



Superconducting Detector Magnet Workshop, Sept 12, 2022

Development of the Large Ultra-Thin HTS Magnet System for the AMS-100 Experiment in Space

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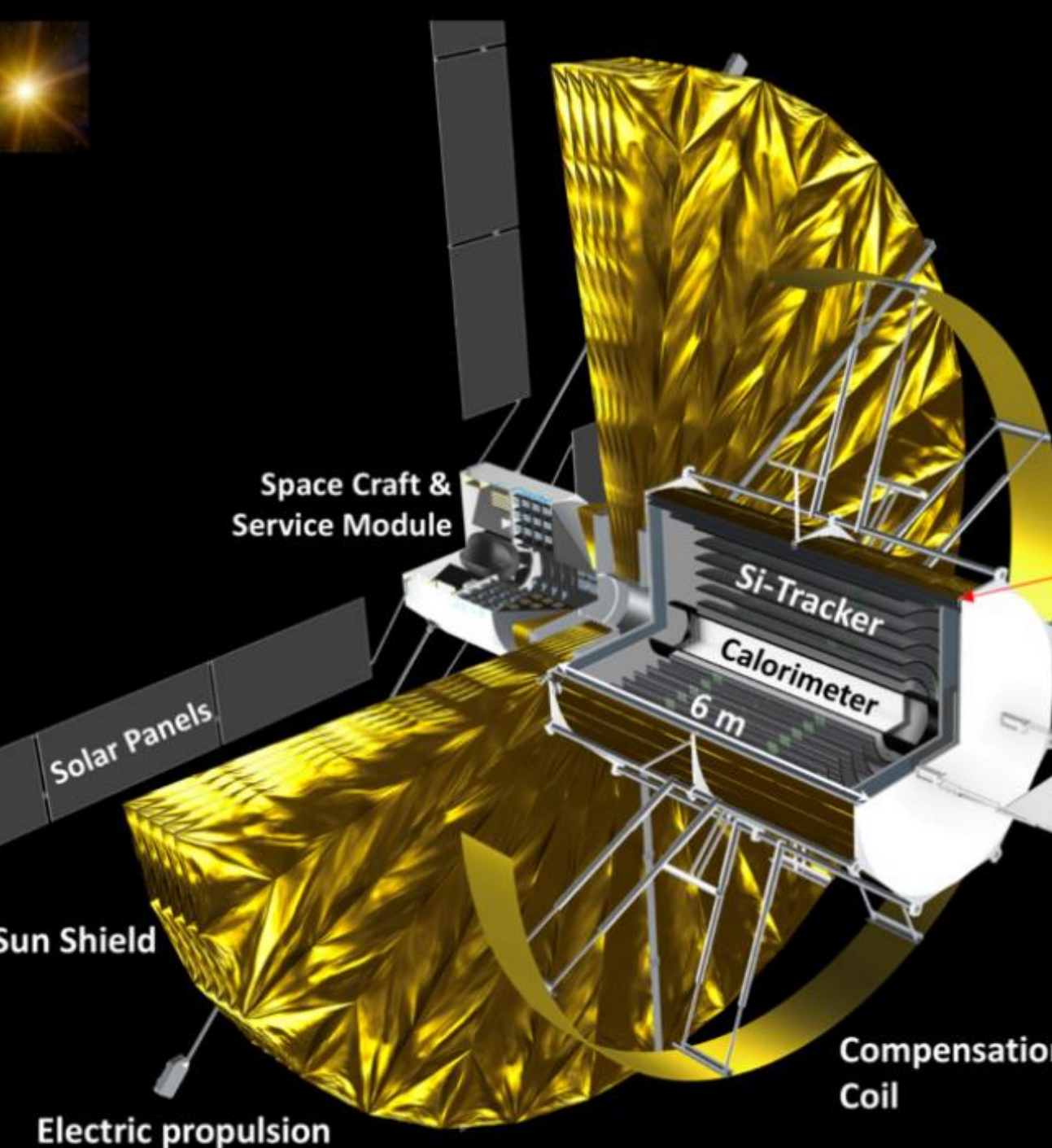
EPFL



**UNIVERSITÉ
DE GENÈVE**

AMS-100 A Magnetic Spectrometer – Successor of AMS-02

The **Alpha Magnetic Spectrometer**
on the International Space Station



AMS-100: Overview

- Magnetic Spectrometer to be send to Lagrange Point 2 (1.5 Mkm from earth).
- Probing high energy cosmic rays, in particular anti-protons and anti-deuterons.
- Geometric acceptance of $100 \text{ m}^2 \text{ sr}$.
- Main magnet is a 6 m long, 4 m diameter HTS ultra-thin solenoid.
- No active cooling, only passive cooling using radiators.
- Electrical/thermal/mechanical challenge.
- Compensation coil needed to correct magnetic torque during operation.
- Comp. coil is a 1.5 m long, 8 m diameter HTS solenoid.

The Expedition to Lagrange Point 2

Vehicle and Launch:

- Target launch year: 2039.
- Operational for 10+ years.
- Total estimated mass of AMS-100: 40 Tons¹
 - ~4 Tons for the magnet system,
 - ~16 Tons of detector equipment,
 - ~20 Tons of auxiliary equipment and cabling.
- Launched with SpaceX's Starship rocket.

AMS Pathfinder mission:

- First radiation cooled HTS magnet in space of such size.
- Test the operation at L2.
- Controls, radiation cooling etc.

¹ AMS-100: The Next Generation Magnetic Spectrometer in Space – An International Science Platform for Physics and Astrophysics at Lagrange Point 2, S. Schael et al., Nucl. Instrum. Methods Phys. Res. A: Accel. Spectrom. Detect. Assoc. Equip., vol. 944, 2019



AMS-100: Design Challenges

- Radiation transparency is important -> thin ~3 mm thick aluminium stabilized conductor. The coil pack needs to provide mechanical stability for the magnet system. [Major mechanical challenge](#).
- Coil needs to survive stresses caused by launch, cool down and magnet powering.
- The AMS-100 magnet system will have a large stored energy of approximately 14 MJ -> 9 kJ/kg. HTS materials are very stable and are difficult to quench. However, they also have the downside that HTS magnets are also difficult to protect, mainly due to the low NZPV.
- Controlled resistance coil -> turns are shorted with a controlled resistance. High enough resistance to charge to coil in a timely fashion, but low enough to protect the magnet in case of a failure event. [Major electrical and thermal challenge](#).
- Magnetic torque needs to be compensated. Can be achieved with a compensation coil. However, given the diameter of the main solenoid, difficult to fit in the rocket's cargo bay.

AMS-100: A Magnetic Spectrometer

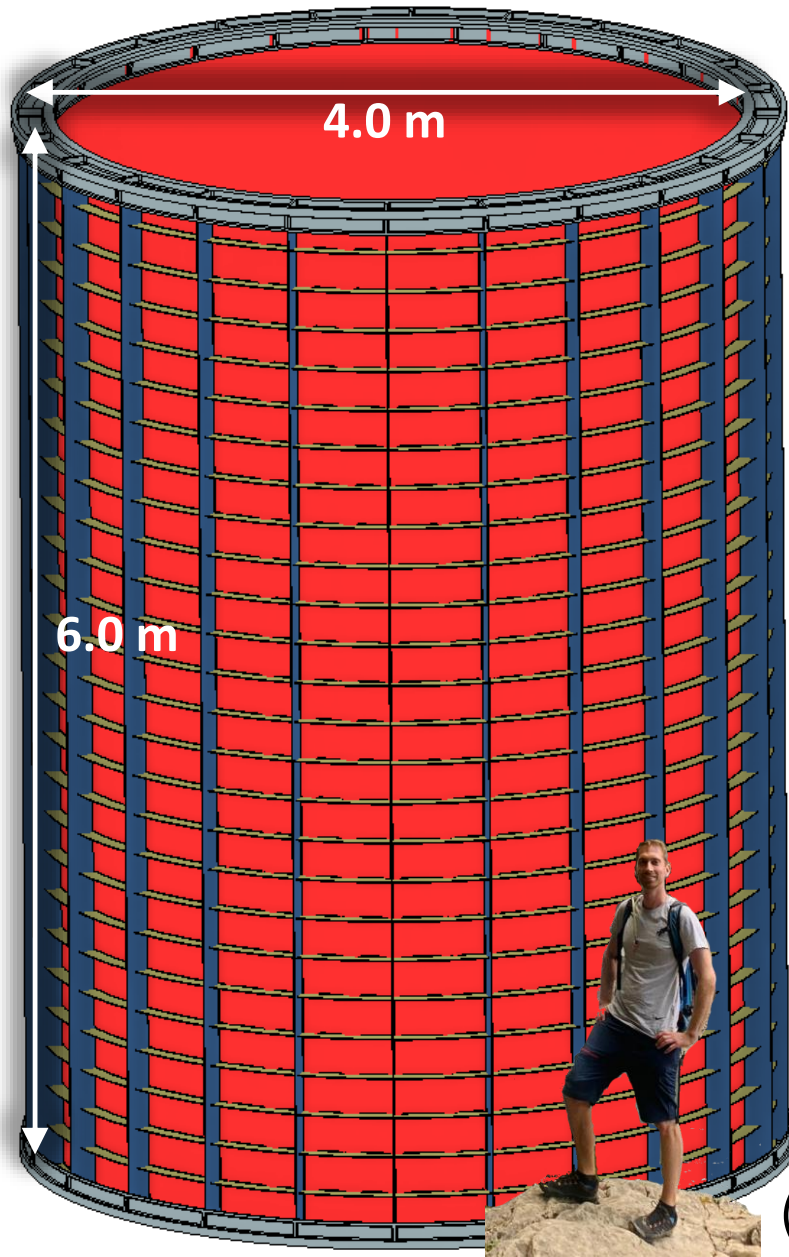


Table of properties for the AMS-100 main solenoid and compensation coil.

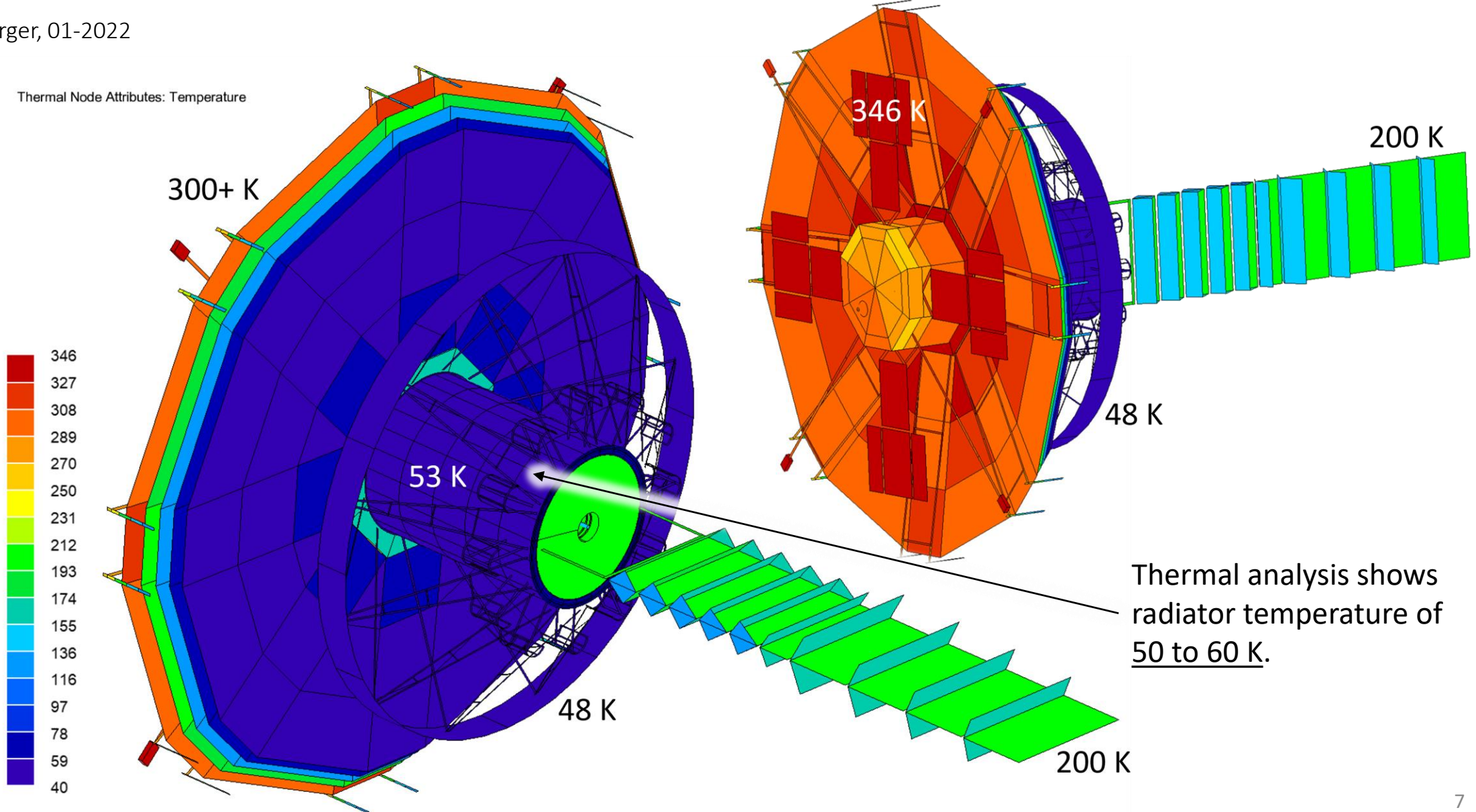
	<u>Main</u>	<u>Compensation</u>	<u>Combined</u>	Unit
Coil radius	2.0	4.0		m
Coil length	6.0	1.5		m
Tape width	12	12		mm
Stabilizer	Al-6063	Al-6063		
Cable thickness	2.85	2.85		mm
Cable width	16	16		mm
Layers	1	1		-
Turns	376	94		-
Inductance	286	114	287	mH
Number of tapes	18	18		-
Total tape length	85	43	128	km
Operating current	10.0	-10.0		kA
Cable mass	1090	545	1635	kg
Stored Energy	14.3	5.7	14.4	MJ
Energy Density*	14	11	9	kJ/kg

*Considering only the mass of the cable.

(me, for scale)

Thermal Analysis of AMS-100

D. Kohlberger, 01-2022



Magnetic Field and Stability

Design **B-field of 0.65 T** in the center, ~ 1 T on the conductor at the edge of the solenoid.

B-field of 0.5 T when then compensation coil is on.

Operating temperature range of 50 to 60 K:

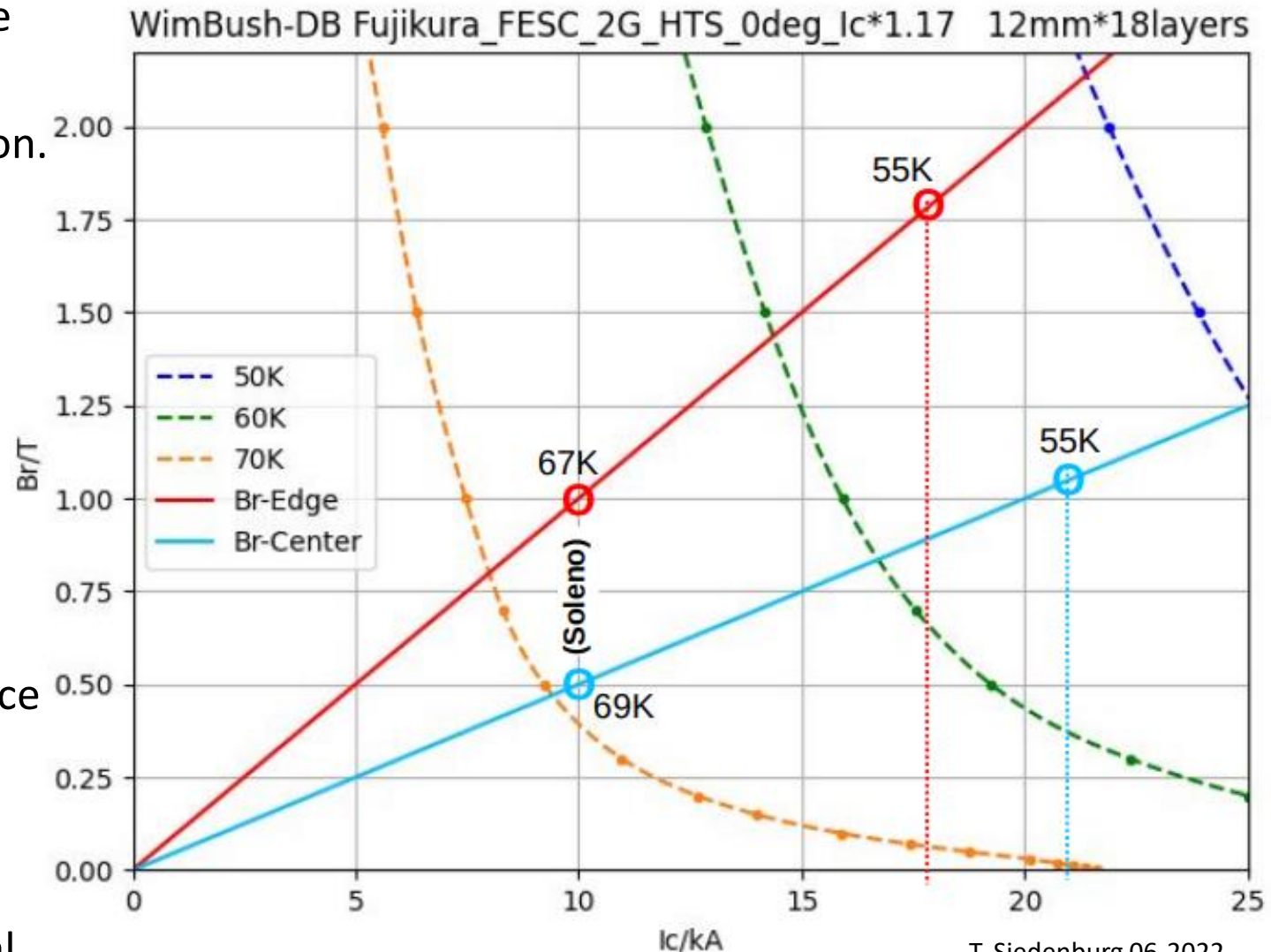
- ΔT of 12 K @ 55 K

Large temperature margin is important:

- cooling power is very limited,
- high energy density,
- no intervention possible.

Smart spacing of the conductor / additional HTS tape is envisioned at the coil extremities to reduce the peak field. And allow possible operation at higher current/magnetic field.

The field homogeneity is not an operation critical parameter.



Dangers of Space: Micrometeorite Impact

ø2.8 mm Al sphere
7.06 km/s

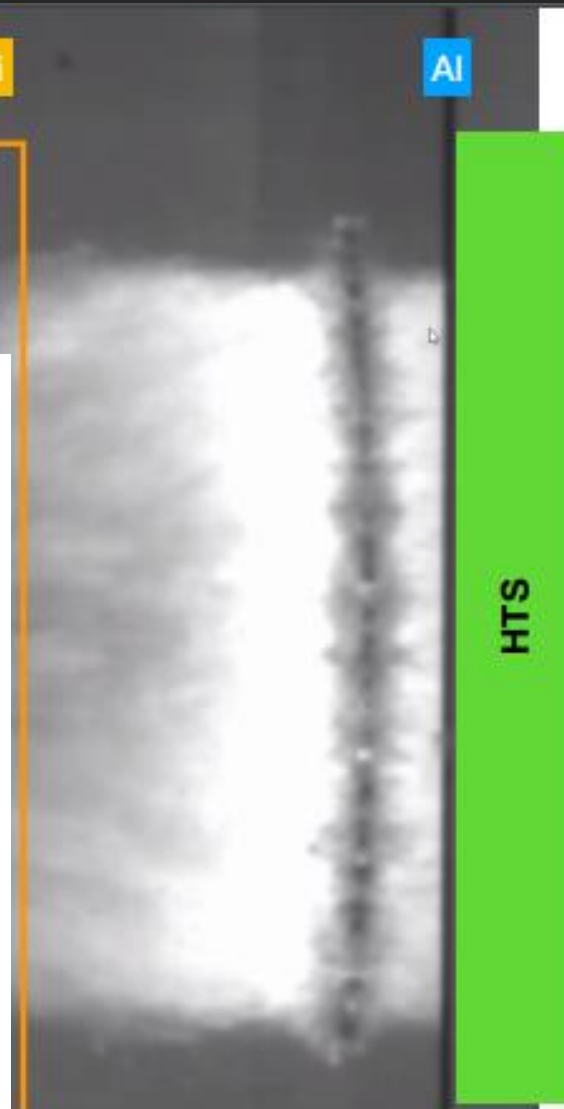
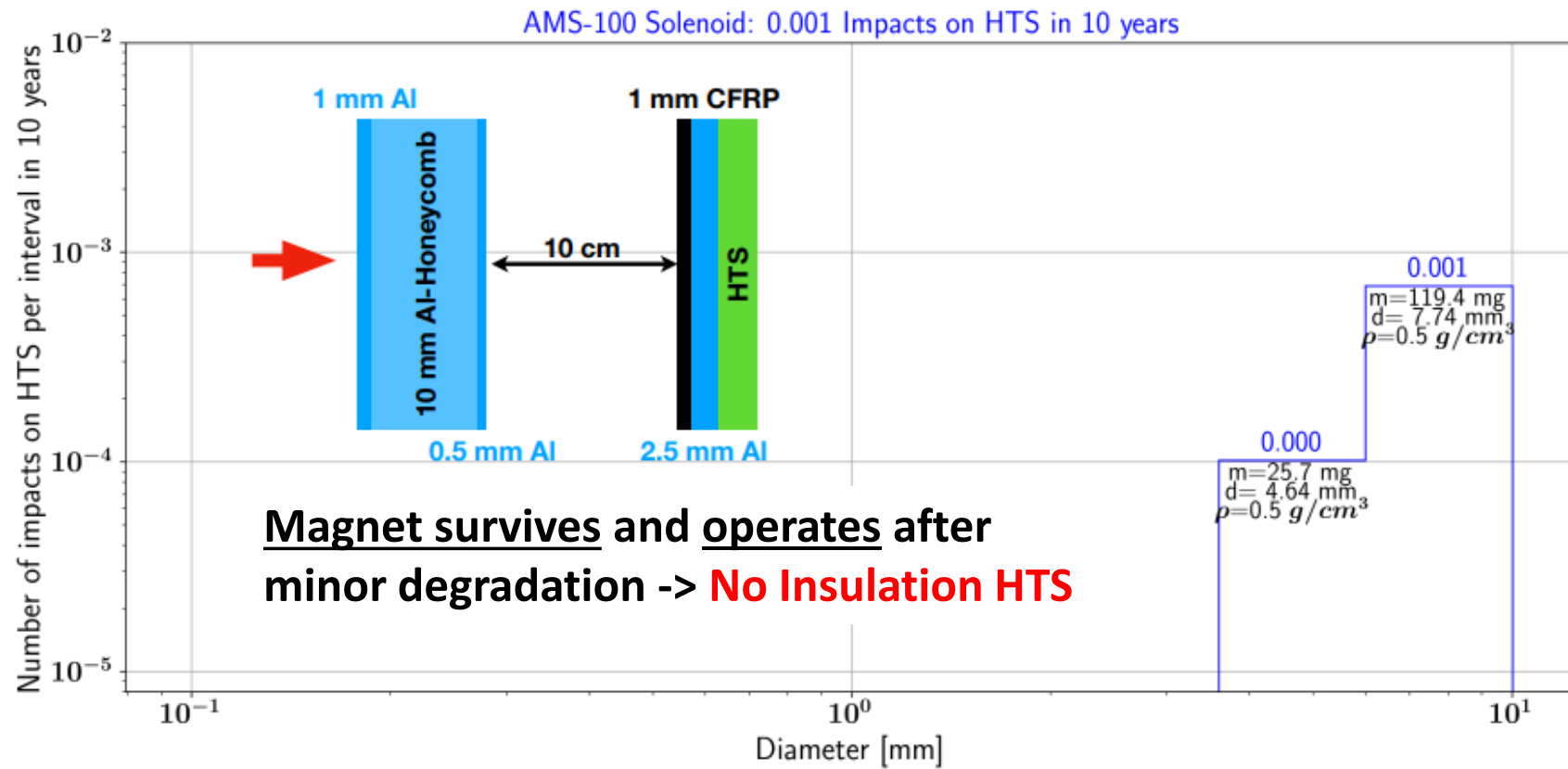
Micro-Meteoroid

Radiator

Whipple Shield?

SciFi

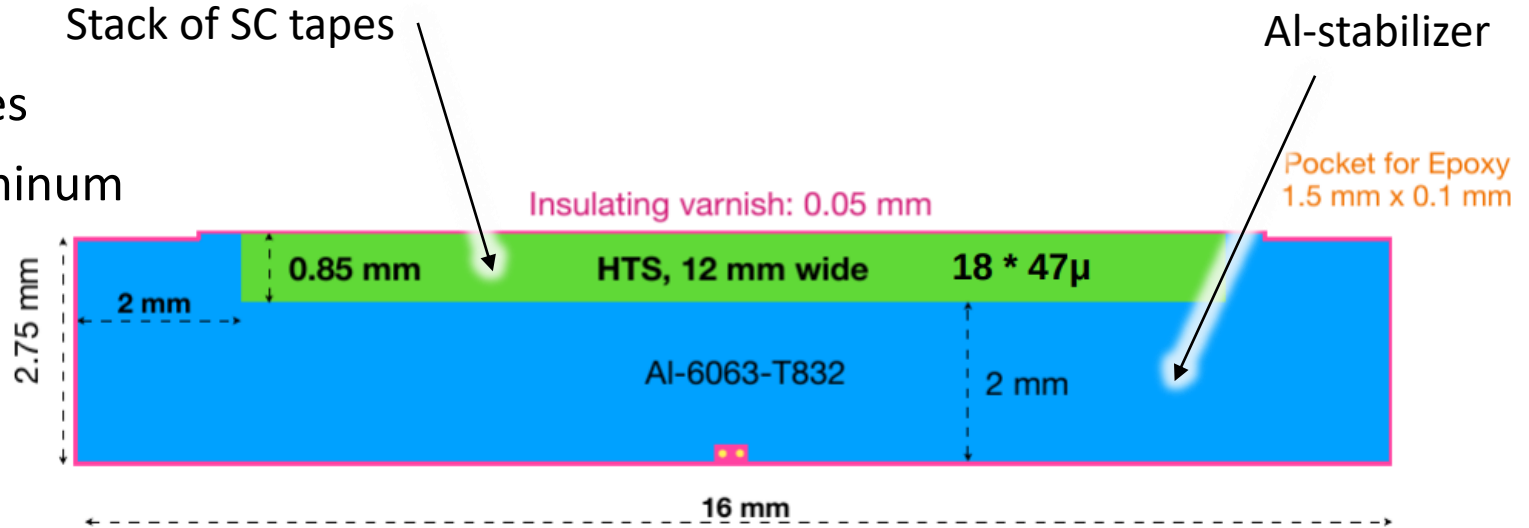
Al



Conductor and Coil Layout

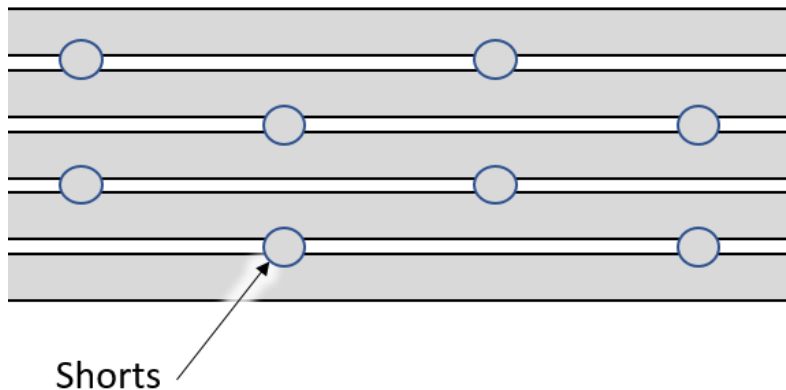
Current conductor layout:

- Stack of eighteen 12 mm wide HTS tapes
- HTS stack is soldered to tin-coated aluminum (6000 series) conductor stabilizer.
- Conductor thickness of 2.75 mm.
- Outer surface anodized / varnished to provide turn-to-turn insulation.



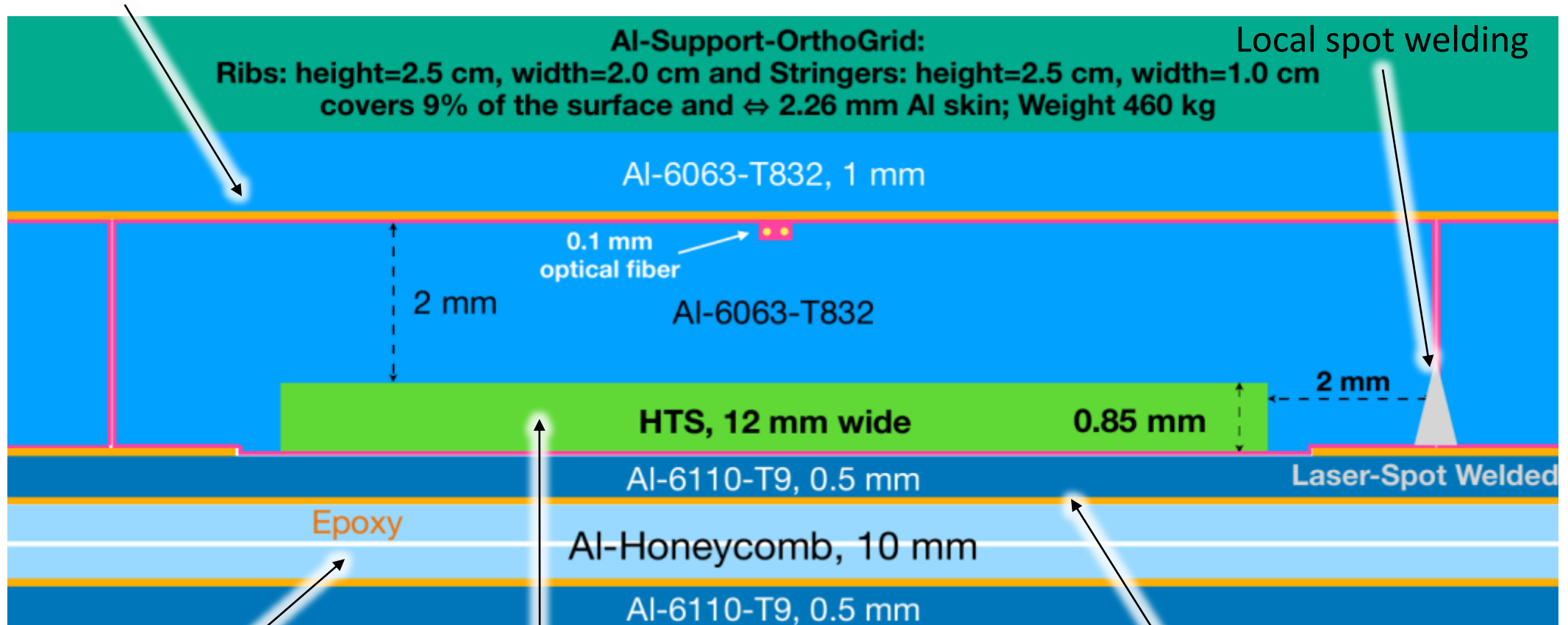
Shorting turns by (EB / laser) point welding.

- 1 mm² weld provides a turn-to-turn resistance of about 3e-5 Ω.
- AMS-100 -> 1250 mm² per turn (10 % of the circumference) covered with point welds of 1 mm² -> τ = 10 hours.
- Provides mechanical strength and provides thermal/electrical path.
- Shorts are within the envelope of the conductor pack.
- To be tested and to be demonstrated.



Structure of the Main Solenoid

Al-alloy skin for mechanical strength and axial thermal conductivity




Honeycomb for mechanical stiffness

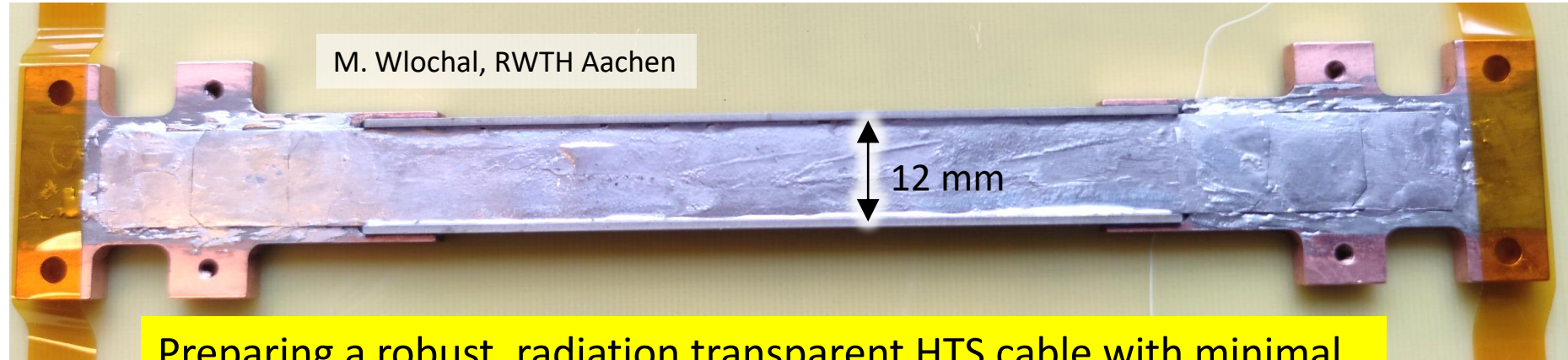
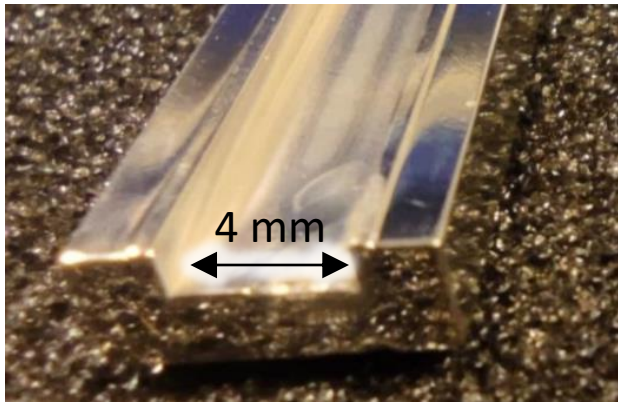
Stack of HTS tapes

Epoxy between layers

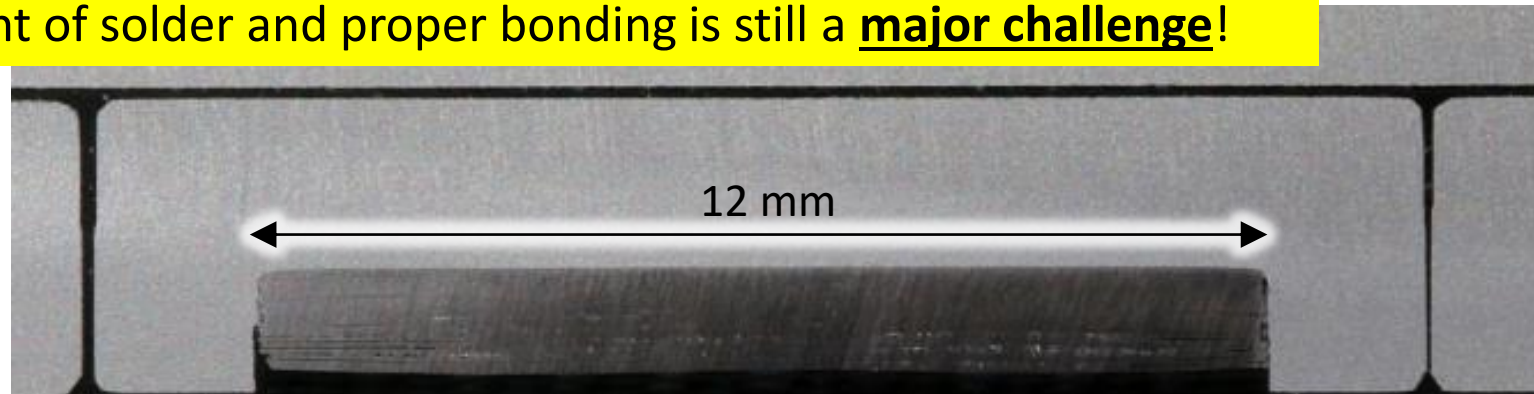
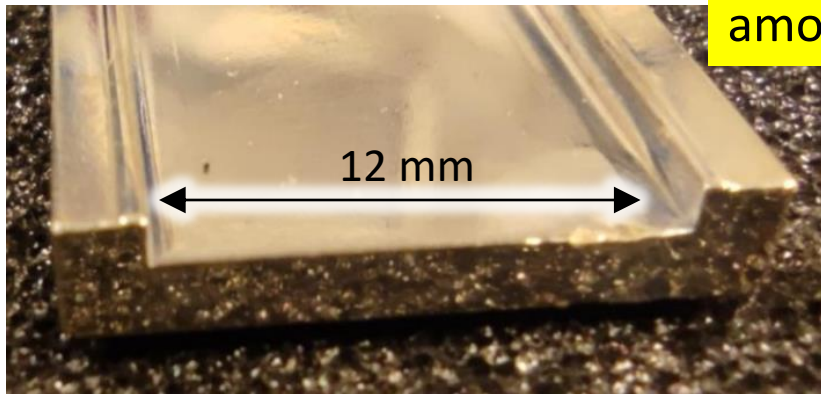
$$X_0 = 10.2\% = \text{Thickness of structure} / \text{Radiation length}$$

Conductor Testing: Single- and Multi-Tape Samples

- Single tapes have been extensively characterized .
- Many short samples of Al-alloy stabilized multi-tape HTS conductors are in preparation.
- Few-tape samples in good agreement with expectations.
- Next: more tapes, bending, micro-meteorite impact testing, etc.

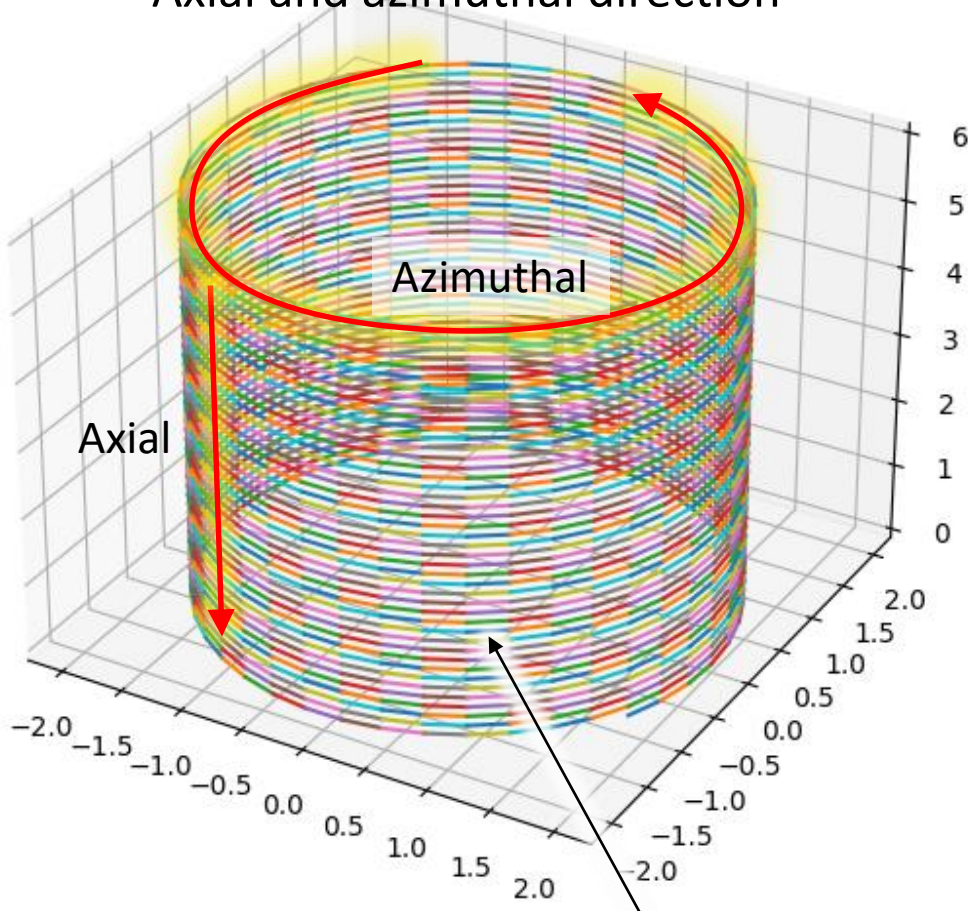


Preparing a robust, radiation transparent HTS cable with minimal amount of solder and proper bonding is still a **major challenge!**



Thermal-ElectroMagnetic Quench Model

Thermally and electrically connected:
Axial and azimuthal direction

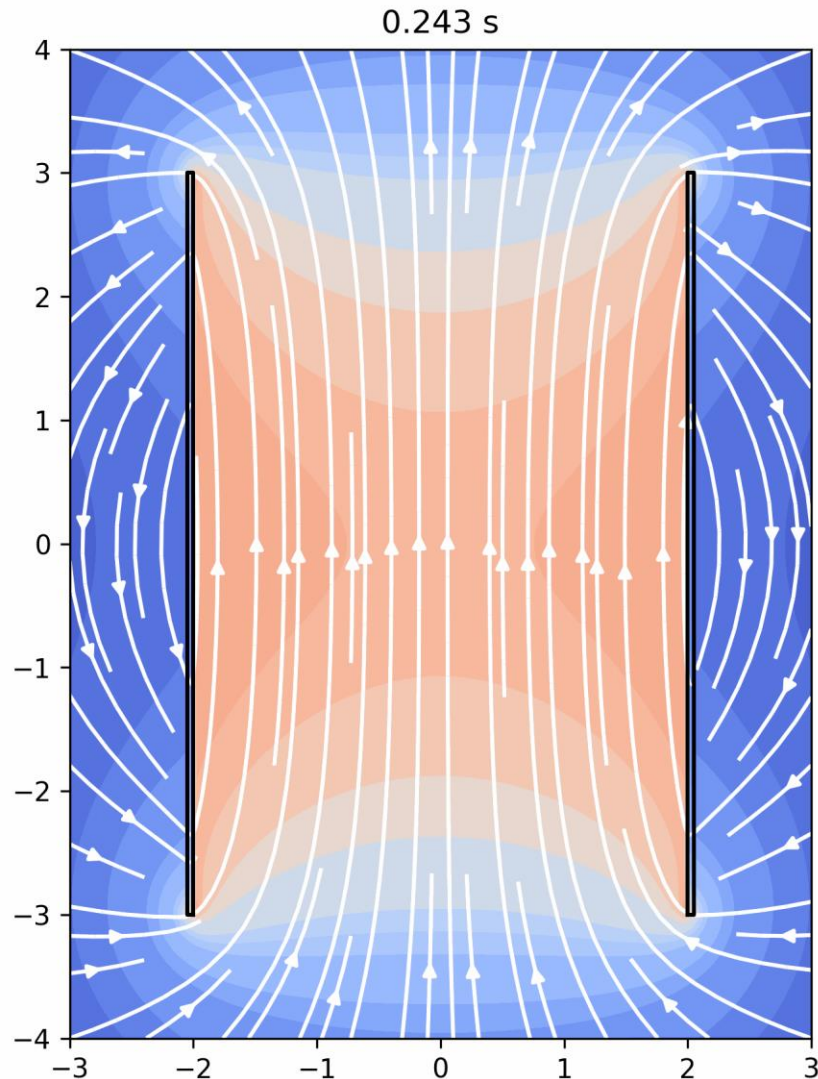


Turns divided in
to line elements

Quench behavior of the non-insulated AMS-100 main solenoid

- Quench behavior of the AMS-100 main solenoid is studied for several quench scenarios.
- Quasi 3D thermal, electrical and magnetic nodal-network model is built using python.
- Results from this model are analyzed in ANSYS/Abacus to evaluate the resulting mechanical response.
- Model studies the effect of slow thermal runaway and consequently a fast quench as function of a small defect.
- Not enough resolution at the moment for sudden and very local defects (due for e.g. micrometeorite impact).
- Other structural elements, such as end-flanges and ribs, are not yet included in the model.
- Simulations performed using a previous design iteration: 428 turns, an operating current of 13.5 kA and a field of 1 T.

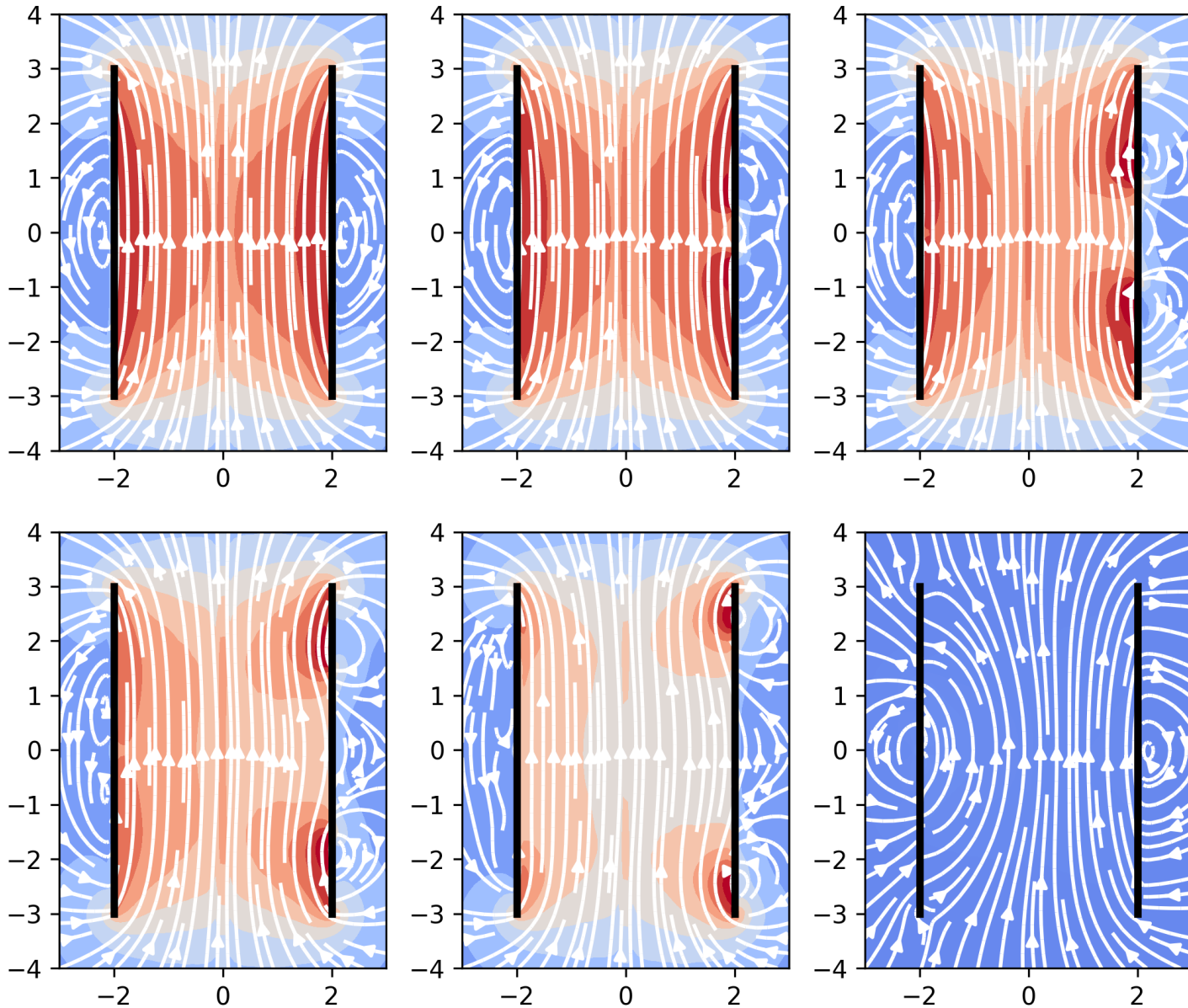
Thermal-ElectroMagnetic Quench Model



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Quench Behavior and Survival



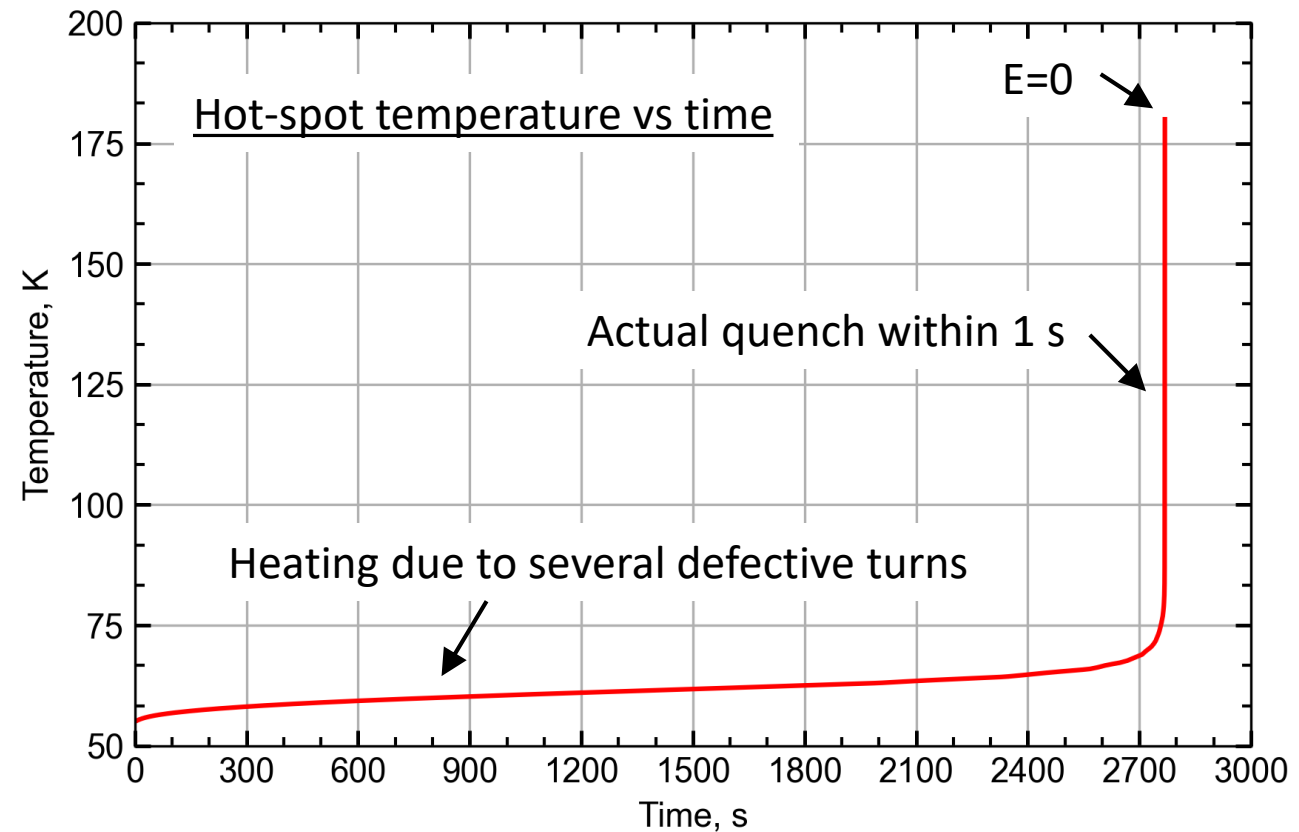
Quench behavior of the non-insulated AMS-100 main solenoid:

- Quench propagation is driven by inductive effects instead of thermal NZP.
- Current is pushed to adjacent turns, these turns reach I_c and consequently quench themselves.
- Hot-spots are observed near the coil extremities as current (and thus energy) is pushed towards those.
- Mechanical ripple follows the normal zone.
- Thermal run-away is slow, but the quench is fast < 1 s.

Quench Behavior and Survival

Investigated slow thermal runaway due to defective turn(s):

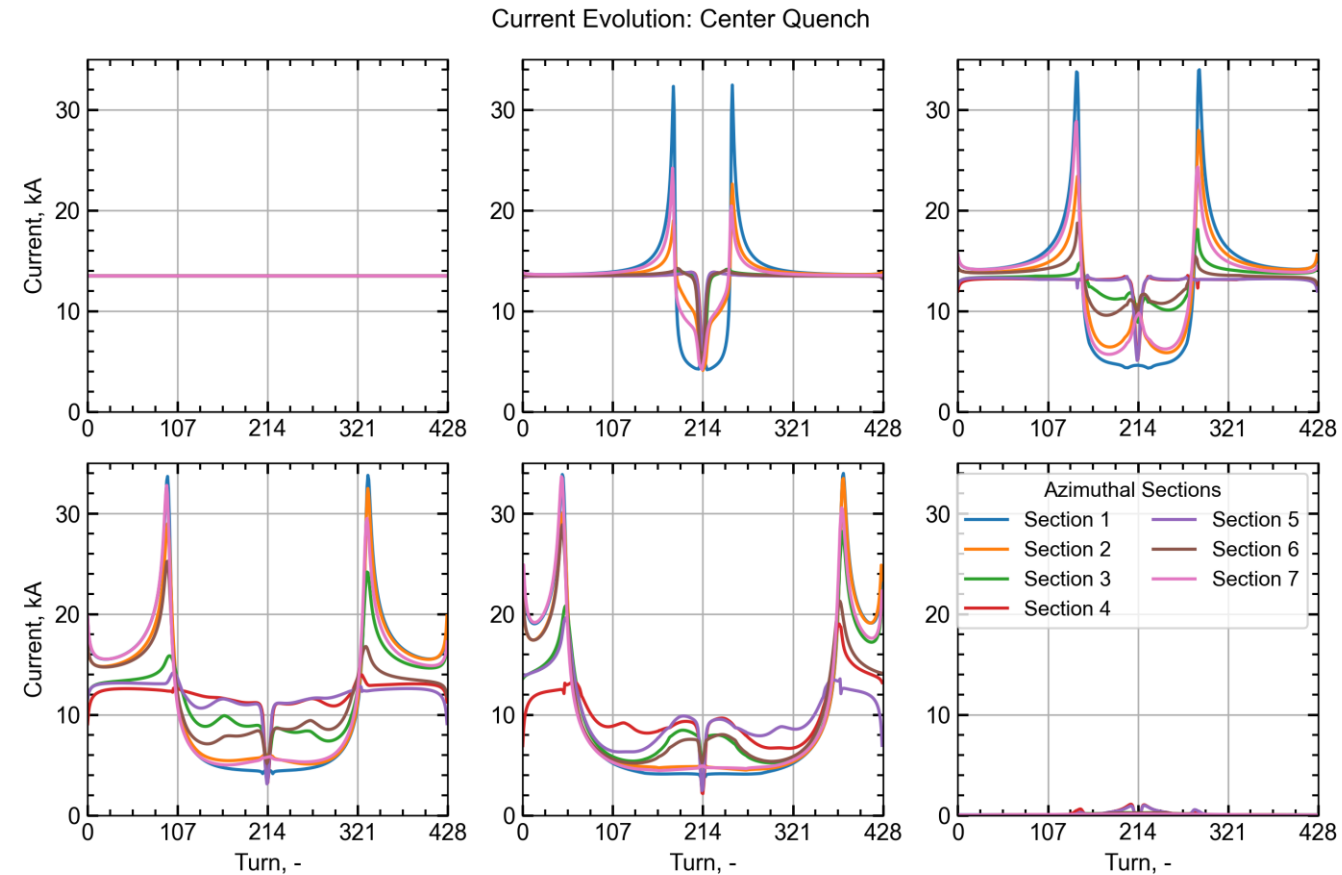
- Current bypasses the defective turn via its turn-to-turn resistance.
- ~ 4 W of heating in the axial resistance per turn.
- Slow thermal runaway in the order of hour(s).
- All energy dissipated within the magnet during the quench, no external extraction.
- Conventional protection methods, such as quench heaters, are ineffective.
- Quench at an extremity gives the highest hot-spot temperature of ~ 190 K.
- Last iteration of the conductor layout is able to cool away heat from one defective turn.



Quench Behavior and Survival

Investigated slow thermal runaway due to defective turn(s):

- NZP is driven by inductive components.
- Current is pushed to adjacent turns, these turns reach I_c and consequently quench themselves.
- High current of $> 2.5 \times I_{op}$ reached for a short period of time.
- Local Lorentz force doubled during a quench.
- Mechanical ripple follows the front of the NZ.
- A quench starting in the center gives the lowest hot-spot temperature.
- A quench starting in an extremity gives the highest hot-spot temperature on the other extremity (all current/energy is pushed towards the other extremity).



Quench in the center of the magnet
in azimuthal section 1.

Simulated Quench Behavior and Survival

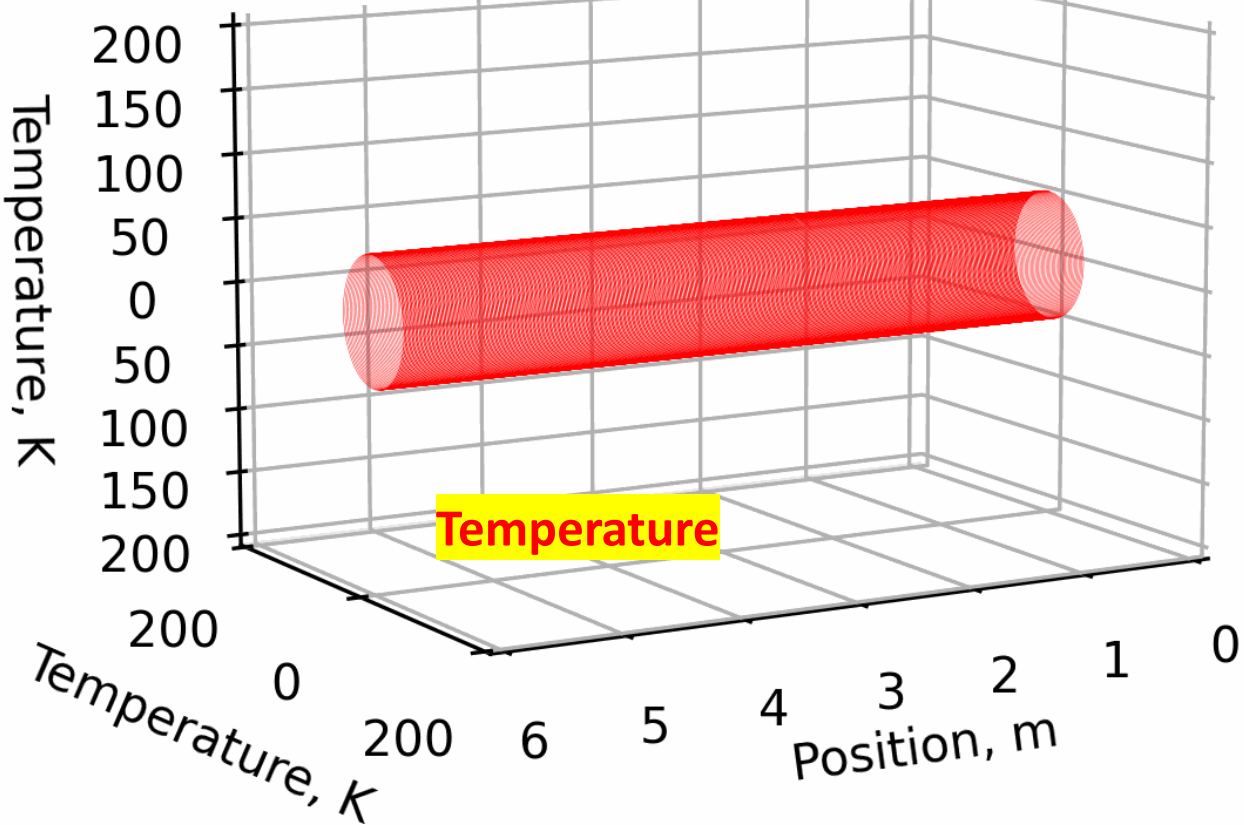
428 turn main solenoid, $I_{op} = 13.5$ kA, $B = 1$ T.

$t = 0.2428$ s

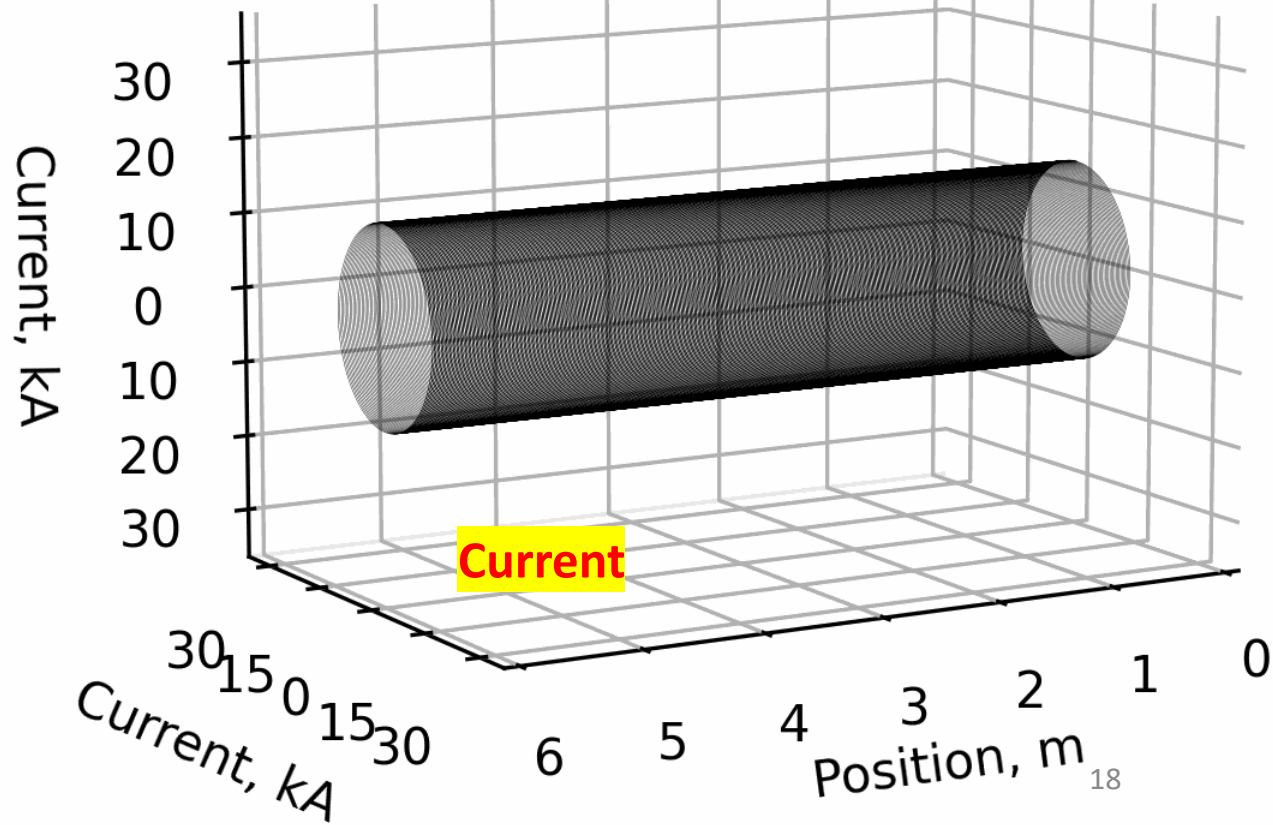
$t = 0.2428$ s

Simulations indicate that the main solenoid is thermally self-protected.

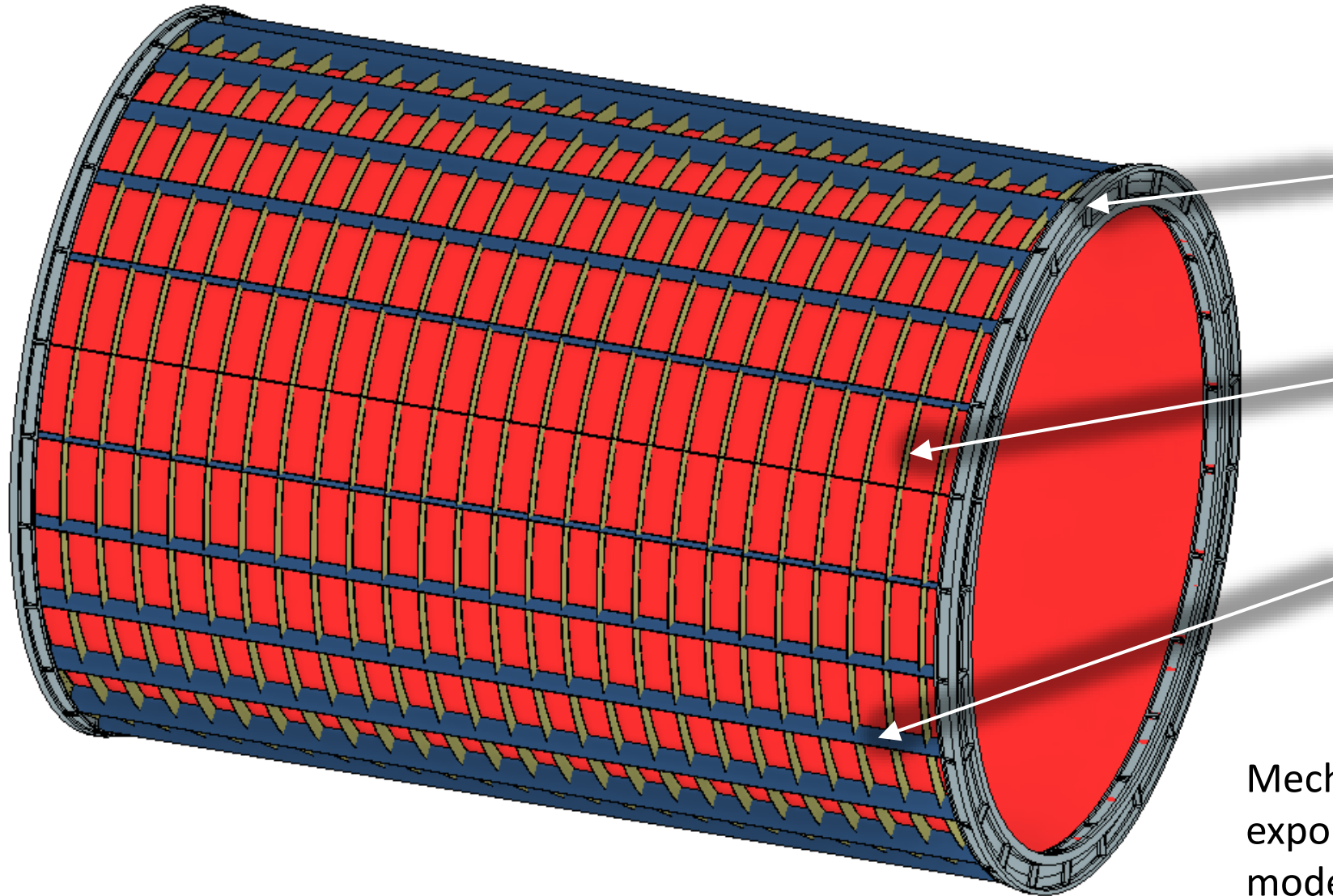
Peak hot-spot near extremities



NZPV of $\sim 4-8$ m/s



End-flanges, Ribs and Stringers



End-Flanges (grey): Mechanical support of the magnet during manufacturing, launch and operation.
Circular, allows quench-back.

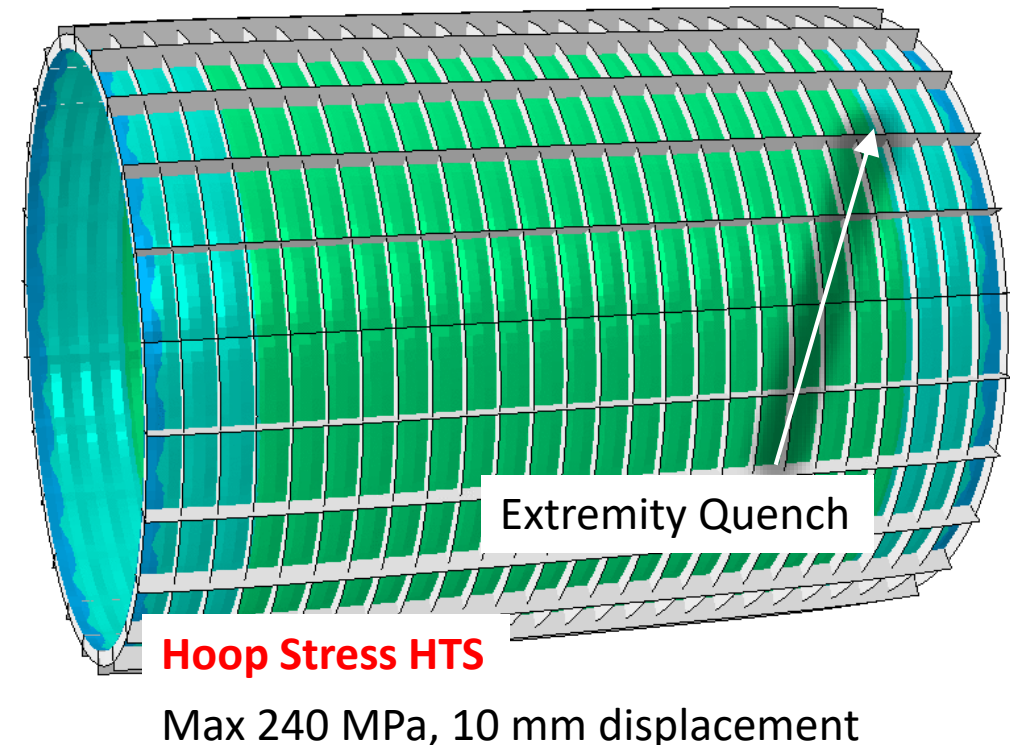
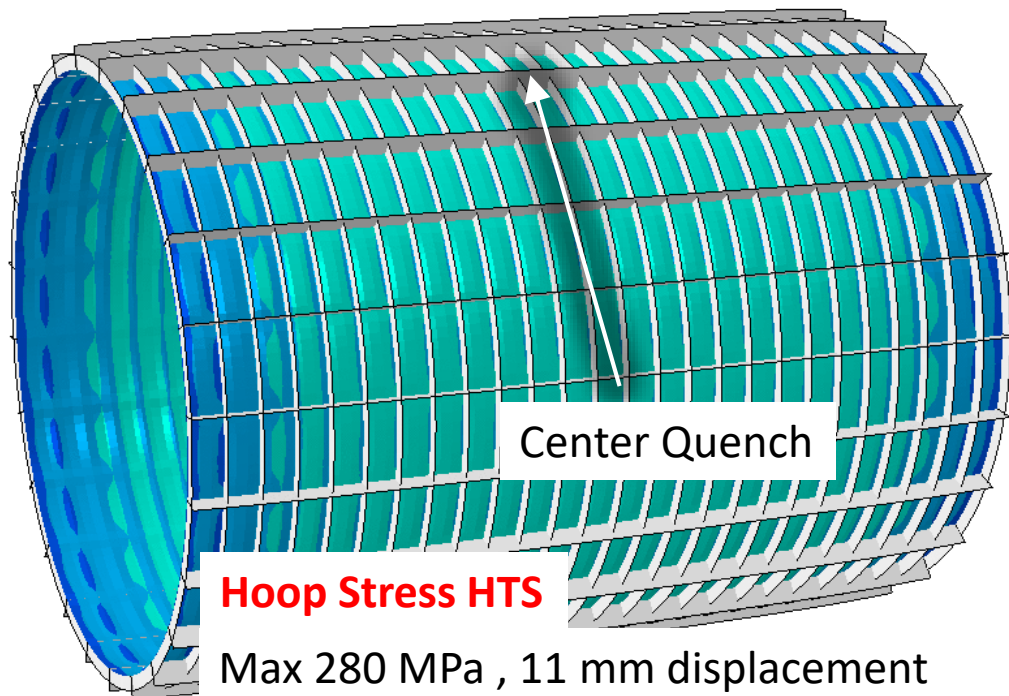
Ribs (yellow): Mechanical support of the magnet during operation and quench events.
Circular, allows quench-back.

Stringers (blue): Mechanical support during launch.

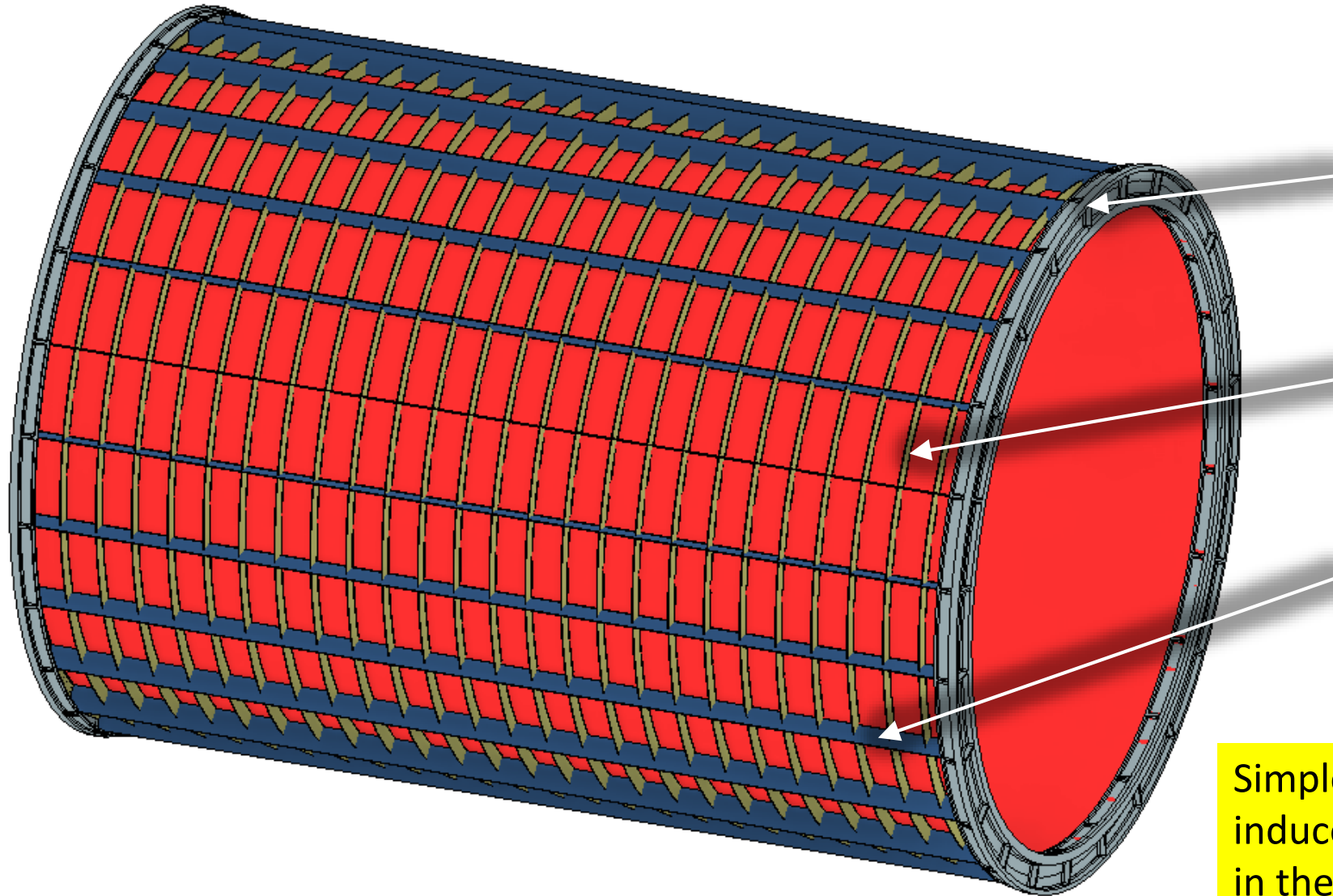
Mechanical load on the conductor is exported from the thermal-electrical model to Abaqus.

Mechanical Quench Analyses

- Shell model set-up in Abaqus to calculate stress in the HTS, Al-alloy conductor and structural components.
- Model includes the conductor, ribs and stringers.
- Stress in the conductor is almost tripled during a quench due to enormous induced current.
- Ribs locally reduce the stress in the conductor.
- Stress in the conductor due to thermal gradients not critical, strength of the epoxy to be validated experimentally.
- Peak stress (~ 300 MPa) caused by radial Lorentz force.
- Support structure requires optimization.



End-flanges, Ribs and Stringers



End-Flanges (grey): Mechanical support of the magnet during manufacturing, launch and operation.
Circular, allows quench-back.

Ribs (yellow): Mechanical support of the magnet during operation and quench events.
Circular, allows quench-back.

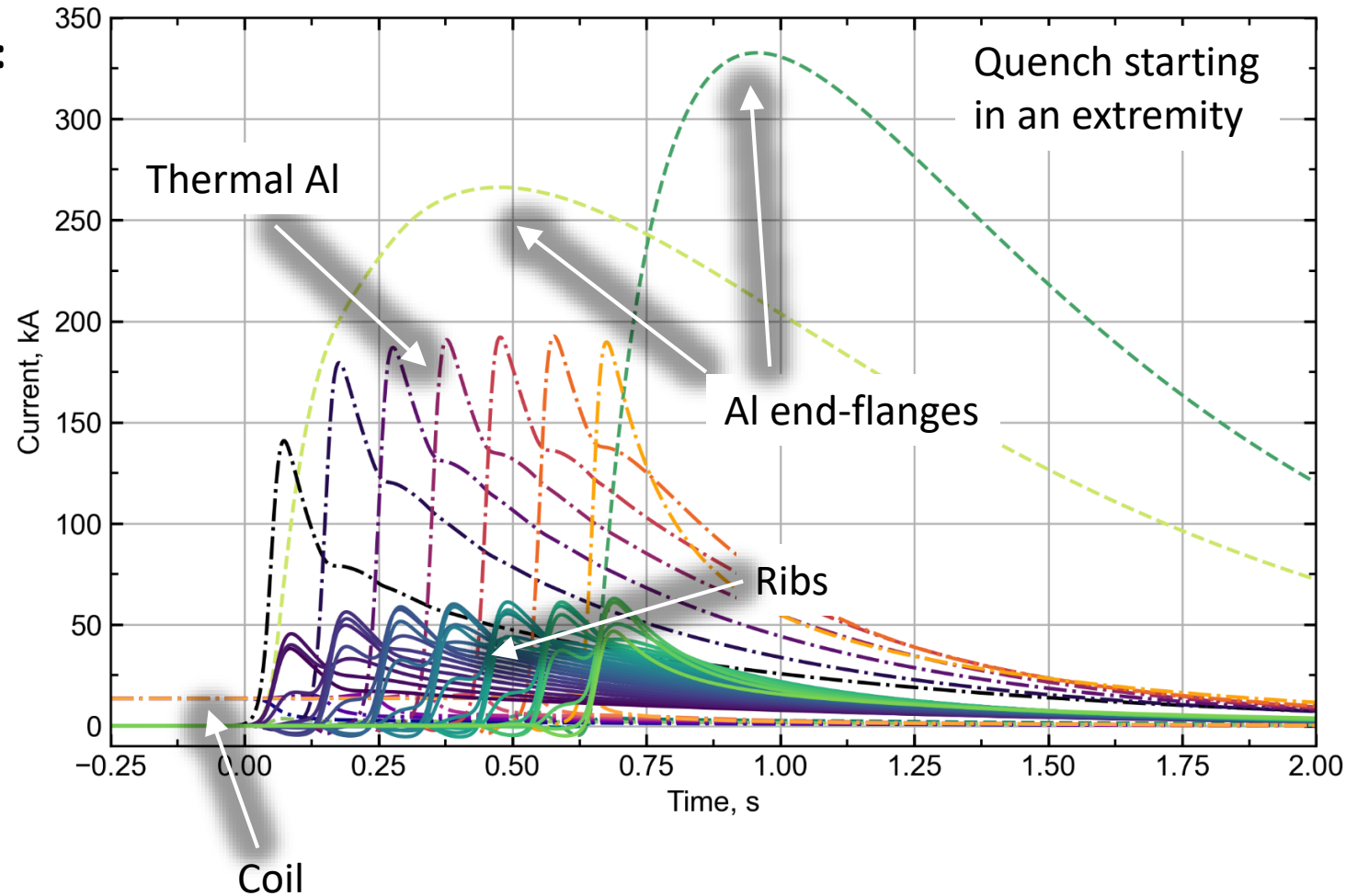
Stringers (blue): Mechanical support during launch.

Simpler 2D Model is set up to estimate induced currents and energy dissipation in the (structural) circular components.

AMS-100 - End-flanges and Ribs: Quench-back

Simpler 2D model to estimate induced currents and losses in other components:

- Coil (divided in to 7 sections)
 - Thermal aluminum (7 sections)
 - End-flanges (2x Al alloy, 2x SS)
 - Ribs (25 pieces, Al alloy)
- Coil sections are quenched 0.1 s after each other.
 - **250-350 kA** induced in the Al end-flanges.
 - 50 kA induced in each of the ribs (**1 MA** total).
 - **1 MA** induced in the thermal aluminum.



Optimization of the ribs and end-flanges is ongoing, results are preliminary.

AMS-100 - End-flanges and Ribs: Quench-back

Quench from the center:

Total stored energy: 33.7 MJ
In conductor: 17.2 MJ (**51%**)
In ribs: 7.4 MJ (**22%**)
In flanges: 2.7 MJ (**8%**)
In thermal aluminum: 6.4 MJ (**19 %**)

Quench from an extremity:

Total Energy: 33.7 MJ
In conductor: 17.5 MJ (**52%**)
In ribs: 7.1 MJ (**21%**)
In flanges: 2.7 MJ (**8%**)
In thermal aluminum: 6.4 MJ (**19%**)

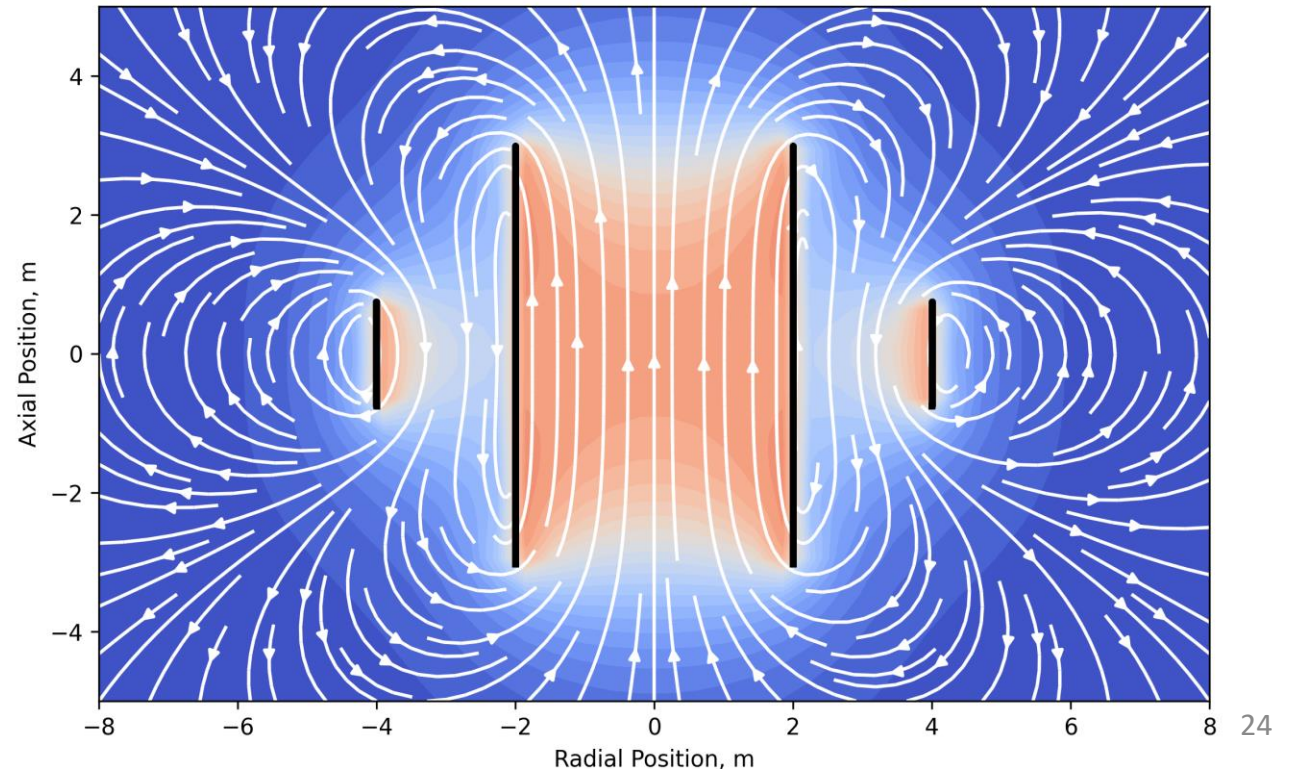
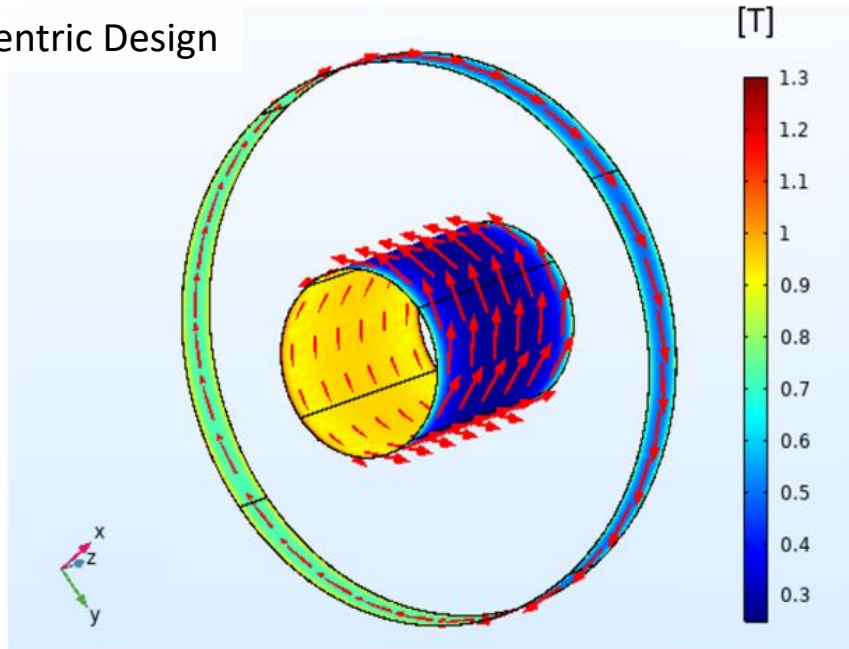
- Large fraction (50 %) of the stored energy is dissipated in the ribs, flanges and thermal aluminum.
- Will result in a much lower hot-spot temperature.
- Induced current in the structural elements result in a lower mechanical load on the conductor itself.
- Slows down the quench process.
- Expected that further mechanical optimization will reduce the mass and dimensions of the ribs and end-flanges.
- Expected reduction of energy dissipated in the ribs and end-flanges from 30% -> 10-20 %.
- One model that includes the main solenoid, compensation coil and all structural components needed.

Remaining Challenge: Compensation Coil

There is a non-zero background field at L2, magnetic torque -> magnet tries to align to this B-field.

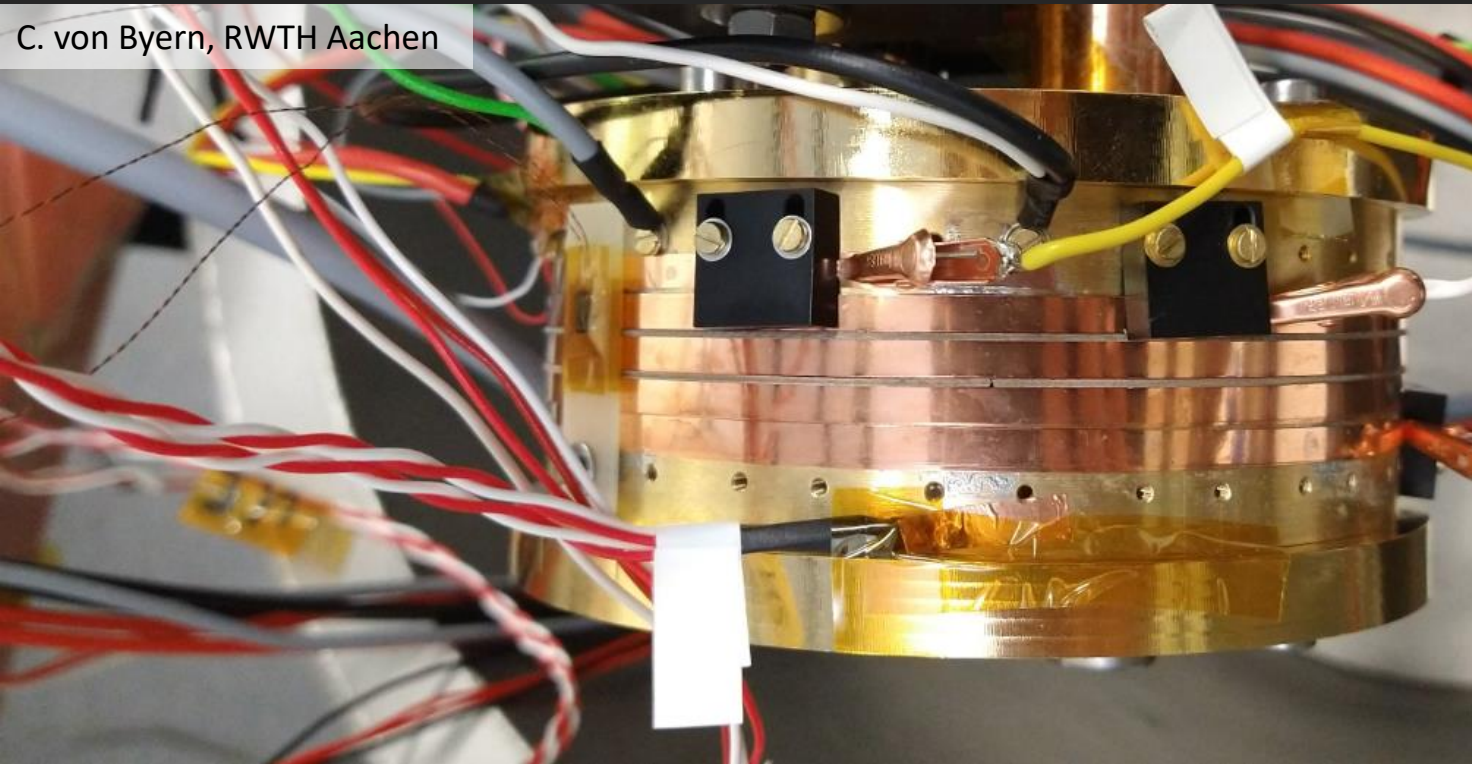
- Compensation coil is needed to have **altitude control**.
- Concentric comp. coil with a radius of 4 m under investigation (to fit in to the cargo bay of the rocket).
- All current designs are inherently unstable and a strong support structure is essential.
- 2D axisymmetric quench model including the compensation coil and struct. components in development.

Concentric Design



AMS-100 Demonstrator Coils

C. von Byern, RWTH Aachen

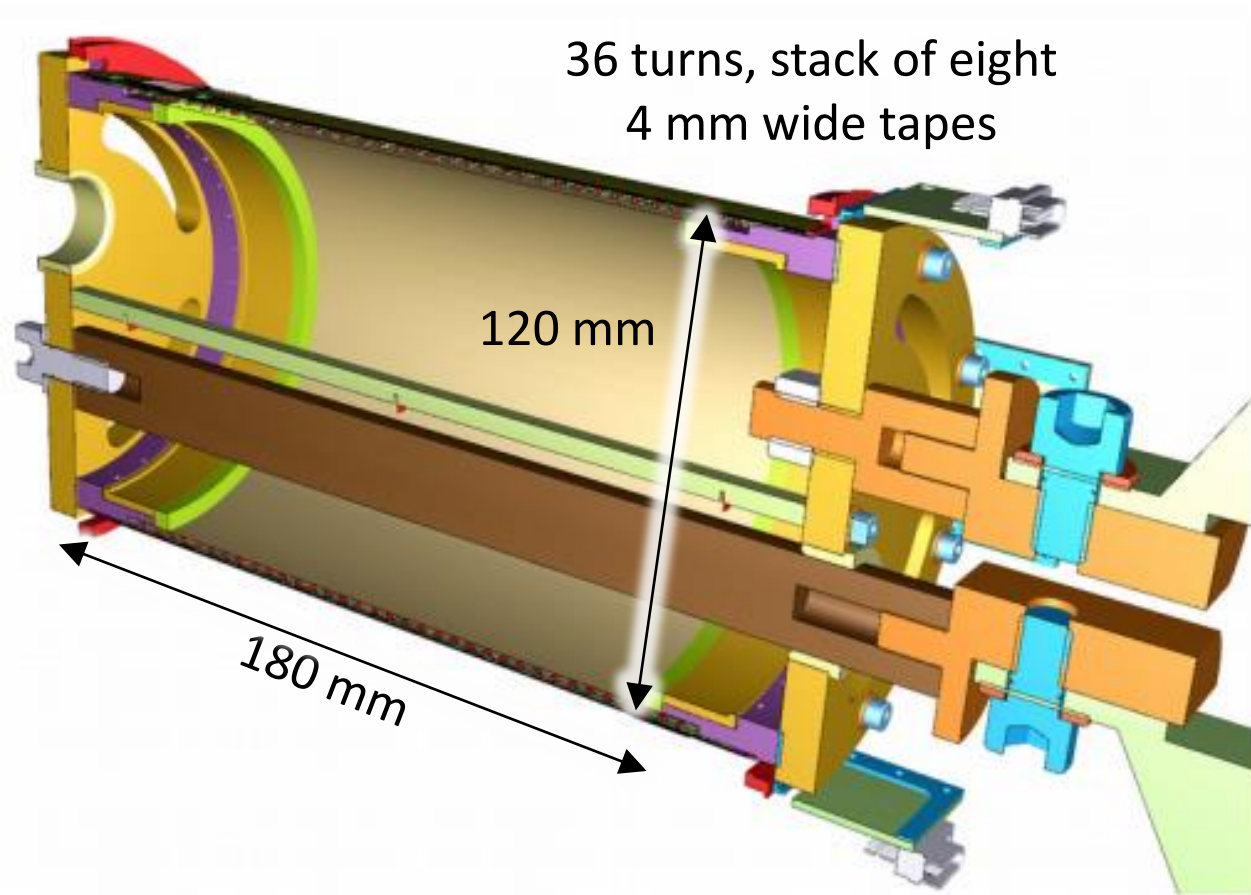


D. Uglietti, EPFL

Several compact demonstrator coils are envisioned and in preparation.

- Test preparation procedures and components.
- Validate models and results (mechanical, electrical and thermal).
- Coils will be pushed to their limits.
- Starting with small, few turn demonstrator coils, later moving to larger coils and the coil for the pathfinder mission.

AMS-100 Demonstrator Coils



- Compact demonstrator in preparation for validating the thermal-electromagnetic model.
- Testing all preparation methods.
- Heavily instrumented.
- Testing at 4 K up to 5 T and 20 – 60 K in S.F.



Conclusions

- AMS-100 magnet system faces many design challenges due to its
 - ultra-thin 0.5 T HTS coil,
 - large stored energy of 14 MJ,
 - very limited cooling via external radiators,
 - requirement to survive high-vibration launch conditions,
 - requirement to fit the magnet and its compensation coil(s) inside a rocket.
- Quench model is under development that predicts the quench behavior of the main solenoid, the resulting hot-spot temperature and mechanical load on the conductor.
- Testing of materials and preparation procedures is ongoing. Several small demonstrator coils are in preparation to test different design aspects of the AMS-100 magnet.
- The magnet system is in its early design phase. Several iterations are expected to fine-tune its design.

