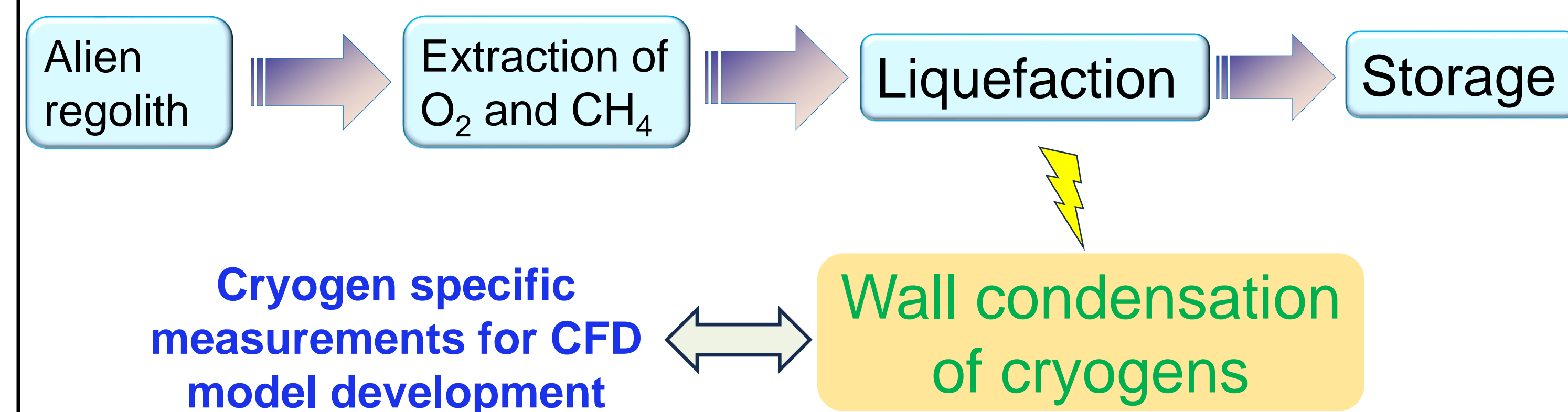


## 1 Introduction

- In-space Resource Utilization (ISRU) technologies are key to long-term and mass efficient lunar and mars missions<sup>1</sup>
- ISRU includes design of hardware/processes for recovery, **liquefaction**, transport, and storage of cryogens

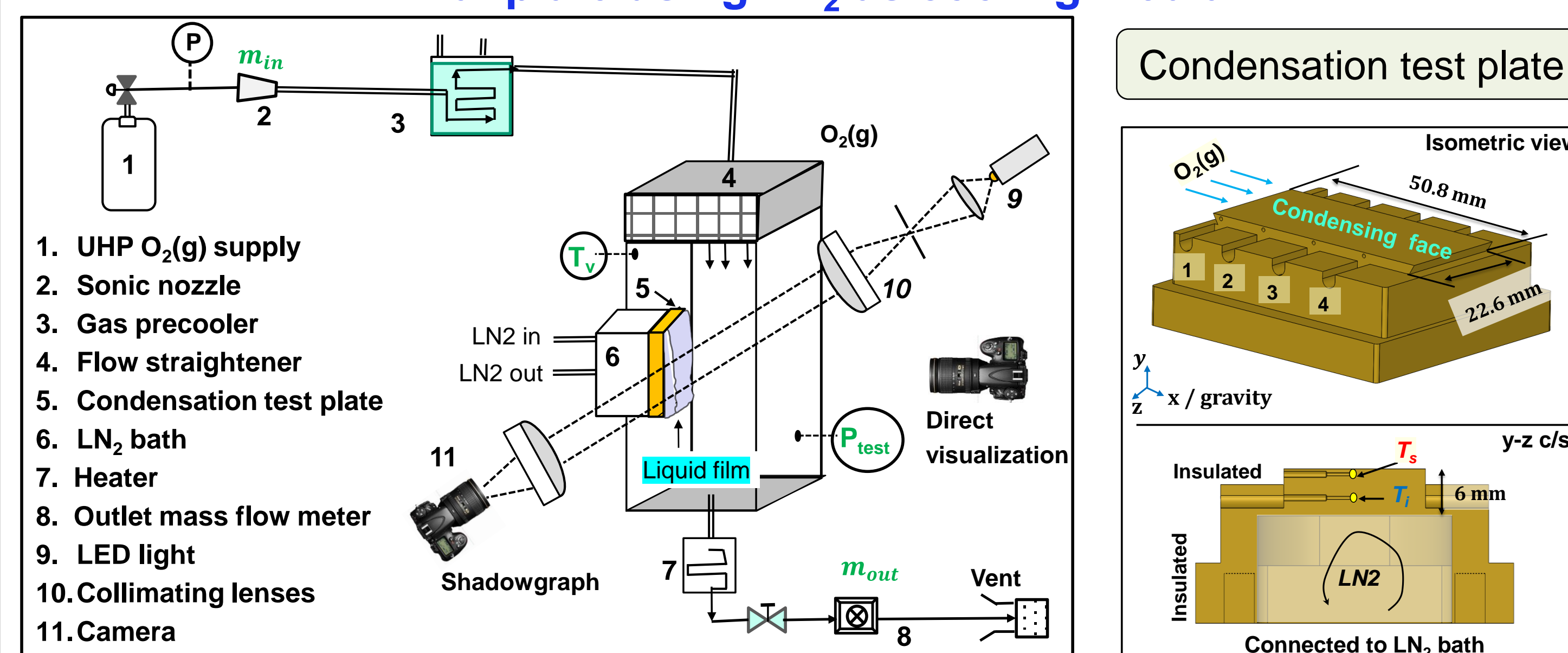


## 2 Research objectives

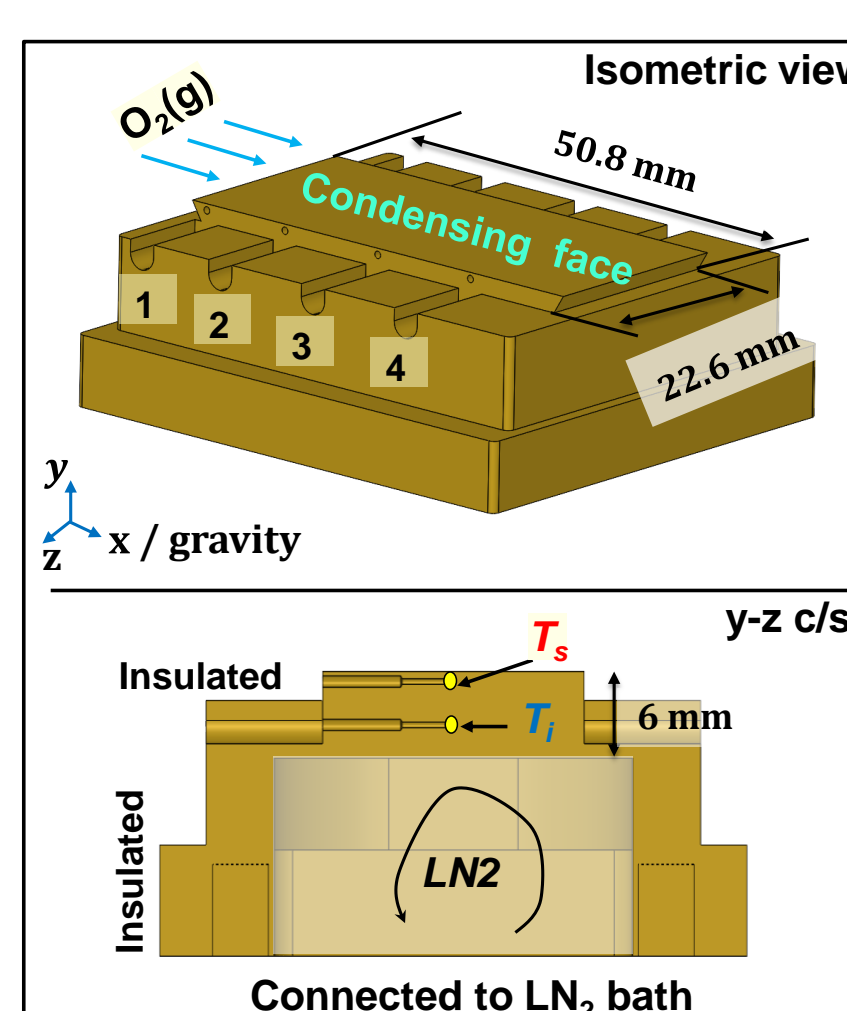
- Develop a test rig to induce wall condensation of ISRU relevant cryogens : O<sub>2</sub>, CH<sub>4</sub>, and N<sub>2</sub>
- Visualization study of onset of condensation with inline process measurements
- Estimation of steady state condensate film thickness using measured heat flux and a visualization technique

## 3 Experimental setup

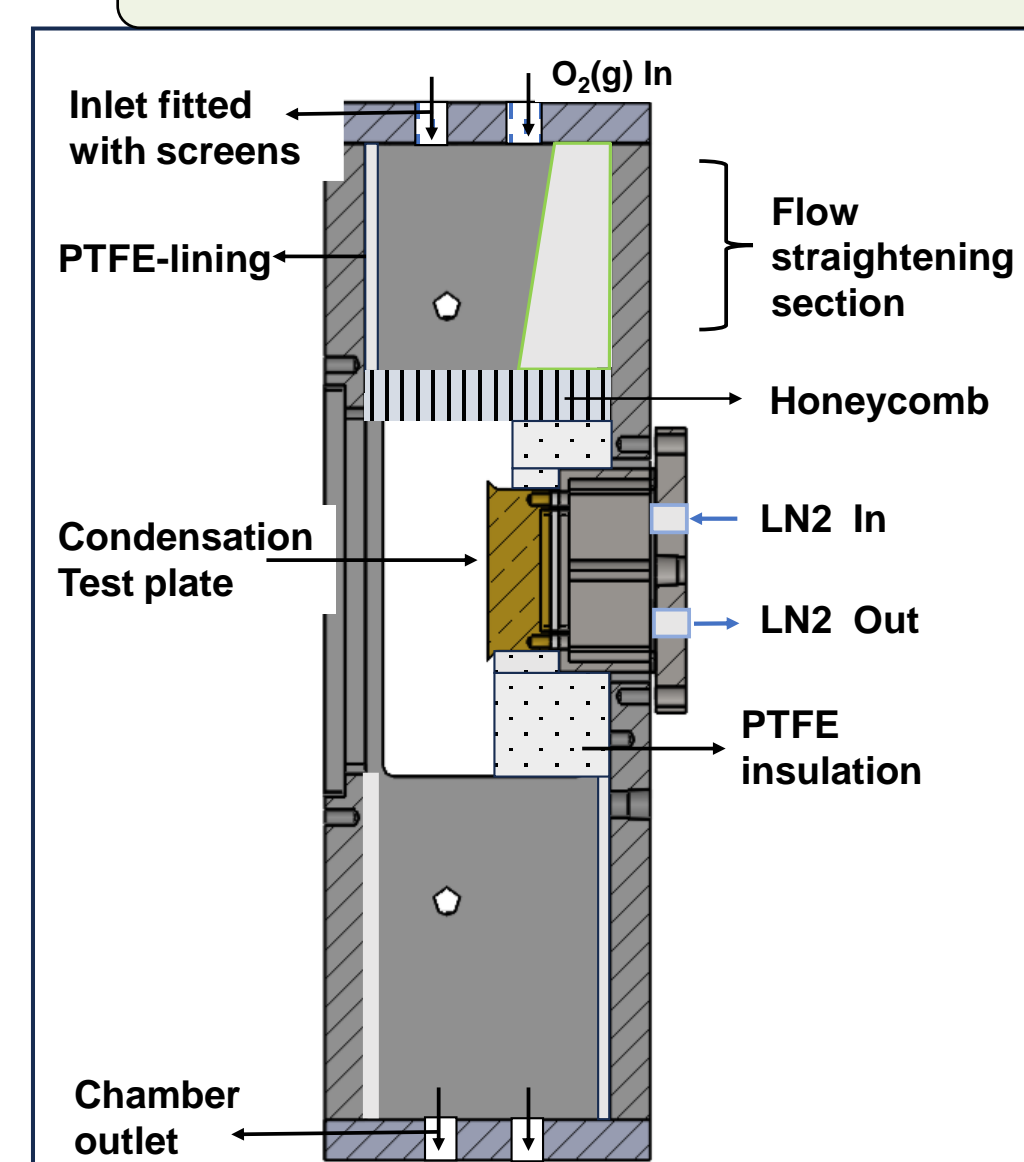
Test-rig to induce laminar film condensation of oxygen on a vertical flat plate using LN<sub>2</sub> as cooling media



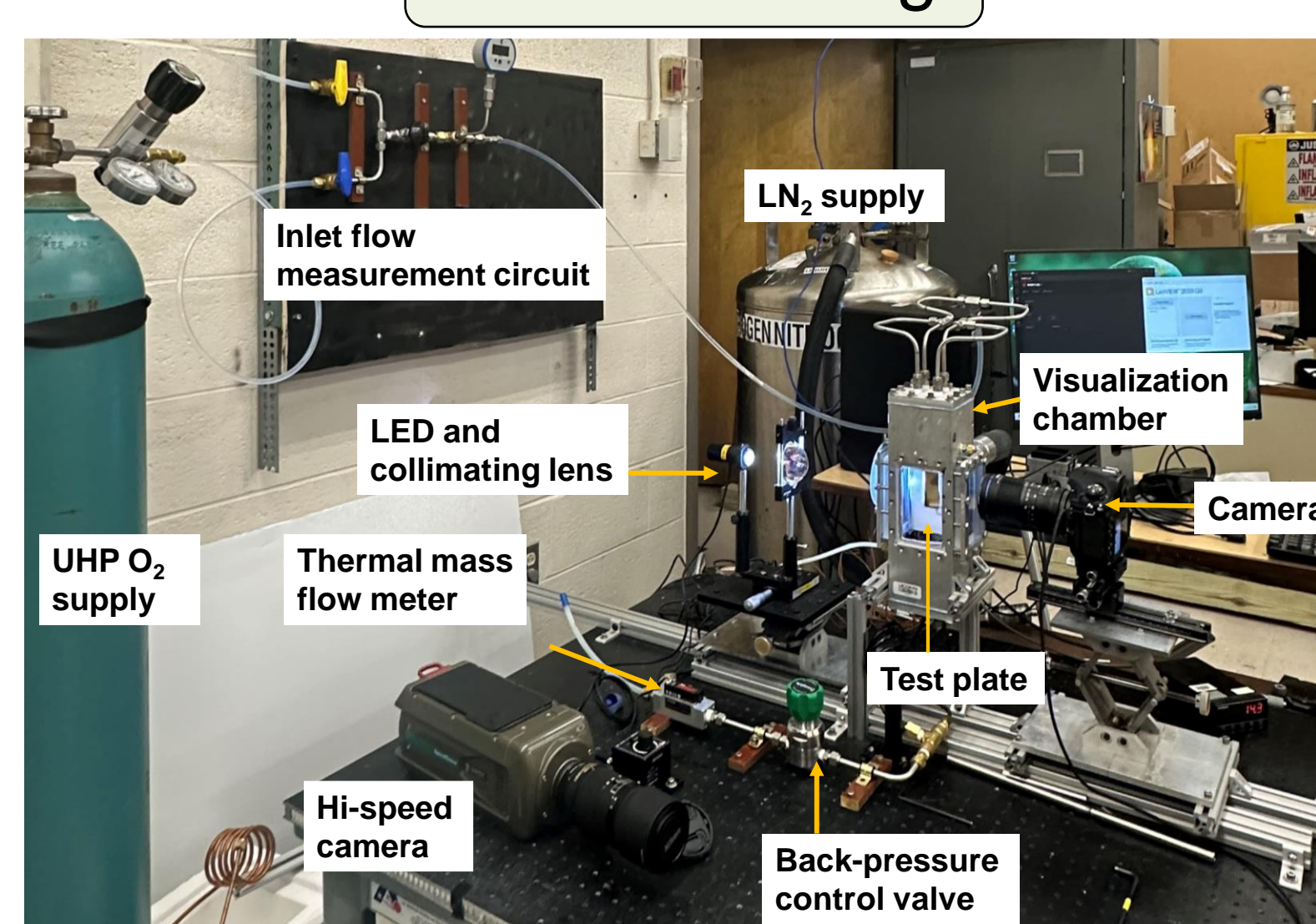
Condensation test plate



Visualization chamber c/s

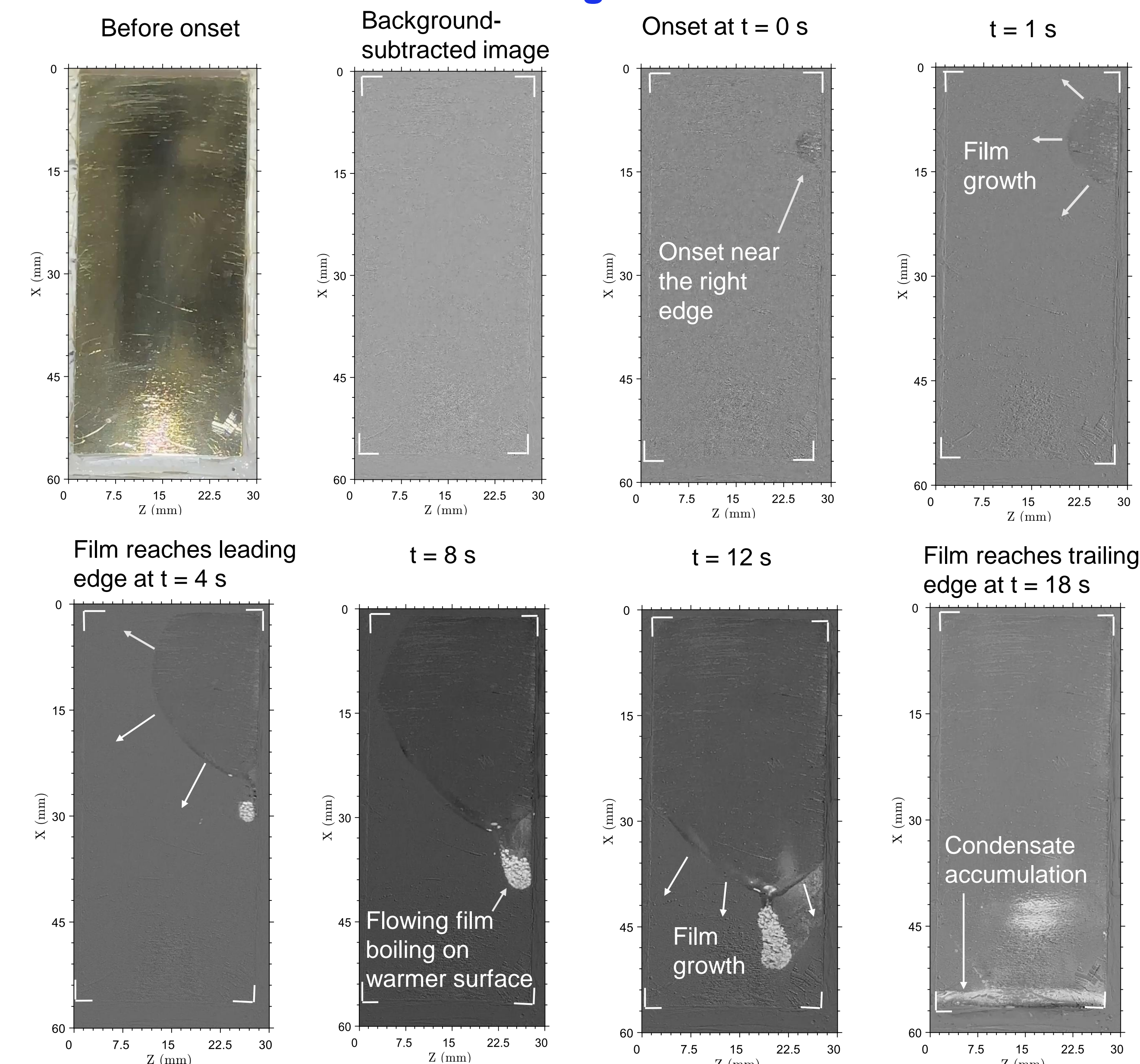


UConn test-rig

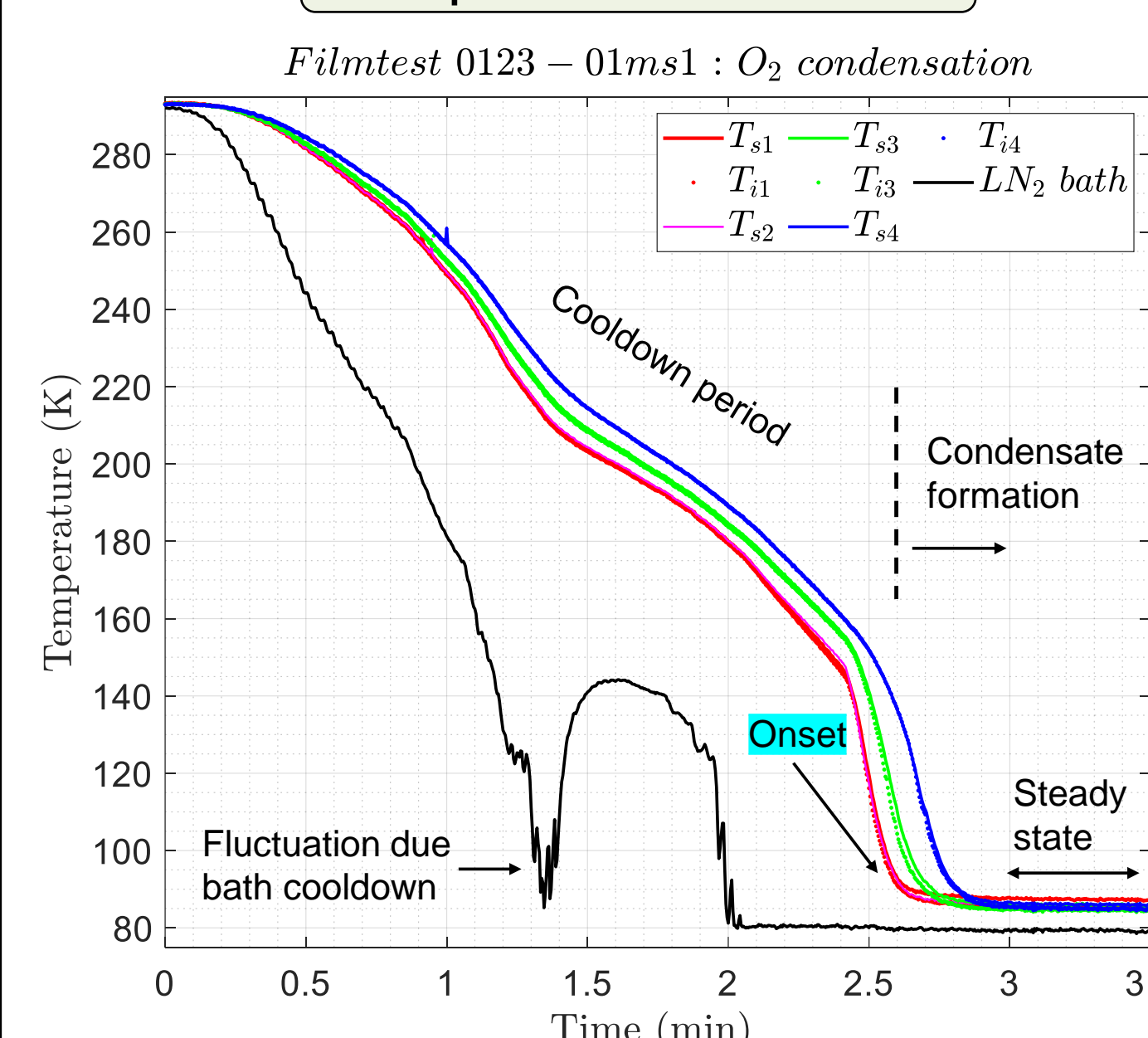


## 4 Results: Visualization of condensation onset

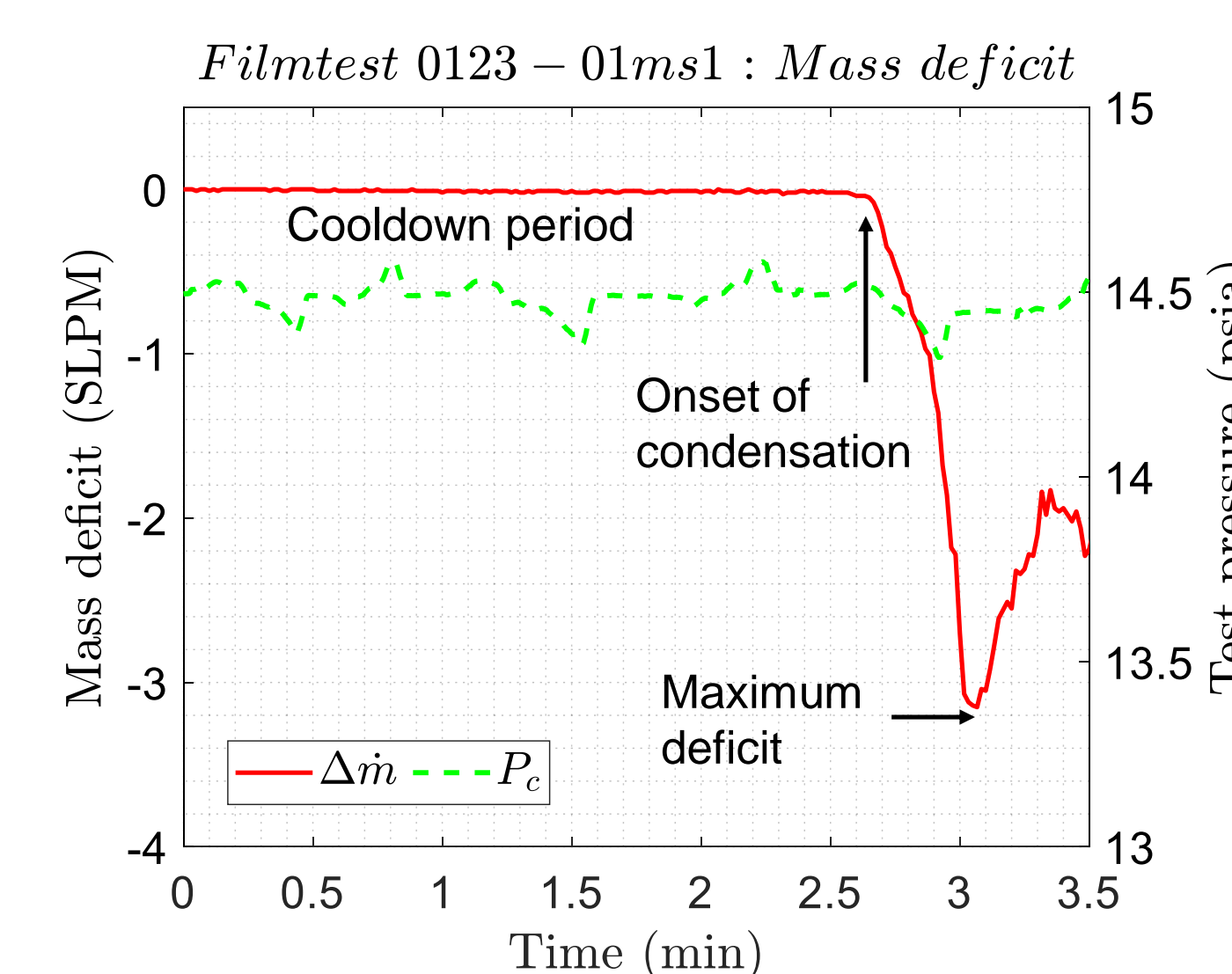
Condensate front grows from a local spherical kernel to a thin film covering entire surface



Temperature evolution



Mass deficit



Condensation onset is coupled with LN<sub>2</sub> bath cooldown

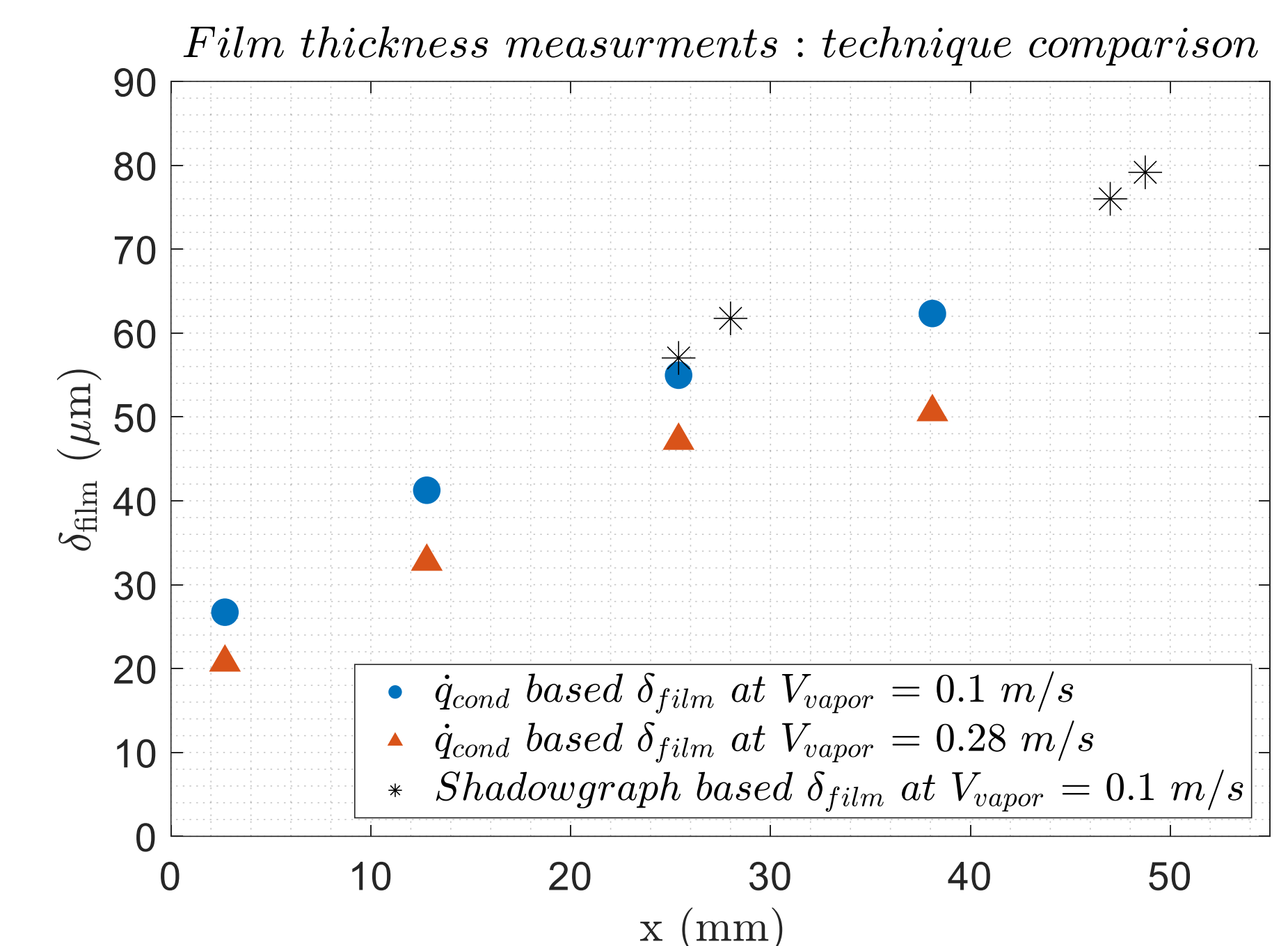
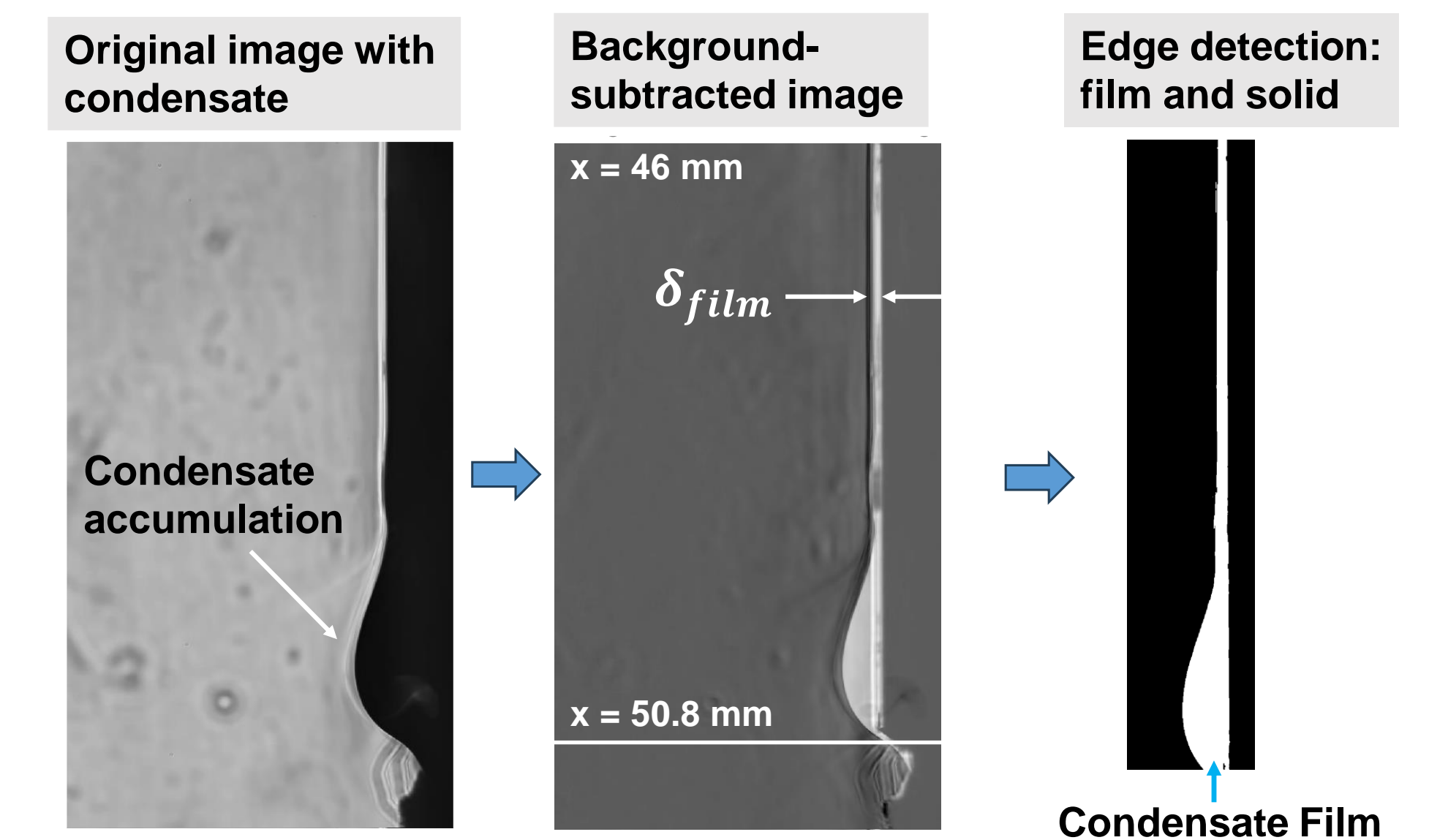
Estimated heat transfer coefficient and film thickness at steady state

Inlet conditions	Stream wise location (mm)	Surface TC T <sub>s</sub> (K)	Internal TC T <sub>i</sub> (K)	HTC (Wm <sup>-2</sup> K <sup>-1</sup> )	Film thickness (μm)
V <sub>vapor</sub> = 0.1 m/s	z = 2	T <sub>s1</sub> = 86.44	T <sub>i1</sub> = 84.06	7231	21.2
T <sub>vapor</sub> = 289 K	z = 12.7	T <sub>s2</sub> = 85.13	T <sub>i2</sub> = 83.27	4094	37.8
P <sub>test</sub> = 14.5 psia	z = 25.4	T <sub>s3</sub> = 85.08	T <sub>i3</sub> = 83.73	2949	52.4
	z = 38.1	T <sub>s4</sub> = 85.08	T <sub>i4</sub> = 83.95	2486	62.2

LOx laminar condensate film is of microscale thickness

## 5 Results: Film thickness visualization

Film thickness is extracted from high-resolution shadowgraph



Shadowgraph technique capture microscale LOx condensate film thickness

## 6 Conclusions and future work

- Test-rig is developed to induce condensation of cryogen vapor using LN<sub>2</sub> as cooling media. Visualization of condensation phenomena and film thickness measurement are conducted
- Onset of condensation occurs at local cold spot. Condensation front forms and grows as a spherical kernel initially. As the condensate mass builds up, gravity causes flow of condensate
- Steady state film thickness measurement shows microscale nature of formed condensate film for cryogens
- High resolution shadowgraph technique is developed for film thickness measurement. Measurement of film thickness using visualization-based technique compares well with heat flux-based estimation

Future work

- Parametric studies: Impact of wall temperature and orientation on condensation phenomena
- Computational modelling of condensation of cryogens

## 7 References

- Johnson et al. (2018). Comparison of oxygen liquefaction methods for use on the Martian surface. *Cryogenics*, 90, 60-69.
- Settles, G. S. (2001). *Schlieren and shadowgraph techniques: Visualizing phenomena in transparent media*. Springer Science & Business Media.

## Acknowledgment

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