Probing Fifth Force Interactions by Planetary Science Data

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Beyond the Standard Model and Fifth Force

• In BSM physics, new light, weakly-coupled degrees of freedom are introduced in the form of hypothetical particles and interactions.

 $\bullet~$ New fundamental interactions introduced by these particles $\rightarrow~$ "Fifth Force"

• These particles can arise from string theory.

• Can also be dark matter or dark energy candidates.

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The Fifth Force and its Modeling

• Phenomenological model for fifth force is given by:

$$V(r) = -\frac{GMm_*}{(1 + \alpha_1) r} \left(1 + \alpha_2 \exp\left(-\frac{r}{\lambda}\right)\right)$$

$$\lambda \rightarrow \text{fifth force range}\left(\propto \frac{1}{\text{mediator mass}}\right)$$
(1)

$$m_* \rightarrow \text{mass of celestial body orbiting object of mass } M$$

$$\alpha_1, \alpha_2 \rightarrow \text{Yukawa parameters (model dependent)}$$

• First term: deviation from Newtonian gravity term

$$G \to \widetilde{G} \equiv \frac{G}{1 + \alpha_1}$$
 (2)

Second term: Yukawa term due to the fifth force

$$V(r) \propto -\alpha_2 \frac{exp(-r/\lambda)}{r} \tag{3}$$

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Probing the Fifth Force

- Can be probed by tracking the deviations of Keplerian orbits of celestial objects.
- Many such probes are similar to the tests for General Relativity (GR).
- Deviations can be quantified in terms of orbital parameters:



 $(a, e, i, \Omega, \omega, \nu)$

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Past Literature

In the literature, constraints on the fifth force due to deviations from Newtonian trajectory are given by:

- Lunar Laser Ranging (Earth \leftrightarrow Moon) (Bergé et al., 2018)
- Timing of pulsar around Sagittarius A^* (Dong et al., 2022)
- Orbital precession of:
 - Near-Earth Object (NEO) asteroids (Tsai et al., 2023)
 - Planets in the Solar system (Poddar et al., 2020)
 - S2 star around Sagittarius A^* (Borka et al., 2016)

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Jupiter Near-Polar Orbiter (JUNO) Mission

The JUNO mission aims to perform a comprehensive study on the planet Jupiter by:

• Mapping its gravitational and magnetic fields.

• Exploring its structure of polar magnetosphere and auroras.

• Mapping variations in atmospheric composition, temperature and cloud opacity.

• Determining the abundance of water and place an upper limit on the mass of Jupiter's possible solid core.

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Jupiter Near-Polar Orbiter (JUNO) Mission





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Probing the Fifth Force with JUNO

- Jupiter's Gravity field data reconstructed in the form of:
 - Central mass value $\rightarrow G \times M$
 - **2** Gravity anomaly coefficients $\rightarrow J_i$ etc.
 - 3 Tidal Love numbers $\rightarrow h$
- Fifth force perturbation \Longrightarrow perturbation in orbital parameters
 - $(\delta a, \delta e, \delta \Omega, \delta \omega, \delta i)$
- We constrain fifth force via the precession angle of JUNO's orbit (ω):

$$\Delta \omega_{\text{fifth-force}}^{2} < \sum_{i=2}^{12} \left| \frac{\partial \omega}{\partial J_{i}} \right|^{2} \sigma_{J_{i}}^{2}$$

$$\sigma_{J_{i}} \rightarrow \text{Uncertainty in } J_{i}$$

$$(4)$$

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• Unique from geocentric/ heliocentric celestial body probes.

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Precession of Orbits



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Fifth Force as Perturbation Force

 Perturbed force in Satellite-Normal co-ordinate system (RTN) is of the form:

$$\vec{F} = R\hat{e}_R + N\hat{e}_N + T\hat{e}_T$$

• For Yukawa force term we have:

$$R = -\alpha_2 \frac{GM}{(1 + \alpha_1)r^2} \left(1 + \frac{r}{\lambda}\right) \exp(-r/\lambda)$$

$$N = T = 0$$
⁽⁵⁾

- For Satellite-Normal CS (R-T-N):
 - \hat{e}_R points along the Jupiter \rightarrow JUNO.
 - \hat{e}_N along \vec{h} (angular momentum)
 - \hat{e}_T is defined by the right hand rule: $\hat{e}_T \cdot v > 0$



The Zonal Harmonics

- The non-spherical feature of the planet, can be accounted by expanding Newtonian potential in spherical harmonics basis.
- For our work, we only focus on zonal harmonics:

$$U_{\text{zonal}}(r,\phi) = \frac{GM}{r} \sum_{n=2}^{\infty} J_n \left(\frac{R_J}{r}\right)^n P_n(\sin\phi)$$
(6)

- Coordinates given by:
 - $\phi \rightarrow \,$ declination from equatorial plane.
 - $r \rightarrow radius$
 - $\lambda \rightarrow$ right ascension (from a meridian)



• Plugging Perturbation force in:

$$\dot{\Omega} = \sqrt{\frac{a\left(1-e^2\right)}{\mu}} \frac{N\sin(\omega+\nu)}{\sin i(1+e\cos\nu)}$$
$$\dot{\omega} = -\dot{\Omega}\cos i + \sqrt{\frac{a\left(1-e^2\right)}{e^2\mu}} \left(-R\cos\nu + T\frac{\left(2+e\cos\nu\right)\sin\nu}{1+e\cos\nu}\right)$$

• Average change over an orbit:

$$\frac{d\omega}{d\nu} = \frac{\dot{\omega}}{h/r^2}$$

$$\Rightarrow \left. \frac{d\omega}{d\nu} \right|_{AV} = \frac{1}{2\pi} \int_0^{2\pi} \frac{d\omega}{d\nu} d\nu \tag{7}$$

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Constraints on Yukawa Parameters



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- Fifth force arises due to new light, weakly-coupled degrees of freedom in BSM physics.
- They can be modeled in the form of Yukawa potential.
- Fifth force can be constrained via deviations from Newtonian orbits of various celestial objects.
- We constrain fifth-force via precession angle (ω) of JUNO's orbit around the Jupiter.
- Constraints obtained are shown to be most relevant in the force range region $\lambda \approx 10^{-4}-10^{-2}~{\rm AU}$

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Curious to know more? Stay tuned!

Supplementary Slides

Orbital Parameters

- a → semi-major axis (half the distance between the apojove and perijove)
- $e \rightarrow$ eccentricity of the ellipse
- $\bullet \ \nu \rightarrow$ defines the position of the orbiting body along the ellipse at a specific time



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Orbital Parameters

- $i \rightarrow$ vertical tilt (inclination) of the ellipse with respect to the reference plane, measured at the ascending node
- $\bullet \ \omega \rightarrow$ angle measured from the ascending node to the perijove of the orbit
- $\bullet~\Omega \rightarrow$ Longitude of the ascending node



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Orbital Parameter Perturbations

Taking Newtonian orbit to be ellipse (unperturbed): r = \frac{a(1-e^2)}{1+e\cos\nu}
 Perturbation in orbital parameters:

$$\dot{a} = 2\sqrt{\frac{a^3}{\mu(1-e^2)}} \left[eR\sin\nu + T(1+e\cos\nu) \right]$$

$$\dot{e} = \sqrt{\frac{a(1-e^2)}{\mu}} \left[R\sin\nu + T\left(\cos\nu + \cos\left(\sqrt{\frac{1-e}{1+e}}\tan\frac{\nu}{2}\right)\right) \right]$$

$$\frac{d}{dt}i = \sqrt{\frac{a(1-e^2)}{\mu}} \frac{N\cos(\omega+\nu)}{1+e\cos\nu}$$

$$\dot{\Omega} = \sqrt{\frac{a(1-e^2)}{\mu}} \frac{N\sin(\omega+\nu)}{\sin i(1+e\cos\nu)}$$

$$\dot{\omega} = -\dot{\Omega}\cos i + \sqrt{\frac{a(1-e^2)}{e^2\mu}} \left(-R\cos\nu + T\frac{(2+e\cos\nu)\sin\nu}{1+e\cos\nu} \right)$$

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(9)

- Coordinates given by:
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 - $\lambda \rightarrow$ right ascension (from a meridian)



Precession due to J_n Perturbation

$$U_{J_n}(r,\phi) = \frac{GM}{r} J_n\left(\frac{R_J}{r}\right)^n P_n(\sin\phi)$$
(10)

- In Jupiter-centered inertial (JCI) frame: $\phi = \frac{z}{r}$
- Perturbation force due to J_n term:

$$\vec{F} = -\frac{\partial U_{J_n}}{\partial r}\hat{e}_R + \frac{\partial U_{J_n}}{\partial z}\hat{e}_z$$
(11)

• For JCI \longrightarrow RTN frame:

$$\hat{e}_z = \sin i \sin(\omega + \nu) \hat{e}_R + \sin i \cos(\omega + \nu) \hat{e}_T + \cos i \hat{e}_N$$

$$z = r \sin \phi = r \sin i \sin(\omega + \nu)$$
(12)

• For brevity: $\theta = \omega + \nu$

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$\mathsf{JCI} \longrightarrow \mathsf{RTN} \ \mathsf{Coordinates}$



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• Plugging Perturbation force in:

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$$\dot{\omega} = -\dot{\Omega}\cos i + \sqrt{\frac{a\left(1-e^2\right)}{e^2\mu}} \left(-R\cos\nu + T\frac{\left(2+e\cos\nu\right)\sin\nu}{1+e\cos\nu}\right)$$

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(13)

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The Harmonics

• Spherical harmonic expansion of potential is given by:

$$U(\phi, \lambda, r) = rac{GM}{r} + U_{zonal} (r, \phi) + U_{sectorial} (r, \phi, \lambda) + U_{tesseral} (r, \phi, \lambda)$$

$$(14)$$

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- Coordinates given by:
 - $\phi \rightarrow \,$ declination from equatorial plane.
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$$U_{\text{zonal}}(r,\phi) = \frac{GM}{r} \sum_{n=2}^{\infty} J_n \left(\frac{R_J}{r}\right)^n P_n(\sin\phi)$$
(15)

- $P_n \rightarrow$ Legendre Polynomial
- Zonal harmonics vary only with latitude.



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$$U_{\text{sect}}(r,\phi,\lambda) = \frac{GM}{r} \sum_{n=2}^{\infty} \left(C_{n, \text{ sect}} \cos(n\lambda) + S_{n, \text{sect}} \sin(n\lambda) \right) \left(\frac{R_J}{r} \right)^n P_n\left(\sin\phi\right)$$
(16)

- Divides globe into slices by longitude.
- Varies only with longitude.



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$$U_{\text{tesseral}} (r, \phi, \lambda) = \frac{\mu}{r} \sum_{i,j=2}^{\infty} \left(C_{i,j} \cos(i\lambda) + S_{i,j} \sin(i\lambda) \right) \left(\frac{R_J}{r} \right)^i P_{i,j} (\sin \phi)$$
(17)

• Divides globe into slices by longitude and latitude.



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