

THE HIGH ENERGY PARTICLE DETECTOR (HEPD-02) FOR THE SECOND CHINA SEISMO-ELECTROMAGNETIC SATELLITE (CSES-02)

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on behalf of the CSES-Limadou Collaboration

Innovative Particle and Radiation Detectors (IPRD23)

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Siena, Italy



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DI TRENTO

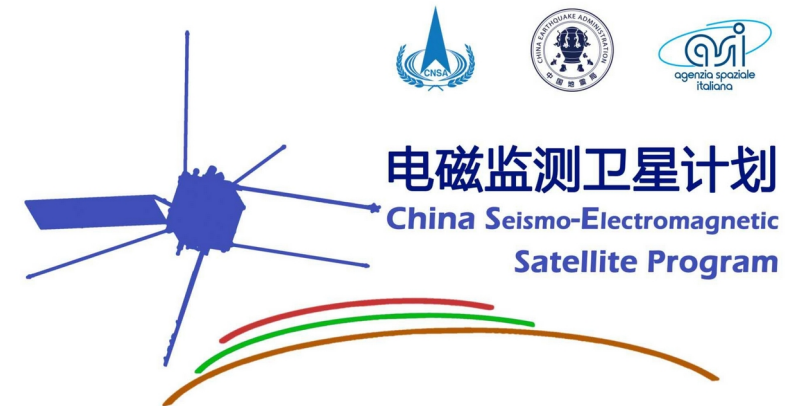


UNIVERSITÀ DEGLI STUDI DI TORINO



CSES MISSIONS – SCIENTIFIC OBJECTIVES

- Monitoring of the **electromagnetic near-Earth space environment**
- Analysis of the **ionospheric and plasmaspheric fluctuations**
- Measurements of iono-magnetospheric perturbations possibly due to **seismo-electromagnetic phenomena**
- Study of **fluxes of high & low energy charged particles** precipitating from the Inner Van Allen radiation belt
- Measurements of **magnetospheric and solar activity**
- Monitoring of the **e.m. anthropic effects** at LEO altitude
- Observations of e.m. transient phenomena caused by **tropospheric activity**



CSES-01 - HIGH ENERGY PARTICLE DETECTOR (HEPD)

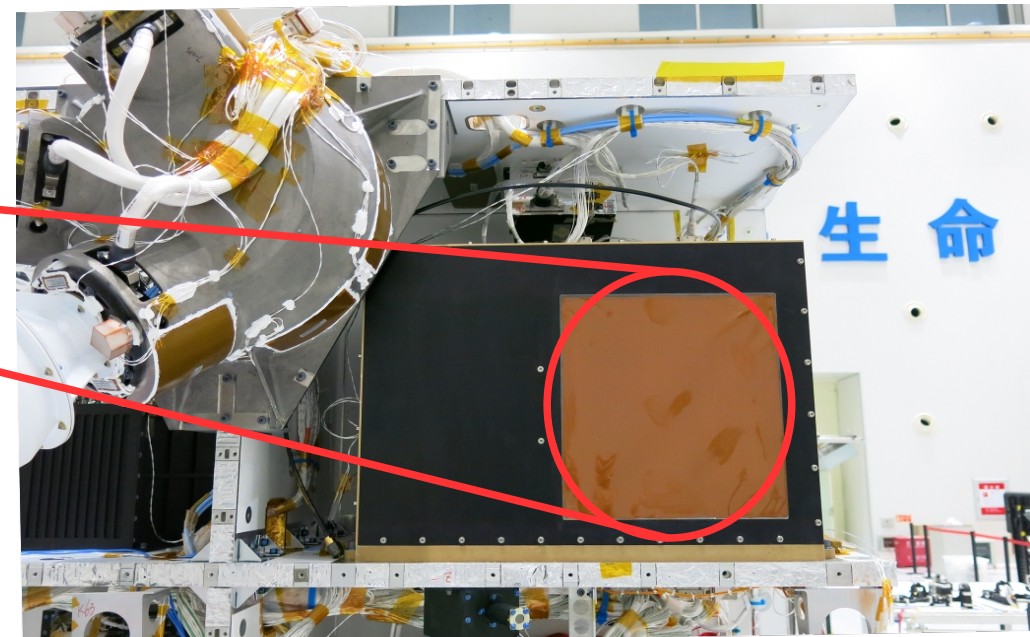
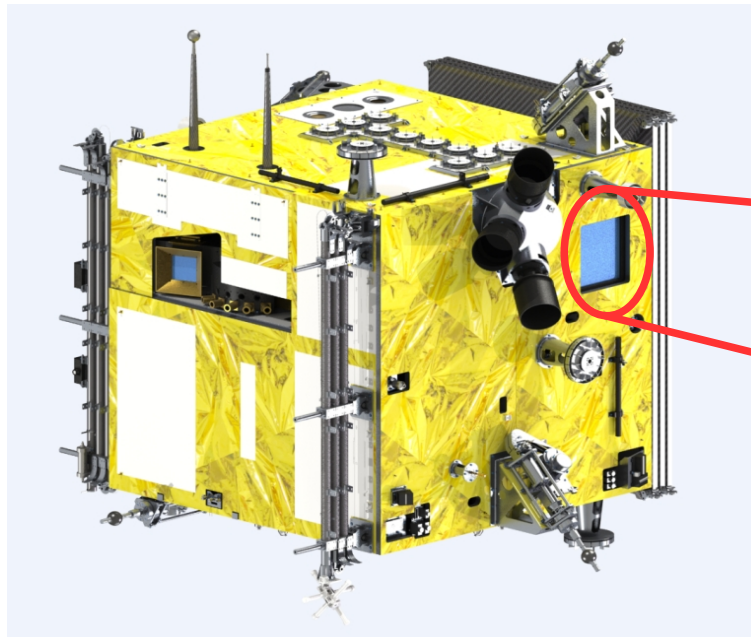
CSES-01: launch February 2nd, 2018 (97.4° sun-synchronous circular orbit, altitude 507 km)

The High-Energy Particle Detector (HEPD) on board CSES-01 can:

- measure the increase of the electron and proton fluxes due to short-time perturbations of the radiation belts caused by solar, terrestrial and anthropic phenomena
- detect different particle populations (solar, trapped, galactic, etc.) according to the satellite position and energy

The energy range explored is 3 - 100 MeV for electrons and 30 - 200 MeV for protons

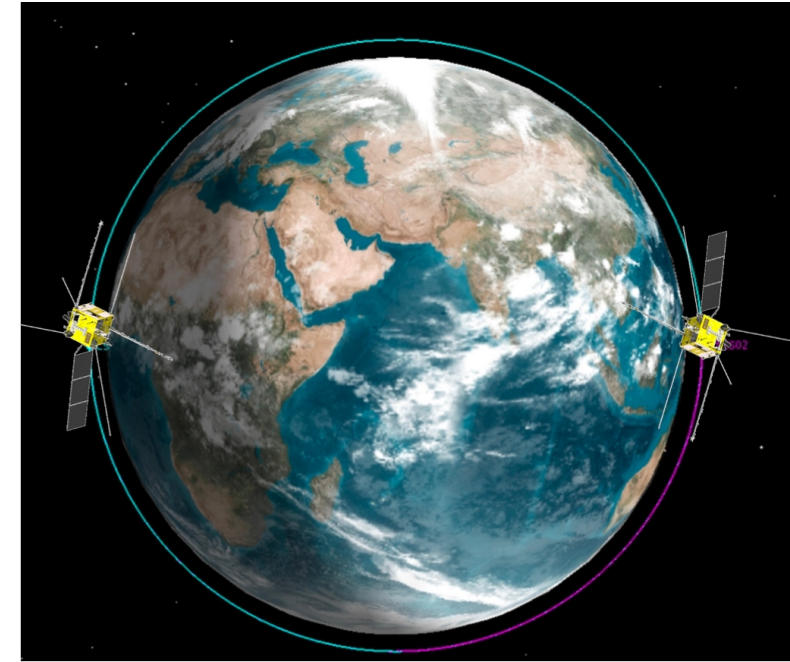
HEPD-01 is installed on board the satellite with its entrance window pointing to the zenith



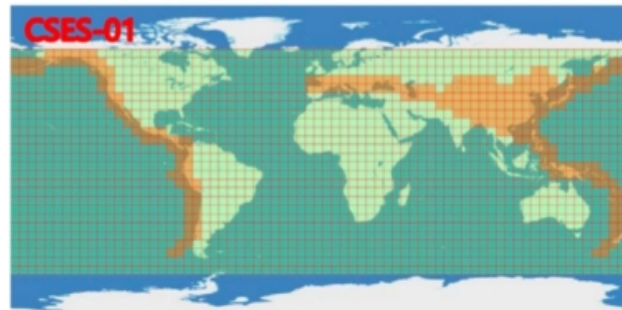
HEPD-01 installed on-board CSES-01 (Credits: DFH)

CSES-02 SATELLITE

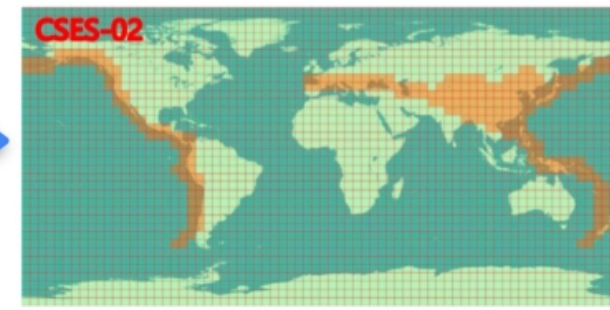
- The second CSES satellite (CSES-02) is expected to be launched in 2024
- CSES will be a sophisticated multi-satellite space observatory (*F. M. Follega The CSES mission: a sophisticated multi-point space observatory, 27/09 11:50*)
- Same DFH CAST-2000 platform of CSES-01 with some upgrades
 - Earth oriented 3-axis stabilization system with orbit maneuver capability
 - X-Band Data Transmission 120Mbps → **150Mbps**
 - Storage 160Gb → **512Gb**
 - Total Mass: 730kg → **900kg**
 - Peak Power Consumption: ~900W
 - Design Life-span: 5 years → **6 years**
- Complementary Ground Track wrt CSES-01
 - Identical Orbit Plane **with 180° Phase Difference**
 - Track interval: 5° → **2.5°**
 - Return cycle: 5 days → **2.5 days**
- Operation mode: Full time operational



Operation area between lat [-65,65]



Full coverage at extreme latitudes

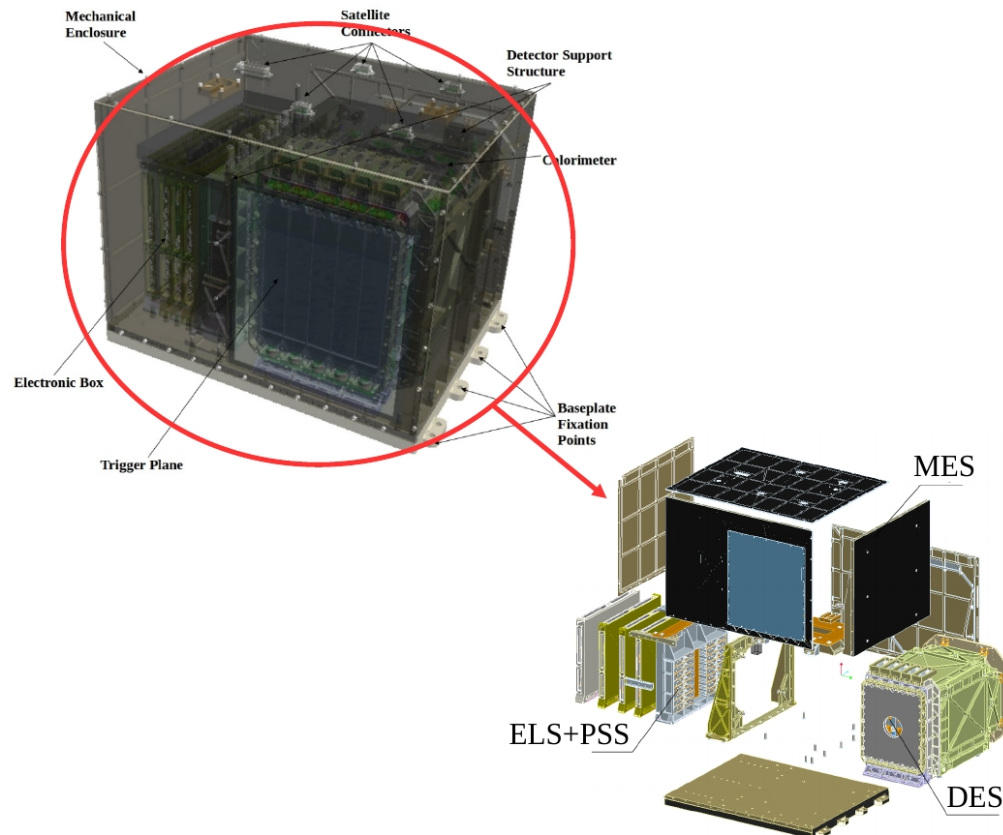


CSES-02 – PAYLOAD CONFIGURATION

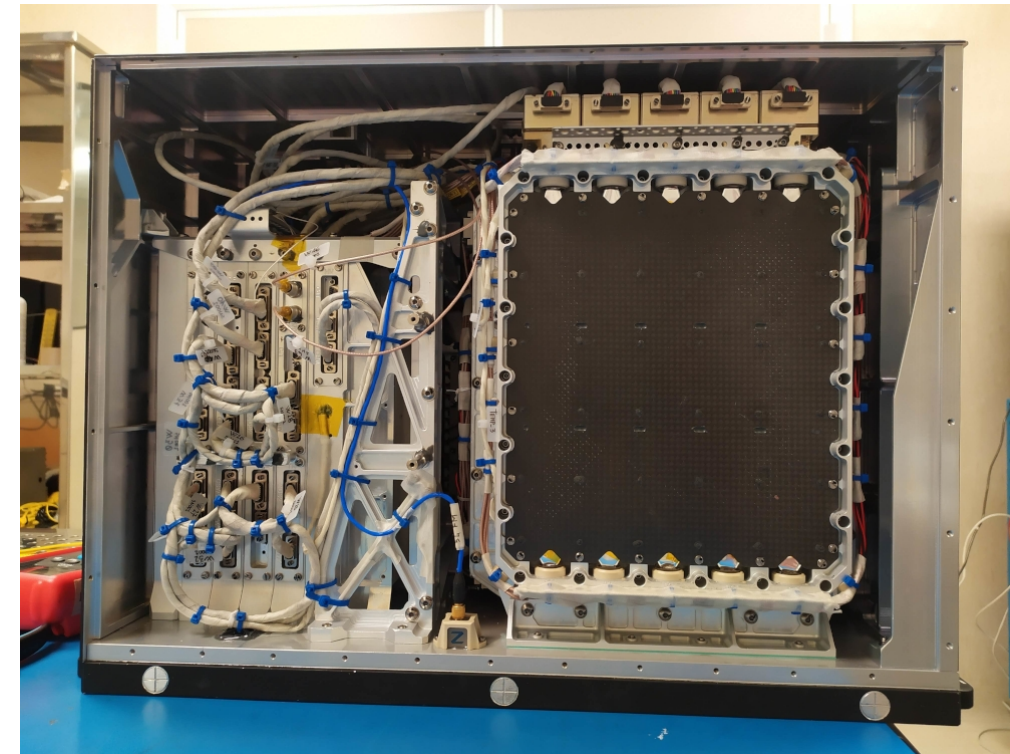
<i>Category</i>	<i>Payload Name</i>	<i>Observation Targets</i>
<i>Energetic Particle</i>	High Energy Particle Detector (HEPD) Italy Medium Energetic Electron Detector (MEED)	Proton : 2MeV~200MeV Electron : 30keV~50MeV
<i>Electro-Magnetic Field</i>	Electric Field Detector (EFD) Italy	Electric Field: DC ~ 3.5MHz
	High Precision Magnetometer (HPM): FGM1, FGM2, CDSM Austria	Magnetic Field: DC ~ 15Hz
	Coherent Population Trap (CPT)	Magnetic Field: DC ~ 15Hz
	Search Coil Magnetometer (SCM)	Magnetic Field: 10Hz ~ 20kHz
<i>In Situ Plasma</i>	Plasma Analyzer Package (PAP)	Composition : H^+, He^+, O^+ $N_i : 5 \times 10^2 \sim 1 \times 10^7 cm^{-3}$ $T_i : 500K \sim 10000K$
	Langmuir Probe (LP)	$N_e : 5 \times 10^2 \sim 1 \times 10^7 cm^{-3}$ $T_e : 500K \sim 10000K$
<i>Plasma Profile</i>	GNSS Occultation Receiver	TEC by GNSS Occultation Signal
	Tri-Band Beacon (TBB)	TEC by transmit VH/U/L Signal
	Ionospheric Photometer (IP)	135.6nm and N_2LBH airglow

HEPD-02

- Compact and lightweight payload (40.36x53x38.15 cm³, 47.2 kg)
- Low power consumption (~43 W)
- Acceptance and calibration campaign completed
- To be delivered to China October/November 2023



Operating temperature	-10 °C ÷ +35 °C
Operating pressure	$\leq 6.65 \cdot 10^{-3}$ Pa
Data budget	≤ 100 Gb/day
Mass budget	≤ 50 kg
Power budget	≤ 45 W
Electron kinetic energy range	3 MeV ÷ 100 MeV
Proton kinetic energy range	30 MeV ÷ 200 MeV
Angular resolution	$\leq 10^\circ$ for e^- with $E > 3$ MeV
Energy resolution	$\leq 10\%$ for e^- with $E > 5$ MeV
Pointing	Zenith
Scientific data bus	RS-422
Data handling bus	CAN 2.0
Life cycle	> 6 years



HEPD-02 DETECTOR LAYOUT AND UPGRADES WRT HEPD-01

TRigger plane TR1 ($200 \times 180 \text{ mm}^2$) segmented in 5 plastic scintillator bars (2 mm thick) → *upgrade wrt HEPD-01: additional trigger plane, decrease energy threshold and increase redundancy*

Direction Detector DD ("tracker") made of five standalone tracking modules ("turrets"), each composed of three sensitive planes ("staves")

TRigger plane TR2 ($150 \times 150 \text{ mm}^2$)

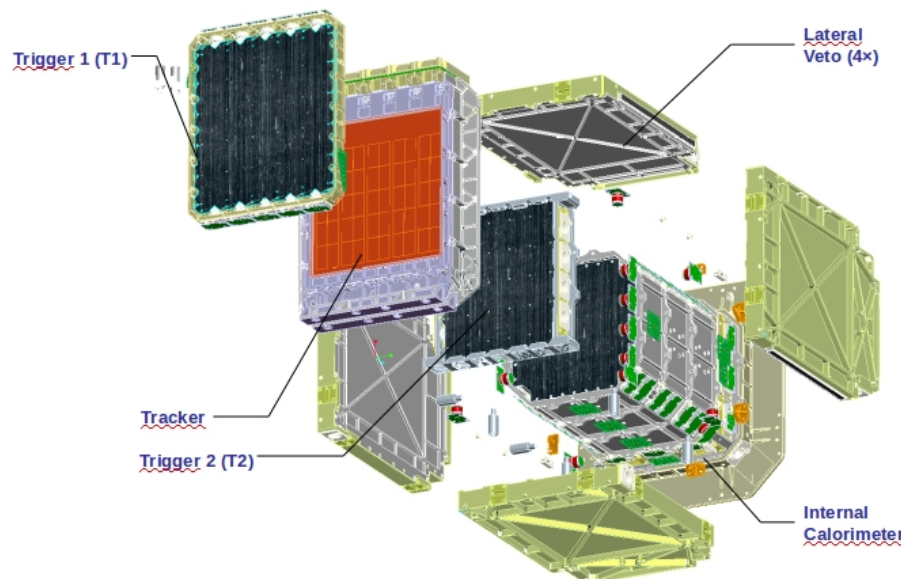
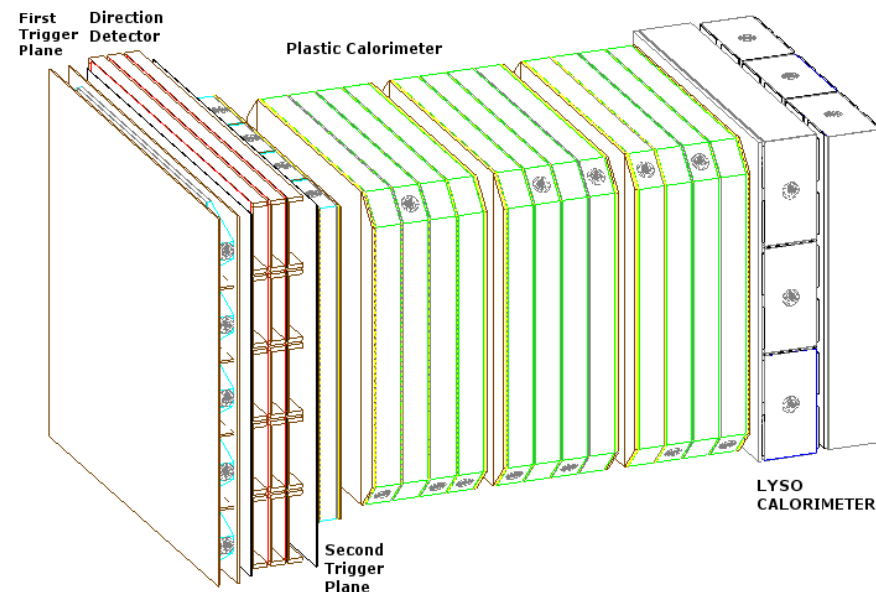
Energy Detector ED ("calorimeter") composed of:

12 plastic scintillator planes ($150 \times 150 \times 10 \text{ mm}^3$) → *16 planes in HEPD-01*

2 crystal (LYSO) scintillator planes ($150 \times 150 \text{ mm}^2$ segmented in 3 bars (50 mm thick) → *upgrade wrt HEPD-01: 6 bars instead of 9 cubes, increase energy range, position sensitivity and redundancy.*

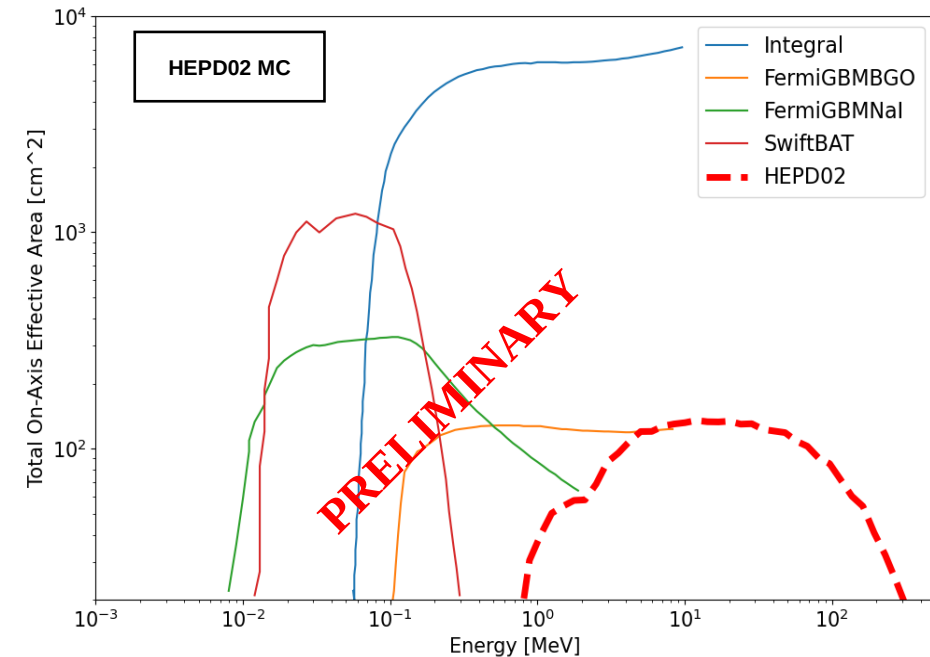
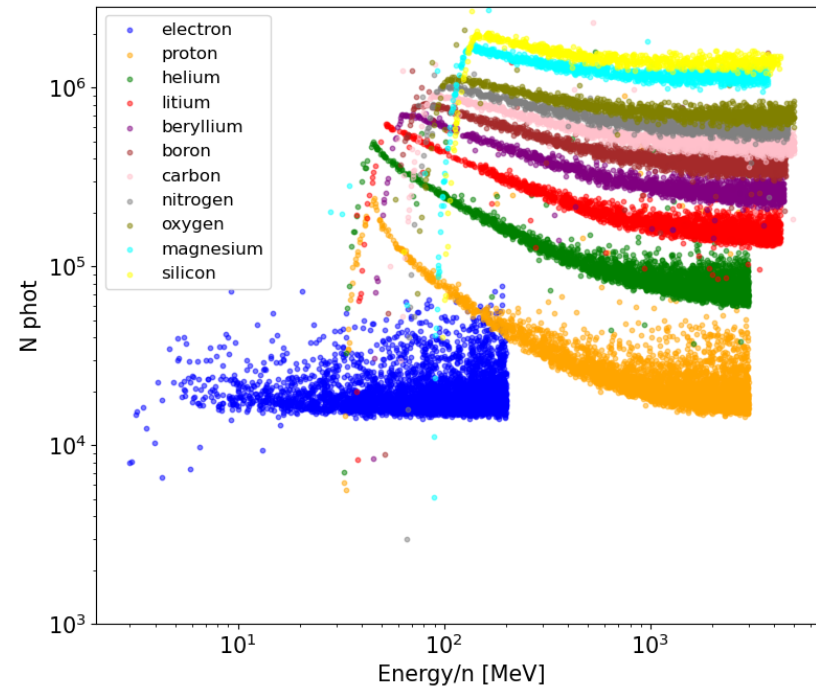
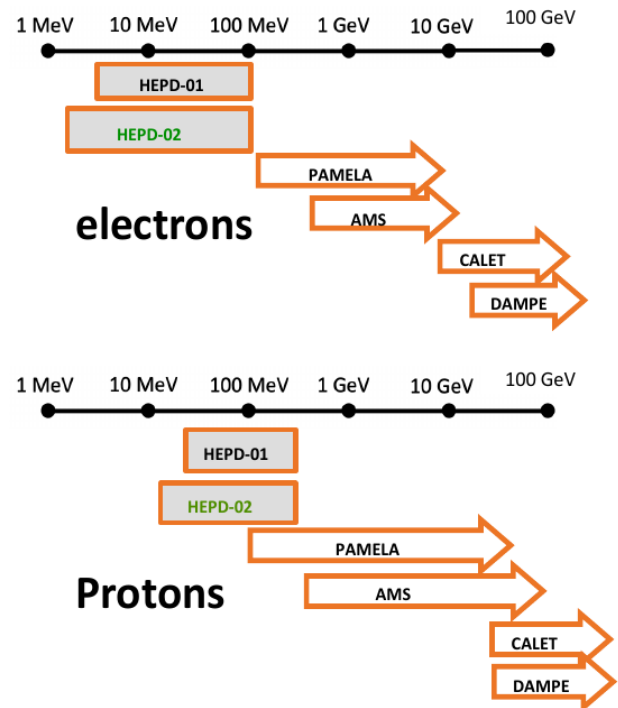
Containment Detector CD surrounding the calorimeter on 5 sides, made of plastic scintillator planes (4 lateral and 1 bottom plane), 8 mm thick.

Plastic scintillators: Eljen EJ-200; PMTs: Hamamatsu R9880-210



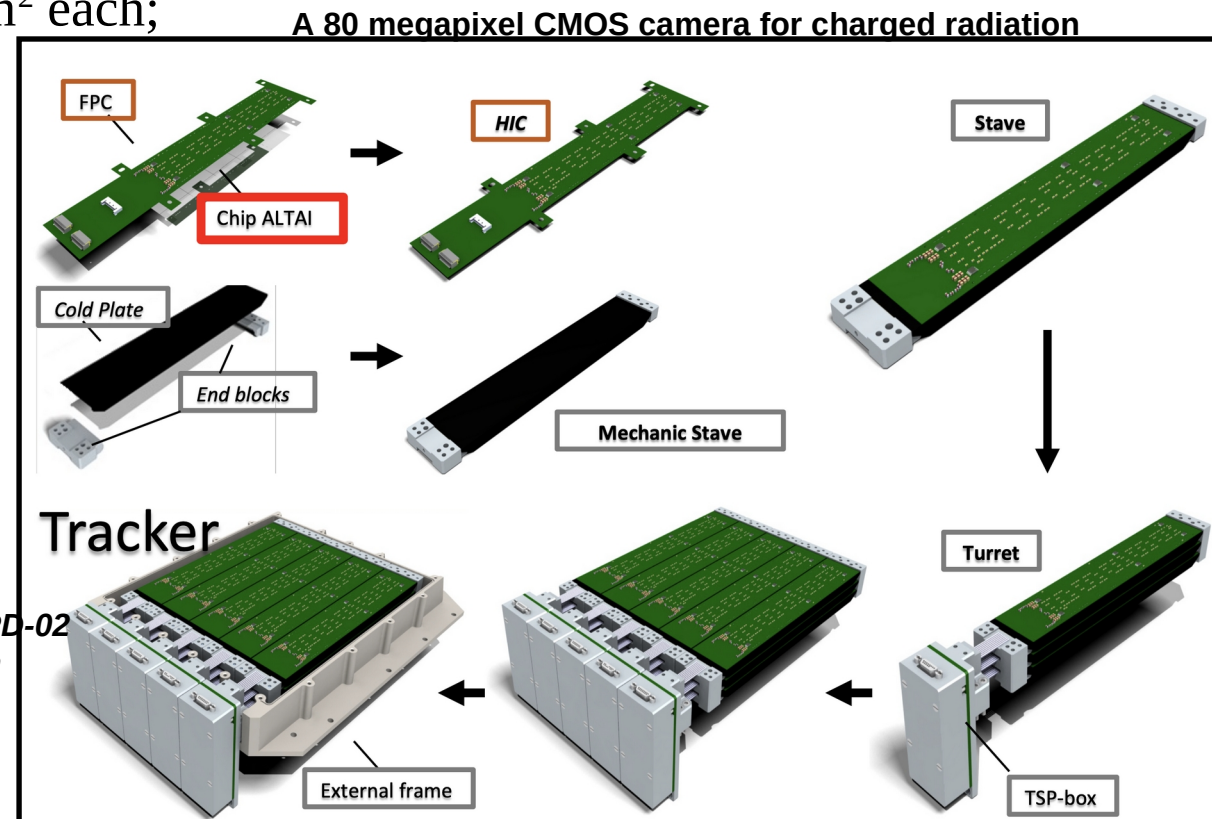
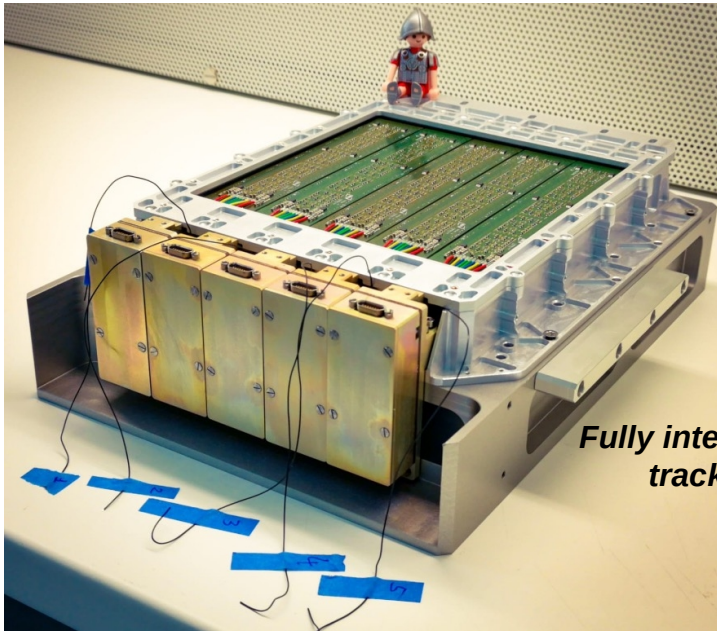
HEPD-02 DETECTOR CAPABILITIES

- Designed to measure fluxes of electrons, protons and nuclei in a wide energy range
- Implements a logic to trigger on sudden increases of GRB masks counts, integrated @ 200 Hz
(*V. Scotti, The DAQ and trigger of the High Energy Particle Detector (HEPD-02) for the CSES-02 space satellite, 26/09 18:00*)



HEPD-02 DIRECTION DETECTOR

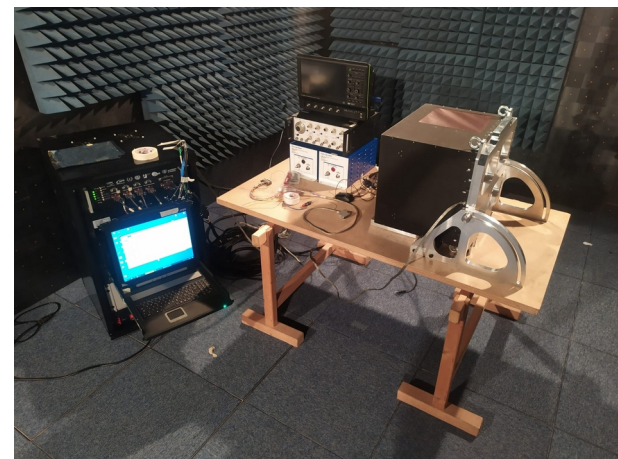
- Based on the MAPS developed for the ALICE experiment at CERN (*E. Ricci, Design and characterisation of the HEPD-02 MAPS-based tracker for operations in space, 25/09 09:50*)
- Pixel size $29.24 \mu\text{m} \times 26.88 \mu\text{m}$ ($\sim 4 \mu\text{m}$ single-hit resolution)
- ALTAI: 512×1024 pixels \rightarrow 10 chips per stave;
- 5 turrets, each made of 3 staves with active area $15 \times 3 \text{ cm}^2$ each;



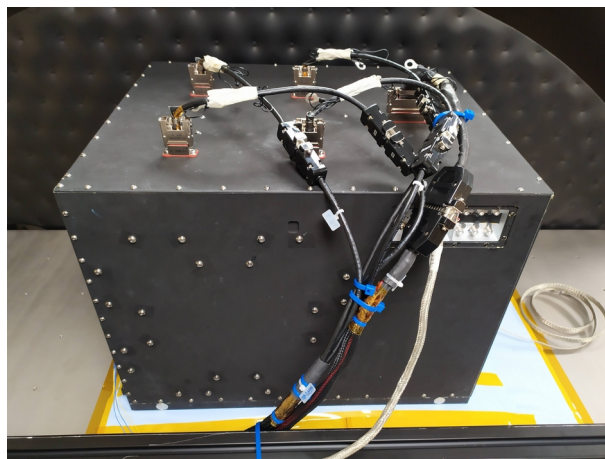
HEPD-02 ACCEPTANCE TEST CAMPAIGN

Environmental test campaign February-July 2023

- Vibration Test (sine 8 g & random 7.55 GRMS)
- Thermal Cycling Test (14.5 cycles, -20 °C ~ + 45 °C)
- EMC Test (CE102, CS101, CS114, RE102, RS103)
- Thermal Vacuum Test (3.5 cycles, -20 °C ~ + 45 °C)



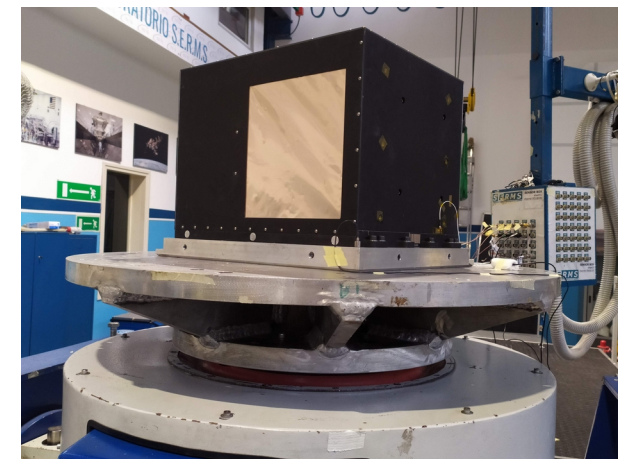
HEPD-02 FM EMC test @IFAC-CNR



HEPD-02 FM in thermal vacuum chamber @SERMS



HEPD-02 FM in thermal chamber @SERMS

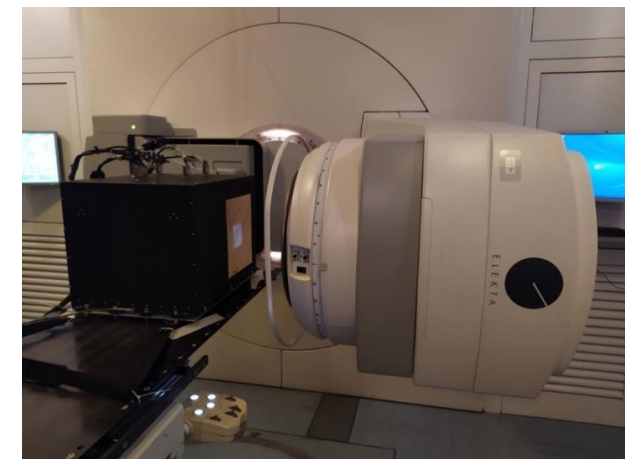


HEPD-02 FM vibration test @SERMS

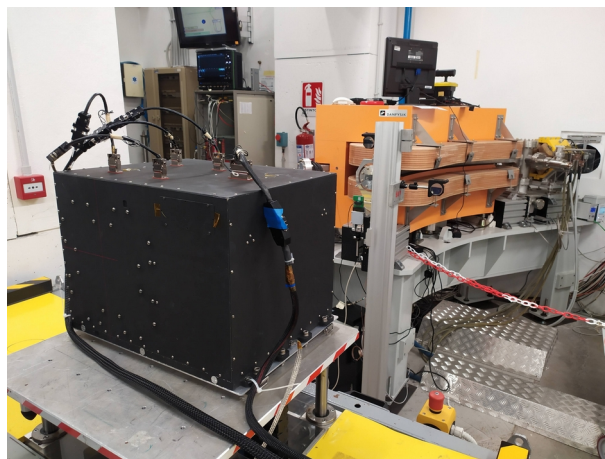
HEPD-02 BEAM TEST CAMPAIGN

Beam test campaign June-July 2023

- e^- (6-12 MeV)/gamma @ LINAC S. Chiara (Trento, Italy)
- e^- (>30 MeV) @ BTF (Frascati, Italy)
- Proton (70-230 MeV) @ APSS (Trento, Italy)
- Carbon/proton (115-398 MeV/amu) @ CNAO (Pavia, Italy)



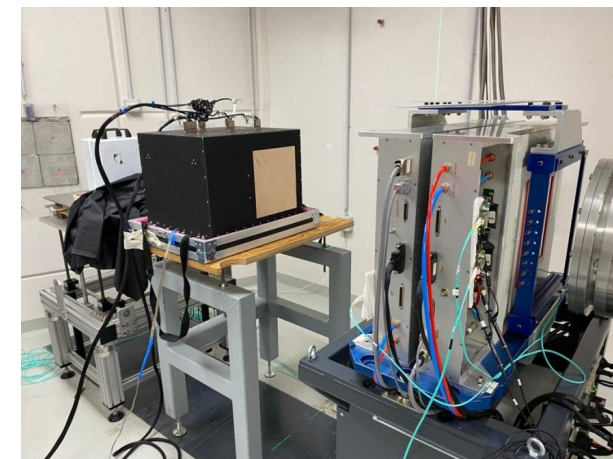
HEPD-02 FM beam test w/ electrons and gammas
@LINAC S. Chiara



HEPD-02 FM beam test w/ electrons @BTF



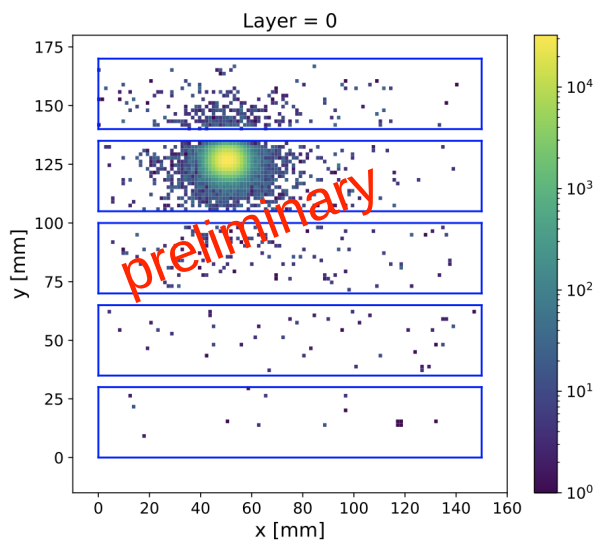
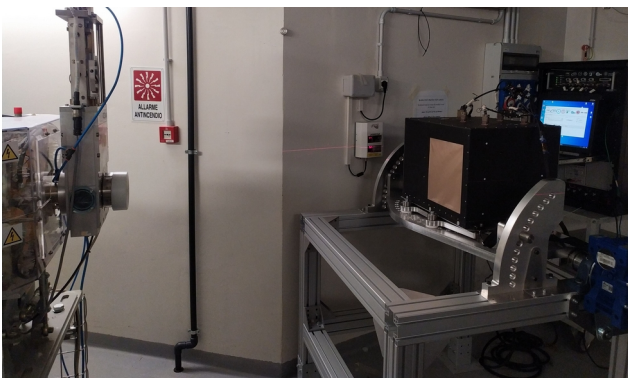
HEPD-02 FM beam test w/ protons @Proton Therapy
Trento



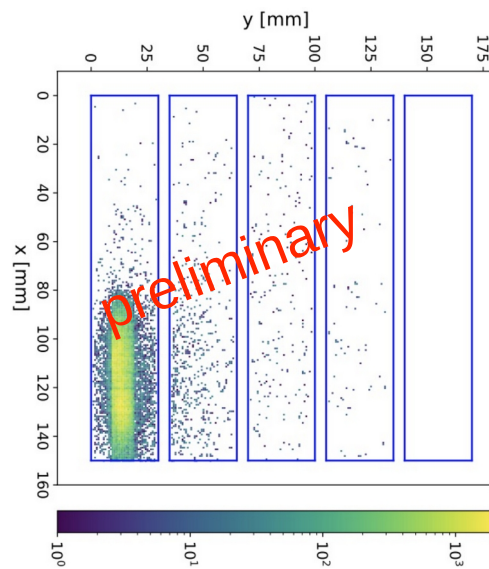
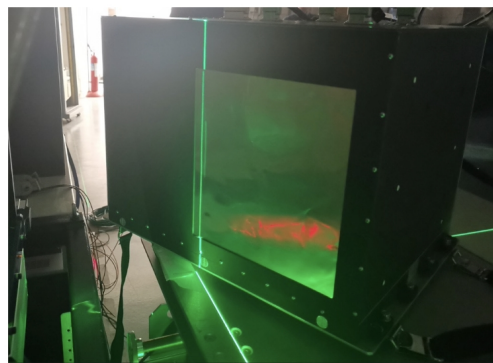
HEPD-02 FM beam test w/ protons+carbon nuclei
@CNAO

HEPD-02 TEST BEAM RESULTS - POSITION AND TRACKING

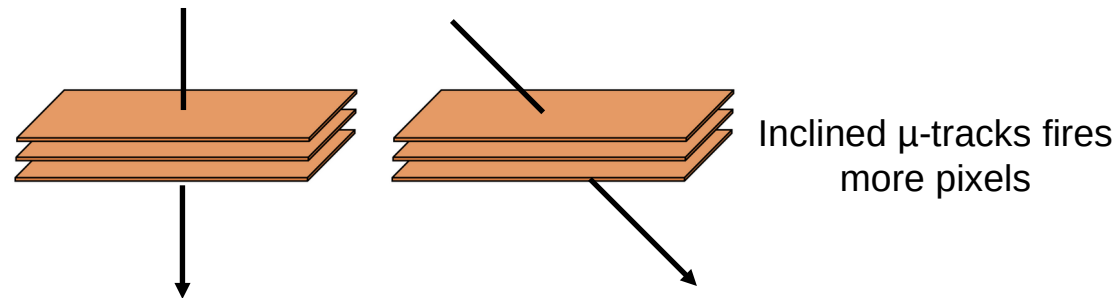
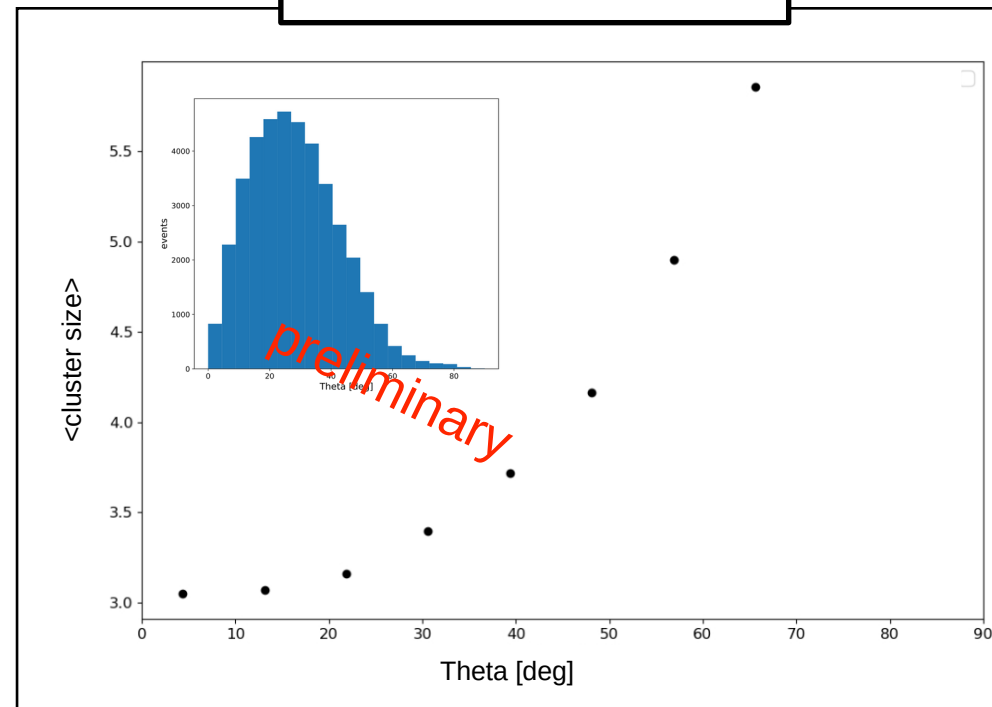
Proton beam



Carbon beam

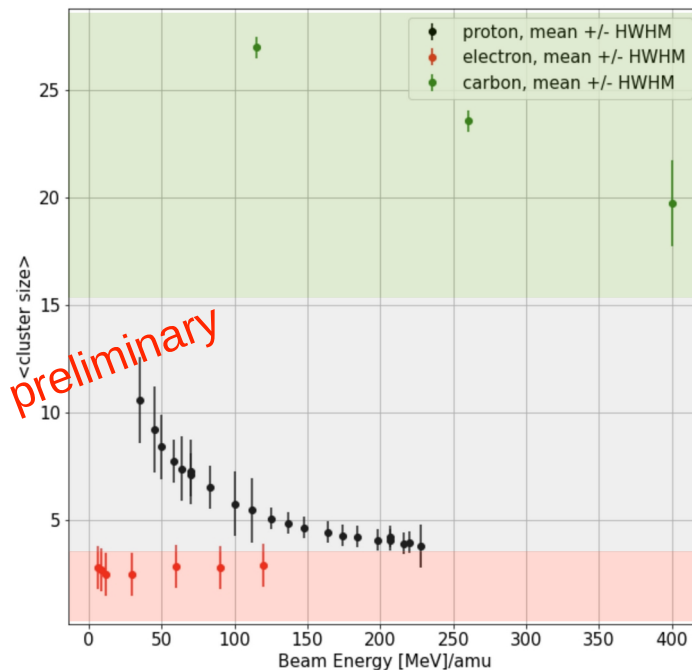
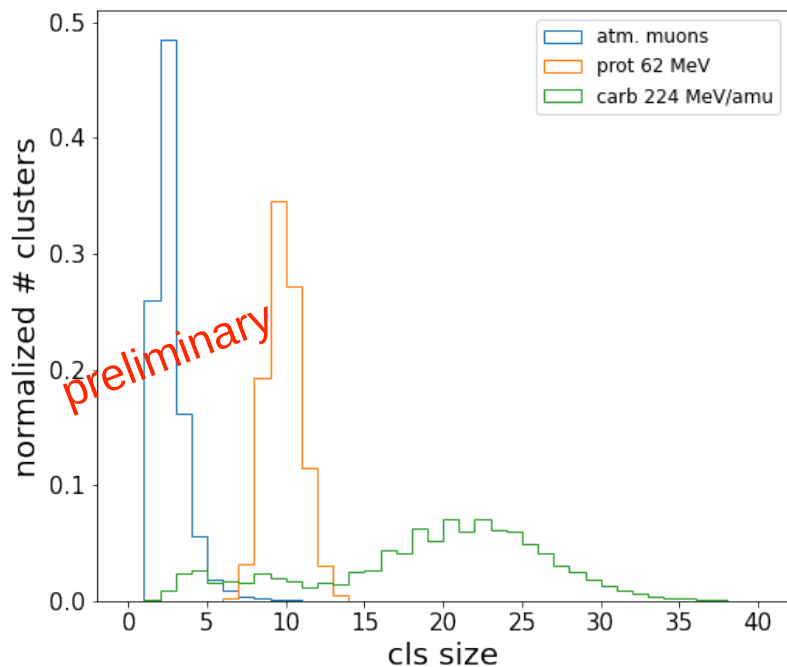
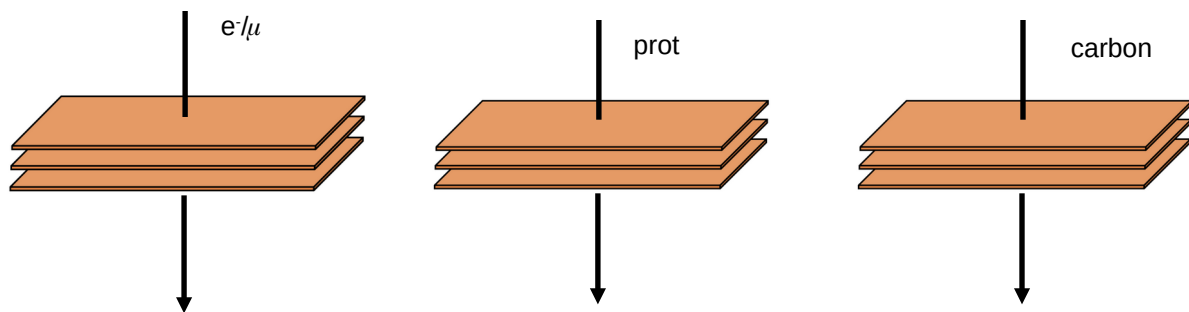


Atmospheric muon data

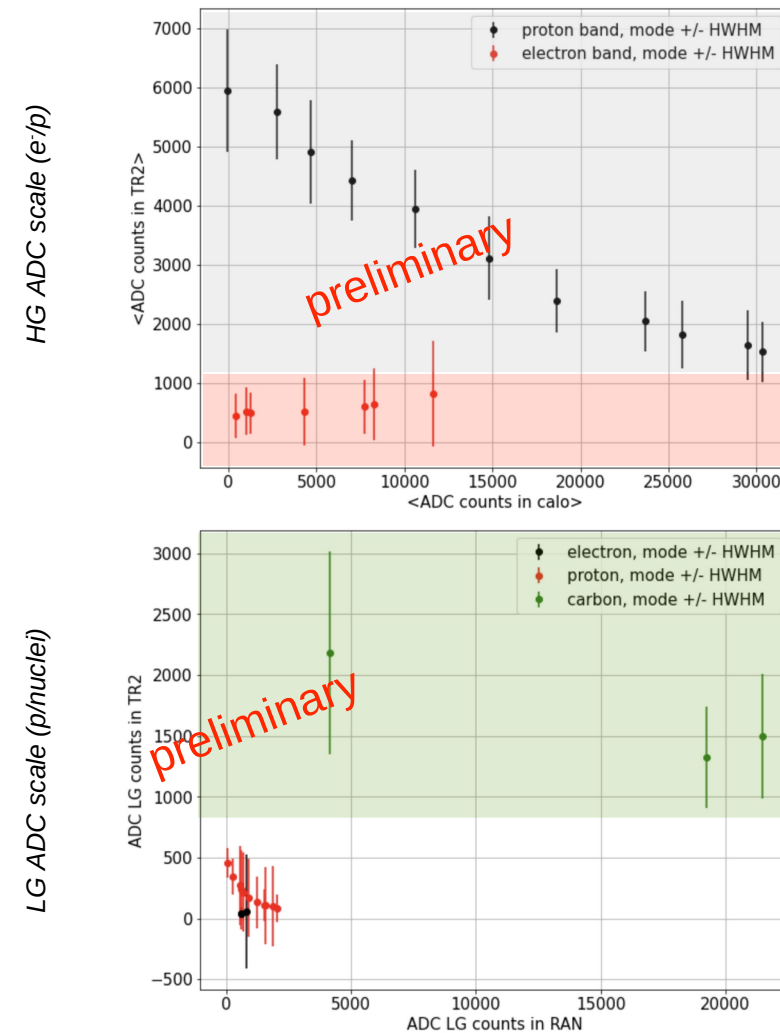


HEPD-02 TEST BEAM RESULTS – PARTICLE IDENTIFICATION

Particle identification using cluster size (Tracker information)

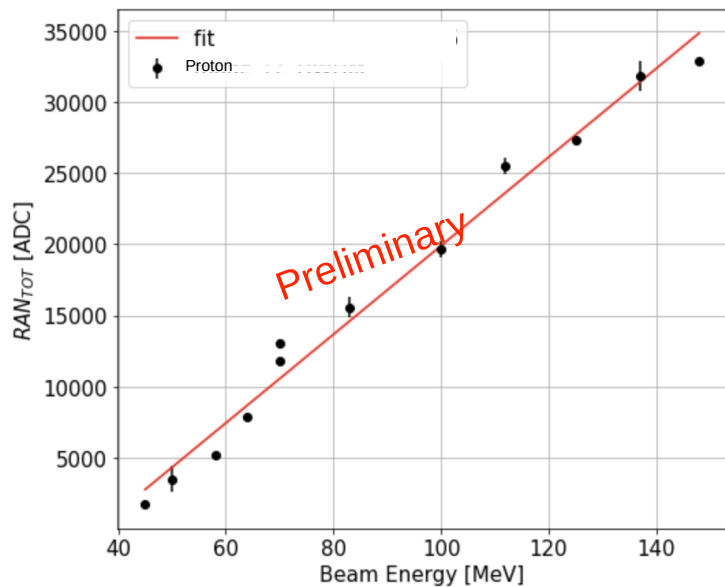


Particle identification calo signal

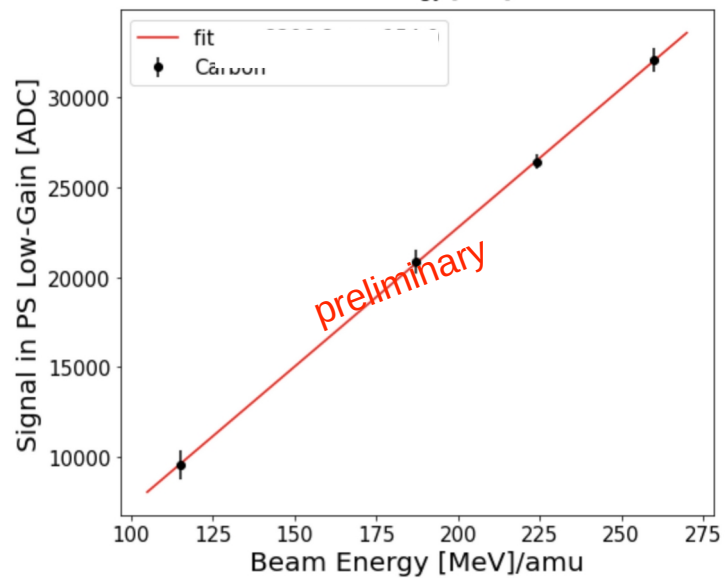


HEPD-02 TEST BEAM RESULTS – ENERGY RECONSTRUCTION

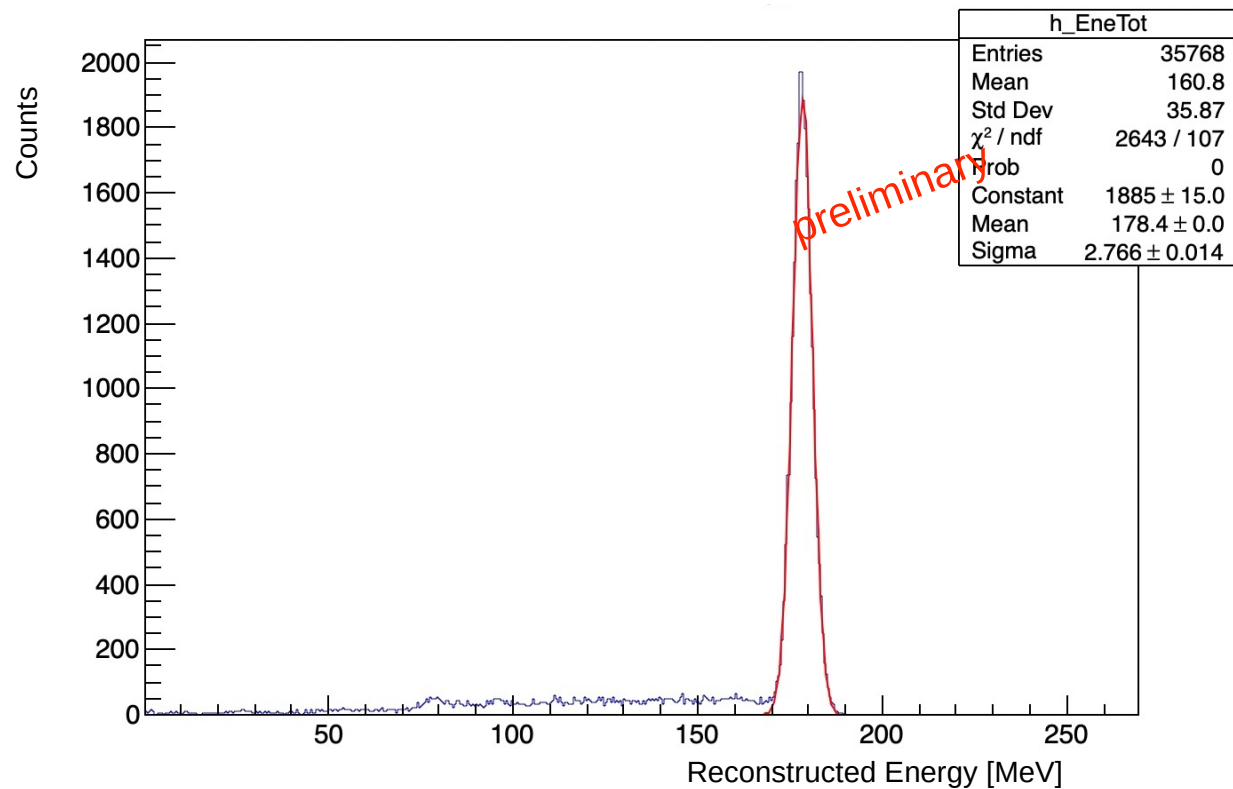
Proton beam



Carbon beam



Proton beam @ Proton Therapy Center (Trento)



< 5% resolution on high energy protons

CONCLUSIONS

- The High Energy Particle Detector (HEPD-02) has been developed to be launched on board of the second China Seismo-Electromagnetic Satellite (CSES-02) in 2024
 - Detector design and capabilities have been improved wrt HEPD-01
 - Designed to measure fluxes electrons, protons and nuclei in a wide energy range
 - A dedicated GRB trigger logic has been implemented and tested
- Acceptance and beam test campaign completed → ready for delivery to China
- Preliminary beam test analysis demonstrates that the performance is compliant with simulations and main mission requirements