# SiPM-based CubeSat for Measuring the Cosmic X-ray Background

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1 Introduction 2 Synthesis model of the CXB 3 CXB measuring approaches 4 EQUATOR CXB detector 5 Summary and outlook

## 1 Introduction

#### Rocket experiment: before 1962

- X-ray emission from the Sun
- o rocket flight beyond atmospheric barrier (~100 km)
- O Geiger (gas) counters
- one can see Sun X-rays modulation
- no filtering of background
- not able to find other sources

#### Rocket experiment: since 1962

- O Geiger counter + active anticoincidence shield
- was able to recognize discrete sources
- O the first extrasolar X-ray source Sco X-1
- the CXB was revealed at the same time

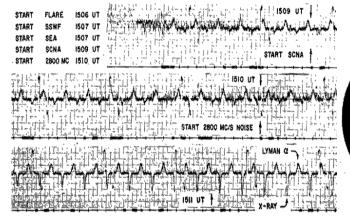
#### ROSAT moon observation

- the bright side: solar X-ray reflection;
- the dark side: a distinct shadow of CXB.
- suggested an extrasolar origin

### CXB Intensity over the sky

- o measured by multi missions: *Uhuru*, *HEAO-1*, *ROSAT*, etc
- nearly isotropy over the sky
- suggested an extragalactic origin

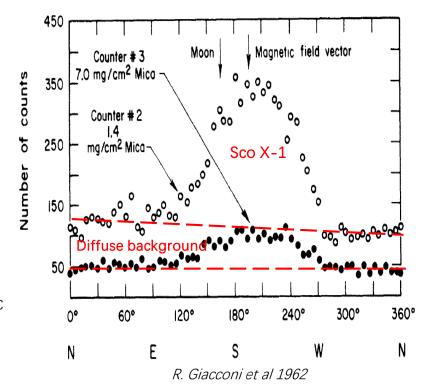


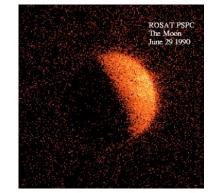


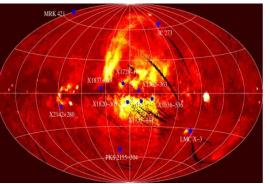


V-2 Rocket, NRL @ NASA

1963IAUS...16...45F



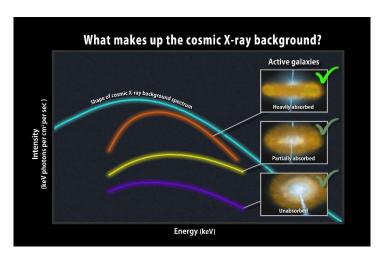


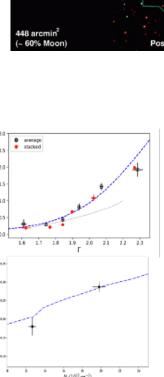


Copyright @ ROSAT

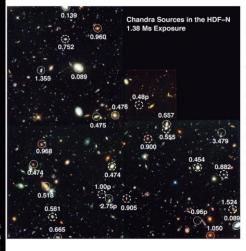
## 2 Synthesis model of the CXB

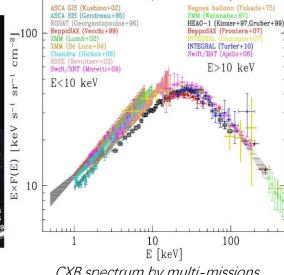
- Deep-field observations:
- Chandra, XMM-Newton, etc.
- resolve the CXB into discrete sources
- below 2 keV, 80-90% are already resolved
- resolved fraction decreases with increasing energy
- Active galactic nucleus (AGN) population
- unobscured AGN ( $N_H$ ~0), most observed
- mildly obscured AGN
- Compton-thick AGN ( $N_H > 10^{25}$ ), later proposed
- Synthesis model of the CXB
- superposition of AGN population
- Challenge of the modelling: poor knowledge of obscured AGN
- measured CXB spectrum uncertainty: ~ 20%.





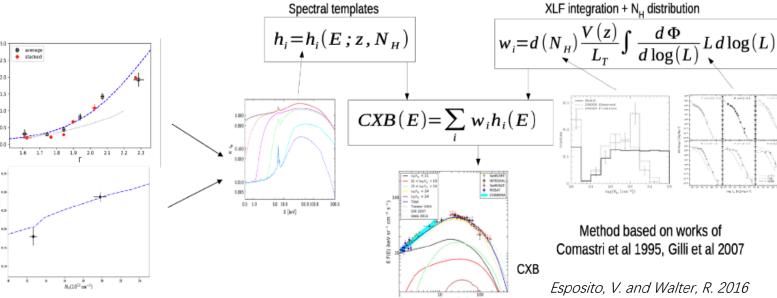
1.95 Ms over 2.3 yr





CXB spectrum by multi-missions

Chandra deep-field observations



Credit: NASA

## 3 CXB measuring approaches

#### General idea:

- o expose the sky regions (preferably blank sky),
- filter out relevant contamination:
  non-X-ray background and known discrete sources;
- O distinguish all components registered in the detector.

#### Approach 1: ASCA/SIS, Beppo-SAX and RXTE/PCA

- deep exposure on high latitude blank sky regions to register the CXB flux;
- same level of exposure on the dark side of the Earth employed as an estimation of the instrumental background.

#### Approach 2: HEAO-1

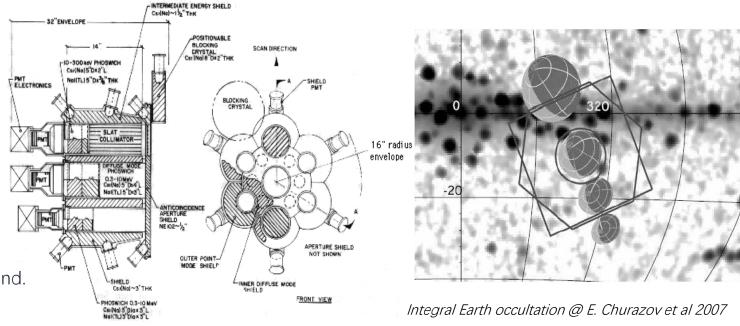
- o active collimator: Csl active anticoincidence shield;
- o movable shutter (Csl/Nal) to introduce modulation.

## Approach 3: INTEGRAL, Swift/BAT

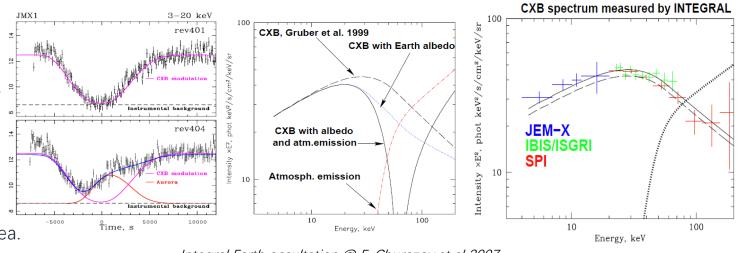
O Earth occultation to modulate all components.

#### Remaining difficulties:

- residual background;
- inaccurate energy calibration;
- o poor knowledge of detection efficiency and collecting area.



HEAO-1 A4 instrument layout @ HEASARC



Integral Earth occultation @ E. Churazov et al 2007

## 4 EQUATOR CXB detector

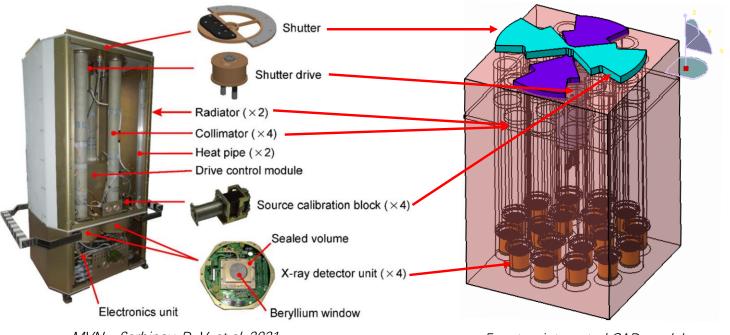
- Monitor Vsego Neba (MVN) proposed to the ISS
- o passive collimator / shutter : Sn-Cu-Al sandwich;
- 1 mm CdTe crystal (6-70 keV) + charge amplifier ASICs;
- 4 tubes, ARF~4.5 cm²/tube at 30 keV, FoV 8.55 deg².
- Equator (inspired by MVN)
- same idea of passive collimator / shutter;
- 2 cm CeBr<sub>3</sub> crystal (10-100 keV) + SiPM readout;
- 18 tubes, ARF ~4.5 cm²/tube at 30 keV, FoV 26 deg².

#### • Improvements:

- bigger collecting area, more active exposure;
- 2 counter-rotating shutters: compensate the torque;
  reveal the emissivity of shutter/collimator; and handle calibration sources;
- wider energy coverage, possible extending to 511 keV even to MeV.
- Calibration: sources attached beneath the top shutter
- $\circ$  AM-241+ tagged  $\beta$ <sup>+</sup> source -> energy, detection efficiency calibration.

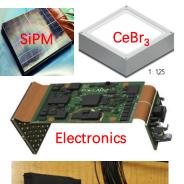
#### Resource requirements

- o dimension: 12-16 Units (CubeSat standard);
- o mass: ~16 kg; mission time: 2-5 years;
- o power: ~20 Watts; data volume: 2 GB / day;
- orbit preference: equatorial / low-inclination orbit;
- o readiness: 1-1.5 year-> flight model (ground calibration + space qualification).



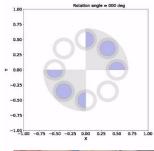
MVN: Serbinov, D. V. et al. 2021

Equator integrated CAD model









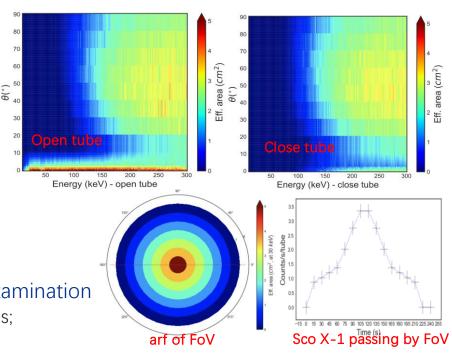


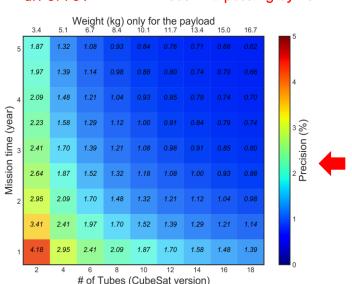
Spectrometer prototype Collimator tube prototype

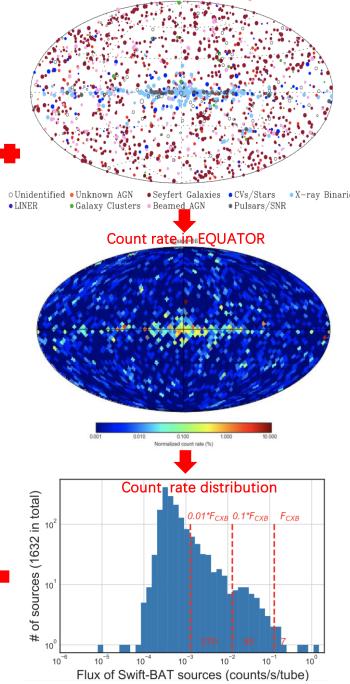
Wheel system prototype

## 4 EQUATOR CXB detector

- Effective area (arf)
- maximum arf ~5 cm<sup>2</sup>;
- collimator cut < 100 keV off-axis photons;
- shutter cut < 150 keV on-axis photons.
- How to measure/subtract Background (B)
  - close tubes monitoring B constantly;
- open tubes minus close tubes;
- o simulation: 0.2 cnts/cm<sup>2</sup>/s/keV/tube
- O How to measure/subtract Sources (S) contamination
- opassing by FoV leaves modulation of the sources;
- simulation: Swift-BAT 105-month catalogue;
  cut |Glat| < 10; cut SNR > 200.
- O How to measure/extract CXB (C)
- All registered counts (A)
- $\circ$  C  $\cong$  A-B-S (systematics can be well studied by tube array)
- O Simulation: 0.128 counts/s/tube
- Statistical/systematic precision (P):  $1\% \leftarrow \frac{\sqrt{C + B + C}}{C}$
- O Mission time (T), Number of tubes (N)
- Gradient: P~ a/sqrt(T) / P~ a/sqrt(N)







Swift-BAT catalog

## 4 EQUATOR CXB detector

#### Isotropic CXB measurement:

- reach 1% precision for 10-100 keV :
  resolving the CXB into AGNs + improve cross-calibration precision among instruments;
- observation at >100 keV, up to MeV:
  offering a sight on the CXB contribution of time-integrated Hawking radiation (HR) of primordial black holes (PBHs);
- o measuring 511 keV from the Galactic center/plane and its vicinity region : a wealth of information on the temperature, density, composition, and dynamics of the source region.

#### Anisotropic CXB measurement (dipole):

- $\circ$  Compton-Getting effect (dipole amplitude  $\Delta$ = 0.0042);
- $\circ$  remaining large-scale structure of the local Universe (0.0023  $\lesssim \Delta \lesssim 0.0085$ ).

#### Monitoring luminous X-ray sources:

- monitoring variability of sources -> to gain knowledge about fundamental astrophysics;
- o calibrate "standard candle" (like the Crab) for cross-calibration

#### Gamma-Ray Burst (GRB) detection

- Short GRBs have generally harder spectra and peak energies (~490 keV)
- o possibly associated with Gravitational Wave event (like Binary Neutron Star merger, 2017ApJ...848L..13A);
- o such GRBs could penetrate the platform structure/instrument housing, and reach the sensitive detector;
- Missions like Integral and Insight-HXMT have successfully employed this idea to monitor GRBs with nearly omnidirectional FoV

# Summary and outlook

Measuring CXB requires precise instrumental calibration and background modelling

© Science topics behind gain us knowledge of the accretion power in the Universe

Optical =

(+AGN?)

© EQUATOR will employ collimated spectrometer with moving shutter approach

○ A 12-16 U CubeSat mission with >2 year could reach 1% precision of the CXB measurement

2 keV

○ We need 1-1.5 year to be prepared for launching such a CubeSat mission