

DIRECTIONS FOR THE FUTURE

SUCCESSIVE ACCELERATION OF POSITIVE AND NEGATIVE IONS APPLIED TO SPACE PROPULSION



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Acknowledgments



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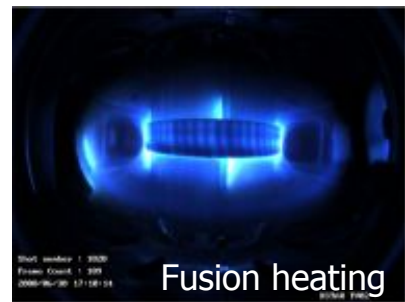
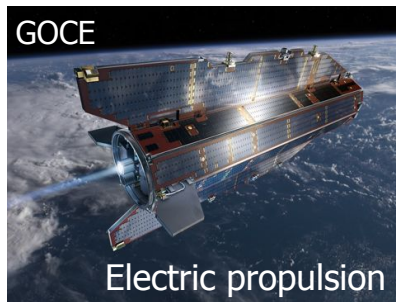
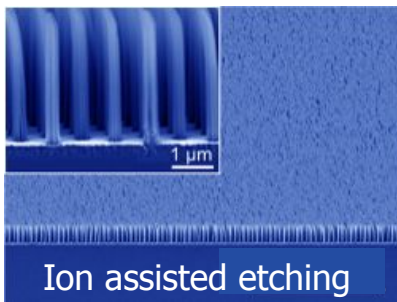


Electrons – Pros & Cons



Electrons required for plasma **generation** and beam **neutralization**

Electrons cause problems by surface and differential **charging** and slow ion-electron **recombination**



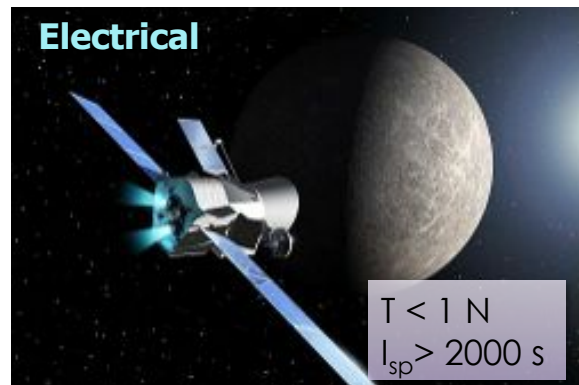
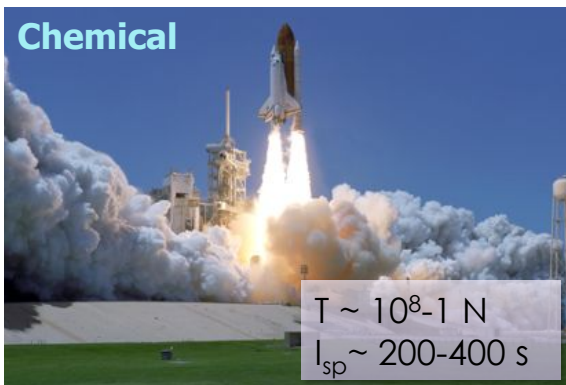
Space propulsion acceleration by ejecting mass



$$m \frac{dv}{dt} = - \frac{dm}{dt} v_g$$

}

Trust $T = \frac{dm}{dt} v_g$
Specific Impulse $I_{sp} = \frac{v_g}{g_0}$

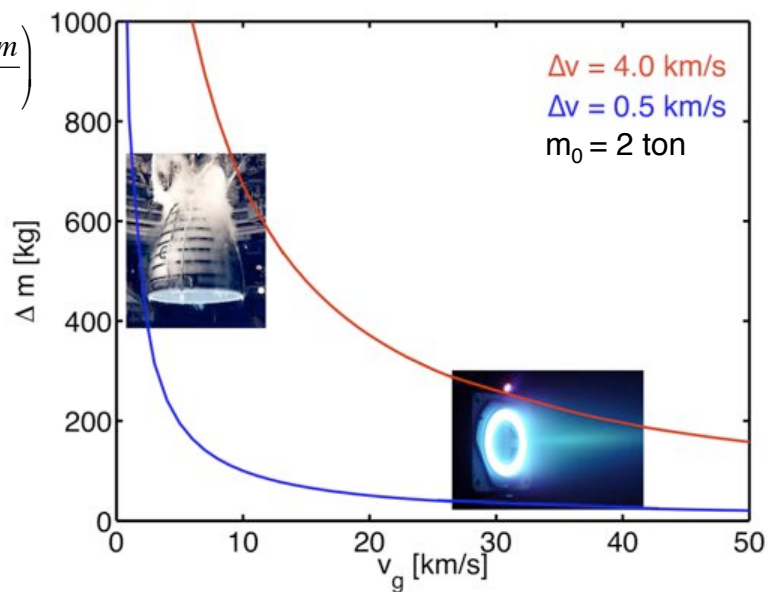


Mass consumption

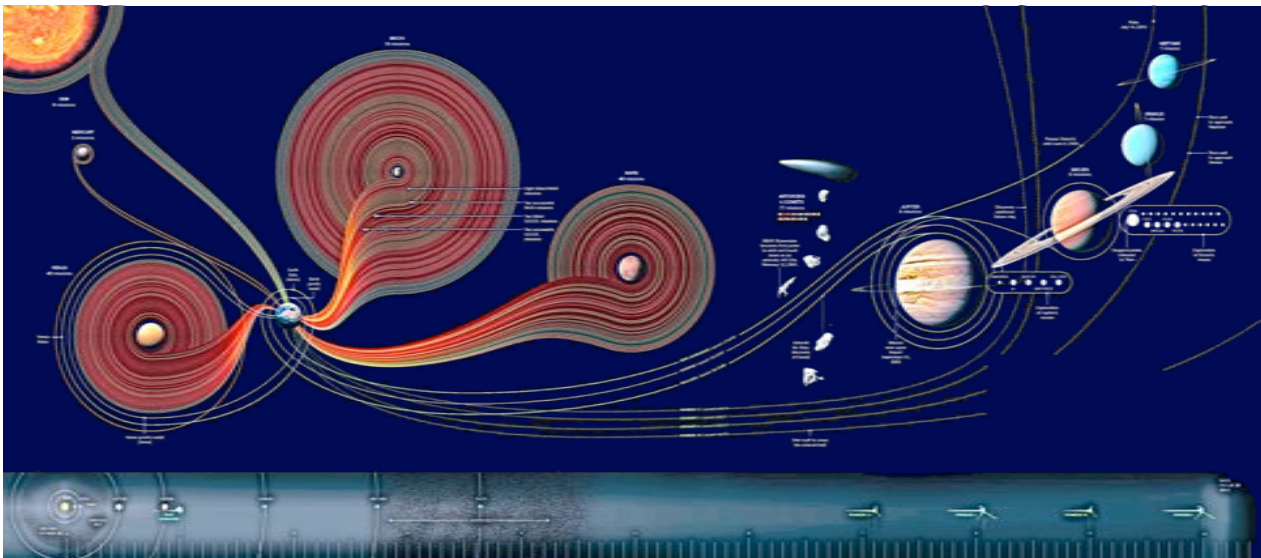
Chemical versus Electric Propulsion



$$\Delta v = v_g \ln\left(\frac{m_0 - \Delta m}{m_0}\right)$$



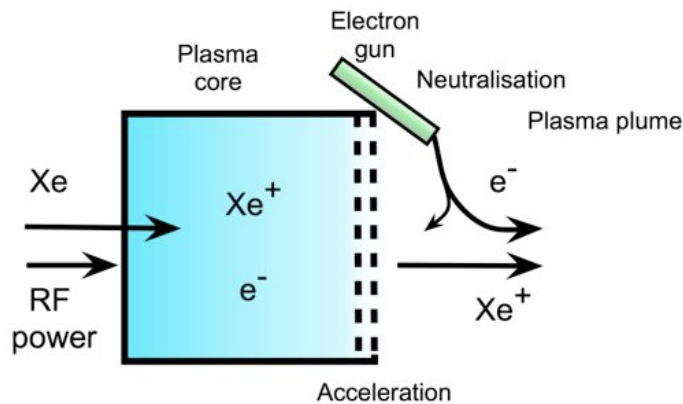
Cost to sent 1 kg to LEO is ~ 20 k€ !



Source: the big picture

SPACE MISSIONS SINCE THE 1950'S

Principle of Electric Propulsion



Two weak points:

- 1) Hollow cathode – limited lifetime and stability
- 2) Back scattering and sputtering

PEGASES

Plasma propulsion with electronegative gases



Stage 1

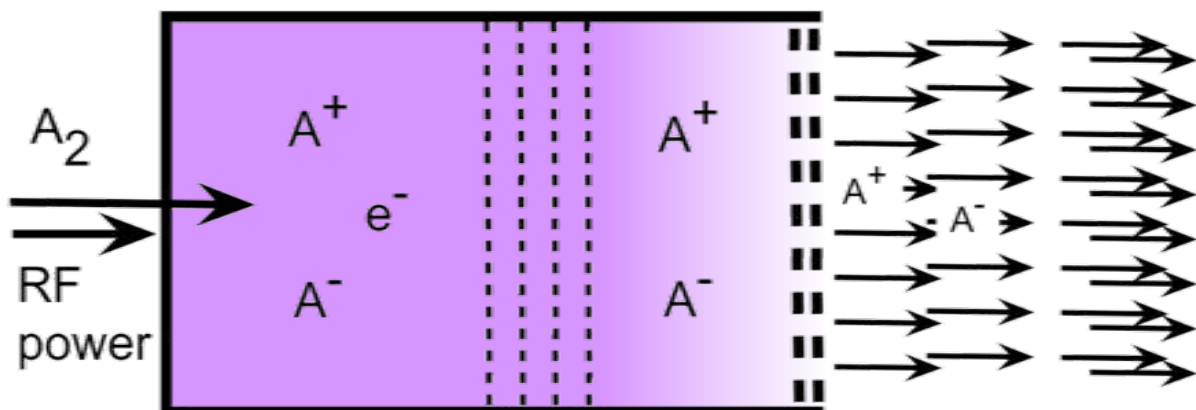
Plasma discharge, power coupling

Stage 2

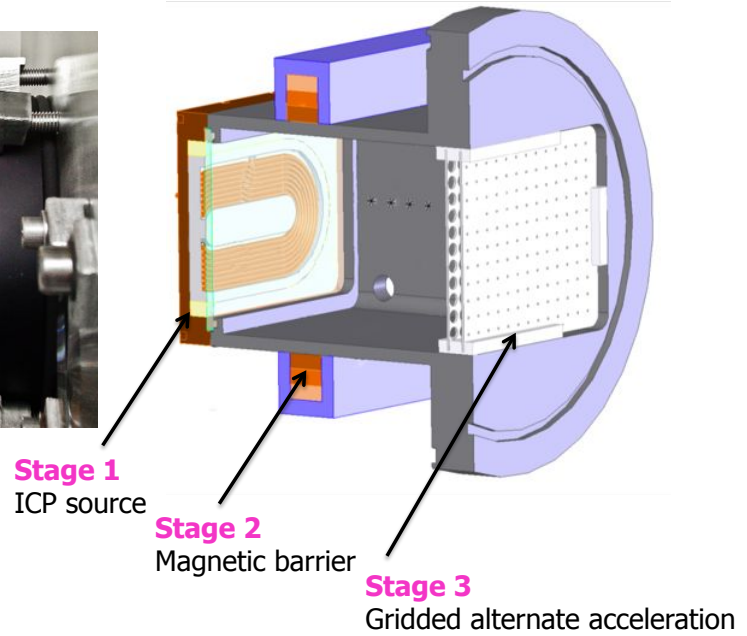
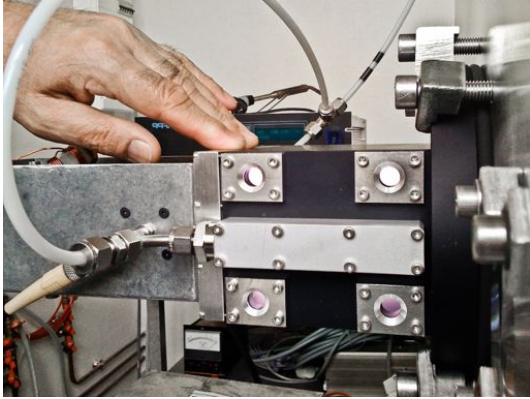
Electron filtering, ion-ion formation

Stage 3

Acceleration and recombination



PEGASES Prototype

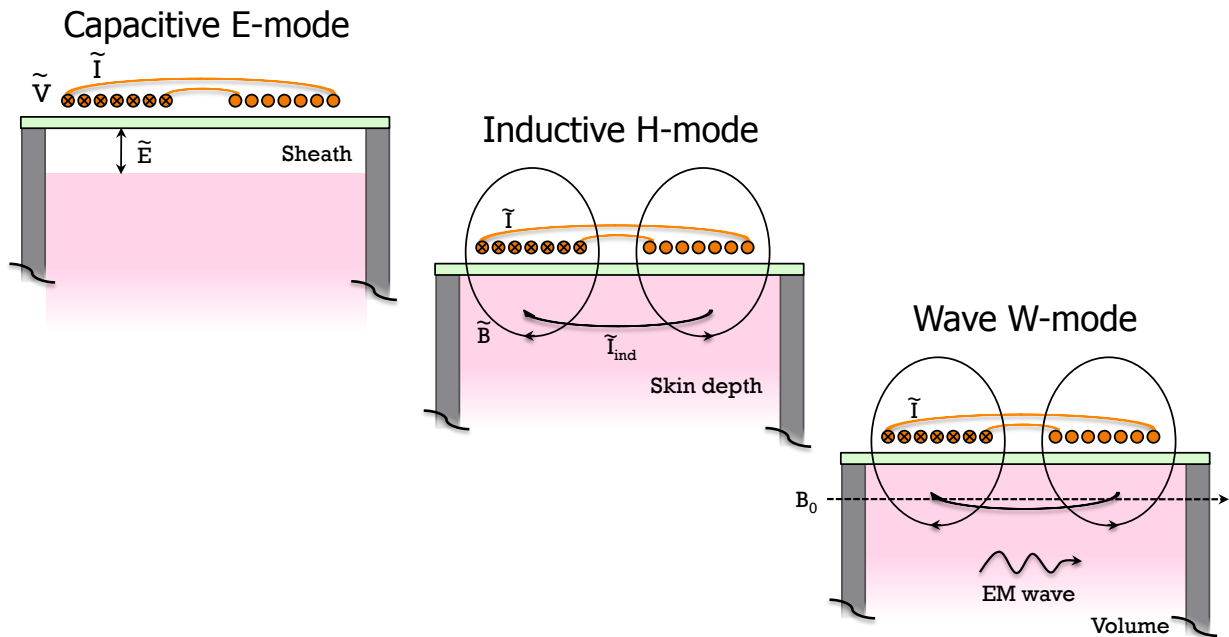


STAGE 1 PLASMA DISCHARGE WITH ELECTRONEGATIVE GASES



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RF plasma discharges

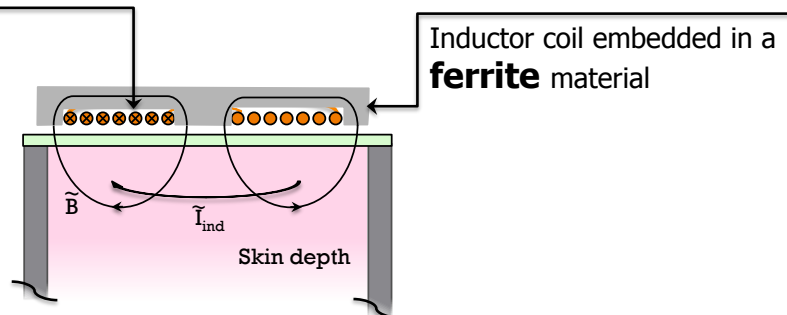
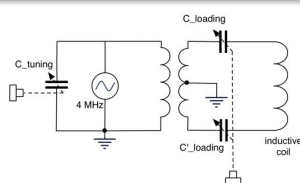


Inductively coupled plasma (ICP) with high efficiency



RF frequency: **4 MHz**

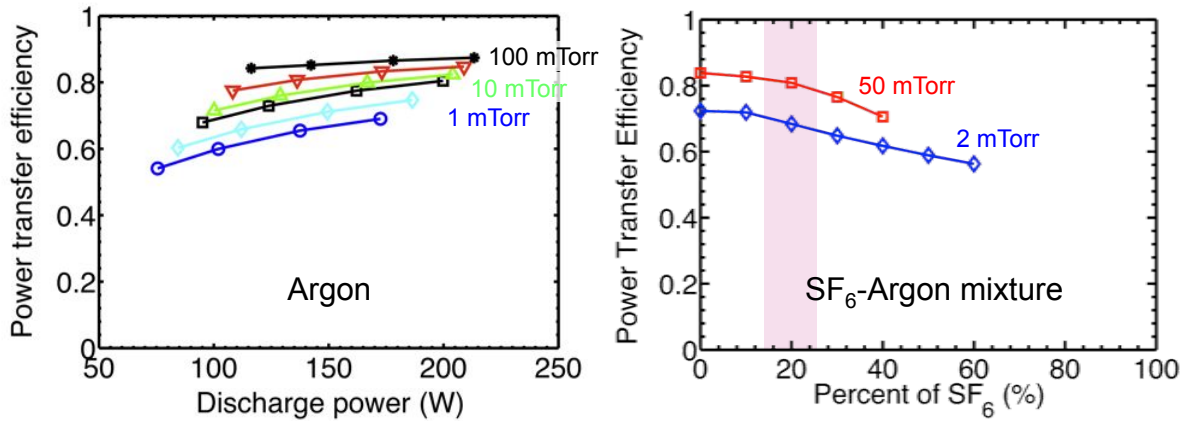
Z-matching: **Step-down transformer**



Requirements for space applications:

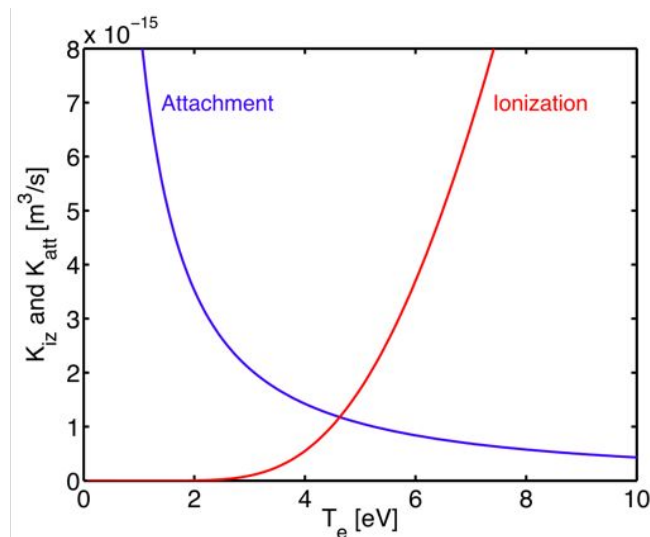
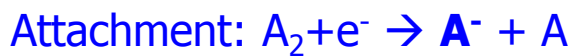
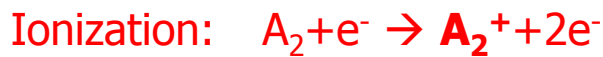
- Small and light
- Minimal energy loss
- Large parameter space in T and Isp

Power transfer efficiency in PEGASES



Up to 90 % power transfer efficiency in Argon
70 – 85 % efficiency in current PEGASES condition

Electronegative volume produced plasma

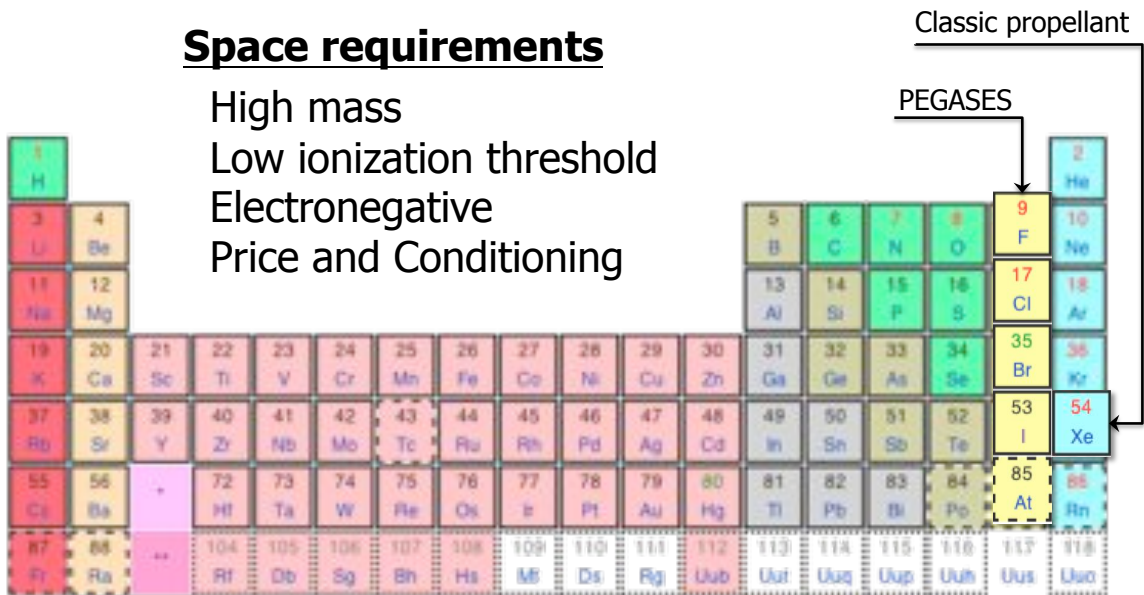


Propellant in the PEGASES thruster



Space requirements

- High mass
- Low ionization threshold
- Electronegative
- Price and Conditioning

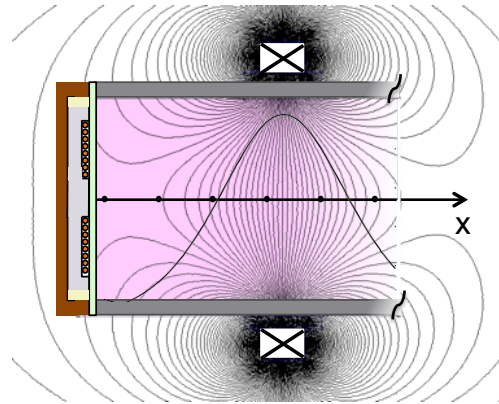
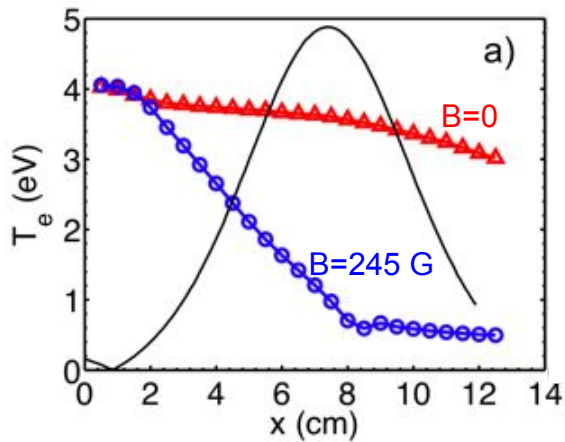


STAGE 2 MAGNETIC FILTER AND ION-ION FORMATION



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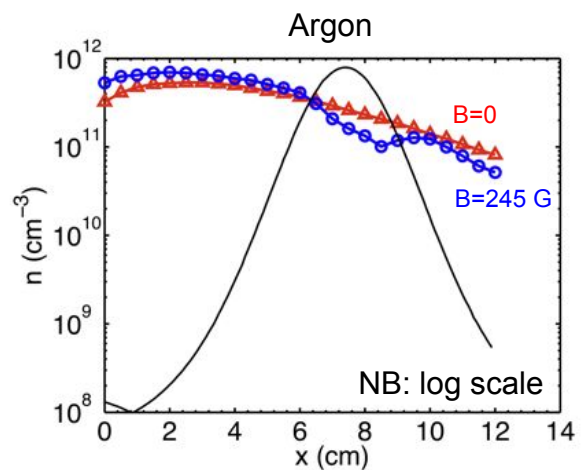
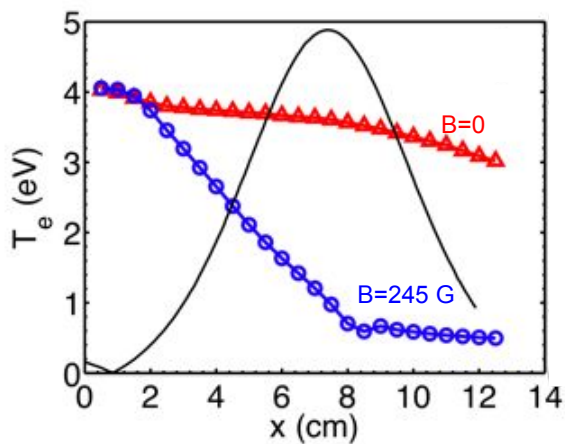
Control of the electron temperature



Cool down the electrons

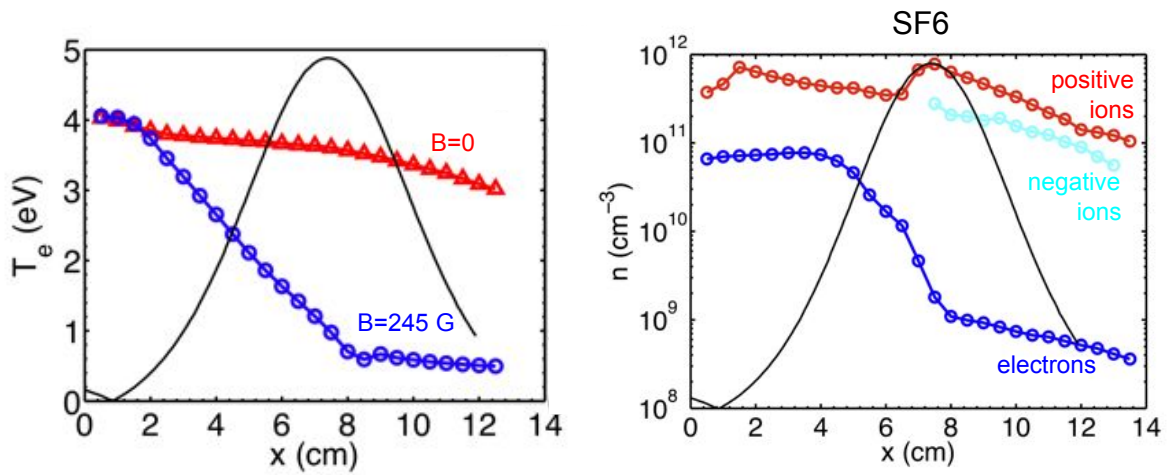
Control of high and low T_e region by pressure and B-field

Plasma density in the magnetic barrier



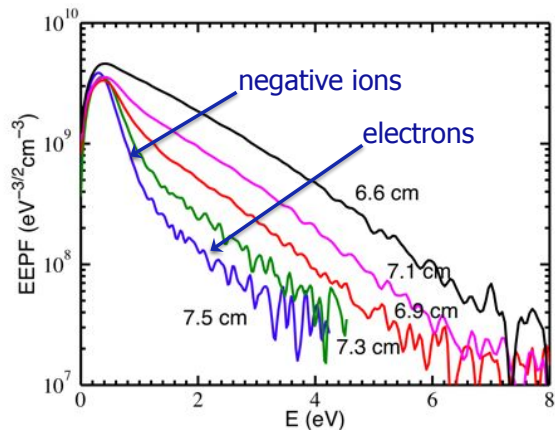
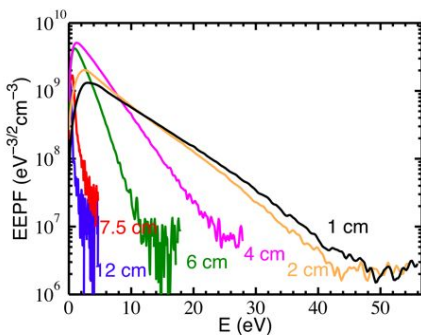
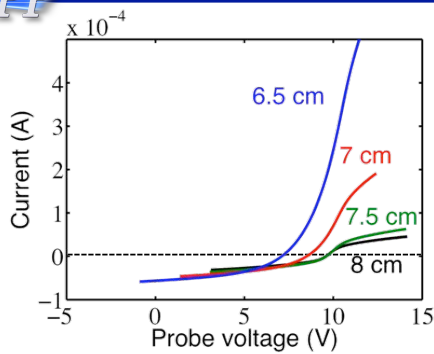
In **electropositive plasmas** the plasma density decreases strongly in the filter region

Ion-Ion plasma in the magnetic barrier



In **electronegative plasmas** the ion density remain high
 $n_i \sim 5 \times 10^{11} \text{ cm}^{-3}$ at 150 W

Langmuir probes in electronegative plasmas

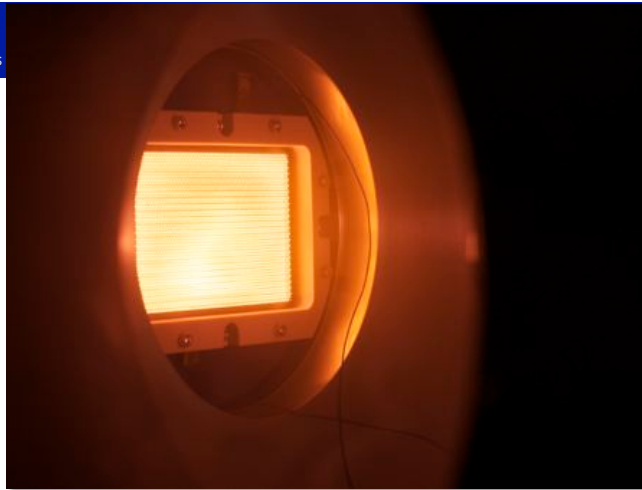


STAGE 3

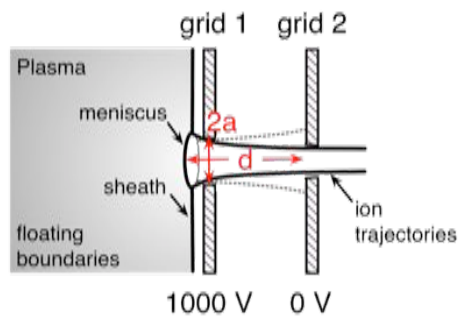
ALTERNATE ION ACCELERATION



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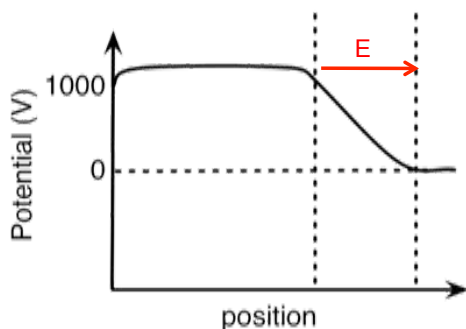


Classical gridded acceleration



The Child-Langmuir space charge limited current controls the **maximum Thrust**

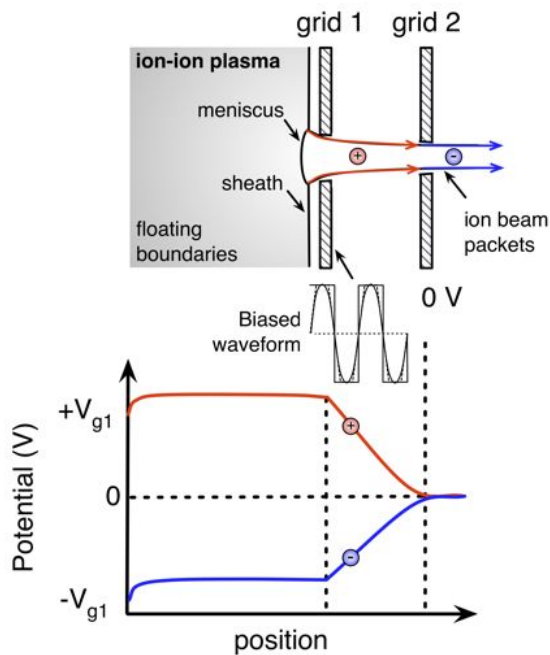
$$J_{CL} = \frac{4\epsilon_0}{9} \left(\frac{2e}{M} \right)^{1/2} \frac{V^{3/2}}{d^2}$$



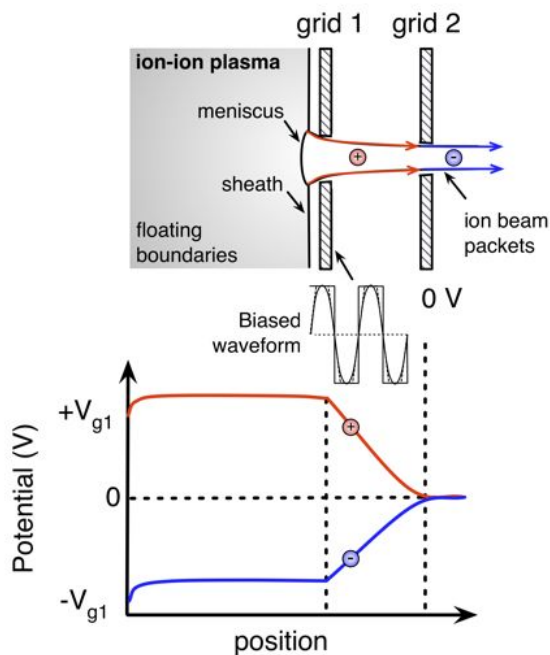
$$T \equiv v_i \frac{dm}{dt} = v_i \frac{I_b M_i}{e} = I_b \sqrt{\frac{2M_i V_b}{e}}$$

$$T_{\max} = \frac{8\epsilon_0}{9} \frac{A_g T_g}{d^2} V_b^2$$

Alternate acceleration – Concept



Alternate acceleration – Requirements



Waveform requirements

Upper limit:

$$\omega < \omega_{pi} \sim 10\text{-}20 \text{ MHz}$$

$$\omega < 1/\tau_{tof} \sim 1 \text{ MHz}$$

Lower limit:

Beam packet blowup

Beam oscillations

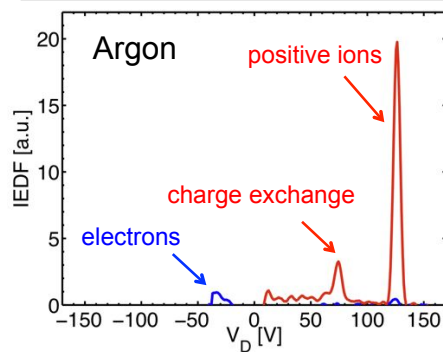
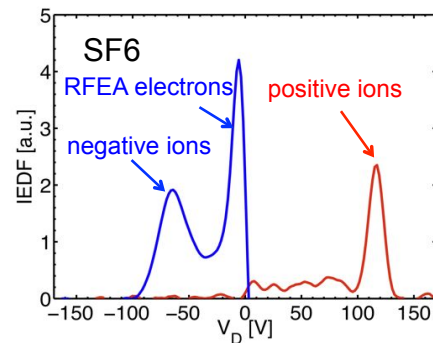
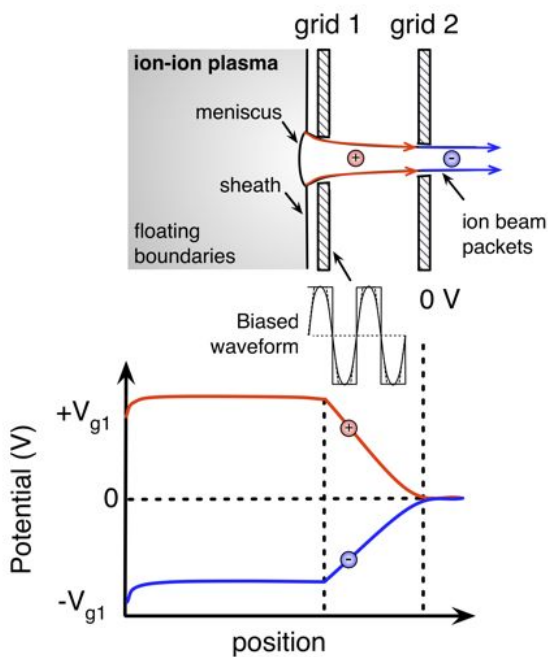
Estimated $\omega > \text{kHz}$

Optimization:

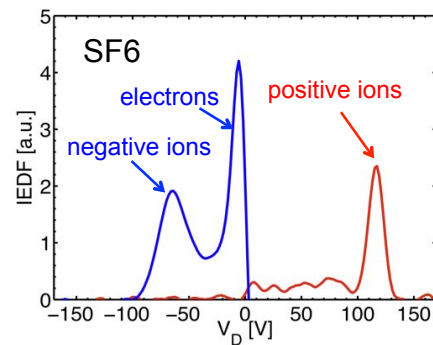
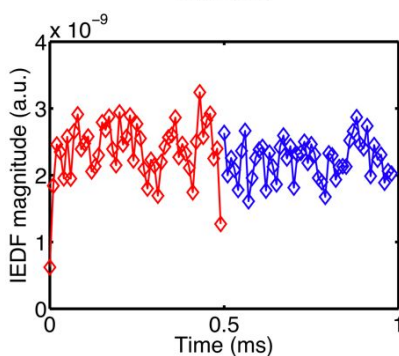
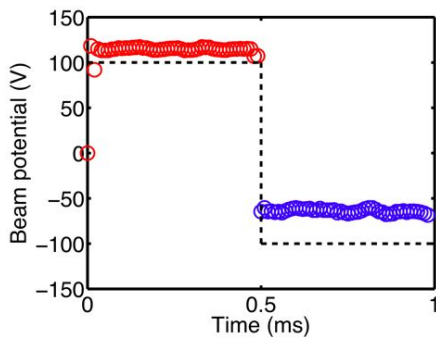
square waveforms

variable rise time and periods

Alternate acceleration Proof-of-Concept with ± 100 V at 1 KHz

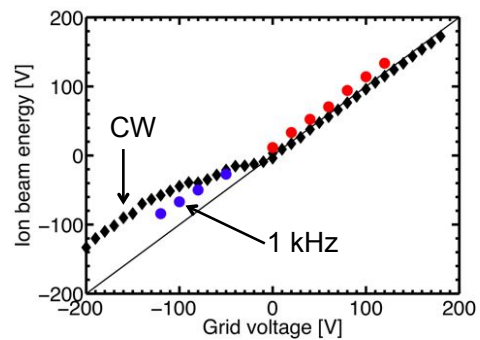
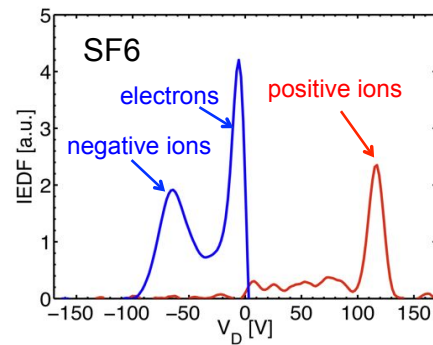
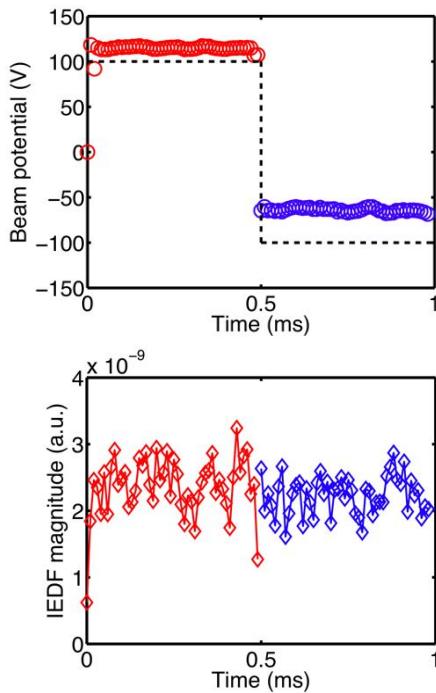


Alternate acceleration Ion beam energy



Positive ion beam at **+114 V**,
Negative ion beam at **-67 V**

Alternate acceleration Ion beam energy versus grid potential



PEGASES – from concept towards reality



Estimated Thrust

$$T = A_g \Gamma_i M_i v_b = A_g e n_i \sqrt{2T_i V_b}$$

$$n_i \sim 2 \times 10^{17} \text{ m}^{-3}, V_b = 200 \text{ V}, T = 0.5 \text{ eV}$$

$$T = 20 \text{ mN/kW}$$

with efficient ion-ion recombination

Stage 1

- Ferrite enhanced ICP source
- **70-85 %** power efficiency in current PEGASES conditions

Stage 2

- Segregation of the electronegative plasma
- Formation of ion-ion plasma
- High density **$5 \times 10^{11} \text{ cm}^{-3}$** at only 150 W

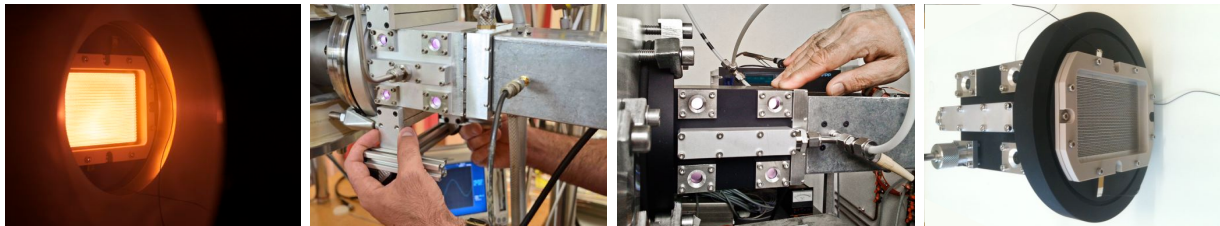
Stage 3

- Dual ion acceleration
- First **proof-of-concept**

THANK YOU FOR YOUR ATTENTION



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