

# **RF** discharges

**Principle:** 

- Acceleration of electrons in an oscillating electric field with amplitudes < source dimensions</li>
- Electrons gain energy, if there is "friction" (i. e. collisions)
- Ionizing collisions
- Equilibrium between ionisation and loss rates

Frequency range: 0.1 – 30 MHz

Power range: 50 W - 800 kW

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# Illustration of the RF absorption



# Three different ways of RF coupling



P<sub>abs</sub> decreases at high frequency

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### Skin effect

The e.m wave vary as		$\propto {f e}^{i(kx-\omega t)}$	
If the RF frequency is much smaller than the plasma frequency the wave decays		$\omega << \omega_{ m p}$	
exponentially		$\propto {m e}^{-{m x}/\delta_{m s}-i\omega t)}$	
Decay length is the Skin depth Collisionless		$\delta_{e} = \mathbf{c}/\mathbf{\omega}_{e} = (\frac{m_{e}}{2})^{1/2}$	
	Collisional	$\delta_s = (\frac{2}{\omega\mu_0\sigma})^{1/2}$	Typically 0.5 – 2 cm, decreases at high frequency(!) and conductivity σ
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# **Electron temperature**

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# **Electron temperature distribution**

- Depends on the distance to the coil
- Determined by the gas density (pressure)
- At high energy reduced by inelastic collisions



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# Energy loss per electron-ion pair created



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$$P_{loss} = \frac{n_e}{\tau_p} \cdot (m \cdot E_{ion}) = P_{abs}$$

m represents energy losses by excitation of vibrational and rotational energy levels, molecular dissociation, energy loss at the wall



*mE*<sub>ion</sub> : energy needed for an electron-ion pair

- Can be measured by the decay time of the plasma
- Is for molecular gas one order of magnitude higher than the ionisation energy

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- Local  $E_{RF}$  field in has to be known
- E<sub>RF</sub> field homogenous (no skin effect)
- E<sub>RF</sub> field constant (not dependent on the plasma parameters)
- Low power, because  ${\rm B_{RF}}$  field not considered (50 -100 G at 100 kW)
- No Coulomb collisions

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# Capacitive discharges

High RF voltage drop in the cathode sheath

RF frequency << ion plasma frequency

 $\Rightarrow$  lons are accelerated in the sheath

⇒ most of the RF power goes for the ion acceleration bombardement of the electrodes by energetic ions some keV

=> Used for surface treatment in the plasma technology

- RF frequency >> ion plasma frequency
- $\Rightarrow$  lons cannot follow the RF field
- $\Rightarrow$  Low ion energy of some 10 eV

Transition region 5 – 10 MHz



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# Schematic of a CCNP Processing Chamber



IPP





# Pressure balance at high power



$$n_{n,w}kT_w = n_nkT_n + n_ekT_e + n_ikT_i$$



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# **ICP Enhanced with Ferrite Core**



V. Godyak, PSST 20, 025004, 2011



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#### **Faraday shield**

- Shields capacitive coupling
- protects insulator from chemical and physical sputtering

#### Ferrites

- Shields RF fields
- · Improves the coupling to the plasma

# Internal antenna

- Better coupling to the plasma
- Lower wall losses due to larger area of magnetic cusps

Problem: Lifetime of the insulation

- Porcelain coating
- Quartz tubing



Type V rf source



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#### Internal Faraday shield for high power ICPs Starter filament B<sub>RF</sub> field penetrates through the Internal Faraday shield slits even when they are Z-shaped 550505 • No power load on the insulator • $H_{\alpha}$ radiation in the driver not changed $\Rightarrow$ No additional power losses by eddy currents 0.5 z-shaped slits 0.4 /P<sub>RF</sub>) / (a.u.) 0.3 0 · · 0.2 ormal slits Driv 0.1 E 0.0 30 40 50 60 70 80 90 100 110 120 RF Power / (kW)

# Ignition of the plasma



breakdowns

**RF** Antenna

# **RF** sources for accelerators

Up to 100 kW at 2 MHz in small volumes (L ~10 cm,  $\emptyset$ ~ 5 cm) Pulse duration 0.5 ms with a repetition rate of 4 - 60 Hz 40 -80 mA H<sup>-</sup>current produced by surface conversion on Caesium surfaces



(ORNL Oakridge National Laboratory)

100kW/2MHz RF source of the LINAC4 accelerator (CERN)

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Tsiolkovsky rocket equation: $v_e = v_T \ln \frac{m_0}{m_e}$						
Maxim	um speed = exhaust velocity	x In(Initial mass/final mass)				
Chemical thrusters	small	large				
Electrical thrusters up to	25 x larger large	small				
<ul> <li>Small thrust (0.1 - 1 N) but</li> <li>Very reliable</li> <li>High propellant capacity</li> <li>propulsion energy provided by an electric source</li> <li>exact control of the thrust <ul> <li>=&gt; used for</li> <li>space missions, space probes</li> <li>orbit control of satellites</li> </ul> </li> </ul>						
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RF ion thrusters		IPP				
$Xe^{\circ}$	RIT 10 Giessen university					
Propellant: <b>Xenon</b> (high mass => high momentun 10 cm diameter, Thrust: 0.01 – 1 N	n => high thrust) Accele Power 4 MHz	eration voltage: ca 2 kV <sup>r</sup> supply: solar z, few 100 W,				

# RF ion sources for the Neutral Beam Injection systems of Fusion Reactors



# Helicon wave sources

RF antenna launches a wave, the **helicon wave**, that propagates along an static **Bfield** with a phase velocity comparable of a 50 – 200 eV electron

- Very efficient ionisation
- · Plasma density one order of magnitude higher than in ICPs

Helicon waves are **whistler waves** confined to a cylinder RH polarized e. m. waves propagating along  $B_0$ , wave vector k at an angle  $\Phi$  to  $B_0$ Dispersion relation of whistler waves



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# Helicon dispersion relation



At fixed  $\omega$ , radius of the source (=>  $k_{\perp}$ ), wavelength  $2\pi/k$ => Density n<sub>e</sub> proportional to the magnetic field

Boundary conditions in a cylindrical discharge for the wave which varies like

$$\vec{B} = \vec{B}(r) e^{i(m\phi + k_z z - \omega t)}$$

fulfilled with azimutal wave numbers m

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# Helicon modes in a cylindrical discharge

m = 0: changes from electrostatic (radial E) to electromagnetic (azimuthal E-lines)



m = 1: rotating E-field pattern, mostly right hand polarized observed (m=+1)







# Helicon antennas and energy transfer



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# Conclusion

# Chen & Chang ("Principles of plasma processing")

"In (source) plasma physics classical treatments like the above are doomed to failure, since plasmas are tricky and more often than not are found experimentally to disobey the simple laws of electromagnetics."

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