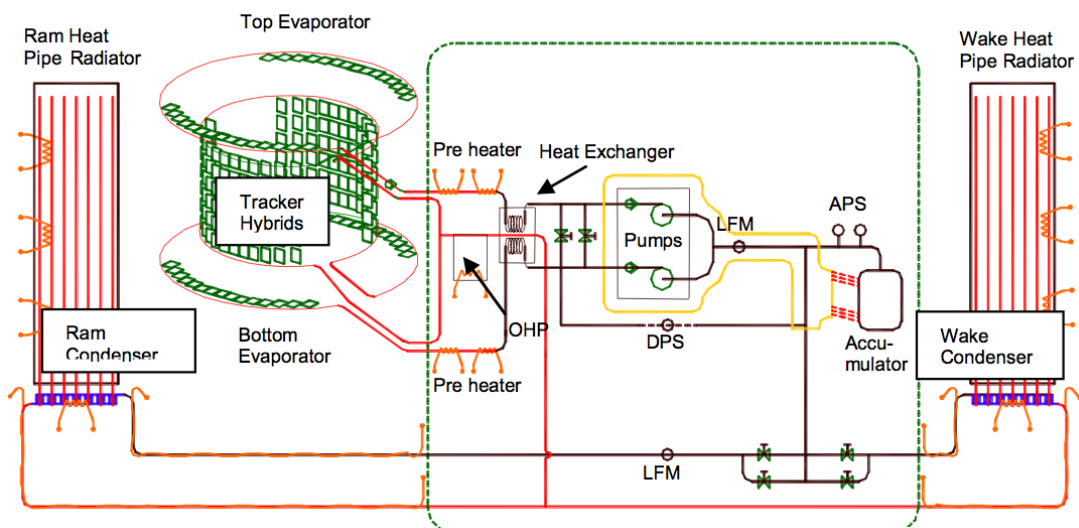


in flight experience





cooling: 2 phases CO2 pumped loop



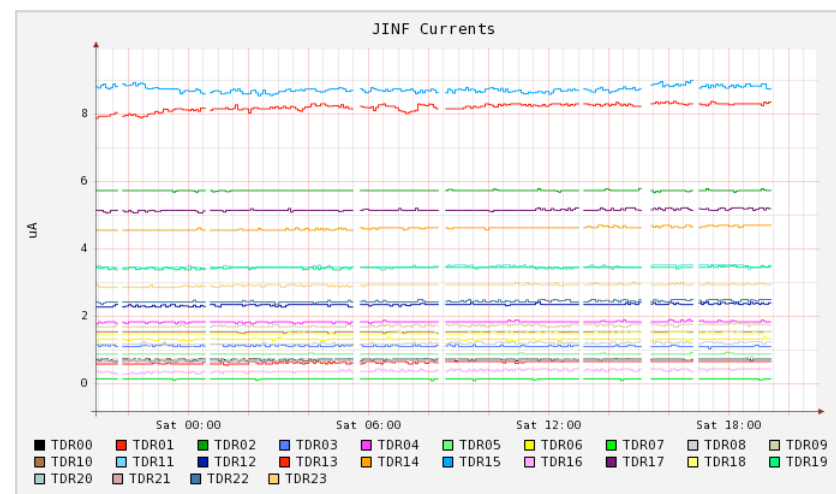
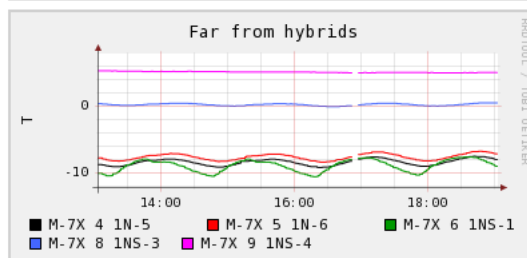
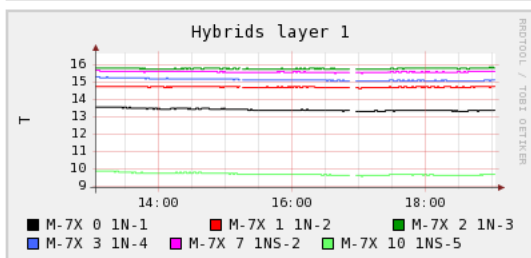
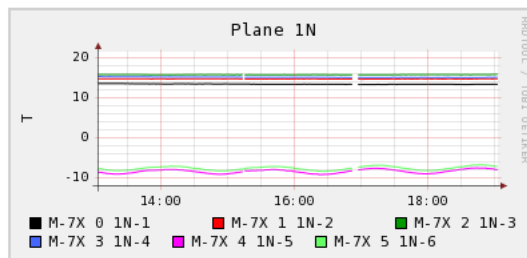
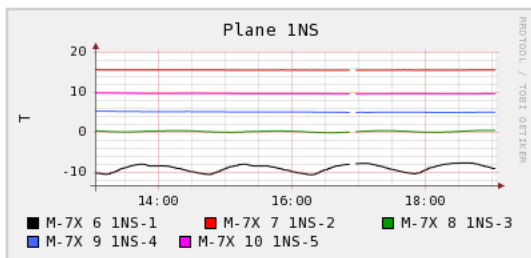
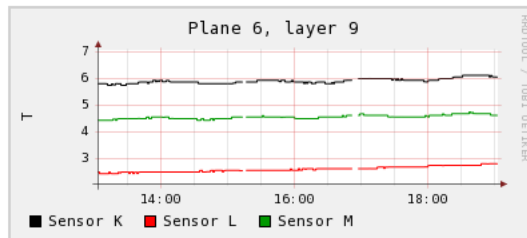
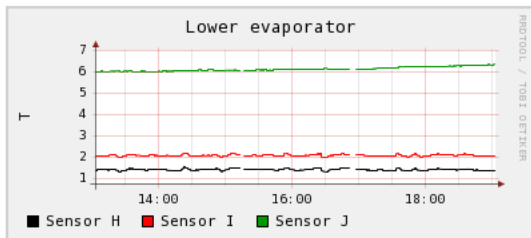
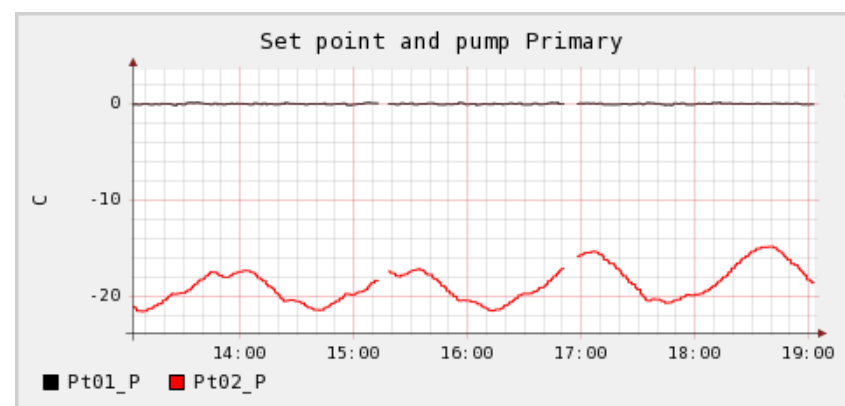
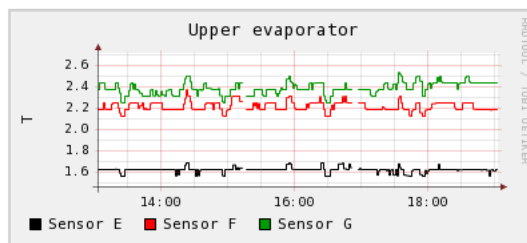
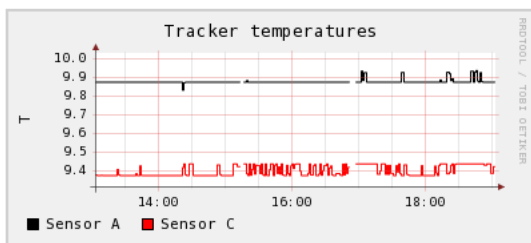


in flight experience: cooling and currents



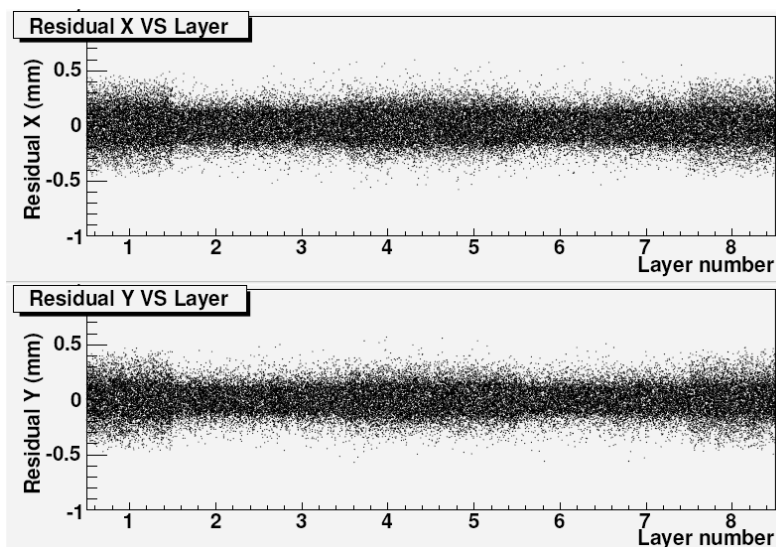
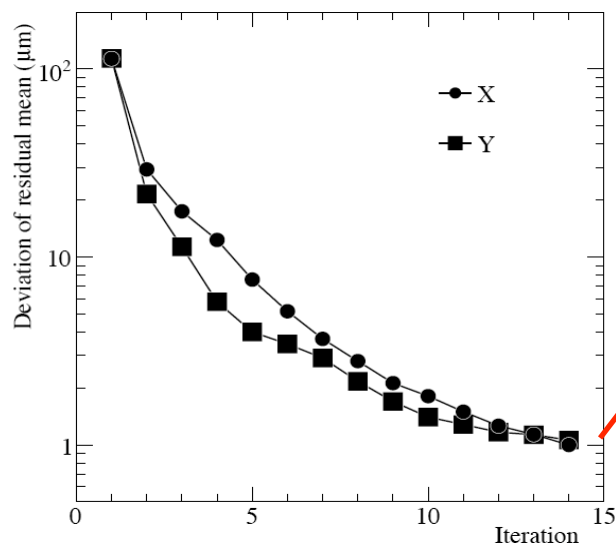
Showing last **hour 3_hours 6_hours day 3_days week**

Until **now** or

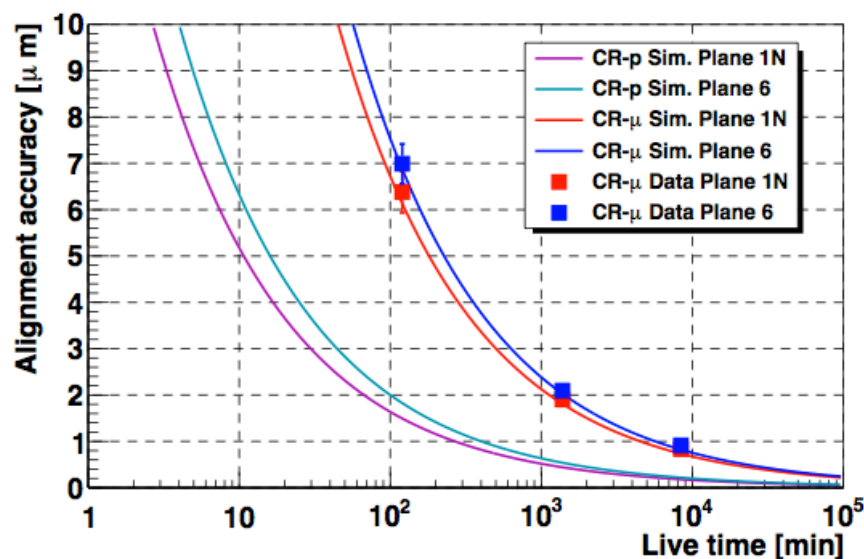




experience to come: alignment



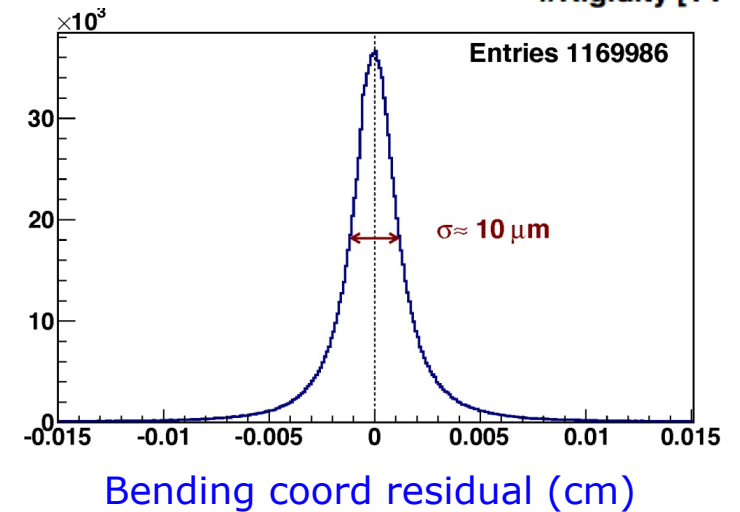
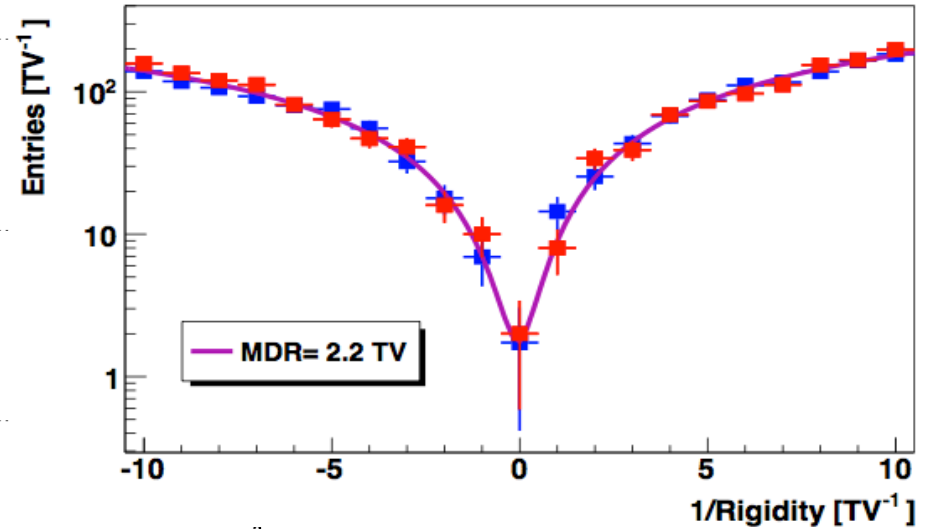
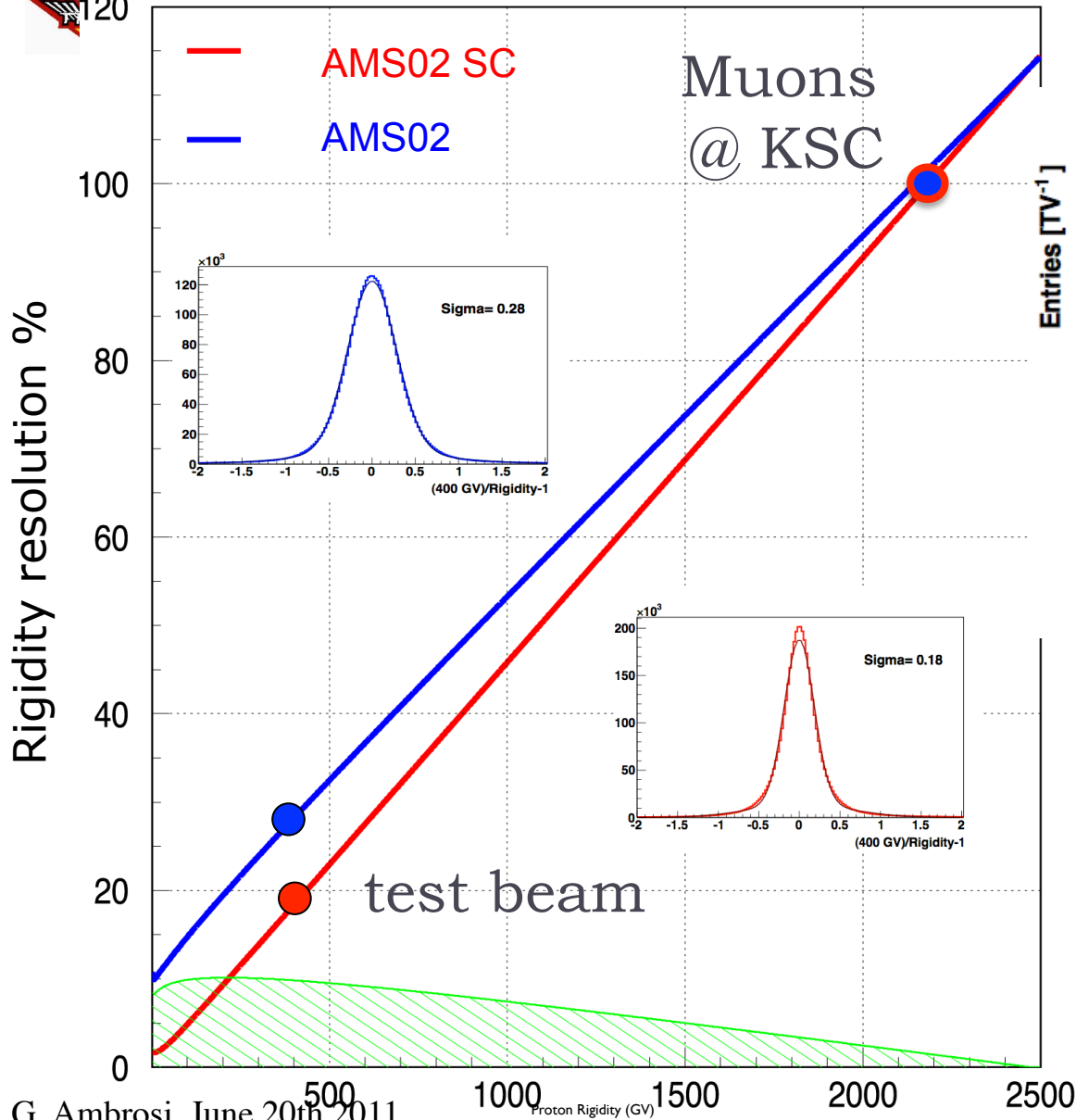
on ground results
inner planes



external planes
expected results



the performance we do expect





Conclusions

- AMS02 is in orbit since May 16th 2011
- No damage due to the launch stress or to the space environment, all the system are working in both the primary and redundant part
- All the detectors are properly functioning with DAQ in nominal conditions since May 19th 2011 (1.3 billions events)
- Tracker behavior is as expected in term of signal and noise levels
- 10+ years on board the ISS: great discovery potential, lot of work ongoing (alignment!)



G. Ambrosi, June 20th 2011

Science will come soon!

37