

# Development of cryogenic cooling system for low emission future aircrafts

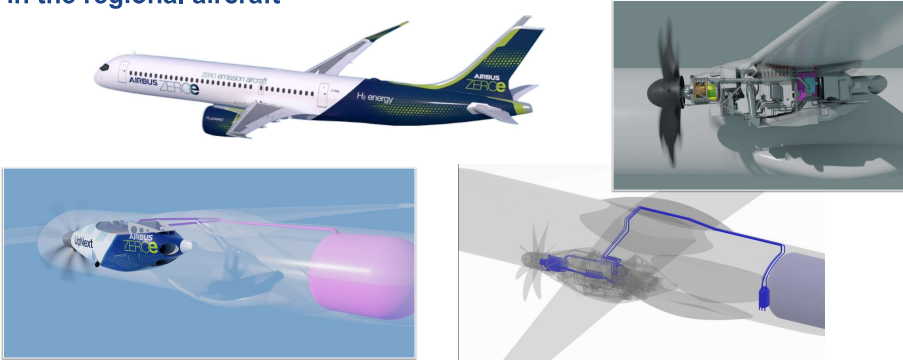
Matteo Tassisto<sup>1</sup>, Timin Jacob<sup>2</sup>, Swapnil Kharche<sup>1</sup>, Souhaib Boukayoua<sup>1</sup>, Ludovic Ybanez<sup>1</sup>

<sup>1</sup>Airbus UpNext SAS, <sup>2</sup>Airbus UpNext GmbH



## CRYOPROP (2024/2026)

- From building a **MW-class cryogenic powertrain demonstrator** to its integration in the regional aircraft



- Develop a **Cryogenics & Superconducting supply chain** for the key components in the powertrain

- Superconductive motor + cryogenic Motor Control Unit
- Hi Power Superconductive DC distribution
- Propulsion Control System
- Cryocooling system based on LH2 as primary cold source**

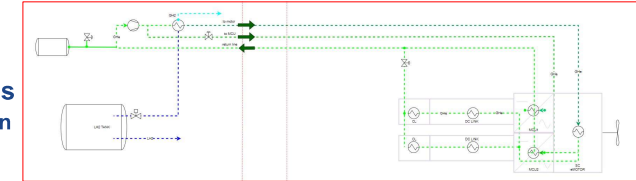
## Cryocooling System Main objectives

- Operability of the electrical components of a 2 MW powertrain at the target working temperatures
  - Up to 4 kW at 40 K for superconductive motor
  - Up to 1 kW at 60 K for DC distribution lines
  - Up to 20 kW at 120 K for Motor Control Unit
- Optimization of the LH2 consumption for flight phase
  - steady state
  - Transients
  - Failure scenarios
- Optimization of weight, dimensions, space allocation at system and component level
- Increase maturity (TRL) at system and component level
- Assess operability and maintainability at aircraft level

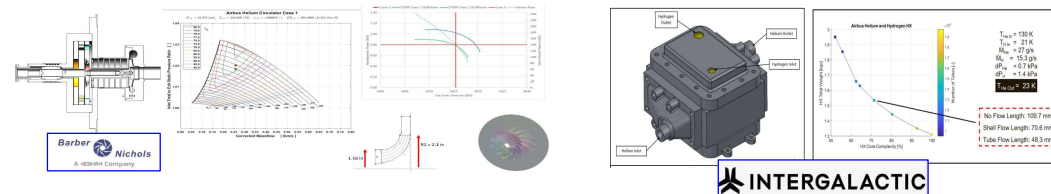
## Cryogenic cooling system First results

### At system level

- Closed recirculation loop of GHe as secondary coolant @ 15 to 30 bara
- Architecture and system trades
  - Cooling system mass minimization 220kg to 290 kg considering 14 m cryolines
  - Component operating temperature vs H2 consumption < 32 g/s at nominal power for motor temperature < 50K, DC cables < 60K, MCU < 100K



### At component level



INTERGALACTIC

## Main learnings so far ...

- Potential of an embeddable cooling system based on LH2 is confirmed
- Figures of Merit in terms of mass, efficiency and LH2 are compatible with AC requirements
- Technology to be matured, no showstoppers identified at system and component level

## Future Challenges and next steps...

- Development of mature components with optimized figures of merit (valves, heat exchangers, cryofans, cryolines) taking into account aircraft constraints and qualification requirements.
- Manufacture and test a full representative cryogenic cooling loop Demonstrator in 2026

## References

- [1] L. Ybanez et al., "ASCEND: The first step towards cryogenic electric propulsion", Proc. IOP Conf. Ser.: Materials Sci. Eng., vol. 1241, 2022.  
[2] L. Ybanez et al., "Cryogenic electric propulsion system: ASCEND main results and perspectives", Conference: MEA2024

UpNext

©Airbus, 2024 - FR\_EC\_NotListed

CRYOPROP