



# Space-Borne Accelerators

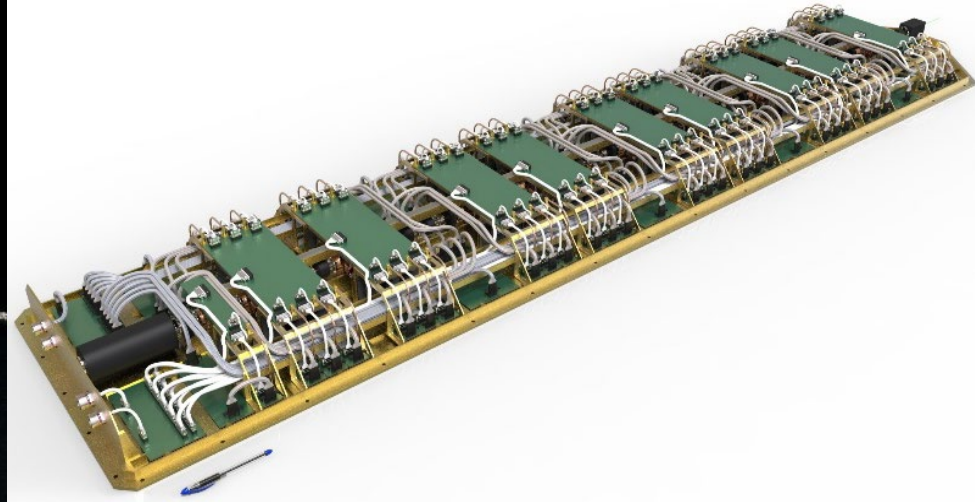
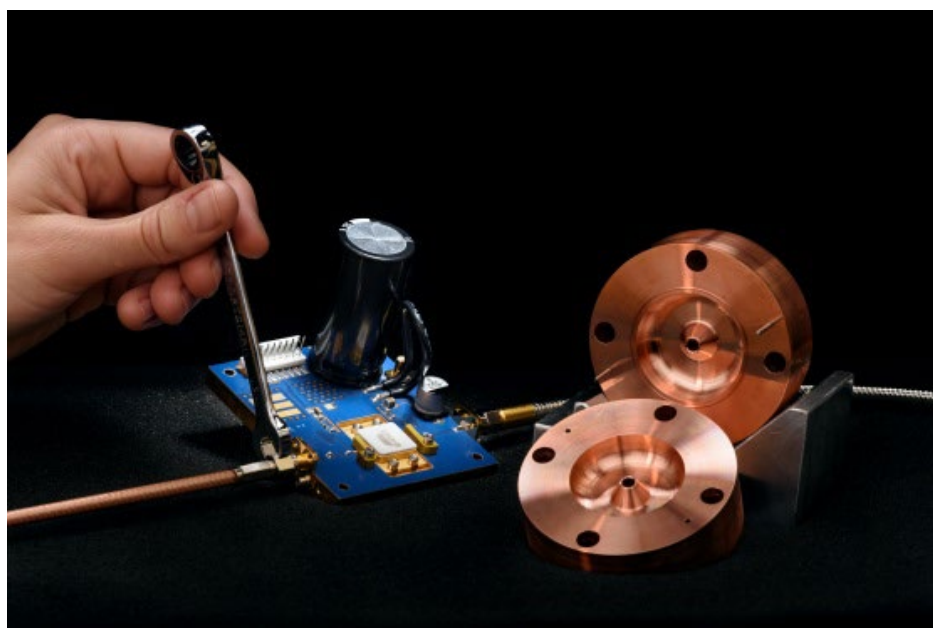
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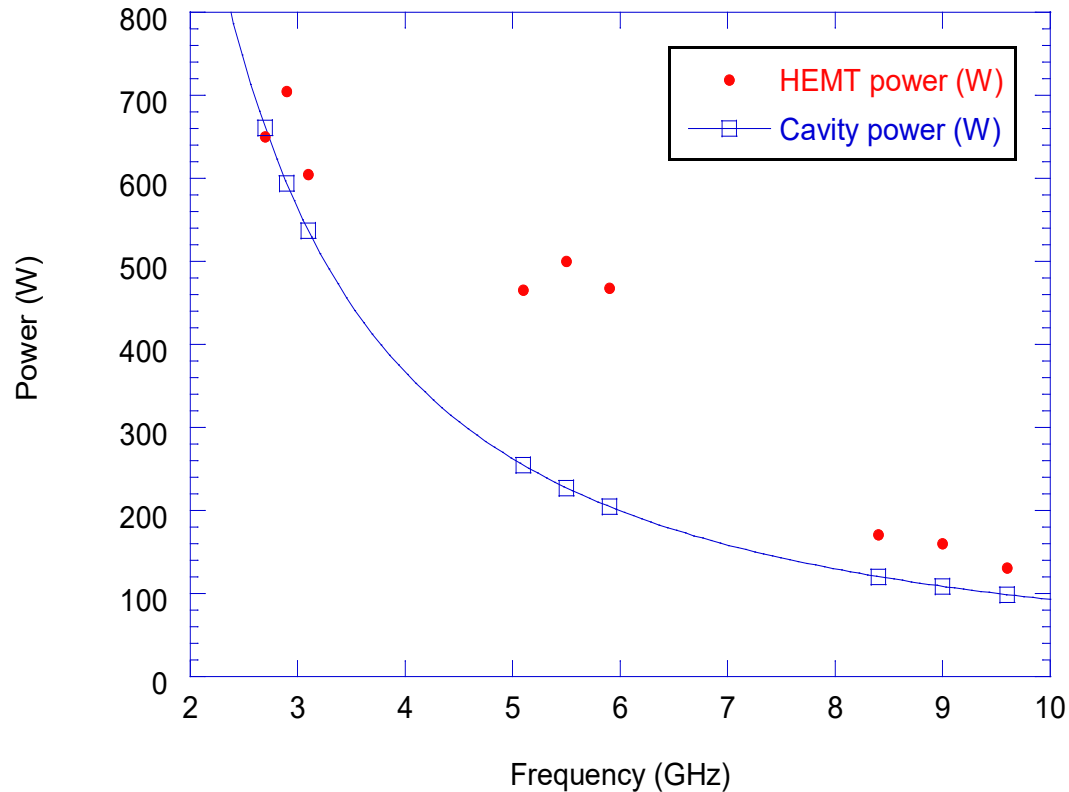
# Compact Accelerators in Space Using HEMTs

# HEMT RF Sources for Compact Acceleration



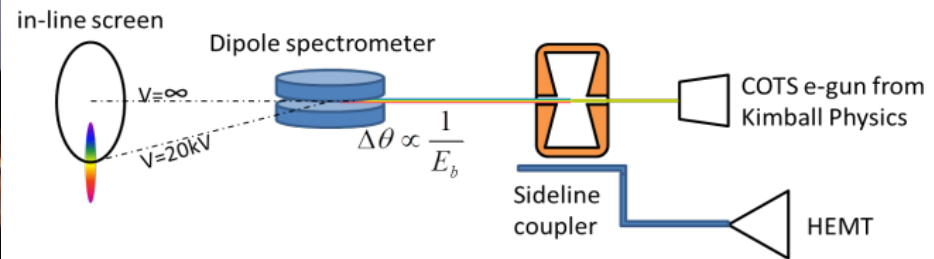
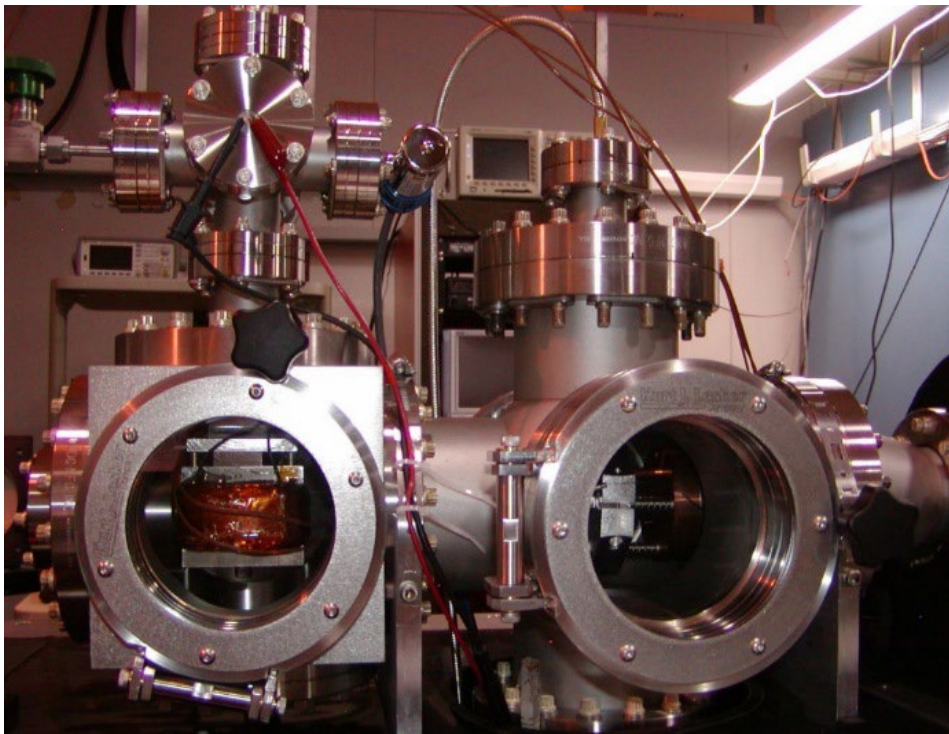
- High Electron Mobility Transistors (HEMTs) used to drive small C-band (5 GHz) cavities.
- The HEMTs are compact, solid state devices that can generate up to 500 W of RF power.
- These allow for compact acceleration of electrons without relying on bulky devices such as Klystrons.

# Frequency Chosen to Optimize Available Power



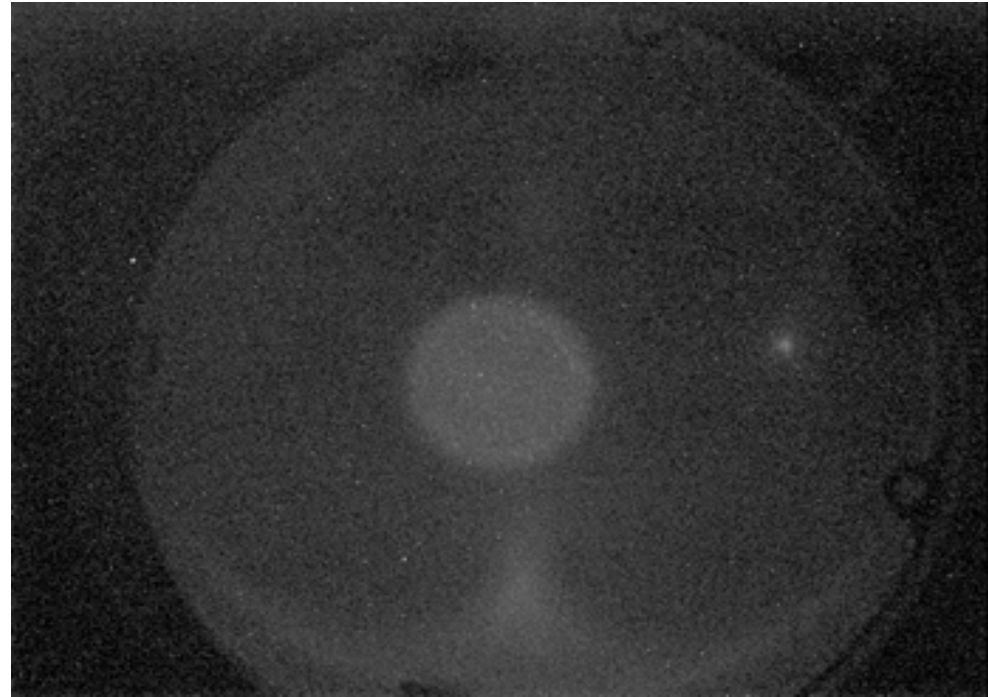
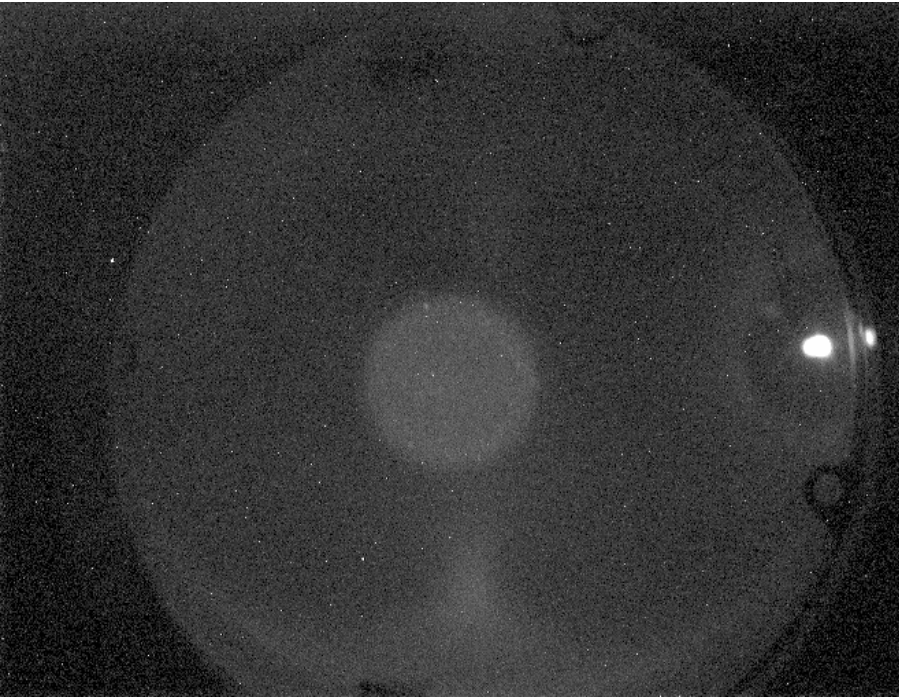
- At low frequencies, HEMTs produce high power, but cavity resistive loss is high so there is not much power left.
- At high frequencies, cavity loss is low but HEMT power is also low.
- Around 5 GHz, HEMT power is higher than cavity loss, leaving a large amount of power to accelerate beam.

# Experimental Demonstration of HEMT Acceleration



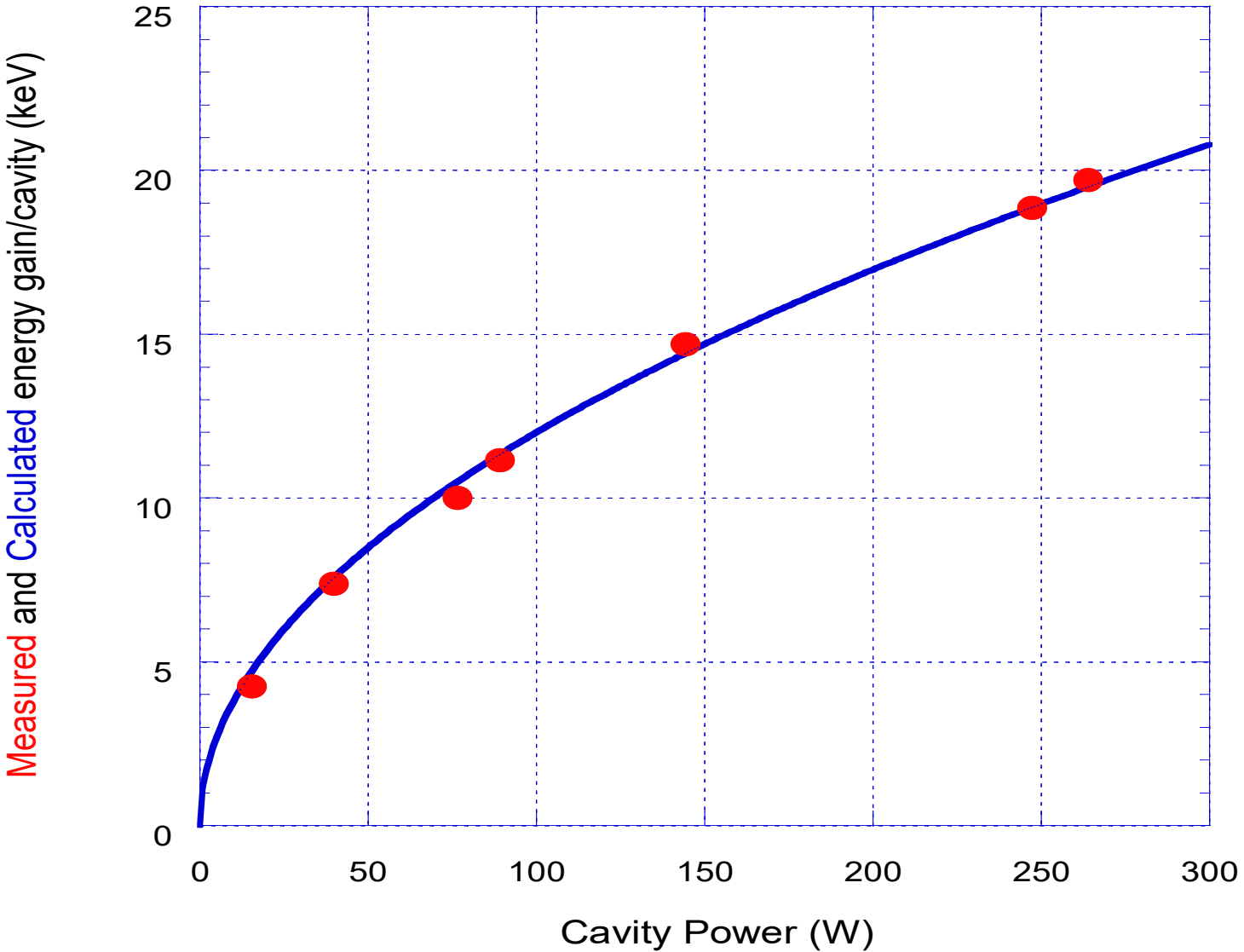
- Use a dipole spectrometer to measure energy gain from a single cavity driven by a HEMT.

# Experimental Demonstration of HEMT Acceleration (2)



- Left is without cavity RF power, right is with. Change in location can be used to measure energy gain.

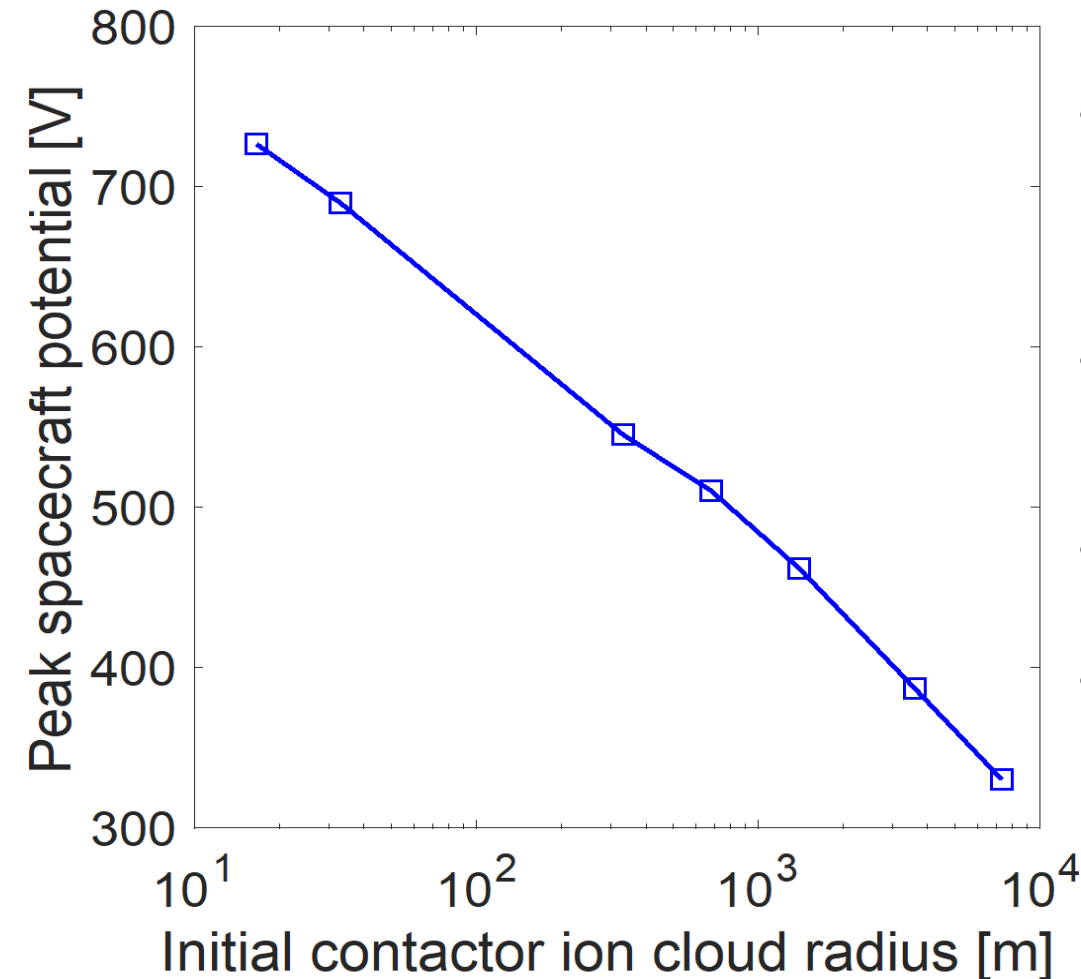
# Experimental Demonstration of HEMT Acceleration (3)



# Plasma Contactor

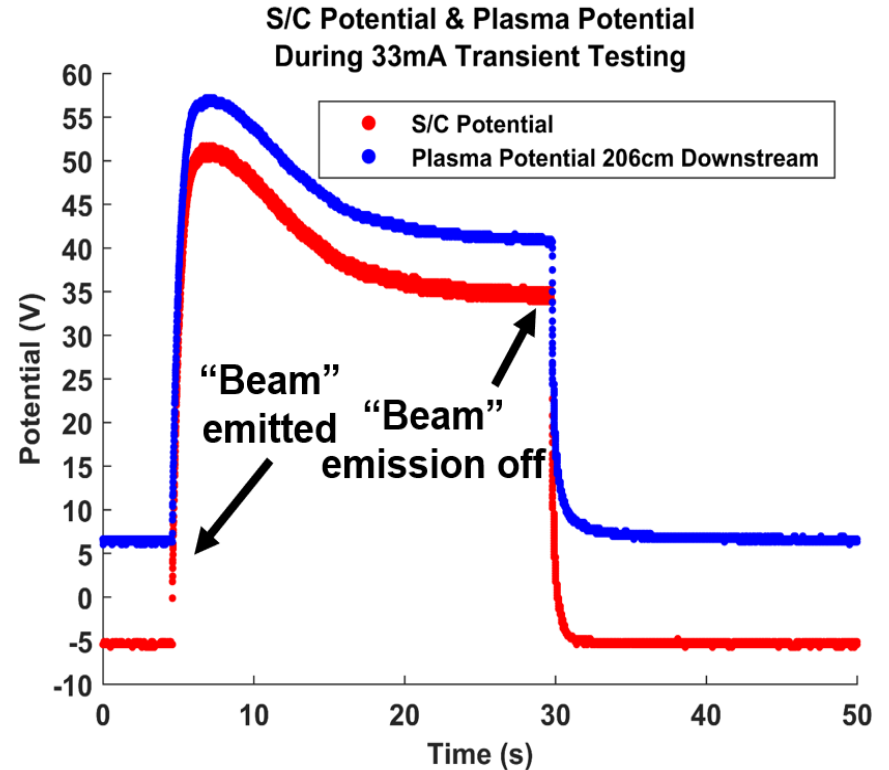
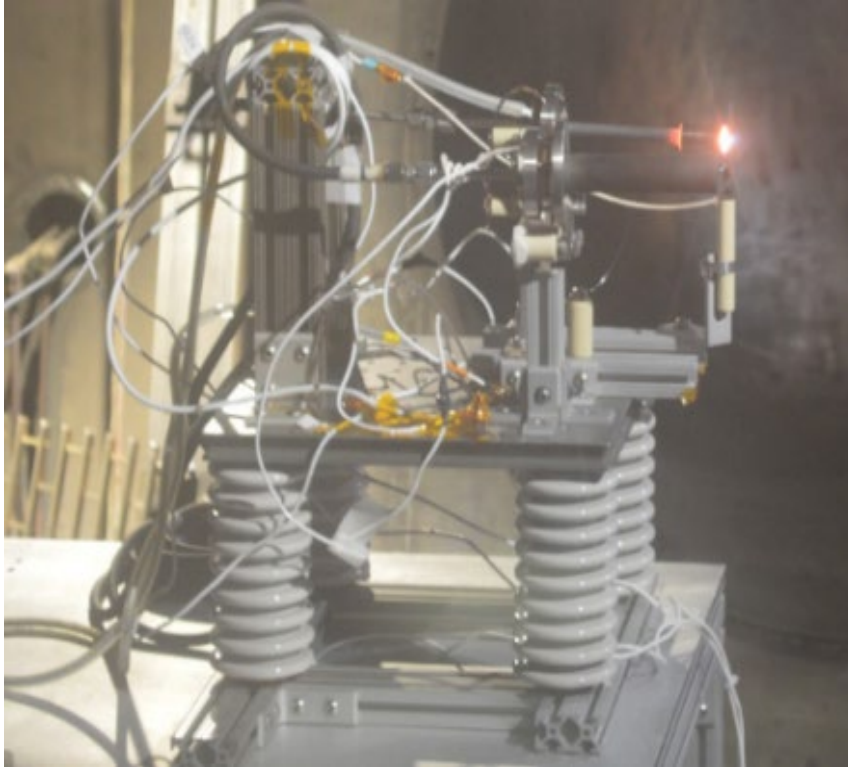


# Plasma Contactor to Mitigate Spacecraft Charging



- In the magnetosphere, the plasma is too low density to recharge spacecraft after electrons (or ions) are emitted away.
- A 1 mA beam could charge up spacecraft to 100 kV, which could damage the spacecraft.
- Use a plasma contactor to eliminate this problem.
- Simulations show this can reduce spacecraft charging to below 1 kV.

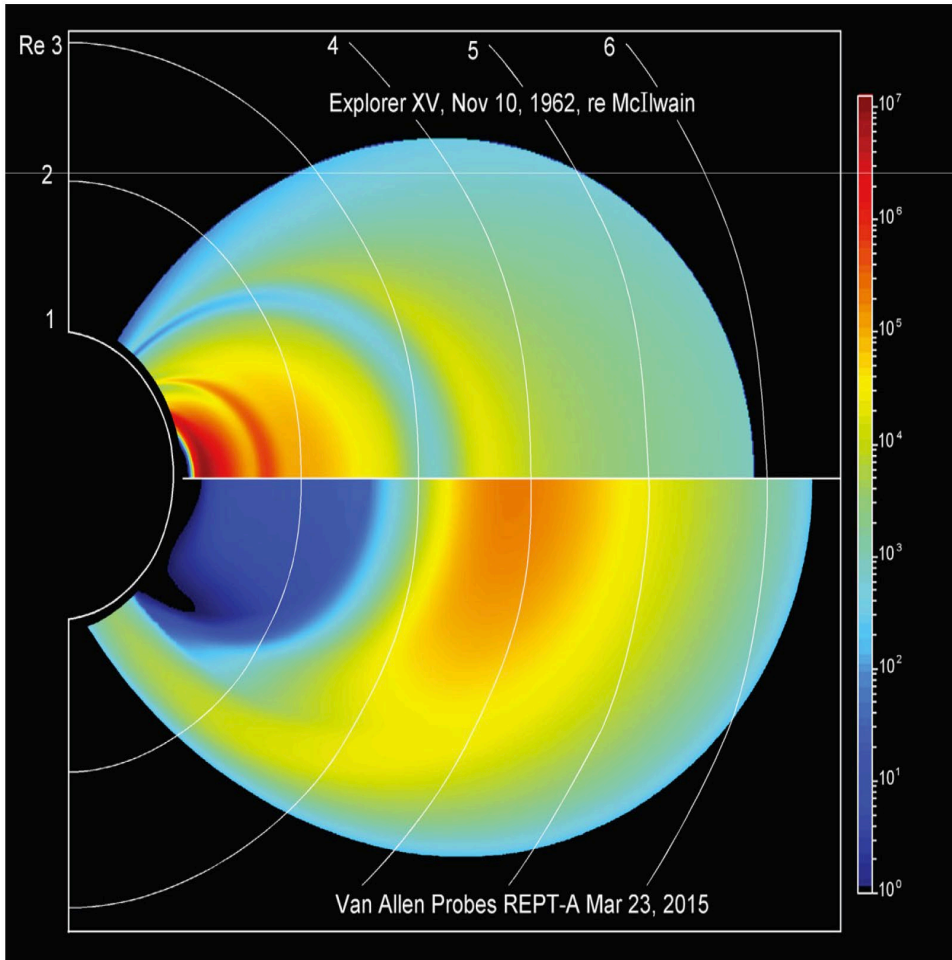
# Validation of Spacecraft Charging Model



- Experiments at U. Mich's Large Vacuum Test Facility validate models of spacecraft charging.

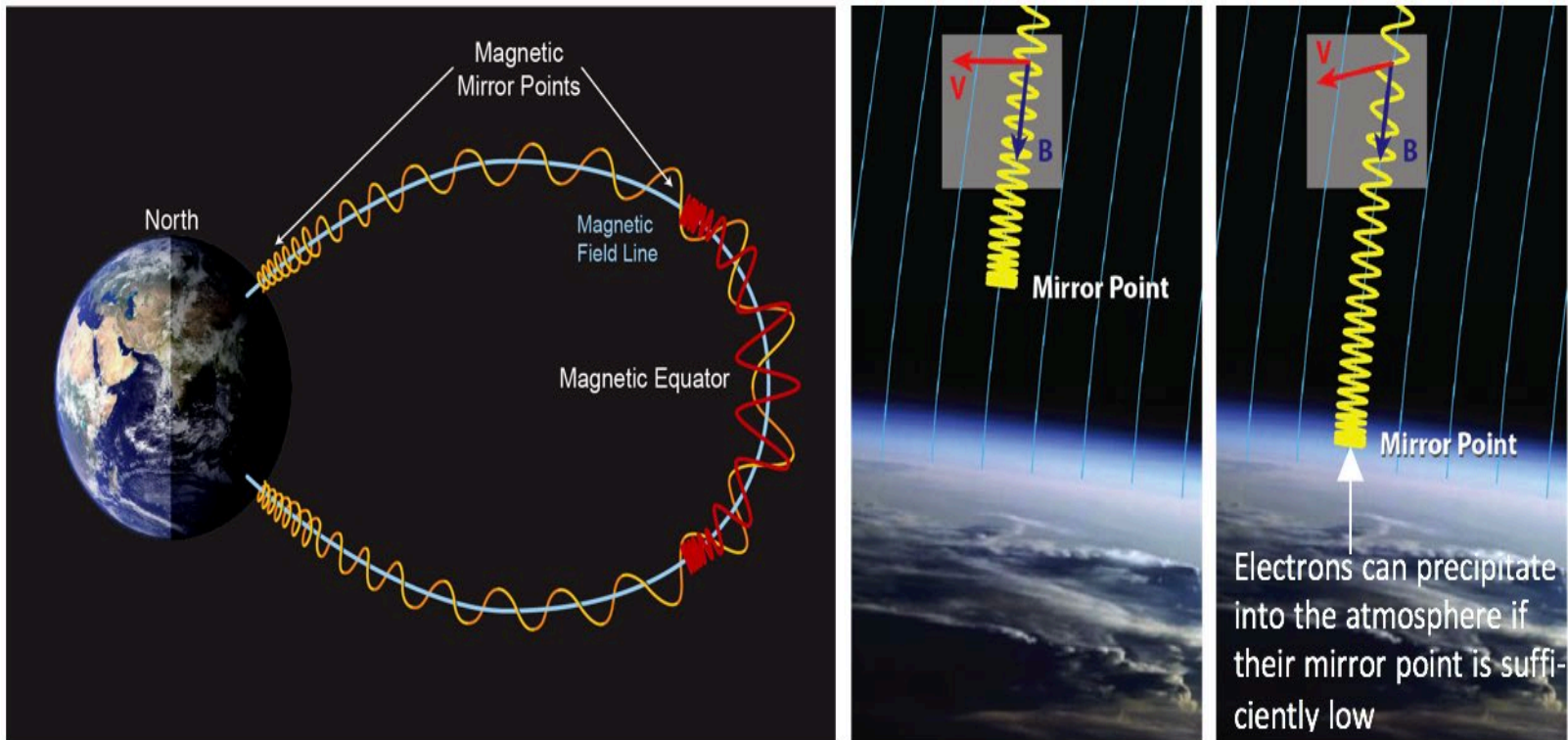
# Radiation Belt Remediation

# A HANE could destroy most LEO satellites



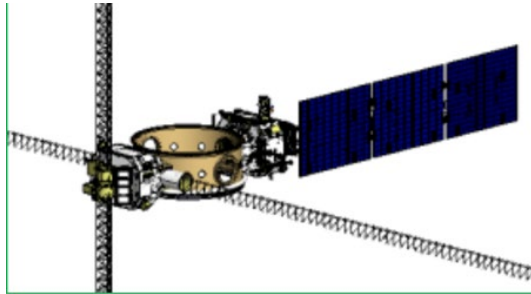
- In 1962, the Starfish Prime high altitude nuclear test produced an artificial HANE belt of high energy electrons.
- These “killer electrons” destroyed a third of all LEO satellites deployed at the time.
- In modern times, we have many more LEO satellites, which would be threatened by a rogue nation performing a HANE test.
- A method is needed to quickly reduce the number of energetic electrons trapped in the artificial belt.

# Use Artificial Waves to Precipitate Electrons into the Atmosphere

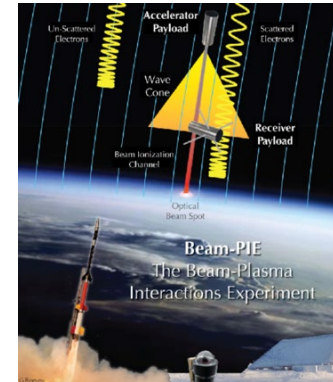


- Directly inject whistler waves into HANE belt using either an antenna or an electron beam.
- These waves cause diffusion in the trapped electrons' trajectories, which causes them to align with the loss cone and travel into the atmosphere.

# Exploring 3 different methods of wave generation



Shown with optional tripod mount



## Dipole Antenna

- Most studied, both theoretically and experimentally
- Likely not very efficient at generating whistler waves.

## Loop Antenna

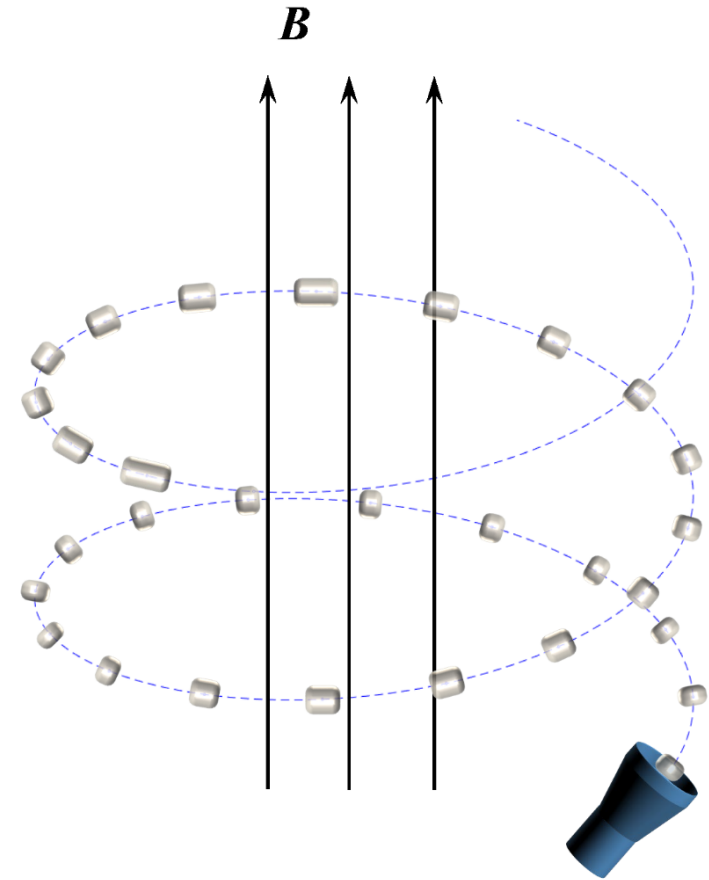
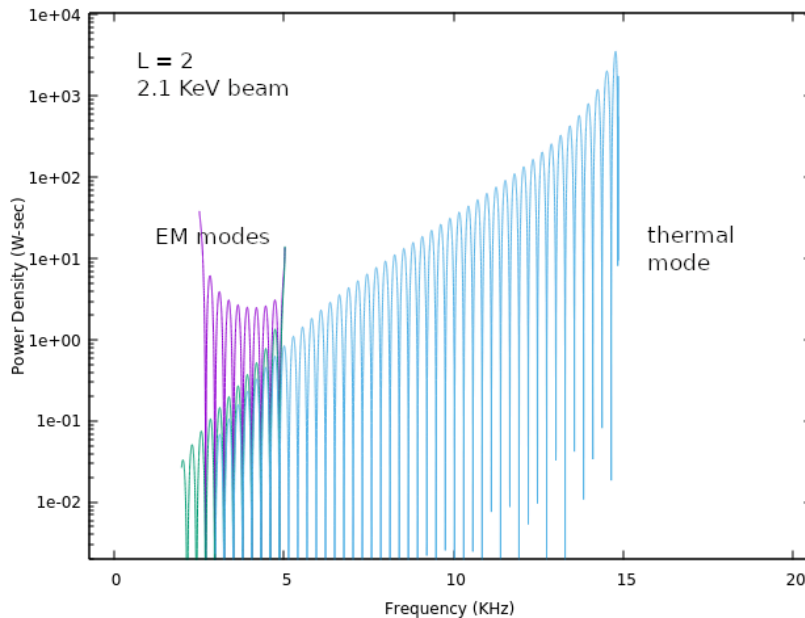
- Not very well studied – some theory and lab experiments, no space experiment.
- Much more efficient than dipole.

## Modulated electron Beam

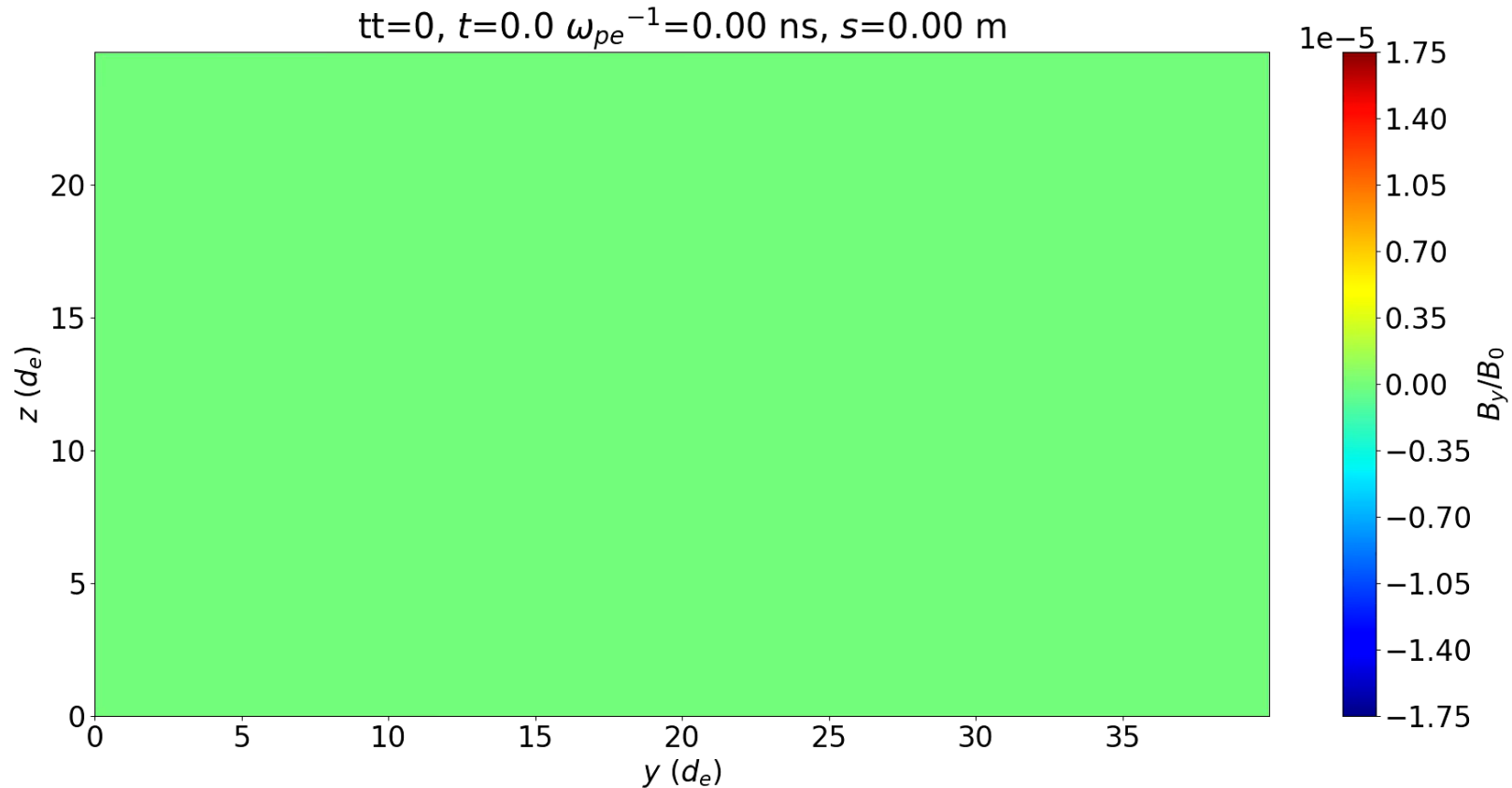
- Planned BeamPIE experiment 1<sup>st</sup> for wave gen.
- LANL studies show promise.

# Linear Theory Calculates Wave Generation for Beams

- Can model finite beam length and beams that are not aligned with B-field.
- Find lots of power in complicated thermal mode, showing need for SPS simulations.



# LANL (T-5) developed code SPS used to model an ideal electron pulse

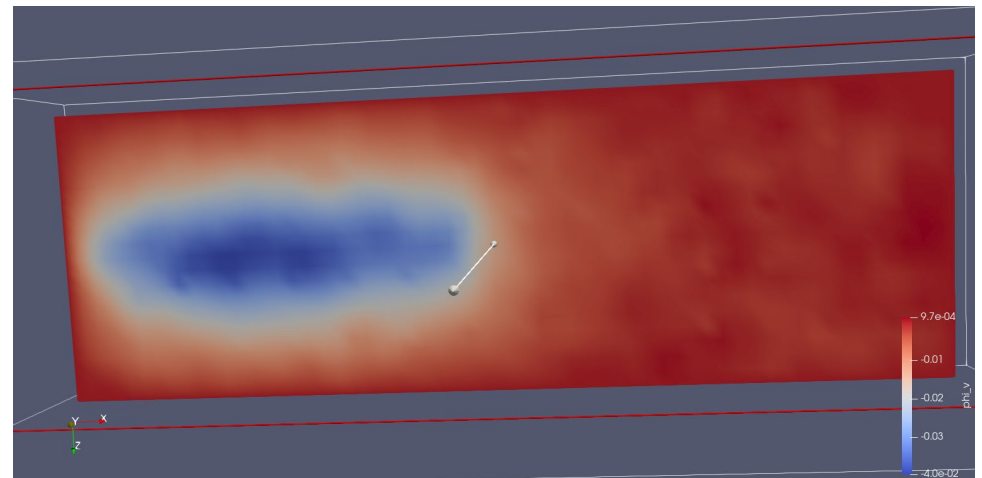
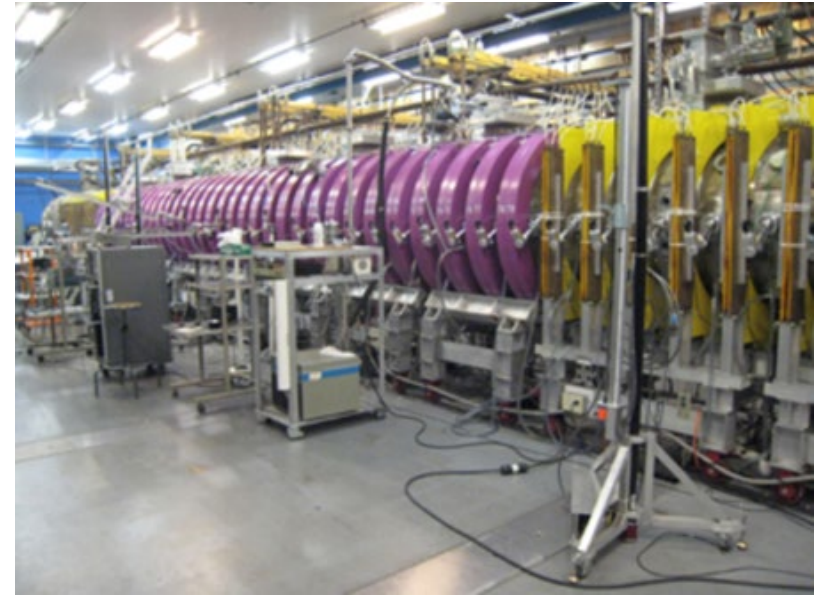


- SPS used to model waves generated from ideal electron beam.
- Currently comparing these results to linear theory predictions.



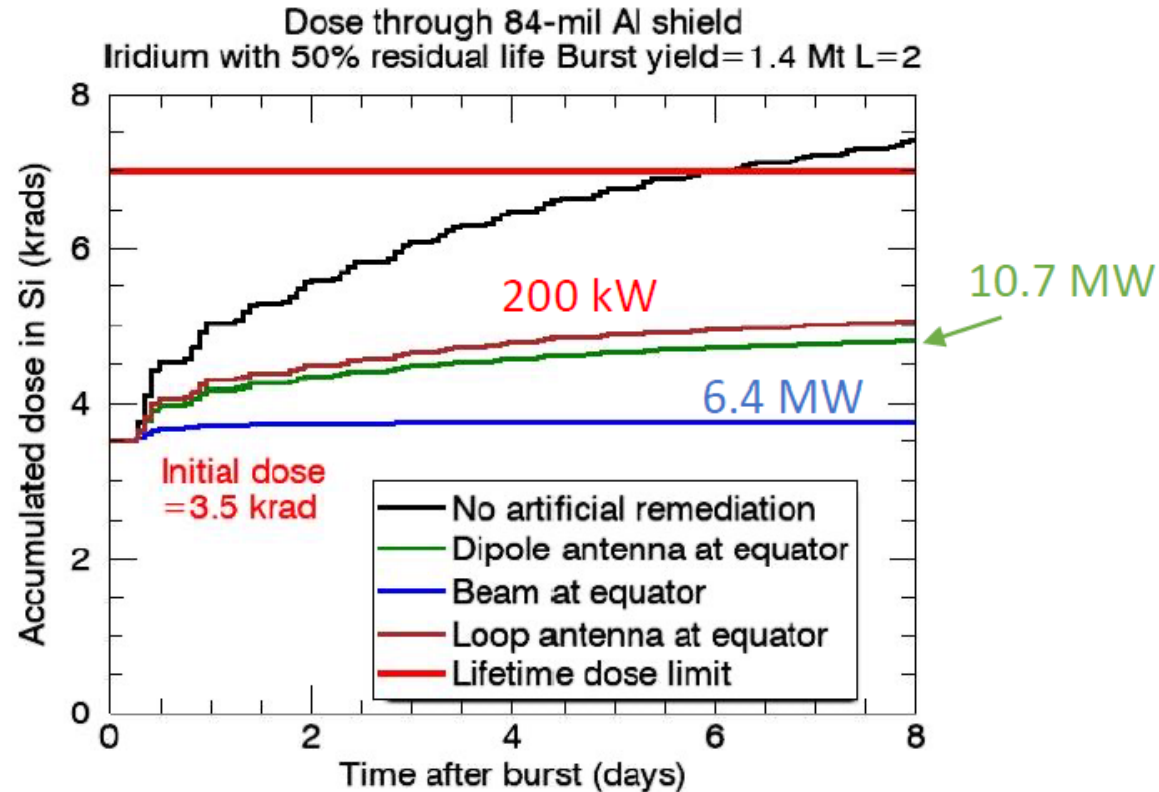
# Lab validation experiment at LAPD

- A validation experiment is taking place at the Large Area Plasma Device (LAPD) at UCLA (Gekelman 2016).
- For this experiment, a 20 keV electron beam was injected into the plasma, and the resulting plasma waves were measured.
- These results are being compared to analytical theory and first principles modeling based on the CPIC code.



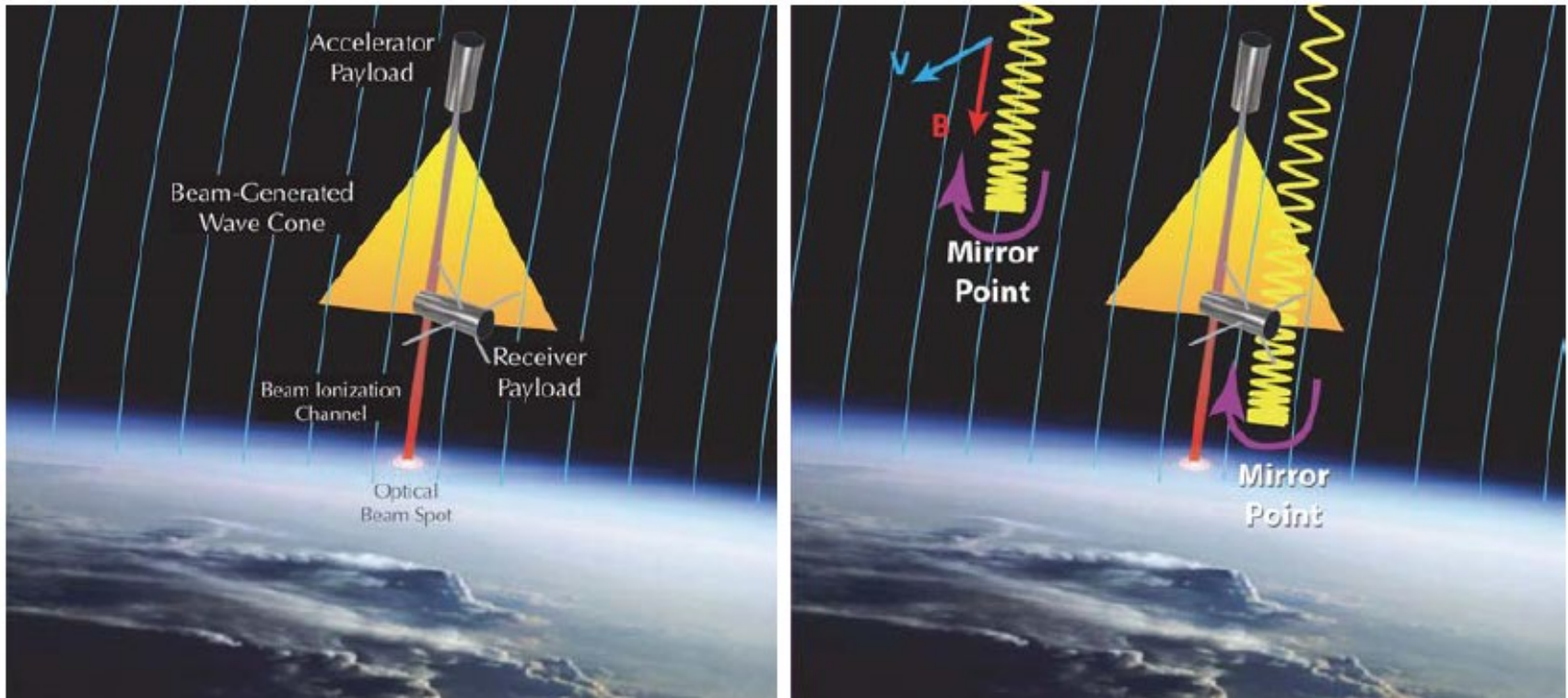
# Our model suggests RBR concept is feasible

- We have done S2E models, including the generation and propagation of waves, as well as the interaction of the waves with MeV electrons.
- Our models suggest that an electron beam can remediate the belt fast enough to save most satellites.
- Have compared the beam with a dipole and loop antenna.



# The Beam Plasma Interactions Experiment: BEAM PIE

# BEAM-PIE will test the models of radiation from a beam



- BEAM-PIE will field a 6 mA, 10-50 keV electron beam aboard a sounding rocket.
- Uses a HEMT cavities to accelerate electron beam.

# BEAM-PIE Diagnostics

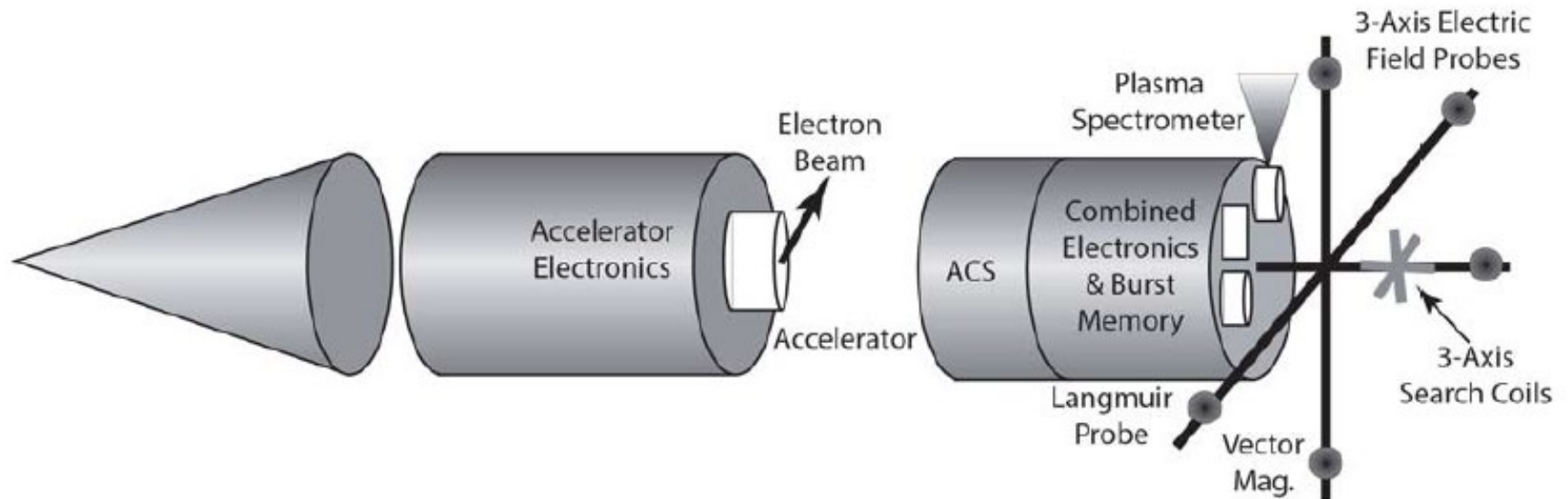
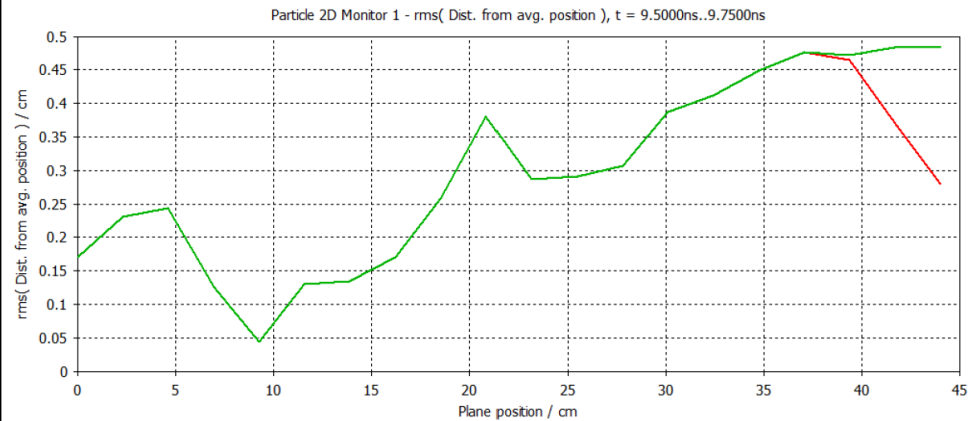
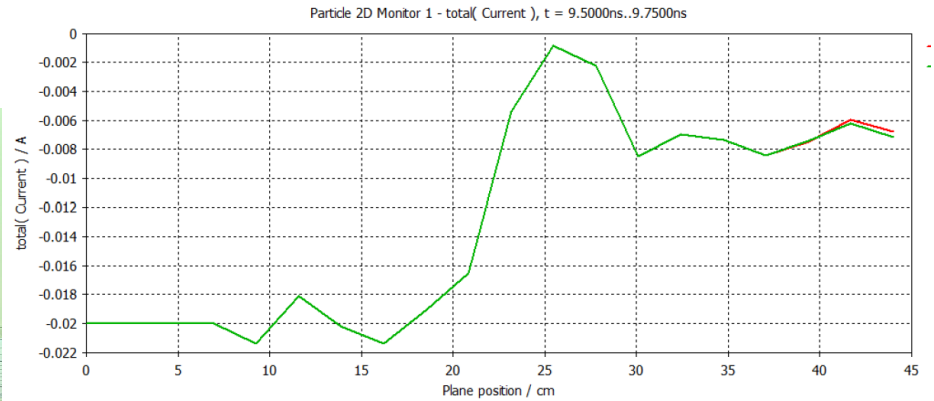


Figure 9. Schematic of the Beam-PIE payload and instrument locations.

- “Mother-Daughter” rocket configuration allows a diagnostic section to split off from electron beam.
- Measurement of plasma waves will be used to validate the models.

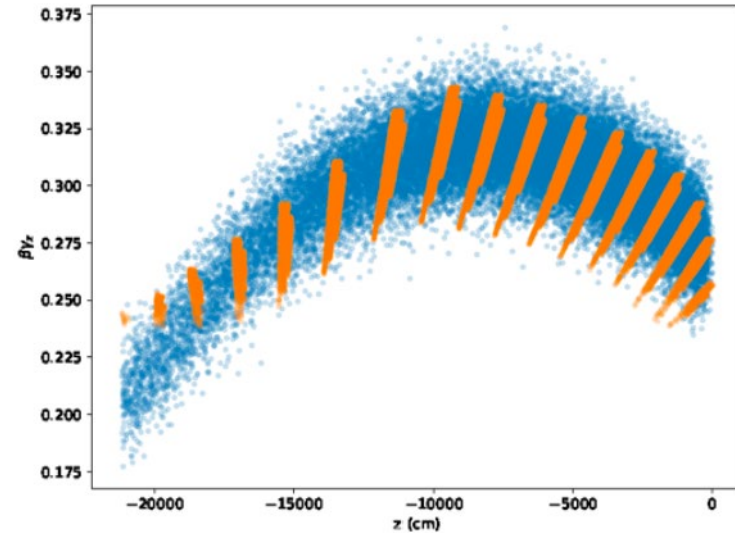
# BEAM-PIE accelerator designed, being built in lab



- The BEAM-PIE accelerator has been designed and modeled using microwave studio.
- Being assembled in lab to verify that it works, then will be put in rocket body.

# Model propagation of BEAM-PIE beam into space

- Take outputs of many microwave studio simulations to make one distribution of a single BEAM-PIE bunch.
- Use a new microwave studio simulation to model the beam propagating into space.
- Beam debunches more quickly than ideal beam because of energy spread.



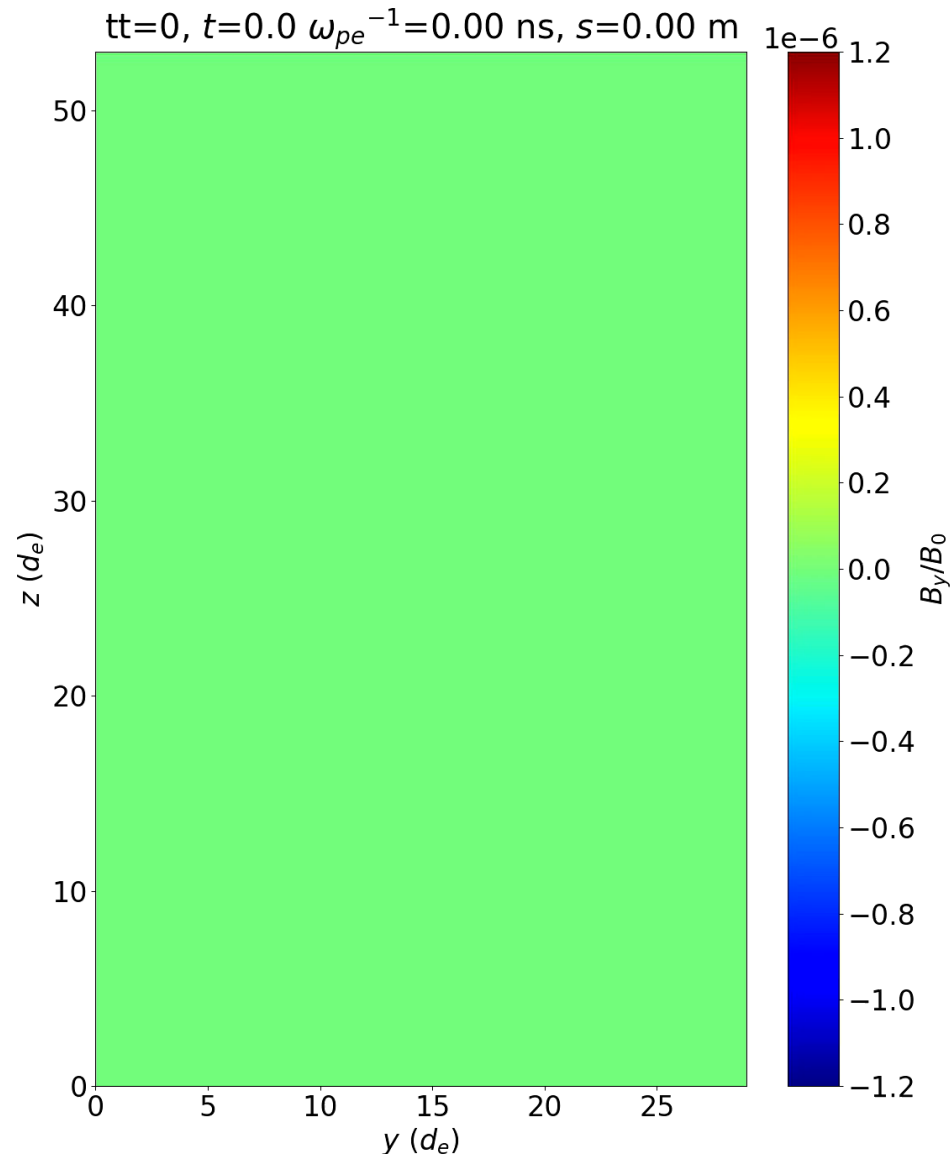
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SIMULIA CST Studio Suite®



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Output Energy  
Sample 1/231  
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Particles 34815  
Maximum (Sample) 33750.4 eV  
Maximum (Global) 33794.7 eV

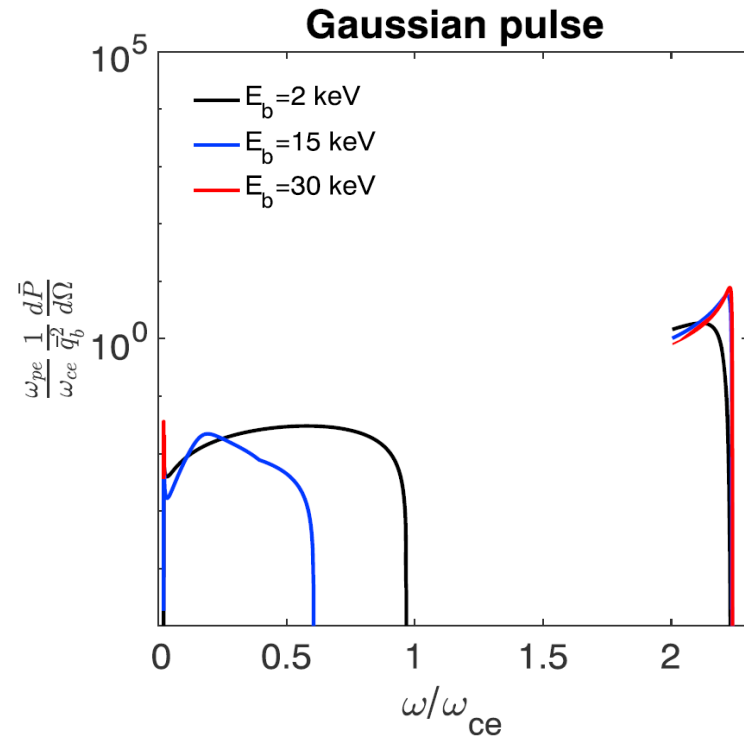
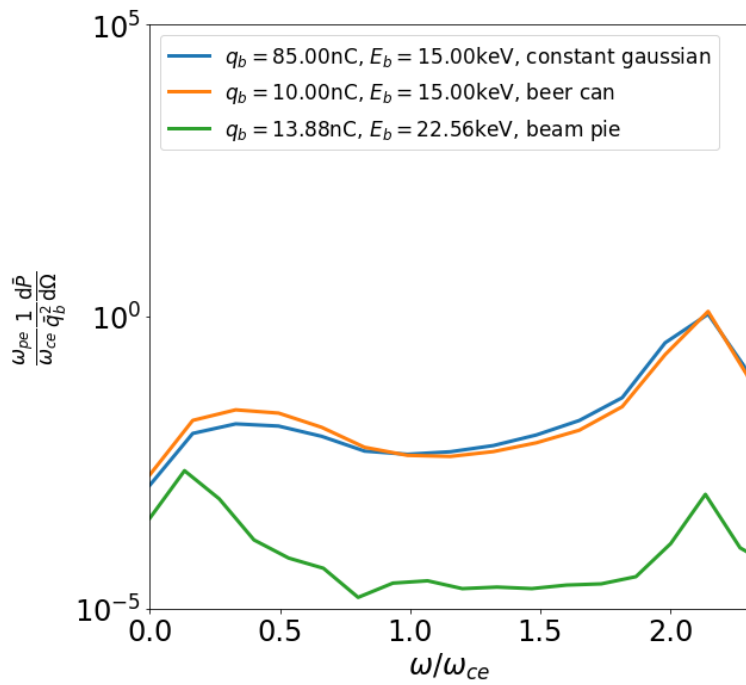
# Model BEAM-PIE beam with SPS to predict waves generated from experiment

- Beam spreads out because of energy spread.
- This reduces the high frequency oscillations (x-mode) significantly (scale of colorbar is different here).
- Low frequency waves (whistler) are only slightly reduced.
- Can compare these results to experiment.





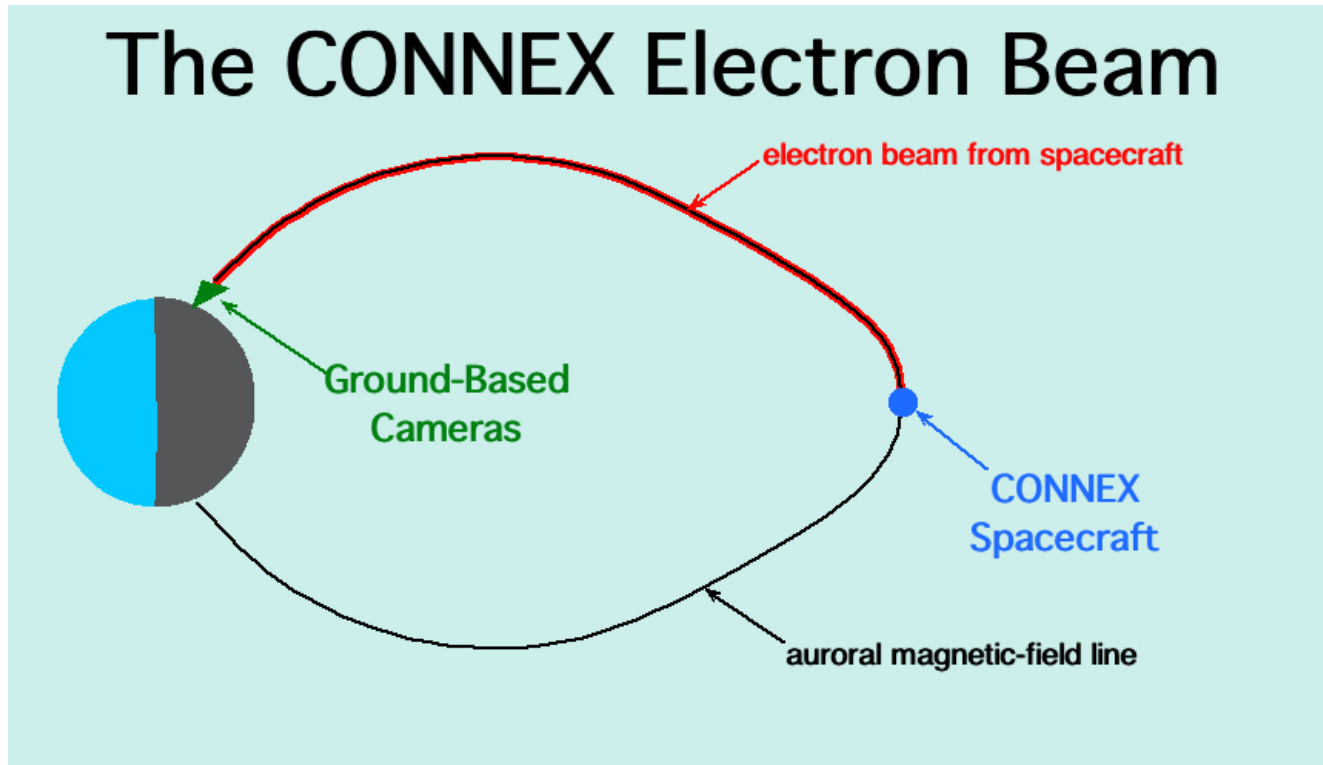
# Comparison of SPS to linear theory



- Compare total wave power from BEAM-PIE and ideal beam in SPS to analytical theory (for an ideal beam).
- Find reasonable agreement between ideal beam and analytical theory for ideal beam, BEAM-PIE significantly lower at high frequencies.
- Allows for quantitative comparison of SPS results.

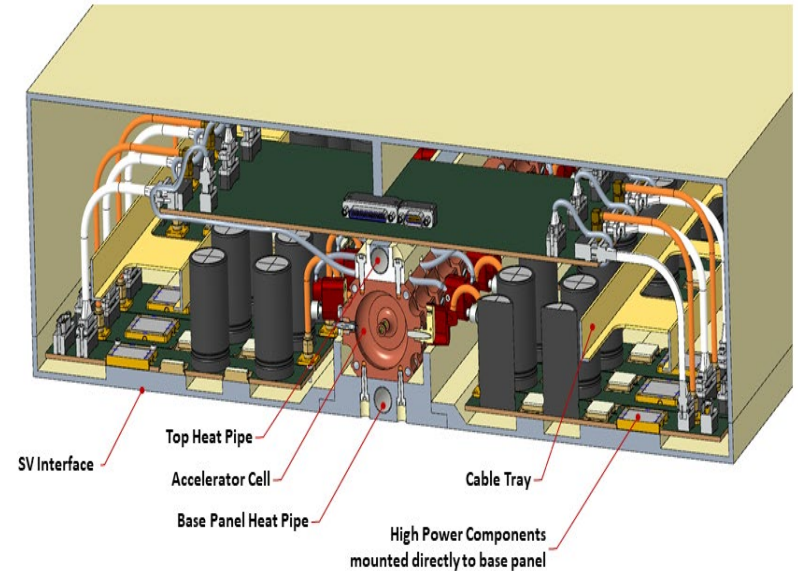
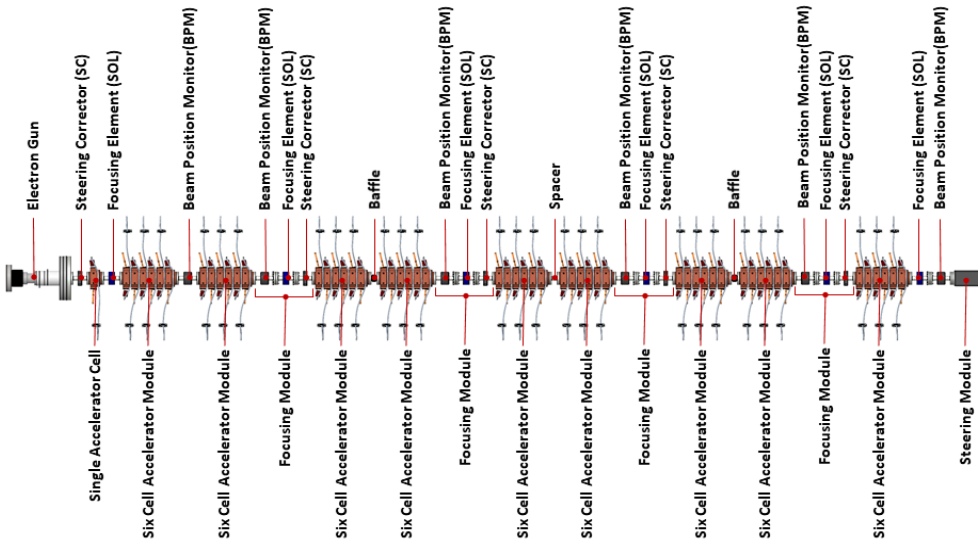
# The CONNEX Experiment

# The CONNEX experiment: map out magnetic field



- Experimentally map magnetic field between the magnetosphere and the ionosphere using a 1 MeV electron beam.
- Image the electron beam hitting the atmosphere by using ground based cameras.
- Answers important questions about the auroras.

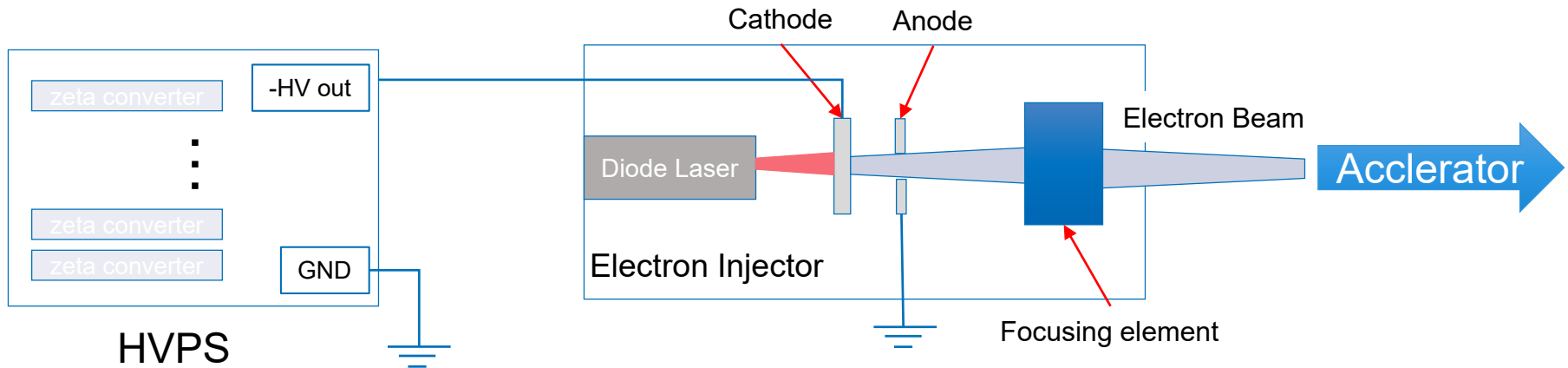
# Experimental design of CONNEX



- The CONNEX experiment will be driven with a 1 MeV, 1 mA electron beam.
- Modular design simplifies things and allows for easy adjustment of total energy of beam.

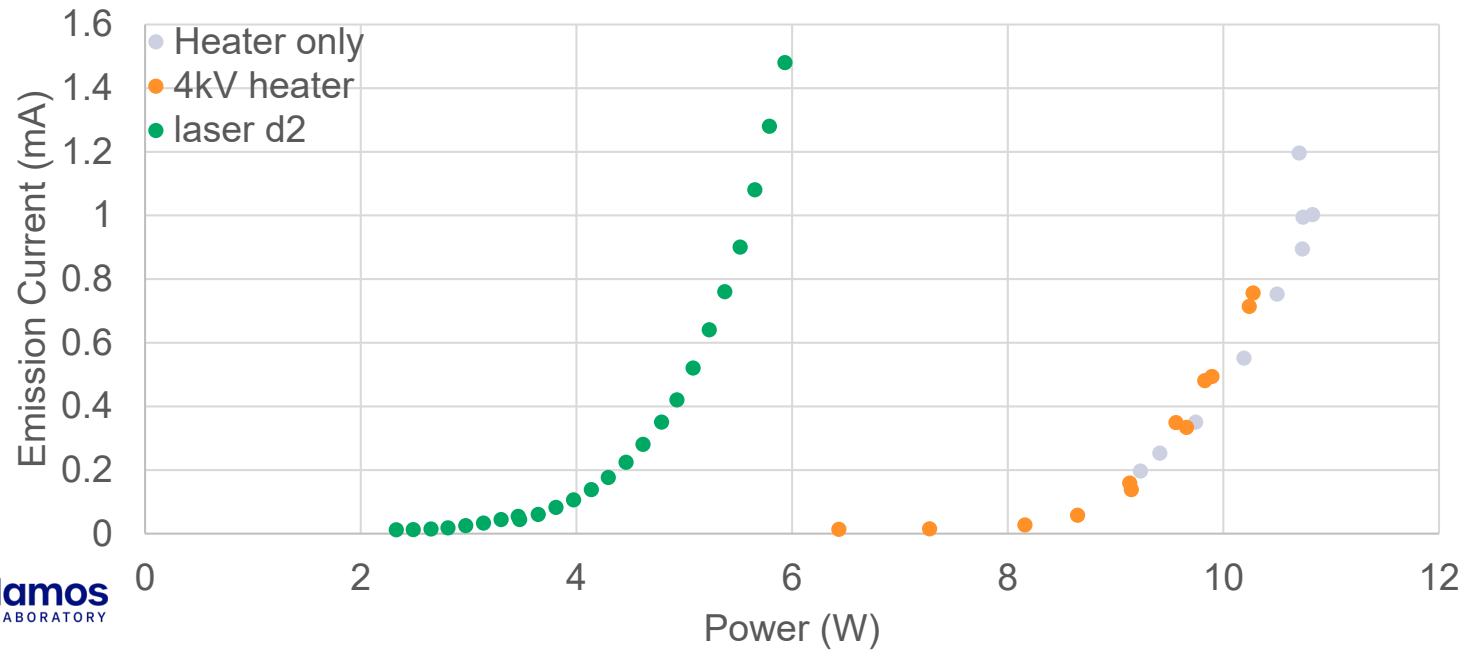
# Low Power Electron Injector for Space Applications

# Low power electron injector for space applications



- Use a diode laser to heat the cathode. This electrically decouples the heating from the high voltage power supply.
- Design a high voltage power supply specifically qualified for space that is modular and scalable.

# Experimental results show laser heating is highly efficient

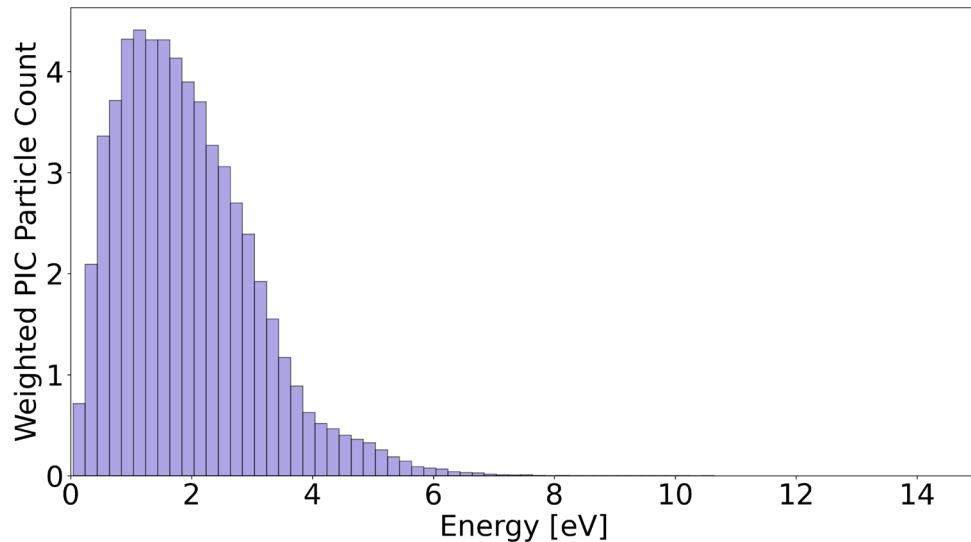


# **A Simple Electron Beam for Plasma Diagnostics in Space**

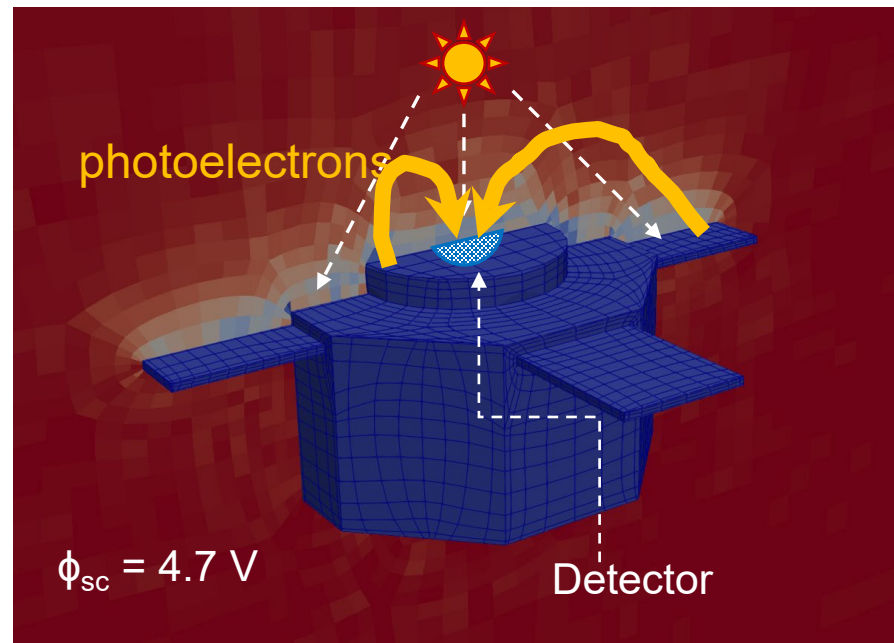


# Detector in sunlight is blinded by photoelectrons

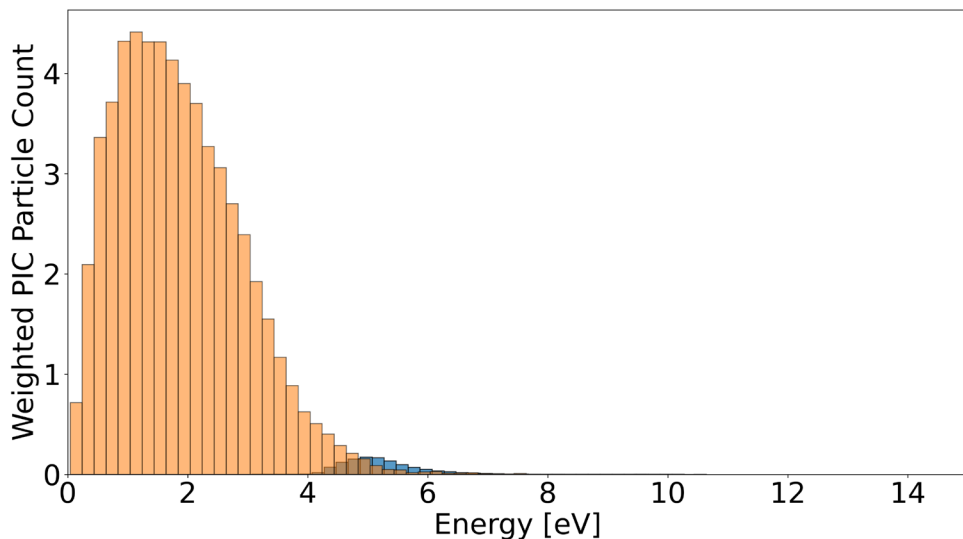
Detector Count: what the detector sees



CPIC Simulation



In simulations we can separate both species



Material: Aluminum (in space)

$J_{ph} = 100 \mu\text{A}/\text{m}^2$  ( $40 \mu\text{A}/\text{m}^2$  in the lab)

$T_{ph} = 1 \text{ eV}$

Density =  $13 \text{ cm}^{-3}$

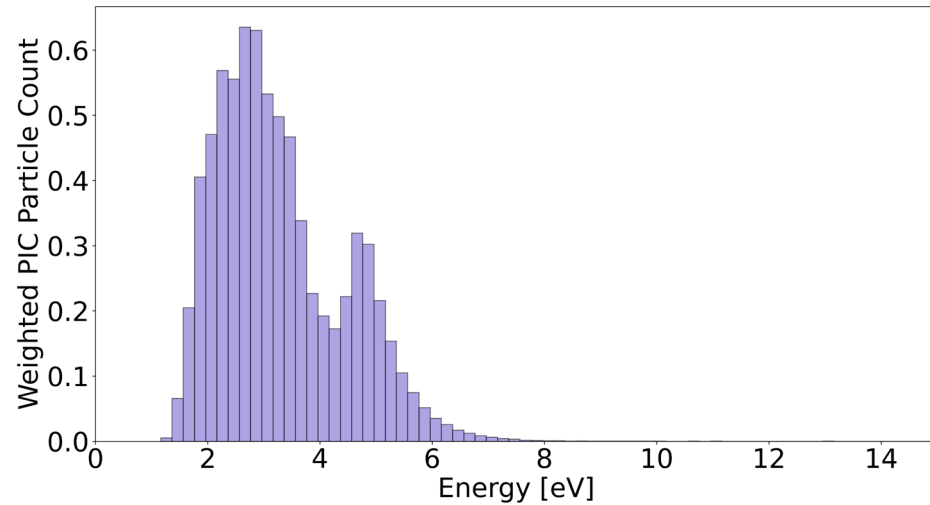
$T_e = 0.5 \text{ eV}$

Debye Length  $\sim 1.4 \text{ m}$

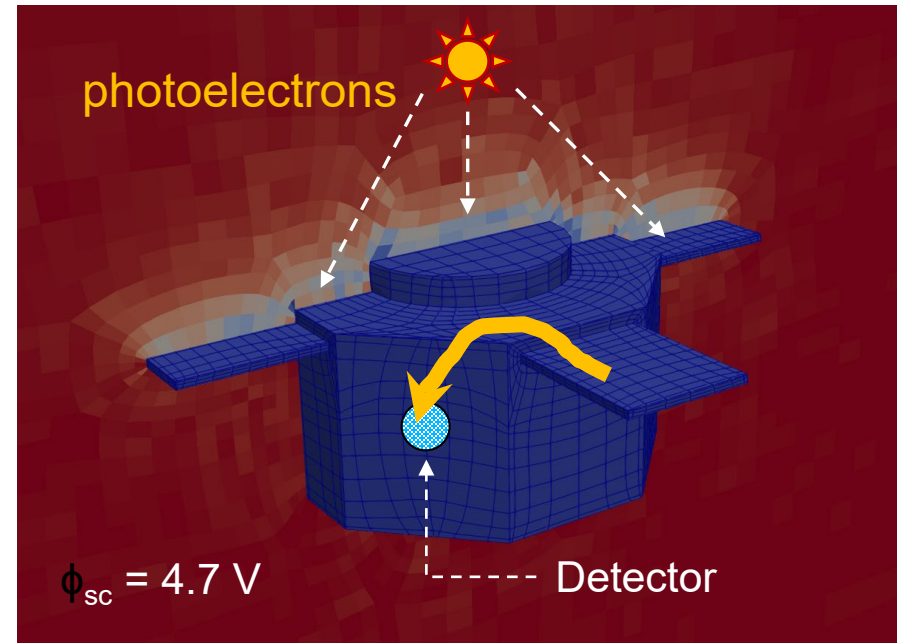
# Having the detector in the shade helps, but in a real case it would still be hard to distinguish the two signals

Note: this is idealized, uncertainties in photoelectron spectrum could still mask the signal!

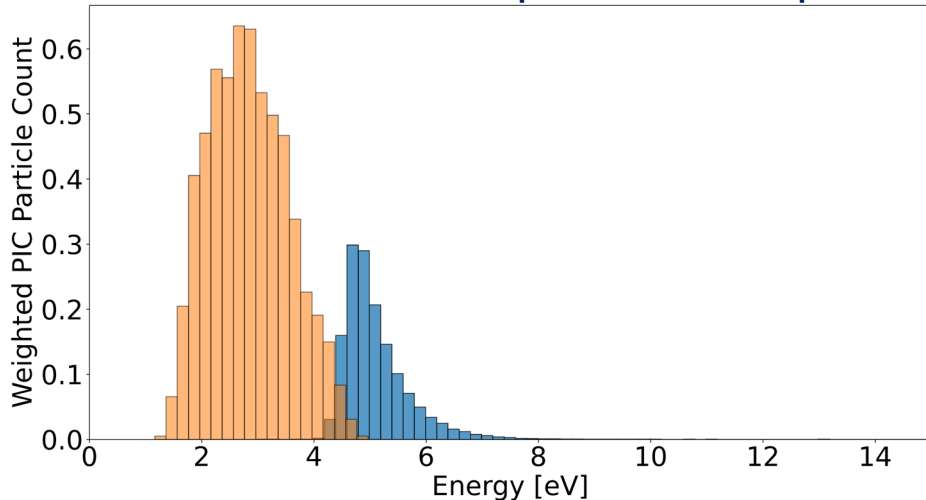
Detector Count: what the detector sees



CPIC Simulation



In simulations we can separate both species

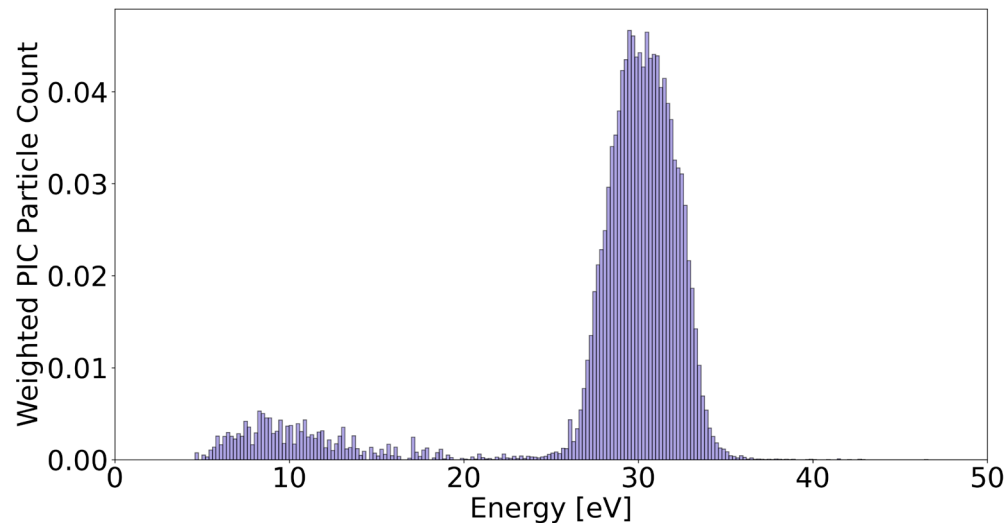


Material: Aluminum (in space)  
 $J_{ph} = 100 \mu A/m^2$  ( $40 \mu A/m^2$  in the lab)  
 $T_{ph} = 1 eV$

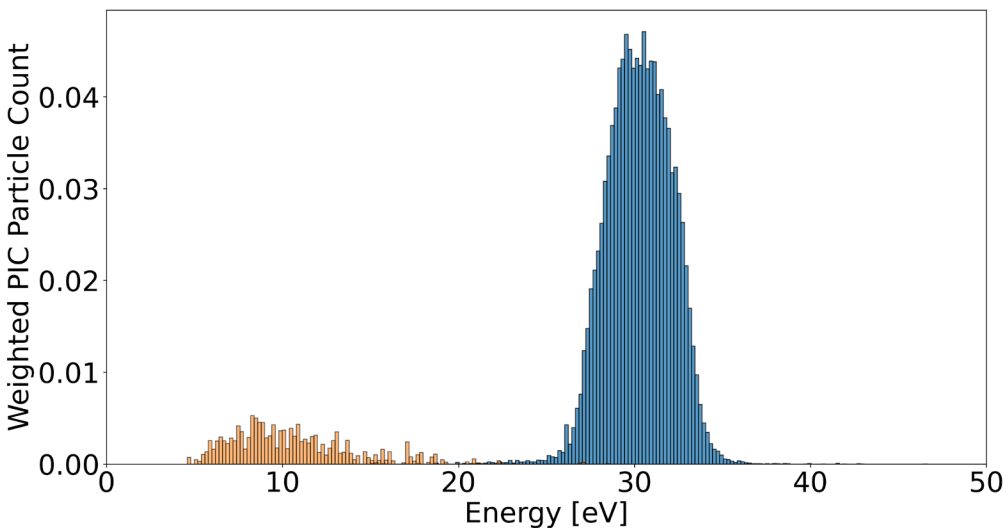
Density =  $13 cm^{-3}$   
 $T_e = 0.5 eV$   
Debye Length  $\sim 1.4 m$

# Here we study a case where we bias the whole spacecraft to +30 V. It works, we can separate the two signals!!!

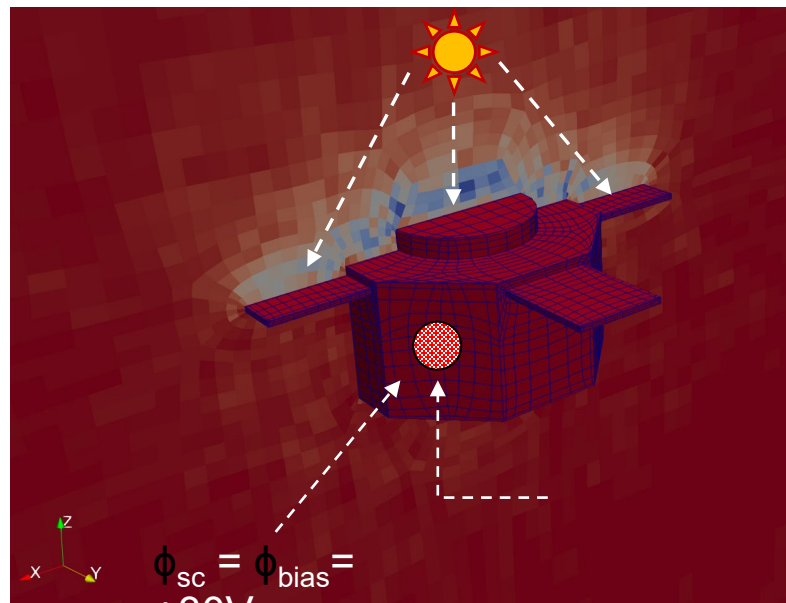
Detector Count: what the detector sees



In simulations we can separate both species



CPIC Simulation



Material: Aluminum (clean)  
 $J_{ph} = 40 \mu A/m^2$  ( $\sim 100 \mu A/m^2$  aged)  
 $T_{ph} = 2 eV$

Density =  $13 cm^{-3}$   
 $T_e = 0.5 eV$   
Debye Length  $\sim 1.4 m$

# Conclusion

- A novel method is being developed and tested at LANL for using HEMTs to power cavities. This allows for a compact electron accelerator that is well suited for space applications.
- In order to mitigate spacecraft charging, a plasma contactor scheme is being developed.
- An electron beam can be used to generate whistler waves in the magnetosphere, which can then mitigate a radiation belt.
- This concept is being tested with the BEAM-PIE experiment.
- Several other important concepts related to accelerators in space are currently being developed at LANL.