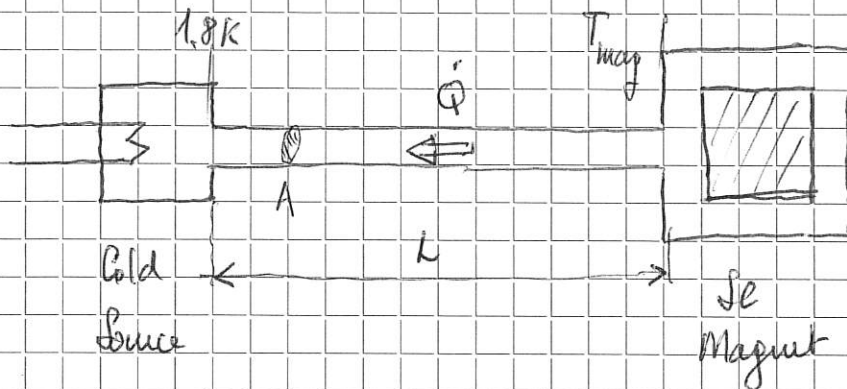




Problem 1 = Steady-state conduction in He II



1/ given $\left\{ \begin{array}{l} T_{\text{mag}} = 1.9 \text{ K} \\ \dot{Q} = 20 \text{ W} \end{array} \right.$ $l = 8 \text{ m} = 800 \text{ cm}$

Equivalent conduction law $\dot{q}^{3.4} L = X(1.8 \text{ K}) - X(1.9 \text{ K})$

From the $X(T)$ plot, in appropriate units $\left\{ \begin{array}{l} X(1.8 \text{ K}) = 360 \\ X(1.9 \text{ K}) = 200 \end{array} \right.$

Hence $\dot{q}^{3.4} = \frac{\Delta X}{L} = \frac{160}{800} = 0.2$

$\dot{q} = 0.2^{1/3.4} = 0.623 \text{ W/cm}^2$

Since $\dot{q} = \frac{\dot{Q}}{A} \Rightarrow A = \frac{\dot{Q}}{\dot{q}} = \frac{20}{0.623} \text{ cm}^2 = 32.1 \text{ cm}^2$

And $D^2 = \frac{4A}{\pi} = 40 \text{ cm}^2 \Rightarrow D = 64 \text{ mm}$

2/ Same $\left\{ \begin{array}{l} T_{\text{mag}} = 1.9 \text{ K} \\ \dot{Q} = 20 \text{ W} \end{array} \right.$ but new $L = 16 \text{ m}$

Hence $\dot{q}^{3.4} = \frac{\Delta X}{L} = \frac{160}{1600} = 0.1$

$\dot{q} = 0.1^{1/3.4} = 0.508 \text{ W/cm}^2$



$$A = \frac{\dot{q}}{\dot{q}'} = \frac{20}{0,508} = 39,4 \text{ cm}^2$$

$$D^2 = \frac{4A}{\pi} = 50,1 \text{ cm}^2 \Rightarrow D = 7,1 \text{ mm}$$

\Rightarrow Doubling the conduction length does not half the heat flux between fixed temperatures, as it would have done for "normal" conduction

3/ given $L = 8 \text{ m}$ and $D = 64 \text{ mm}$ ($A = 32,1 \text{ cm}^2$)

The heat flow from the magnet is now increased to 25 W.

$$\dot{q}' = \frac{\dot{Q}}{A} = \frac{25}{32,1} = 0,779 \text{ W/cm}^2$$

$$\dot{q}'^{3,4} = 0,427 \Rightarrow \dot{q}'^{3,4} L = 342$$

From the equivalent conduction law, in appropriate units

$$\dot{q}'^{3,4} L = 342 = X(1,8\text{K}) - X(T_{\text{mag}})$$

$$\text{Hence } X(T_{\text{mag}}) = 360 - 342 = 18$$

$$T_{\text{mag}} = 2,07 \text{ K}$$

\Rightarrow The magnet temperature is now just below the lambda point.



4 / given $L = 8 \text{ m}$ and $D = 64 \text{ mm}$
($A = 32,1 \text{ cm}^2$)

The maximum heat flux is obtained when the magnet temperature reaches the lambda point

$$T_{\text{mag}} = T_{\lambda} \Leftrightarrow X(T_{\text{mag}}) = X(T_{\lambda}) \equiv 0$$

The maximum heat flux is given by

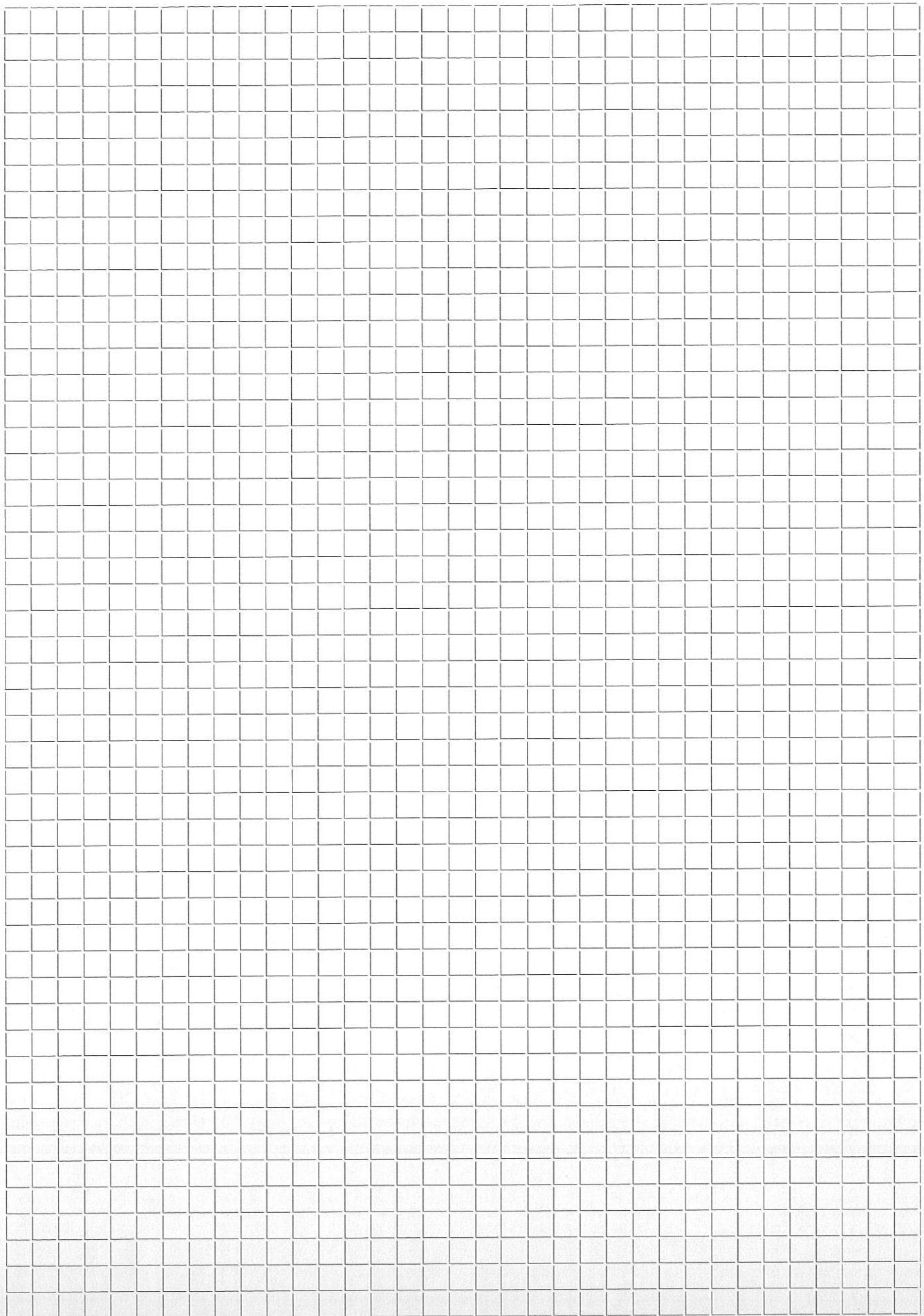
$$\overset{-3,4}{q_{\text{max}}} L = X(1,8) - X(T_{\lambda}) = 360$$

$$\overset{-3,4}{q_{\text{max}}} = \frac{360}{800} = 0,45$$

$$\overset{\cdot}{q_{\text{max}}} = 0,791 \text{ W/cm}^2$$

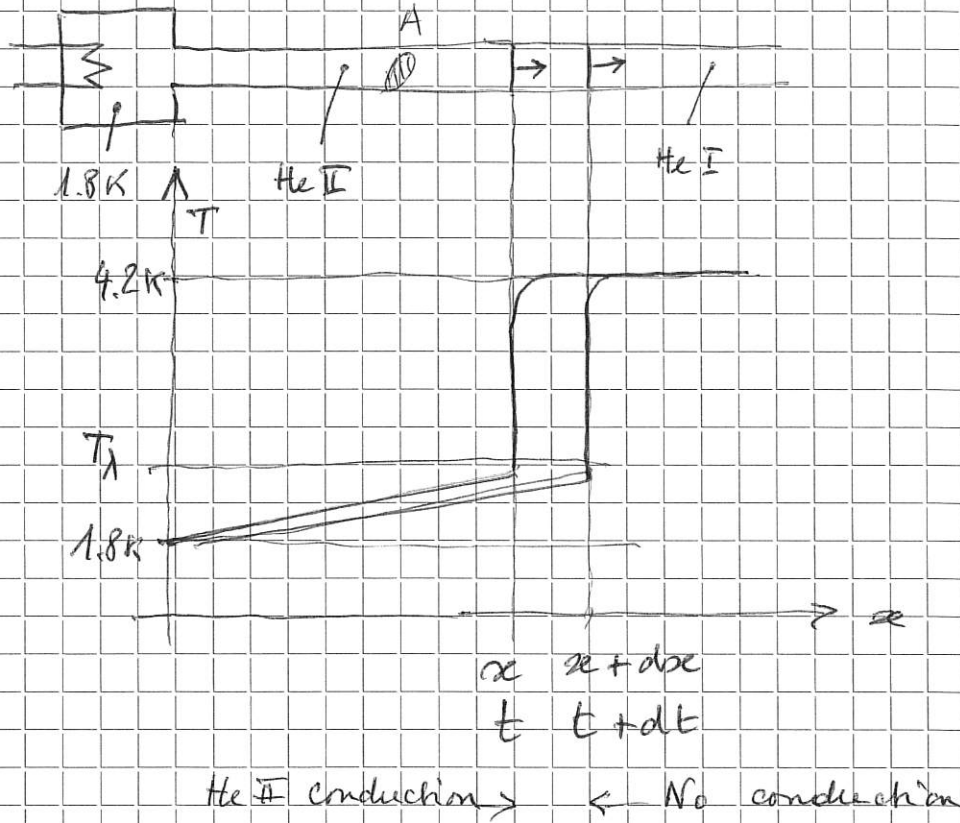
$$\overset{\cdot}{Q_{\text{max}}} = 0,791 \times 32,1 = 25,4 \text{ W}$$







Problem 2 = Cooldown of liquid helium duct to superfluid



1/ The interface between normal and superfluid helium will be a front moving along the duct at T_λ

The heat conducted to the cold source in He II will be used to cool the helium from its initial temperature of 4.2K to the lambda point T_λ .

2/ • He II conduction $\left(\frac{\dot{Q}}{A}\right)^{3.4} x = X(1.8K) - X(T_\lambda) = X(1.8K)$

Hence $\dot{Q} = \left(\frac{X(1.8K)}{x}\right)^{1/3.4} A$



Over time dt , the front progress of dx

$$dQ = \dot{Q} dt = \left(\frac{X(1.8K)}{x} \right)^{1/3.4} A dt$$

• Cooling of slice dx $d\dot{Q} = \Delta H dm = \rho A dx \Delta H$

↑ enthalpy difference of helium ($4.2 - T_{1.8}$)

Energy conservation =

$$\left(\frac{X(1.8K)}{x} \right)^{1/3.4} A dt = \rho A \Delta H dx$$

$$\Rightarrow \frac{dx}{dt} = \frac{X(1.8K)^{1/3.4}}{\rho \Delta H} \times \frac{1}{x^{1/3.4}} \quad 0.294$$

⇒ the velocity of the front does not depend on A

⇒ it slows down as distance x increases.

A.N. $X(1.8K) = 360$ $\rho = 147 \text{ kg/m}^3 = 147 \text{ g/cm}^3 = 0.147 \text{ g/cm}^3$

$\Delta H = 9902 - 3684 = 6268 \text{ J/kg} = 6.27 \text{ J/g}$

$$\frac{dx}{dt} = \frac{360^{0.294}}{0.147 \times 6.27} \times \frac{1}{x^{0.294}} \text{ cm/s}$$

$$\frac{dx}{dt} \approx 6.12 \times \frac{1}{x^{0.294}} \text{ cm/s} \quad (x \text{ in cm})$$

For $x = 1 \text{ m} = 100 \text{ cm}$, $\frac{dx}{dt} = 1.6 \text{ cm/s}$

$x = 10 \text{ m} = 1000 \text{ cm}$, $\frac{dx}{dt} = 0.8 \text{ cm/s}$



3/

$$t = \int_0^T dt$$

$$dt = \frac{\rho \Delta H_{1.8}^{4.2}}{X (1.8K)^{0.294}} \cdot 2x^{0.294} dx$$

$$t = \frac{\rho \Delta H_{1.8}^{4.2}}{X (1.8K)^{0.294}} L^{1.294}$$

$$t \approx \frac{0.147 \times 6.27}{360^{0.294}} L^{1.294}$$

$$t \approx 0.163 L^{1.294}$$

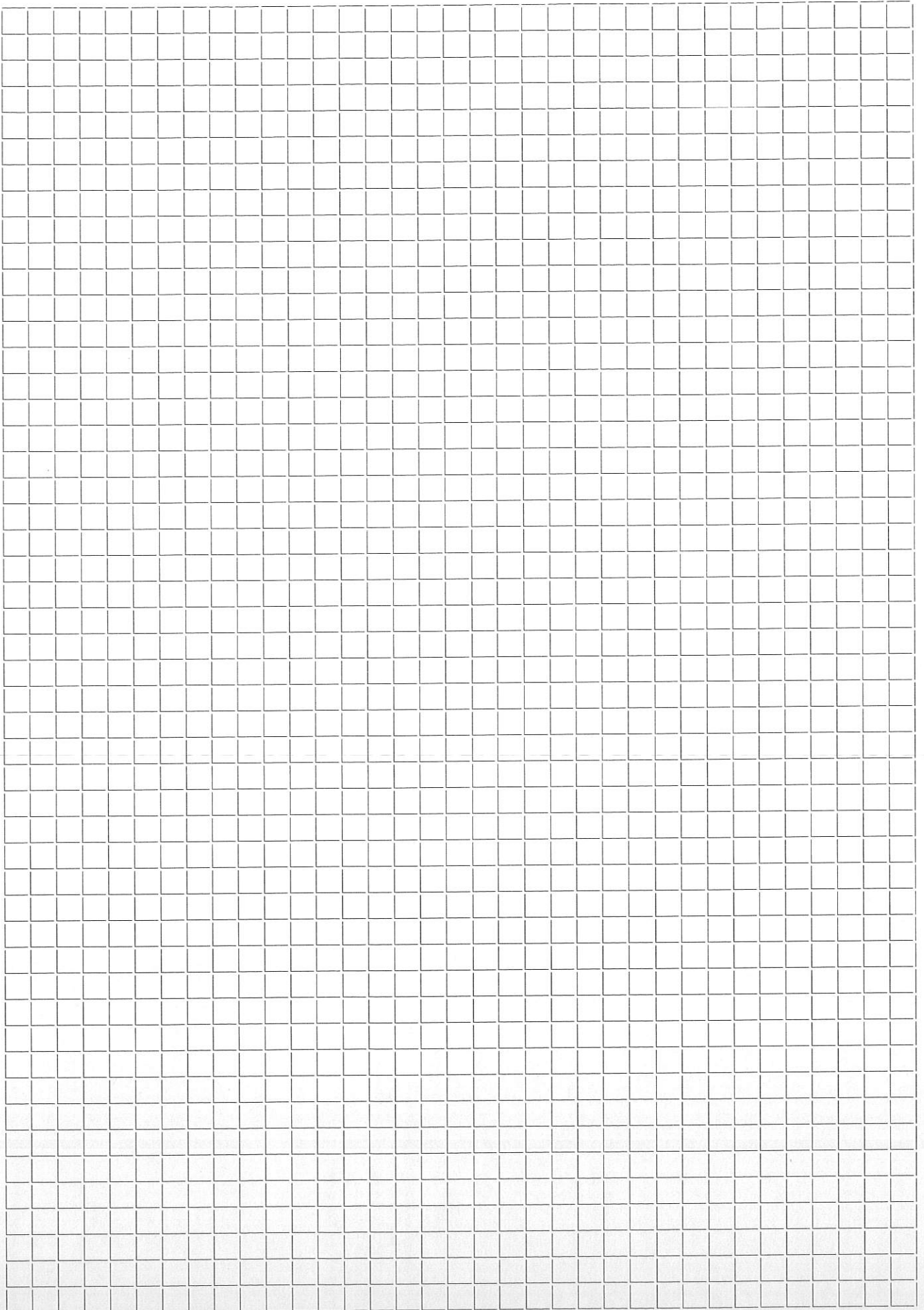
\uparrow sec \uparrow cm

For $L = 1m = 100cm$, $t = 387s = 6 \text{ min } 27s$

$L = 10m = 1000cm$, $t = 7621s = 127 \text{ min}$

$= 2 \text{ h } 7 \text{ min}$







3

Problem 4 = Adiabatic compression of the vapor

1/ $\dot{Q} = 120 \text{ W at } 1.8 \text{ K}$

$$\dot{Q} = \dot{m} \Delta H_L^V(1.8 \text{ K})$$

$$\Delta H_L^V(1.8 \text{ K}) = 24201 - 842 \approx 23360 \text{ J/kg}$$

$$\Delta H_L^V(1.8 \text{ K}) \approx 23.36 \text{ J/g}$$

$$\text{Hence } \dot{m} = \frac{\dot{Q}}{\Delta H_L^V} = \frac{120}{23.36} \approx 5.14 \text{ g/s} = 5.14 \times 10^{-3} \text{ kg/s}$$

2/ Volumetric flow \dot{V} is given by $\dot{V} = \frac{\dot{m}}{\rho}$

$$\rho(16.4 \text{ mb, } 4 \text{ K}) = 0.198 \text{ kg/m}^3$$

$$\rho(16.4 \text{ mb, } 290 \text{ K}) = 0.00272 \text{ kg/m}^3$$

$$\text{Hence } \dot{V}_{4 \text{ K}} = 0.026 \text{ m}^3/\text{s} = 93.5 \text{ m}^3/\text{h}$$

$$\dot{V}_{290 \text{ K}} = 1.89 \text{ m}^3/\text{s} = 6803 \text{ m}^3/\text{h}$$

3/ For an ideal gas

$$W_{\text{adiab}} = \frac{\gamma}{\gamma-1} P_1 V_1 \left[\left(\frac{P_2}{P_1} \right)^{\frac{\gamma-1}{\gamma}} - 1 \right]$$

$$\text{For He, } \gamma = \frac{5}{3} = 1.67 \Rightarrow \frac{\gamma-1}{\gamma} = 0.4$$

$$\frac{\gamma}{\gamma-1} = 2.5$$

$$W_{\text{adiab-4K}} = 2.5 \times 1640 \times 0.026 \times (61^{0.4} - 1) \approx 446 \text{ W} = 0.446 \text{ kW}$$

$$W_{\text{adiab-290K}} = 2.5 \times 1640 \times 1.89 \times (61^{0.4} - 1) \approx 32300 \text{ W} = 32.3 \text{ kW}$$



4/ Real helium

Adiabatic reversible $\rightarrow \delta Q = 0 \rightarrow \Delta S = 0$ isentropic

$$\Delta H + \dot{V}K = \Delta S + \Delta W$$

The compression work is equal to the enthalpy difference, at constant entropy.

$$\left. \begin{aligned} h(16.4 \text{ mbar}, 4 \text{ K}) &= 26.00 \text{ kJ/kg} & s(16.4 \text{ mbar}, 4 \text{ K}) &= 14.13 \text{ kJ/kg K} \\ h(16.4 \text{ mbar}, 290 \text{ K}) &= 1511 \text{ kJ/kg} & s(16.4 \text{ mbar}, 290 \text{ K}) &= 36.39 \text{ " "} \end{aligned} \right\}$$

Hold s constant \Rightarrow $h(1 \text{ bar}, s=14.13) = 112.46 \text{ kJ/kg}$
 Corresponding temperature 20.8 K

$h(1 \text{ bar}, s=36.39) = 7804 \text{ kJ/kg}$
 Corresponding temperature 150.2 K

Hence, with inlet @ 4 K $\Delta h = 112.46 - 26.00 = 86.46 \text{ kJ/kg}$
 and $\dot{W} = \dot{m} \Delta h = 5.14 \times 86.46 = 444 \text{ W}$

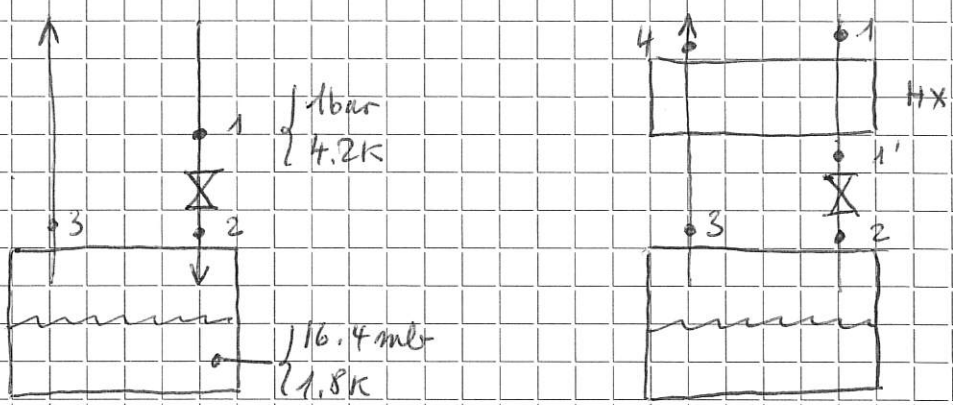
With inlet @ 290 K, $\Delta h = 7804 - 1511 = 6293 \text{ kJ/kg}$
 and $\dot{W} = \dot{m} \Delta h = 5.14 \times 6293 = 32346 \text{ W}$

\Rightarrow The ideal gas approximation is fully justified.



4

Problem 3 = Saturated superfluid by J-T expansion



1/ Saturation pressure of He at 1.8 K = 1638.41 Pa
 $\approx 16.4 \text{ mbar}$

2/ ^{Adiabatic} Expansion without production of motor work \equiv isenthalpic

$$\Delta U + \Delta K = \Delta Q + \Delta W \quad \Delta W = -\Delta(PV)$$

negligible $\frac{U}{c}$

$$\Rightarrow \Delta(U + PV) = 0$$

$$3/ \begin{cases} h(1 \text{ bar}, 4.2 \text{ K}) = 9902 \text{ J/kg} \\ h_L(\text{sat}, 1.8 \text{ K}) = 842 \text{ J/kg} \\ h_V(\text{sat}, 1.8 \text{ K}) = 24204 \text{ J/kg} \end{cases}$$

Conservation of enthalpy can be written

$$h(1 \text{ bar}, 4.2 \text{ K}) = x h_V(\text{sat}, 1.8 \text{ K}) + (1-x) h_L(\text{sat}, 1.8 \text{ K})$$

with x = vapour fraction produced in the expansion.

Hence $x = 0.39$

$\left. \begin{array}{l} 39\% \text{ vaporized} \\ 61\% \text{ remaining liquid} \end{array} \right\}$



4/ - In the high-pressure stream of the HX, the liquid can be subcooled not lower than the lambda point: ~~the~~ the part of this stream which would contain superfluid would be a thermal short circuit.

Hence, $T(1') = 2.18 \text{ K}$

- The temperature of the returning saturated vapor is by definition 1.8K

Hence, $T(3) = 1.8 \text{ K}$.

$$h(1 \text{ bar}, 4.2 \text{ K}) - h(1 \text{ bar}, 2.18 \text{ K}) = \Delta h_{HP}$$

$$h(16 \text{ mbar}, T_4) - h(16 \text{ mbar}, 1.8 \text{ K}) = \Delta h_{LP}$$

$$\Delta h_{HP} = 9302 - 3634 = 6268 \text{ J/kg}$$

$$\Delta h_{LP} = \Delta h_{HP}$$

$$h(16 \text{ mbar}, T_4) = 6268 + 2420.1 = 30469 \text{ J/kg}$$

Hence $T_4 \approx 3.0 \text{ K}$

- Fraction of vapor produced in expansion x

$$h(1 \text{ bar}, 2.18 \text{ K}) = x h(16.4 \text{ mbar}, 1.8 \text{ K}) + (1-x) h_L(16.4 \text{ mbar}, 1.8 \text{ K})$$

$$3634 = x \times 2420.1 + (1-x) \times 842$$

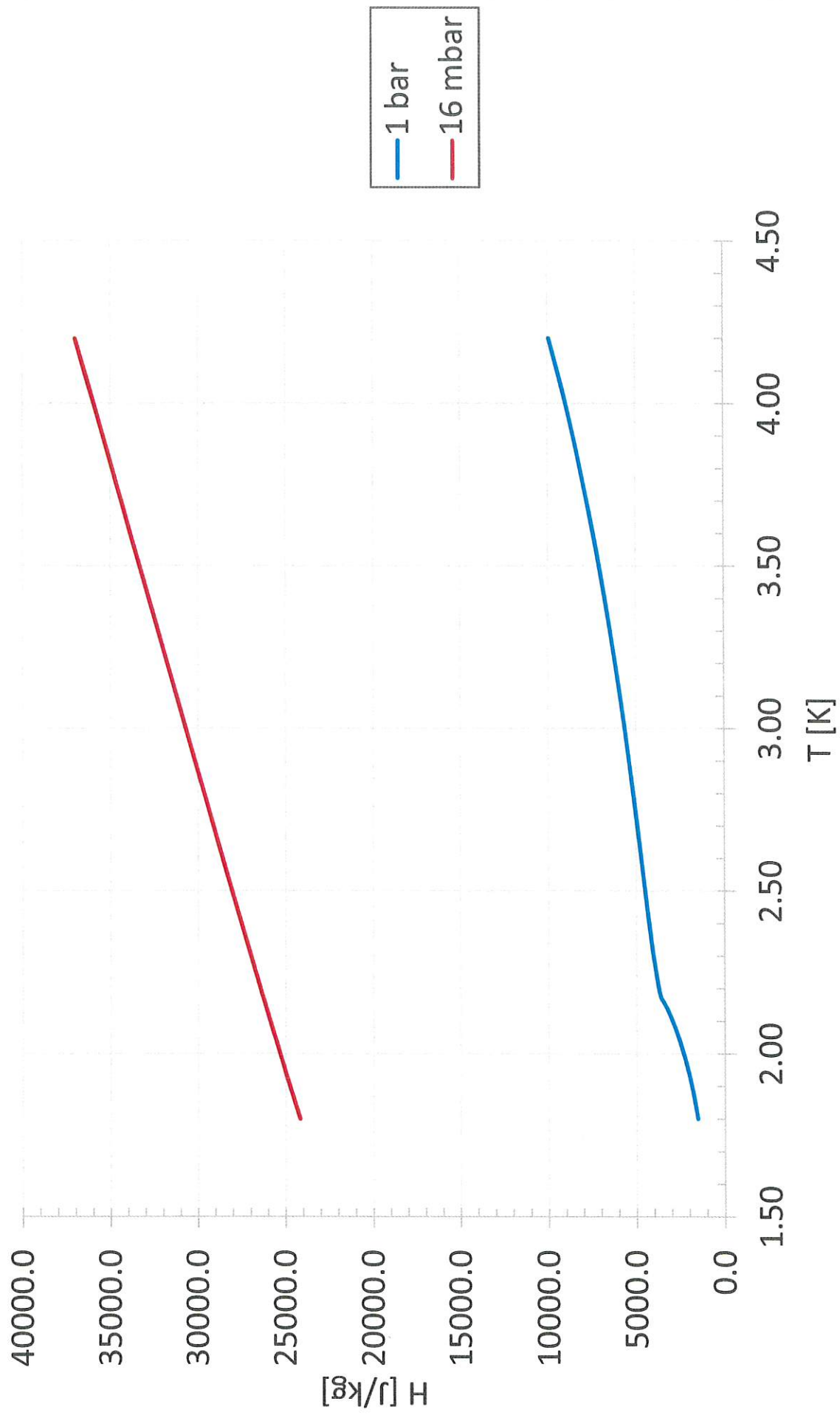
Hence $x = 0.12$

} 12% vaporized
} 88% remaining liquid

5/ Challenges of the HX

- High efficiency (at small T differences between HP and LP streams)
- Very low pressure drop on the LP stream, typically $< 1 \text{ mbar}$.

Enthalpy of helium at 1 bar and 16 mbar



120 505

0.0

0.00001

0.00001

0.00001

0.00001

0.00001

0.00001

0.00001

0.00001

H (N²)

0.00001