



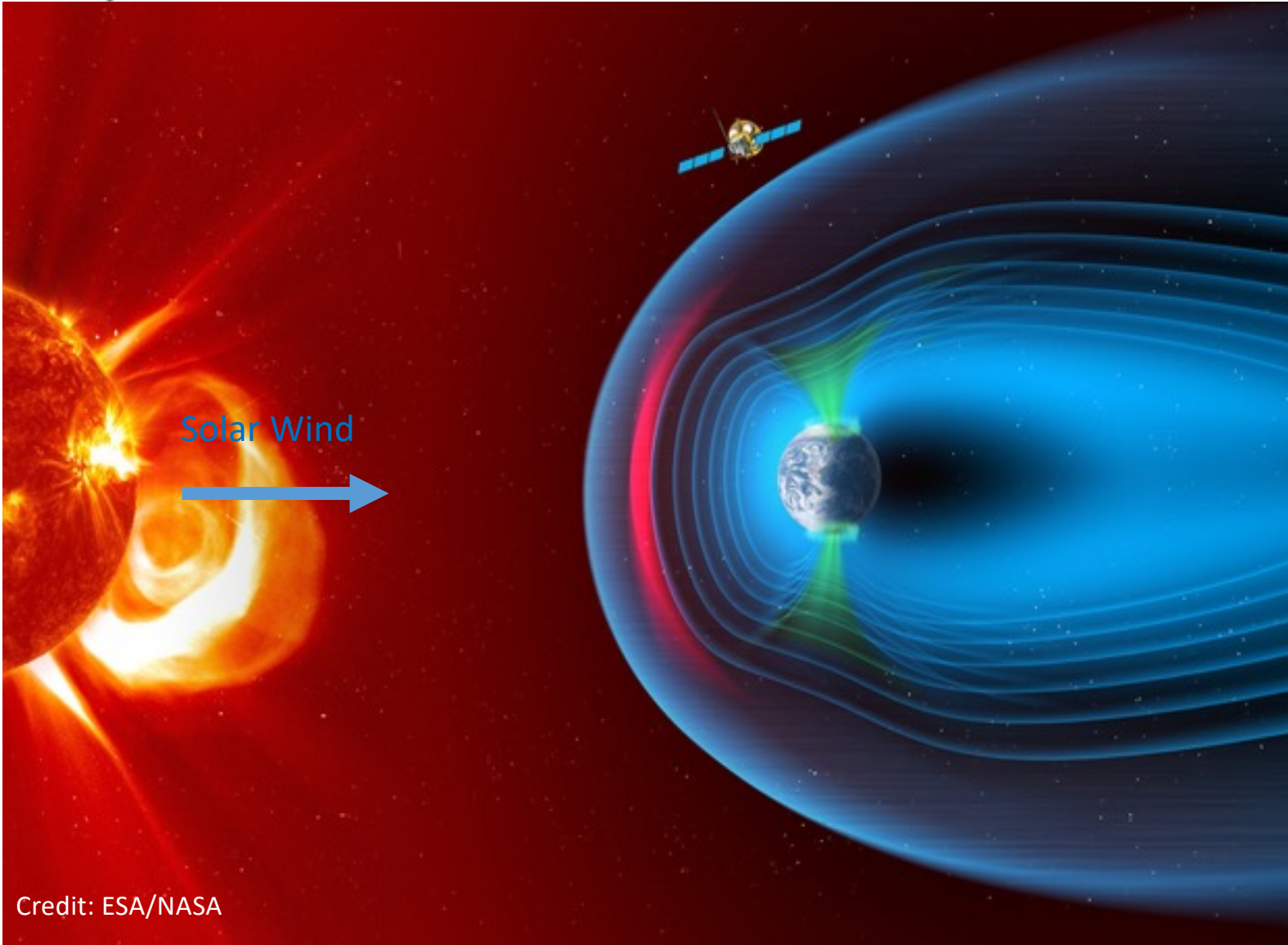
Imaging Earth's magnetosphere with the Soft X-ray Imager (SXI) on the SMILE mission

Authors:

Steve Leach, Steve Sembay (PI) and the SMILE SXI Instrument Team

Solar wind Magnetosphere Ionosphere Link Explorer: SMILE

SMILE is a collaborative science mission between the European Space Agency (ESA) and the Chinese Academy of Sciences (CAS)



Credit: ESA/NASA

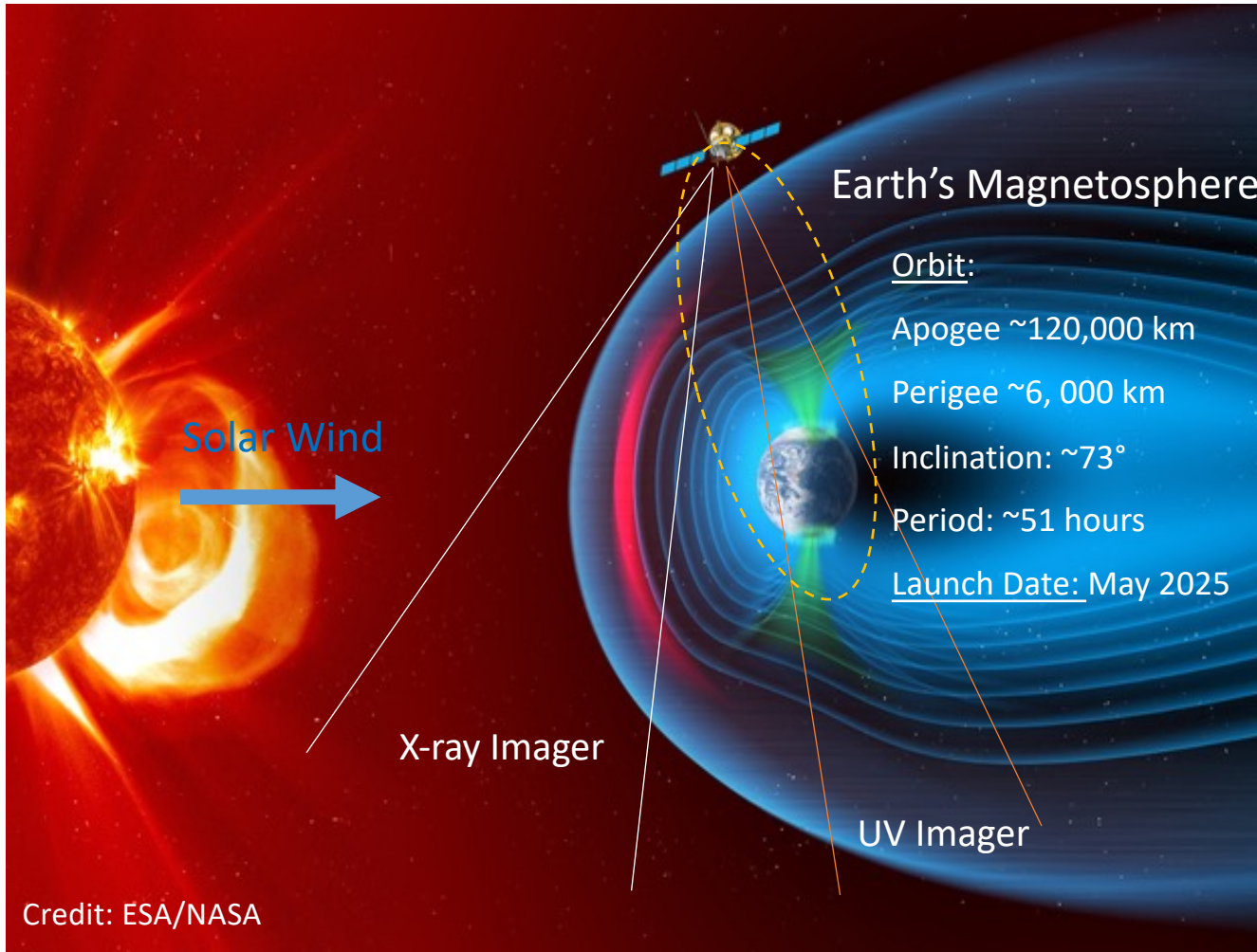
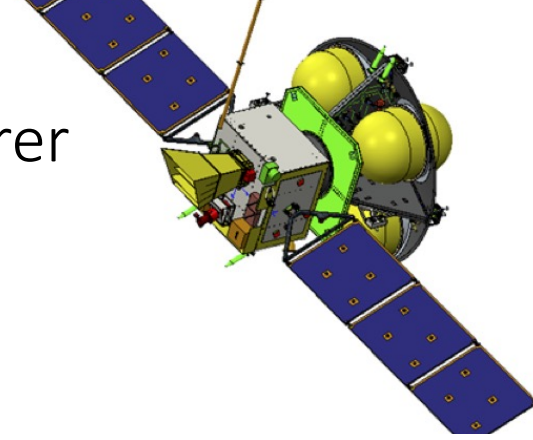
- The **solar wind** and embedded interplanetary magnetic field (IMF) interacts with the Earth's magnetic field (**magnetosphere**)
- Dipolar terrestrial field → direct entry of solar wind plasma into Earth **ionosphere** (ionised upper atmosphere)
- Several ground/space instruments have studied independently
- Simultaneous data required on multiple temporal and spatial scales (**explore the link**)



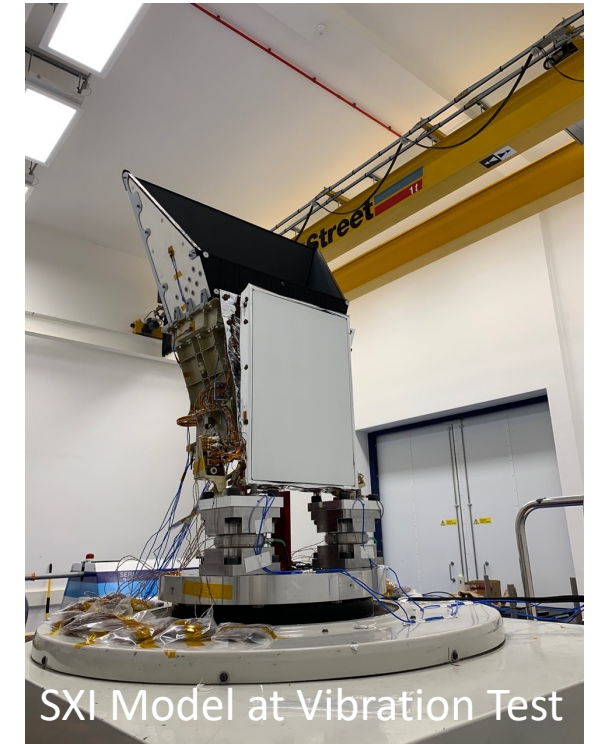
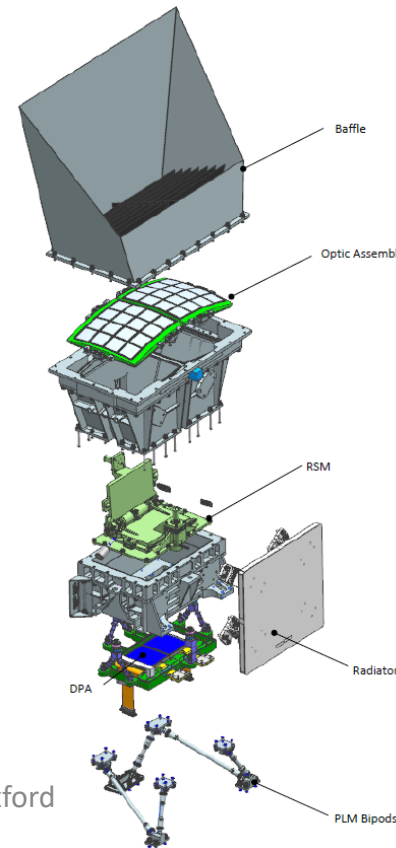
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Solar wind Magnetosphere Ionosphere Link Explorer

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Soft X-ray Imager (SXI) is led by the University of Leicester in collaboration with other UK and European Institutions

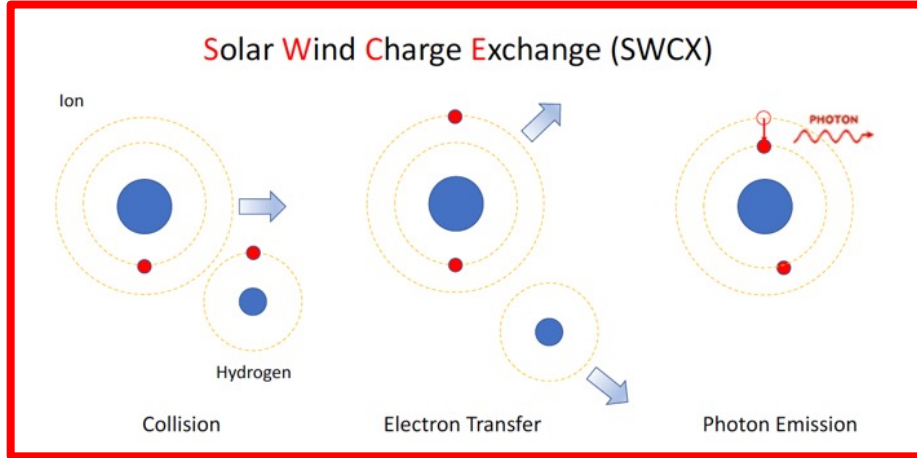




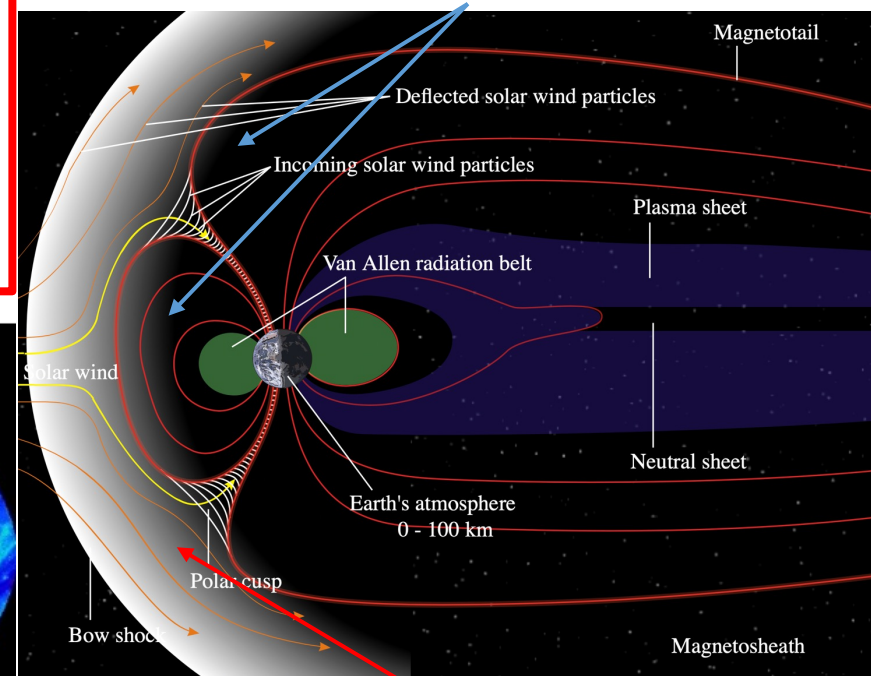
Global imaging using Solar Wind Charge Exchange (SWCX)



SWCX: High charge state solar wind ions in collision with hydrogen in the Earth's exosphere to produce photons at X-ray energies

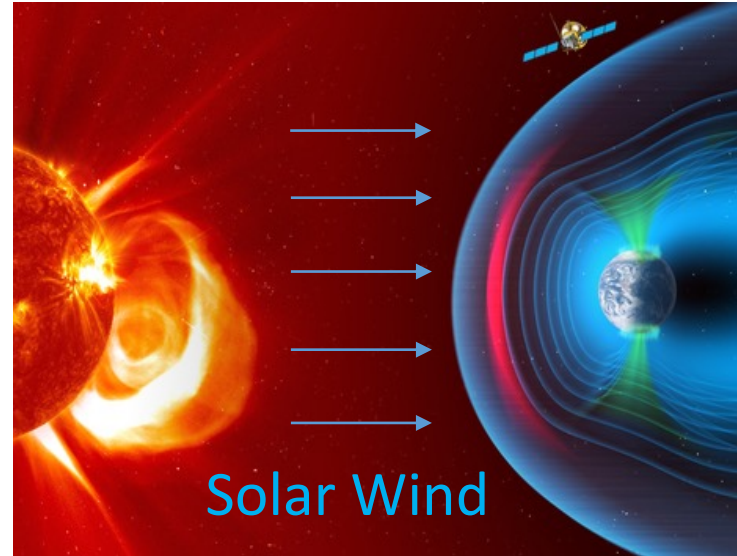


No Solar Wind in these regions therefore no SWCX X-rays

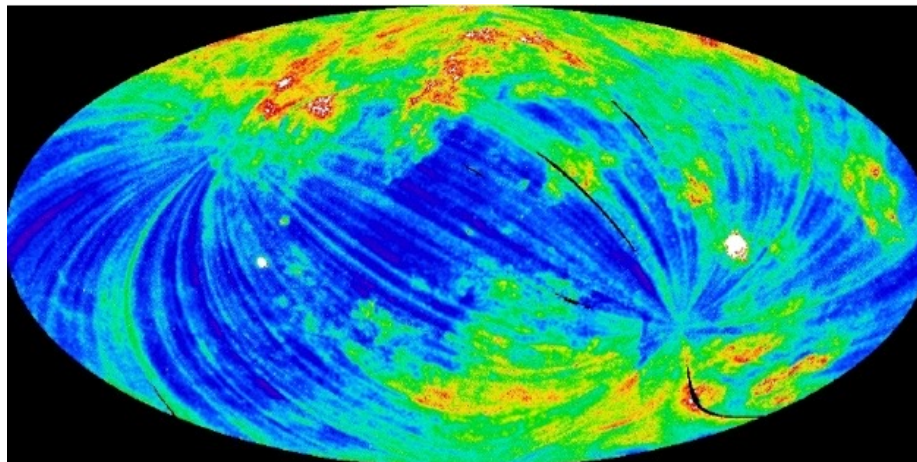


Solar Wind in these regions therefore SWCX X-rays

Hence: magnetopause boundary can be detected using X-ray imaging



Solar Wind: electrons & protons with some (~2%) heavy ions such as carbon, nitrogen and oxygen embedded in the Sun's magnetic field



ROSAT All-Sky Survey in 1990 (Scanning observations) SWCX Can be brighter than X-ray background

Soft X-ray Imager (SXI) CONSORTIUM



Steve Sembay (SXI PI)
University of Leicester, UK



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- All hardware and flight SW is supplied by UK and European Institutions including ESA
- US and China contribute some ground software and science support



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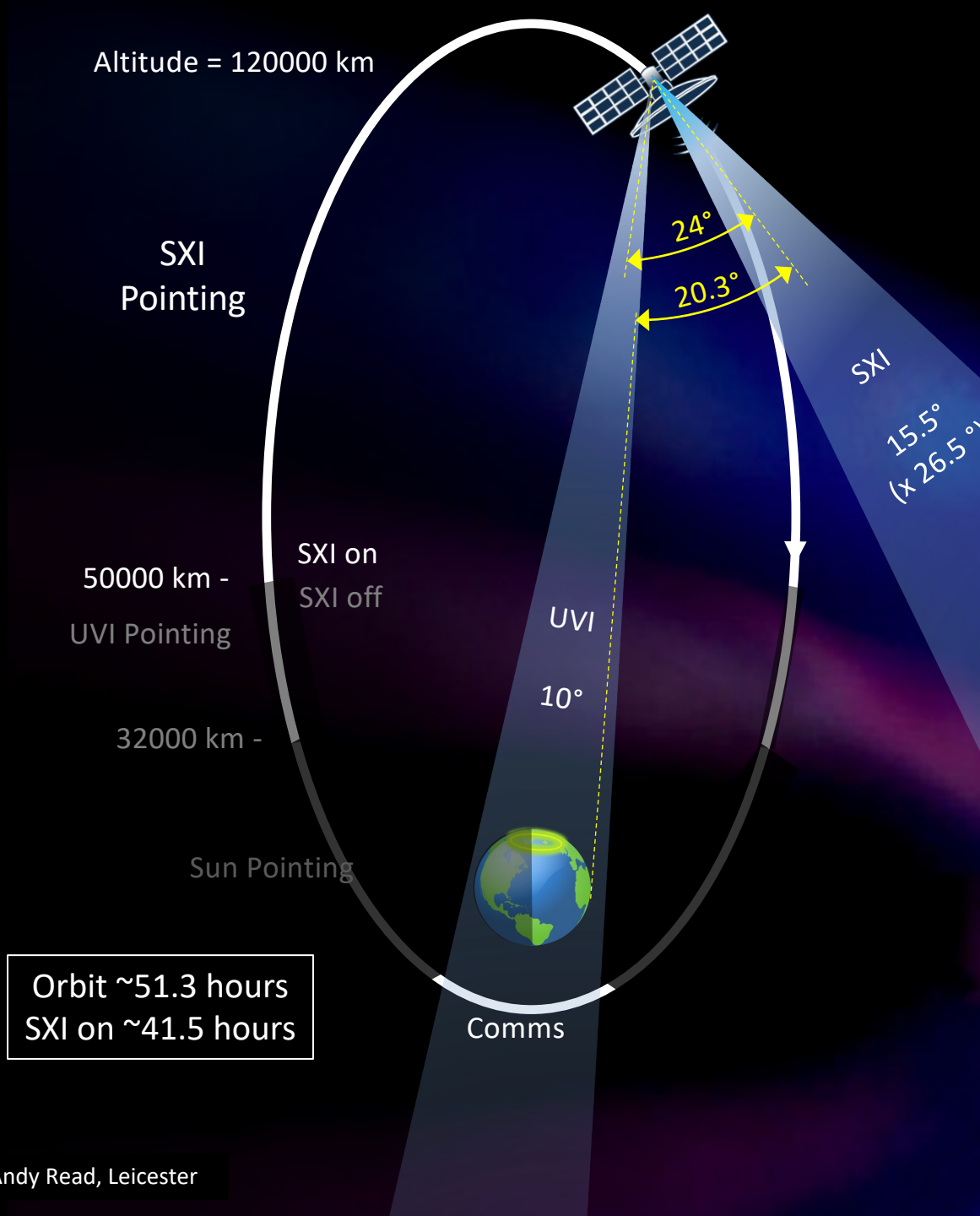
SPACE
ACOUSTICS

Nsse

SMILE

Solar wind Magnetosphere
Ionosphere Link Explorer

- Soft X-ray Imager (SXI) is pointed at a constant angular distance (20.3°) from the Earth limb, and along the Earth-Sun line
- SXI footprint on cosmic X-ray sky is pre-determined - known in advance, & SXI FOV is large
- All SXI background images/spectra have potential for secondary science

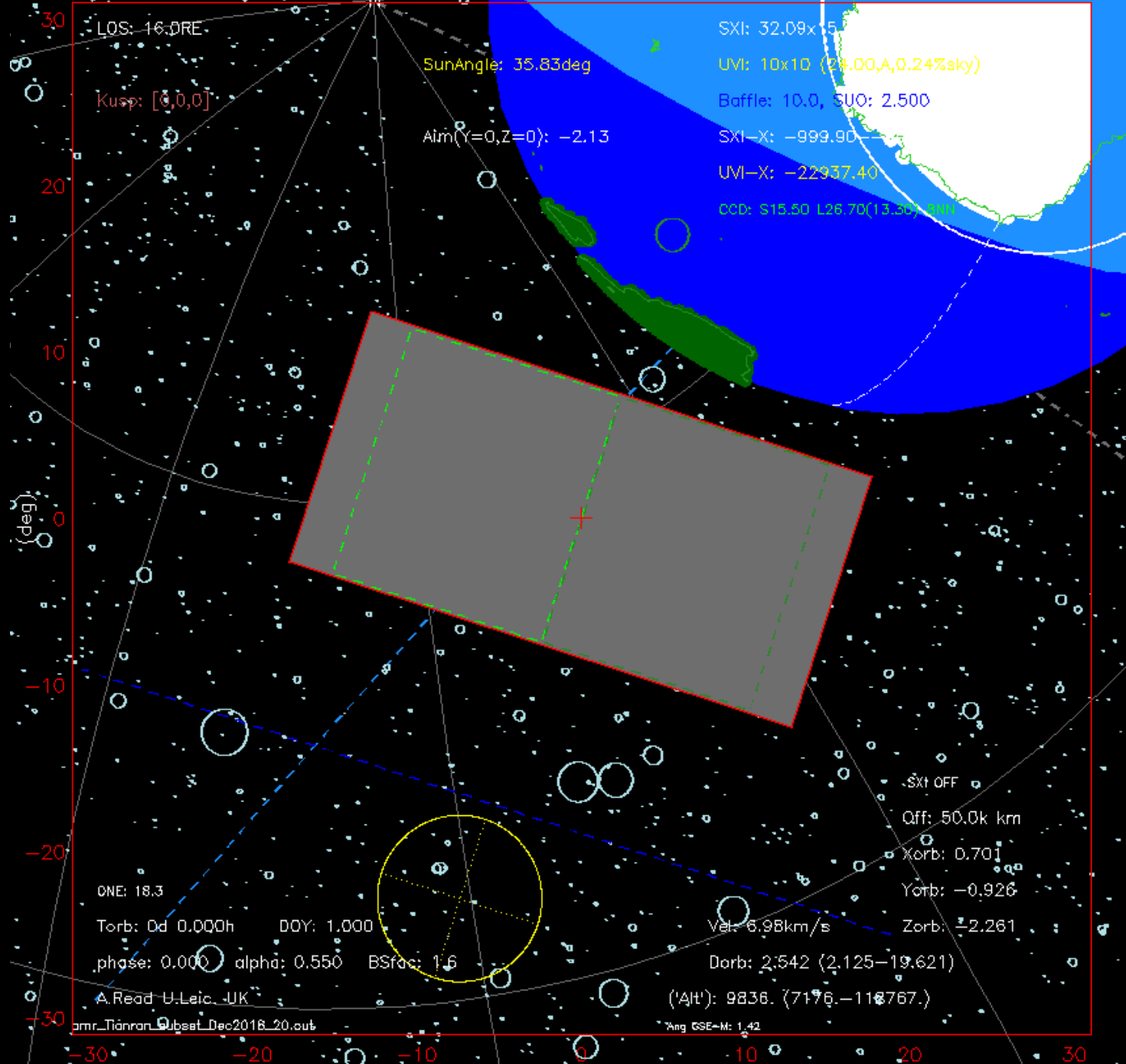


SMILE Orbit simulations

Secondary/serendipity science observations and potential links with other operational space telescopes

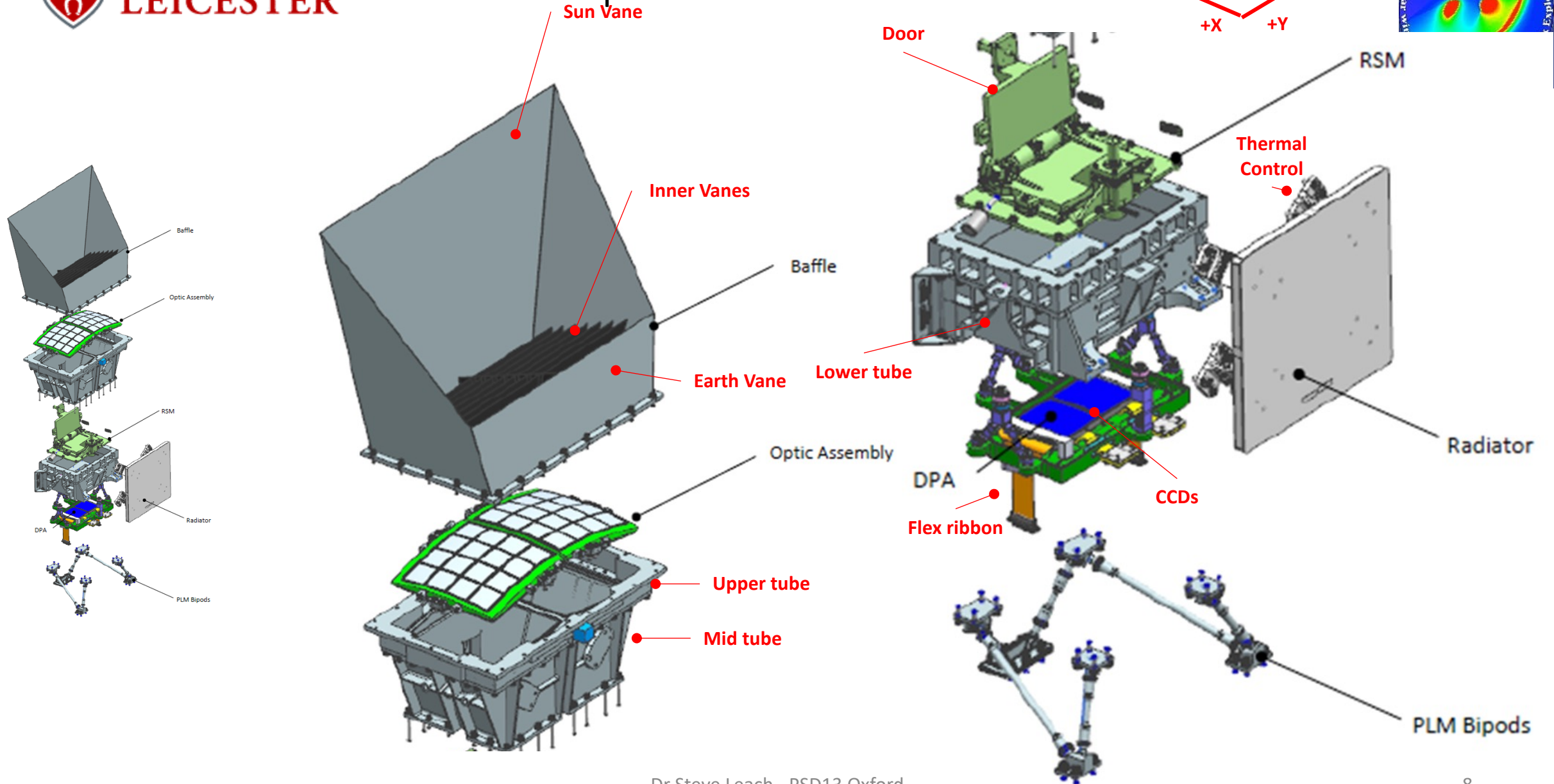
- One typical SMILE orbit (~51.3hr, May 2025)
 - Apogee semi-towards Sun
- Large central rectangle: SXI FOV
 - Active FOVs of 2 CCDs
- Red/orange/yellow emission: Science Targets – SWCX X-ray emission from Magnetosheath & bright Cusps (static MHD model shown)
- UVI (yellow circle) observes Earth
- Many bright cosmic X-ray sources pass through SXI FOV
- White circles: ROSAT Bright Source Catalogue [BSC] (size~brightness)
- Contributing to BG (need to model)

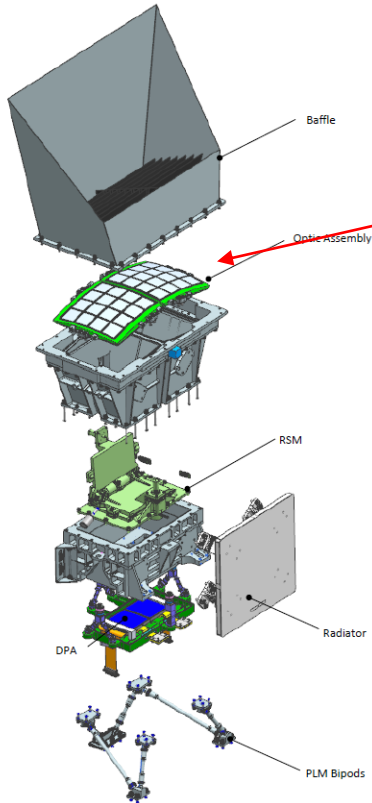
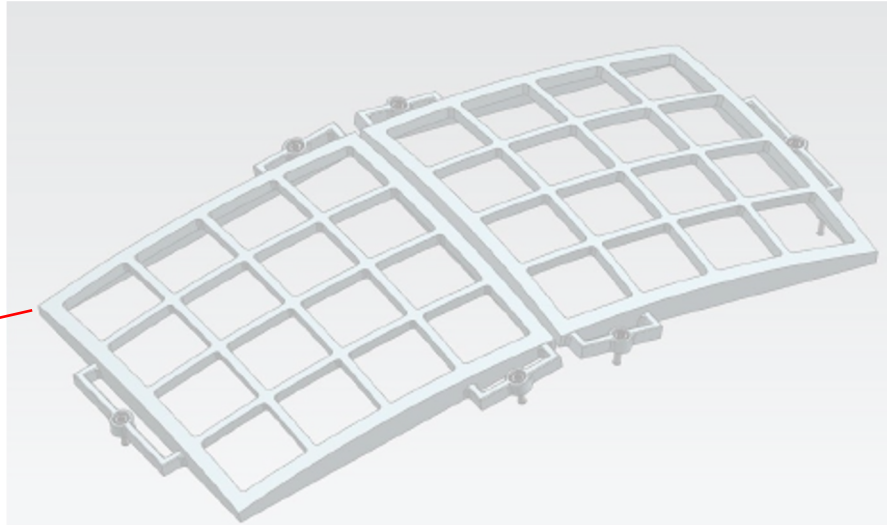
Sources used for dynamic calibration



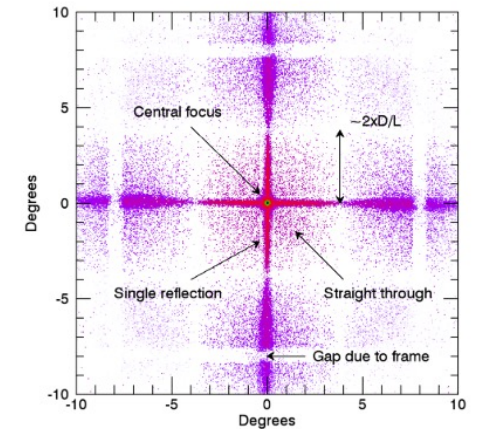


SXI Telescope





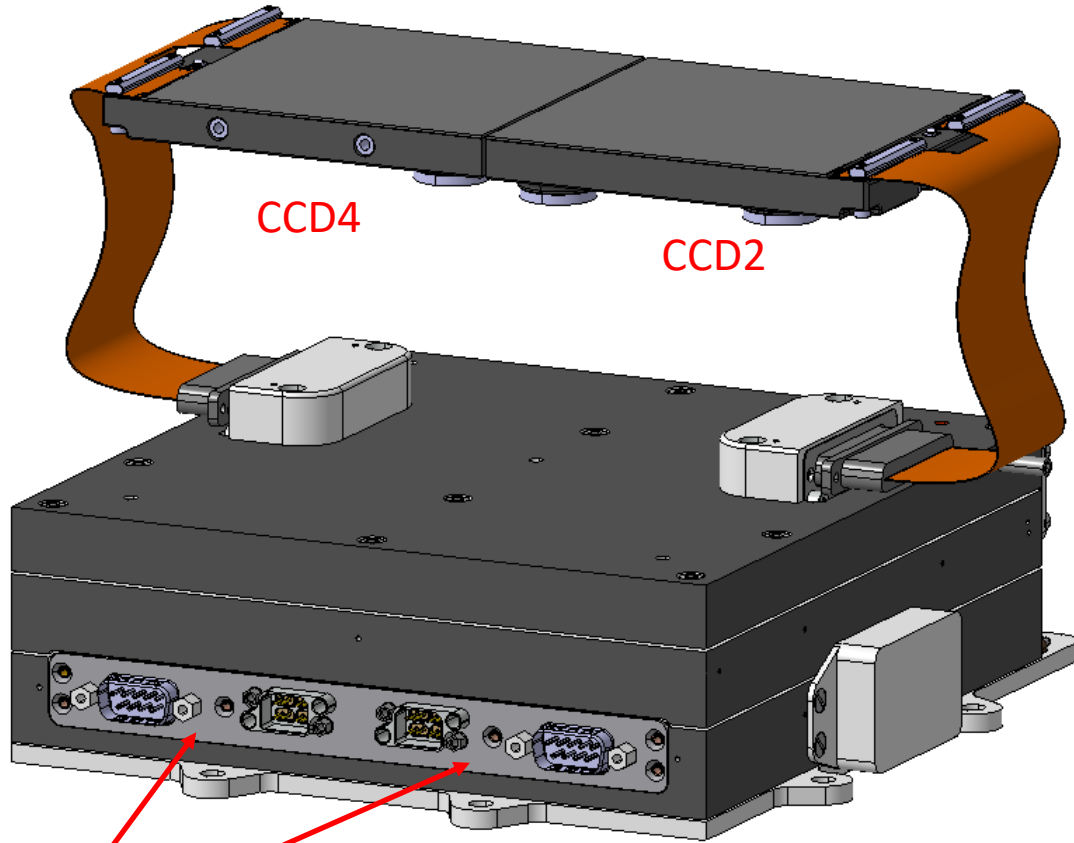
- Soft X-ray focusing using micropore optics (MPOs)
 - Manufactured by Photonis
 - Each MPO 40 x 40 mm²
 - 100 nm Aluminum film (to block optical straylight)
- Optical Assembly holding 16 MPOs (x2)
- SXI wide-field imaging in the soft X-ray (~0.15 to 1.1 keV) band
- Qualification completed, FM calibration in progress at Leicester’s facility



SXI Detector Front-End Electronics (FEE)



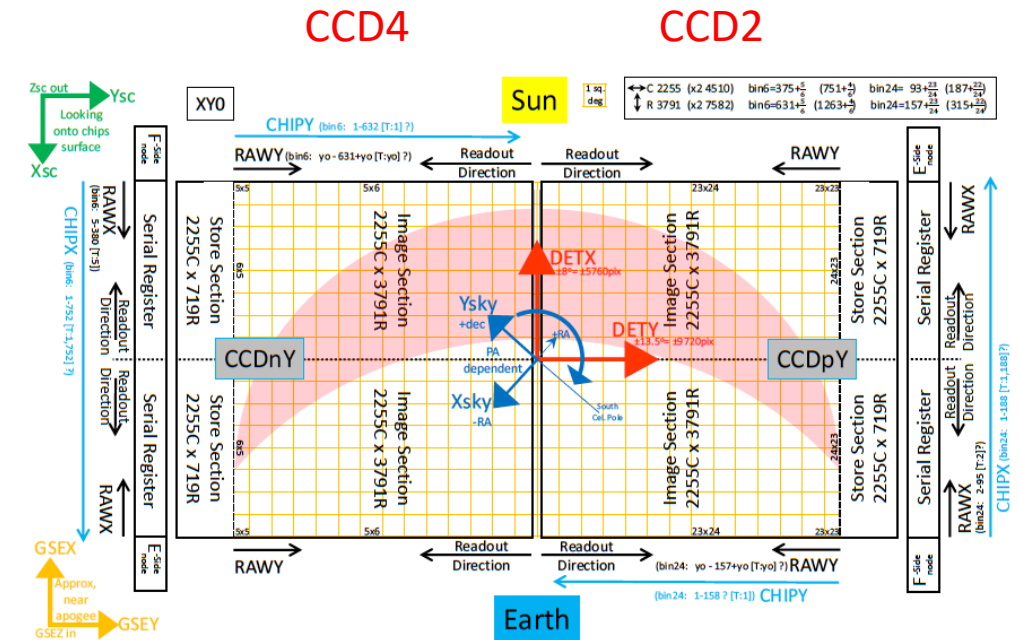
- Two large format Te2v CCDs
- Passively cooled by radiator and heaters
- -95 C to -120 C (-117 C nominal)
- Back-side illuminated
- 4510 x 4510 native 18 μm pixels (FF)
- FT 6x6 binning, 719 frame store + shield



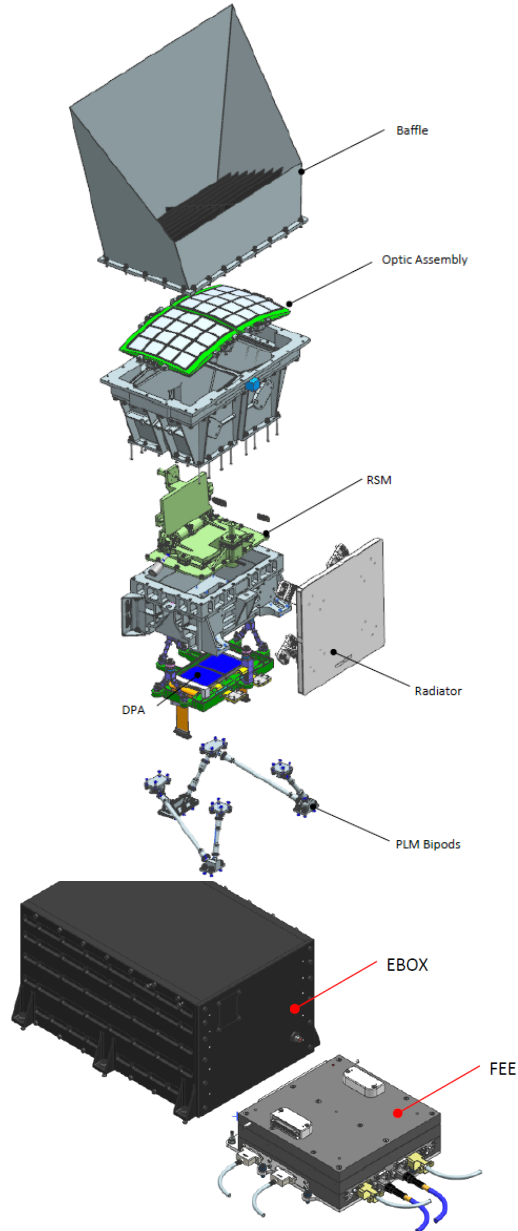
Nominal and Redundant

FEE and CCDs have PLATO heritage (4x CCDs)
FEE is being developed at MSSL

Credit: MSSL



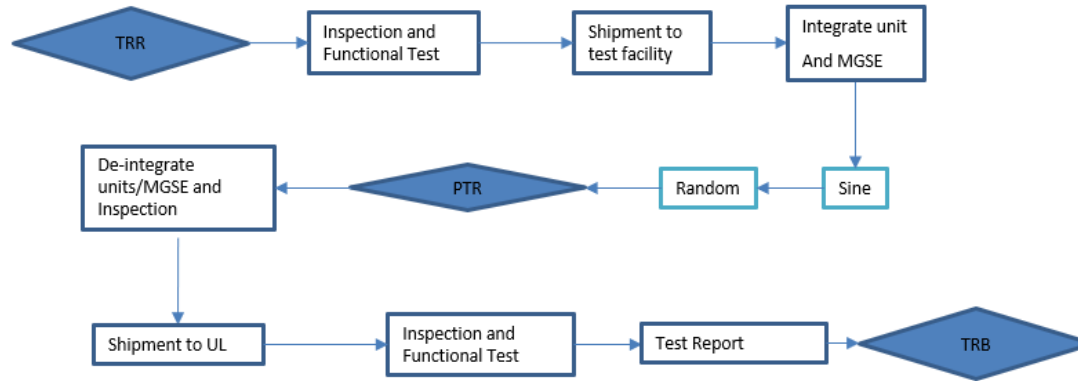
SXI Hardware Model Versions



Model	Description	Objective (Instrument level)	Objective (PLM level) (EID-1903/R-MIS-PROG-9250 AD1)	Delivery
BB Electronics System <ul style="list-style-type: none"> • BBV1 FEE • BBV2 FEE • BB EBOX • BB Harness • BBV1 RSM • BBV2 RSM • BBV1 RSE 	Development models of electronic elements	<ul style="list-style-type: none"> • Development, function, performance, HW/SW interaction 	-	Internal only
STM Instrument <ul style="list-style-type: none"> • STM Telescope • STM FEE • STM EBOX • STM Harness 	Represents structural and thermal system	<ul style="list-style-type: none"> • Structural qualification of the Telescope • Thermal qualification 	STM S/C integration	PLM
EQM Instrument <ul style="list-style-type: none"> • EQM Telescope (partial) • EQM FEE • EQM EBOX • EQM Harness 	Represents electronic system	<ul style="list-style-type: none"> • Functional verification • EMC/EMI verification • Grounding • ESD 	Support the verification of key electrical I/Fs with the PLM	PLM (Loan of FEE, EBOX and Harness)
PFM SXI Instrument <ul style="list-style-type: none"> • PFM Telescope • FM FEE • FM EBOX • FM Harness 	SXI Instrument for flight	<ul style="list-style-type: none"> • Acceptance 	Integration in FM S/C	PLM



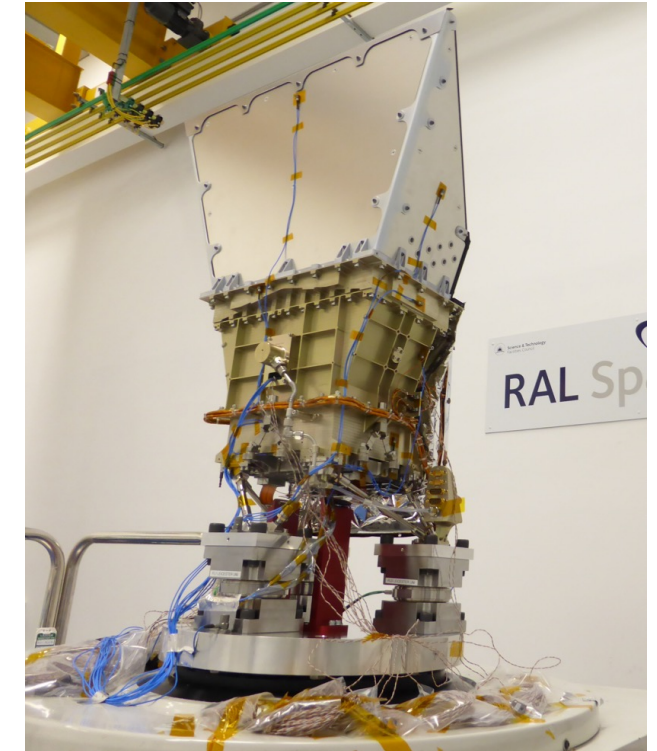
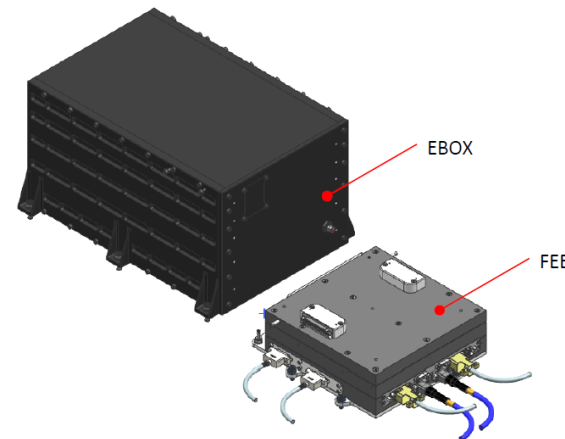
- Vibration, Thermal and EMC



Subsystem level (DPU, FEE), then instrument level

Vibration

- Sine & random vibration in x-y-z axes



STM Telescope vibration test Setup,
PFM scheduled at RAL for
Sep-Oct 23



- Vibration, Thermal and EMC

Thermal balance

- Hardware in vacuum to simulate thermal conditions of space
- Instrument surrounded by thermally controlled plates
- Verify the thermal control subsystem
- Correlate with SXI thermal model

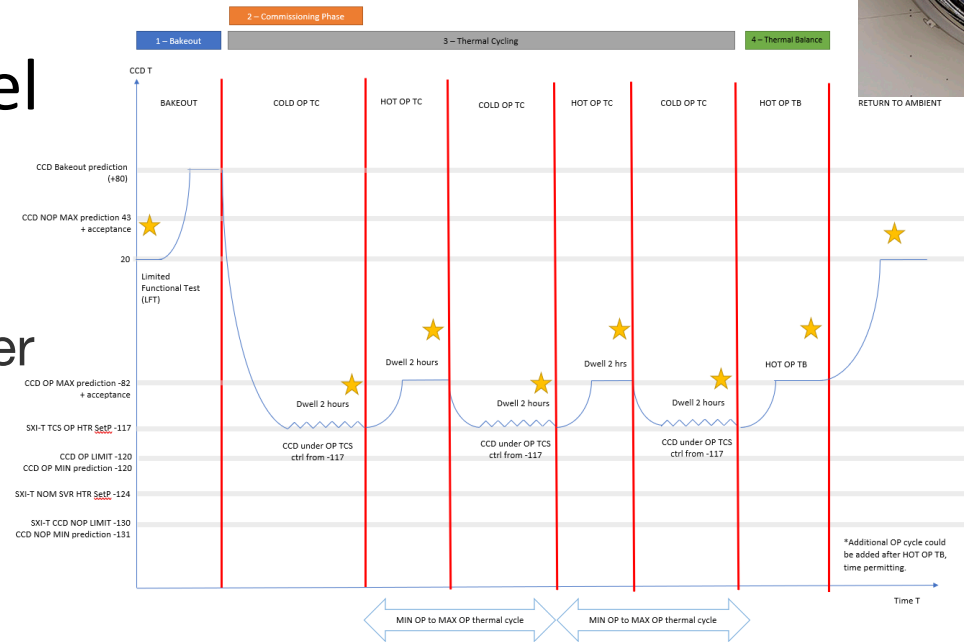


RAL Space TVAC Facility

Now at instrument level

Thermal Balance

- Thermal test cycle
- Verify CCDs before/after



29-day campaign, scheduled at RAL for **Nov-Dec 23**



• Vibration, Thermal and EMC

EQM Instrument 'FlatSat' EMC testing completed Dec '22

Multiple failures. Most significant – Radiated Emissions (RE) in S-Band comms frequency

Unit level EMC tests show both EBox and FEE contribute to RE test failure

Investigation (at ESTEC July '23) making enclosures

'RF tight' via gaskets shows promise.

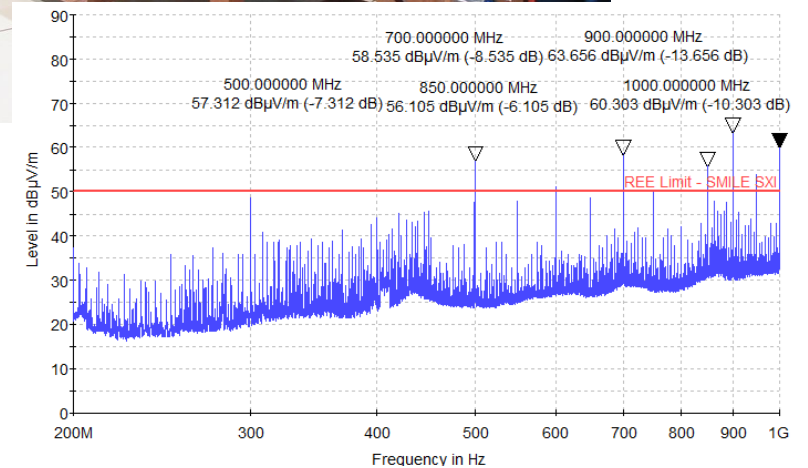
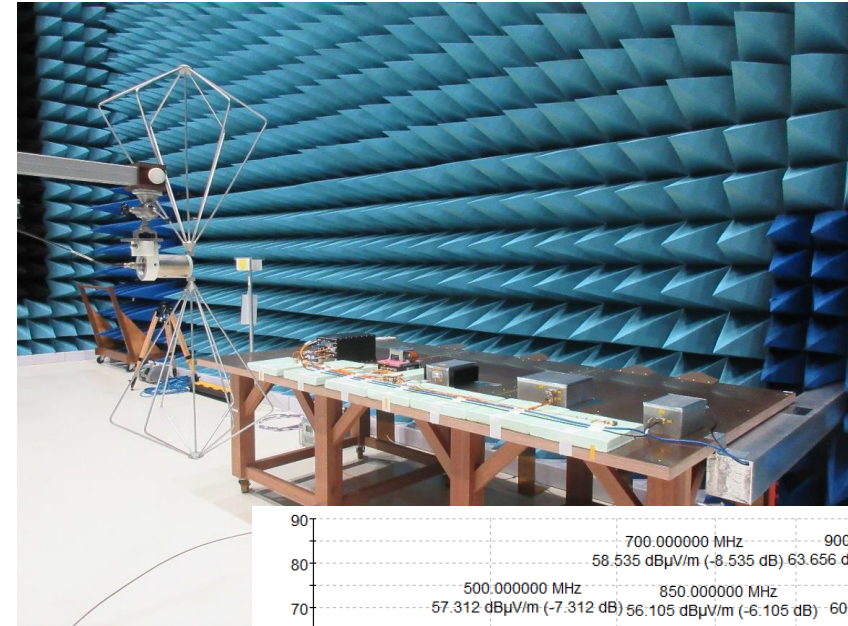
EMC

Unit level system retest of EQM FEE to assess impact of the design changes (in progress)

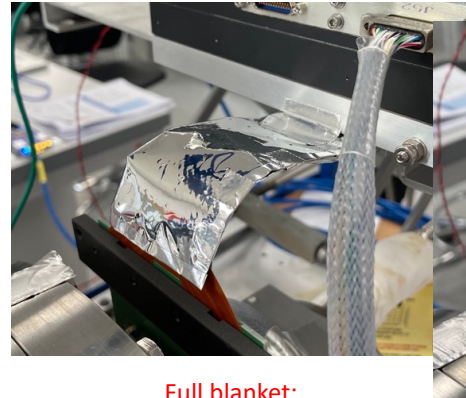
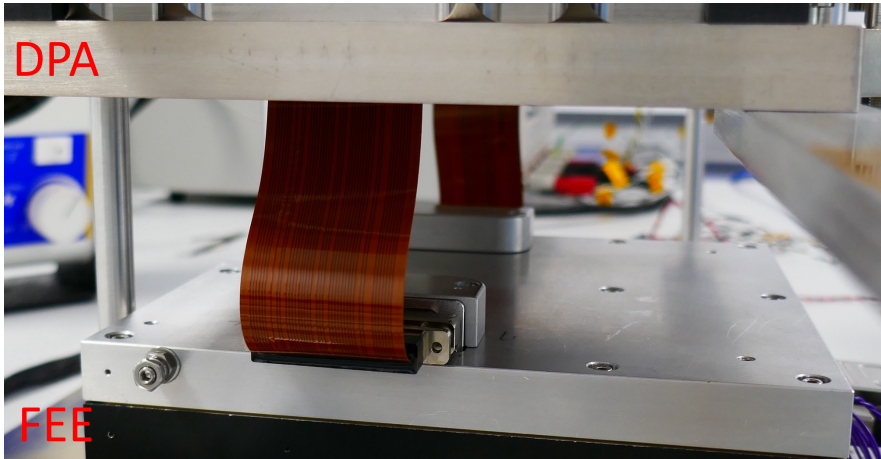
Unit level test of FM FEE with EMC modifications (in progress)

Unit level test of FM EBox with EMC modifications (TBD)

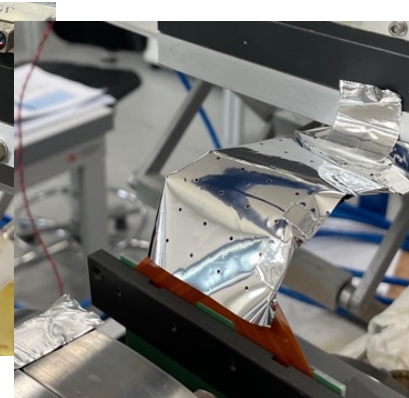
SXI Instrument FM test (~Jan '24)



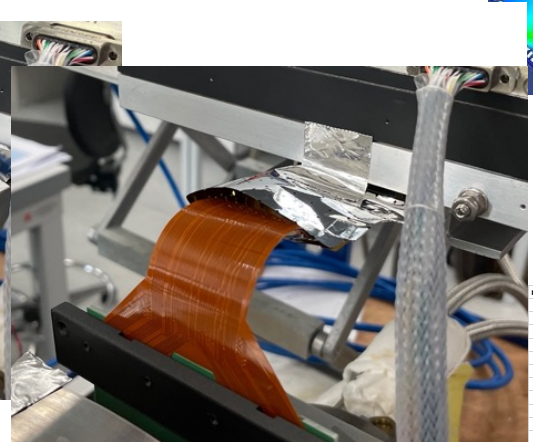
Instrument level verification
EMC planned at ESTEC **~Jan 24**



Full blanket:
Multiple folded layers,
outside GND to FEE

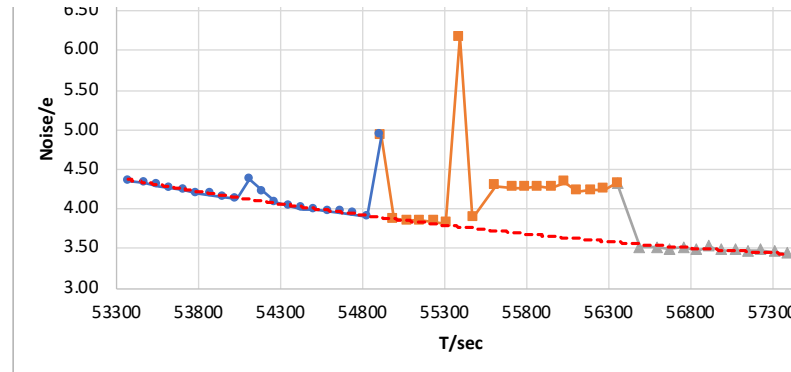
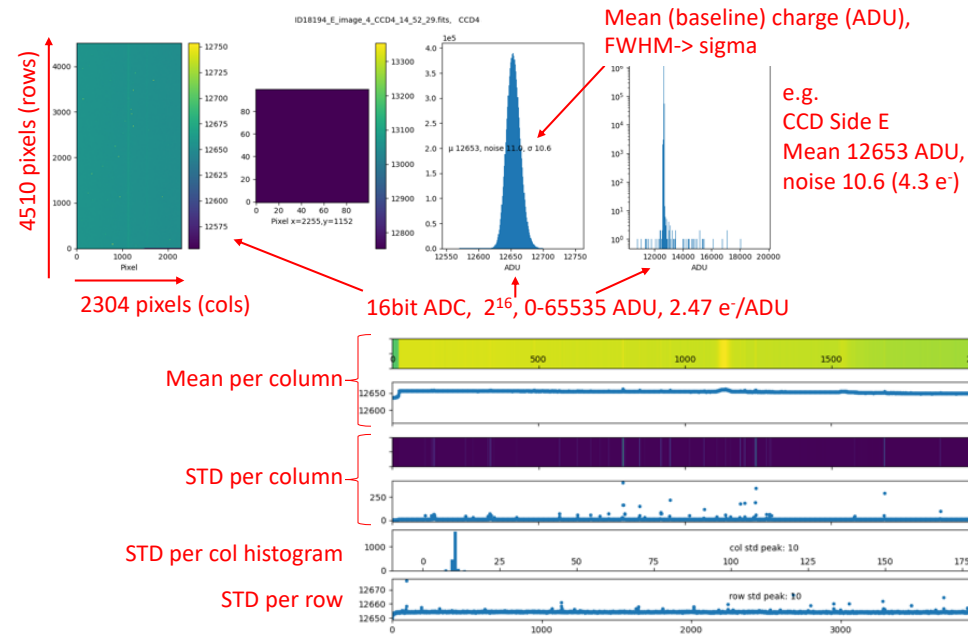


Shield only:
Single layer,
GND to FEE



Half shield only:
Single layer,
GND to FEE

Analysis method, Full-Frame (FF) mode, 2 sides (E & F) per CCD



- Noise follows Residual Charge Decay
- Anomaly caused by handling flex/comms loss
- No measured increase in noise with screening

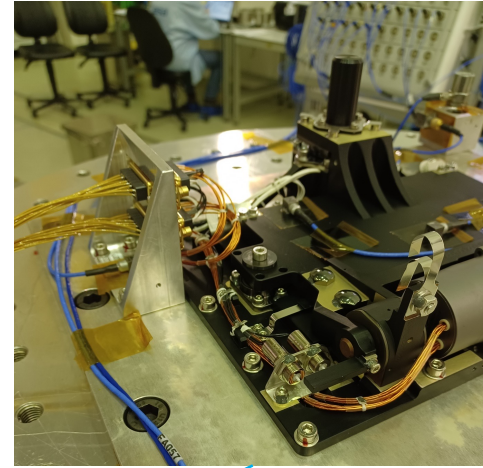
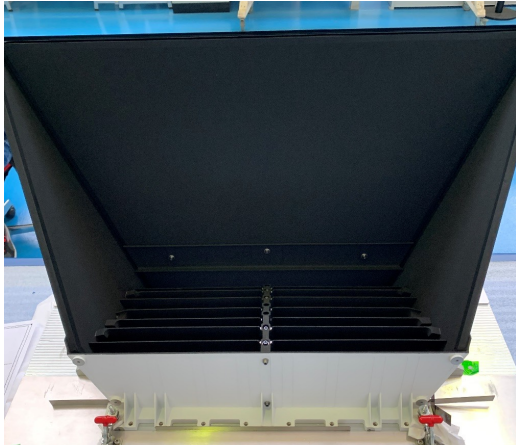
Screen	MeanNoise/ADU	Noise/e
NONE	10.5	4.23
NONE		
BLANKET		9.8
NONE		
NONE	9.5	3.85
SHIELD		
SHIELD	10.6	4.27
NONE		
NONE	10.5	4.24
HALF		
HALF	8.7	3.51
NONE		
NONE	8.6	3.47



SXI Hardware Status (as July)



Baffle



Radiation Shutter Mechanism

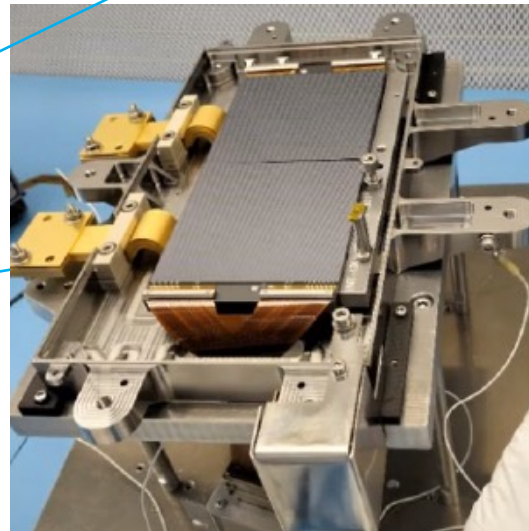
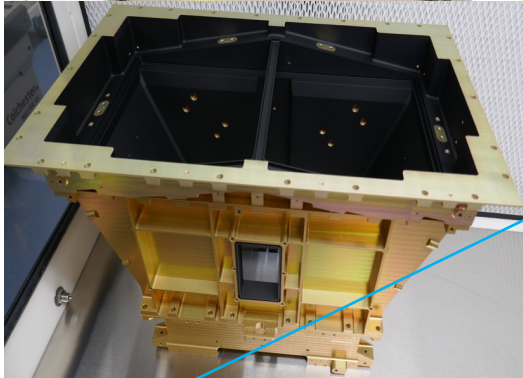


Radiator

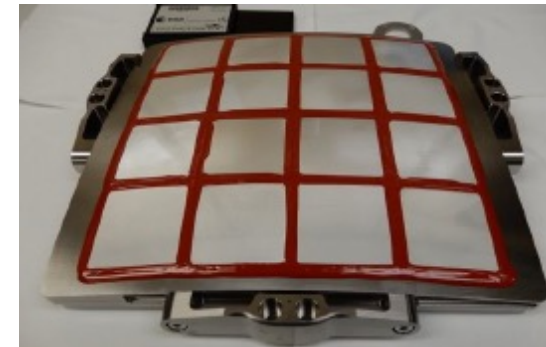


Electronics Box (DPU) (EQM)

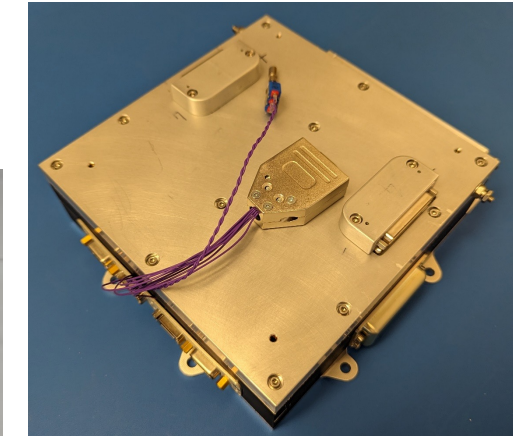
Primary Structure



Detector Plane Assembly

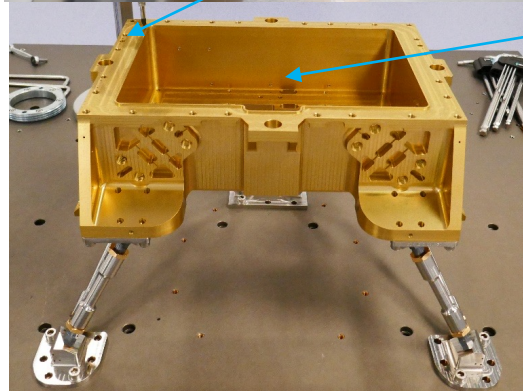


Optic Assembly x2

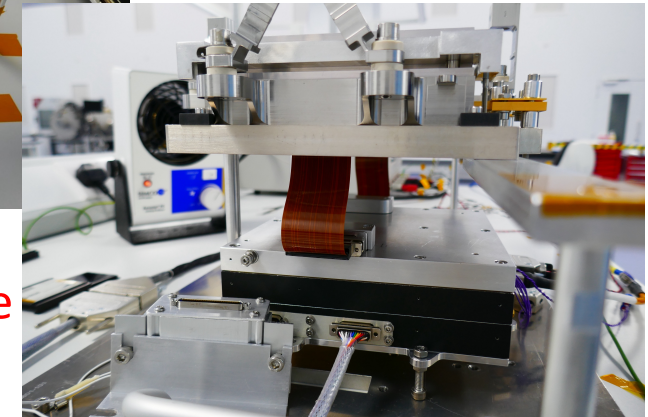
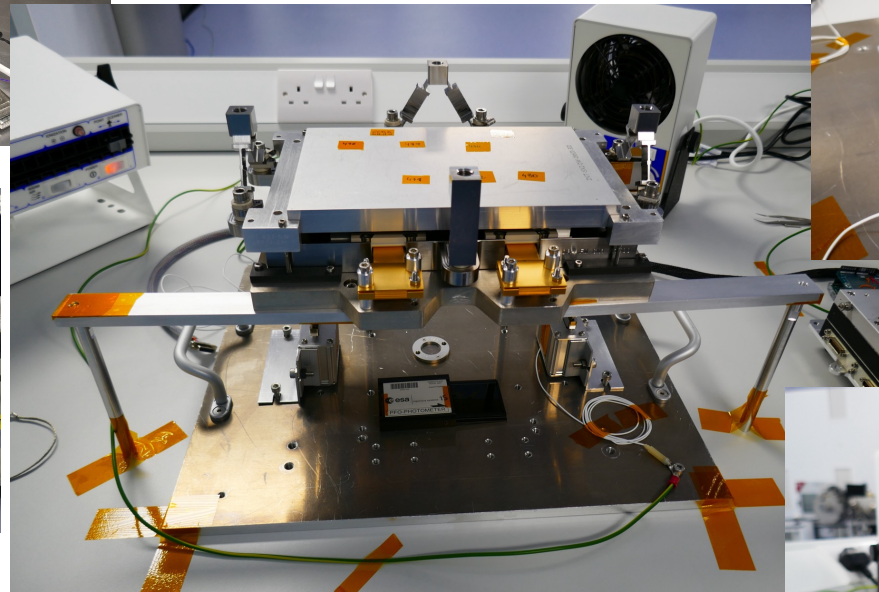
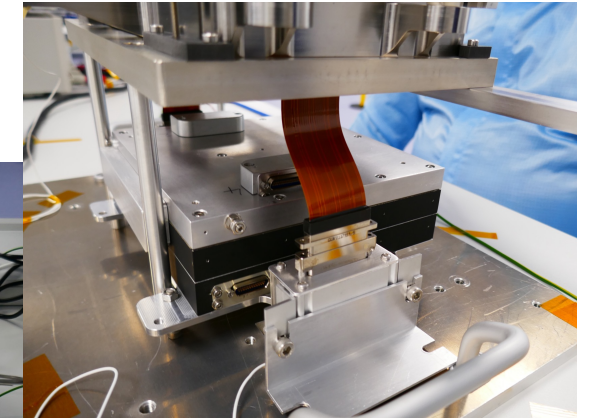
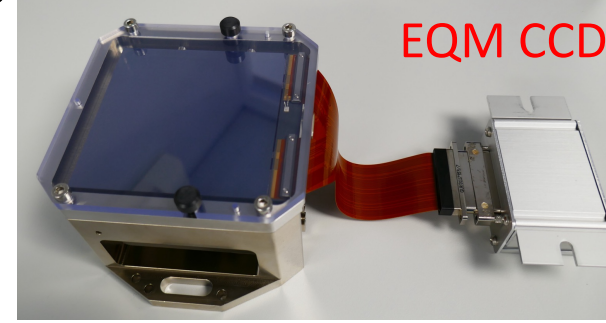
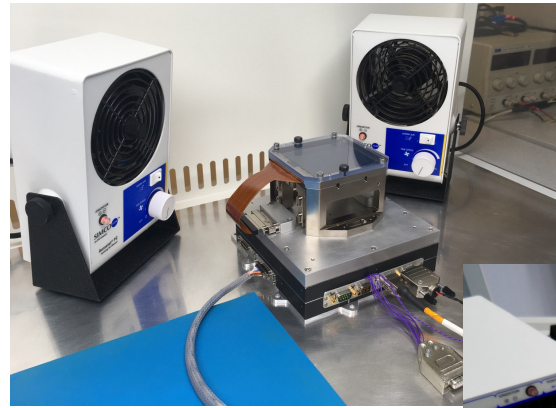


Front-End Electronics (BBv2)

Bipods



SXI Assembly Integration Verification (DPA)



CCD aliveness verification

- Ambient temperature
- Reverse clocked CCDs → noise analysis
- Preparation -> EQM -> Flight Units
- Strict ESD precautions

FM DPA
(Detector Plane
Assembly)



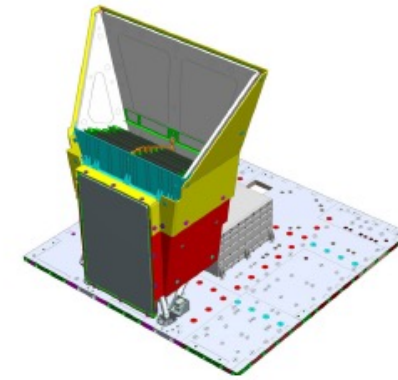


Activity/Milestone	Date
Vibration Test Slot @ RAL	Sept – Oct '23
Thermal Test Slot @ RAL	Nov – Dec '23
EMC Test Slot @ ESTEC	Q1 2024
QAR	Oct through Dec '23
Instrument Delivery to ESA	End Feb '24
PLM and Platform Integration	June – July 2024
Spacecraft AIT	July '24 – March '25
Launch Campaign and Launch	April – May '25
In-orbit Commissioning	Launch + 3 months
Nominal Science	3 years
Extended	2 years
Decommission	

SMILE Instruments

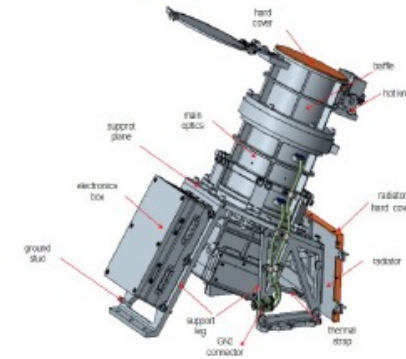
❑ SXI: Soft X-ray imager

- Wide field lobster-eye 0.2-2.5 keV X-ray
- Two large CCD detectors (8.12x8.12 cm).
- 15.5° x 26.5° FOV, Res.: 1-5 min., 0.25-1°, $\Delta E=50$ eV.
- PI: Steve Sembay, Leicester, UK



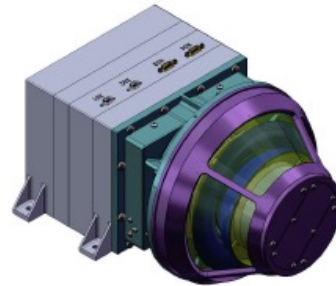
❑ UVI: UV imager

- Four mirror imager, 10° FOV
- Wavelength: 160-180 nm.
- Resolution: 60 s, 150 km at 19 R_E altitude
- CCD detector. Coated mirrors (dayglow rejection)
- PI: ZHANG Xiaoxin, NCSW, CMA, China



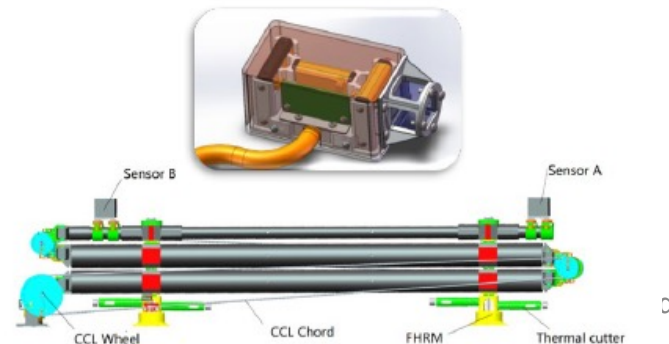
❑ LIA: light ion analyser (2 sensors)

- Top-hat analyser for ions, 3D, density, velocity and temperature
- Energy range: 50 eV - 20 keV
- FOV : 4 π (at up to 0.25s time resolution)
- PI: DAI Lei, NSSC, CAS, China



❑ MAG: Magnetometer

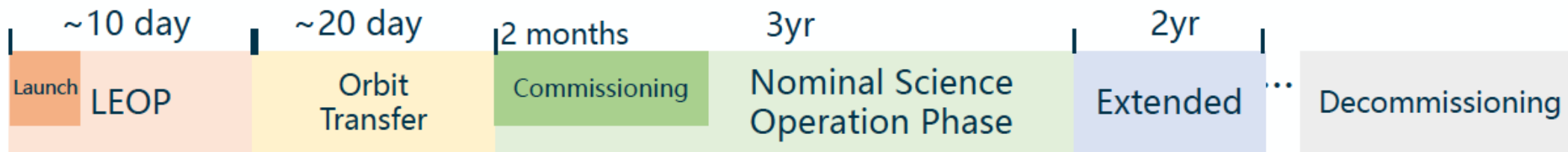
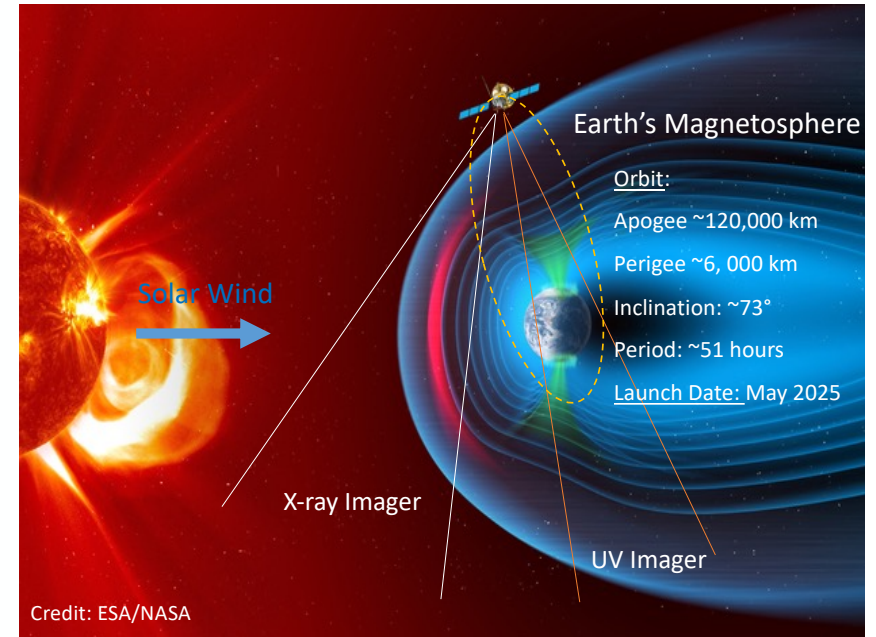
- Fluxgate magnetometer for magnetic field strength and direction up to 40 Hz
- 3 m boom, 2 sensors separated by 0.8 m
- PI: LI Lei, NSSC, CAS, China



Credit: ESA



- 0 Pre-launch Phase (Launch Campaign)
- 1 Launch and Early Operations Phase (LEOP)
- 2 Transfer Phase to HEO
- 3 Commissioning Phase
- 4 Nominal Science Operations Phase
- 5 Extended Science Operations Phase (if any)
- 6 Decommissioning Phase



Credit: ESA



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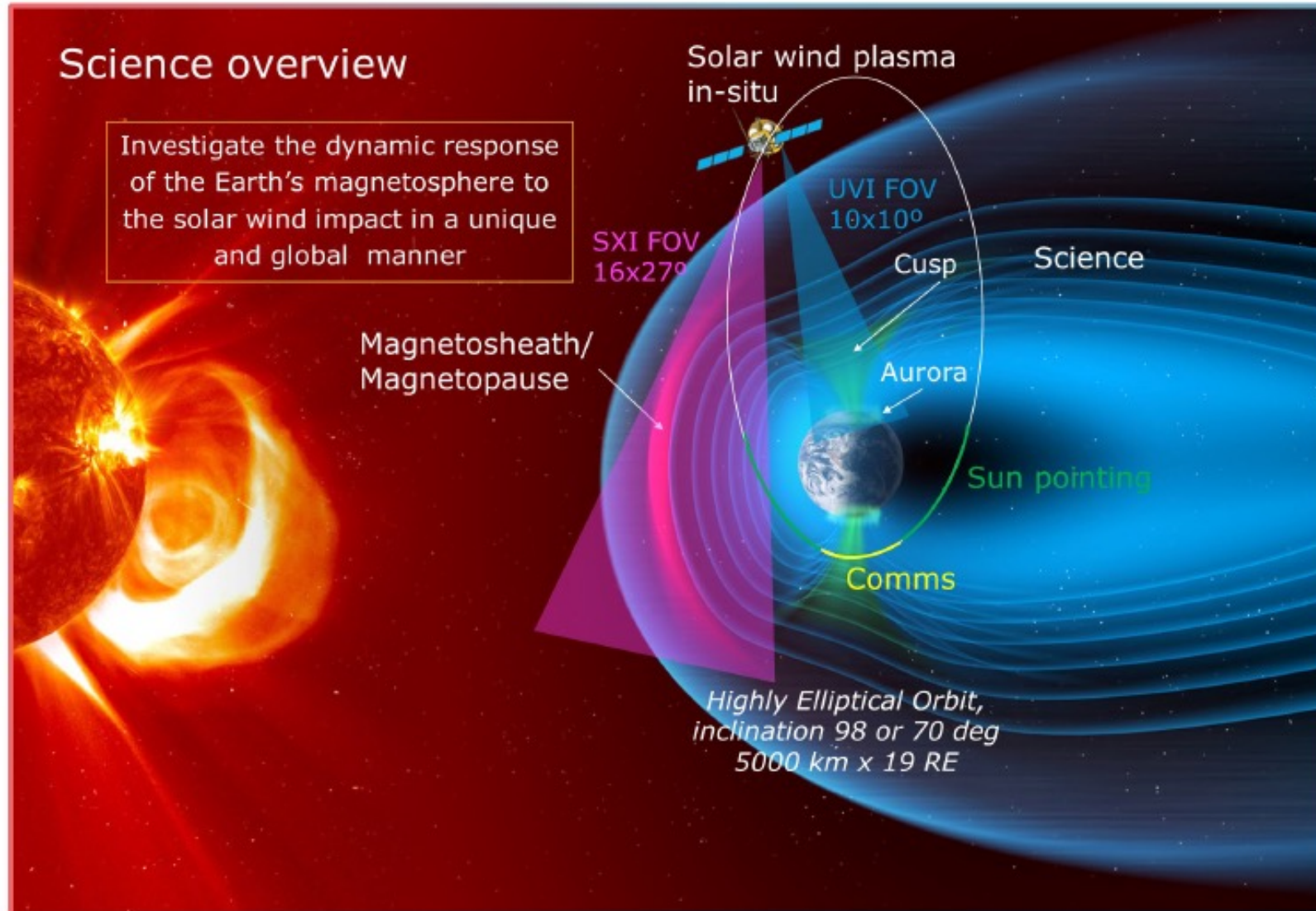
THANK YOU

Solar wind **M**agnetosphere **I**onosphere **L**ink **E**xplorer: **SMILE**

SMILE is a collaborative science mission between the
European Space Agency (ESA) and the Chinese Academy of Sciences (CAS)



EXTRA slides



Credit: ESA

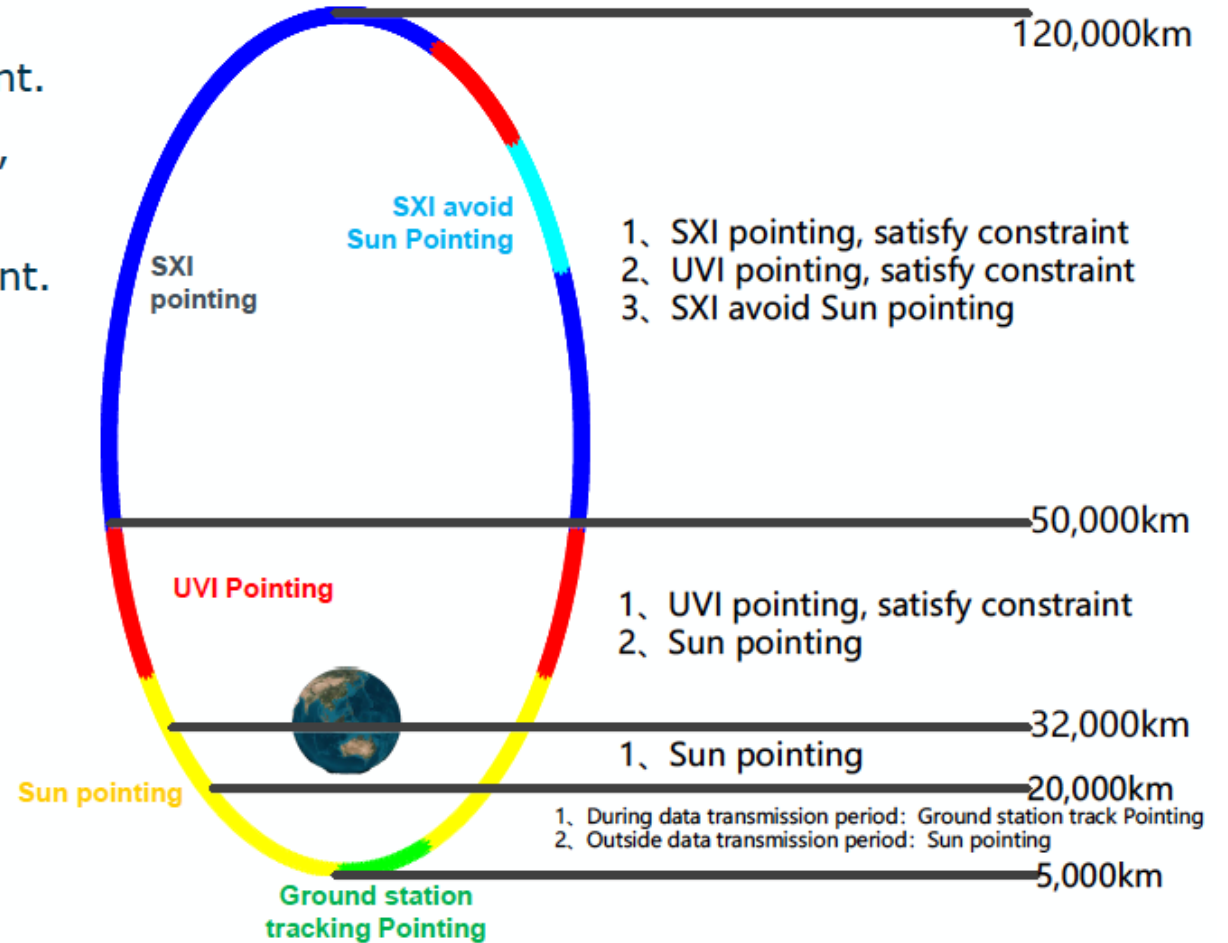
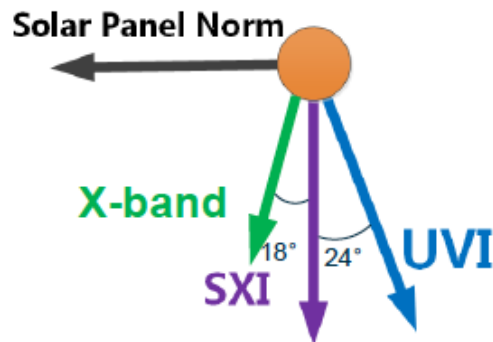


SMILE Complex Pointing Arrangement



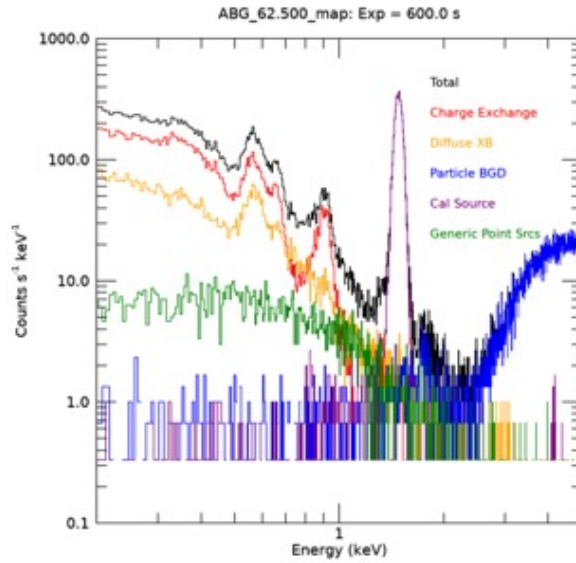
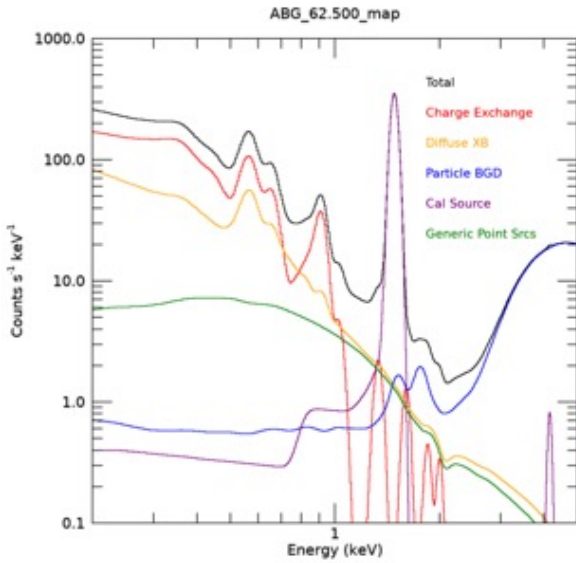
Pointing design

- **SXI pointing:** Driven by SXI pointing requirement.
- **SXI avoid Sun Pointing:** Based on SXI Pointing, S/C rotate to the earth around the body Y axis.
- **UVI pointing:** Driven by UVI pointing requirement.
- **Sun pointing:** Solar array points to Sun.
- **Ground station tracking Pointing:** the X-band antenna points to the relevant Ground Station.



Credit: ESA

SXI Image and spectra simulations



SXI simulation run: Neutral density model: $N_h = 25 \cdot (10 \text{ RE}/r)^3$; $\alpha = 10^{-15}$
 $N_{\text{ex}} = 25.00 \text{ cm}^{-3}$ $V_{\text{ex}} = 399.95 \text{ km s}^{-1}$ Flux: $1.00\text{e}+09 \text{ cm}^{-2} \text{ s}^{-1}$
 $B_x = -0.00 \text{ nT}$ $B_y = -0.00 \text{ nT}$ $B_z = -5.00 \text{ nT}$

Position: 7.68 8.22 13.20 GSE Aim Point: 6.90 0.00 0.00 GSE
 Earth limb angle = 20.39 degrees XPix equivalent size = 0.14 RE

