

Advances in low speed gas cooling and extra ligthweght self-supported mechanics for ALICE ITS-3 modules

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Reported by G.Feofilov

ITS-upgrade WP5 meeting, 10 May 2022, 16300 \rightarrow 17:00 Europe/Zurich

https://indico.cern.ch/event/1158834/





Introduction

Part-I: Low-speed gas cooling of large area thin pixel sensors Space blanket instead of thermally isolated barrel Space blanket and 3 cylinder layers with heaters



Part-II: Ultralighweight self-supported mechanics for ALICE ITS-3 modules



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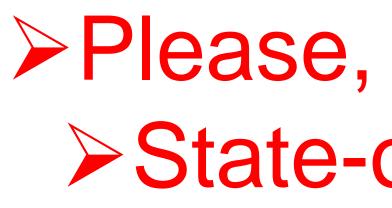
Geometrical parameters of the ITS-3[1]

Beampipe inner/outer radius (mm)

IB Layer parameters

Radial position (mm) Length (sensitive area) (mm) Pseudo-rapidity coverage^a Active area (cm²) Pixel sensors dimensions (mm²) Number of pixel sensors / layer Pixel size (μm^2)

^a The pseudorapidity coverage of the detector layers refers to tracks originating from a collision at the nominal interaction point (z = 0). ^b For the fallback solution the pixel size is about a factor two larger ($O(30 \times 30) \, \mu m^2$).



[1] ALICE-PUBLIC-2018-013



	16.0/16.5			
Layer 0	Layer 1	Layer 2		
18.0	24.0	30.0		
270	270	270		
±2.5	±2.3	±2.0		
305	408	508		
280×56.5	280×75.5	280×94		
	2			
	$O(15 \times 15)^{b}$			
or layers refers to tracks originating from a				

Please, note rather compact dimensions! State-of-the-art solutions are required!



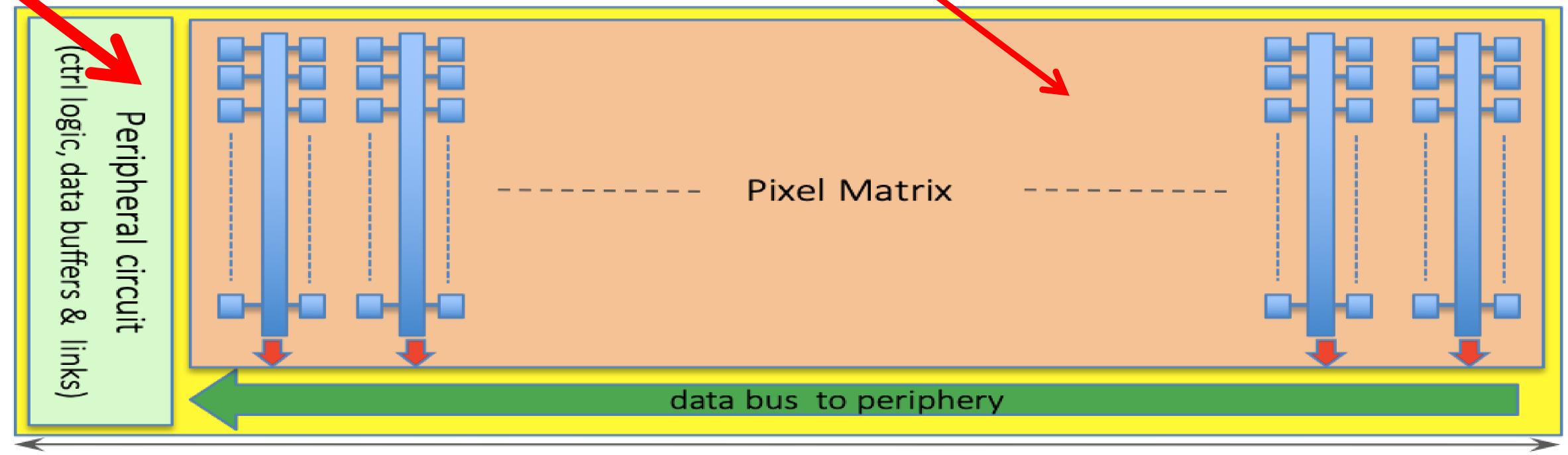






ITS-3 pixel sensor

power density below 140mWcm⁻²



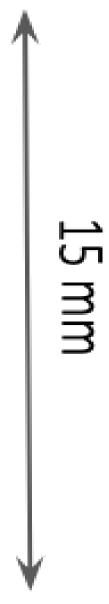
280 mm Diagram of stitched sensor in one direction (horizontal and vertical dimensions not to scale). Stitching in the vertical direction is also possible.[1] [1] ALICE-PUBLIC-2018-013



power density below **20mWcm⁻²**

• We consider the underpressure liquid cooling as the feasible solution for power density **140mWcm⁻²** • Low-speed cold gas is proposed to cool large are thin pixel sensors in order to minimize vibrations







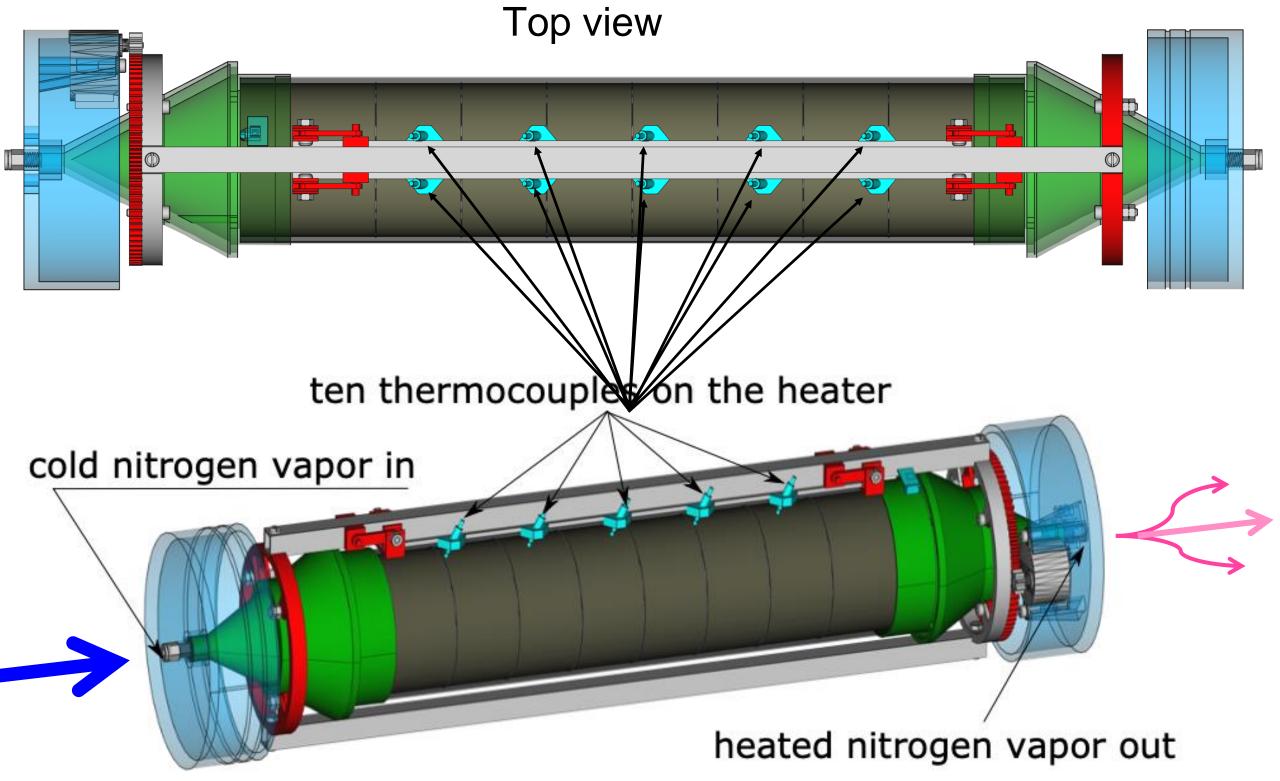
Low-speed gas cooling of large area thin pixel sensors

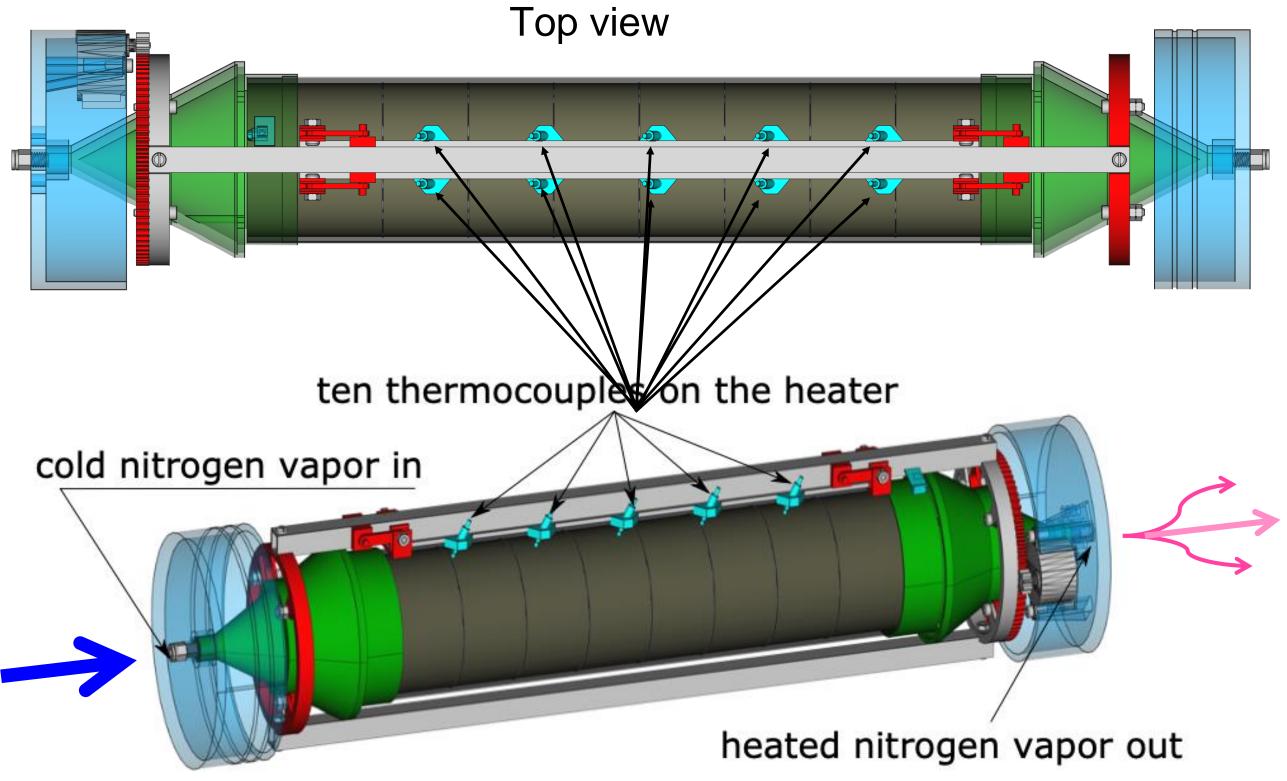


one layer

Cooling system with thermally well isolated barrel Cylinder Layer 2 mock-up





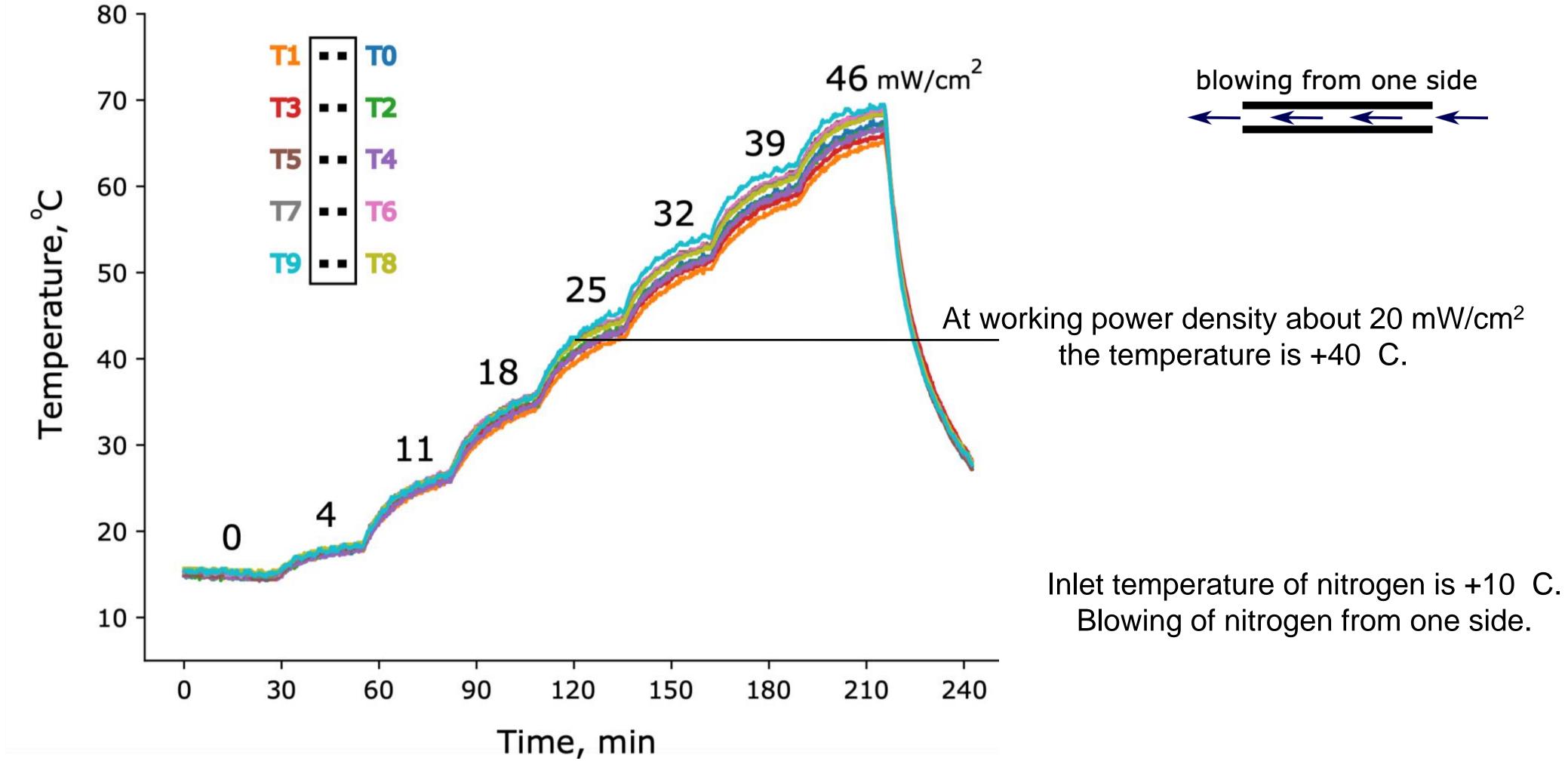


Left: Photo of thermally isolated barrel. Right: a scheme of cylinder Layer 2 mock-up with heater and nitrogen cooling system.

G.A. Feofilov, V.I. Zherebchevsky, et al. **G.Feofilov** ITS3 -WP5 (25_January 2022) · Indico (cern.ch)

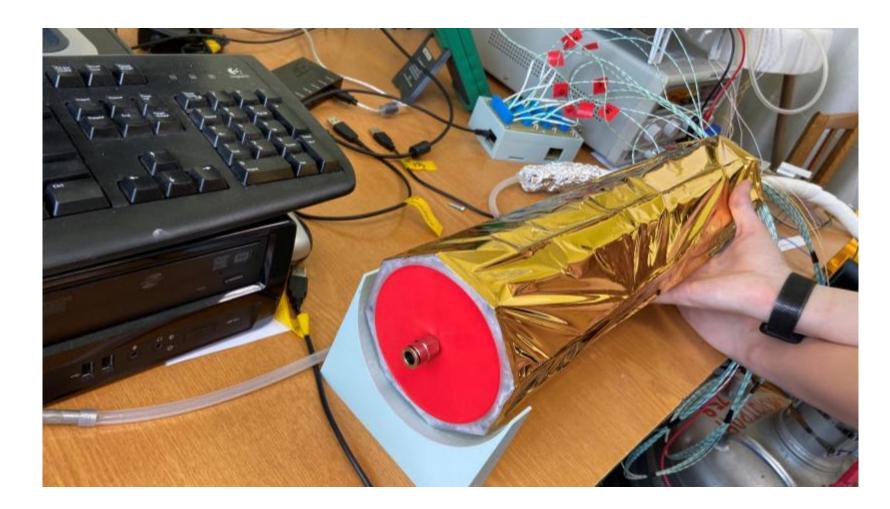


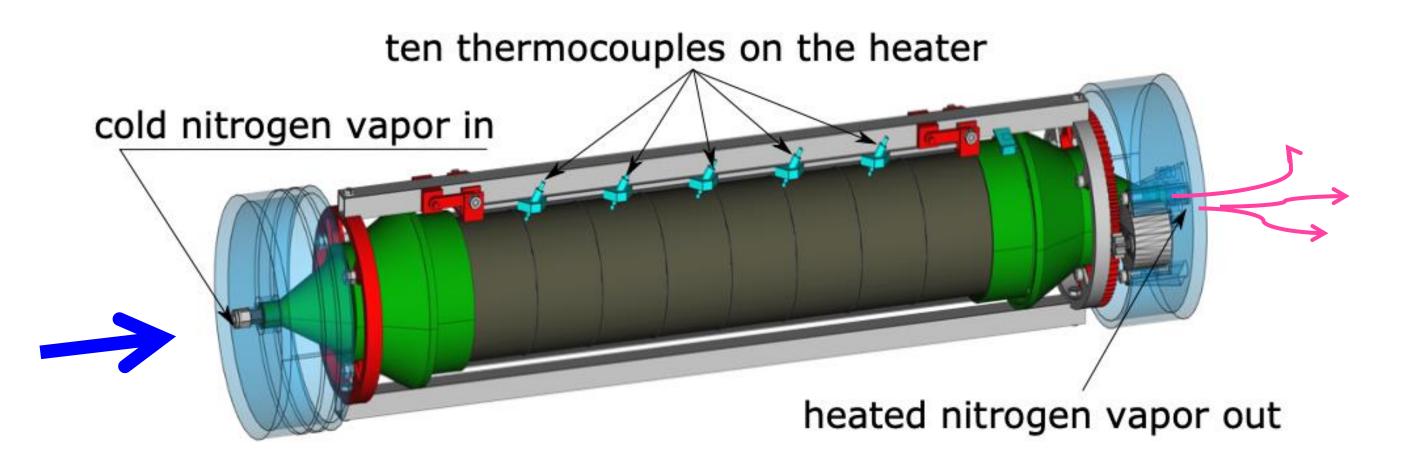
Temperature distributions measured at different heat loads with blowing from one side



120	150	180	210	240
e, m	in			

Space blanket instead of thermally well isolated barrel. Cylinder Layer 2 mock-up







Left: photo of self-supported outer shell with space blanket. It consists of low-weight rigid CF bars, space blanket and plastic support ring.

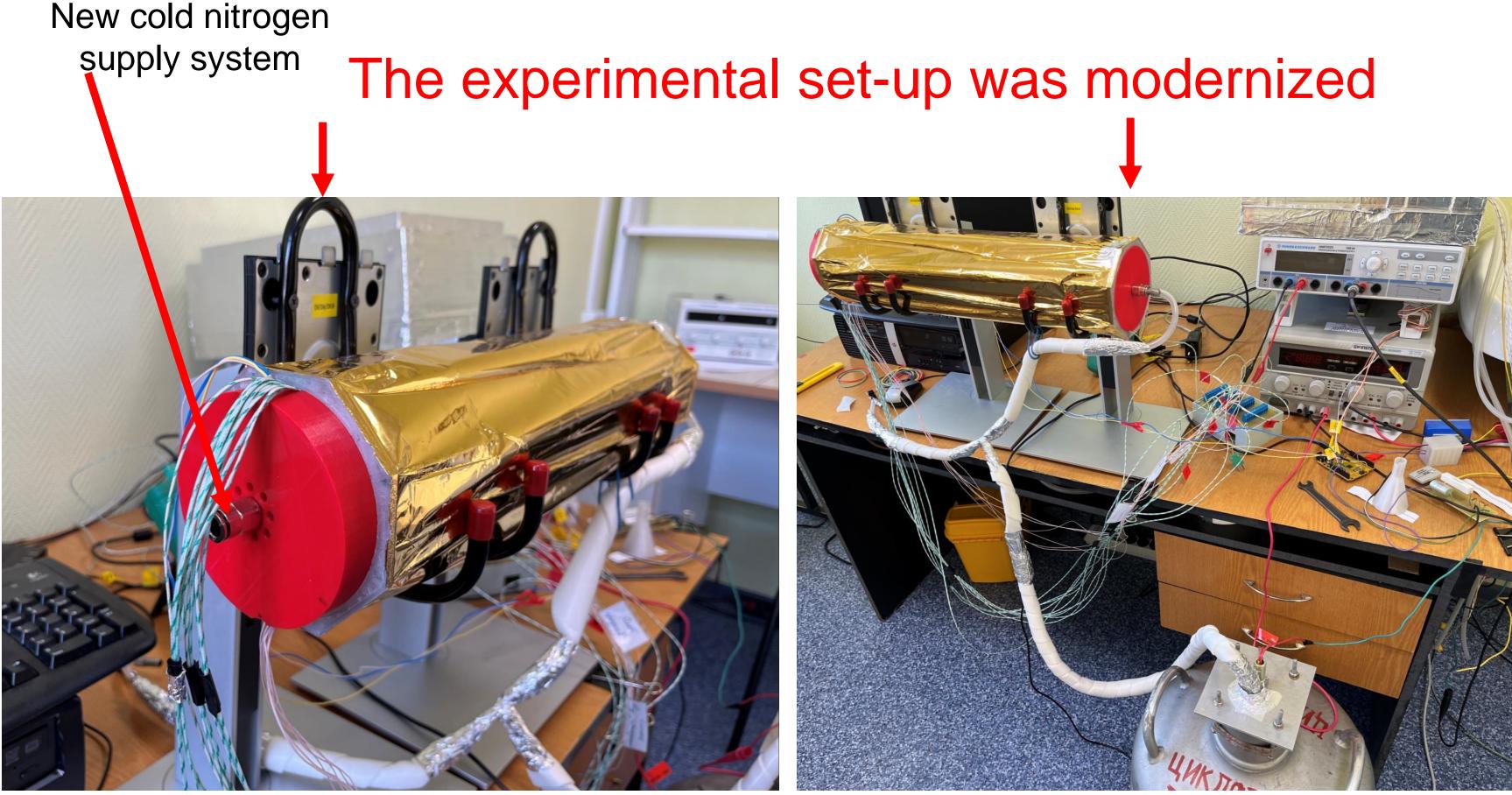
> Right: a scheme of cylinder Layer 2 mock-up with heater and nitrogen cooling system

V.I. Zherebchevsky, G.A. Feofilov et al. **ITS3 Upgrade WP5 (Mechanics and Cooling) meeting 06.07.2021**



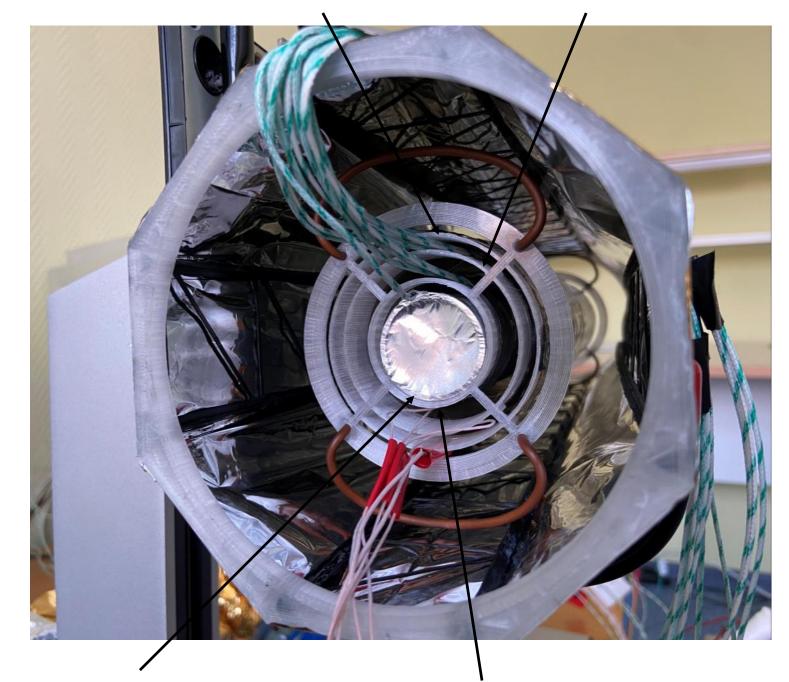


Space blanket and 3 cylinder layers with heaters



3d layer

2nd layer

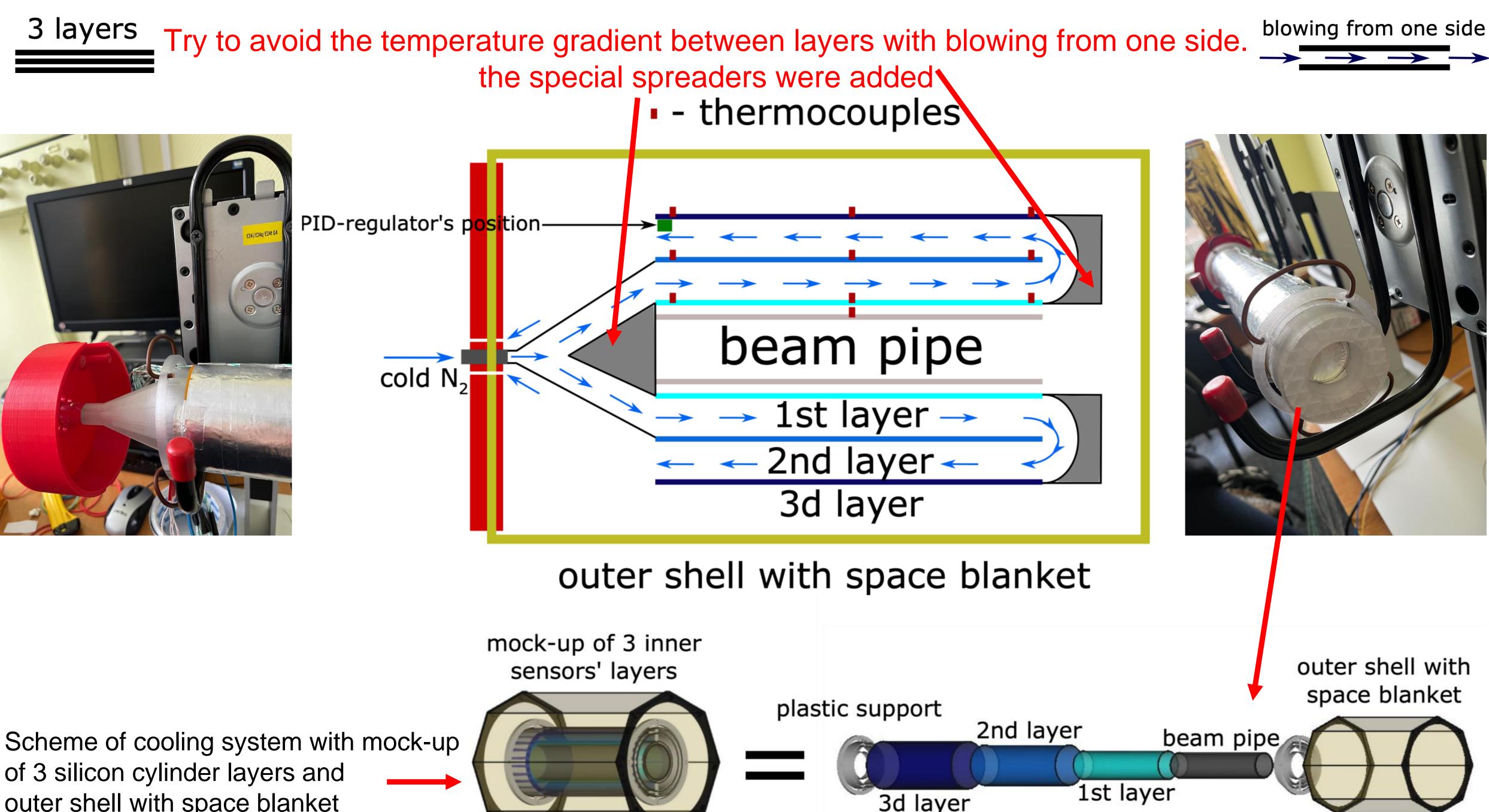


beam pipe

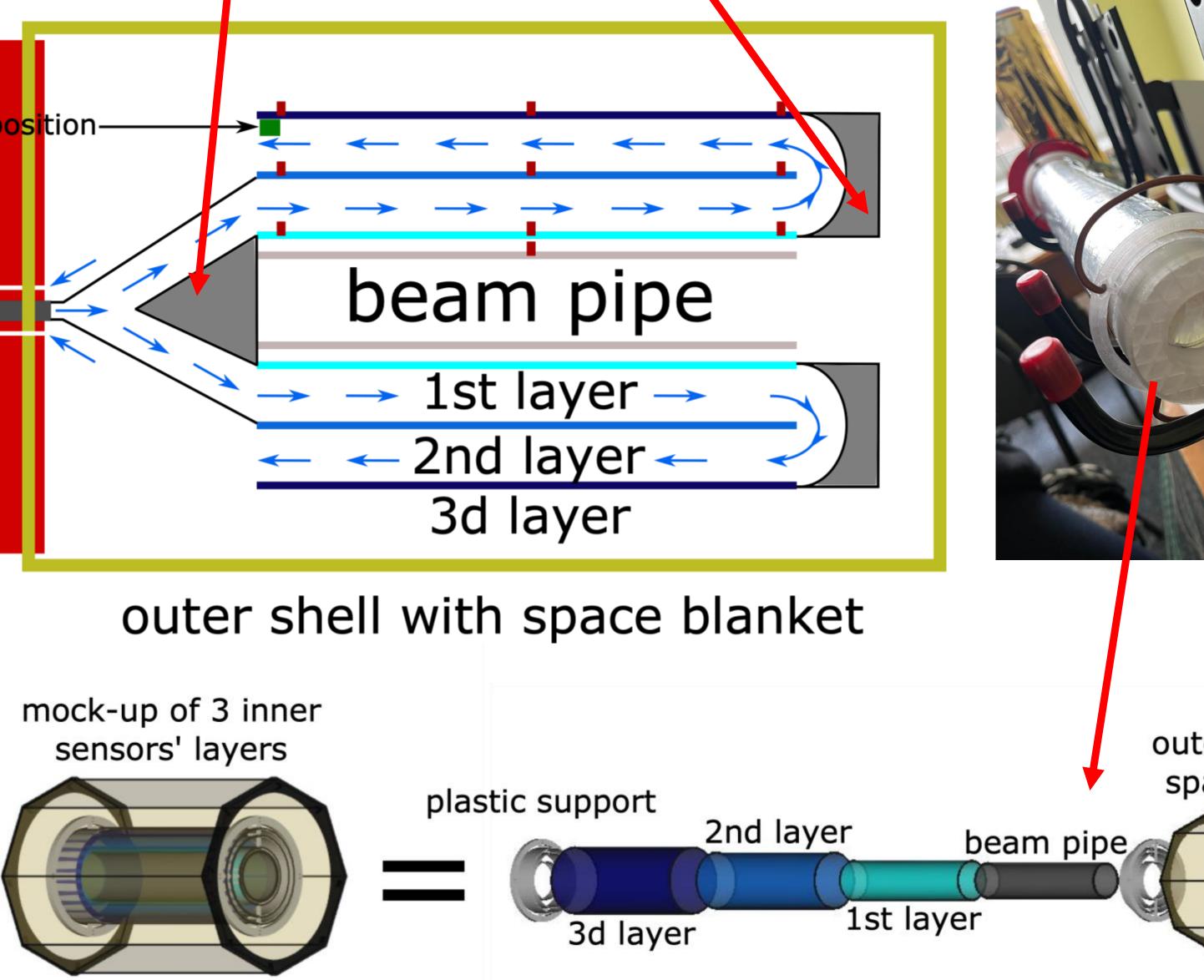
1st layer

Left: a general view Right: view of the interior of the space blanket shell

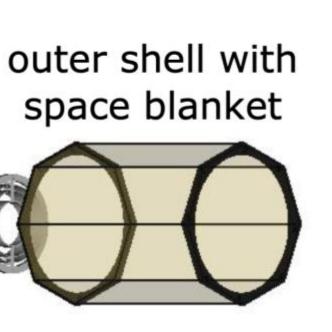




outer shell with space blanket





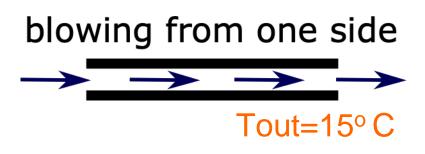




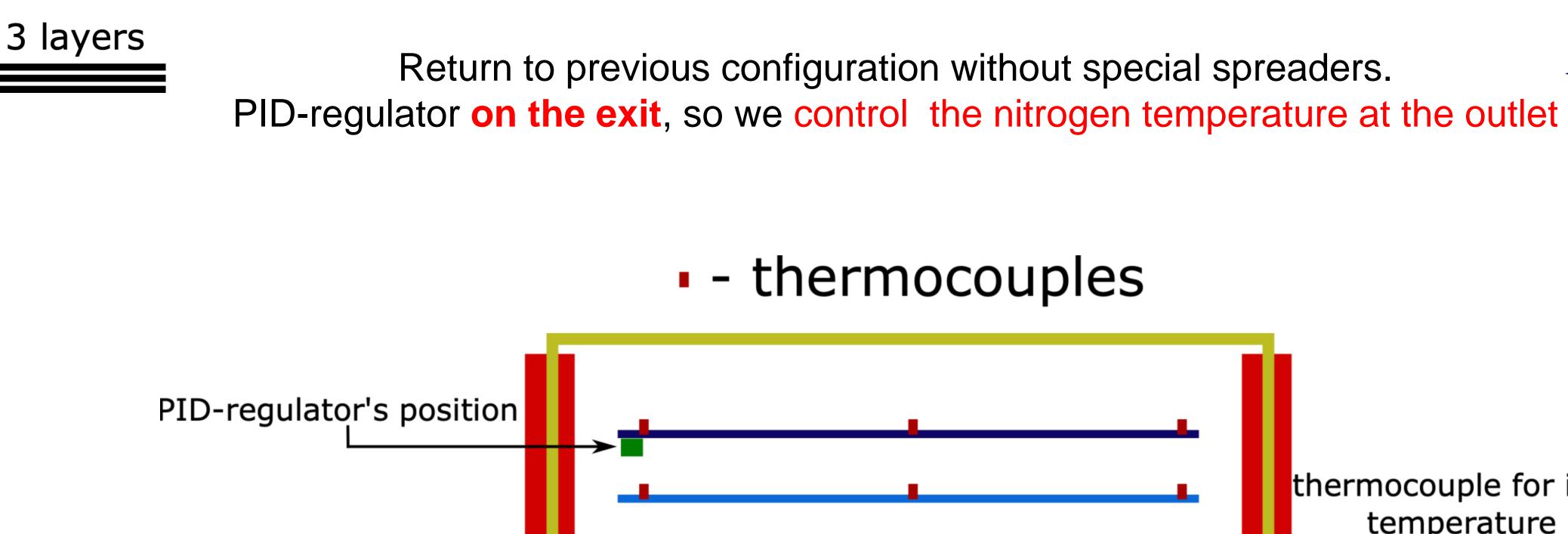


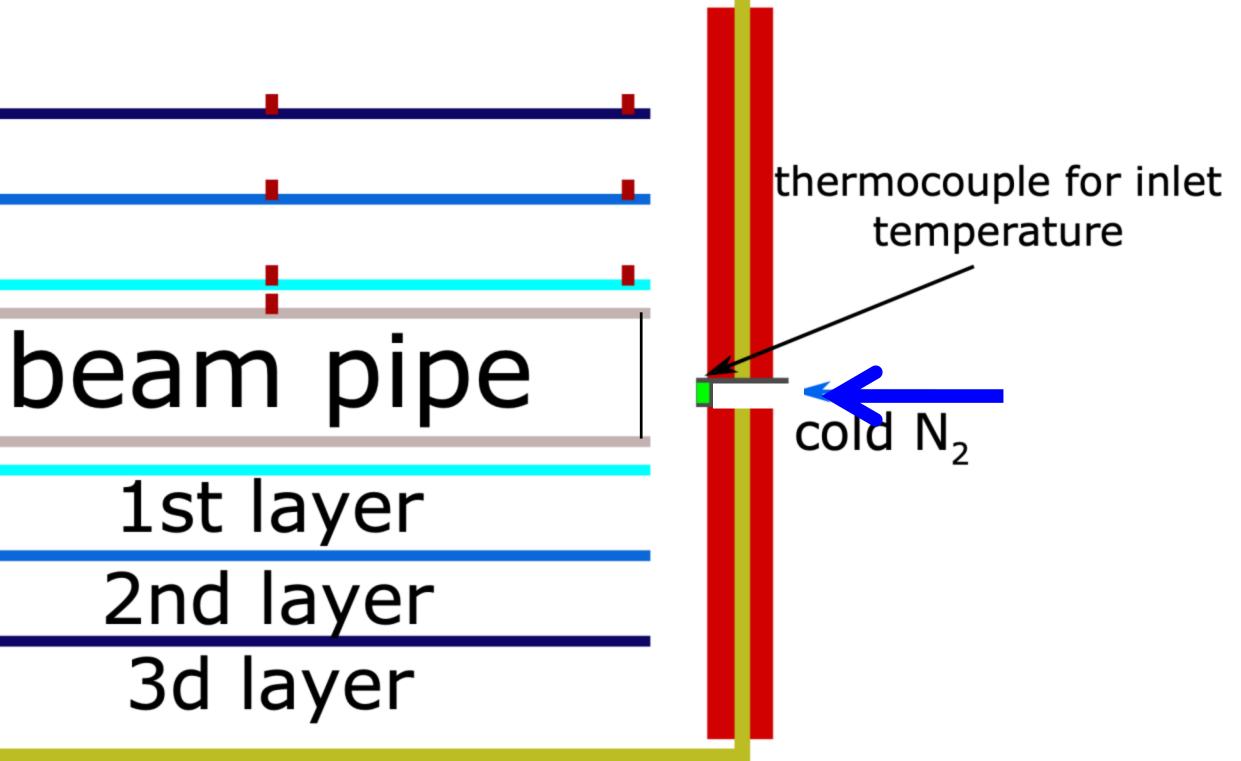
Results of temperature distributions measured at different heat loads with nitrogen cooling from one side with special spreaders - at this moment are not quite conclusive due to the position of the PID-regulator: it was found to be sensitive both to the input and outlet temperatures position to the exit.

Work is in progerss



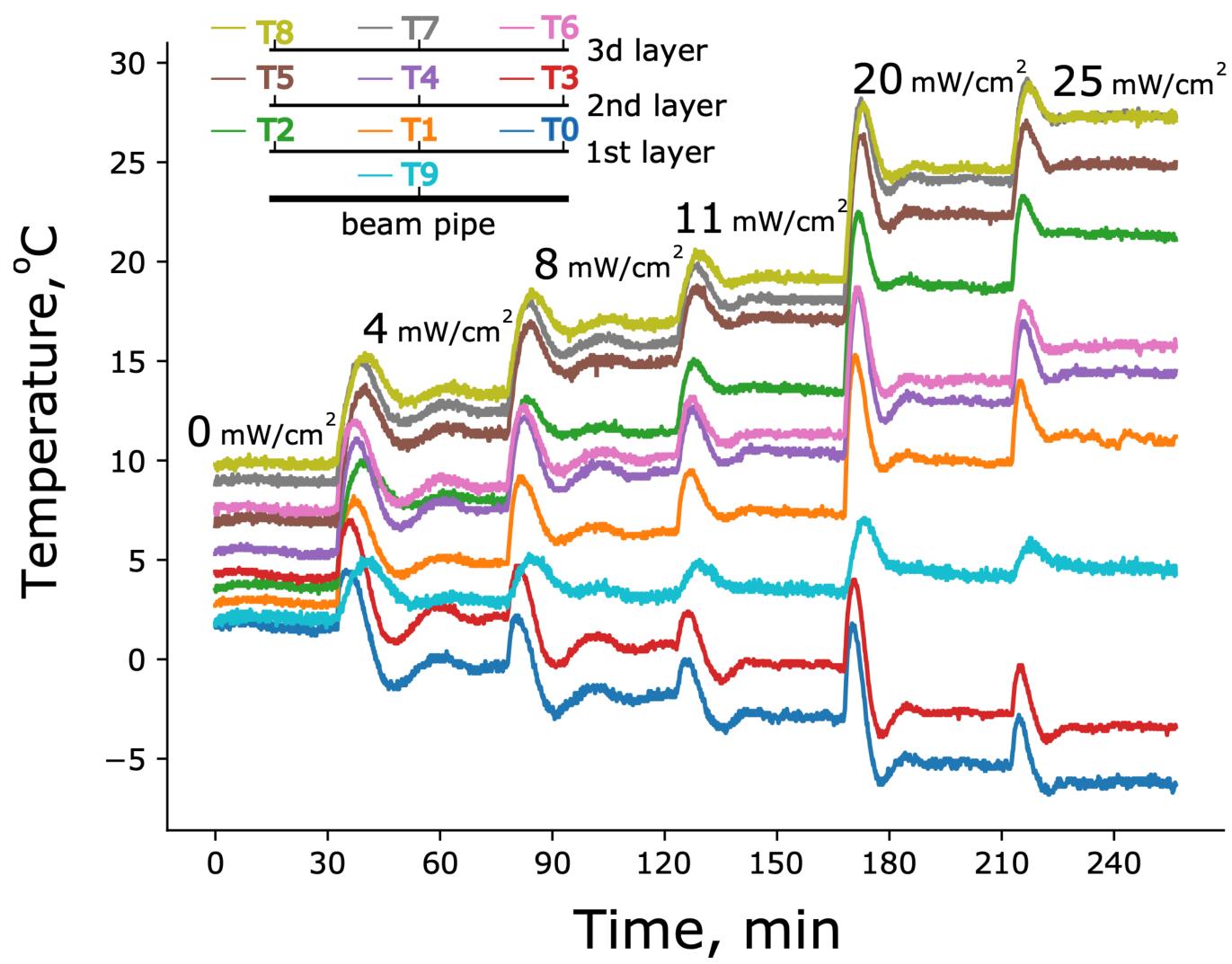


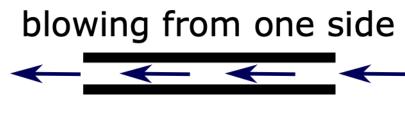




outer shell with space blanket

blowing from one side





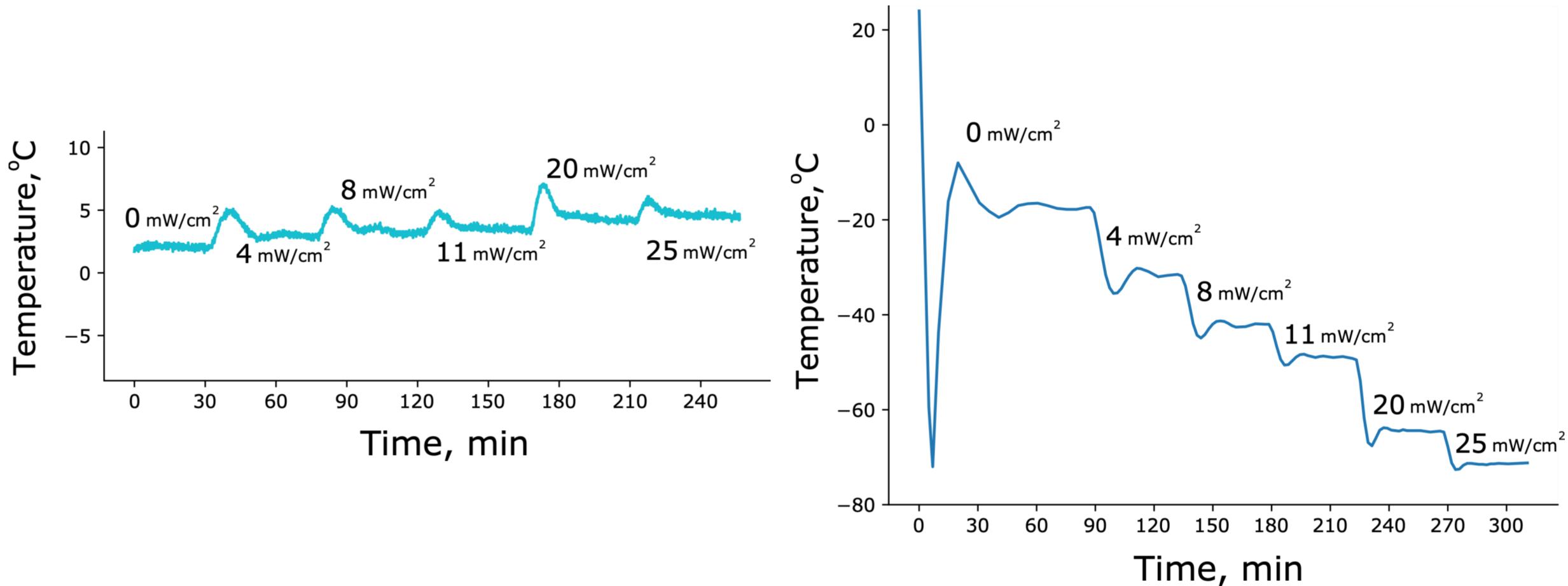
Results of temperature distributions measured at different heat loads with nitrogen cooling from one side and control of the outlet temperature

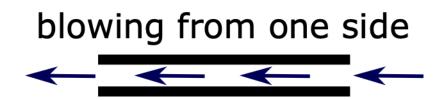
Outlet temperature of nitrogen is 15 °C Blowing of nitrogen from one side. Density of power is the same for each layer.

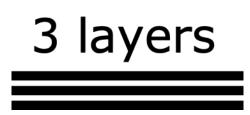




Temperatures vs. different heat loads: on a **beam pipe**(left) and **inlet nitrogen**(right)

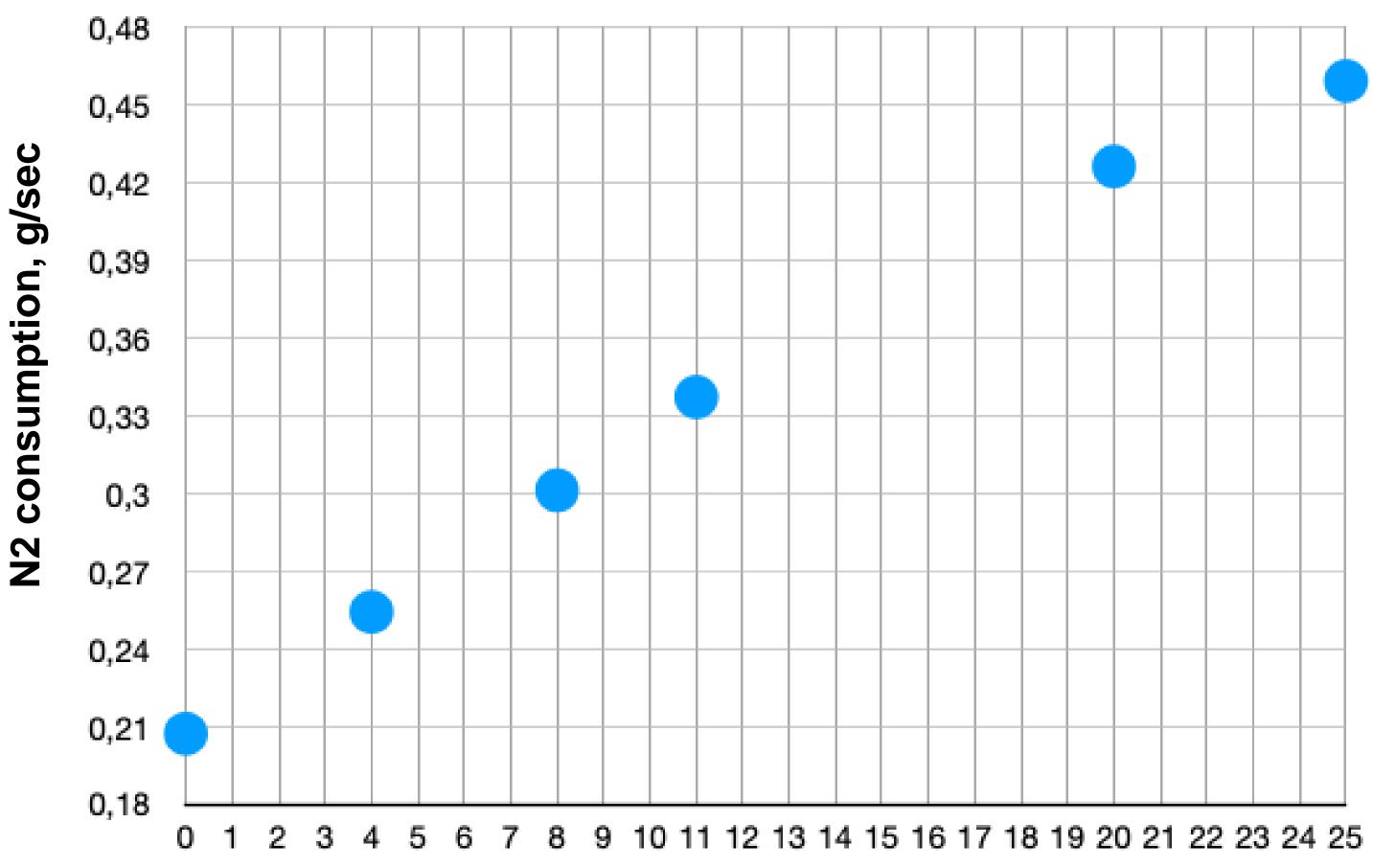


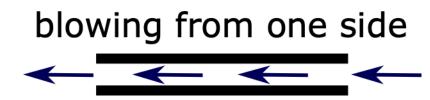




Nitrogen consumption

dependence of nitrogen consumption on density of power on mock-up of 3 silicon cylinder layers \succ Nitrogen consumption on 25 mW/cm² is ~1.6r kg in hour.



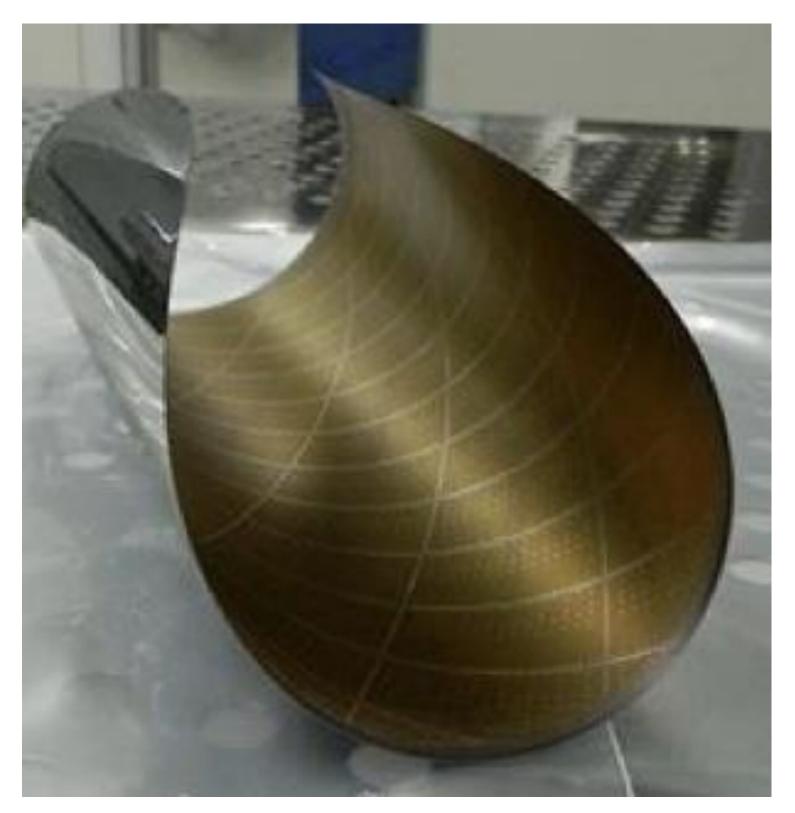


Power Density, mW/cm^2

Ultraligthweight self-supported mechanics for ALICE ITS-3 modules

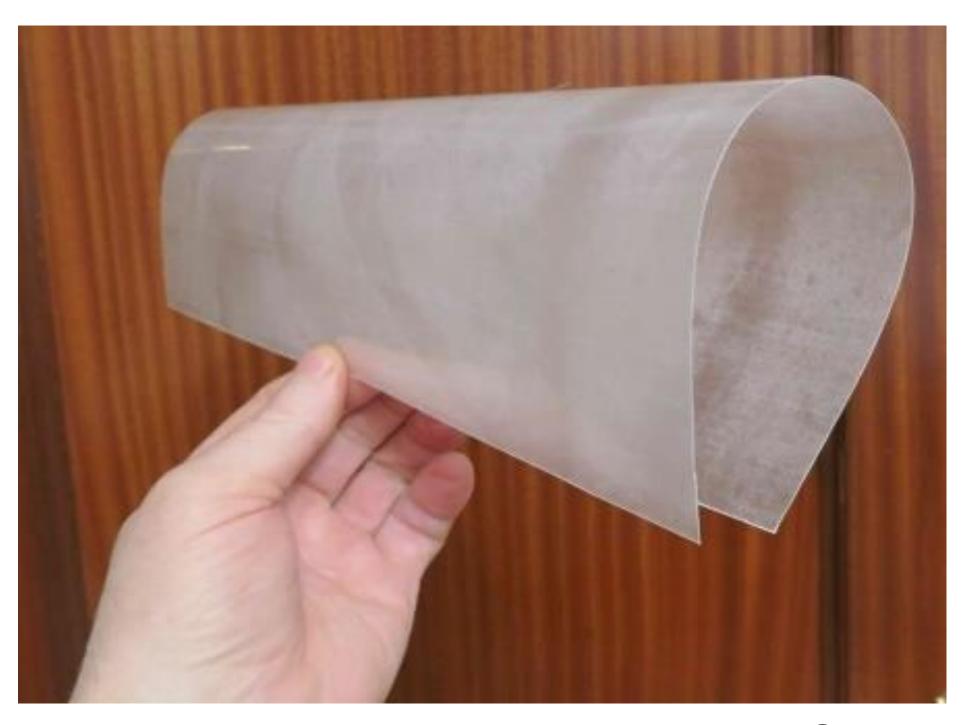


Glass-fiber plate



Bent Si-plate CTE Si = 5.1×10^{-6}

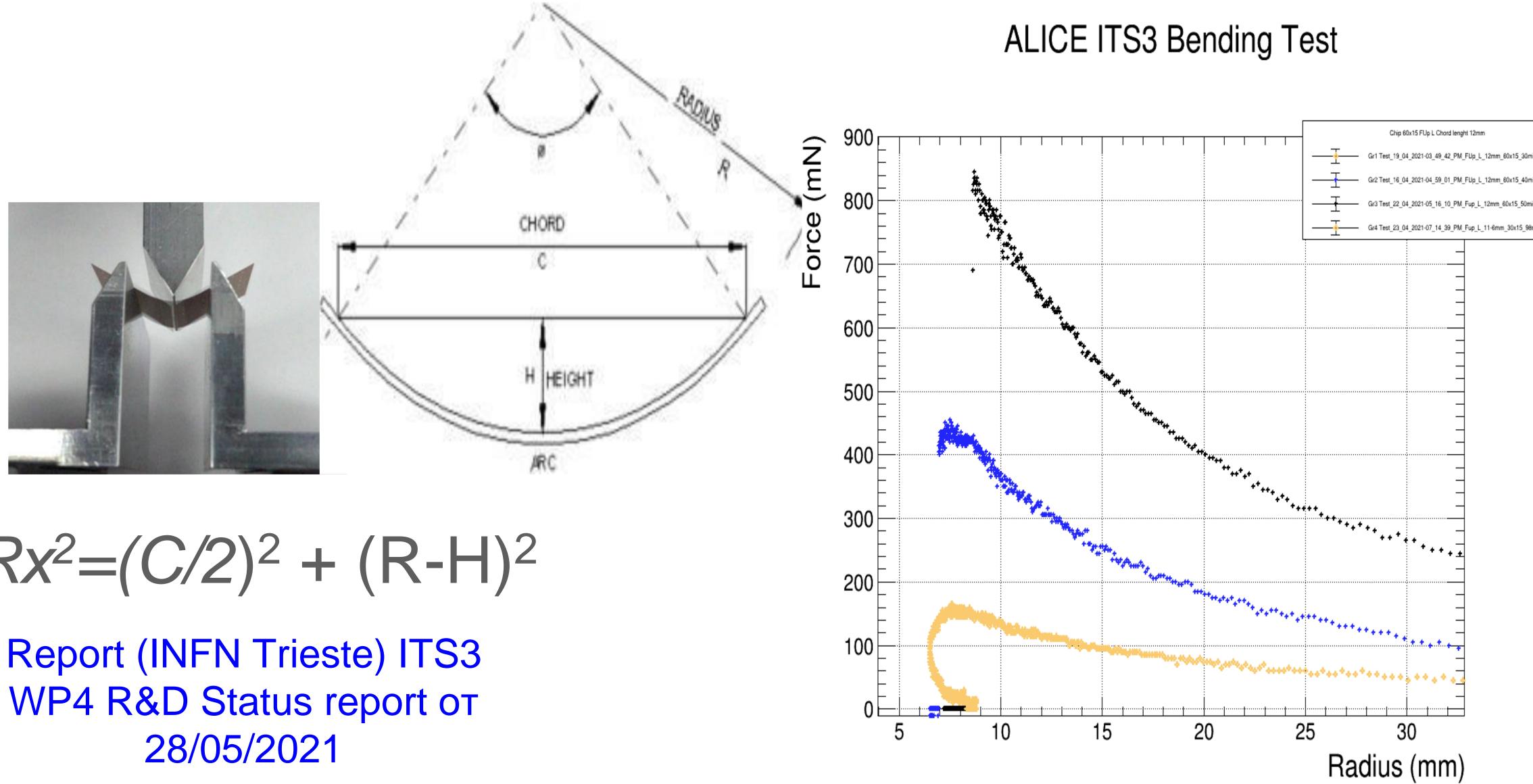




Glass-fiber plate 310 x 240 mm² CTE Glass-fiber = 5.1×10^{-6} Fiber type: А-1-ТД СТТ Thickness of fiber $= 0,1^{+0.01}$ MM Density of fiber = $110g/m^2$ Density of glass-fiber material= $0,0183r/cm^{2}$.

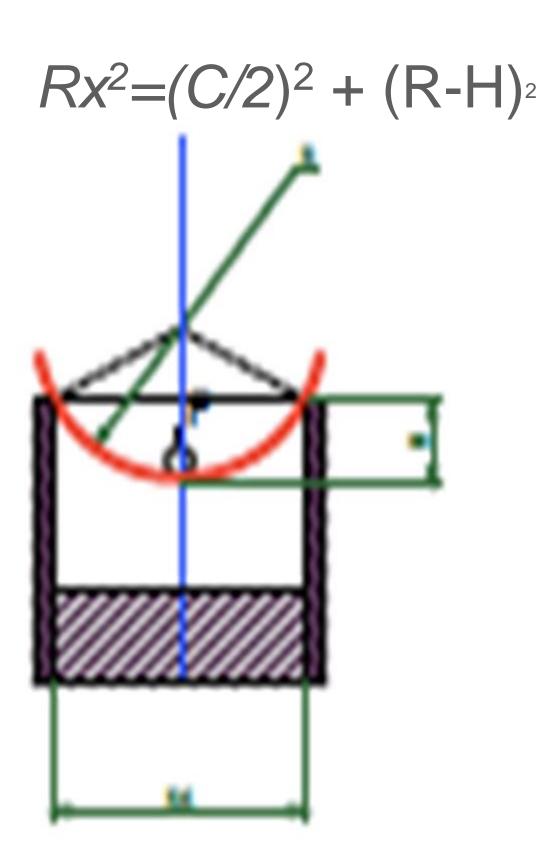


ALICE bending tests of Si-plates 30,40, 50 µ



$Rx^2 = (C/2)^2 + (R-H)^2$





C=28mm

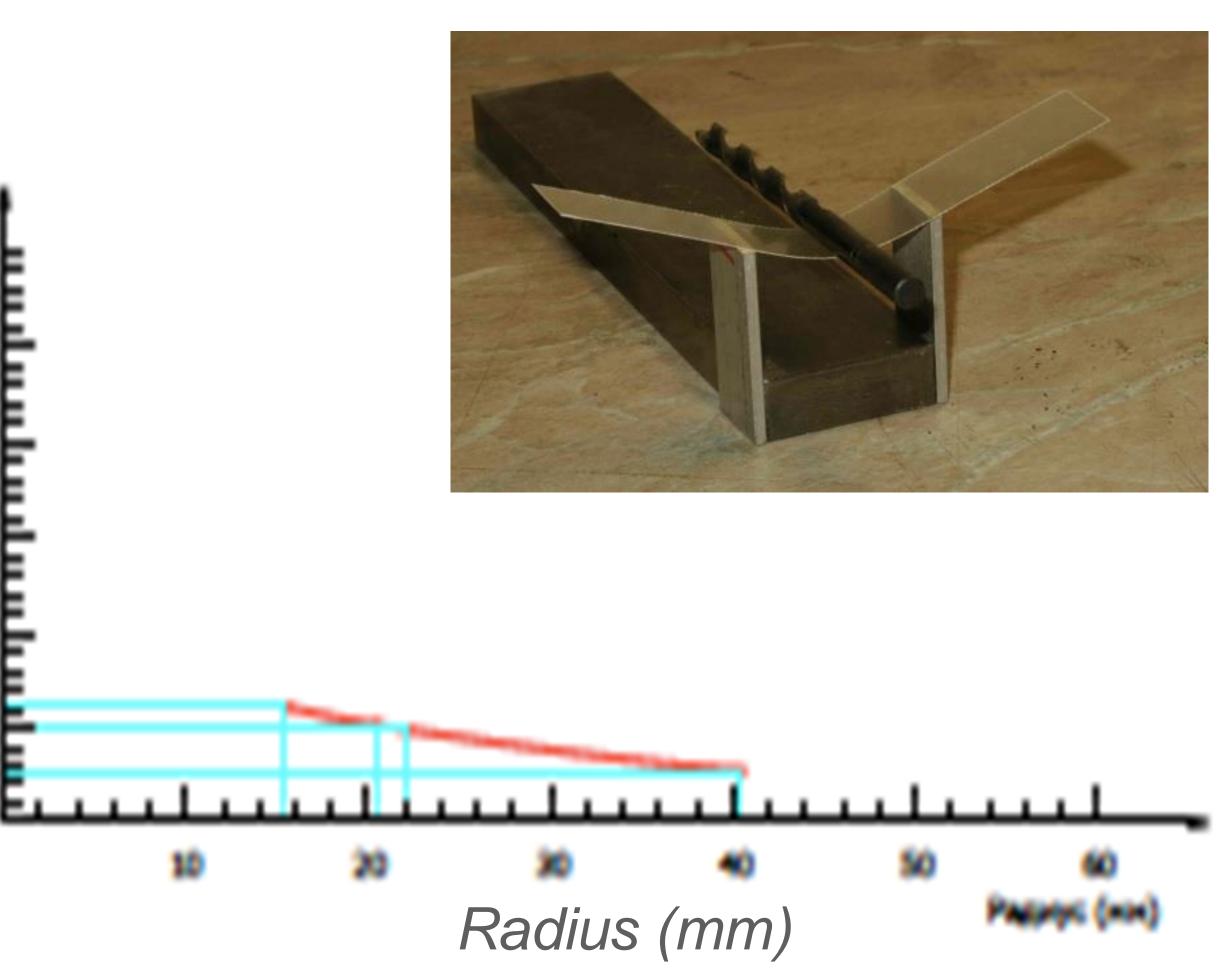
Load, g.	Sage, mm	R мм (plotted)
12,25	9	15,4
10,25	5,5	20,5
9,8	5	22,1
5,4	2,5	40,5



200

100

Bending tests of fiber-glass plate 100µ



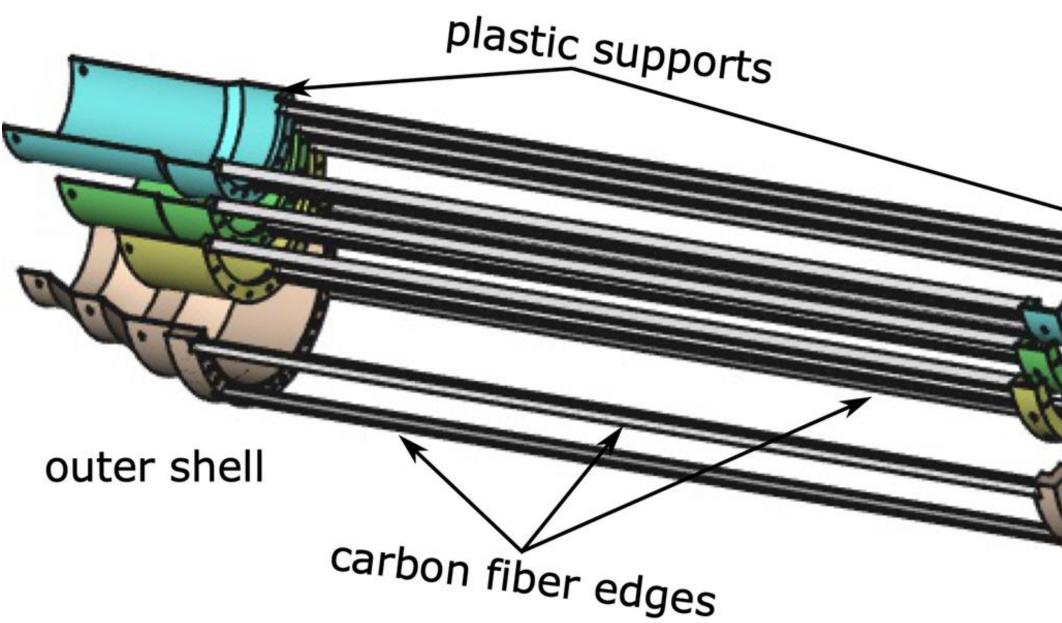
The stifness is estimated to be similar to $\sim 30\mu$ Si







Scheme of carbon fiber support structure for inner tracking system



V-shaped ribs

Length of carbon fiber V-shaped ribs for layers: 284mm

Length of carbon fiber V-shaped ribs for outer shell: 297mm

Radius of layers: layer 1 - 30mm, layer 2 - 24 mm, layer 3 - 18mm

1st layer 2nd layer 3d layer	Part of support Structure	Weight, g
	carbon fiber edge	2
	plastic supports for 1st layer	9,5
	plastic supports for 2nd layer	10
	plastic supports for 3d layer	9,5
	plastic supports for outer shell	24,9

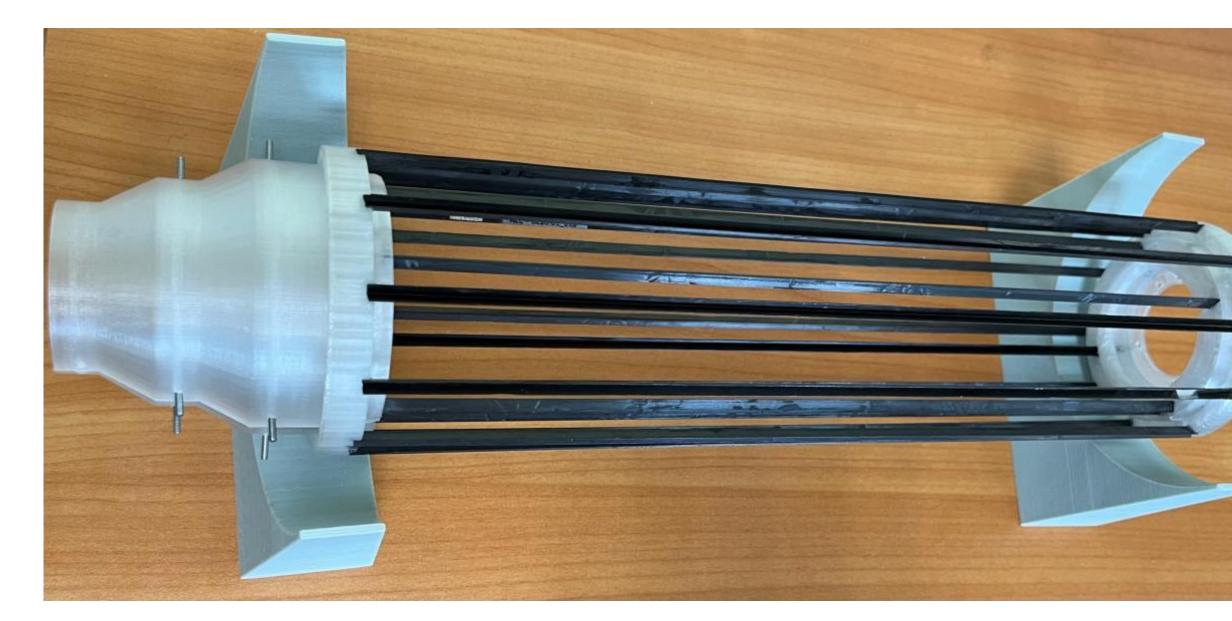


From scheme to real life

Outer shell



Outer shell + 3d layer



Outer shell + space blanket + 3d layer

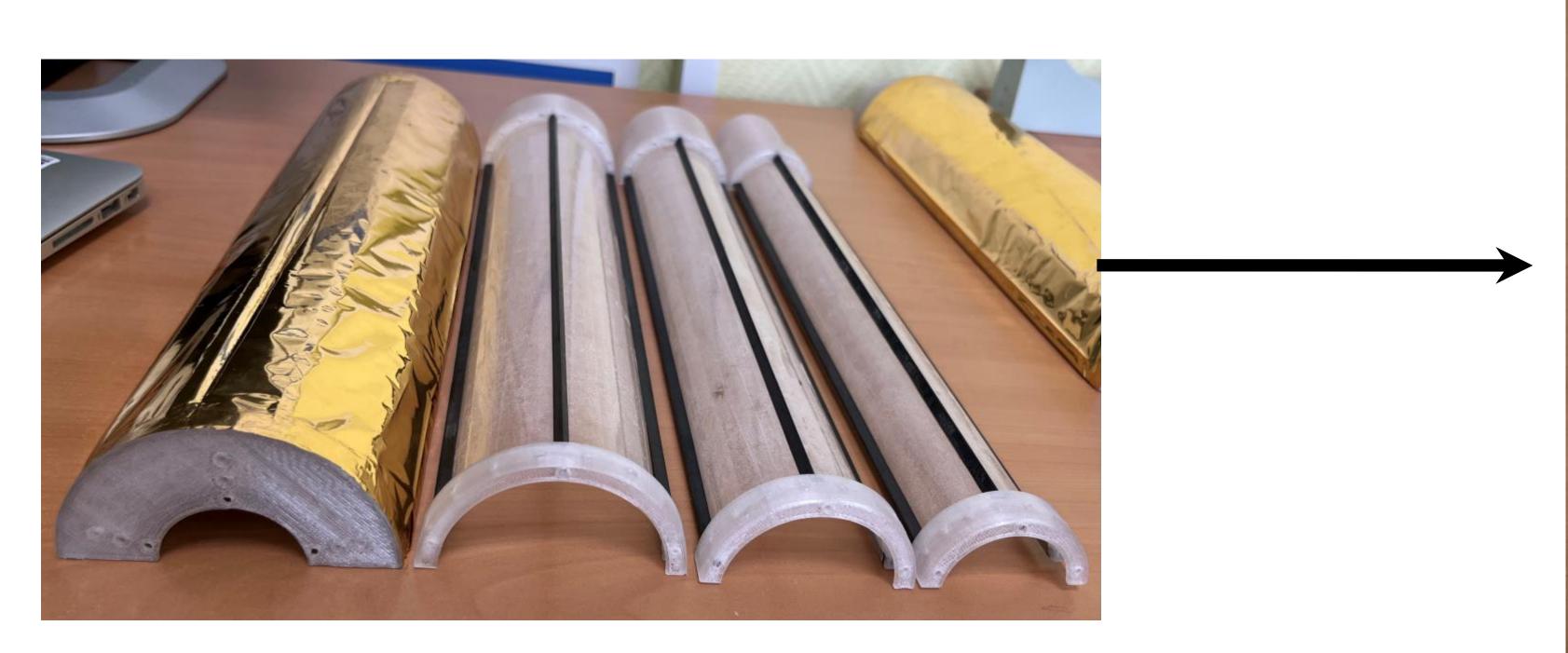






Assembling of mechanical mock-up of ITS-3

We use a fiberglass as a mock-up of silicon thin sensors

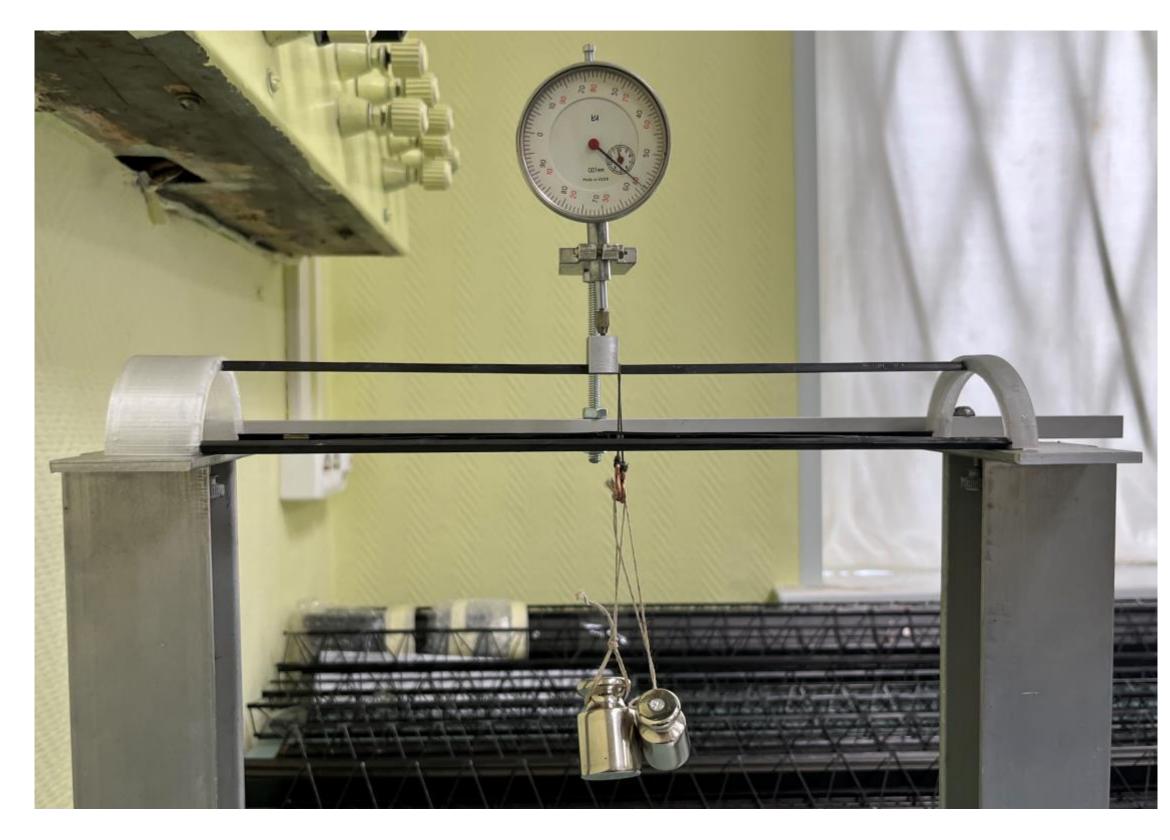




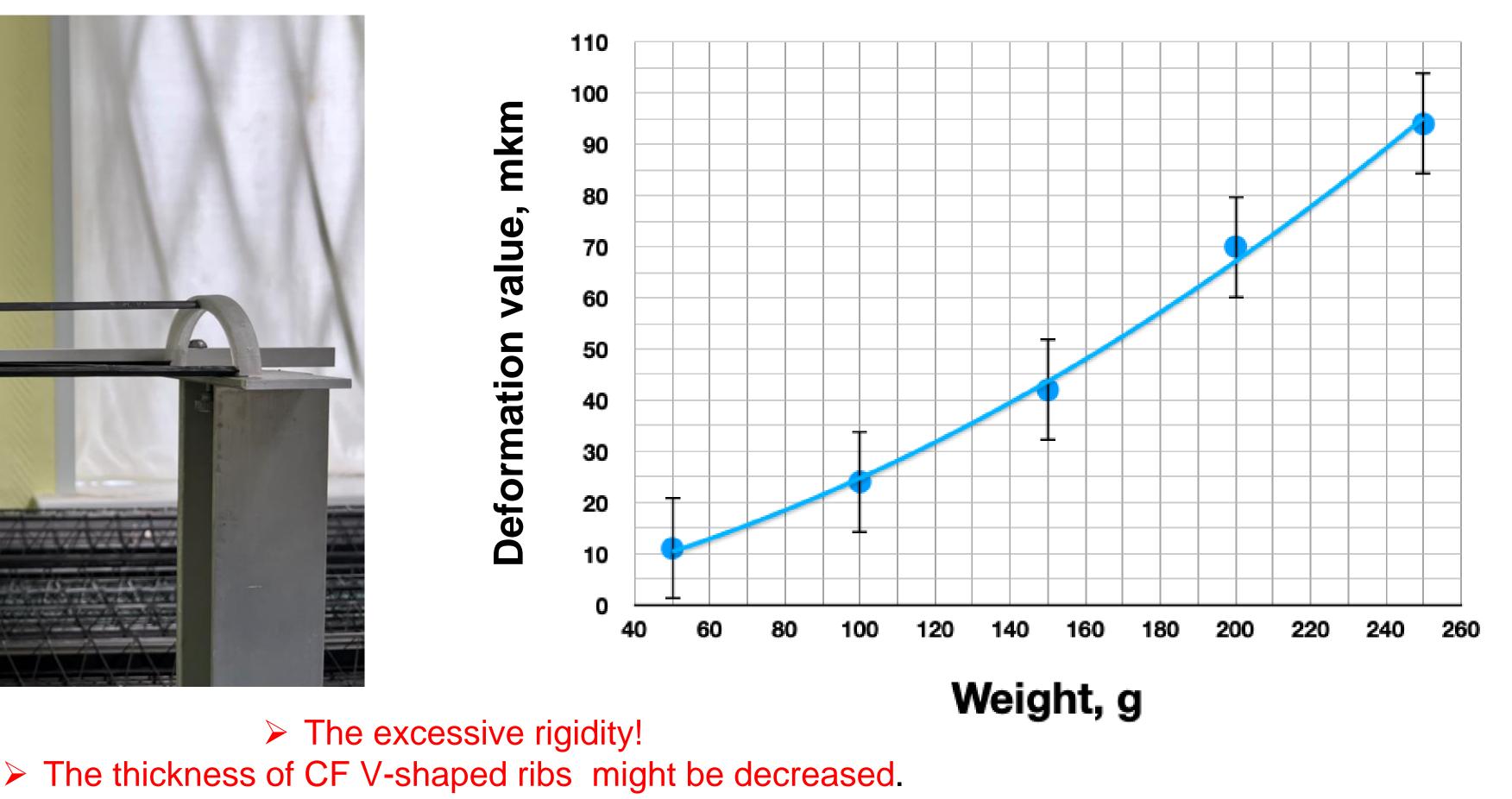


Deformation value of V-shaped CF ribs under the influence of different weights

Measurements were performed by micrometer

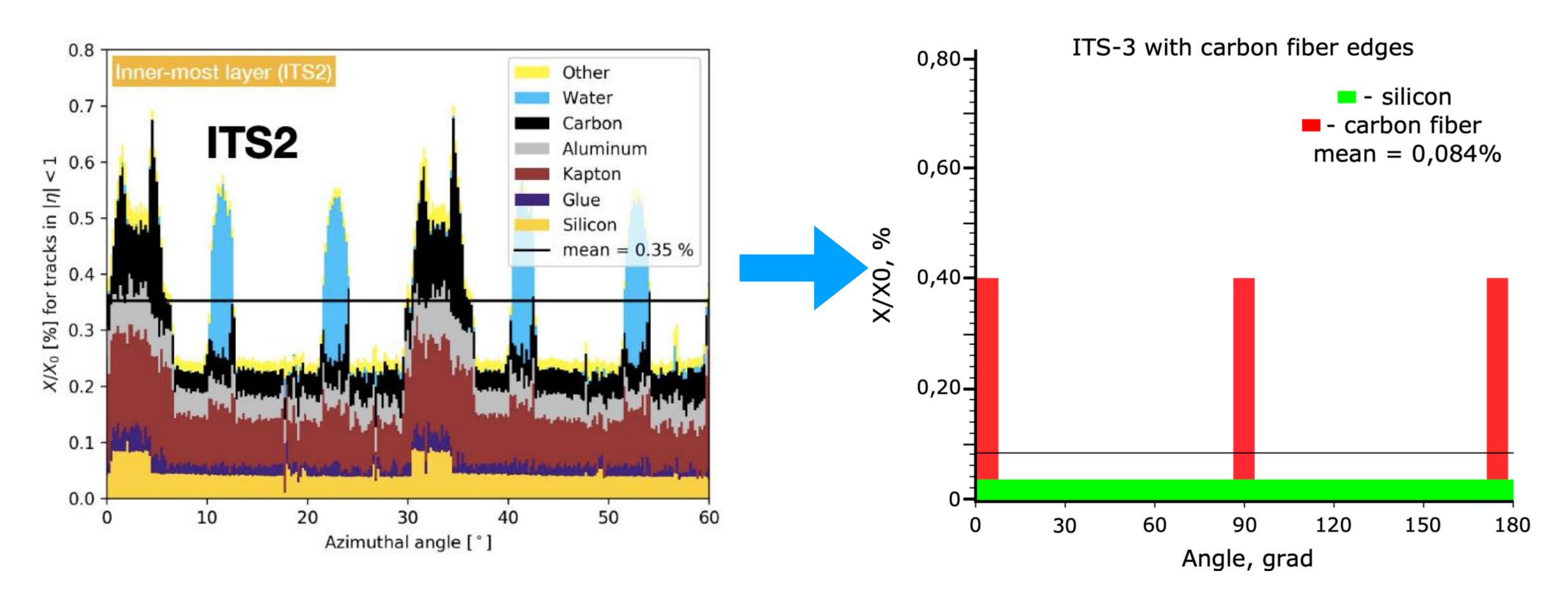








Radiation transparent



The excessive rigidity!
The thickness of CF V-shaped ribs might be decreased



Conclusions

> The performance of low-speed gas cooling of the full scale mock-up of three layers of the ITS 3 was demonstrated > The space blanket instead of thermally well isolated barrel works perfectly well (no any signs of water vapor condensation were noticed)

the ITS 1 and ITS 2 carbon fiber technology > The assembly/disasembly will not require gluing/ungluing > All layers and the outer space blanket shell are self supported be further decreased

> Ultralighweight self-supported mechanics for ALICE ITS-3 modules is based on

 \succ It may allow to use the individual operations with bent modules of LO, L1 and L2

> Thickness of CF V-shaped ribs holding the bent-Silicon and the outer shell might



BACK-UP SLIDES