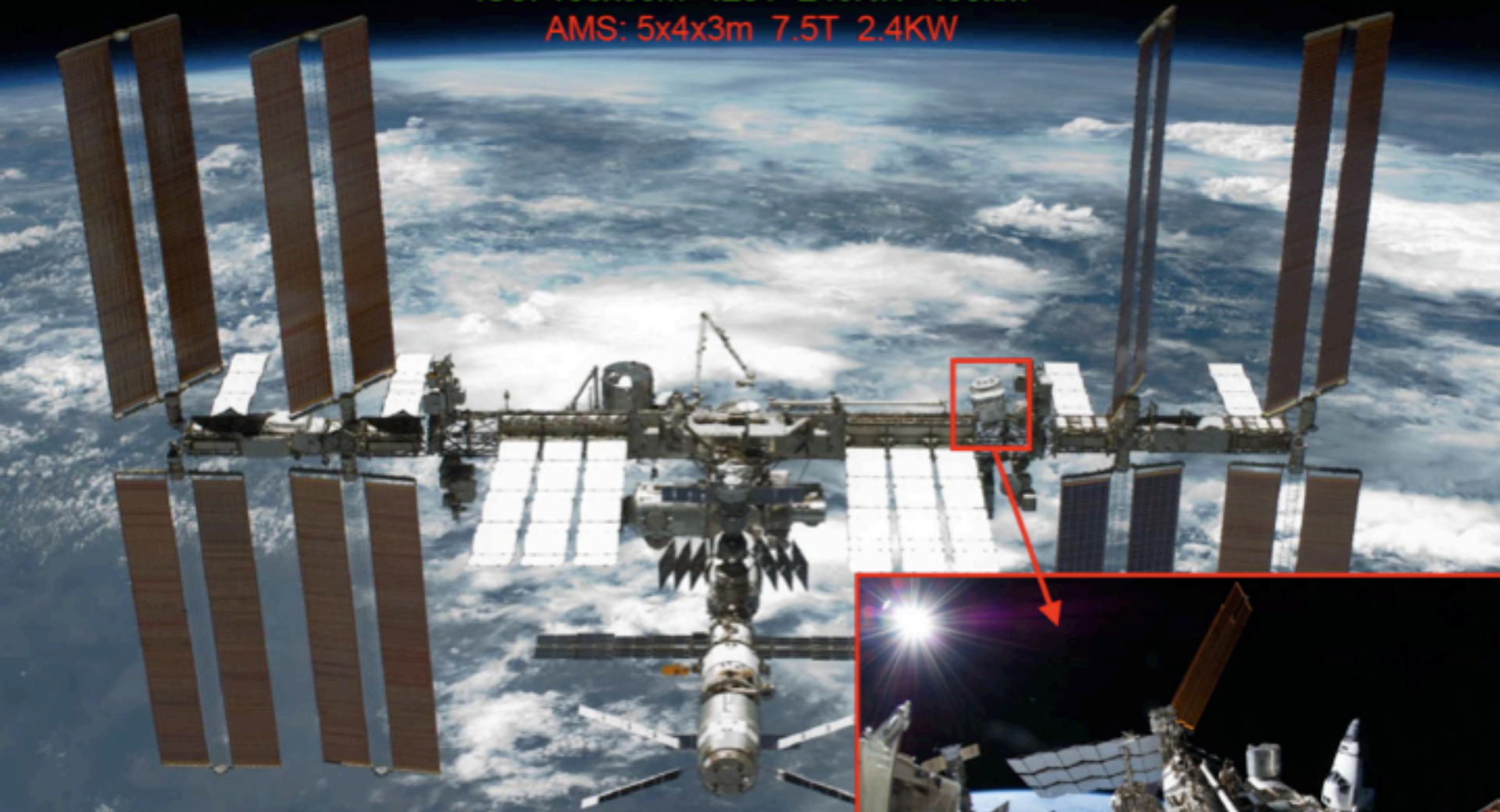


The Alpha Magnetic Spectrometer (AMS)

ISS: 108x80m 420T 240KW 400km

AMS: 5x4x3m 7.5T 2.4KW



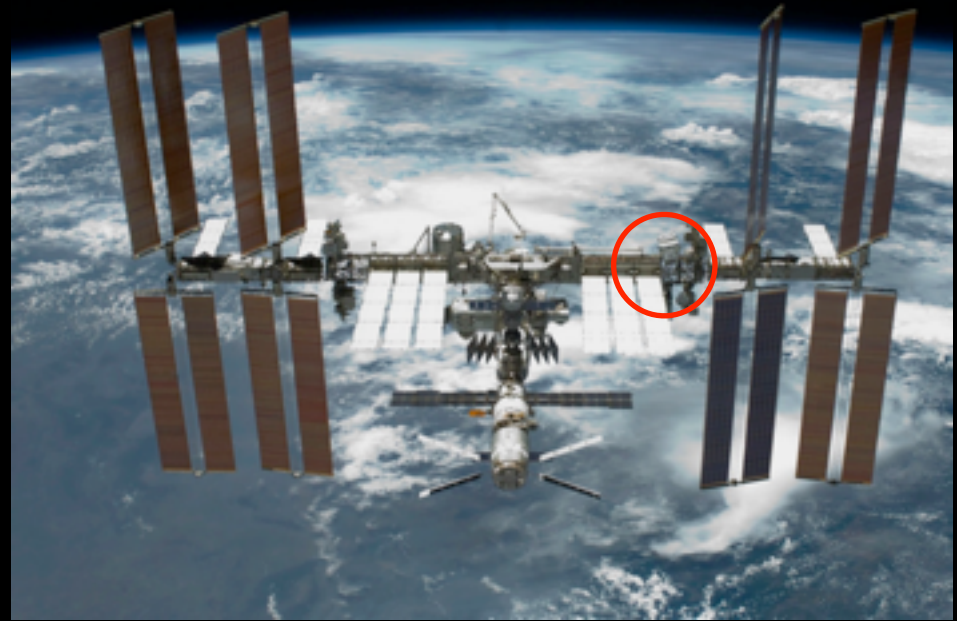
Veronica Bindi

Assistant Professor

Physics and Astronomy Department

University of Hawaii at Manoa

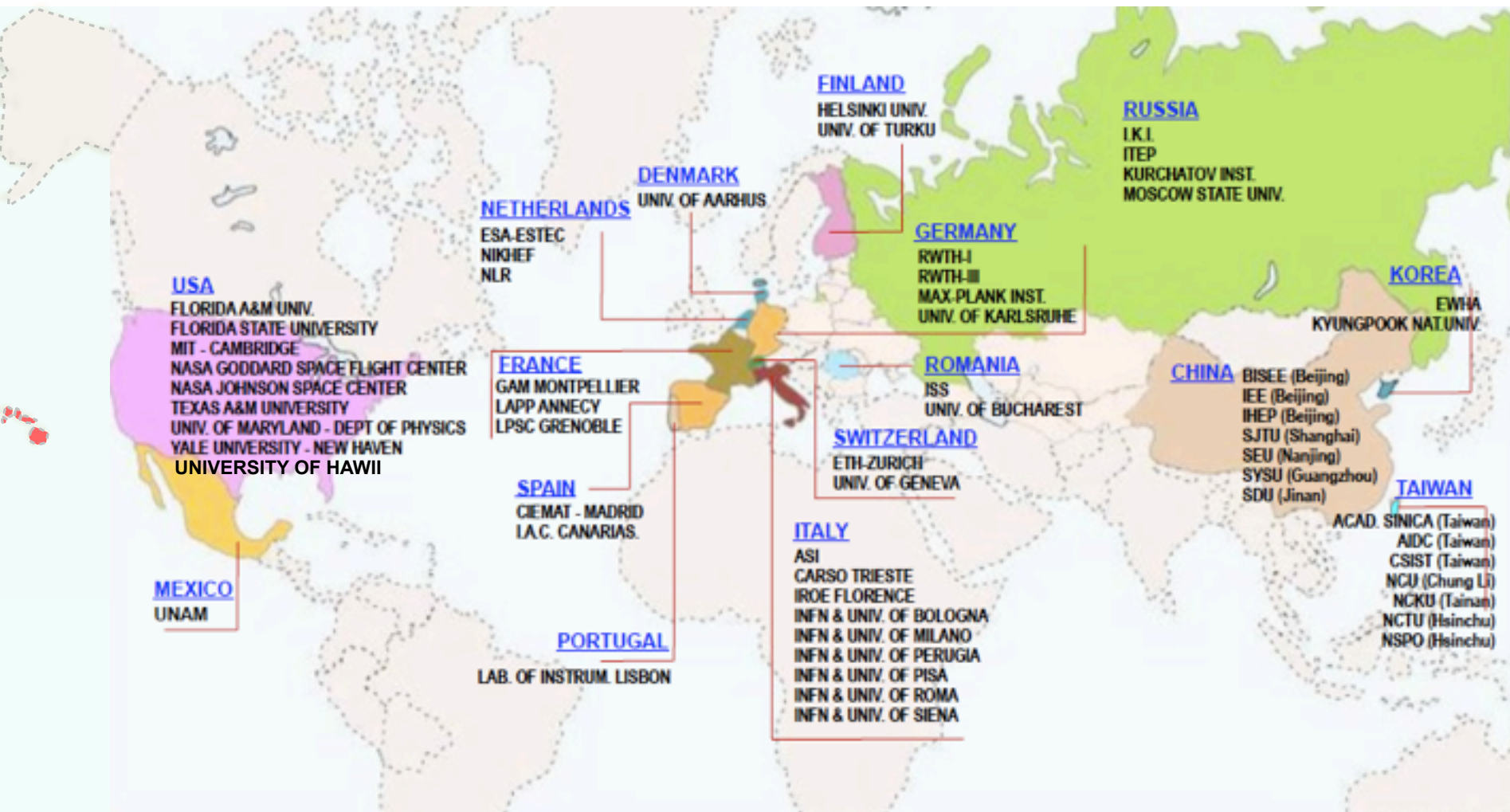
Scientific goals of AMS on the International Space Station



- Direct search of primordial antimatter:
Anti-nuclei: He, C
- Indirect search of Dark Matter:
 e^+ , antiprotons, γ , ... simultaneous observation in several signal channels
- New forms of matter:
strangelets
- Identification of local sources of high energy photons ($\sim \text{TeV}$):
SNR, Pulsars, PBH
- Measuring CR spectra up to the iron— refining propagation models;
- Solar modulation on CR spectra over 11 year solar cycle

AMS is a US DOE led International Collaboration

Spokesperson: Nobel laureate Prof. Dr. S. Ting from MIT
16 Countries, 60 Institutes and 600 Physicists



**The detectors and the electronics were built all around the world
and assembled in CERN, Switzerland**



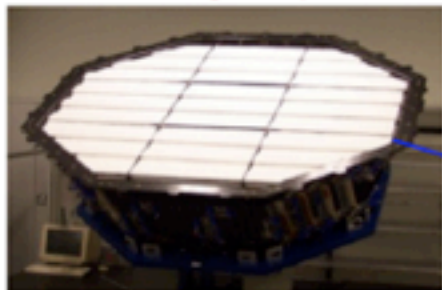
15ft x 12ft x 9ft
5m x 4m x 3m
7.5 tons

**AMS is designed with the same precisions as the CERN LHC detectors
the technology has been miniaturized and upgraded to work on space**

AMS consists of 5 sub-detectors which provide redundant information for particle identification

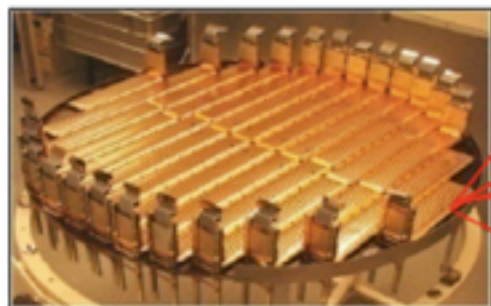
TRD

Identify e^+ , e^-



Silicon Tracker

Z, P

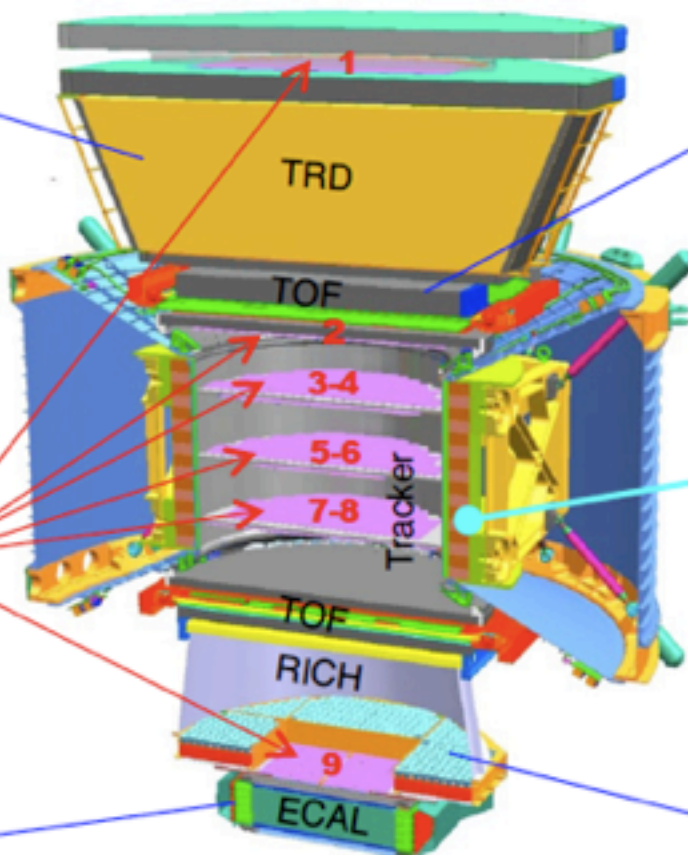


ECAL

E of e^+ , e^- , γ



Particles and nuclei are defined by their charge (Z) and energy ($E \sim P$)



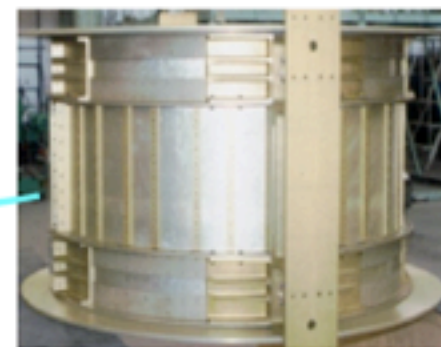
TOF

Z, E



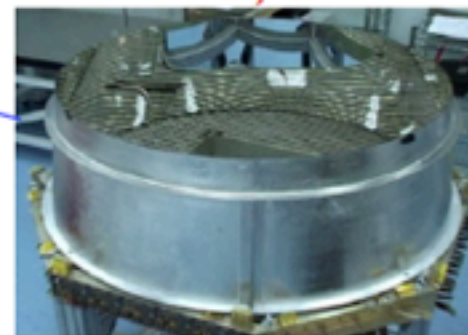
Magnet

$\pm Z$



RICH

Z, E



Z, P are measured independently by the Tracker, RICH, TOF and ECAL

A TeV precision, multipurpose spectrometer

Magnet

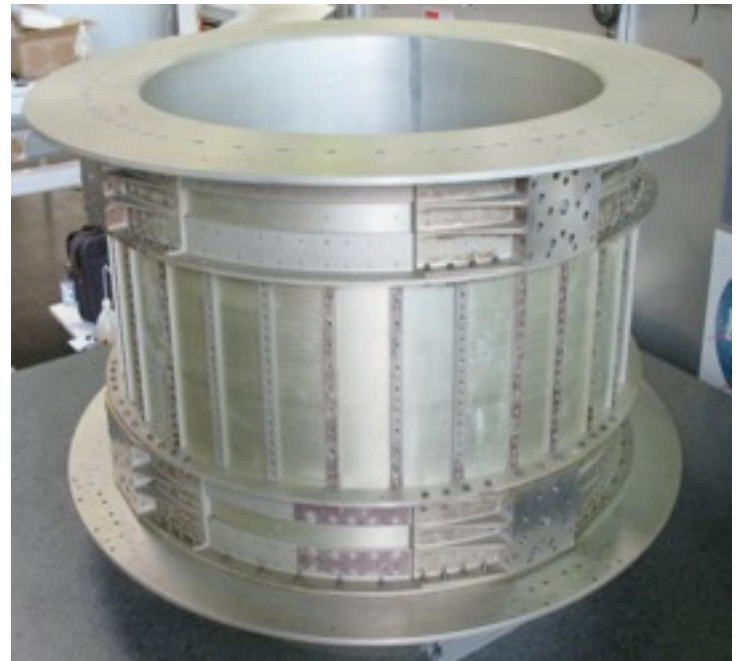
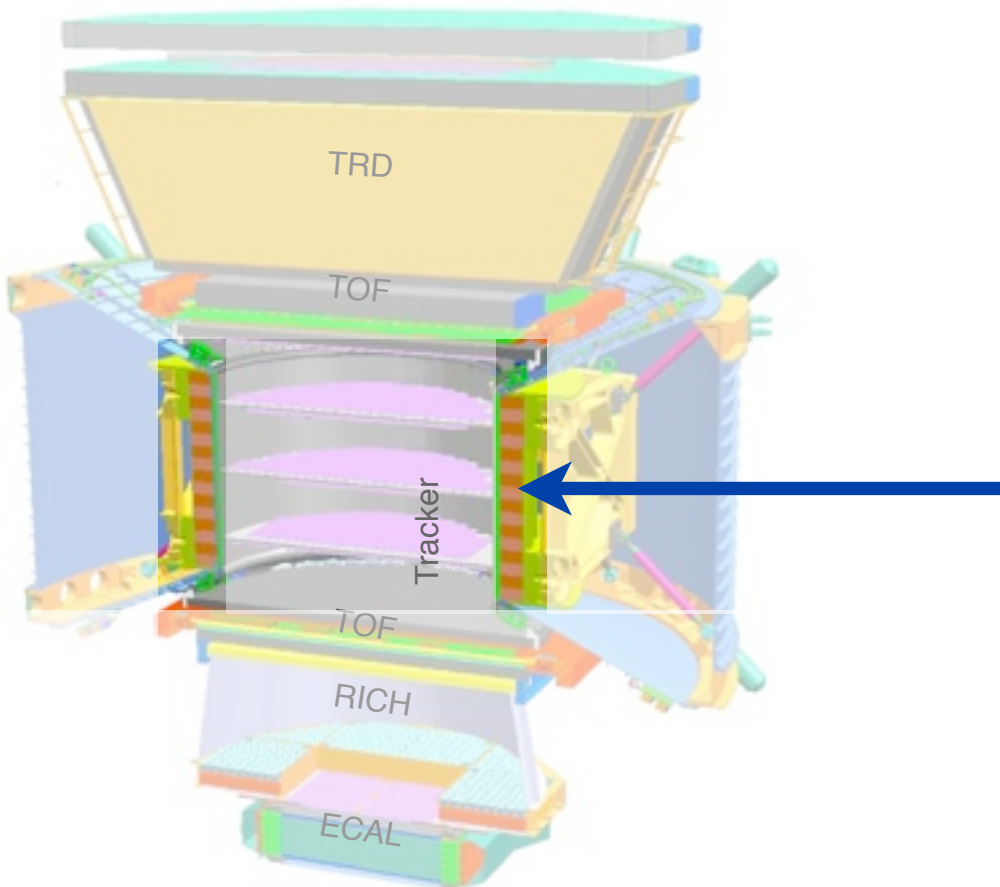
$\pm Z$

same of AMS-01

$B \sim 0.14$ Tesla uniform along x

no field leak outside

no torque outside



A TeV precision, multipurpose spectrometer

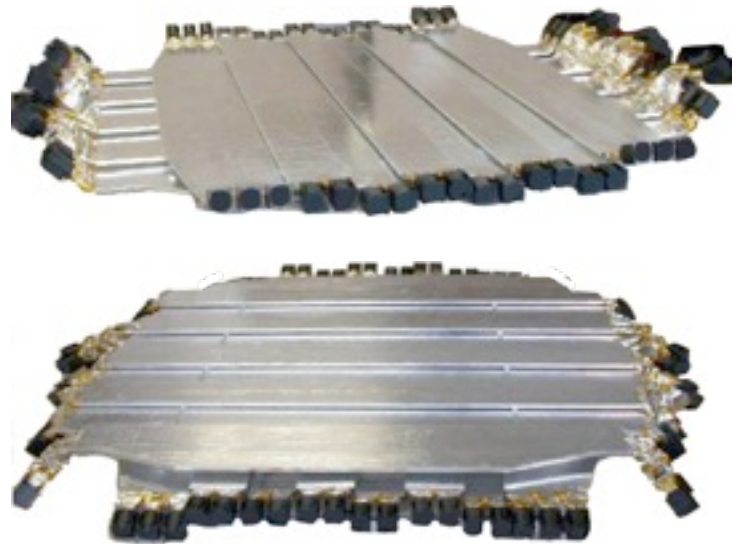
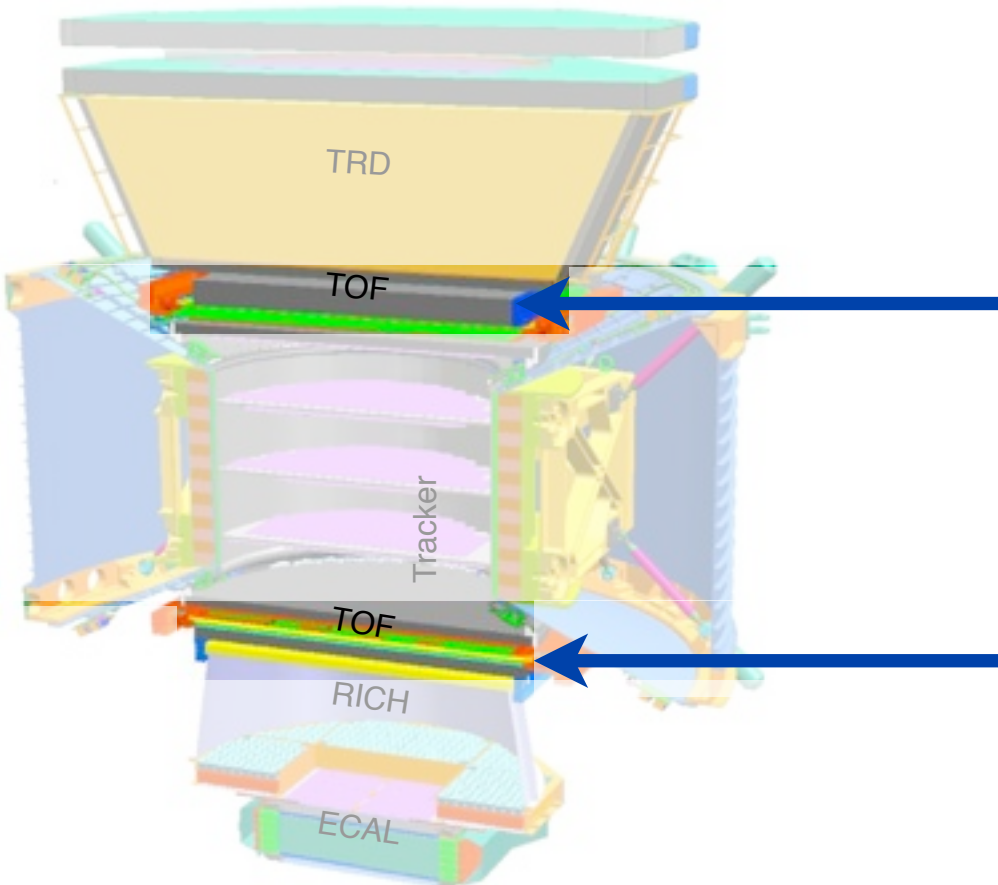
TOF

Z,E

4 layer scintillators

acceptance $0.4 \text{ m}^2 \text{ sr}$

144 photomultipliers



A TeV precision, multipurpose spectrometer

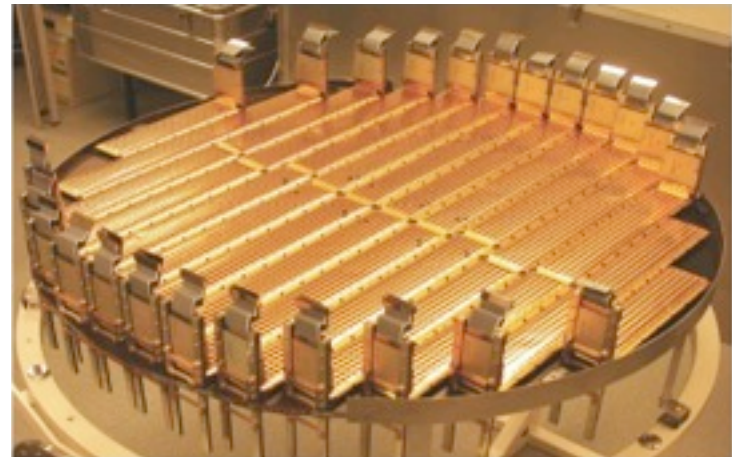
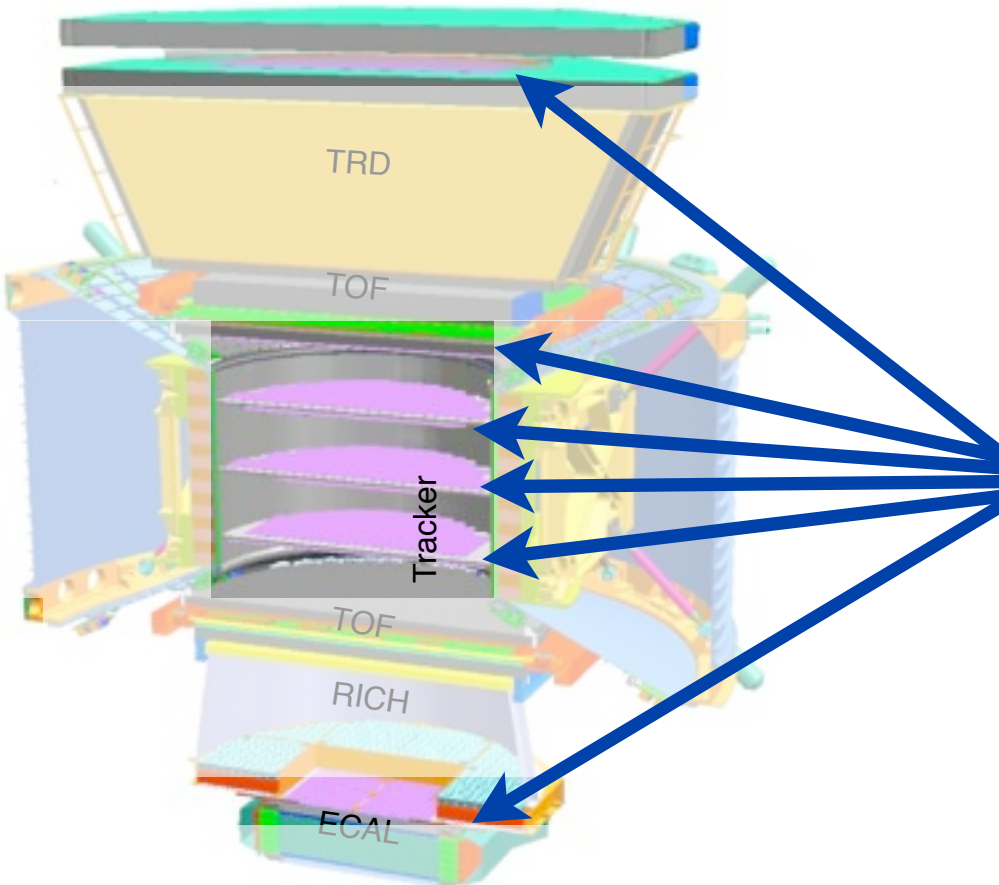
Silicon Tracker

Z, P

9 layers silicon

192 readout channels

resolution of $10\ \mu\text{m}$ in bending direction



A TeV precision, multipurpose spectrometer

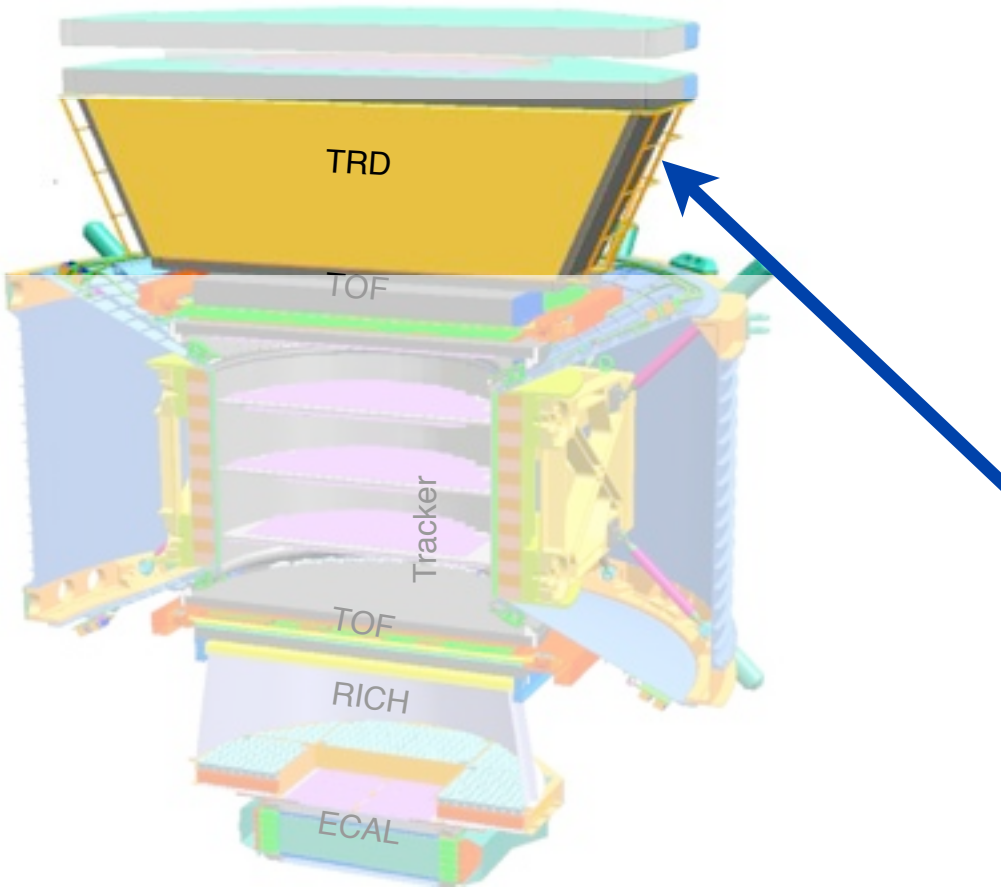
TRD

Identify e^+ , e^-

20 layers of radiators
and 6mm straw tubes

Xe/CO₂ (80%/20%) gas

5248 channels

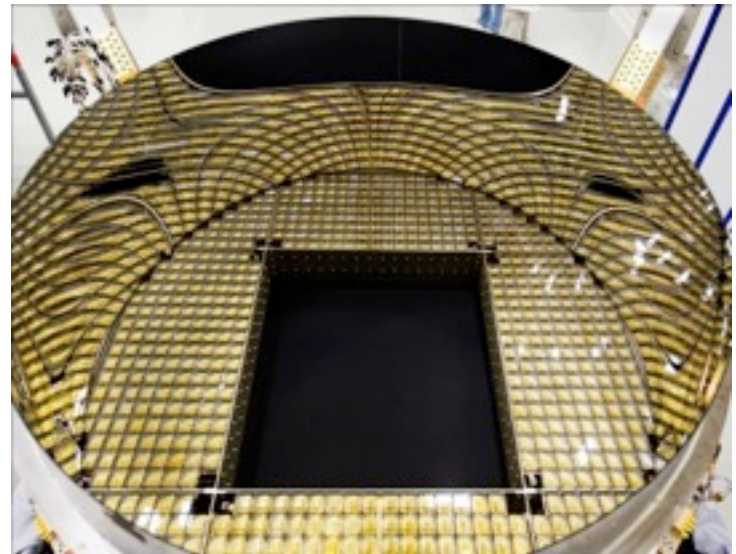
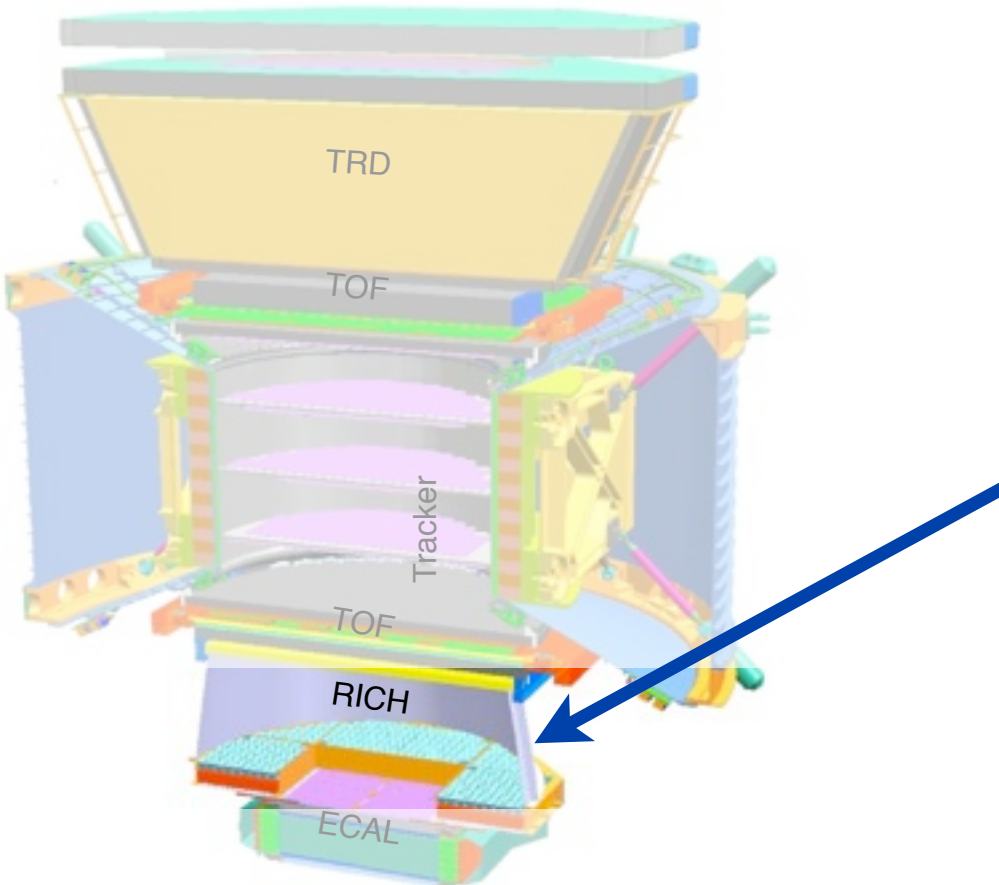


A TeV precision, multipurpose spectrometer

RICH

Z,E

Areogel and NaF radiators
conical reflector
10880 photosensors



A TeV precision, multipurpose spectrometer

ECAL

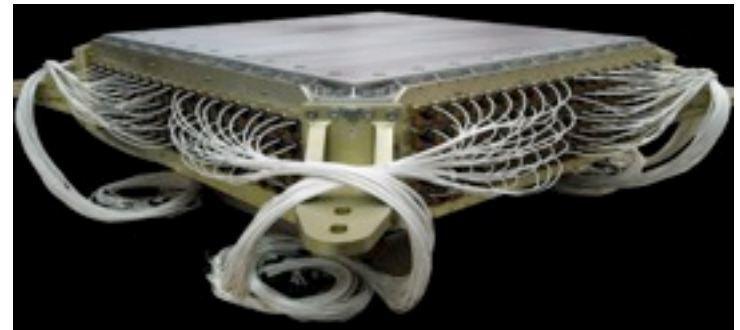
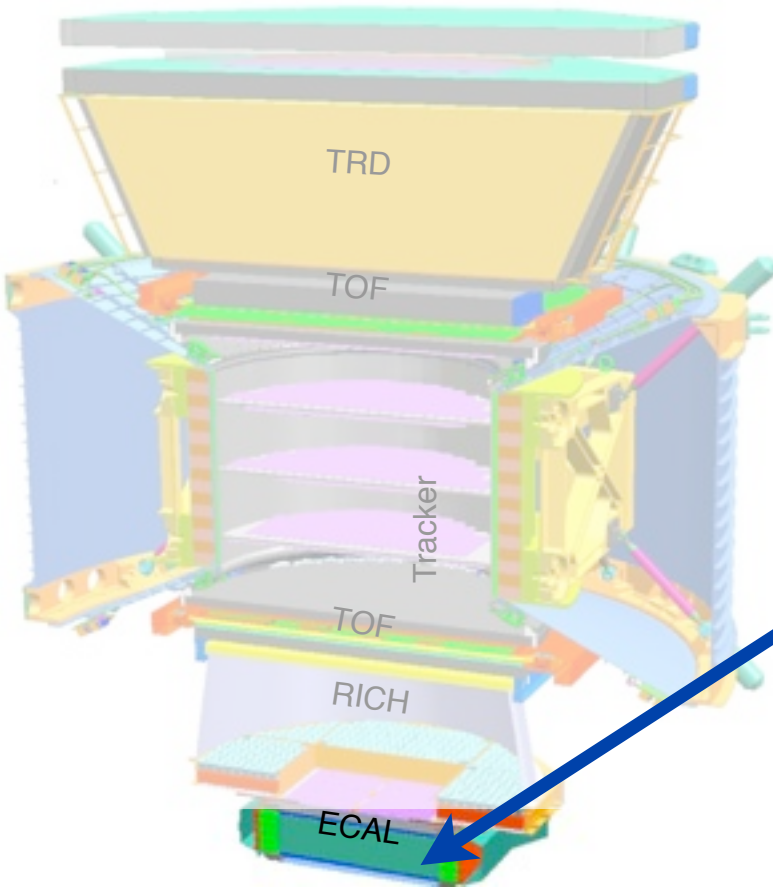
E of e^+ , e^- , γ

3D sampling calorimeter

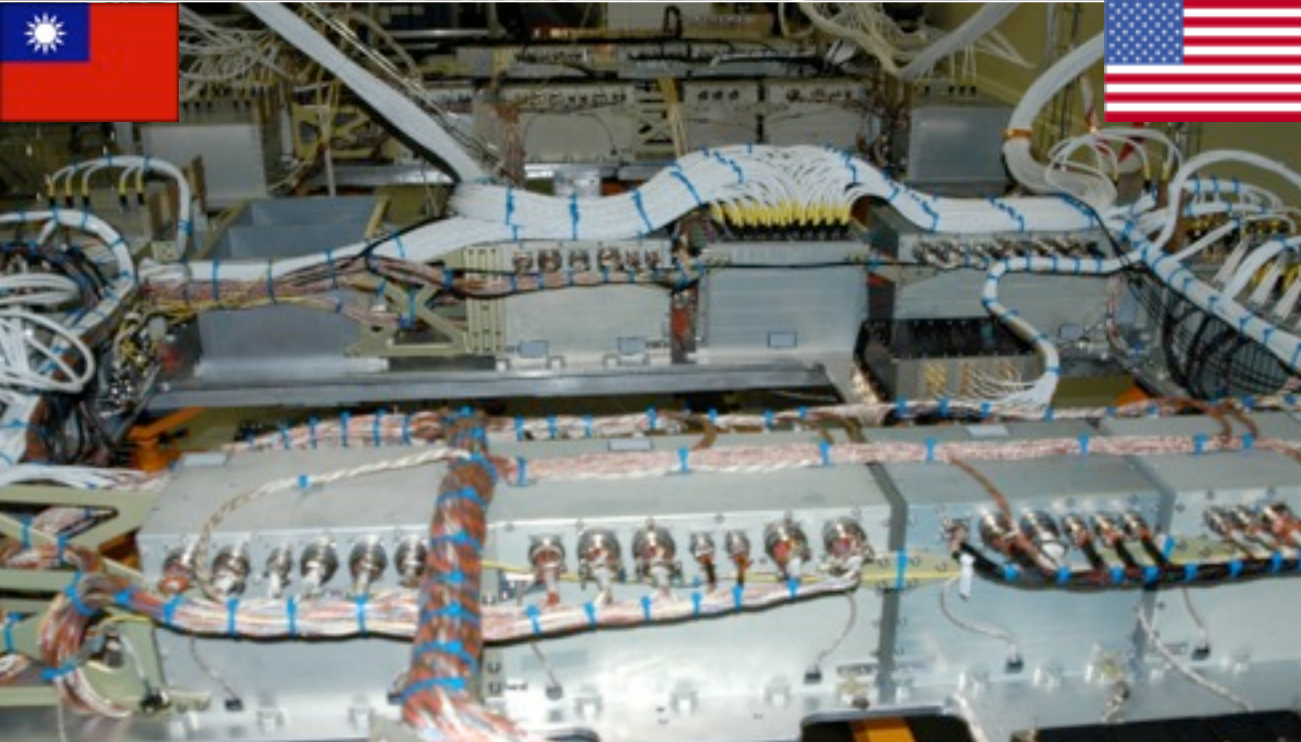
17 X_0

50000 fibers 1mm thick

e/p separation 10^4



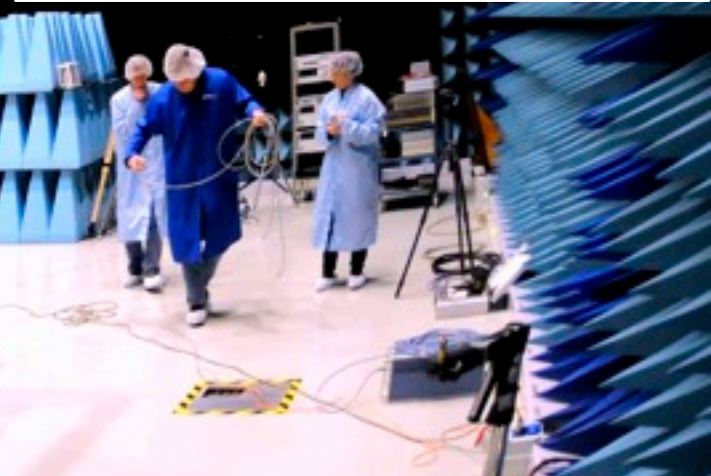
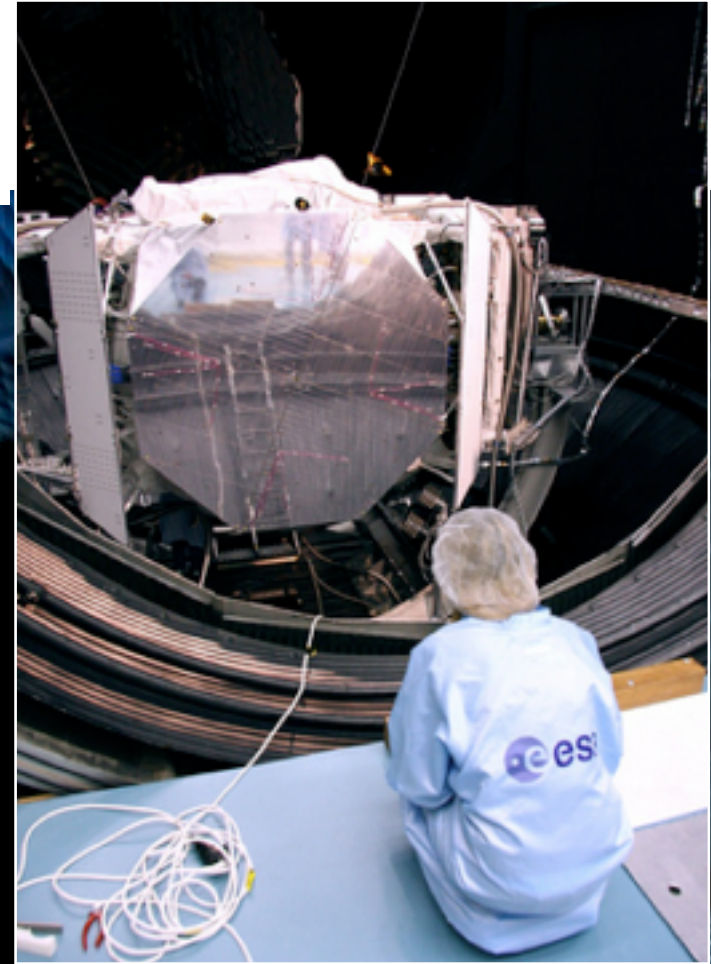
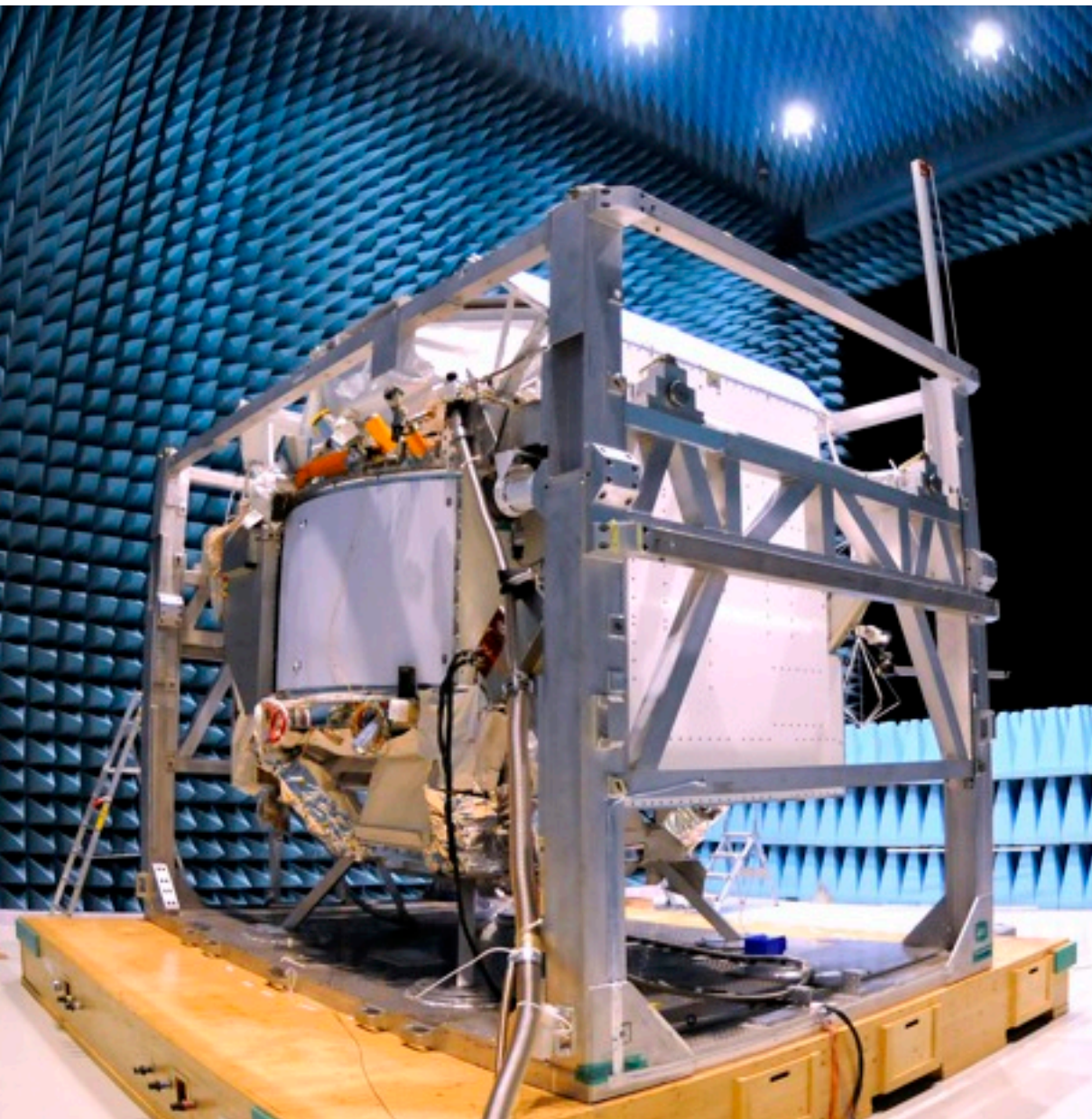
AMS electronics on ISS



**650 computers,
300,000 channels.
A ten year effort by
75 engineers
400% redundancy**

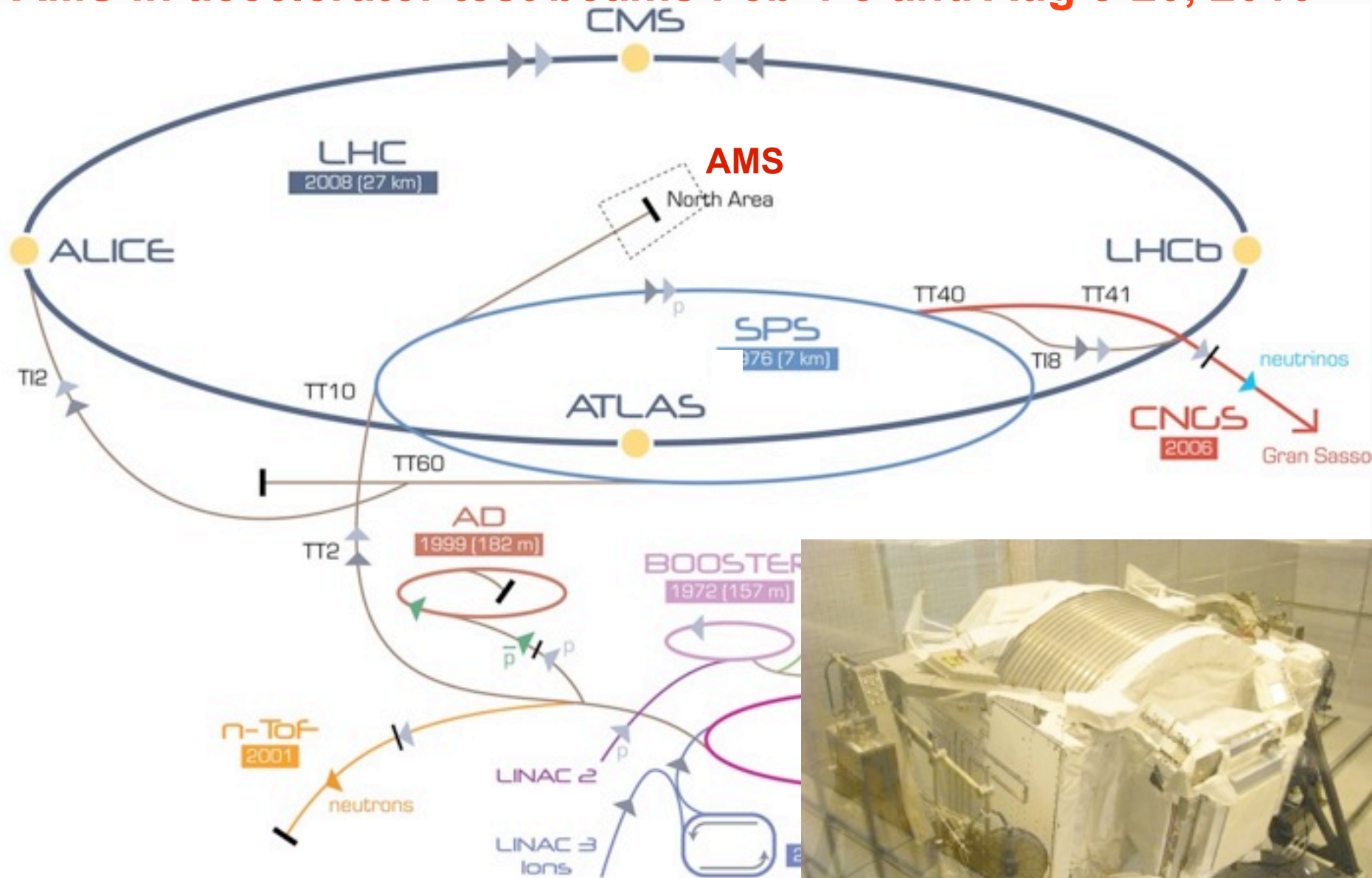
AMS in the ESA

EMI and TVT tests

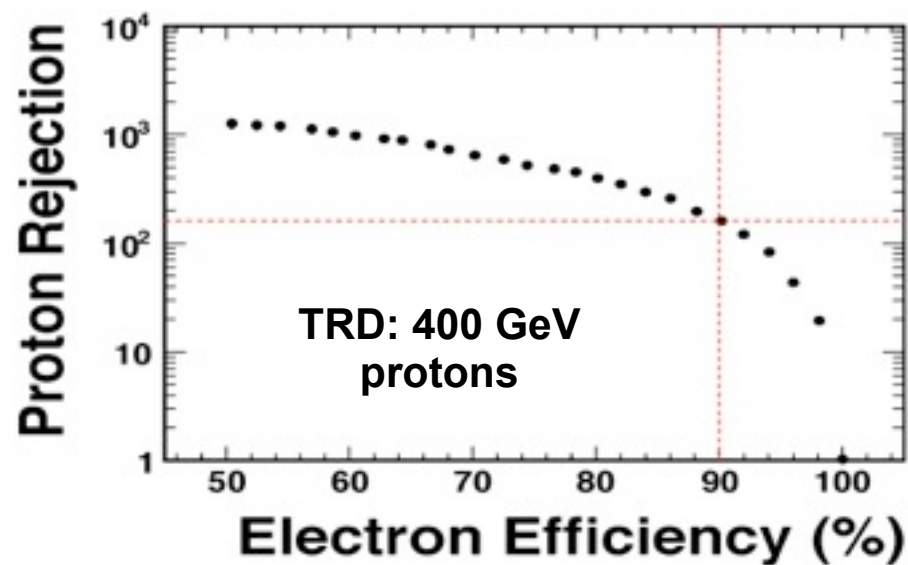
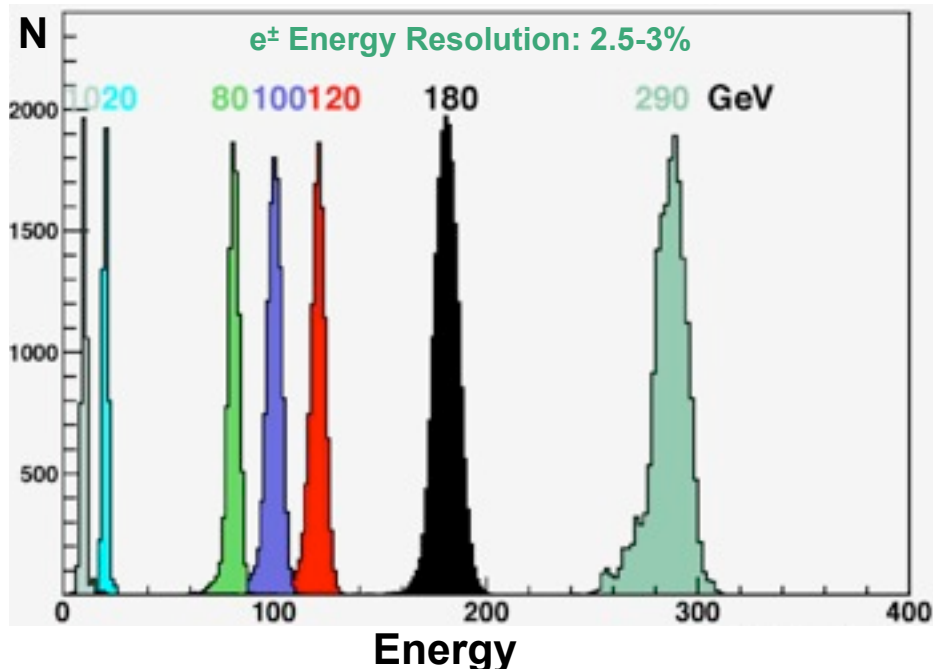
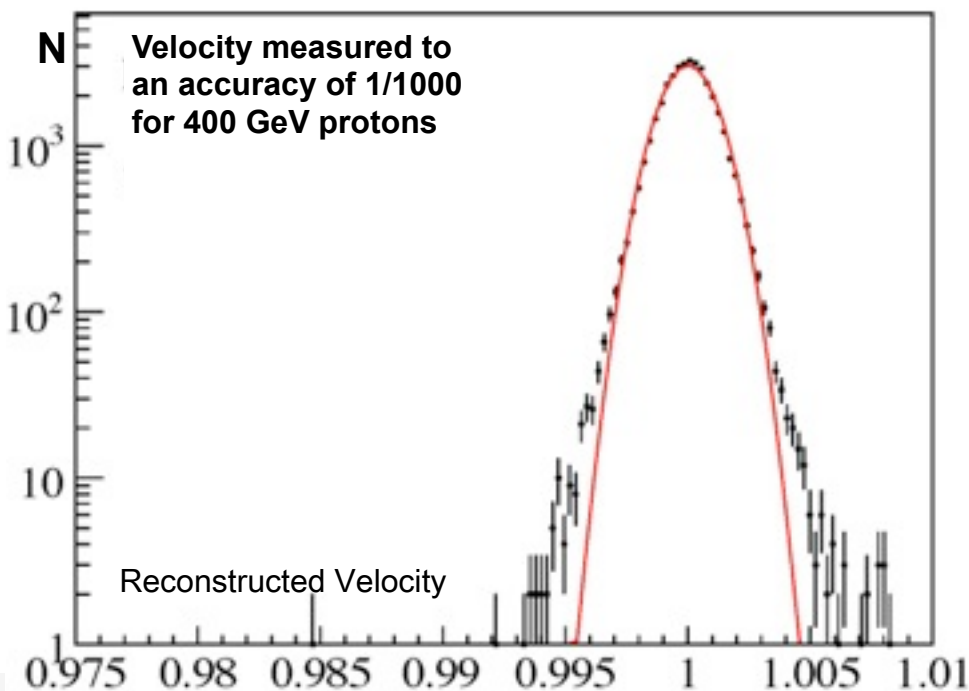
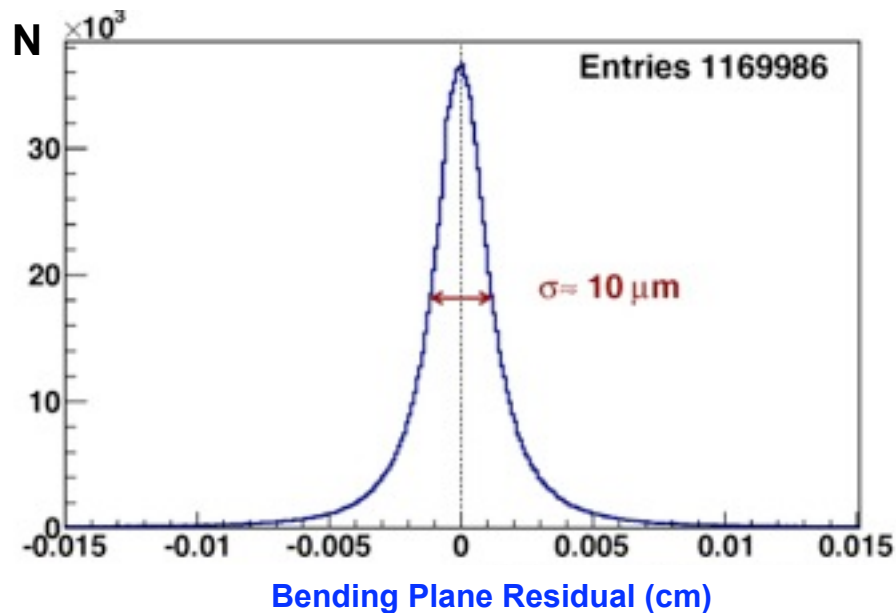


Tests at CERN

AMS in accelerator test beams Feb 4-8 and Aug 8-20, 2010



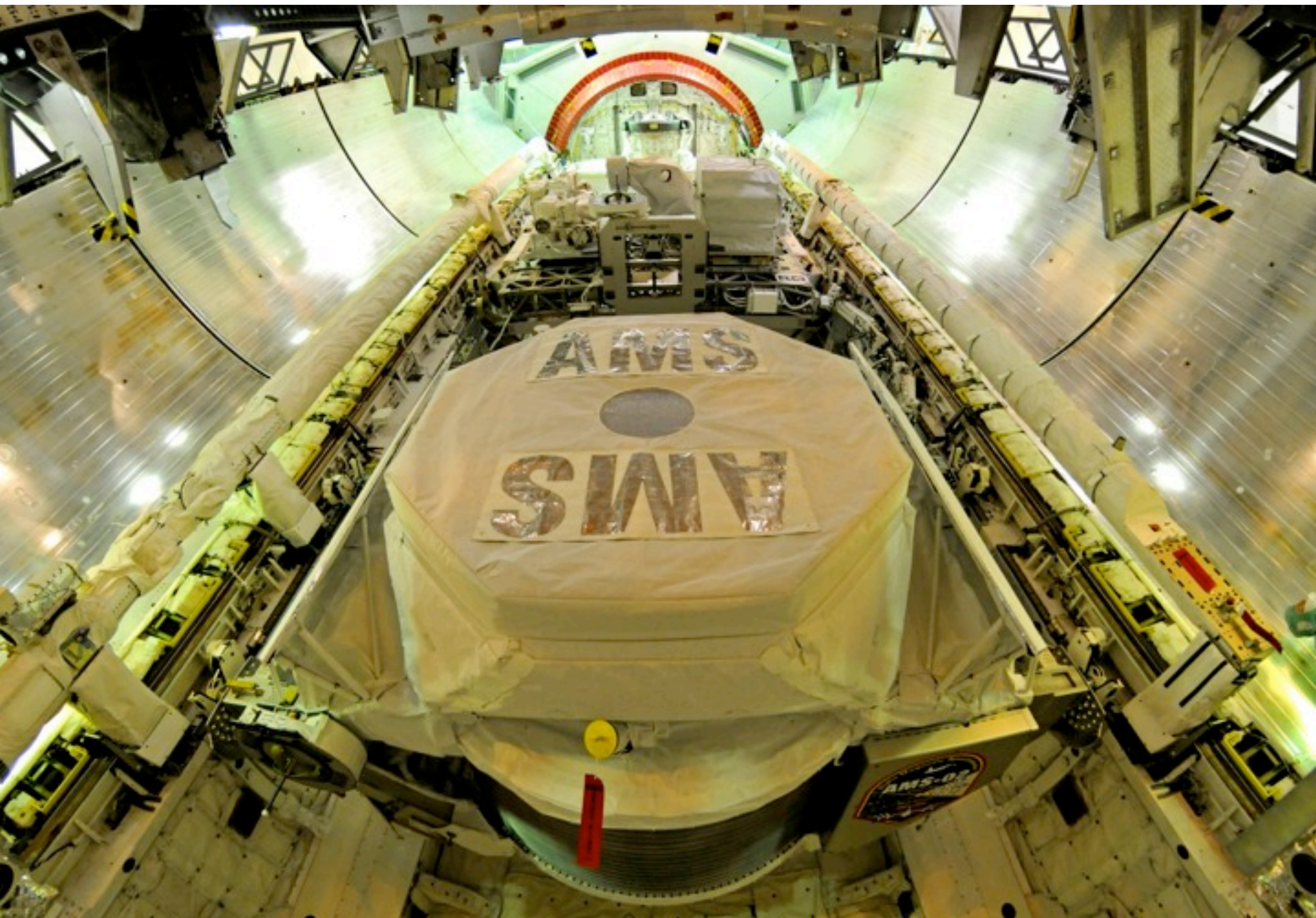
Test Beam Results



A US-AirForce C-5 Galaxy has been used for transport
from Geneva to KSC - August 25th 2010



AMS in Endeavour's Payload Bay





**Launch of the Space Shuttle Endeavour
May 16, 2011 @ 08:56 AM**

Temperatures and slow-control data monitoring started 2.5 h after the launch

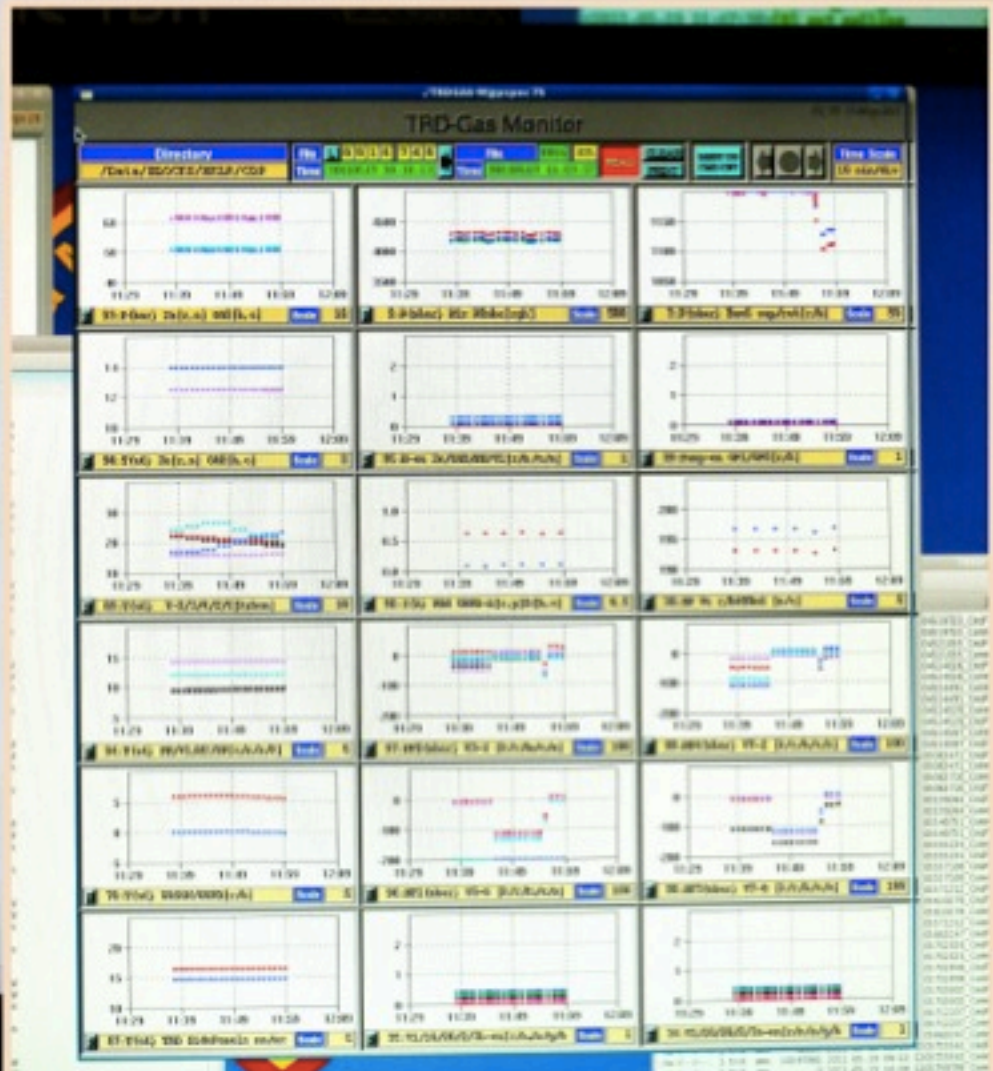
0-A
0-B

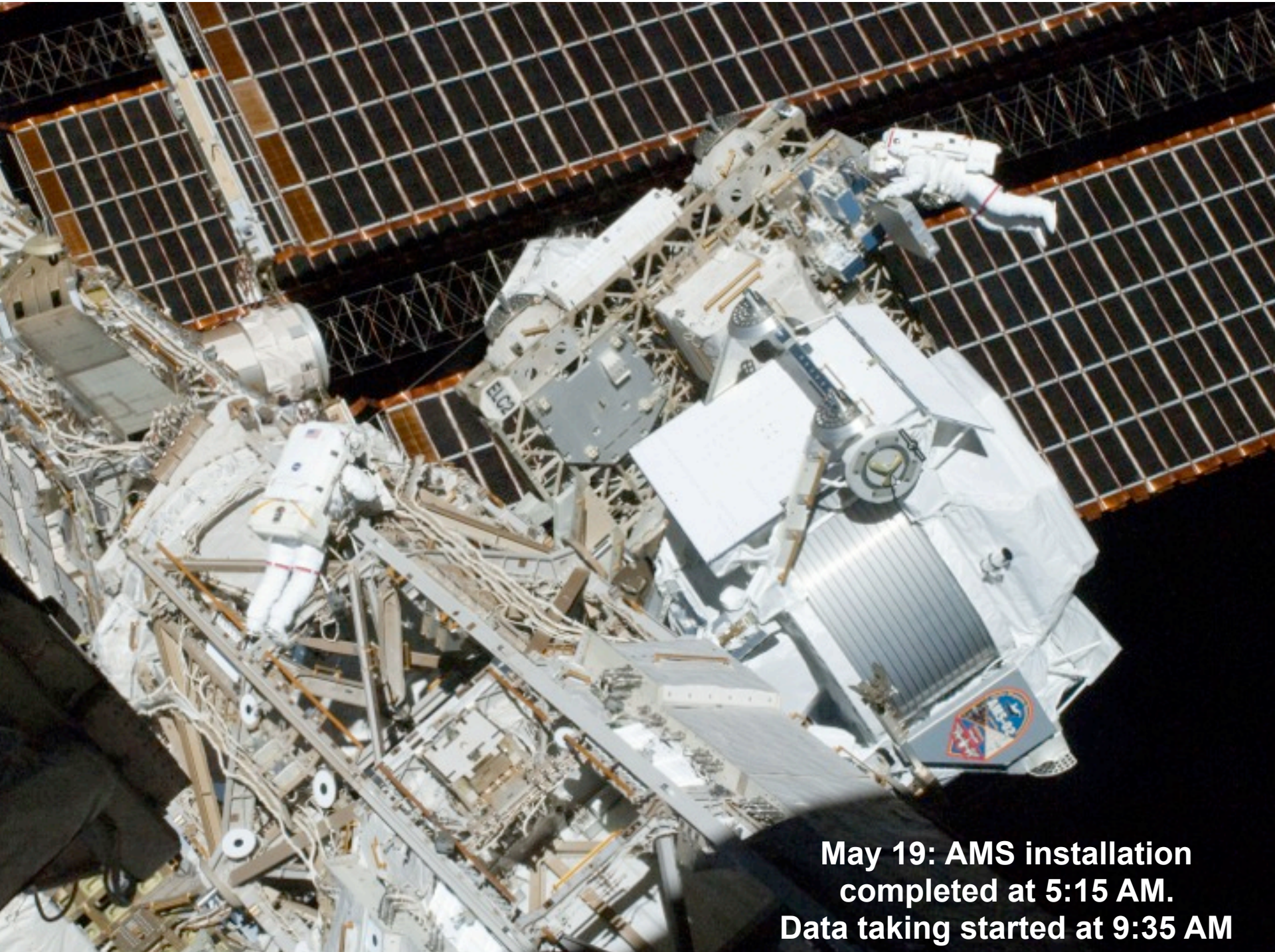
CHECK

MPD @ TMD2	13.875 °C
M	13.8875 °C
GPS	12.5 °C
TT	14.0625 °C
TTCBP	14.0425 °C
TTCBS	14.0625 °C
UGPD	13.75 °C
UG	12.5 °C
CCEB Signal Side	13.625 °C
CCEB Power Side	13.6425 °C
UPD0	13.6875 °C
U0	12.3875 °C
UPD1	13.8125 °C
U1	13.625 °C
SPD0 @ TSPD1	13.6875 °C
S0	13.9375 °C
SHV0	13.8125 °C
SPD1 @ TSPD3	13.5625 °C
S1	12.8 °C
SHV1	13.6425 °C
SPD2 @ TSPD4	13.625 °C
S2	14.0425 °C
SHV2	13.8125 °C
SPD3 @ TSPD6	13.875 °C
S3	14.3125 °C

Unit now or 14.05.2011

Everything OK



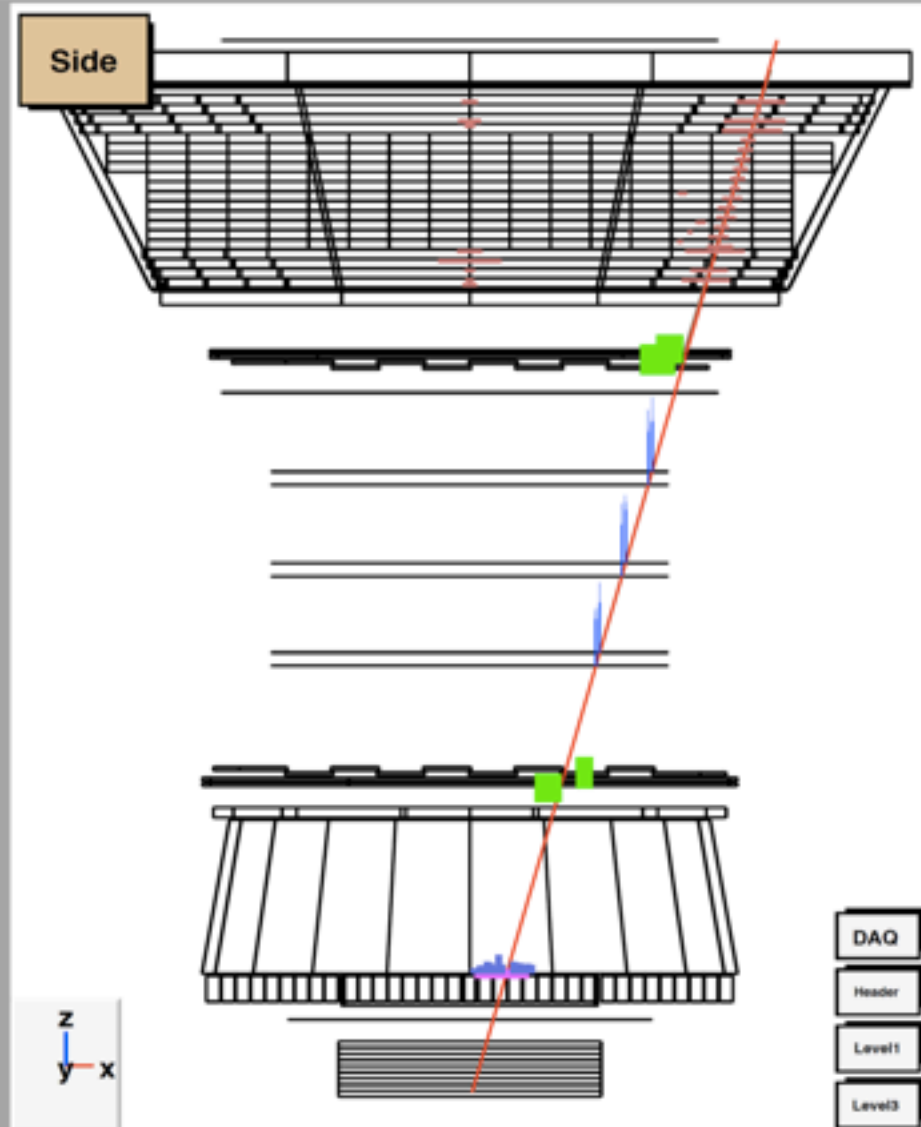
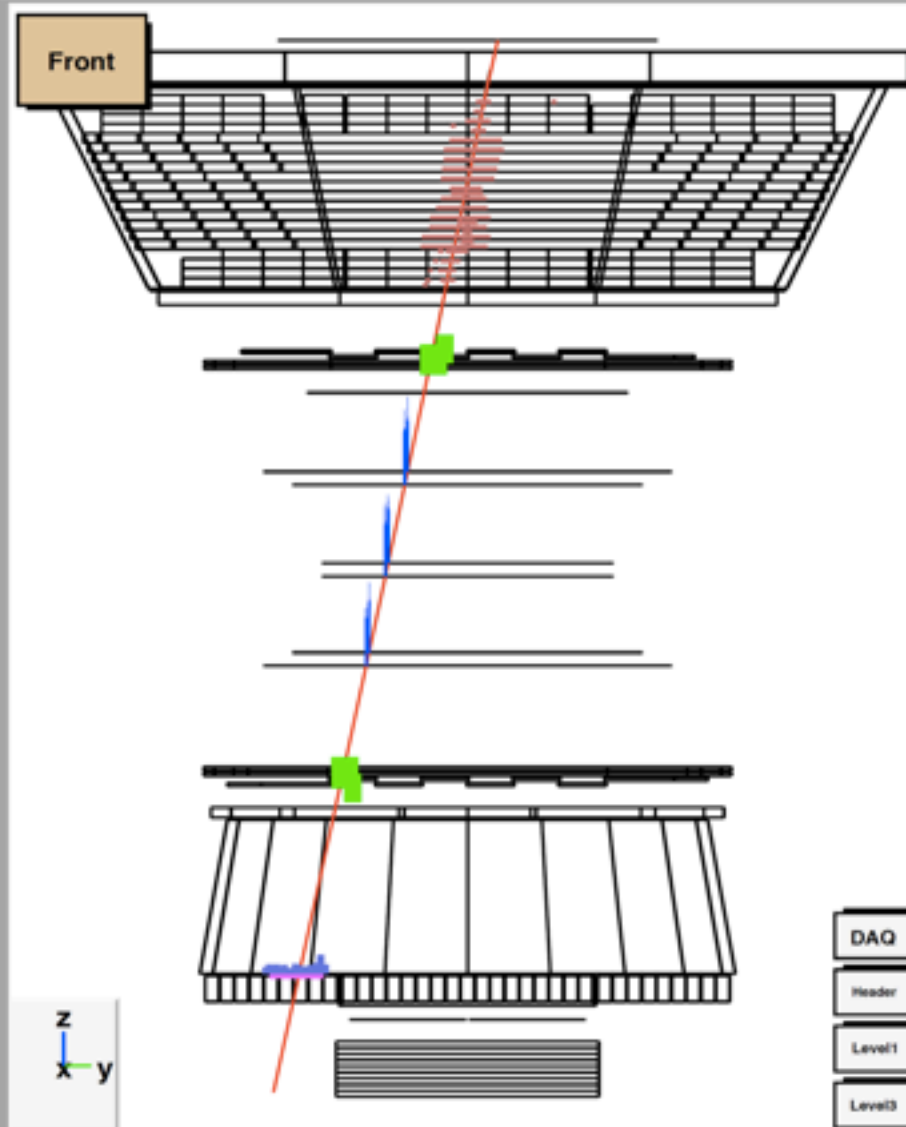


**May 19: AMS installation
completed at 5:15 AM.
Data taking started at 9:35 AM**

42 GeV/c Carbon

AMS Event Display

Run 1305815610/ 224169 Thu May 19 16:42:29 2011



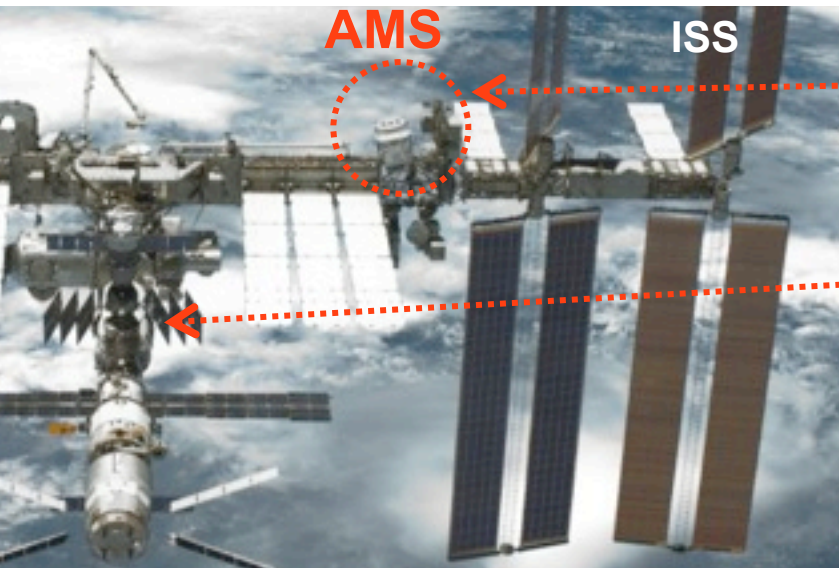
AMS installed on ISS @ 16:15 Geneva time

AMS start data taking @ 16:35 Geneva time



Johnson Space Center, May 19th 2011

AMS Operations

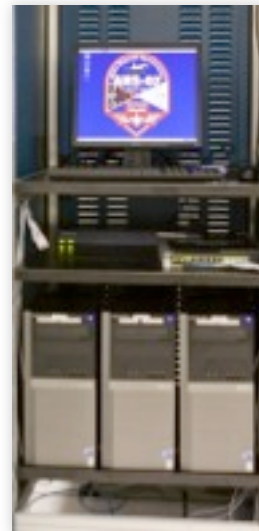


**Ku-Band
High Rate (down):**
Events: 10Mbit/s

Flight Operations

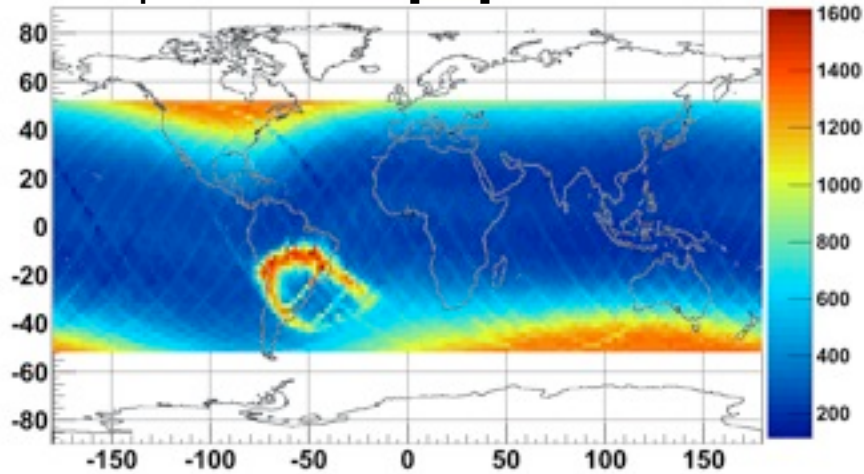
Ground Operations

**S-Band
Low Rate (up & down):**
Commanding: 1 Kbit/s
Monitoring: 30 Kbit/s

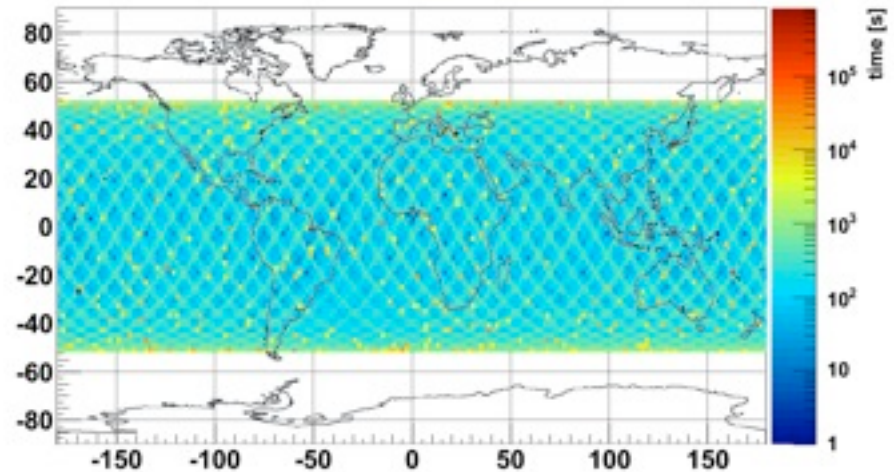


Orbital DAQ parameters

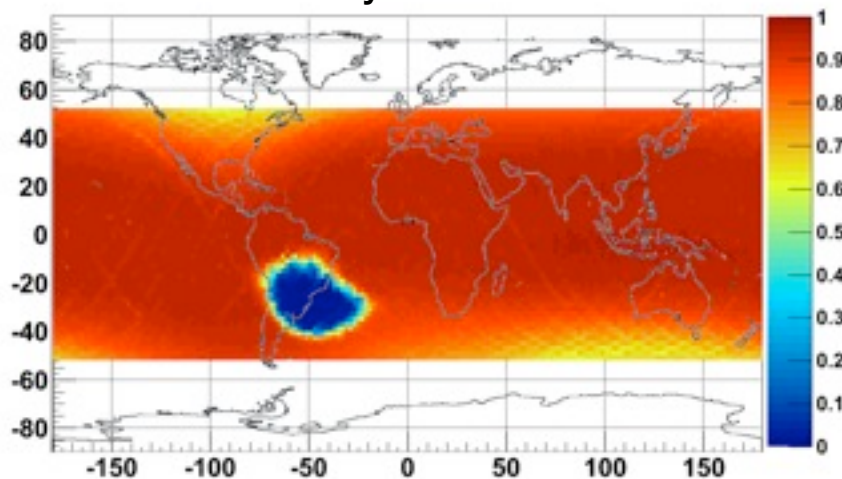
Acquisition rate [Hz]



Time at location [s]



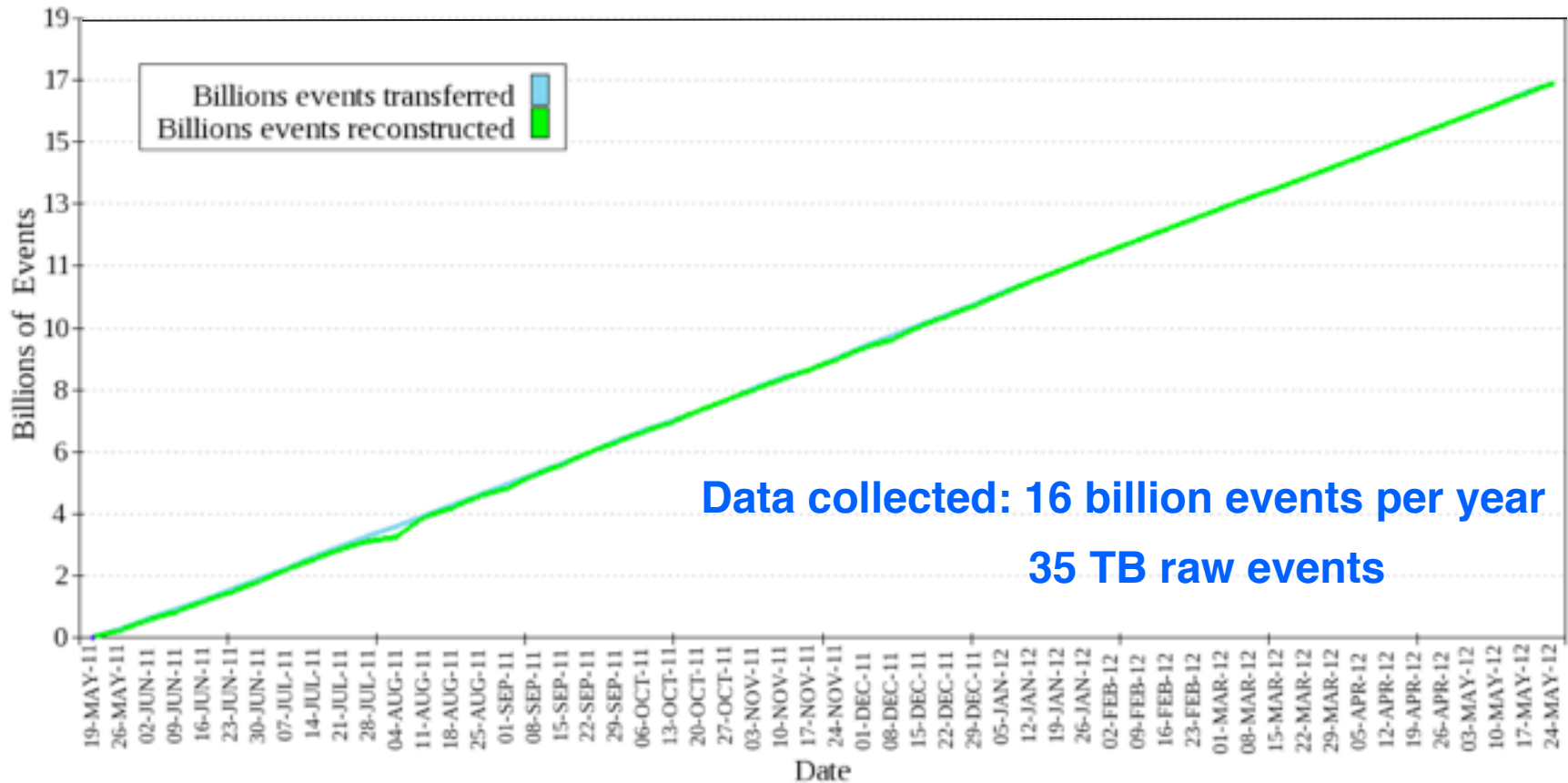
DAQ efficiency



Particle rates vary from 200 to 2000 Hz per orbit

Average DAQ efficiency 85%
Average DAQ rate ~700Hz

Data taking continuously active



In 10 - 20 years AMS will collect from 160-320 billion events.

This will provide unprecedented sensitivity and statistic.

1.03 TeV electron

AMS Event Display

Run/Event 1315754945 / 173049 GMT Time 2011-254.15:31:15

front view

side view

TRD:
identifies
electron

Tracker and Magnet:
measure momentum

RICH
charge of
electron

ECAL:
identifies electron and measures
its energy

DAQ

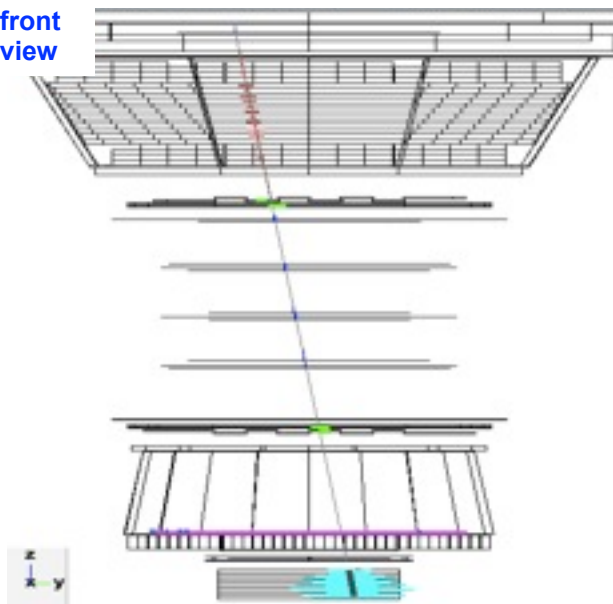
Header

Level1

Level3

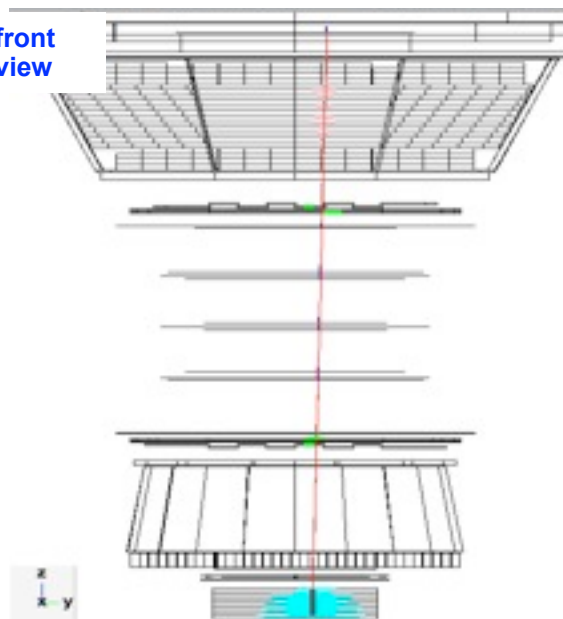
205 GeV Positron

front
view



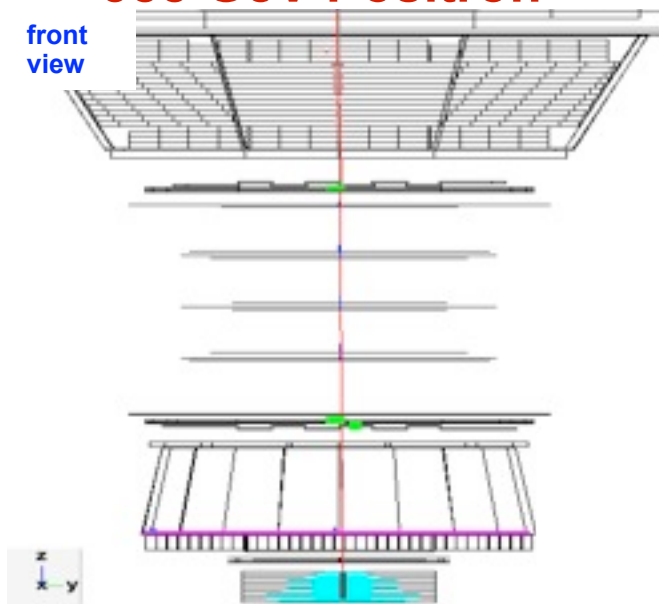
369 GeV Positron

front
view



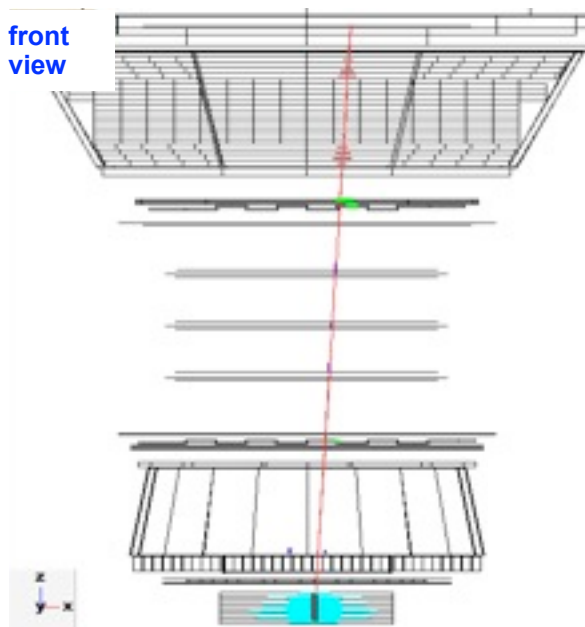
388 GeV Positron

front
view



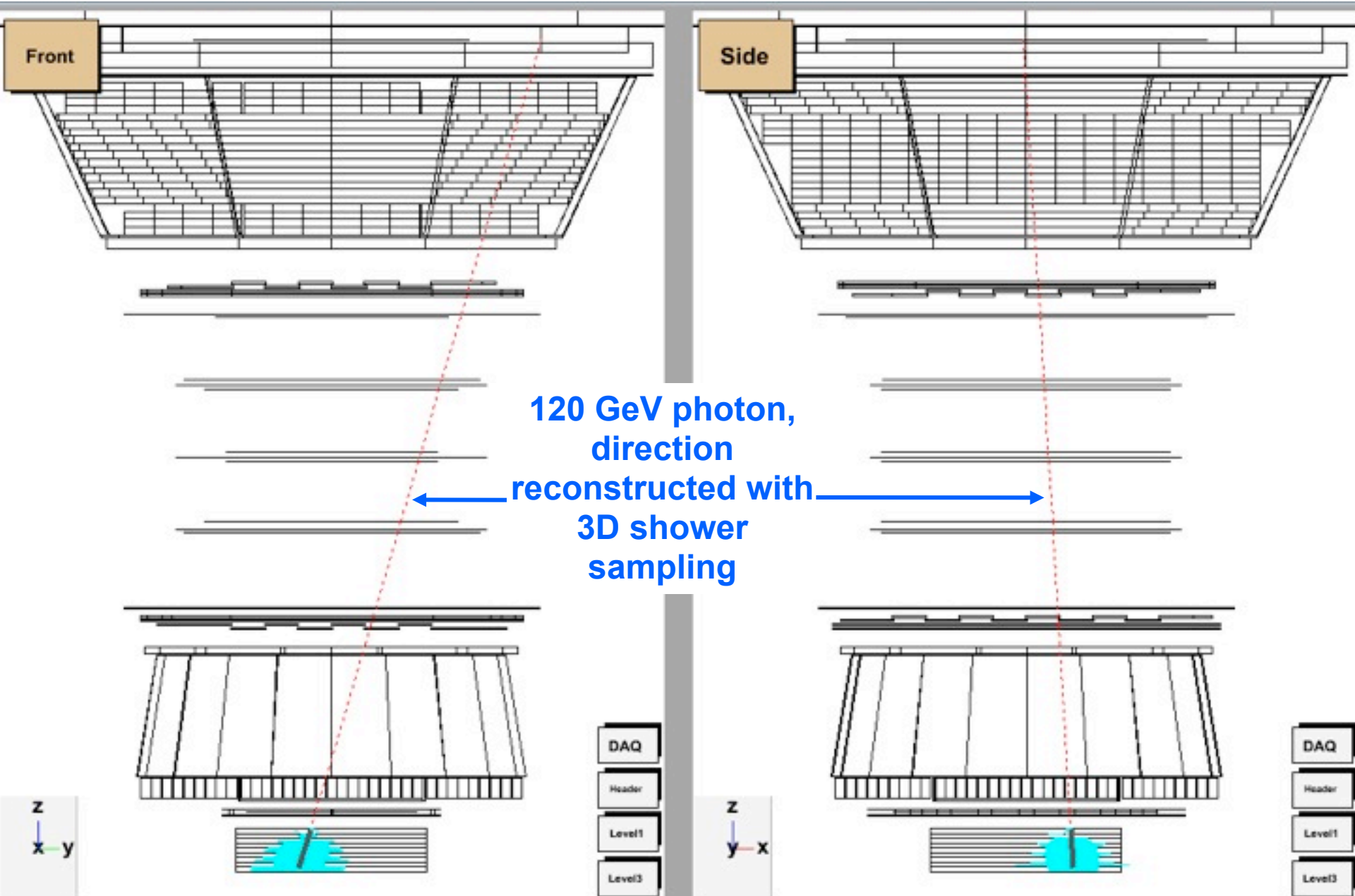
424 GeV Positron

front
view

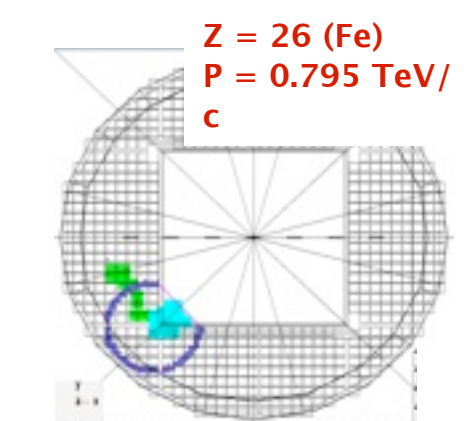
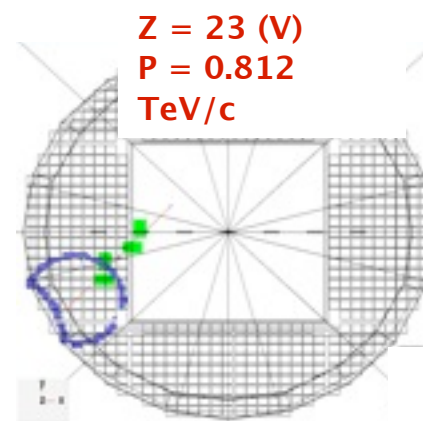
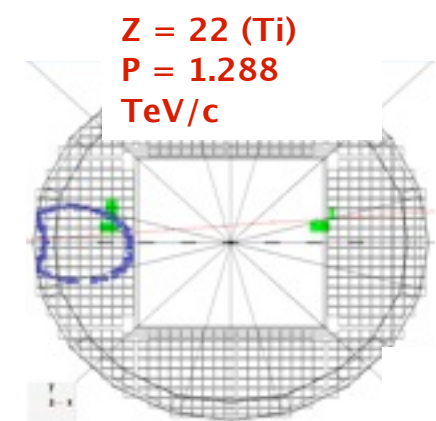
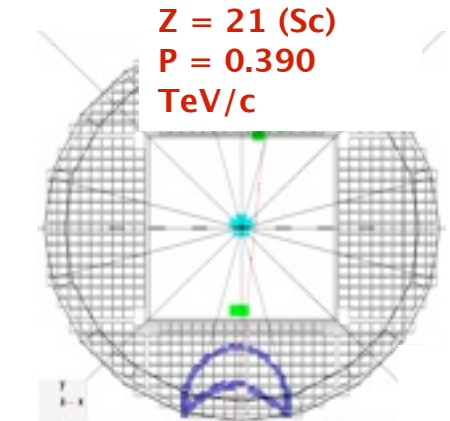
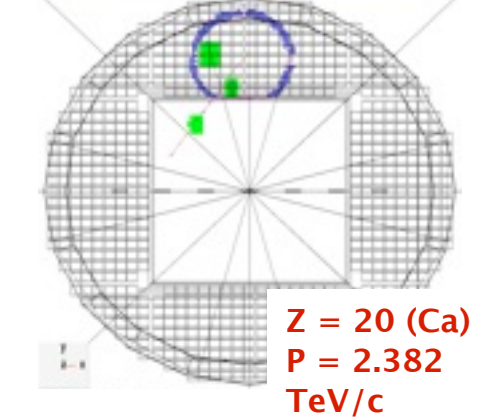
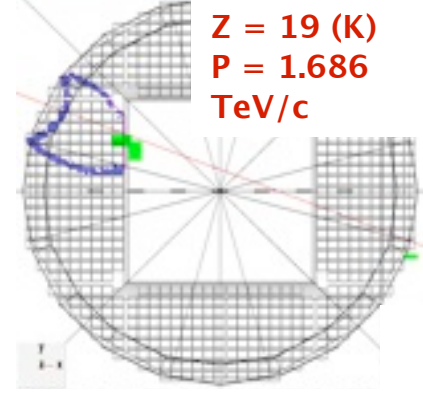
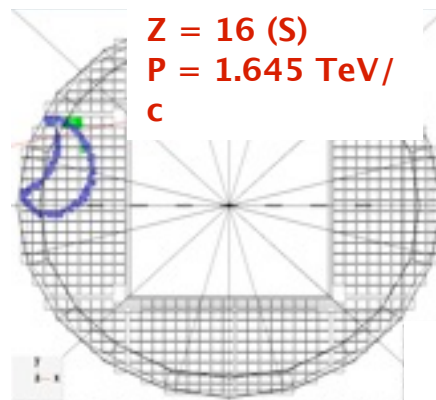
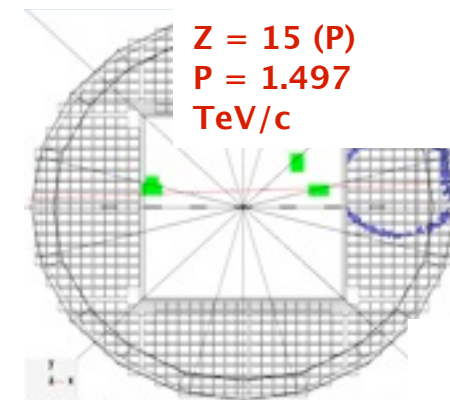
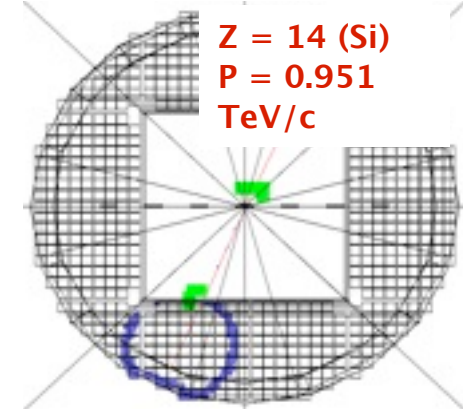
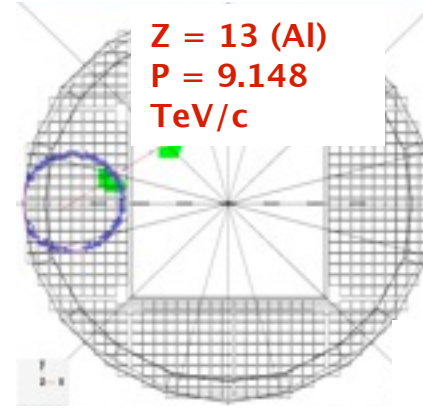
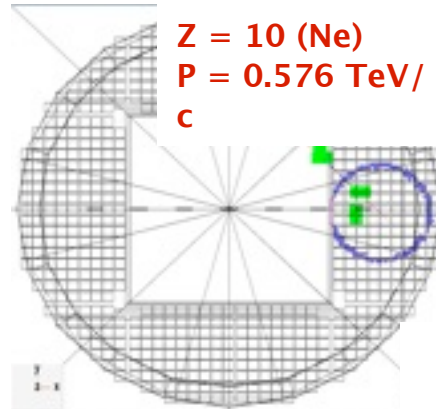
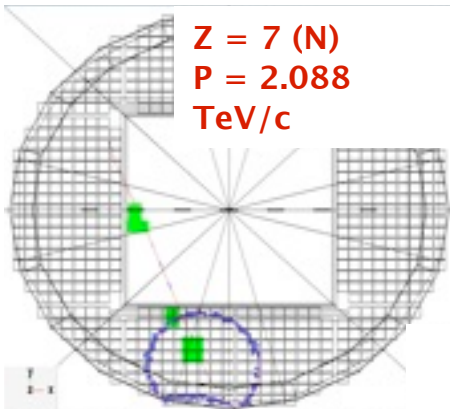


120 GeV Photon

17 X_0 , 3D ECAL, measure γ to 1 TeV, time resolution of 1 μsec



Nuclei in the TeV range

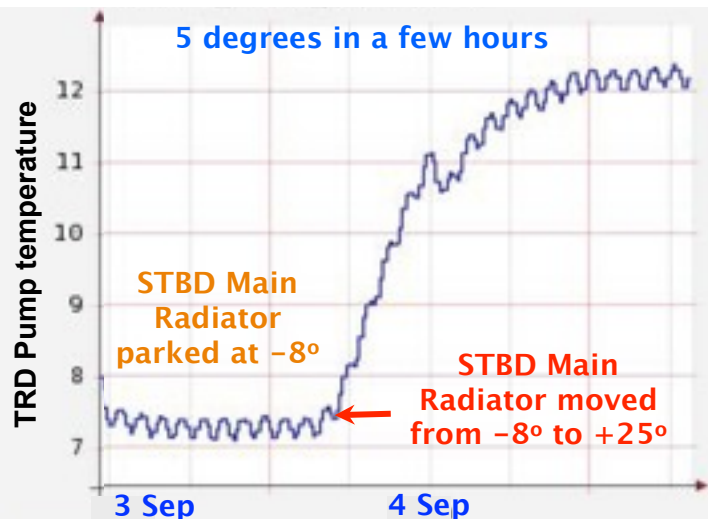
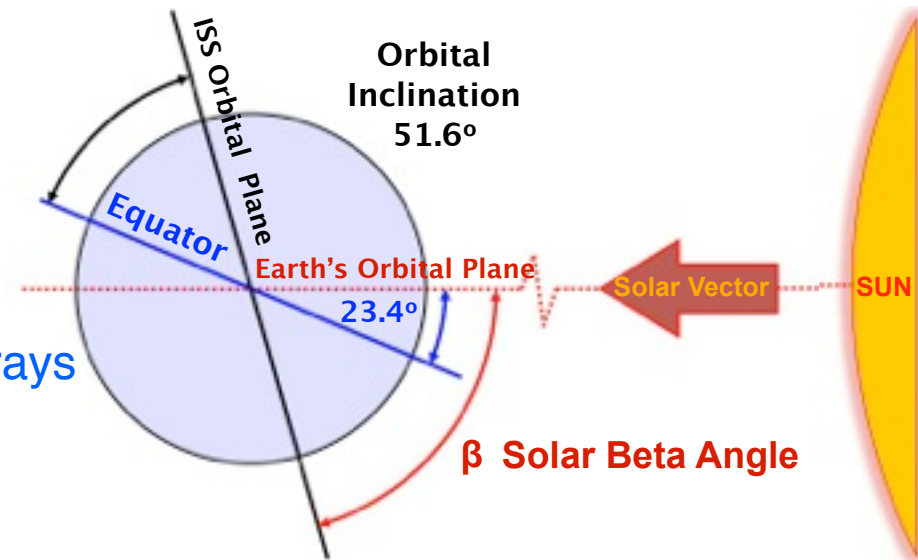


On-orbit thermal control

The thermal environment on ISS is constantly changing due to:

- Solar Beta Angle (beta)
- Position of the ISS Radiators and Solar Arrays
- ISS Attitude

Over 1,100 temperature sensors and 298 heaters are monitored to assure components stay within thermal limits and avoid permanent damage.

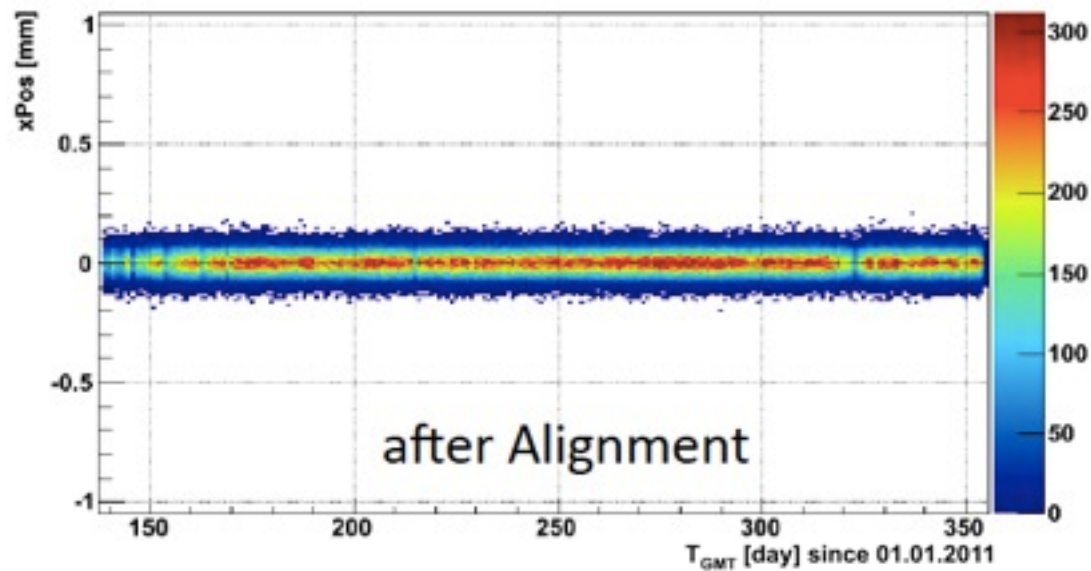
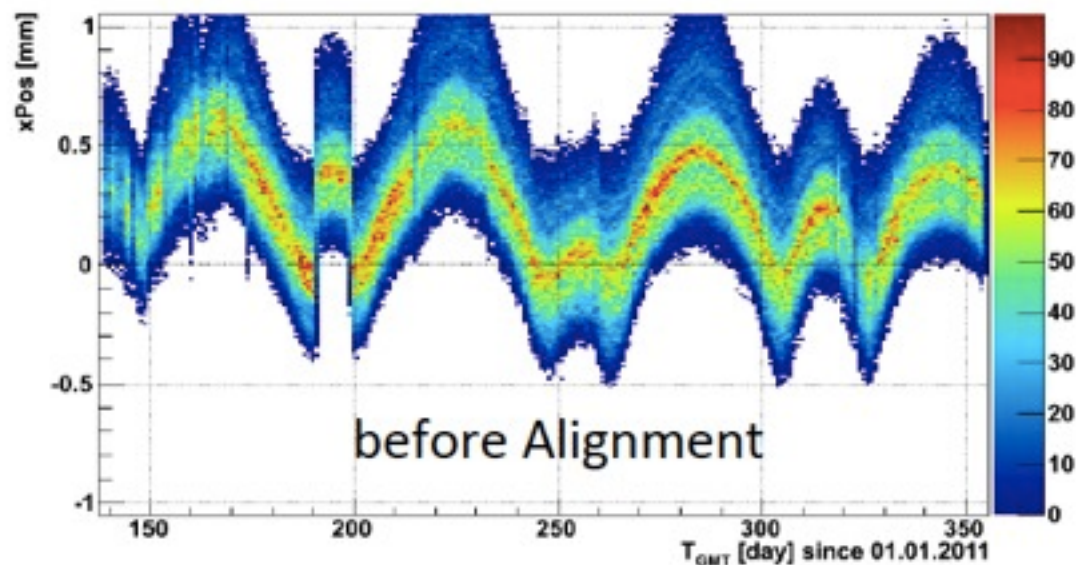


On orbit performance

TRD alignment

Due to temperature variations the TRD is moving on top of the inner tracker by up to 1 mm.

Cosmic protons are used for alignment to an accuracy of 0.04 mm for each straw module.

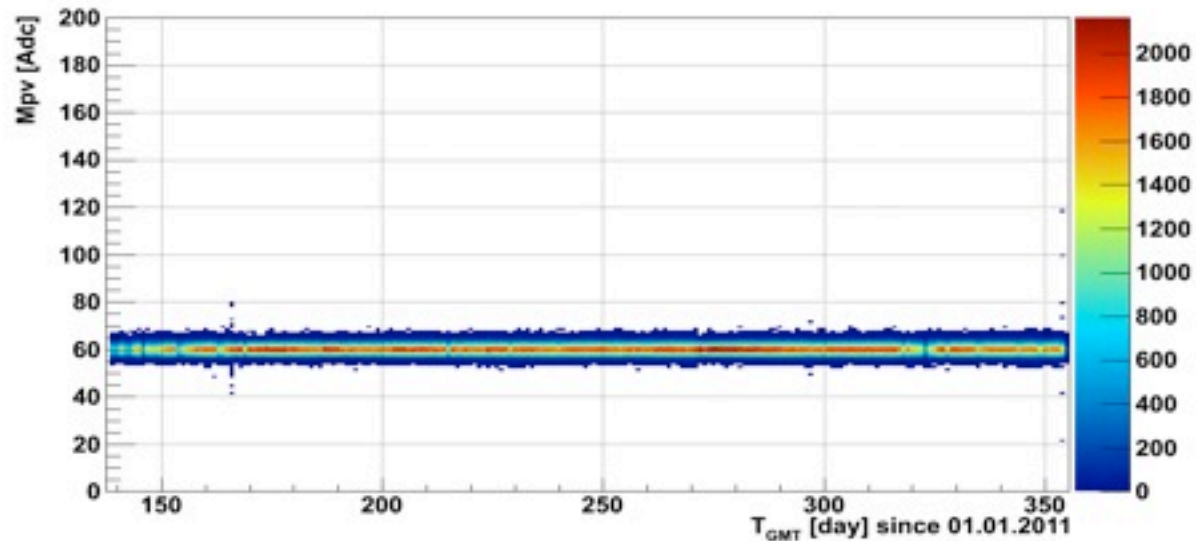
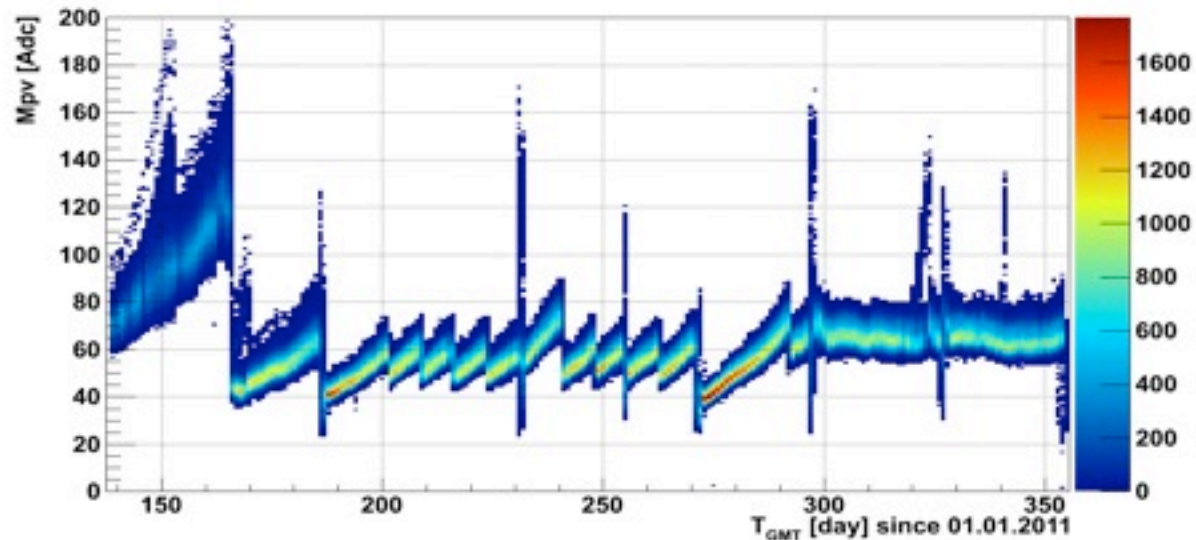


On orbit performance

TRD gain calibration

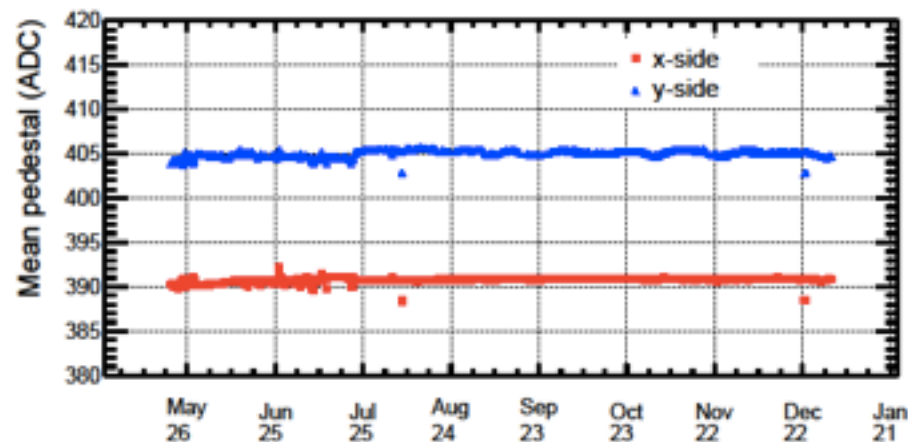
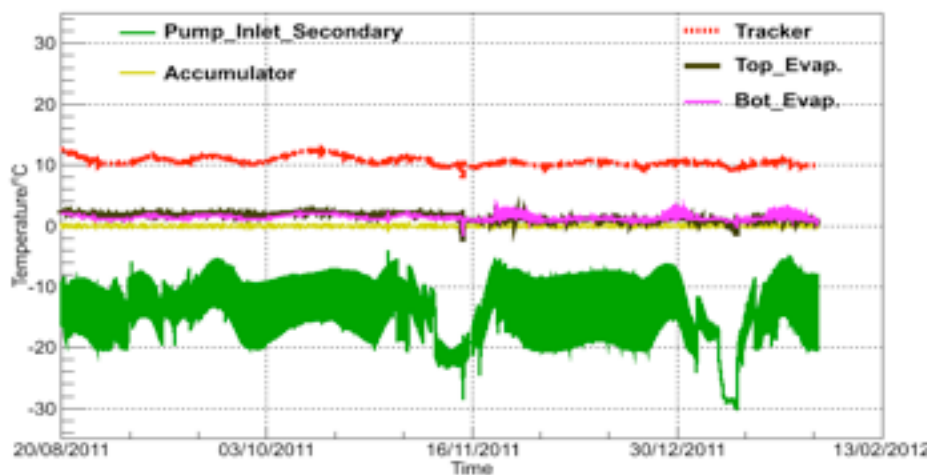
Due to temperature, pressure, gas composition and HV changes, the TRD detector response is changing.

Cosmic ray protons are used to calibrate the detector response to 3% accuracy

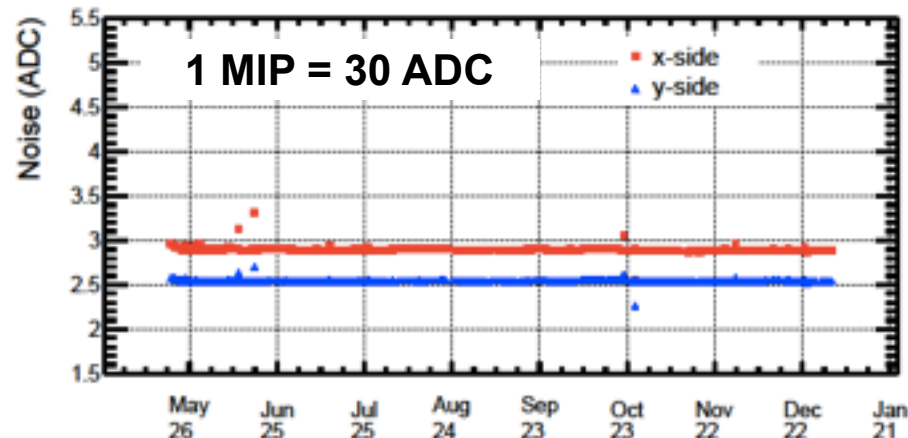
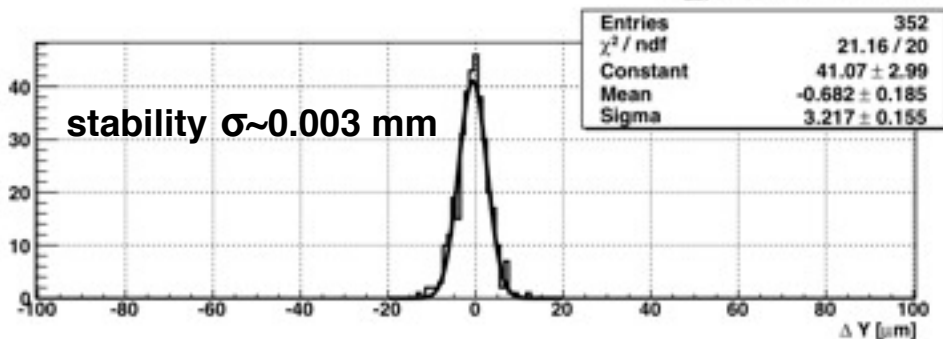
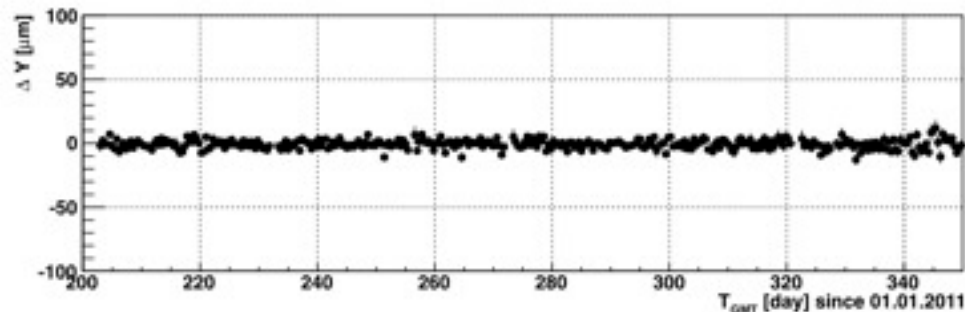


On orbit performance

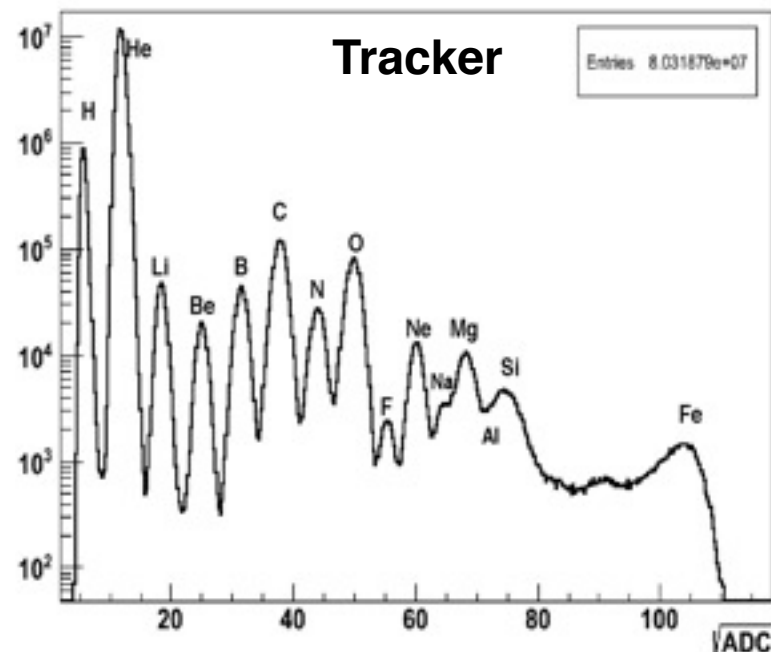
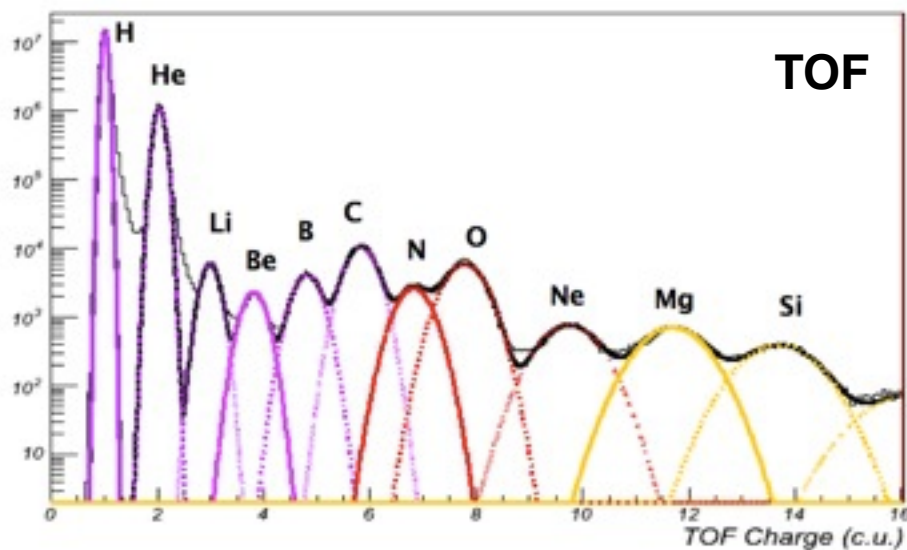
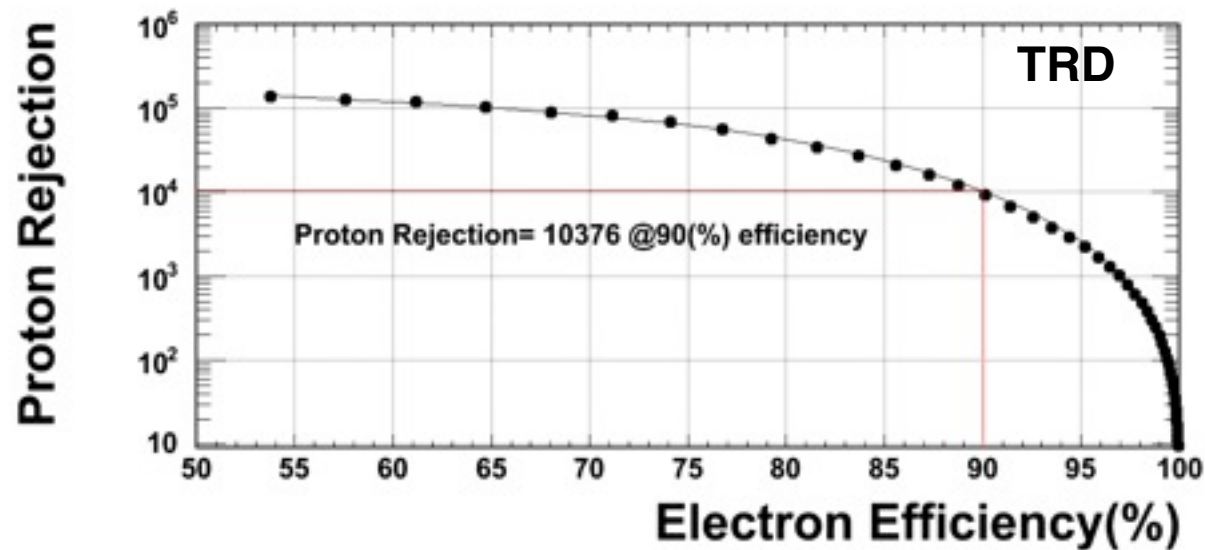
Tracker works as expected



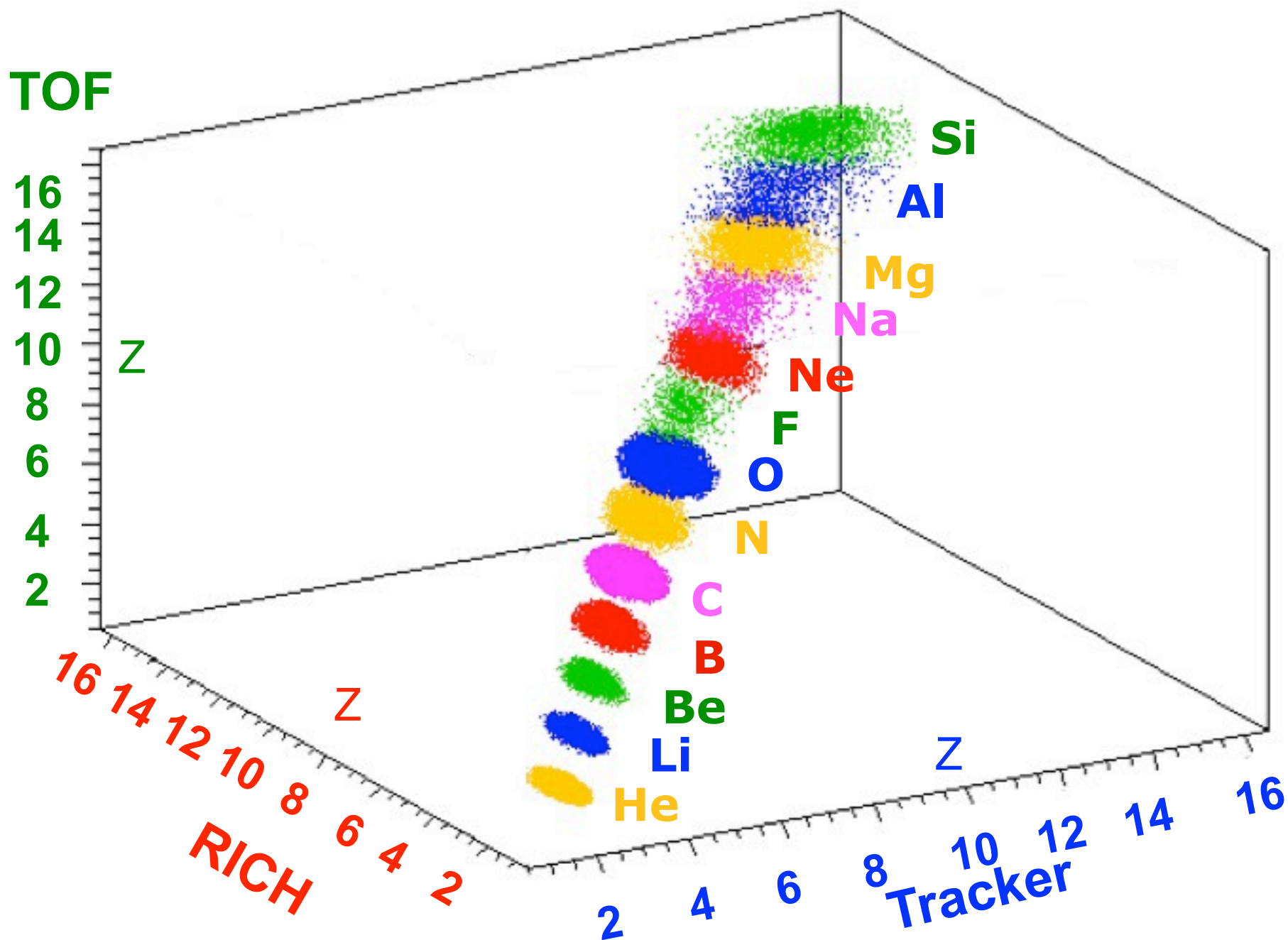
Tracker alignment stability



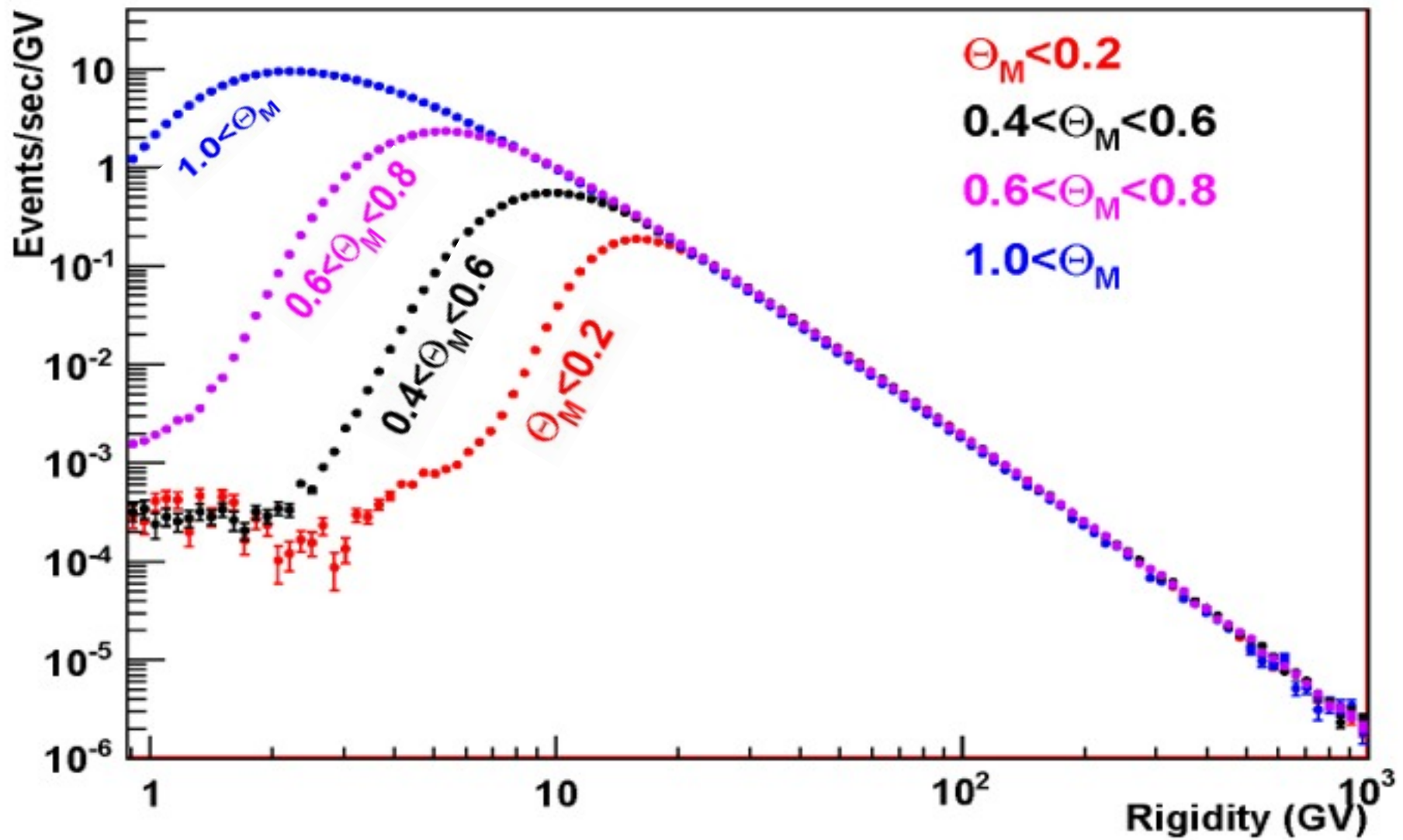
On orbit performance



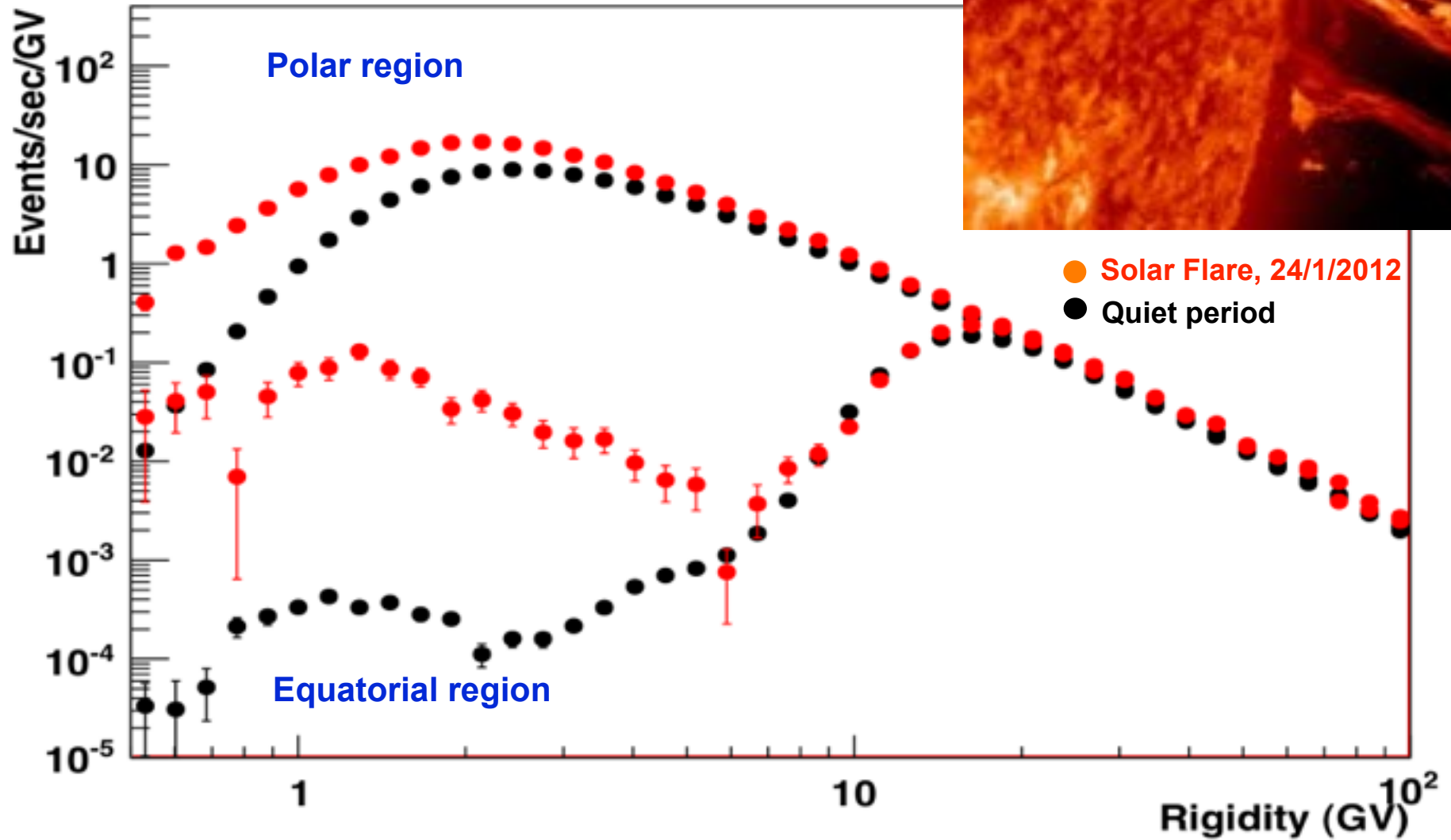
Redundant charge identification capability



He rate



He rate and Solar Flare



Conclusions

The Cosmos is the ultimate Lab:
cosmic rays can be observed at
energies higher than any
accelerator



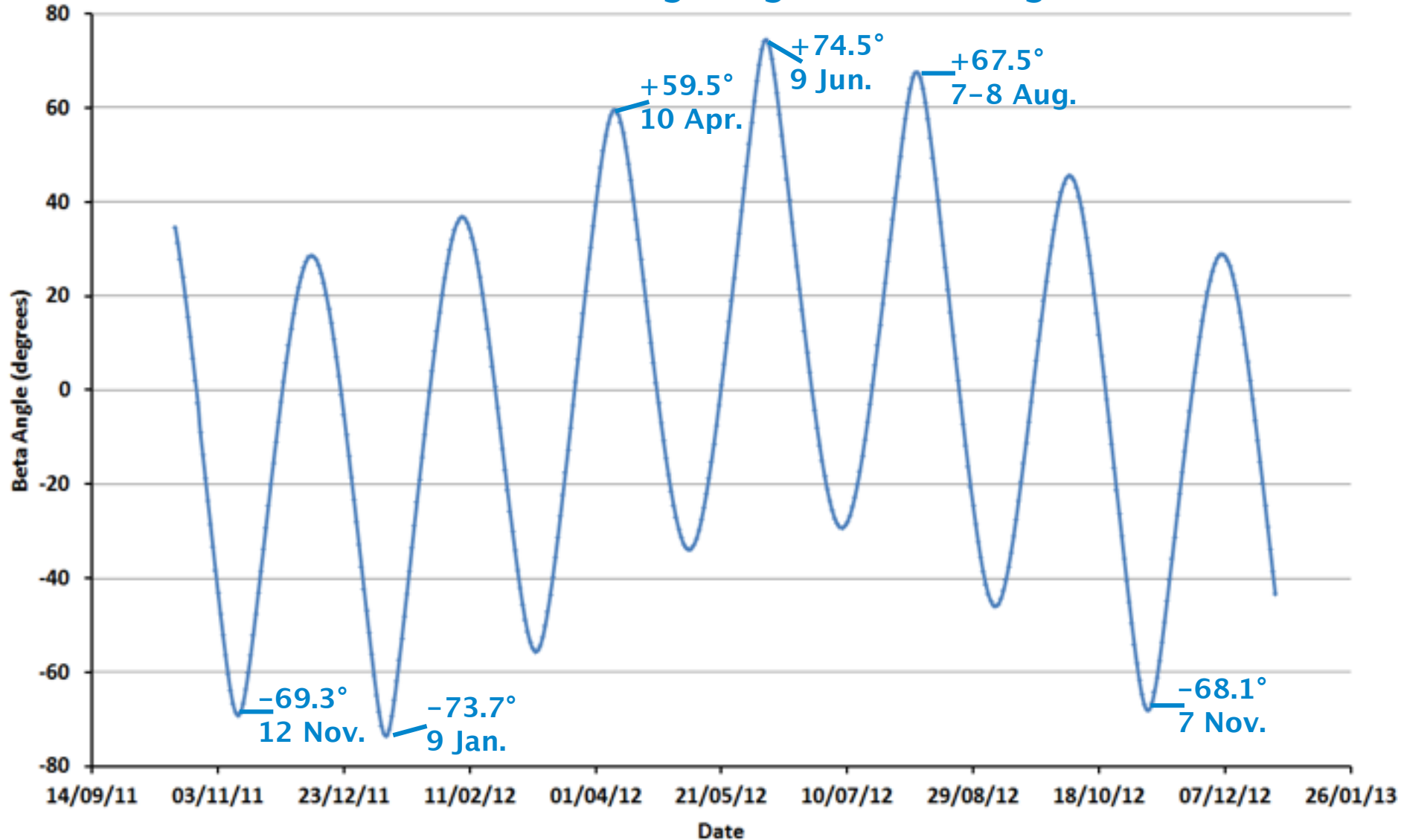
- AMS02 is in orbit since May 16th 2011
- All the detectors are properly functioning with DAQ in nominal conditions since May 19th 2011
- Ground operations (POCC and SOC) run smoothly
- Detector calibration (alignment, e/p rejection, charge id, etc.) are well advanced
- 10+ years on board the ISS at $16 \cdot 10^9$ events/year: great discovery potential

Science coming soon!!

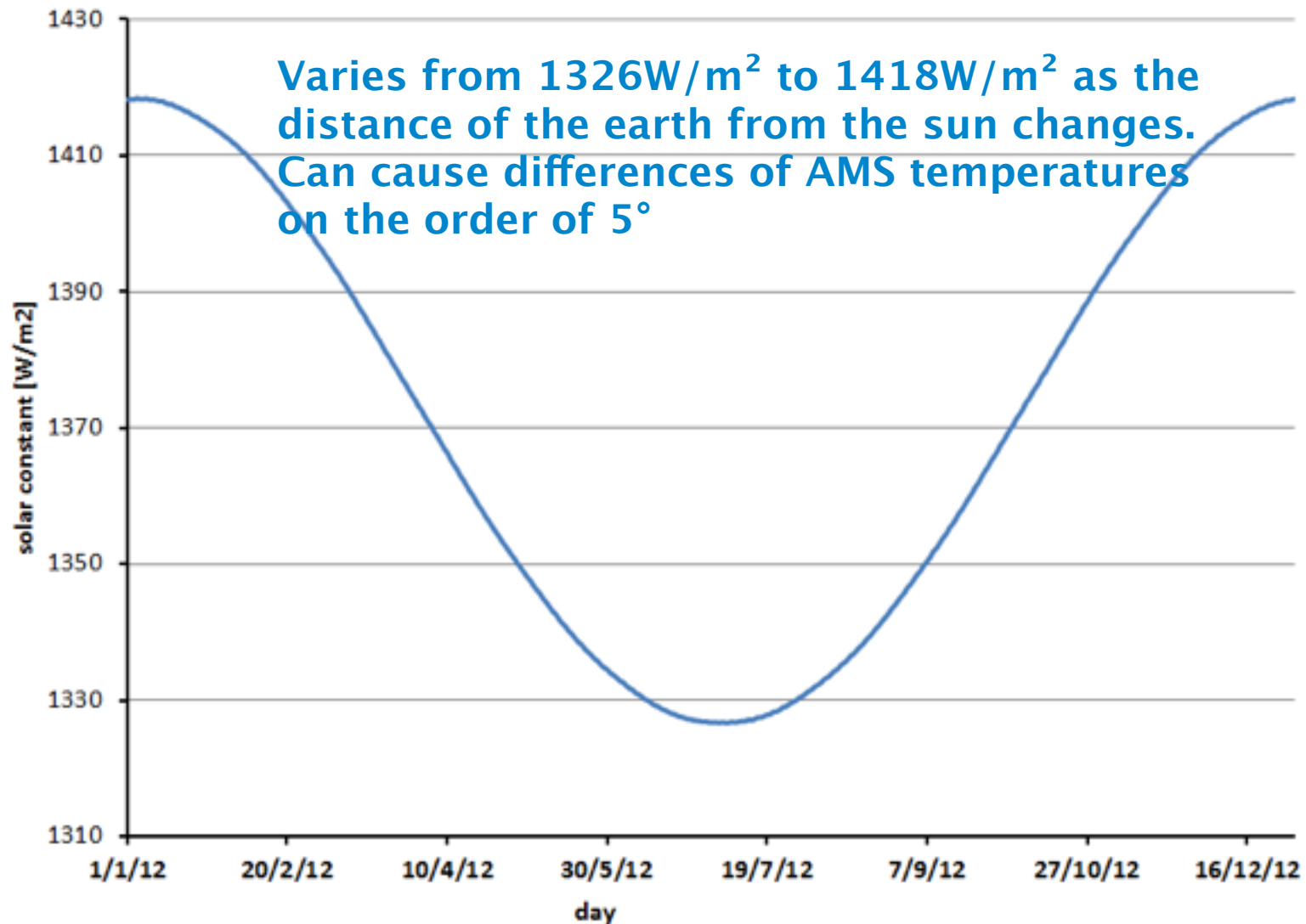
Back-ups

Evolution of the beta angle through 2012, with dates of extreme values
At large positive values, the port side of AMS is hot and the starboard side cold.

Vice-versa for large negative beta angles



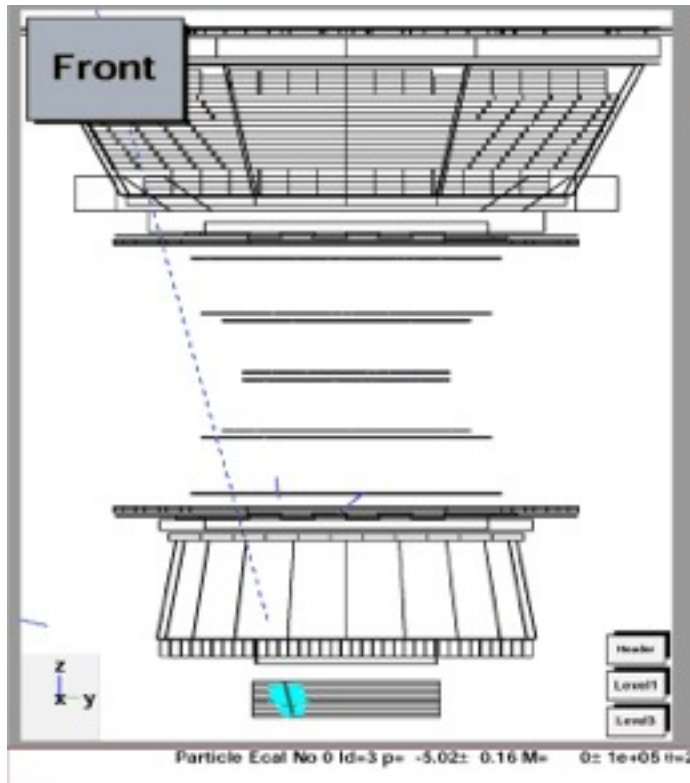
Solar Constant: illumination by the sun



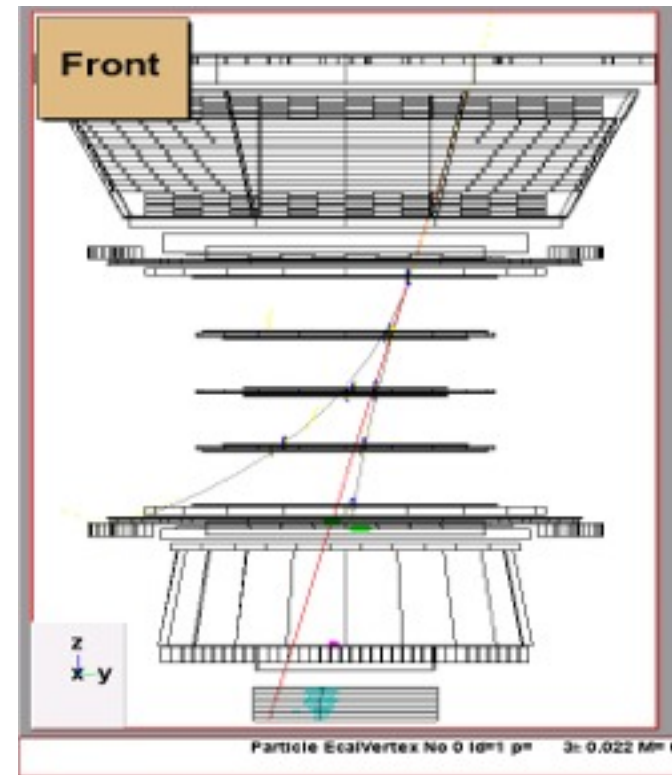
Scientific goals

γ rays astrophysics up to TeV energies

single photon mode



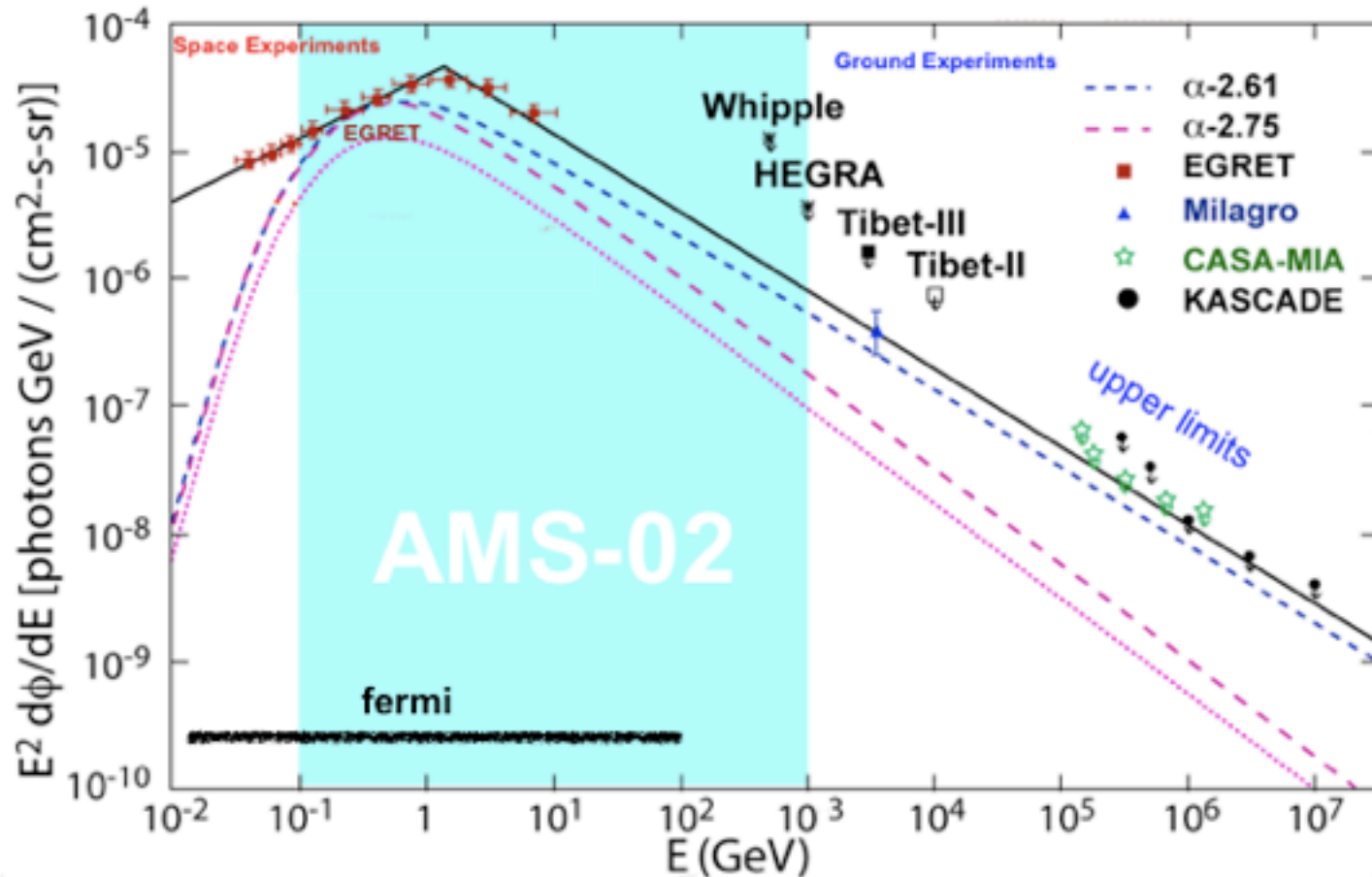
conversion mode



Energy spectrum for pulsars in the 100 MeV – 1 TeV and pulsar periods measured with μsec time precision.

Similar studies can be made for Blazars and Gamma Ray Bursters.

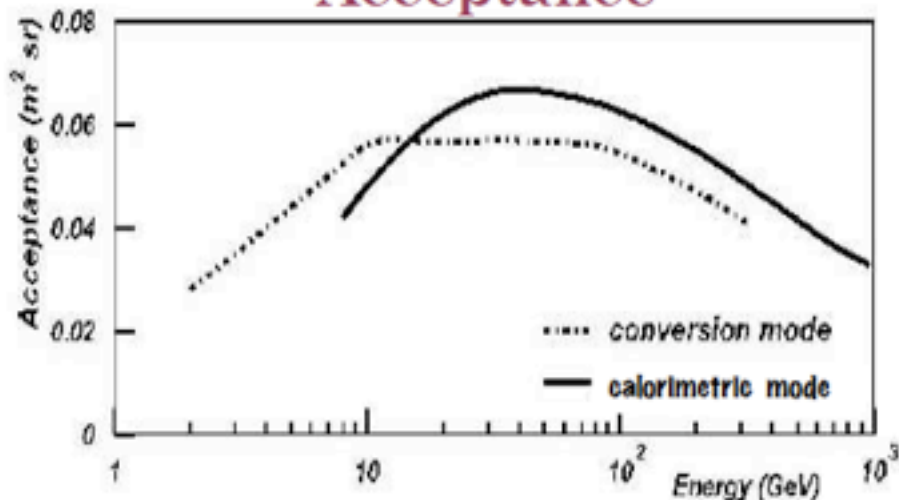
γ rays astrophysics up to TeV energies



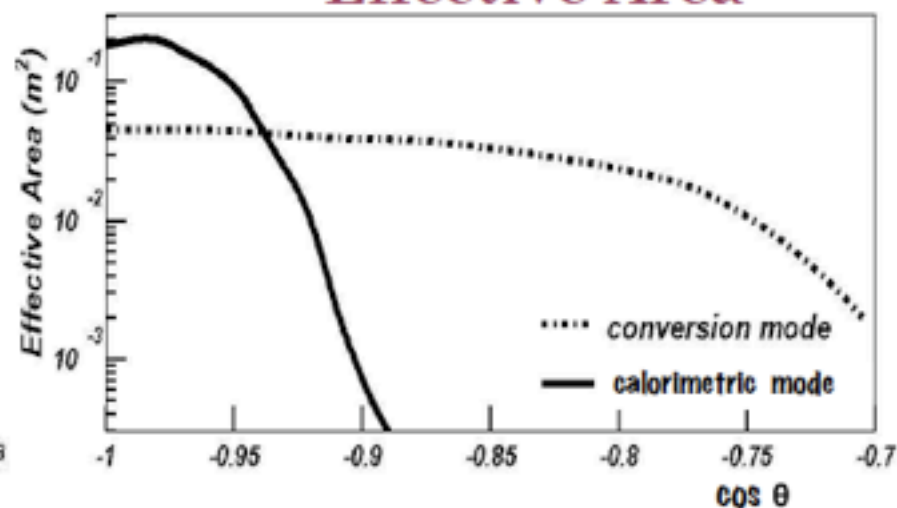
The diffuse gamma-ray spectrum of the Galactic plane

Tracker and Ecal Performances

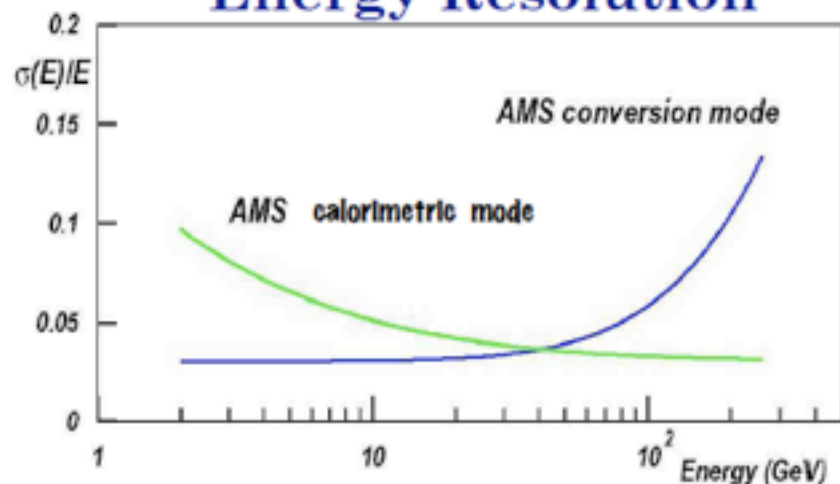
Acceptance



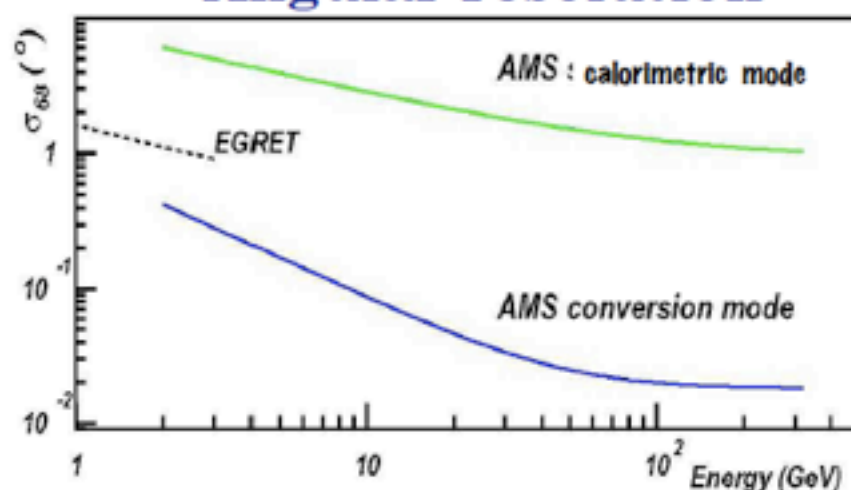
Effective Area



Energy Resolution



Angular resolution



Search for New Matter in the Universe

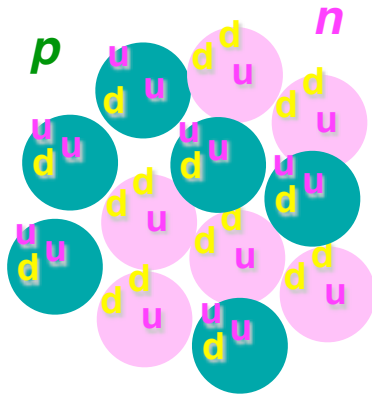
After many years, the question of the existence of strange quark matter still remains without a definitive answer.

There are six types of Quarks found in accelerators (u, d, s, c, b, t).

All matter on Earth is made out of only two types (u, d) of quarks. “Strangelets” are new types of matter composed of three types of quarks (u, d, s) which should exist in the cosmos.

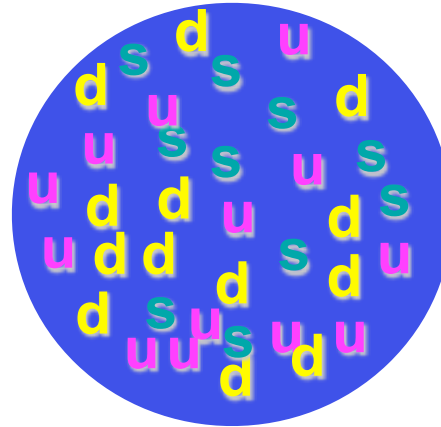
Carbon Nucleus

$Z/A \sim 0.5$



Strangelet

$Z/A \sim 0.1$

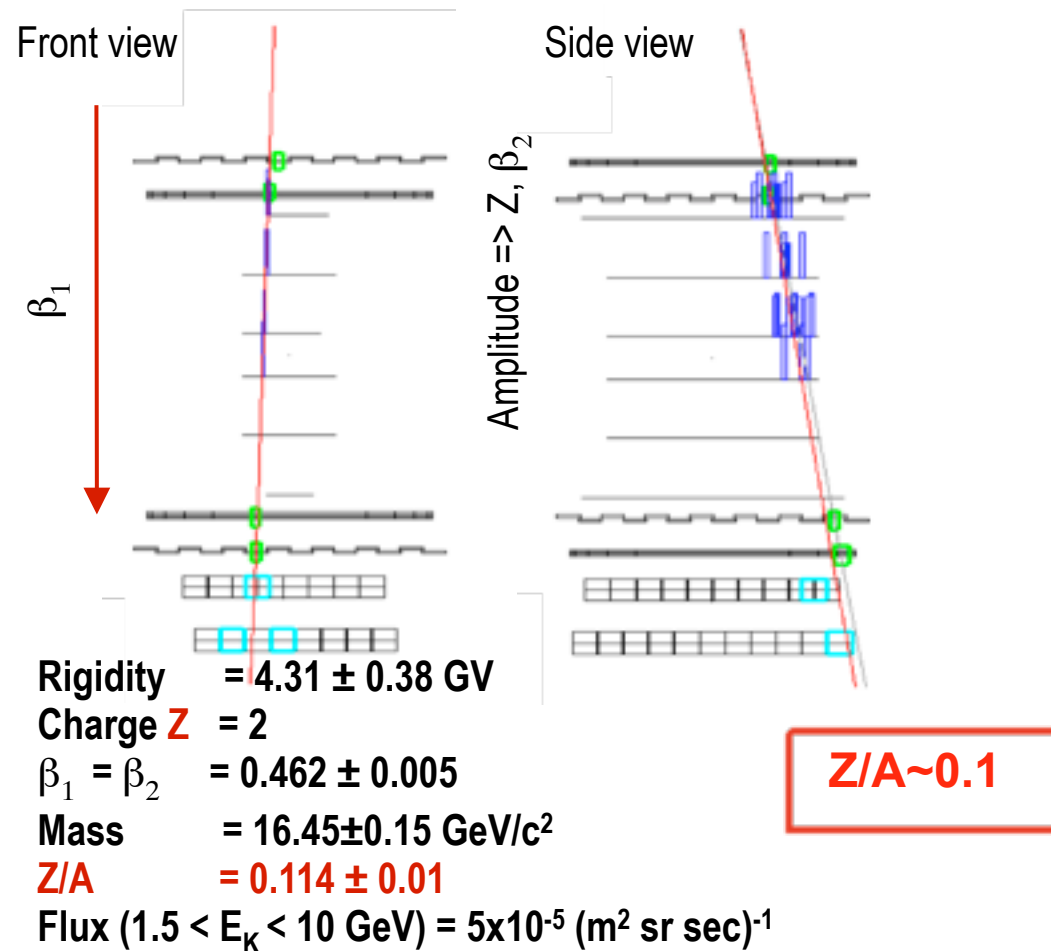


E. Witten, Phys. Rev. D, 272-285 (1984)

Strangelets

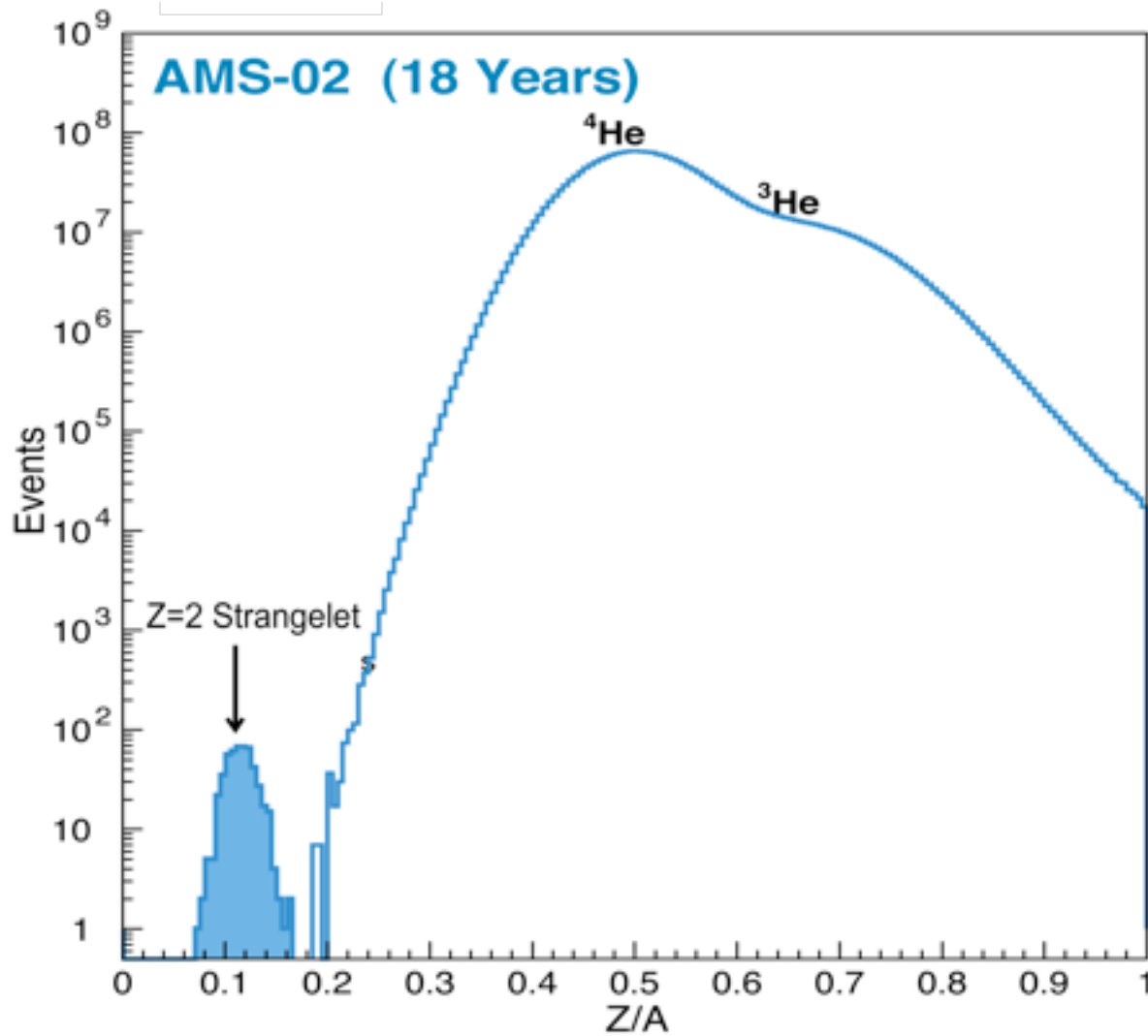
Jack Sandweiss (Yale) is leading the AMS search.

Candidate observed with AMS-01 5 June
1998 11:13:16 UTC

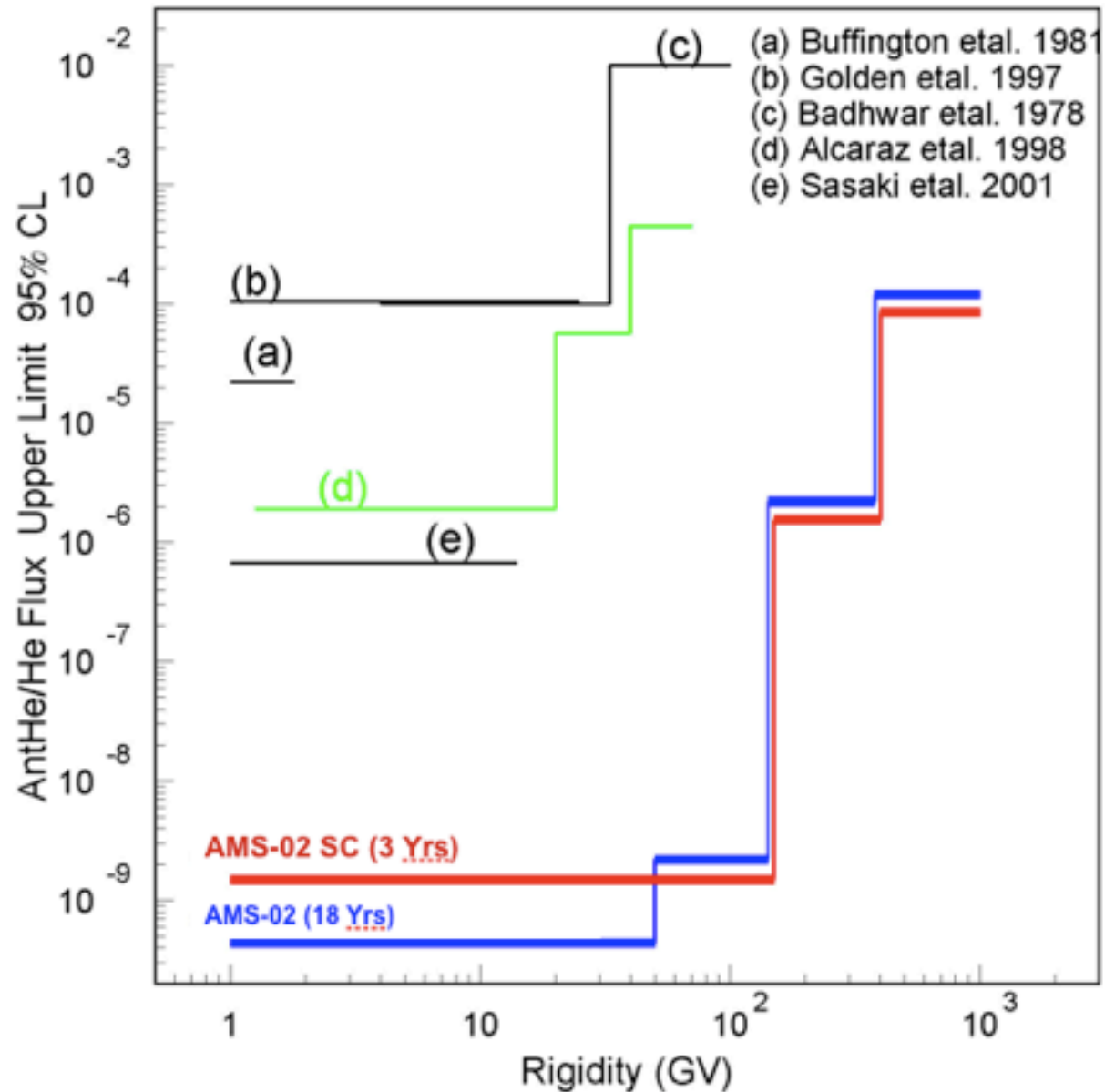


Strangelets

$$\Phi_{\text{strangelets}} = 5 \times 10^{-10} (\text{cm}^2 \text{s sr})^{-1}$$



Antimatter search



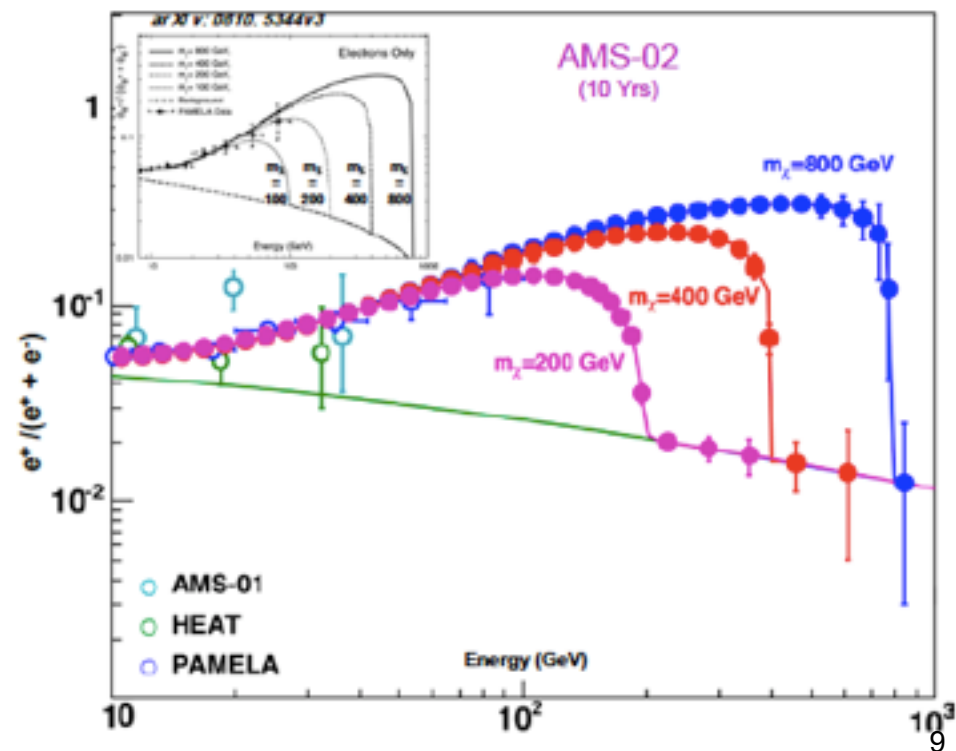
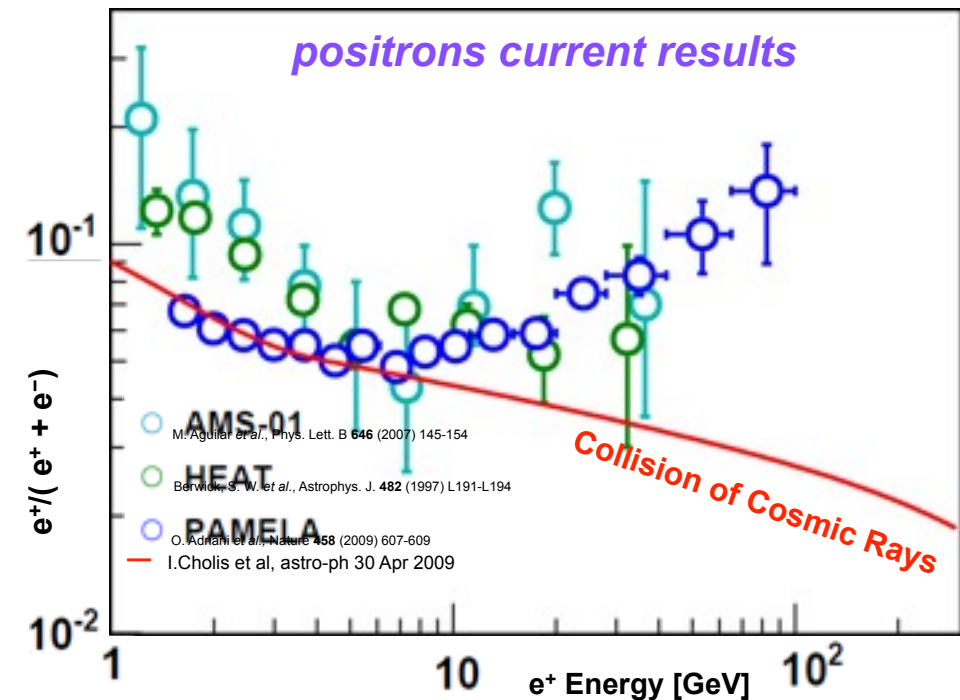
Dark Matter search: positron channel

AMS will explore the indirect detection of dark matter, measuring the dark matter annihilation and collision products. Combining searches in several different channels as:

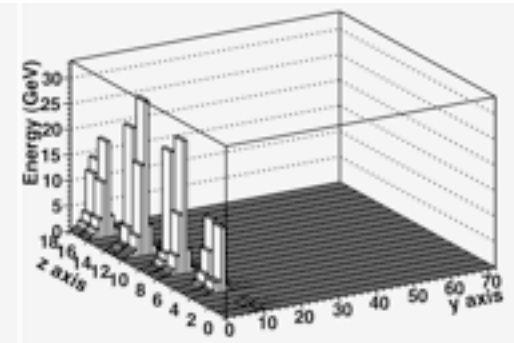
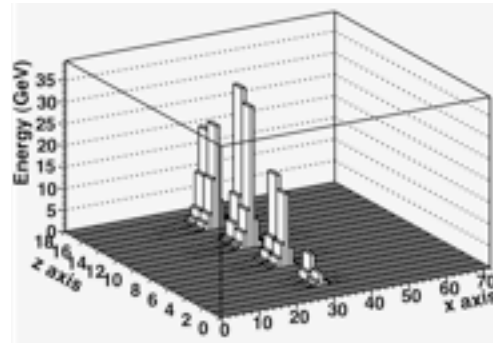
- e^+
- Photons
- p
- Antideuterons

The leading candidate for Dark Matter is a SUSY neutralino (χ^0)

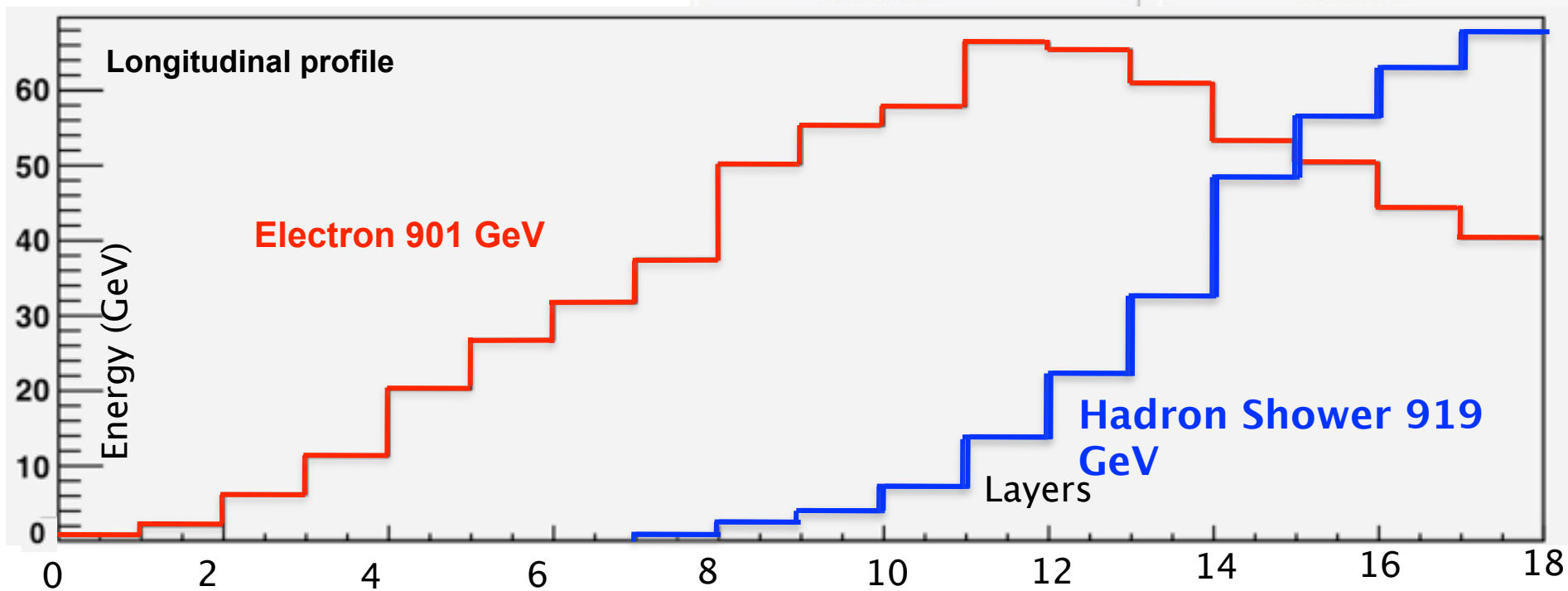
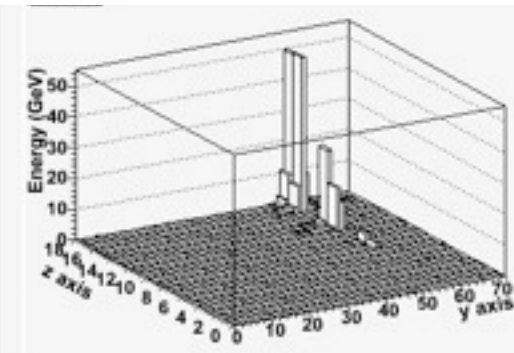
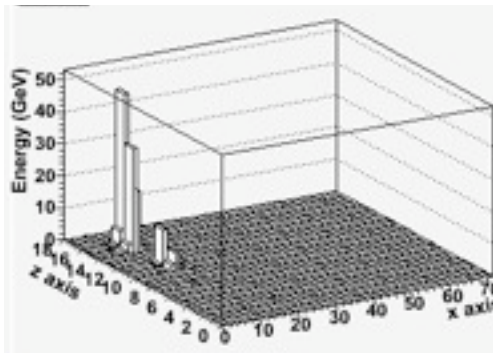
Collisions of χ^0 will produce excess in the spectra of e^+ different from known cosmic ray collisions



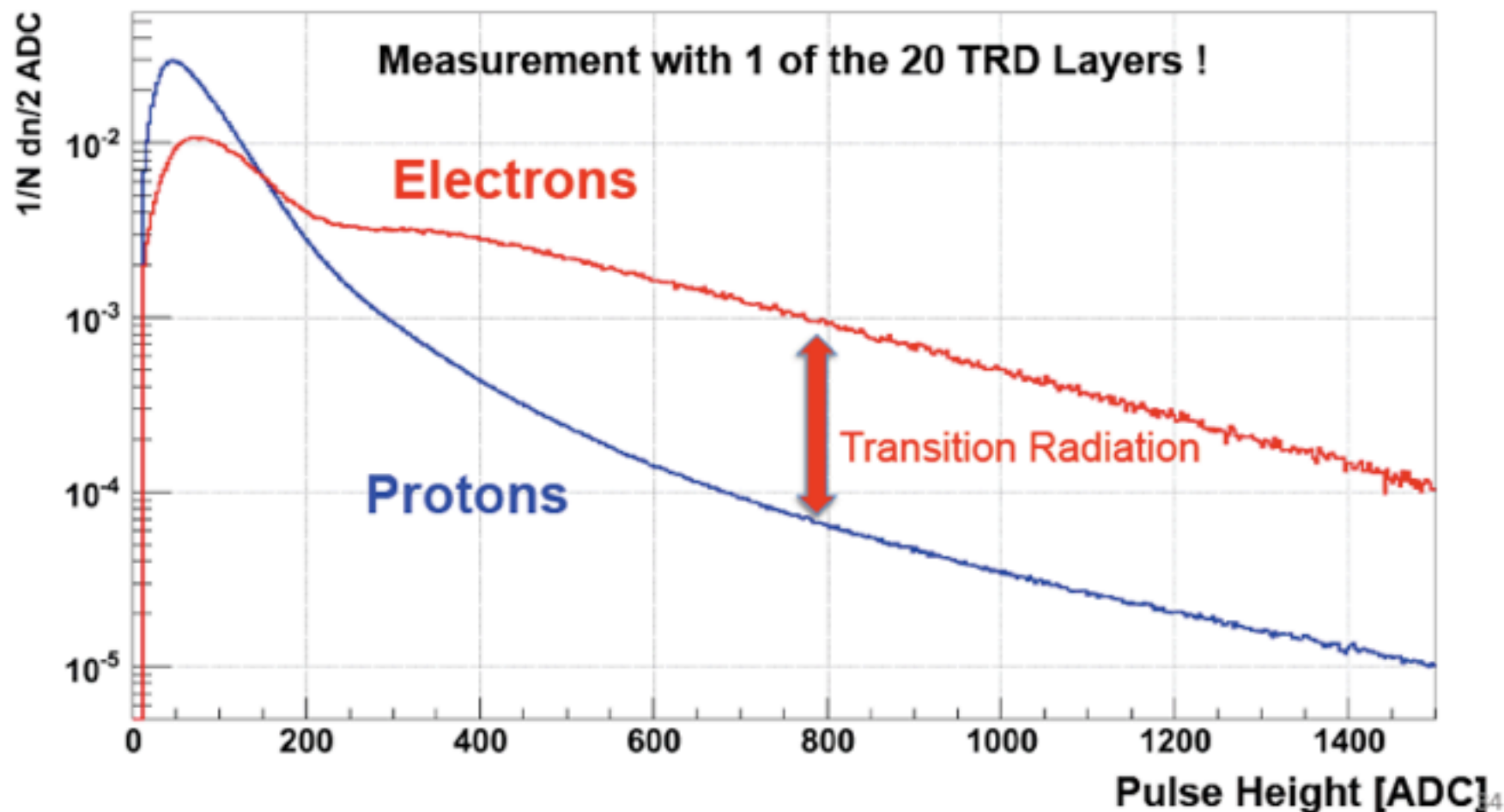
Electron 901 GeV



Hadron Shower 919 GeV

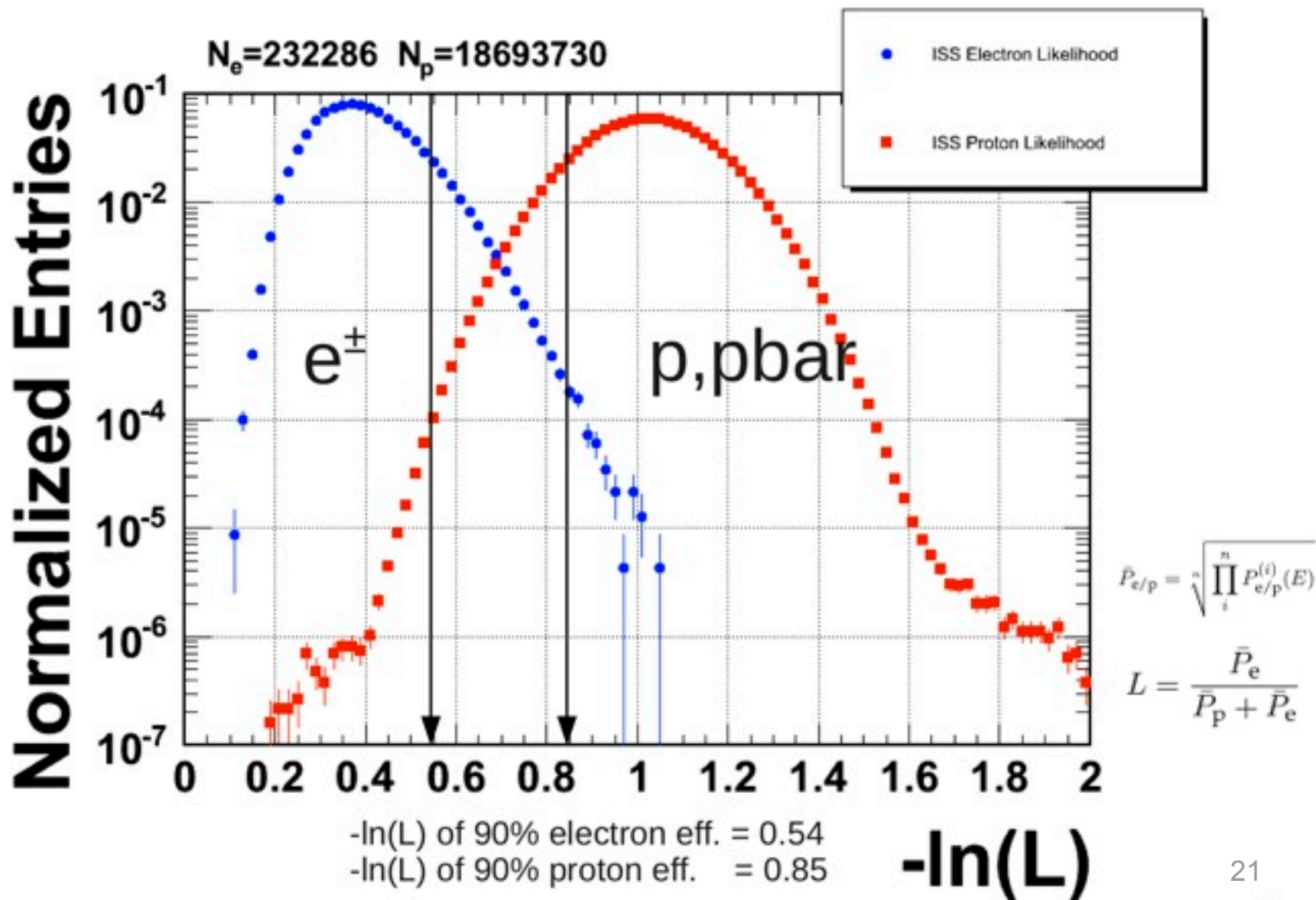


- Use the AMS Tracker and Electromagnetic Calorimeter to define a clean Electron and Proton sample.
- Study the TRD response in Space and determine the particle identification power from space data directly !



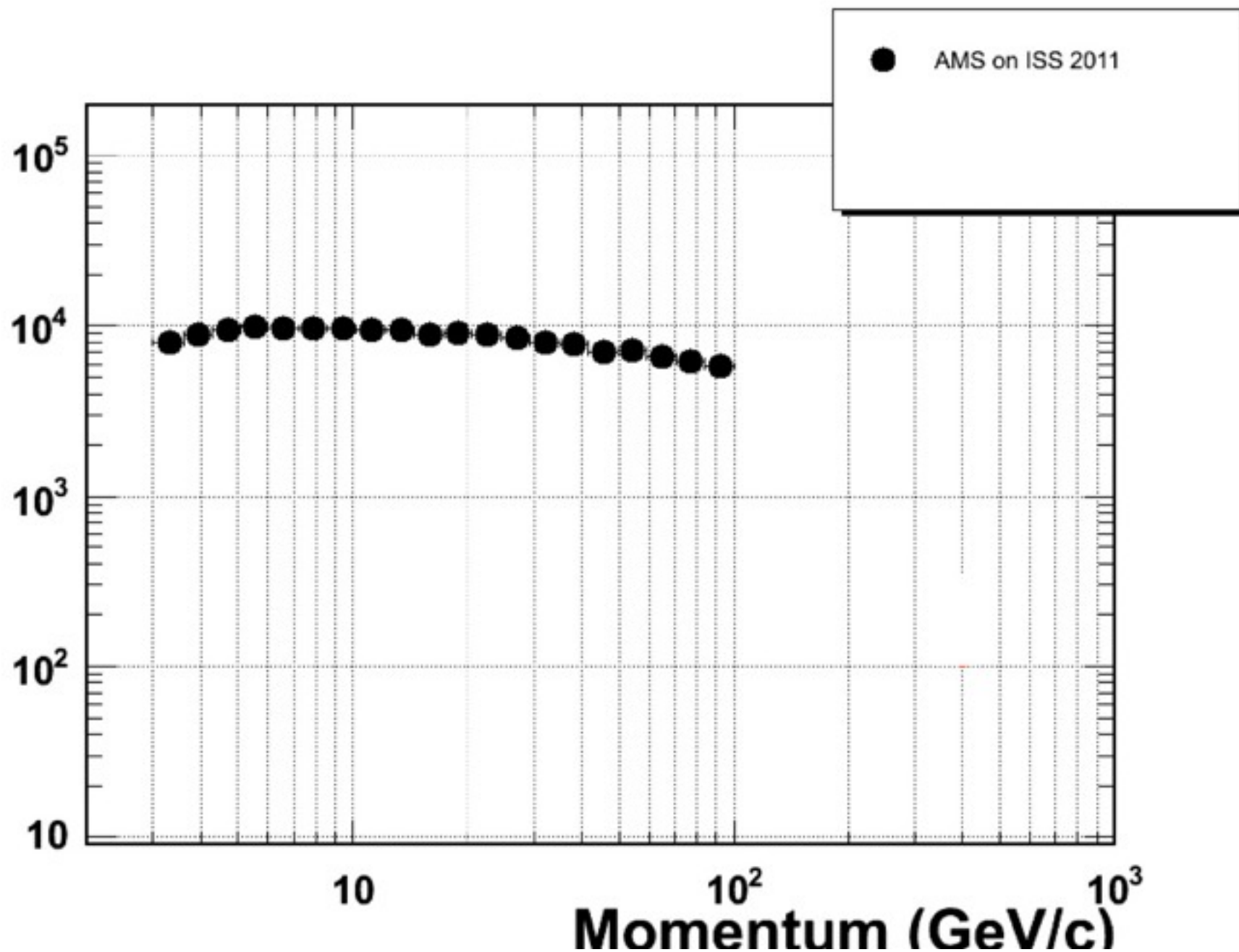
Electron vs. Proton

AMS TRD



TRD Proton Rejection at 90% Electron Efficiency

Proton Rejection

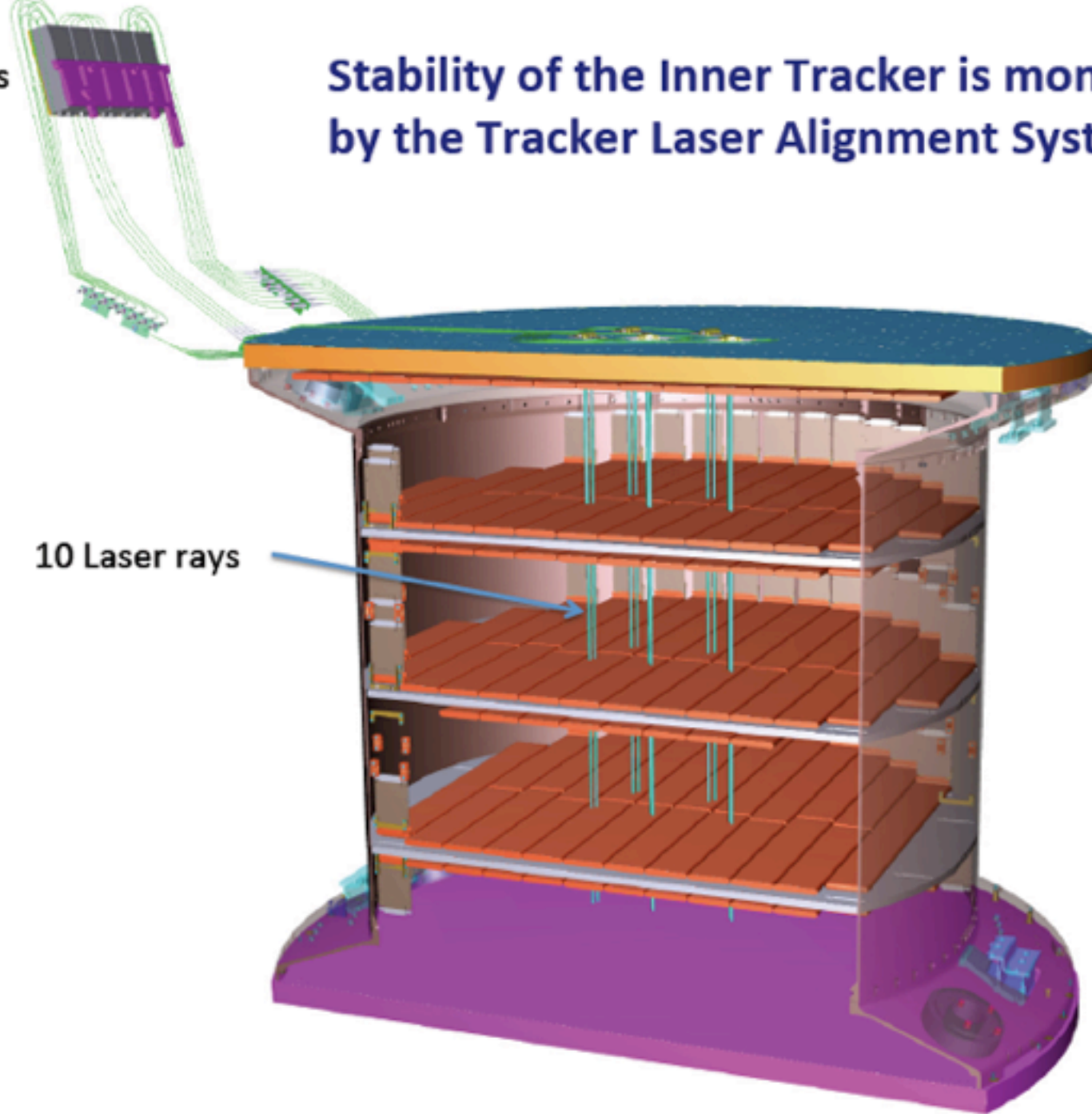


Alignment

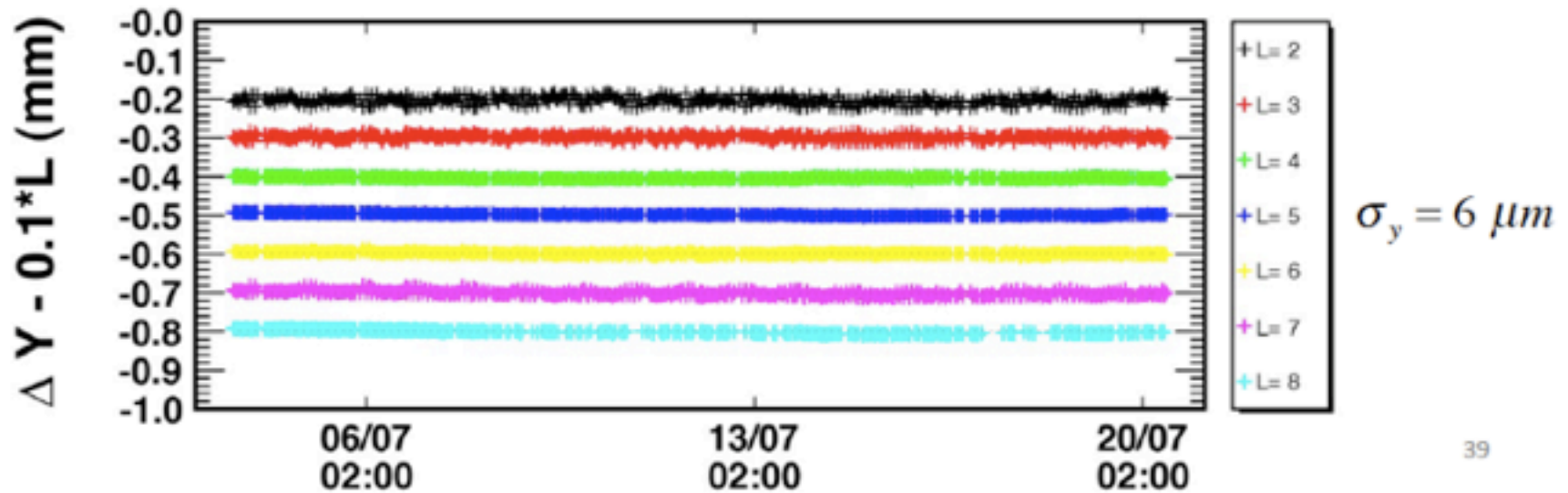
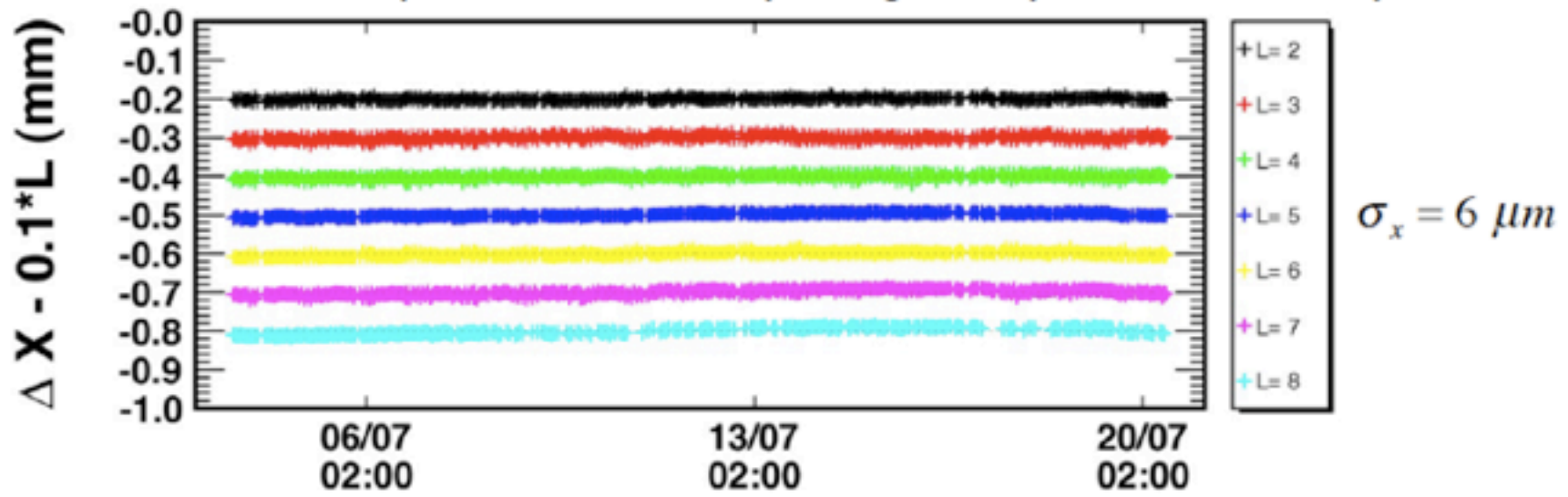
Laser Diodes
 $\lambda=1080\text{ nm}$

**Stability of the Inner Tracker is monitored
by the Tracker Laser Alignment System.**

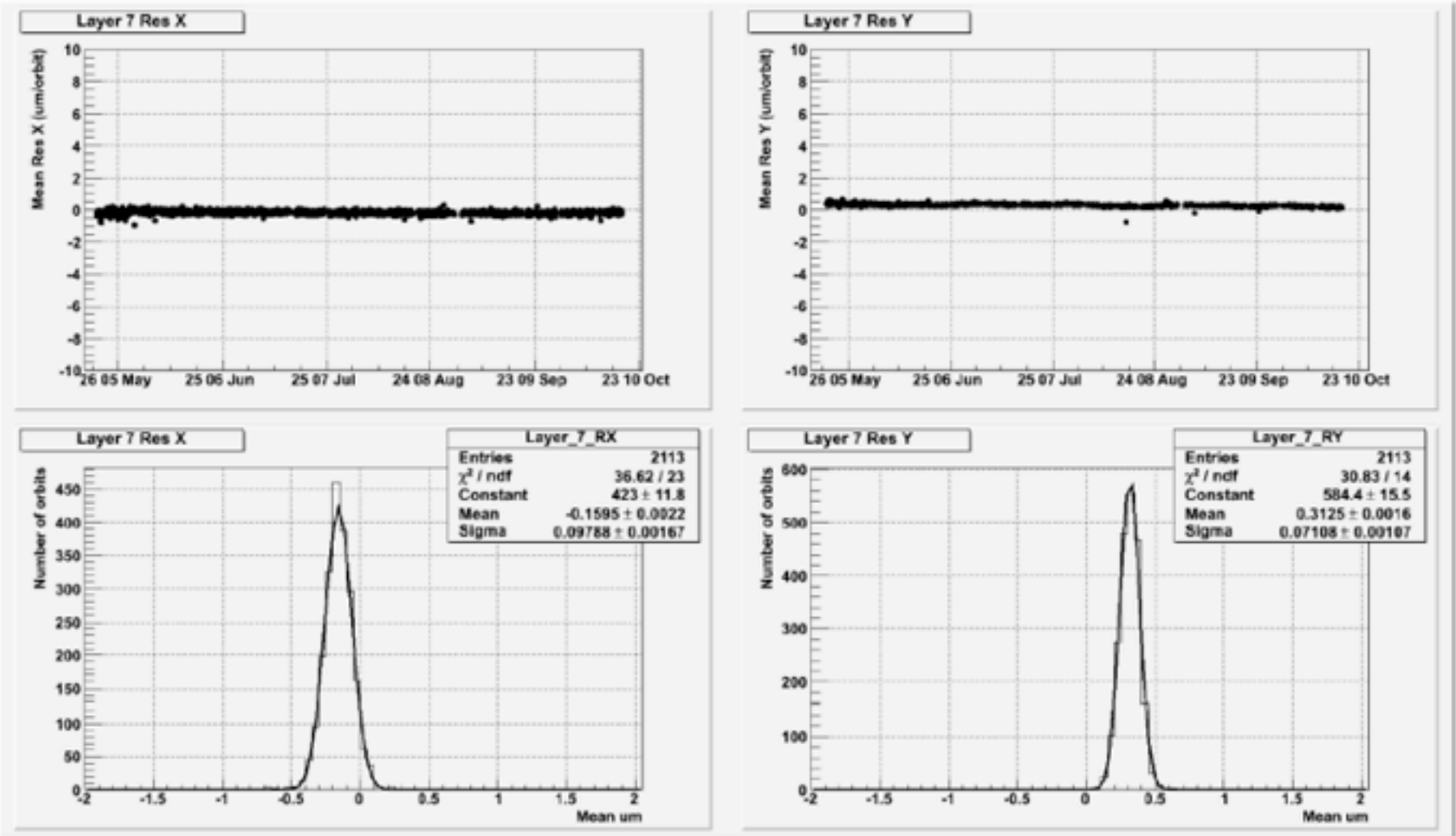
10 Laser rays



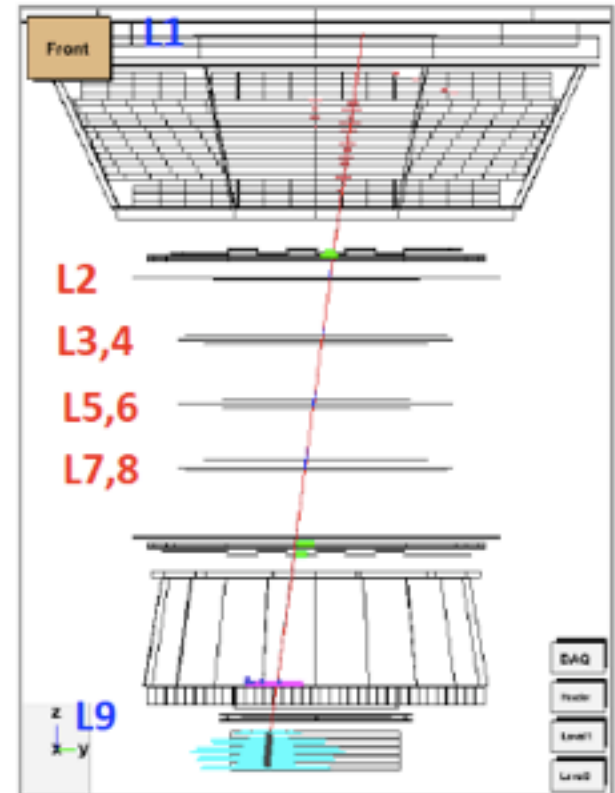
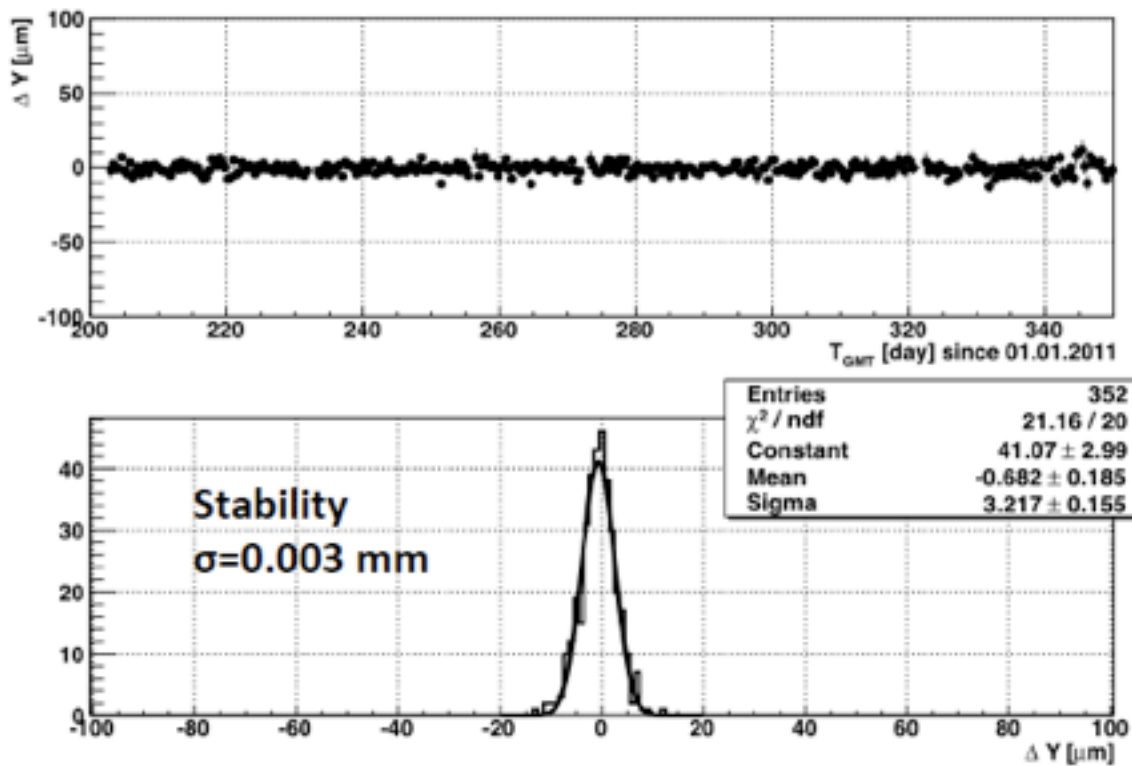
Stability of the Inner Silicon Tracker of AMS (not corrected for the pointing stability of the laser beams!)



Typical stability of an inner tracker plane is 0.1 μm as measured with Protons on the ISS

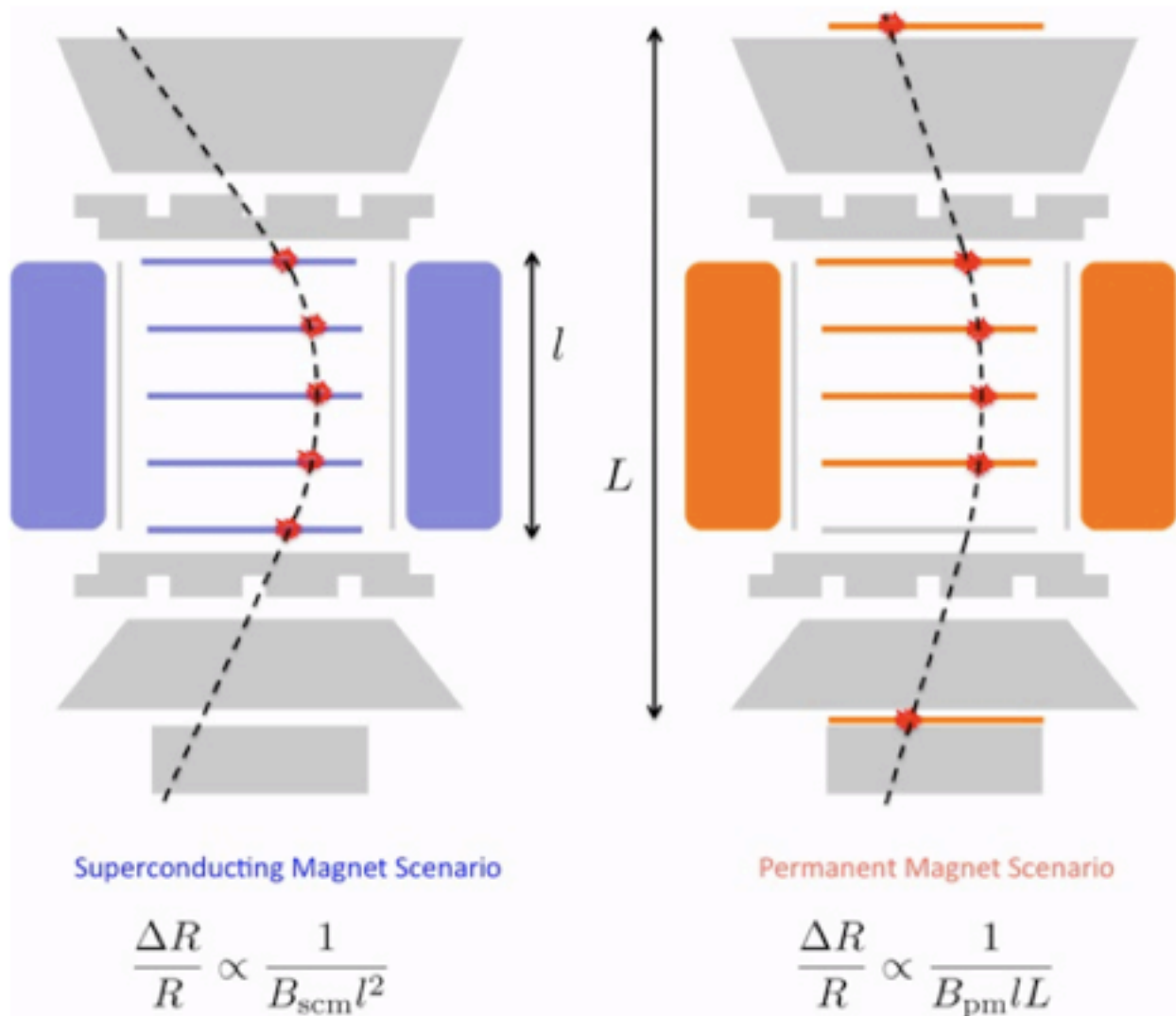


Stability of Layer 1 with respect to the Inner Tracker (bending plane)



Magnet

Magnets Comparison



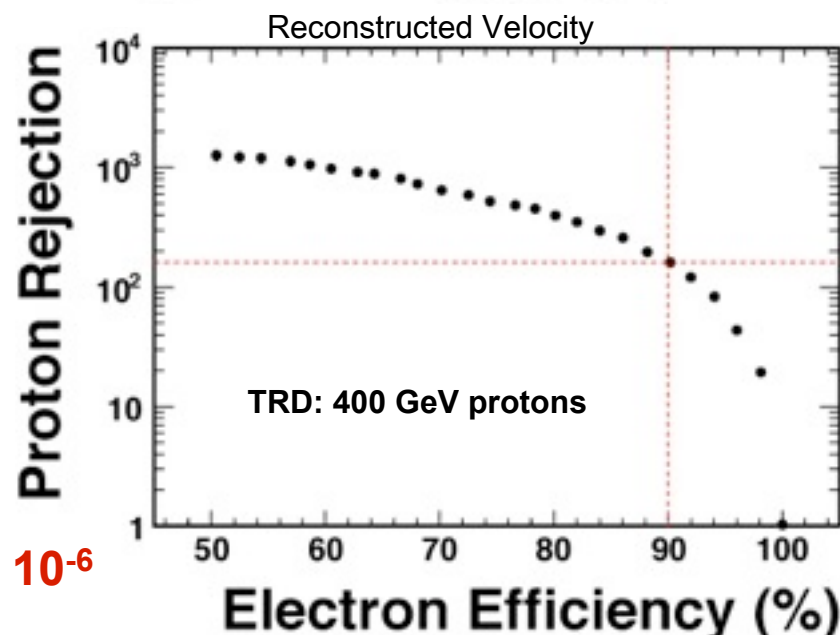
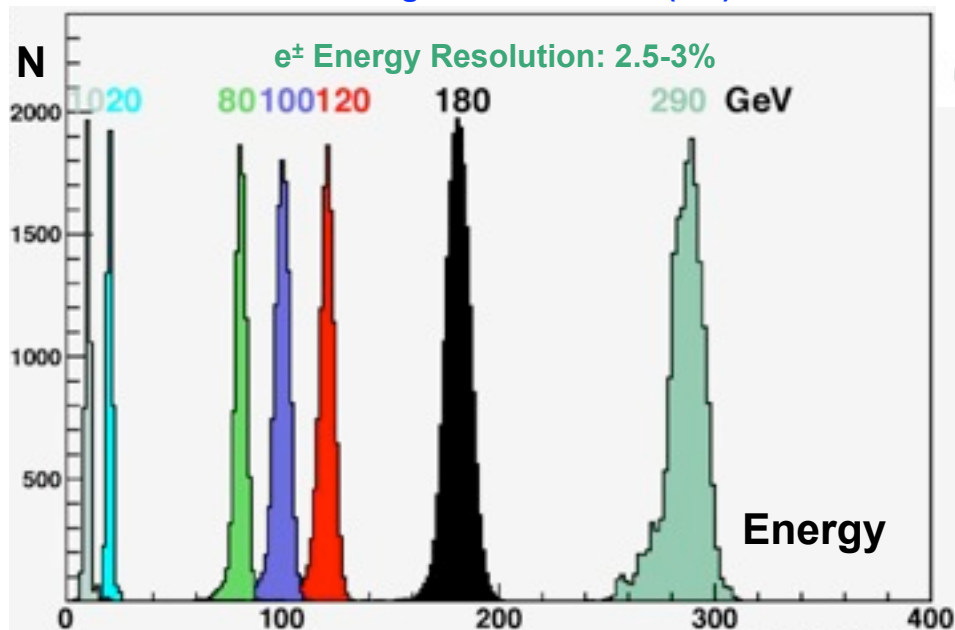
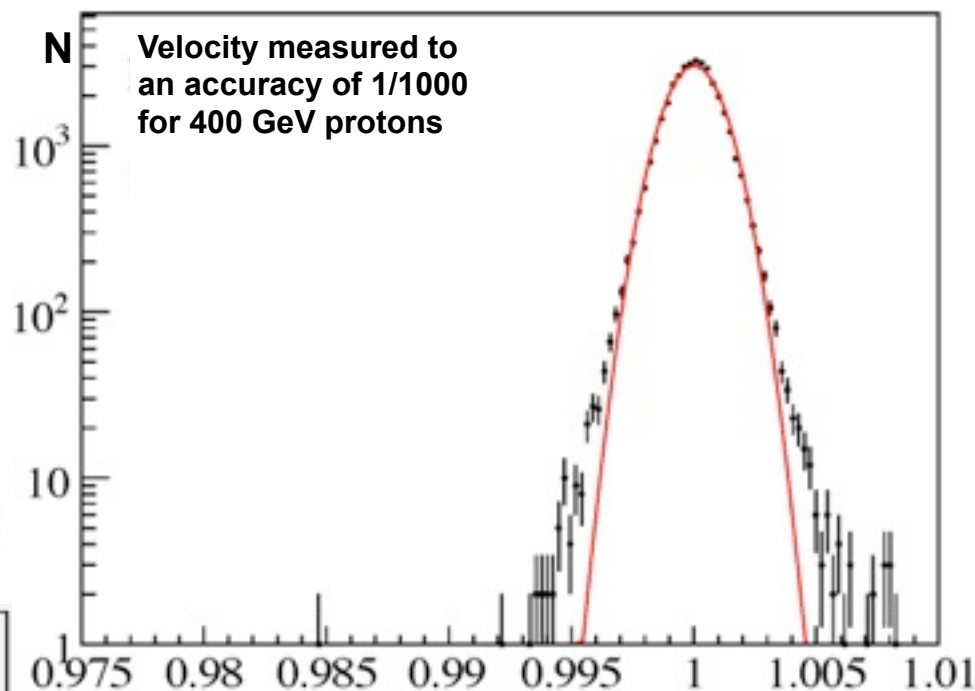
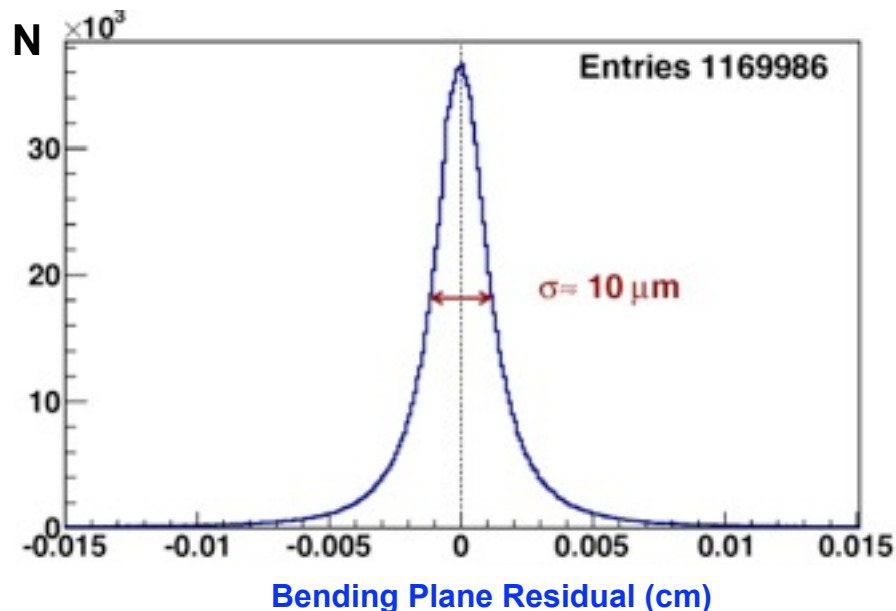
$B_{\text{scm}} \sim 0.87 \text{ T}$

$l \sim 1 \text{ m}$

$B_{\text{pm}} \sim 0.14 \text{ T}$

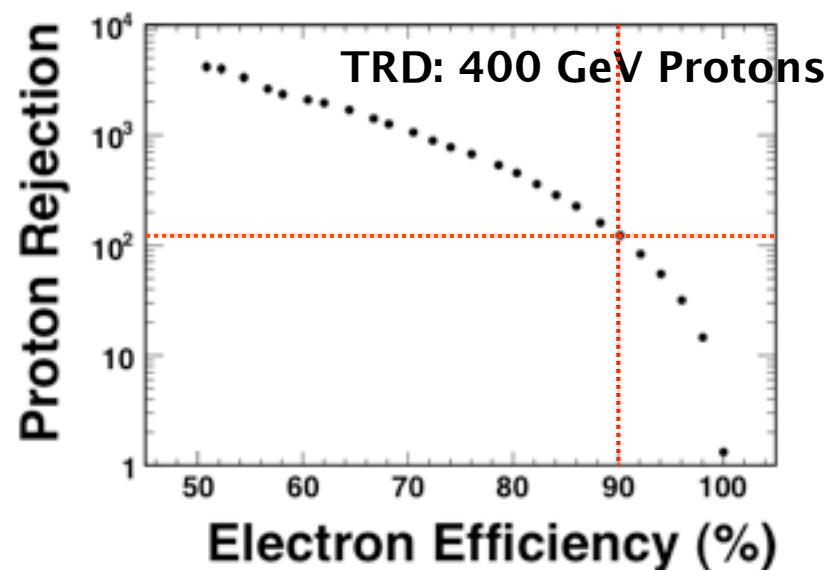
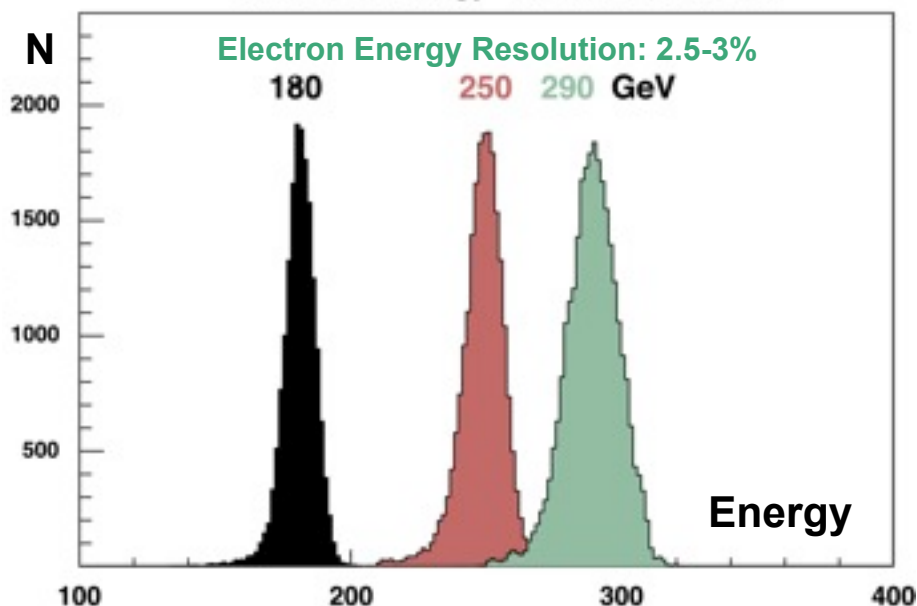
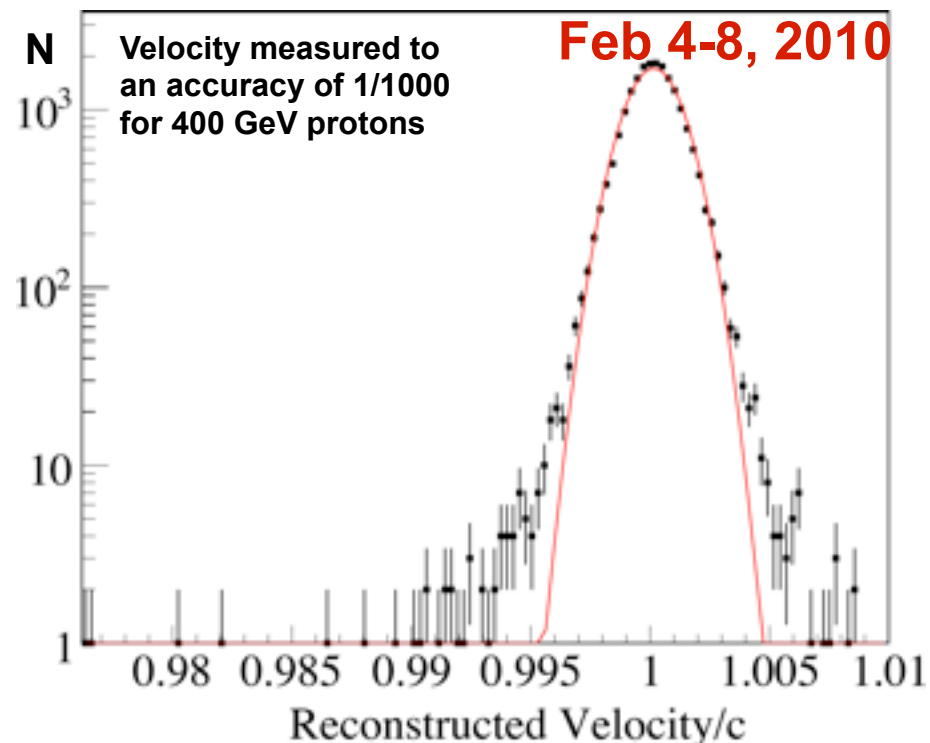
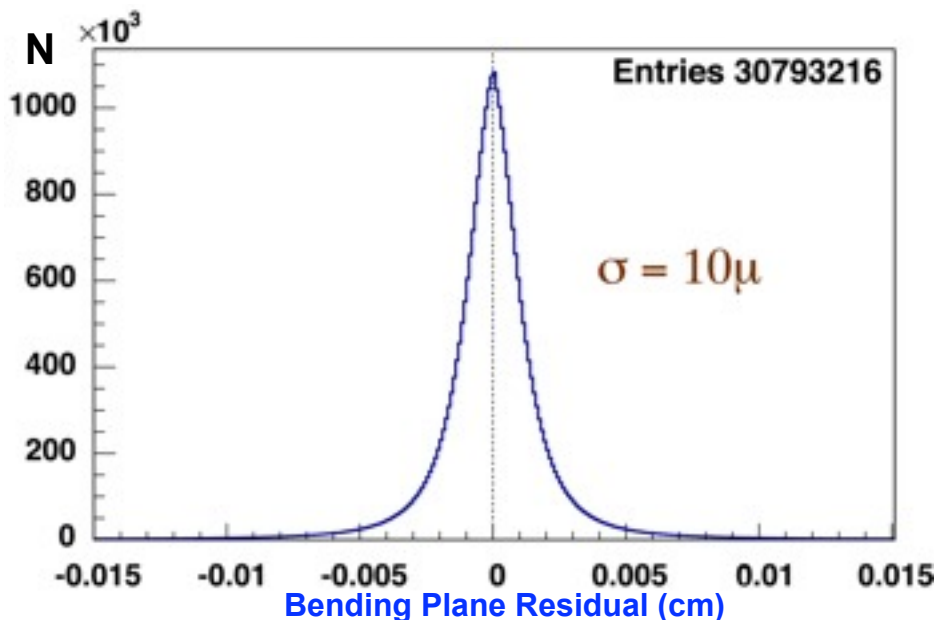
$L \sim 4 \text{ m}$

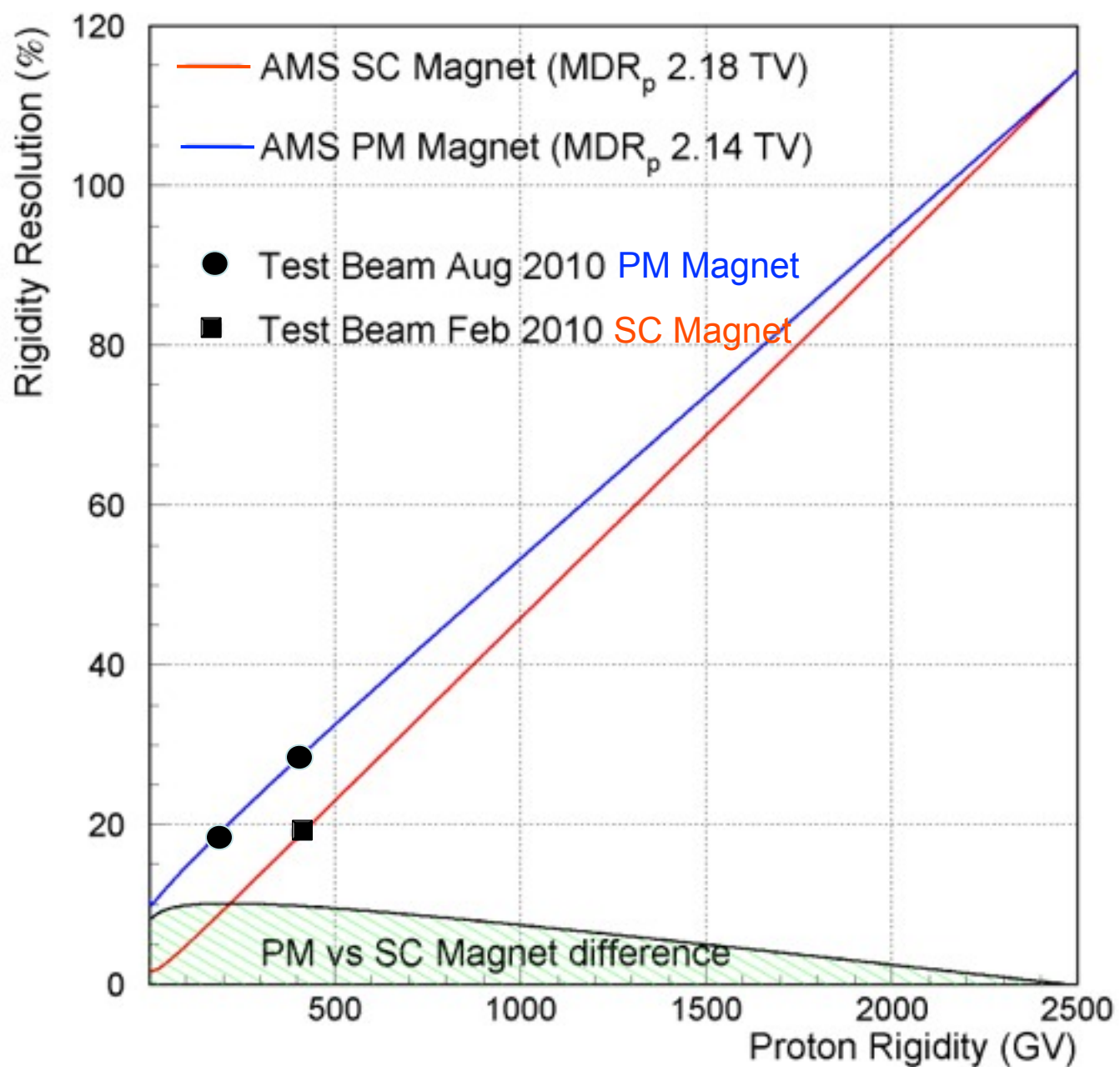
Test Beam Results– 8-20 Aug 2010



Measured combined rejection power at 400 GeV: $e^+/p = 10^{-6}$

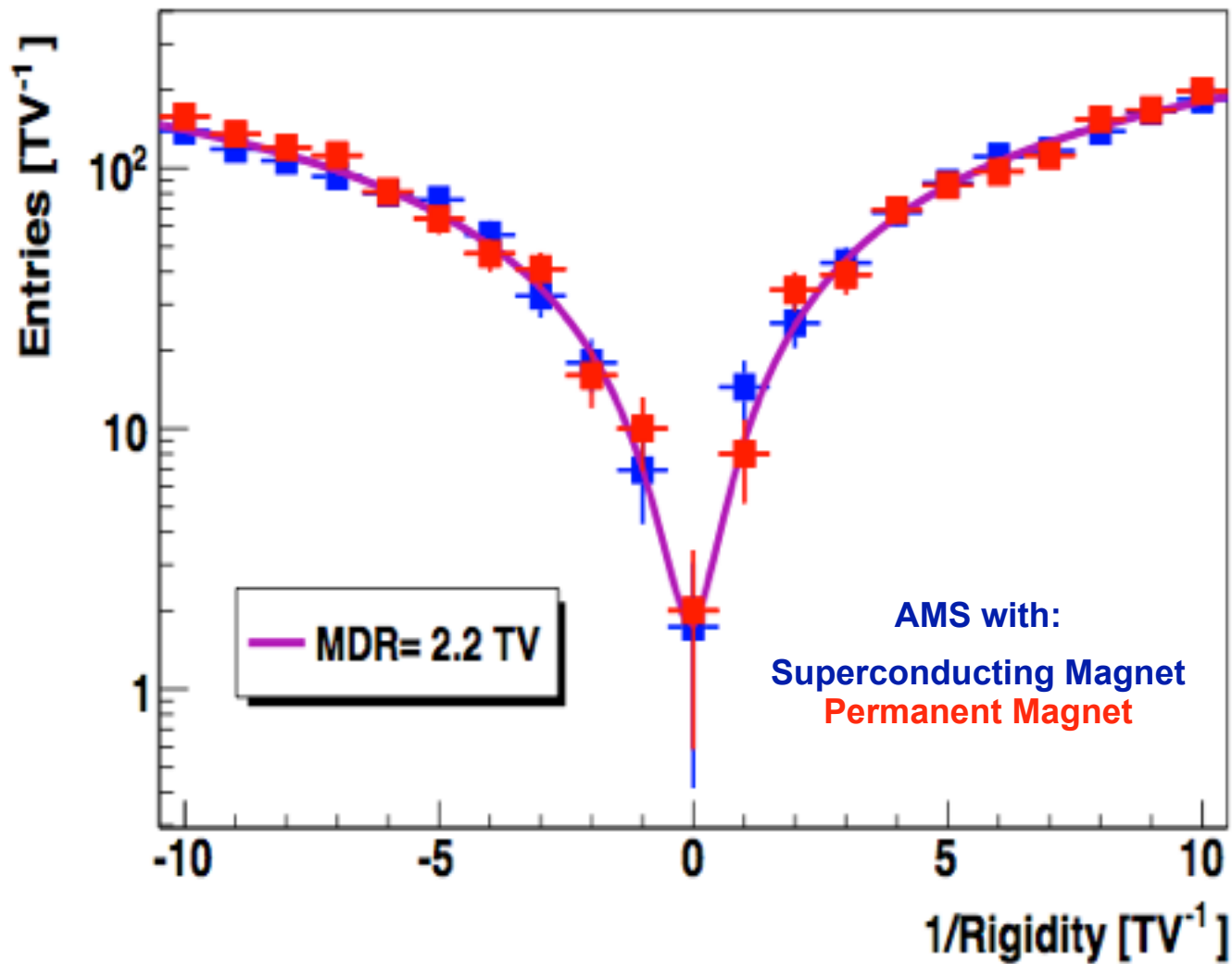
Test Beam Results of detector with **superconducting magnet**





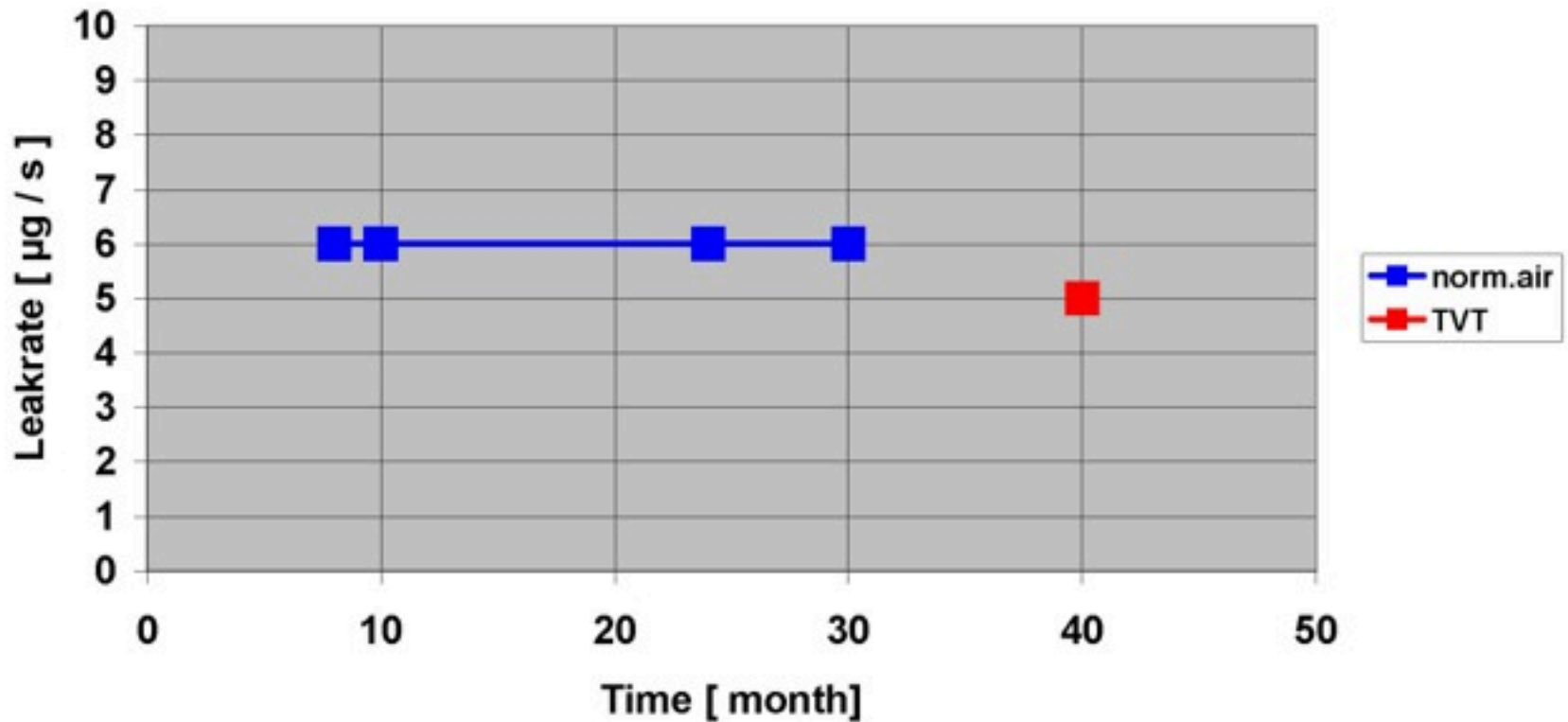
With 9 tracker planes, the resolution of AMS with the permanent magnet is equal (to 10%) to that of the superconducting magnet.

Rigidity resolution



Lifetime

A. The lifetime of consumables – TRD Leak rate



Caused by CO₂ Diffusion

CO₂ Storage at Launch: 5 kg

Leakrate of 5 µg / s corresponds to a

TRD Lifetime of 30 Years

AMS-02 data flow

AMS Data/MC Volumes Projected

DATA

Per Year Of Operation:

- 1.6×10^{10} Events
- 35 TB Raw Events
- 130 TB Reconstructed Ev.

MC

Per Year Of Operation:

- $\sim 2 \times 10^{10}$ Simulated Events
- ~ 200 TB Simulated Data Volume