# STE-QUEST Electronics Thoughts

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### **Electronics functions**

- Provide stable low voltage/high current power for
  - Narrow band lasers
  - Coils/magnets (for controlling the BEC locations)
- Provide high power supplies for
  - Rb sources
  - Mechanical shutters on Rb source
  - [Pumps assumed controlled at spacecraft level]
- Control
  - AOMs, MEMS switches in laser paths
  - Overall thermal system
  - Precision thermal control of some components
- Read out
  - Housekeeping (temperatures, photodiodes, ....)
  - Science data (CCDs, photodiodes)
- Provide payload autonomy
  - Timeline control (microsecond accuracy over tens of seconds)
  - Implement laser tuning and locking algorithms
  - Safing
  - ...



### **Preliminary Block Diagram**



**Sea** 3

### Budgets (current status – still evolving)

	DIMENSIONS			MASS		POWER*	
	size	length x width x height or length x diameter	volume	without margin	with margin	average	peak
		(mm x mm x mm)	(I)	(kg)	(kg)	(W)	(W)
Physics Package	cylinder	800 x 600	226	101.83	122.08	55.19	135.60
Laser System***	box****		42.7	46.15	55.38	156.64	156.64
Electronics	box	450 x 400 x 300	54	20.84	25.96	222.14	267.98
Ion Pump Control**	box	100 x 100 x 100	1	1.00	1.20	1.50	1.50
subtotal			323.7	169.81	204.61	435.47	561.72
additional system level margin (%)					20	20	20
TOTAL					245.54	522.56	674.06

- A kW-level payload requiring precision timing and sequencing of large numbers of components
- Stringent pointing requirements on the spacecraft
- Data volumes relatively low



### Legacy

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- STE-QUEST M3, M4
- German drop tower/sounding rocket, French parabolic flights, CAL
- Large-scale interferometers (Germany, France, UK, Italy?)
- Cold Atom in Space Community Roadmap
- $\rightarrow$  in principle TRL9, but maybe really like TRL4 for STE-QUEST:
- STE-QUEST will fly higher and longer than other cold atom missions
  → Tougher requirements on radiation hardness
  - $\rightarrow$  Can SWAP be improved?



**RAL Space** 

# Space Electronics at RAL Space

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# **RAL Space Heritage**

#### In-house mixed-signal ASIC design and qualification

Eg: SDO CCD Camera Electronics •Atmospheric Imaging Assembly (AIA), Lockheed Martin •Helioseismic and Magnetic Imager (HMI), Stanford University •SDO flight spares are being used on the LM IRIS SMEX mission

#### 180 Years failure-free operation in space







#### Full instrument build and assembly

Eg: Ptolemy (part of Rosetta mission)



**RAL Space** 



Ion Trap MS at the heart of Ptolemy



Ptolemy before assembly in Philae lander

# **Cold Atoms Electronics**

#### Laser Drive and stabilisation using space qualified<sup>\*</sup> components

- Function:
  - Provides a variable laser current drive of few 100s mA
  - Provides a variable TEC current drive of few A
  - Spectroscopy lock (optimized to MTS)
  - Frequency offset lock (beat note detected up to 10GHz offset)
- Successfully tested
  - Laser Temperature stabilization
  - Free running laser linewidth <50kHz @ 1ms (using RIO planex laser diode)</li>
  - MTS Lock with long term stability <<1MHz</li>



Lab testing laser drive module (in PC104 form factor)



Sample MTS error signal



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\*Some times using equivalent components for the lab demonstrators, with space

counterpart identified

# **Cold Atoms Electronics**

#### In-house DDS ASIC using space tested xxx

- Function:
  - Provides a variable RF signal (100 150 MHz) with <100kHz</li>
  - 2 watts of electrical power to drive the AOM devices.
- Successfully output a waveform and able to obtain:
  - Different pulse shapes (square and gaussian amplitude envelopes)
  - Pulse width <1µs and time jitter <10ns</li>
  - Frequency chirps with frequency resolution 0.1Hz

#### Lab prototypes

- Function:
  - Magnetic Coil Drive
  - Digital pulse generator
- Successfully tested
  - Currents up to 5A with fast switch off (<1ms)</li>
  - TTL pulses with jitter <10ns</li>



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Phase 1 (years 1 and 2)

- Building alteration and hardware procurement
- Accelerator delivery and installation

### Phase 2 (year 3)

- Work-up to 30 mA protons [Q1-Q2 2022]
- Fast neutron fluence rate of 1.8x10<sup>11</sup>n/cm<sup>2</sup>/s
- Thermal fluence rate of 6x10<sup>9</sup> n/cm<sup>2</sup>/s
- Develop modified target system for closer location of samples
- Develop associated fast neutron reflector configuration

Phase 2 (year 4)

- Fast neutron fluence of 1x10<sup>12</sup> n/cm<sup>2</sup>/s
- Achieving a 10<sup>18</sup> integrated neutron fluence required operation for 11.5 days

### Phase 3 (years 4+)

- Develop deuteron beam with enhancement of fluence to  $>3x10^{12}$  n/cm<sup>2</sup>/s
- Achieving a 10<sup>18</sup> integrated neutron fluence required operation for 4 days Development of Dual Beam Facility 2023+



### Thermal Management!



Figure 3-26: Thermal management approach for the STE-QUEST ATI.

## TRL4→6 Potential Facilities and Companies Support



Science & Technology Facilities Council Rutherford Appleton Laboratory





Trym Systems Ltd









Satellite Test Facilities Space Systems Engineering, Space Electronics

Satellite Platforms

Sensor System Integration Vacuum Systems

Consultancy

Mechanical and thermal design

Space electronics, wiring and fibre for data handling

Space and specialist CCD and CMOS imaging solutions

Neutron and Proton irradiation facility AI testbeds