# **Data and Diagnostics Subsystem**

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# The Gravitational Astronomy group

- Our research focus on all aspects related to the field of Gravitational Astronomy
- We lead the Spanish contribution to *LISA*. We also contribute to other GW projects as the *Einstein Telescope* or *ELGAR*.



 We contributed to the mission operations and data analysis of LISA Pathfinder (2015-17), the precursor of LISA.









- Payload control and software
- Diagnostics
  - Temperature sensors
  - <u>Magnetometers</u>
  - Radiation monitor





# **LISA Pathfinder: Data & Diagnostics**

#### Data Management Unit

Commanding and control of LTP subsystems. Single command interface to S/C Provides power supply and Processing to Diagnostics Implements Optical Metrology control loops

#### **Radiation Monitor**



2 PIN diodes in telescope conf. Measuring energy deposition for coincidence events

#### Thermal diagnostic subsystem

![](_page_3_Picture_7.jpeg)

24 sensors (10-4 K/√Hz)

16 heaters

Monitor and charac. of thermal sensitive locations

![](_page_3_Picture_11.jpeg)

#### Magnetic diagnostic subsystem

![](_page_3_Picture_13.jpeg)

- 4 fluxgate magnetometers
- 2 coils

Monitor and charac. of test mass magnetic properties

![](_page_3_Picture_17.jpeg)

# Payload control and software

- Following the LPF heritage, the STE-QUEST proposal included a DMU (Data Management Unit)
  - Main digital control unit was based on the LEON processor developed under an ESA program – the LEON 2 processor.
  - Also included data acquisition electronics and housekeeping systems. It includes e.g. interfaces to photodiodes monitoring the fiber harness, to thermistors and to the CCDs.
- In terms of software, the architecture was based on:
  - Boot Software (BSW)
    - assess and report on the overall DMU hardware health status,
    - establish a reliable communication link with the OBC,
    - check and provide access to RAM and EEPROM memory (where the ASW shall be stored),
    - allow remote patching of Application Software.
  - Application Software (ASW)
    - Handling of the AI subsystems
    - Computation of science data
    - System monitoring, including health status and Onboard Monitoring Function

![](_page_4_Picture_14.jpeg)

![](_page_4_Picture_15.jpeg)

# Payload control and software

![](_page_5_Figure_1.jpeg)

![](_page_5_Picture_2.jpeg)

![](_page_5_Picture_3.jpeg)

### **Temperature sensors (in LPF) — motivation**

- Temperature variations are critical for spacebased GW detectors, mostly because they are ubiquity in the **low frequency band**.
  - Notice: the following effects should be minimised by design.
  - Diagnostics should serve as support to noise model and/or data quality.
- In LPF, temperature gradients across the test mass induce
  - Forces in the test mass. Not directly applicable (in the same scale) to STE-QUEST.
  - Temperature variations in optical elements can lead to path-length changes. Could be relevant to STE-QUEST
  - Path-length changes in the structure holding the experiment inside the thermal shield. Relevant to STE-QUEST
    - "coils of the atom interferometer are allowed to change 260 nm due to thermal expansion although they consume an average power of 22 W". *Rev of Sci. Inst 85, 083105 (2014)*

![](_page_6_Picture_9.jpeg)

L Carbone et al. Phys. Rev. D 76 (2007)

![](_page_6_Figure_11.jpeg)

M Nofrarias et al. Class. Quantum Grav. 24 (2007)

![](_page_6_Picture_13.jpeg)

F Gibert et al. Class. Quantum Grav. 32 (2015)

![](_page_6_Picture_15.jpeg)

### **Temperature sensors (in LPF) — motivation**

![](_page_7_Figure_1.jpeg)

![](_page_7_Picture_2.jpeg)

![](_page_7_Picture_3.jpeg)

- Developing LISA temperature diagnostics subsystem under 'Enhanced temperature measurement for LISA' (LETS) ESA contract
  - Team: <u>IEEC (ES)</u>, DLR (DE), SENER (ES)
- The objective is the design of a prototype temperature subsystem for LISA (**TRL4**)
  - Increasing 1 order of magnitude performance: 1uK/√Hz down to 1mHz

#### • Two main components:

- Front-end electronics composed by Analog Frontend Board (AFB), Power Distributing Board (PDB) and Digital Processing Board (DPB)
- Ultra-stable test bench (DLR) composed by concentric AI thermal shield layers inside vacuum tank. Peltier elements for active control.
- Current status:
  - Achieved sensitivity compatible with LISA in most of the band. Studying low frequency excess noise.

![](_page_8_Picture_10.jpeg)

![](_page_8_Picture_11.jpeg)

![](_page_8_Picture_12.jpeg)

![](_page_8_Picture_13.jpeg)

CSIC

![](_page_8_Picture_14.jpeg)

• Setup running inside temperature control cabine (±5mK stability)

![](_page_9_Picture_2.jpeg)

![](_page_9_Picture_3.jpeg)

![](_page_9_Picture_4.jpeg)

![](_page_10_Figure_1.jpeg)

Armano, M et al. 486 MNRAS (2019)

![](_page_10_Picture_3.jpeg)

![](_page_10_Picture_4.jpeg)

- Developing temperature diagnostics subsystem under 'Optical Fibre Micro-Kelvin Temperature Sensor Network for Sensitive Optical Payloads' under ESA contract
  - Team: <u>IEEC-ICE (ES)</u>, INESCTEC (PT)
- The objective is the design of a prototype temperature subsystem (TRL4) using optical fibers
  - Same requirement as before (1uK/√Hz down to 1mHz) but using optical fibers
  - Avoiding, for instance, spurious magnetic fields close to the experiment
- The current baseline is based on a timeof-flight measurement between FP-sensor and FP-stable reference.
- Current status:
  - Passed PDR, moving to DDR (Detailed Design Review)

![](_page_11_Figure_9.jpeg)

![](_page_11_Picture_10.jpeg)

![](_page_11_Picture_11.jpeg)

# Magnetometers — motivation

0.93

0.91

0.89

D.37

D.85

0.83

- In STE-QUEST monitoring and control of the magnetic field is more prominent than, for instance, in LISA PAthfinder.
  - Still, in both cases stray magnetic fields are a source of **spurious acceleration**.
- In STE-QUEST, the baseline is
  - 4-layer mu-metal configuration
  - Active (low frequency, >1s) compensation coil
  - **Magnetometers** (AMR-like)
  - **Degaussing coils** (~mT) to avoid residual magnetic field induced by the mu-metal
- In LISA PAthfinder, the test mass acts as a magnetic dipole. Magnetic field and magnetic field gradients can induce forces in the test mass
  - The dominant contribution couples **local** gradients with interplanetary fluctuations  $\langle S_{R} \cdot \nabla B_{DC} \rangle$

![](_page_12_Figure_10.jpeg)

[T Hz<sup>.1/2</sup>] 10 S<sup>1/2</sup> Bx 10<sup>-B</sup> Notice that magnetometers on-board must monitor 10-5  $10^{-4}$ 10-2  $10^{-3}$ Armano, M et al. (LPF collaboration) 494 MNRAS (2020) CSIC

![](_page_12_Picture_12.jpeg)

two components

# In-flight magnetic experiments in LPF

- In LISA Pathfinder, magnetic force noise contribution was characterised in-flight.
  - Scheduled experiments with coils to obtain magnetic relevant parameters and derive magnetic force noise projection

![](_page_13_Figure_3.jpeg)

![](_page_13_Figure_4.jpeg)

![](_page_13_Figure_5.jpeg)

![](_page_13_Picture_6.jpeg)

# Magnetometers (in LPF) — motivation

![](_page_14_Figure_1.jpeg)

![](_page_14_Picture_2.jpeg)

![](_page_14_Picture_3.jpeg)

# Magnetometer read-out development

- IEEC developed an improved magnetic diagnostic system more compact and avoiding back-action problems
  - based on Anisotropic magneto-resistors (AMR), solidstate, low noise magnetic sensors.
  - AMR is a compact, hence improves magnetic field spatial resolution
  - Reduction of power needed: AMR (20mW) vs. Fluxgate (1W)
- In order to achieve the required performance, some noise reduction techniques are applied
  - Flipping: applying set/reset pulses to keep magnetic moments aligned. Also removing bridge offset and drifts at low frequencies
  - Electro-magnetic feedback: aims to maintain bridge output near, to compensate bridge sensitivity gain due to thermal fluctuations.
- Current implementation fulfils req. LISA performance
- Developing an alternative compact magnetometer based on MEMs (reducing further sensor power dissipation)
  - Magnetic field modulation, using MEMs resonators and high permeability layers, to mitigate 1/f noise
  - TMR used as sensing device

![](_page_15_Picture_12.jpeg)

![](_page_15_Picture_13.jpeg)

![](_page_15_Figure_14.jpeg)

![](_page_15_Picture_15.jpeg)

![](_page_15_Picture_16.jpeg)

# Magnetometer read-out development

 MEMs based solution tested in representative environment, achieving ≤ 1nT/√Hz @ 1mHz

I/Q demod MEMS6 vacuum

pink noise

white noise

noise requirement

 $10^{-3}$ 

 interplanetary magnetic field fluctuations are typically above (10 nT/√Hz @ 1mHz)

 $10^{-2}$ 

Frequency (Hz)

 $10^{-1}$ 

100

![](_page_16_Picture_3.jpeg)

![](_page_16_Picture_4.jpeg)

101

100

 $10^{-1}$ 

 $10^{-4}$ 

nT//Hz]

![](_page_16_Picture_5.jpeg)

# **Radiation monitor (in LPF) — motivation**

- In LPF, high energy environment responsible for test-mass charging
- In LPF/LISA, two main mechanisms need monitoring
  - Galactic Cosmic Rays (GCR)
    - nearly constant low-level charging rate
    - flux modulation: interaction isotropic flux at heliosphere boundary and solar wind (inducing 27d, 13.5d, 9d periodicities)
  - Solar Event Particles (SEP)
    - can last for days, increasing TM charging orders of magnitude. Not measured and unavoidable in LISA.
- IEEC provided the radiation monitor for LPF
  - was a simple particle counter for particles with E > 70 MeV (no capabilities for particle discrimination).
- IEEC is currently running simulations with PENELOPE (able to resolve electrons down to 100 eV) to study/optimise the LISA radiation monitor

![](_page_17_Picture_11.jpeg)

![](_page_17_Figure_12.jpeg)

Armano et al. (LPF collaboration) Astroparticle Physics 98 (2018)

> Geometry under study (PENELOPE)

![](_page_17_Picture_15.jpeg)

![](_page_17_Picture_16.jpeg)

# Summary

- Following M3/M4 proposal, Data and Diagnostics Subsystem in STE-QUEST main aims are:
  - payload processing and control
  - provide precision diagnostics to support data analysis and/or provide data vetoes
- A lot of heritage already existing from LISA Pathfinder and also synergies from current developments towards LISA.
- Still, a lot of work ahead in optimising how this subsystem can be optimised to provide the essential diagnostic information to the mission.

![](_page_18_Picture_6.jpeg)

![](_page_18_Picture_7.jpeg)

# Thanks for you attention

![](_page_19_Picture_1.jpeg)

![](_page_19_Picture_2.jpeg)