S.M.Liuzzo, A.Franchi, D.Martin, G.Gatta, et al., FCC-ee workshop, CERN May 2022



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ESRF tuning algorithms

Outline History of ESRF tuning

New tuning techniques tests FCC-ee tuning simulations Conclusions

- \rightarrow 60µm mag-mag tolerances,
- → simulated survey replace girder-to-girder
- \rightarrow NOECO
- \rightarrow 10 μ m, no IP errors
- \rightarrow tuning not trivial
- → NOECO

CURRENT ESRF, ACCELERATOR CHAIN LAYOUT AND MAIN PARAMETERS



ERSF Storage ring 1994-2018 and 2019-?? C=844 m E=6 GeV *τ*= 5-26 h * $V_{RF} = 6 MV$ ε_x =134 nm rad $\varepsilon_v = 10 \text{ pm rad}$ I = 40-200 mA * (*) according to filling mode

Compute error tolerances for the SR upgrade



TOLERANCE ESTIMATES OBJECTIVE





GUN, LINAC, Transfer Line to Booster

Booster Tr. Line Booster \rightarrow SR

SR : inject beam

make first turn : lattice steerers, BPM, CT, beam losses ...
few turns -> RF on, beam accumulation (few mA)
orbit, tunes, BPM-QUAD offsets, chromaticity
Response matrix (optics/coupling)
Resonance knobs, sextupoles
specific optics adaptations
beam tuning for users
optimize injection efficiency / lifetime
Wait for vacuum, close collimators

days 1-30 2 shifts/day



WHAT IS SIMULATED IN COMMISSIONING-LIKE BEAM TUNING PROCEDURES



From beam threading till accumulation

- Injection on axis (static bump) or off axis (fit injected beam oscillation) available. Start from injection off-axis.
- 2) All magnets at their nominal working point
- Power orbit steerers to achieve first turn (from simulations, beam survives about 3-4 cells without orbit steering, if magnets & alignment within tolerances).
- 4) Measure and correct tune based on few turns data (most relevant for off-axis injection)
- 5) Switch on RF, search for optimal frequency and phase
 - Beam accumulation

Video: first turn correction simulations for beam injected on axis. Large BPM offsets ($500\mu m$) are included.



Measurements here:

Preparation of the EBS beam commissioning S M Liuzzo *et al* 2019 *J. Phys.: Conf. Ser.* **1350** 012022

Use in commissioning here: https://doi.org/10.1103/PhysRevAccelBeams.24.110701

Not included in tuning simulations as it, but could be done. Usually before step 4, AT is able to provide a closed orbit and tunes.

6)

Page 6 FCC-ee tuning workshop | CERN + Zoom | May 2022 | S.M.Liuzzo et al.

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PROCEDURE FOR OPTICS CORRECTION

% RDT+DISPERSION CORRECTION from lattice error model % fit lattice errors model [rfit]=FitResponseMatrixAndDispersionEBSsimple(... rerr,... r0,... inCOD,... indBPM.... indHCor(1:9*2:end),... % 4 correctors, 1 every 8 cells indHCor(1:9*2:end),... % 4 correctors, 1 every 8 cells [neigQuadFit, neigDipFit, neigSkewFit, neigDipFit],... 4,... [speclab 'fitrm']); % get change of strength of correctors fq=atgetfieldvalues(rfit,indQuadCor,'PolynomB',{1,2}); fs=atgetfieldvalues(rfit,indSkewQuadCor,'PolynomA',{1,2}); % correct RDT and dispersion of fitted error model [~,inCOD,fcq,fcs]=atRDTdispersioncorrection(... rfit,... <<--- fitted error model! not lattice with errors! r0,... indBPM,... indQuadCor,... indSkewQuadCor,... inCOD,... [[floor(linspace(1,neigQuad,5)),neigQuad,neigQuad];... [floor(linspace(1,neigSkew,5)),neigSkew,neigSkew]]',... [true],... 1.0.... [0.8 0.1 0.8],... ModelRM);

%fcq=atgetfieldvalues(rfitcor,indQuadCor,'PolynomB',{1,2});
%fcs=atgetfieldvalues(rfitcor,indSkewQuadCor,'PolynomA',{1,2});

% store proposed correction dcq(1,:)=(fcq-fq); dcs(1,:)=(fcs-fs); 33 A Fr Commissioning-like sequence of corrections is implemented using matlab accelerator toolbox.

Fit of "measured" partial Orbit Response Matrix (slow) → FITTED OPTICS MODEL

Computation of normal and skew quadrupoles RDTs + dispersion and correction → Normal and skew quadrupole correction strengths

This is LOCO equivalent (+ RDTs)

Linear problem + generalize potentially different fit and correction locations

33. A. Franchi, L. Farvacque, J. Chavanne, F. Ewald, B. Nash, K. Scheidt, and R. Tomás, *Vertical emittance reduction and preservation in electron storage rings via resonance driving terms correction*, Phys. Rev. ST Accel. Beams 14, 034002 (2011).



LIFETIME AND DYNAMIC APERTURES VS ERROR SOURCES

A CLUSTER is crucial for this analysis. Each error set to analyze requires 1CPUx4h

Each point is average of 5 seeds.



How to determine the "tolerance" ? → DRAW A LINE (relative impact of each error), RESCALE globally.



Each error, on each magnet family, is studied individually looking at the dependence of DA, lifetime, emittances and all relevant parameters vs error amplitude.

	Required:	DX	DY	DS	DPSI	DK
This table		μ m	μ m	μ m	μ rad	10^-4
represents	DL	>100	>100	1000	500	10
centers errors.	DQ, QF[68]	70	50	500	200	5
	Q[DF][1-5]	100	85	500	500	5
No alignment	SFD	70	50	500	1000	35
Survey errors.	OF	100	100	500	1000	

Sextupoles and high gradient quadrupoles are the most relevant limitations, nevertheless, this alignment specifications are currently achievable. (DX=DY=60μm, 84 μm between two magnets).

ALIGNMENT SURVEY ERRORS, OPTION 1 (THE OLD SCHOOL): GIRDER ERRORS



ALIGNMENT SURVEY ERRORS, OPTION 2 (THE LONG ONE): LONG RANGE ERROR WAVES



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THE MEASURED POSITION OF THE LATTICE

This is where the machine is. Let's use this information!





ALIGNMENT SURVEY ERRORS, OPTION 3 (THE MODERN WAY): SIMULATED SURVEYS (REPLACEMENT OF EXISTING SR)



DYNAMIC APERTURE AND LIFETIME

The above curves are interpolated to set the errors in the S28A lattice.



26/07/2013

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Additional random errors are still required to describe the error on the position of the **magnetic centers** of the magnets.



IN PRACTICE

We have no specific beam dynamics interest in the girder-girder tolerance themselves. Those are not precisely defined among different groups and may lead to lack of understanding. From beam dynamics, we can say that: *large, slow, smooth*, variations of position are <u>not detrimental</u>.

In order to model the lattice at the location where it will be installed, we included in all "predictive" (DA, LT, IE,etc) simulations the **Simulated Survey Position provided by the alignment group**.

From a beam dynamics point of view, as long as **magnet-magnet tolerances** (actually, single magnets rms position errors) are achieved <u>at every location in the machine</u>, including between magnets standing on neighboring girders, we will get the foreseen Dynamic Aperture and LIFETIME.

We let the alignment group decide how to achieve this requirement.

For ESRF-EBS It worked extremely well, largely exceeding expectations/simulations.

rms DX DY alignment

Expected magnet-to-magnet

60-70 μm

From steerers/orbit data Achieved magnet-to-magnet 25-55 µm





Sextupoles and high gradient quadrupoles are the most relevant limitations, nevertheless, this alignment specifications are currently achievable. (DX=DY=60μm, 84 μm between two magnets).

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NOECO: SEXTUPOLES TUNING TO MAXIMIZE LIFETIME



After commissioning,

Improve measurement based tuning compared to the schemes used in commissioninglike simulations.



NOECO CORRECTION AT ESRF, MEASUREMENTS WITH BEAM



PHYSICAL REVIEW ACCELERATORS AND BEAMS 23, 102803 (2020)

Function developed based on:

Nonlinear optics from off-energy closed orbits

David K. Olsson[®], Åke Andersson, and Magnus Sjöström MAX IV Laboratory, Lund University, SE-22100 Lund, Sweden

A.Franchi, N. Carmignani, Sextupole calibrations via measurements of off-energy orbit response matrix and high order dispersion, presented at the 25th European Synchrotron Light Source Workshop (ESLS'17), Dortmund, Germany, Nov. 2017, https://indico.cern.ch/event/657829/contributions/2782617/ attachments/1569843/2475779/ESLS17 Carmignani SextCalibration.pdf.

C06 at 50%

SF2

50, dh=1.0

50, dh=1.0, chrom

50, dh=352e6, chrom

25



correction tests

SF2C06a 50%

Initial

chrom

90, dh=352e6,

100, dh=1.0, chrom

130, dh=1.0, chrom

chrom

=1.0,

Чþ 80,



The specific SD sextupole corrector is <u>NOT found</u>. Nevertheless, the correction applied recovers the bad lifetime but does not completely restore the initial condition.





RANDOM SEXTUPOLE CORRECTORS GIVING ~10H LIFETIME



- The specific random sextupole corrector <u>pattern</u> <u>is NOT found</u>.
- 2) Nevertheless, the correction applied recovers the bad lifetime but does not completely restore the initial condition.
- 3) Zero correction strengths are better then 1 iteration of NOECO correction

Data: 19th Apr 2022





FCC-EE



Apply the same tuning strategy for FCC-ee.

Not trivial!

Yet not comparable to existing studies (see T. Charles presentation)



LATTICE CONVERTED BY FELIX CARLIER



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Errors: No multipole errors



SOME MAGNETS ARE described with "**slices**": **MUST MOVE TOGETHER** For now, do not move

Correction matrices

OrbHCor: {[1892×600 double] OrbVCor: {[1892×600 double] OrbHDPP: [1×1892 double] OrbVDPP: [1×1892 double] TrajHCor: {[1892×600 double] TrajVCor: {[1892×600 double] TrajHDPP: [1×1892 double] TrajVDPP: [1×1892 double] kval: 1.0000e-04 delta: 1.0000e-03

+ matrices for RDT correction

Derivatives for optics fit, on computing cluster

fitrmNorm.mat	176 11/05/2	MAT-file	_			
fitrmNorm.mat (MAT-file)		~	·			
Image: Grad trace IndBPM indHCor indVCor Image: Trace Iman	Value 1x1892 double 1x34 double 1x34 double function_handle 130550x210 double 1x210 struct	waiting f waiting f waiting f waiting f waiting f waiting f	or 50/50 or 50/50 or 50/50 or 50/50 or 50/50 or 50/50 or 48/50	processes processes processes processes processes processes	to to to to	finish. finish. finish. finish. finish. finish.
ſ		waiting f waiting f waiting f waiting f waiting f waiting f waiting f	or 47/50 or 16/50 or 11/50 or 10/50 or 10/50 or 10/50 or 10/50	processes processes processes processes processes processes processes	to to to to to to	finish. finish. finish. finish. finish. finish. finish.

Grouped trying to

follow family naming

Correction strategy:

open trajectory (steerers) Tikhonov tune (quadrupoles, 2 families) RF cavity orbit (steerers) Tikhonov tune (quadrupoles, 2 families) chromaticity (sextupoles, 2 families) orbit (steerers) Tikhonov tune (quadrupoles, 2 families) chromaticity (sextupoles, 2 families) Fit Quad+Dip Errors Correct RDT and Dispersion of fitted model orbit (steerers) Tikhonov tune (quadrupoles, 2 families) chromaticity (sextupoles, 2 families) Fit Quad+Dip Errors Correct RDT and Dispersion of fitted model RF cavity tune (quadrupoles, 2 families)



COMMISSIONING LIKE CORRECTION SEQUENCE COMPLETED FOR 3 SEEDS WITH LIMITED ERRORS.



DYNAMIC APERTURE AND MOMENTUM ACCEPTANCE WITH ERRORS



This is only a preliminary plot, only showing that we can compute DA and MA for FCC-ee with errors and correction.

We need to work more time on it!

Assume the recent updates in AT Tune the correction parameters (singular values cuts) Tune the correction loop steps (tune, orbit, etc)

The correction loop is unstable above 20-30 um (no IP errors). In most cases a closed orbit is found but tunes are NaN. May be fixed including an estimate of tunes from few turns, not presently available in the matlab AT toolkit.



NOECO SIMULATIONS (FIT & CORRECTION) FOR FCC-EE





NOECO SIMULATIONS (FIT & CORRECTION) FOR FCC-EE



NOECO SIMULATIONS (FIT & CORRECTION) FOR FCC-EE





Commissioning-like simulations do not include many of the steps that are actually done during commissioning. Space for future developments.

For EBS tuning was not the real issue. It was rather to set the most correct and realistic errors.

Only one number is necessary for magnetic centers tolerances (for EBS 60 um). All other values are useful only if locally this value may not be achieved.

Use simulated survey rather than girder-to-girder tolerances: more information, more realistic, easier to simulate.

NOECO has been tested in MDT and shows good potential, even if it does not improve compared to our best optimized settings for operation.

Contrary to EBS, FCC-ee optics tuning with errors is not trivial at all. Even without radiation!

The matlab AT tuning procedure used for EBS is being adapted to FCC. Seen the amount of changes needed till now and the speed at which AT is developing, it is likely that the work will start over using pyAT. This activity is synergic with several other laboratories working on python AT commissioning like simulations.

NOECO has been used in simulations also for the FCCee lattice, recovering the local momentum acceptance also in presence of errors.



BACKUP



FIRST-TURNS TRAJECTORY CORRECTION PROGRESS ON TBT BPMS



Preparation of the EBS beam commissioning S M Liuzzo et al 2019 J. Phys.: Conf. Ser. **1350** 012022



LOW INTENSITY TUNING (PHASE 1)

Optics tuning/correction

- response matrix
- beam-based alignment
- tune working point
- emittances
- chromaticity
- dynamic apertures
- specific optics tuning (ex: phase advance between sextupoles)

Separated simulations for BBA, DA and special tunings.

For BBA in the tuning loop we assume it is possible by reducing BPM offsets from 500 to 50 um. (not as trivial as it sounds, as changing the offset the orbit has to be recorrected)



Response matrix (ORM) measurements is used for:

- Precise orbit control (full, not compulsory)
- Optics and coupling tuning (partial)
- Check magnet calibrations It is possible to use AC steerers for a fast partial response measurement. At ESRF this was only possible after the initial SLOW measurements, for calibration.



	$\sqrt{\langle ()^2 angle}$	x	$(\eta_x - \eta_x^0)$	$rac{\Deltaeta_x}{eta_x^0}$	ϵ_x	y	η_y	$rac{\Deltaeta_y}{eta_y^0}$	ϵ_y	DA	$ au_{Touschek}$
	error	μm	μm	%	pmrad	μm	μm	%	pmrad	mm	s
	thresholds:	5.0e-05	5.0e-04	1.0e-02	4.0e-12	5.0e-05	5.0e-04	1.0e-02	4.0e-12	-1.0e-02	3.6e + 04
	DL x,y [0, 1.00e-04]	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04
The worst	DL x,y,s,psi [0, 1.00e-04]	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04
crossing of	DQ x,y [0, 1.00e-04]	7.2e-05	6.9e-05	1.0e-04	1.0e-04	8.6e-05	6.4e-05	1.0e-04	1.0e-04	1.0e-04	1.0e-04
the "line" for	DQ x,y,s,psi [0, 1.00e-04]	7.3e-05	6.9e-05	1.0e-04	1.0e-04	8.1e-05	6.6e-05	1.0e-04	1.0e-04	1.0e-04	1.0e-04
	Octupole x,y [0, 1.00e-04]	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04
each	Octupole x,y,s,psi [0, 1.00e-04]	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04
performance	QF68 x, y [0, 1.00e-04]	2.8e-05	3.1e-05	1.0e-04	1.0e-04	8.7e-05	1.0e-04	7.1e-05	1.0e-04	7.0e-05	9.5e-05
portormation	QF68 x, y, s, psi [0, 1.00e-04]	2.6e-05	2.9e-05	1.0e-04	1.0e-04	8.3e-05	1.0e-04	6.5e-05	2.2e-05	8.2e-05	1.0e-04
parameter	QFD15 x,y [0, 1.00e-04	1 0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04
	QFD15 x,y,s,psi $[0, 1.00e-04]$	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04	1.0e-04
	Quadrupole x,y $[0, 1.00e-04]$	2.8e-05	2.5e-05	1.0e-04	1.0e-04	8.3e-05	1.0e-04	6.9e-05	7.1e-05	9.9e-05	8.9e-05
	Quadrupole x,y,s,psi [0, 1.00e-04]	2.6e-05	2.9e-05	1.0e-04	1.0e-04	8.3e-05	1.0e-04	6.3e-05	1.0e-04	8.0e-05	1.0e-04
	Sextupole x,y $[0, 1.00e-04]$	1.0e-04	5.8e-05	8.9e-05	1.0e-04	1.0e-04	1.0e-04	6.8e-05	5.4e-05	6.0e-05	1.0e-04
	Sextupole x,y,s,psi [0, 1.00e-04]	1.0e-04	5.0e-05	8.3e-05	.0e-04	1.0e-04	1.0e-04	7.5e-05	9.1e-05	6.4e-05	1.0e-04

- DQ (combined function dipoles) behave as quadrupoles concerning errors. Evident the impact on vertical dispersion compared to the other quadrupoles (defocussing quadrupoles).
- Quadrupoles have large impact on orbit and horizontal dispersion, also in this case, lifetime and DA are strongly affected. QF6 and QF8 are dominant.
- Sextupoles have the largest impact on DA, they are also the strongest source of beta-beating and emittance as expected.
- Octupoles influence is limited compared to quadrupole and sextupoles, nevertheless they do have an impact on DA. Their effect on lifetime is very small.
- Rotations up to 100 urad have impact on the various parameters but limited.



Last simulations with errors only in arcs and of 30um rms.

Now 100 um rms errors in all the elements, a part IP magnets (50um)

```
errset= 1.0 * [...H, V, rot, field
   [ 100e-6, 100e-6, 100e-6, 1e-4, 0, 0];... B*
   [ 100e-6, 100e-6, 100e-6, 1e-4, 0, 0];... BG*
   [ 100e-6, 100e-6, 100e-6, 1e-4, 0, 0];... Q[FD]#*
   [ 100e-6, 100e-6, 100e-6, 1e-4, 0, 0];... S*
   0.5*[ 100e-6, 100e-6, 1e-4, 0, 0];... S*
   0.5*[ 100e-6, 100e-6, 100e-6, 1e-4, 0, 0];... BPM (offset, rotation, reading, gain h gainv)
   0.5*[ 100e-6, 100e-6, 100e-6, 1e-4, 0, 0];... PQ*
   ]; %
```

radon = 0;



OrbHCor: OrbVCor:	{[1892×1892 double] {[1892×1892 double]	[1892×1892 double] [1892×1892 double]	[1892×1892 do [1892×1892 do	ouble] [1892×1892 ouble] [1892×1892	double]} double]}	orbit
OrbHDPP: OrbVDPP: TrajHCor:	[1×1892 double] [1×1892 double] {[1892×1892 double]	[1892×1892 double]	[1892×1892 do	uble] [1892×1892	double]}	
TrajVCor: TrajHDPP:	{[1892×1892 double] [1×1892 double]	[1892×1892 double]	[1892×1892 do	uble] [1892×1892	double]}	I rajectory (I hreading
DispHCor: DispVCor:	[1×1892 double] {[1892×1892 double] {[1892×1892 double]	[1892×1892 double] [1892×1892 double]	[1892×1892 do [1892×1892 do	ouble] [1892×1892 ouble] [1892×1892	double]} double]}	
DispHDPP: DispVDPP:	[1×1892 double] [1×1892 double]					
DispQCor: DispSCor: TuneQCor:	{[1892×1892 double] {[1892×1892 double] {[1×1892 double] [1	[1892×1892 double] [1892×1892 double] ×1892 double] [1×18	[1892×1892 do [1892×1892 do 392 double] [1	uble] [1892×1892 uble] [1892×1892 ×1892 double]}	double]} double]}	Optics correction

Long quads, small kicks (not integrated 1/m)!



Solution:

add more correctors : NOT OK. The corrector indexes look correct, first corrector before first BPM.

Nevertheless the first block of the Trajectory response is ZEROS, thus the trajectory correction never starts.



RECOMPUTE TRAJECTORY RM.

BEAM THRE	ADING IS RU	JNNING C	K			V PLANE
All magnets	ON at their	nominal				correcting available V trajectory X: 104.969 -> 52.523 um Y: 43.905 -> 21.887 um Search closed orbit Trajectory correction: nbpms= 1875 ncor: 1875, 1875, computing ORM for available trajectory
Steerers por there is bear	wered as so m on the ne	oon as ext BPMs				H PLANE correcting available V trajectory V PLANE correcting available V trajectory X: 53.389 -> 26.960 um Y: 54.155 -> 27.303 um Search closed orbit
Steering till of the orbit w	one turn an vith the last	d closure 2 steere	ſS.			Trajectory correction: nbpms= 1885 ncor: 1885, 1885, computing ORM for available trajectory H PLANE correcting available V trajectory V PLANE
Allow for 0.5 amplitude ac has to be re	omm trajecto ccepted. Ma visited.	ory ay be this				Correcting available V trajectory X: 96.471 -> 48.678 um Y: 81.014 -> 40.532 um Search closed orbit Trajectory correction: nbpms= 1891 ncor: 1891, 1891, computing ORM for available trajectory
	Yanna hilikihidan wana di tamilikan ku	many have been been been been been been been be	raaffaataalaridaada		w town have been a feature of the	vertical horizontal
<u>9</u> _1					¶ 	
0 1	2	3	4	5 s [m]	6	Found closed orbit X: 12.973 -> 12.973 um
38 FCC-ee tuning workshop 0	CERN + Zoom May 202	2 S.M.Liuzzo et al.				Finished: Correction Step: 1/16 Elapsed time is 317.886735 seconds. The European Synchrotron ESRF

Sommand window

Once a closed orbit is found, correct it using all available H/V steerers.

No limits set to steerers for now.

Truncation of SVD to 500 singular vectors / >1800, to keep forces required to a minimum.

There is a closed orbit but tunes are several units different from nominal.







Found closed orbit X: 23.076 -> 23.076 um Y: 13.526 -> 13.526 um Finished: Correction Step: 1/16 Elapsed time is 136.982589 seconds. Correction Step: 2/16

Tune Matching Nominal tune: 222.17213, 222.39982, Initial tune: 222.18332, 222.30962, Going to tune: 222.17213, 222.39982, Single correction step Could not match Tune Finished: Correction Step: 2/16 Elapsed time is 6.000173 seconds. Correction Step: 3/16 Set RF cavity. 120 MV, 121657 buckets, 0 radiation Finished: Correction Step: 3/16

Finished: Correction Step: 4/16 Elapsed time is 11.936607 seconds. Correction Step: 5/16 Tune Matching Nominal tune: 222.17213, 222.39982, Initial tune: 222.05755, 213.78119, Going to tune: 222.17213, 222.39982, Could not match Tune Finished: Correction Step: 5/16 Elapsed time is 10.077279 seconds. Correction Step: 6/16 - - - - chromaticty correction - - - -All SF and All SD moved by a constant value Nominal chrom: -0.130, 0.348,

O: Add orbit in in all functions, it must be given to atlinopt, atfittune, atm





TRY REDUCED ERRORS AND SIMPLIFIED ORBIT CORRECTION (2 STEPS ONLY)





Girder displacements and rotations

Girder displacements and rotations + random errors Quad. and Sext.

4 girders, BPMs are moved with the girder.

How

GIRDER



26/07/2013 | WP1 ESRF upgrade | May 2014 | S.M.Liuzzo The European Synchrotron

ESRF

		dyna	mic apert	ure		emitt	tance	accep	otance		were
error	Н	V	area	+2%	-2%	ϵ_x	ϵ_y	Η	V	mi	defined
AllMisal	39.0	35.8	27.2	f 25.2	27.4	100.0	100.0	31.5	36.1	25.	2
AllMisalBPM	3869.3	3867.1	2585.4	f2517.6	2867.8	10000.0	10000.0	3312.4	3496.4	2517	7.6
$\Delta b_1/b_1$	1000.0	1000.0	f1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	10^{-6}
$\Delta b_2/b_2$	1000.0	1000.0	f1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	1000.0	10^{-6}
$\Delta b_3/b_3$	10000.0	10000.0	f7427.8	7826.8	7841.3	10000.0	10000.0	10000.0	10000.0	7427.8	10^{-6}
$Dipole\Delta\psi$	500.0	500.0	f 500.0	500.0	500.0	500.0	500.0	500.0	500.0	500.0	μrad
$Girder\Delta\psi$	500.0	500.0	f 500.0	500.0	500.0	500.0	500.0	500.0	500.0	500.0	μrad
$Girder\Delta x$	100.0	100.0	f 100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	μm
$Girder\Delta y$	100.0	100.0	100.0	f 73.2	77.7	100.0	100.0	100.0	100.0	73.2	μm
$Quadrupole\Delta\psi$	500.0	500.0	f 500.0	500.0	500.0	500.0	500.0	500.0	500.0	500.0	μrad
$Quadrupole\Delta x$	66.9	100.0	59.0	f 46.4	50.1	100.0	100.0	59.7	100.0	46.4	μm
$Quadrupole\Delta y$	100.0	100.0	f 100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	μm
$Sextupole\Delta x$	100.0	100.0	f 100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	μm
$Sextupole\Delta y$	100.0	100.0	f 100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	μm
	<u> </u>					Ŷ	Î	[tion
	Strong	n imna	ct of			No influ	ience	L		JI COFFECTION	
		y mpa									
	quadr	upole (errors			on emit	tances				



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THE MEASURED POSITION OF THE LATTICE

How GIRDER tolerances This is where the machine is. Let's use this information! were excluded x 10⁻³ 0 z [m] nominal -1 current -2 -200 ESRF 1994-2018 EBS 2019-?? We placed EBS where the old SR used to be. Everyone is happy. 100 150 100 50 0 0 -100 -50 -100 -200 -150 y [m] x [m]



SURVEY ERRORS PLUS RANDOM ERRORS



How GIRDER tolerances were excluded

- The blue curve represents the current ESRF storage ring position.
- The x symbols are the interpolated positions on this curve for the S28A lattice.
- BPM are positioned on the same curve.
- DX and DZ and girder rotations are applied.
- Errors are ramped in steps of 20% up to the nominal value.
- Added random errors to account for unknown magnetic center position

Random Gaussian DX DY in quadrupoles and sextupoles on top of the survey errors.



LONG RANGE ERRORS

Error "waves" applied to the lattice: DX, DY, Rotations for quad and sext.



Correction performed for each frequency and amplitude set as:

- Open trajectory correction
- Orbit correction
- RDT correction

Simulations performed on the OAR asd cluster, 8h-600cpu



How

GIRDER

tolerances

were









MISALIGNMENT PATTERN, EVERY 2 CELLS





Impact of error "waves" on horizontal orbit

At a wavelength of L/128~6m 30um errors amplitude give 30 um residual orbit distortion. At a wavelength of L/2~422m 7mm errors amplitude give 30 um residual orbit distortion.



How

GIRDER

ESRF

The European Synchrotron

THE MEASURED POSITION OF THE LATTICE

How GIRDER and LONG RANGE tolerances were excluded



This is where the machine is. Let's use this information!







	total std corrector values applyed	
	HK (288) [1/m]: 0.00e+00 -> 1.24e-	04
	VK (288) [1/m]: 0.00e+00 -> 1.63e-	04
	SK (288) [1/m2]: 0.00e+00 -> 0.00e	+00
	OK (514) $[1/m2]: 0.00e+00 \rightarrow 0.00e$	+00
	HKL (288) [rad]: 0 00e+00 \rightarrow 1 59e	-05
	VKL (288) [rad]: 0.00e+00 \rightarrow 1.14e	-05
	SKI (288) [T/m] = 0.000+00 -> 0.000+00 -> 0.000+00 -> 0.000+00 -> 0.000+00-00-	+00
	OKI (200) [1/m]: 0.000+00 -> 0.000	+00
	QKL (314) [1/m]: 0.000+00 -> 0.000	+00
	residual orbit and dispersion	
	OH (288) [m]: 6.11e-05 -> 1.40e-05	
	OV (288) [m]: 1.37e-04 -> 1.84e-06	
	DH (288) [m]: 7.00e-04 -> 2.83e-04	
	DV (288) [m]:1.72e-03 -> 3.68e-04	
	BBH (288) %: 3.8 -> 2.4	
	BBV (288) %: 1.8 -> 2.5	
	PhH (288) : 2.23e-02 -> 1.34e-02	
	PhV (288) : 1.19e-02 -> 1.38e-02	
	tune and emittance	
	Ox [0.580]: 0.583 -> 0.580	
	$Ov [0.620]: 0.622 \rightarrow 0.620$	
	EX [147.528 pm]: 147.809 -> 147.69	2
	EY $[0 \ 0 \ 0 \ 0 \ m] \cdot 1 \ 477 => 0 \ 133$	_
	Li [0.000pm]. 1.1// / 0.100	
	$(e_{x}, e_{y}) \cdot 147, 53$	0.0
arrora	$(a_{X}, a_{Y}) \cdot 1/7 81 $ 1	18 18
errors	$(e_x, e_y): 147.01, 1.$	40 ,
COLLECI	$(e_x, e_y): 14/.03, 0.$	1J ,

Freq 8, every 4 cells





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COMPARISON OF VARIOUS ERROR SOURCES (OTHER PARAMETERS)



DRAFT TABLE OF TOLERABLE RMS OF ERROR DISTRIBUTIONS

	New value	TDS value	Error (S28 Table)	New value	TDS value	Error (S28 Table)
DS = longitudinal	500 μ rad	300	DL DPSI	Splitted	300 mrad	Dipole DPSI
displacement	100 μm	50	DL DX	Splitted	50 µm	Dipole DX
DX = radial	100 μ m	50	DL DY	Splitted	50 µm	Dipole DY
displacement	1000 µm	-	DL DS	Splitted	50 µm	Quadrupole DX
DY = vertical	200 μ rad	300	DQ and QF6-8 DPSI	Splitted	100 µm	Quadrupole DY
displacement	50 μm	50	DQ and QF6-8 DX	Splitted	350 µrad	Quadrupole DPSI
DPSI = rotation	70 μm	50	DQ and QF6-8 DY	70	50 µm	Sextupole DX
about beam axis	500 μm	-	DQ and QF6-8 DS	50	75 μm	Sextupole DY
	100 μ m	-	Octupole DX	500 μrad	-	Sextupole DPSI
	100 μm	-	Octupole DY	Removed	50 µm	Girder DX
	500 μ rad	-	Octupole DPSI	Removed	50 µm	Girder DY
Girder errors	1000 μm	-	Octupole DS	Removed	200 µrad	Girder DPSI
replaced by	500 μ rad	3 <mark>50</mark>	Q [D-F] [1-5] DPSI	Separated	50 µm	BPM X-offset
summing first 4	100 μm	5 <mark>0</mark>	Q [D-F] [1-5] DX	Separated	50 µm	BPM Y-offset
long wavelengths	85 μm	1 <mark>00</mark>	Q [D-F] [1-5] DY	10 10-4	10 10-4	Dipole DK0/DK0
amplitude.	500 μm	-	Q [D-F] [1-5] DS	5 10 ⁻⁴	5 10 ⁻⁴	Quadrupole DK1/K1
1	1000 μ m	-	Sextupole DS	35 10-4	35 10-4	Sextupole DK2/K2

ESRF



D.A. AND LIFETIME: TABLE ABOVE + WAVES + BPM ERRORS 50 UM RMS





DA about 7mm at S3 septum. Some seeds have bad DA and some are lost. Work in progress to recover them. this may change the above list of tolerable errors. Lifetime is about 7h.





ALIGNMENT SURVEY ERRORS, OPTION 3 (THE MODERN WAY): SIMULATED SURVEYS (IDEAL POSITION)





Simulated survey errors about the nominal position of the lattice.

Rotations are random, but for a 21 urad systematic outward tilt.

Data provided in 320 points for the current lattice.

ADDITIONAL RANDOM ERRORS



X-RAY BEAM POSITION AT ID (60 METERS) : DECISION TO PLACE THE NEW SR WHERE THE OLD WAS.



All ID are assumed to be at 60m from the source.

The position of the beam after 60m is very similar for ESRF and S28A considering the current survey measurement.

The position if the ring was aligned on the reference circumference would be about (0,0) for al ID.

ESRF

The European Synchrotron



LARGER ERRORS ON MORE MAGNETS





% RDT+DISPERSION CORRECTION from lattice error model

```
% fit lattice errors model
[rfit]=FitResponseMatrixAndDispersionEBSsimple(...
    rerr,...
   r0,...
   inCOD,...
   indBPM....
   indHCor(1:9*2:end),... % 4 correctors, 1 every 8 cells
   indHCor(1:9*2:end),... % 4 correctors, 1 every 8 cells
    [neigQuadFit, neigDipFit, neigSkewFit, neigDipFit],...
    4,...
   [speclab 'fitrm']);
% get change of strength of correctors
fq=atgetfieldvalues(rfit,indQuadCor, 'PolynomB', {1,2});
fs=atgetfieldvalues(rfit,indSkewQuadCor,'PolynomA',{1,2});
% correct RDT and dispersion of fitted error model
[~,inCOD,fcq,fcs]=atRDTdispersioncorrection(...
   rfit,... <<--- fitted error model! not lattice with errors!
    r0,...
   indBPM,...
   indQuadCor,...
   indSkewQuadCor,...
   inCOD,...
   [[floor(linspace(1,neigQuad,5)),neigQuad,neigQuad];...
   [floor(linspace(1,neigSkew,5)),neigSkew,neigSkew]]',...
   [true],...
   1.0....
   [0.8 0.1 0.8],...
   ModelRM);
```

%fcq=atgetfieldvalues(rfitcor,indQuadCor,'PolynomB',{1,2});
%fcs=atgetfieldvalues(rfitcor,indSkewQuadCor,'PolynomA',{1,2});

```
% store proposed correction
dcq(l,:)=(fcq-fq);
dcs(l,:)=(fcs-fs); 33 A France
```

Fit of "measured" partial Orbit Response Matrix (slow) → FITTED OPTICS MODEL

Computation of normal and skew quadrupoles RDTs + dispersion and correction → Normal and skew quadrupole correction strengths

This is LOCO equivalent (+ RDTs)

Linear problem + generalize potentially different fit and correction locations

33. A. Franchi, L. Farvacque, J. Chavanne, F. Ewald, B. Nash, K. Scheidt, and R. Tomás, *Vertical emittance reduction and preservation in electron storage rings via resonance driving terms correction*, Phys. Rev. ST Accel. Beams 14, 034002 (2011).

