

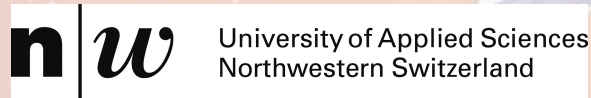
# The Spectrometer Telescope for Imaging X-Rays on-board the ESA Solar Orbiter satellite

Oliver Grimm  
Institute for Particle Physics, ETH Zürich

— for the STIX collaboration —

13<sup>th</sup> Vienna Conference on Instrumentation

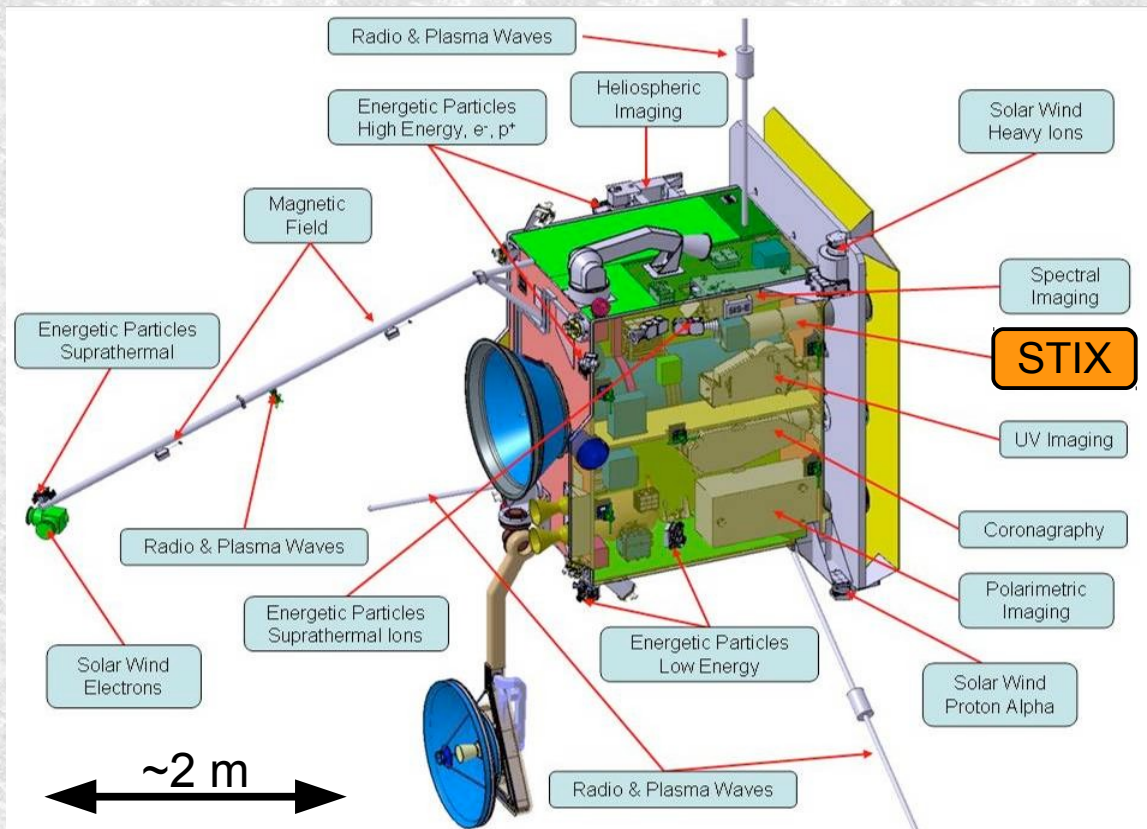
12 February 2013, Vienna



# ESA Solar Orbiter

“How does the Sun create and control the heliosphere?”

- Sun-heliosphere interaction
- Energetic solar phenomena
- Solar transients, heliospheric variability
- Solar wind accelerating mechanisms
- Solar wind plasma, coronal magnetic fields
- Solar dynamo working principle



**10 instruments**  
(remote-sensing and in-situ)

Mass **1.8 t** (payload 180 kg)  
Power **180 W**  
Telemetry **150 kbps** (@ 1 AU)

Launch January **2017**  
Mission duration **4+3 years**

Resonance orbit with Venus

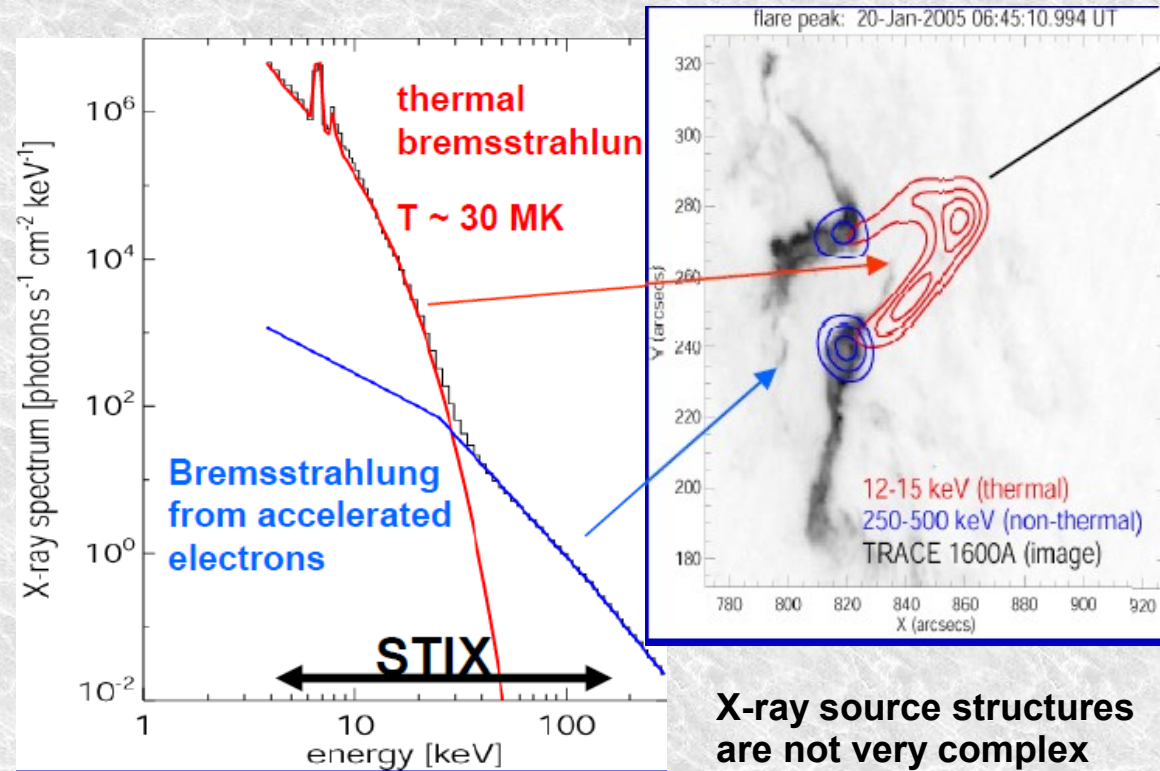
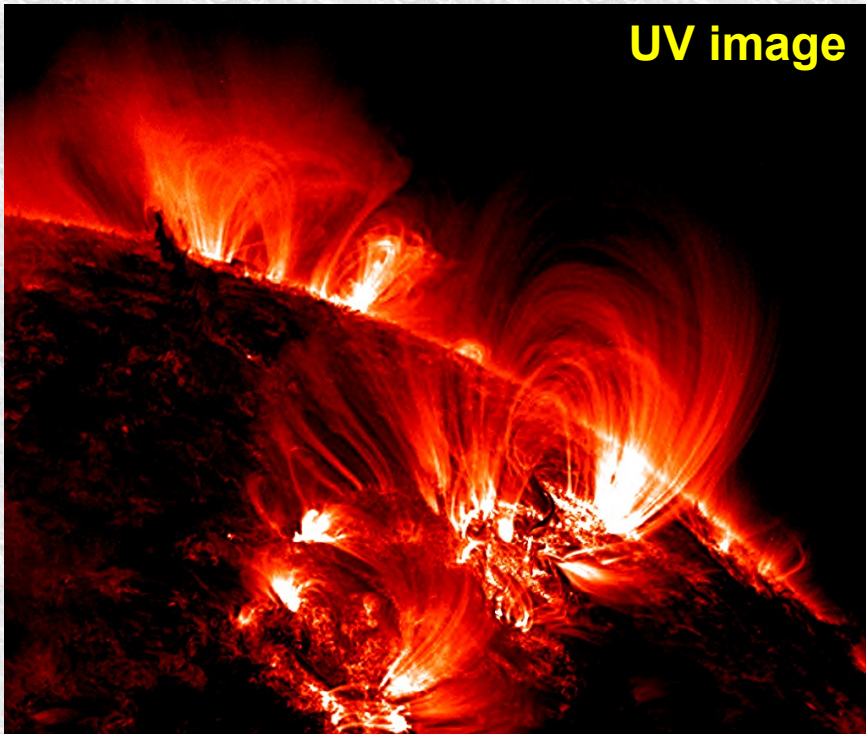
Currently in realization phase C

# STIX Science Goal

Imaging 4-150 keV x-rays

determines intensity, spectrum, timing and location of accelerated electrons near the Sun.

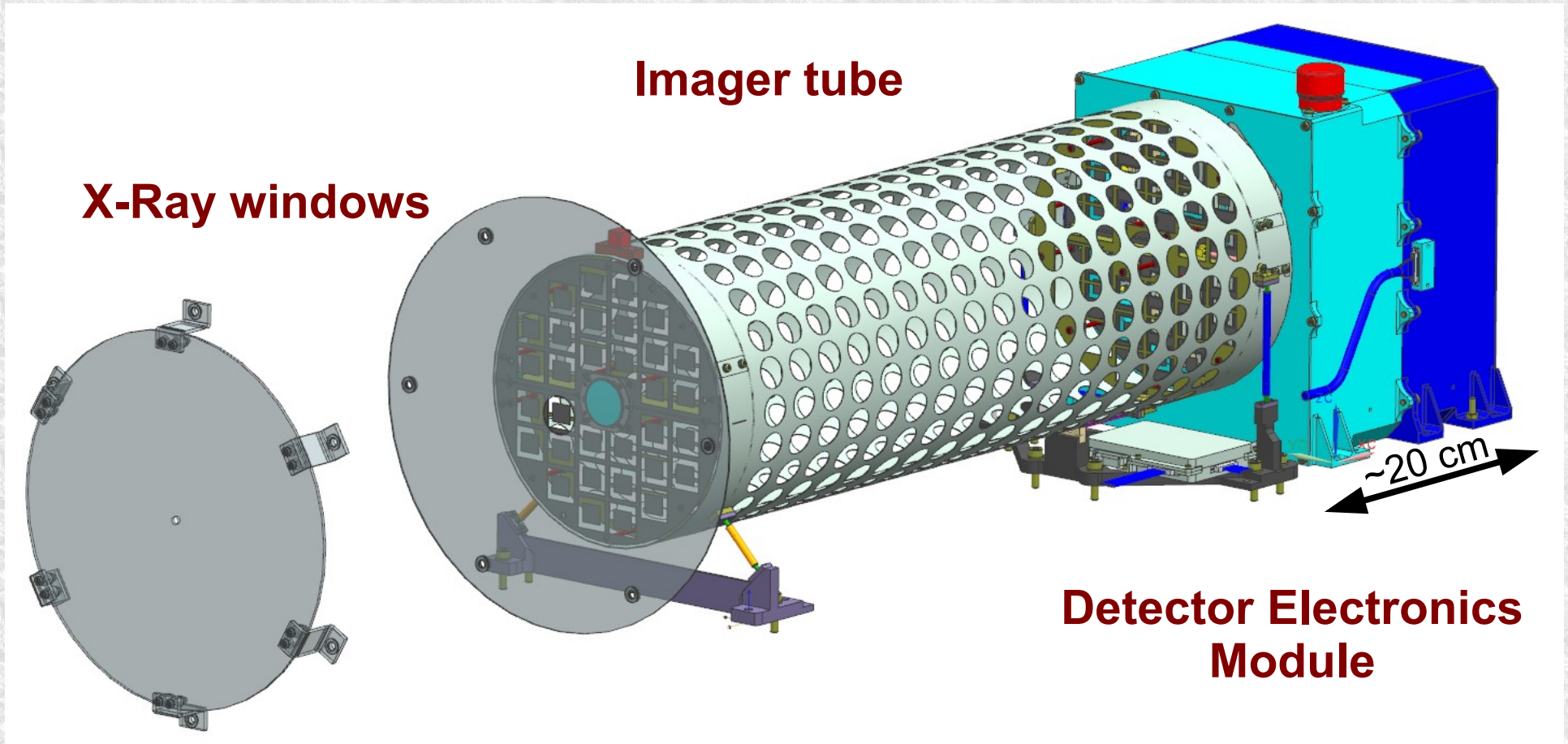
- Study
- acceleration mechanism of electrons at the Sun
  - electron transport into interplanetary space



# STIX Design

Instrument allocation is **4 W** power, **7 kg** mass and **700 bits/s** telemetry.

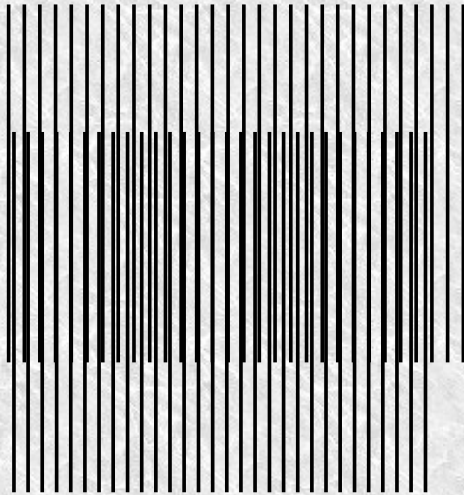
→ only indirect Fourier imaging feasible at X-ray energies for required parameters



Energy range      **4-150 keV**  
Energy resolution   **1 keV (FWHM @5 keV)**  
Effective area      **6 cm<sup>2</sup>**

Angular resolution      **7 arcsec**  
Pointing accuracy      **4 arcsec**  
Field of view          **2°**  
Time resolution        **0.1s (statistics limited)**

# STIX imaging: Basic principle

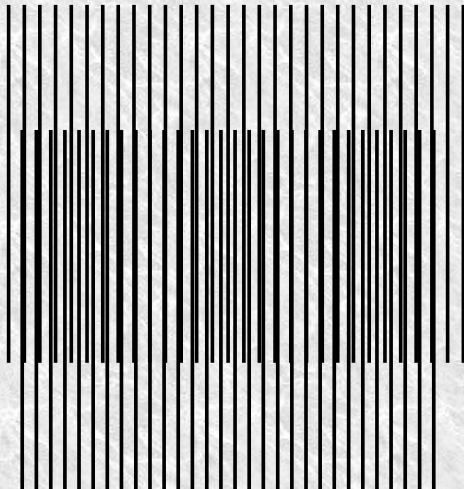


2 grids with slightly different pitch

Illuminated by point source at large distance

→ **Moire transmission pattern**

Large-scale Moire structure encodes direction to point source



One grid shifted by  $\frac{1}{2}$  period

→ **Large-scale pattern moved by  $\frac{1}{2}$  period**

Shift equivalent to source moved in angle by  $\frac{\frac{1}{2} \text{ grid pitch}}{\text{grid separation}}$

→ **Coarsely pixelized detector sufficient for high angular resolution**  
(if fine grids with large separation used)

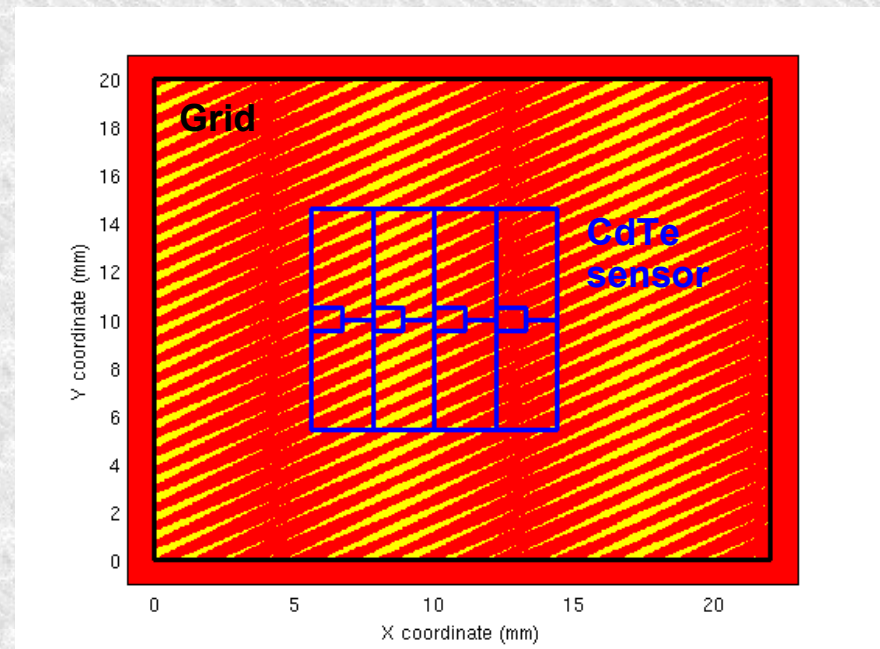
**Number of grid pairs** determines allowable **source complexity**

Slightly different angles on grids generates Moire pattern

→ orientation of Fourier component and Moire pattern decoupled

Pitch 666 / 690  $\mu\text{m}$   
Angle  $60^\circ$  /  $64^\circ$   
Vertically impinging x-rays

→ Large scale Moire pattern period 10 mm

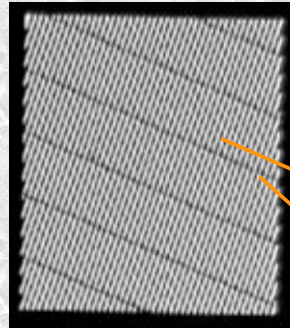


**Pixel count rate differences encode source direction**

# Imaging in STIX

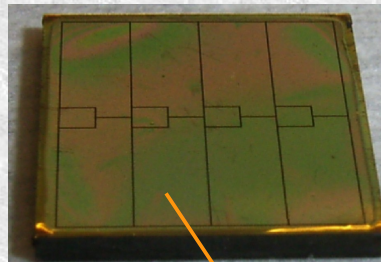
## 32 Tungsten grid-pairs

Pitch 38  $\mu\text{m}$  – 1 mm  
400  $\mu\text{m}$  thick  
30 Fourier components in three directions  
2 special counters



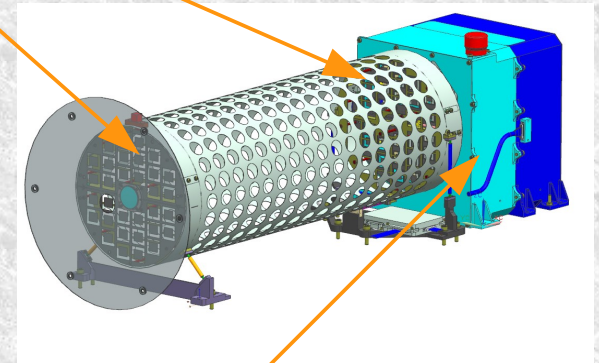
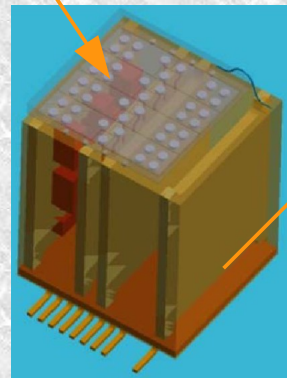
## 32 CdTe semiconductor sensors

10x10x1 mm<sup>3</sup>  
Sensors from Acrorad (Japan)  
Patterning and testing at PSI, Switzerland



## 32 Caliste-SO hybrids

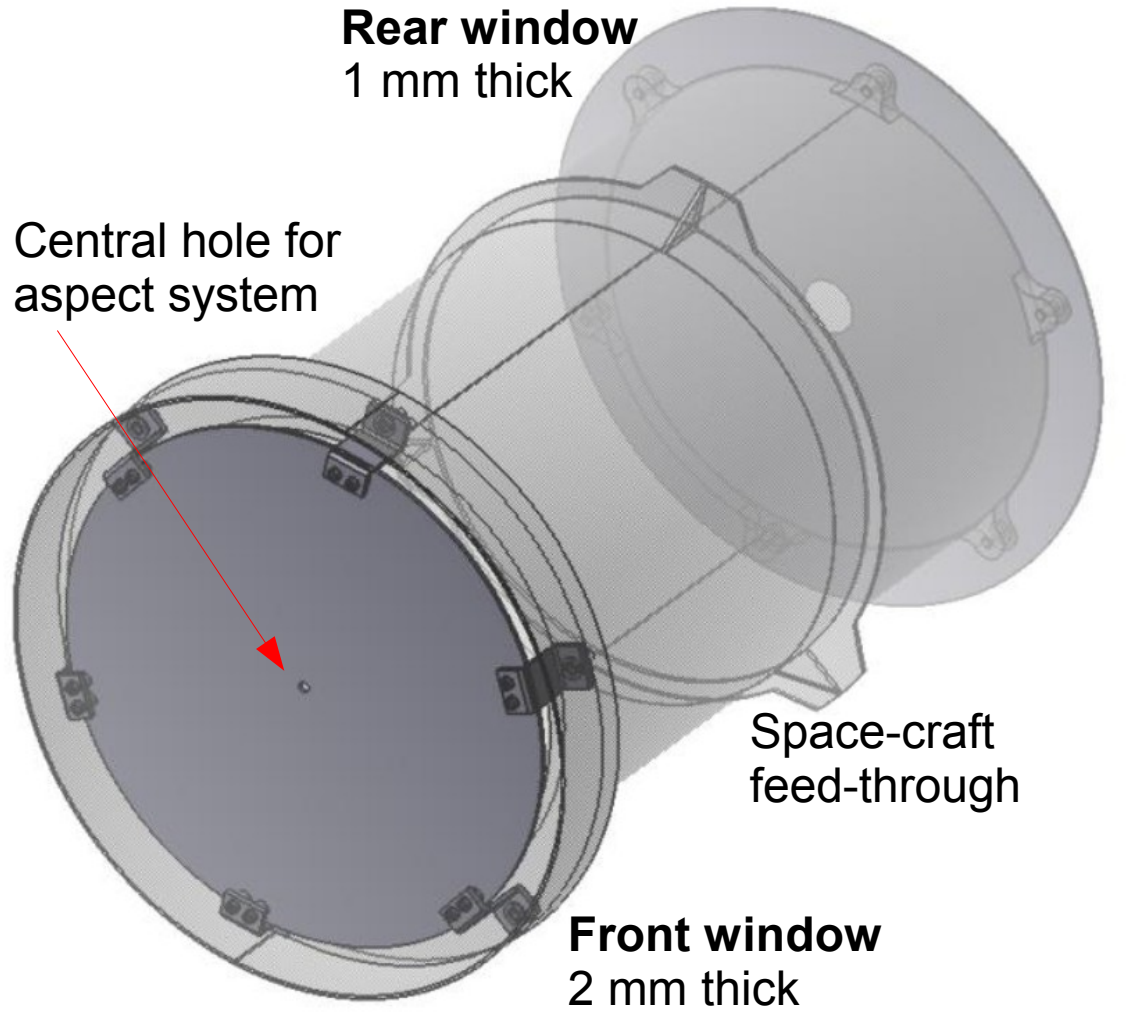
ASIC, voltage filtering and bias routing  
ASIC and hybrid developed at CEA Saclay, France



# Beryllium X-ray windows

**Rear window**  
1 mm thick

Central hole for  
aspect system



Space-craft  
feed-through

**Front window**  
2 mm thick

Al-SiO<sub>x</sub> coating

Redundant against failure of  
one window (e.g. micrometeorite)

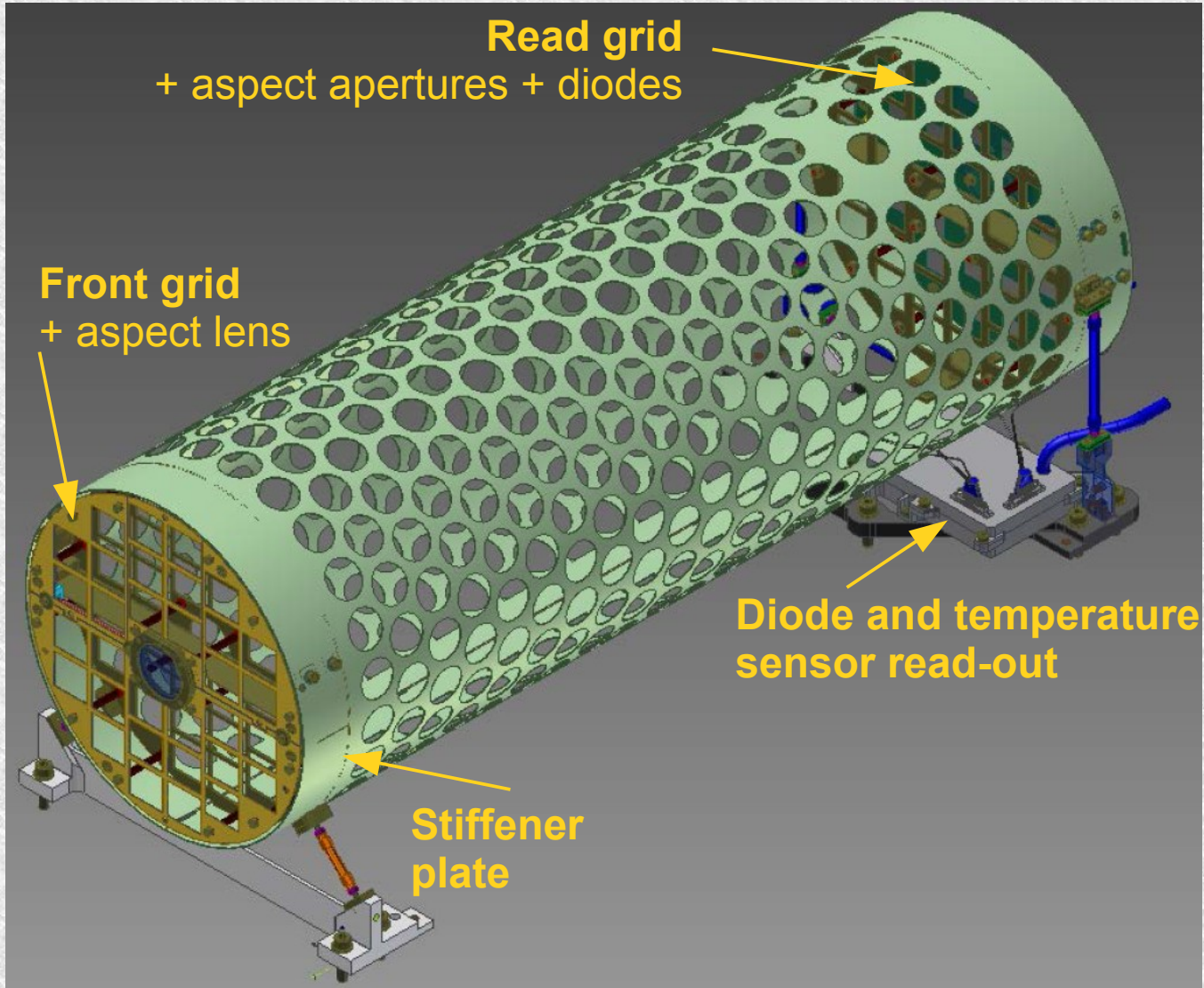
Mass 360 g (without feedthrough)

Temperatures at perihelion

- 300°C front
- 180°C rear



# Imager



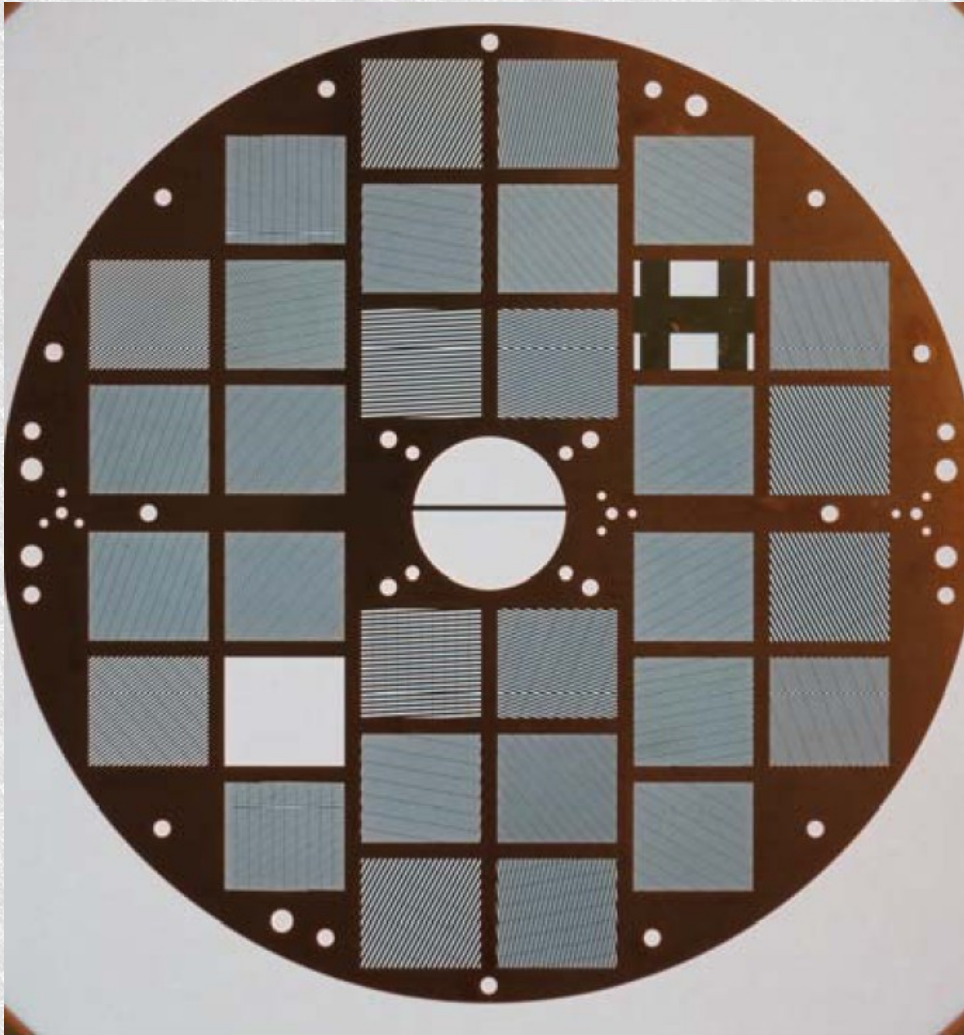
Tube supports front and rear grid assembly

Prevents twisting of grids

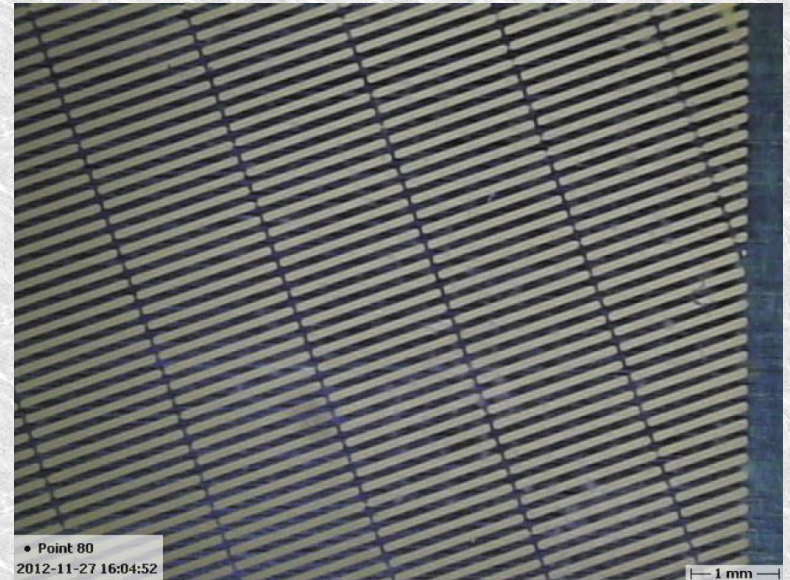
Mass ~1.6 kg

Aspect system to establish line of sight

# 400 $\mu\text{m}$ thick Tungsten grids



Front grid copper dummy



32 subcollimators  
Pitch 38  $\mu\text{m}$  to 1 mm

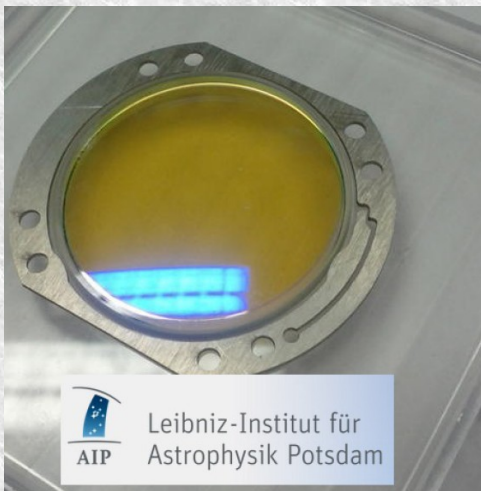
Mounting & alignment apertures

Produced from etched and  
stacked Tungsten foils

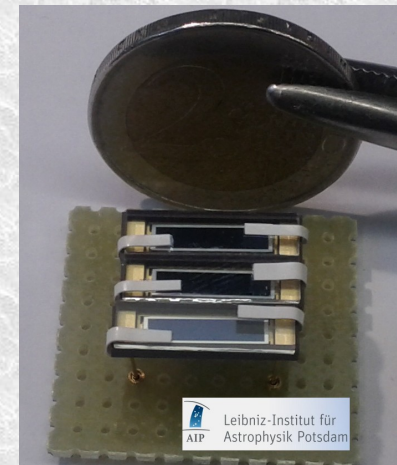
# Aspect system

Lens in front grid casts solar image on rear grid

Linear photodiodes illuminated through holes in rear grid



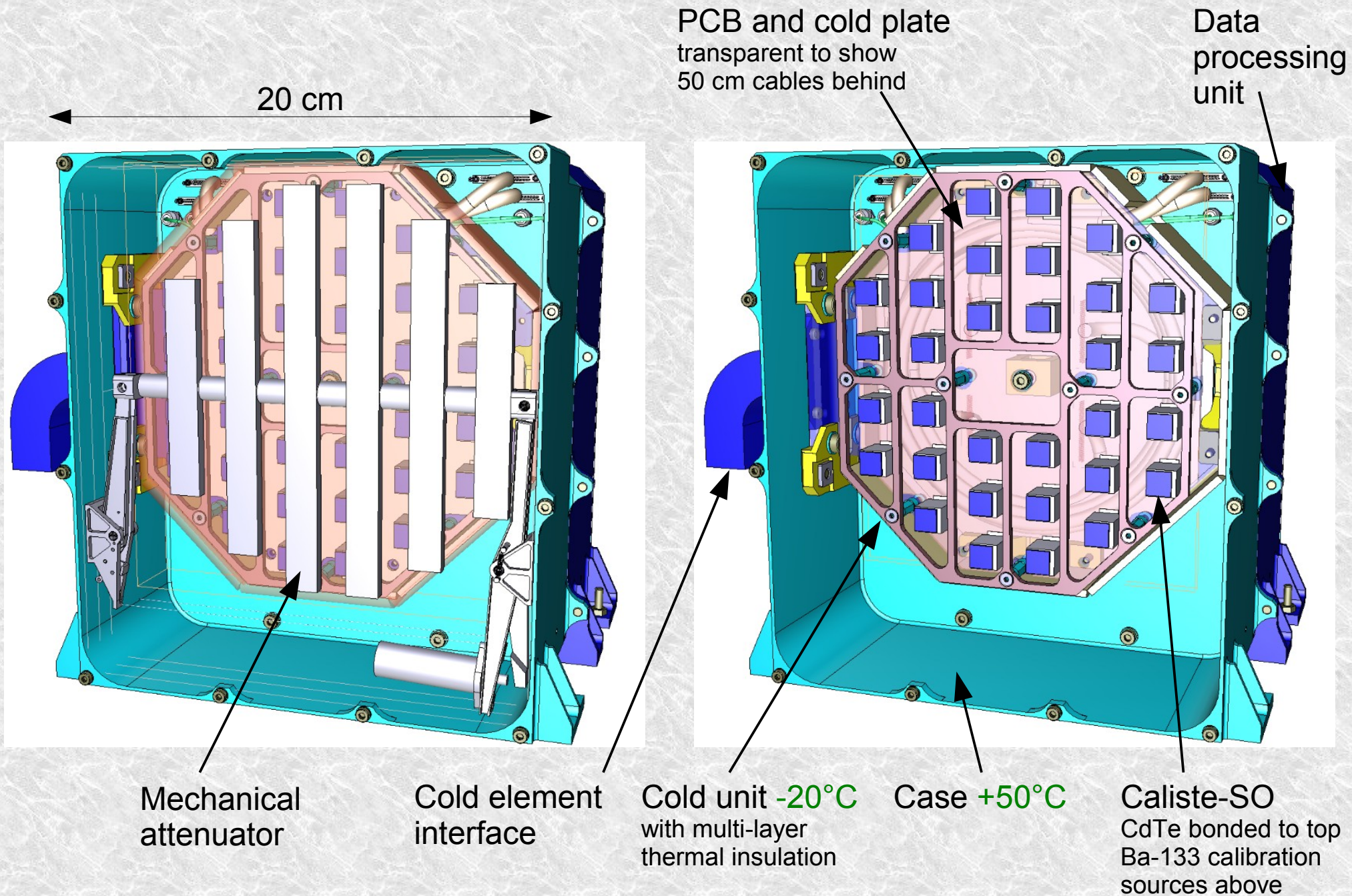
- **Step-wise change in diode signal** when solar limb crosses hole
- **Signal decrease due to diode aging** not significant for limb detection
- **3 points define solar disk**  
4<sup>th</sup> point for redundancy/error determination



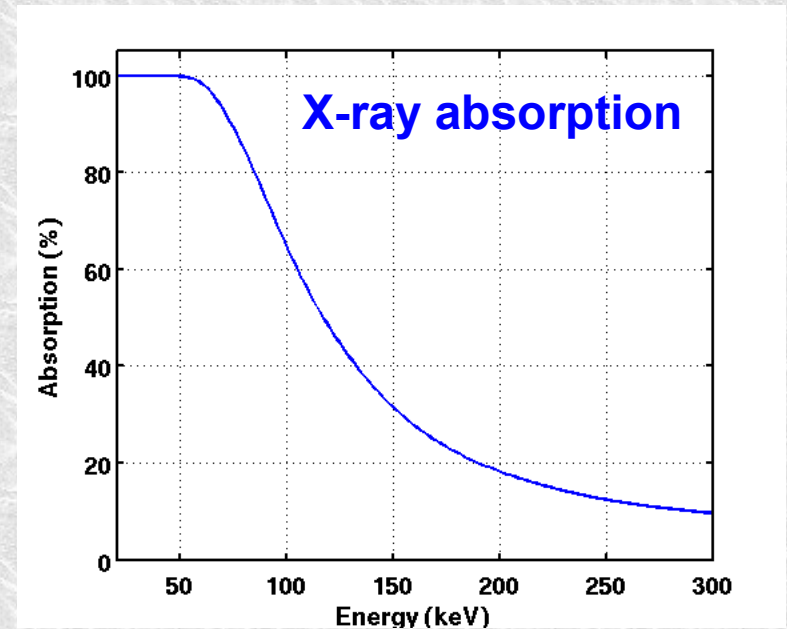
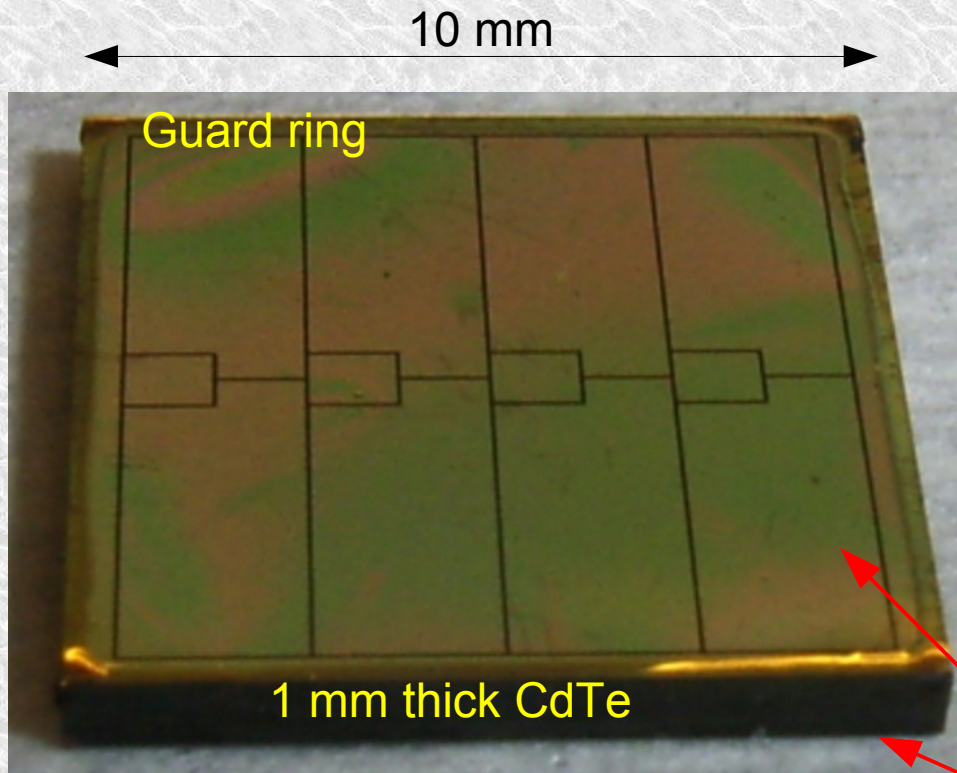
**Imager line-of-sight determined to 4 arcsec**

**→ X-ray images can be accurately placed on Sun**

# Detector Electronics Module (DEM)



# CdTe semiconductor sensors



Pixilized Au-Ti-Al Schottky electrode  
(connected to ASIC inputs)

Monolithic Platinum Electrode  
(-300 V)

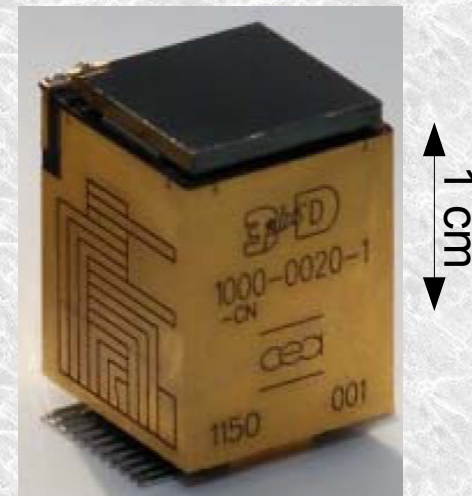
Pair creation energy	4.43 eV (870 pairs at 4 keV)
Leakage current (large pixel)	<60 pA (-20°C, 300 V)
Charge absorption length	10 cm / 0.6 cm (electron/hole)

Non-ionizing dose (NIEL) will degrade charge collection over mission duration  
Total ionizing dose of 30 krad otherwise not severe

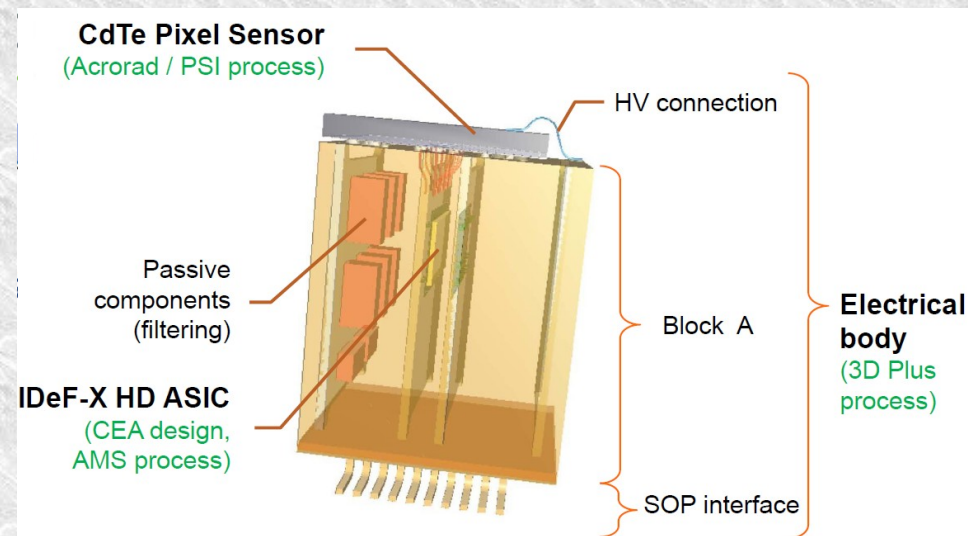
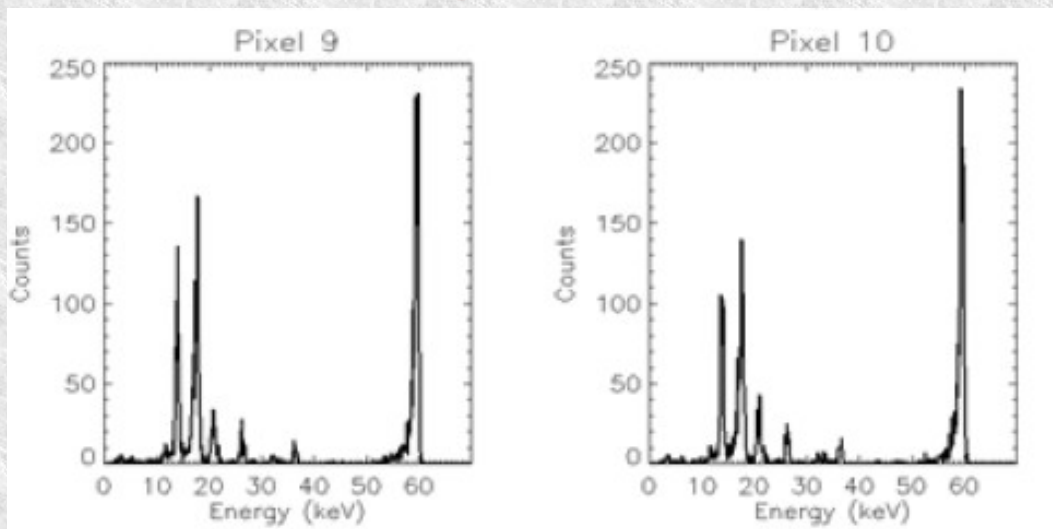
# Caliste-SO read-out hybrid / ASIC

Caliste-SO hybrid contains CdTe bonded to top  
IDeF-X HD ASIC  
bias routing, filtering

ASIC 32 charge-sensitive amplifiers, ENC  $\sim 80 e^-$   
multiplexed readout ( $\sim 6.6 \mu s$  per hit @ 20 MHz clock)  
 $\sim 1$  mW/active channel



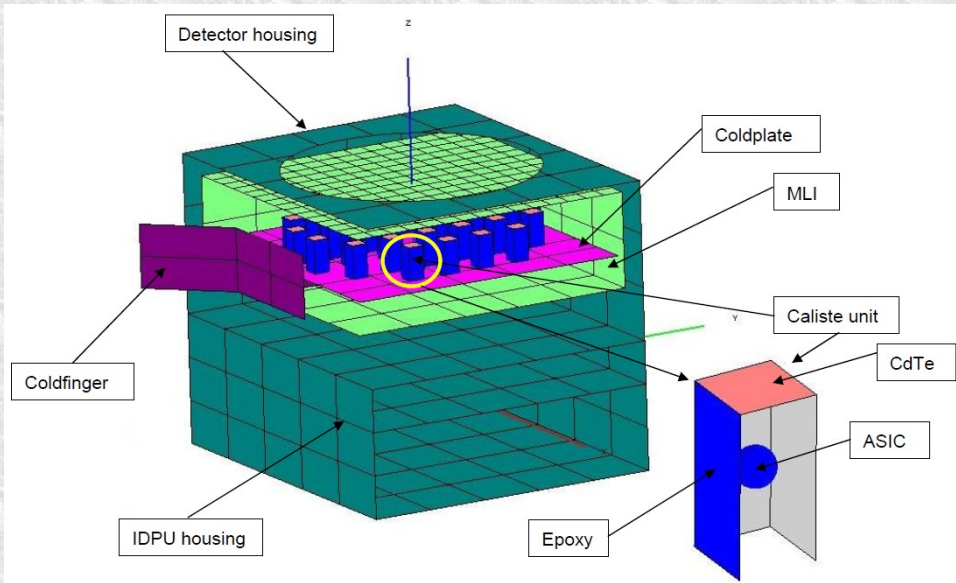
## Americium-241 spectrum



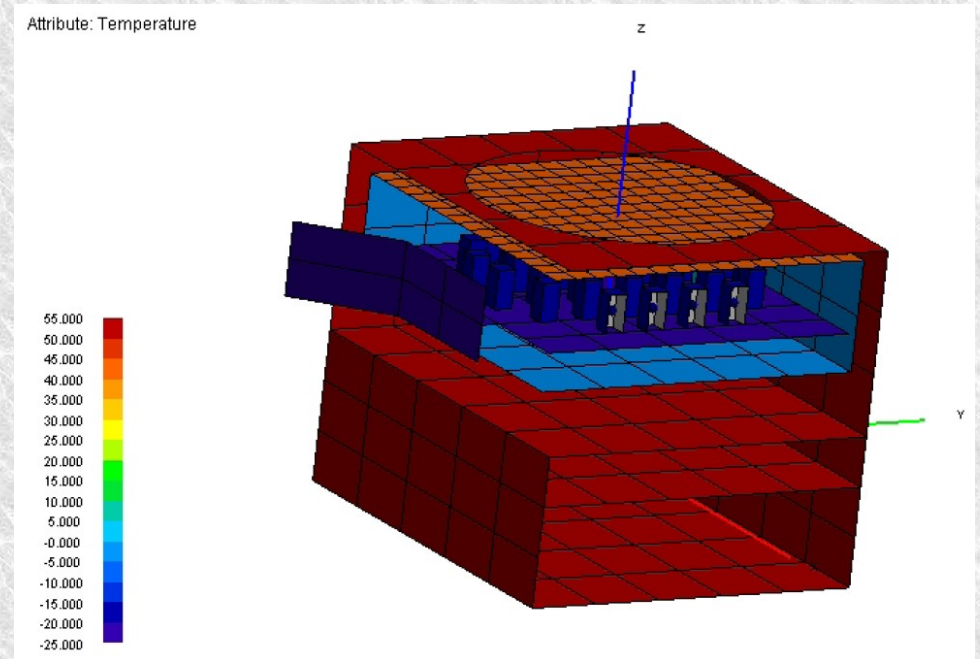
Resolution FWHM  $< 1.2$  keV @ 14 keV

# Thermal simulation

## ESATAN model



## Temperatures (hot operational case)



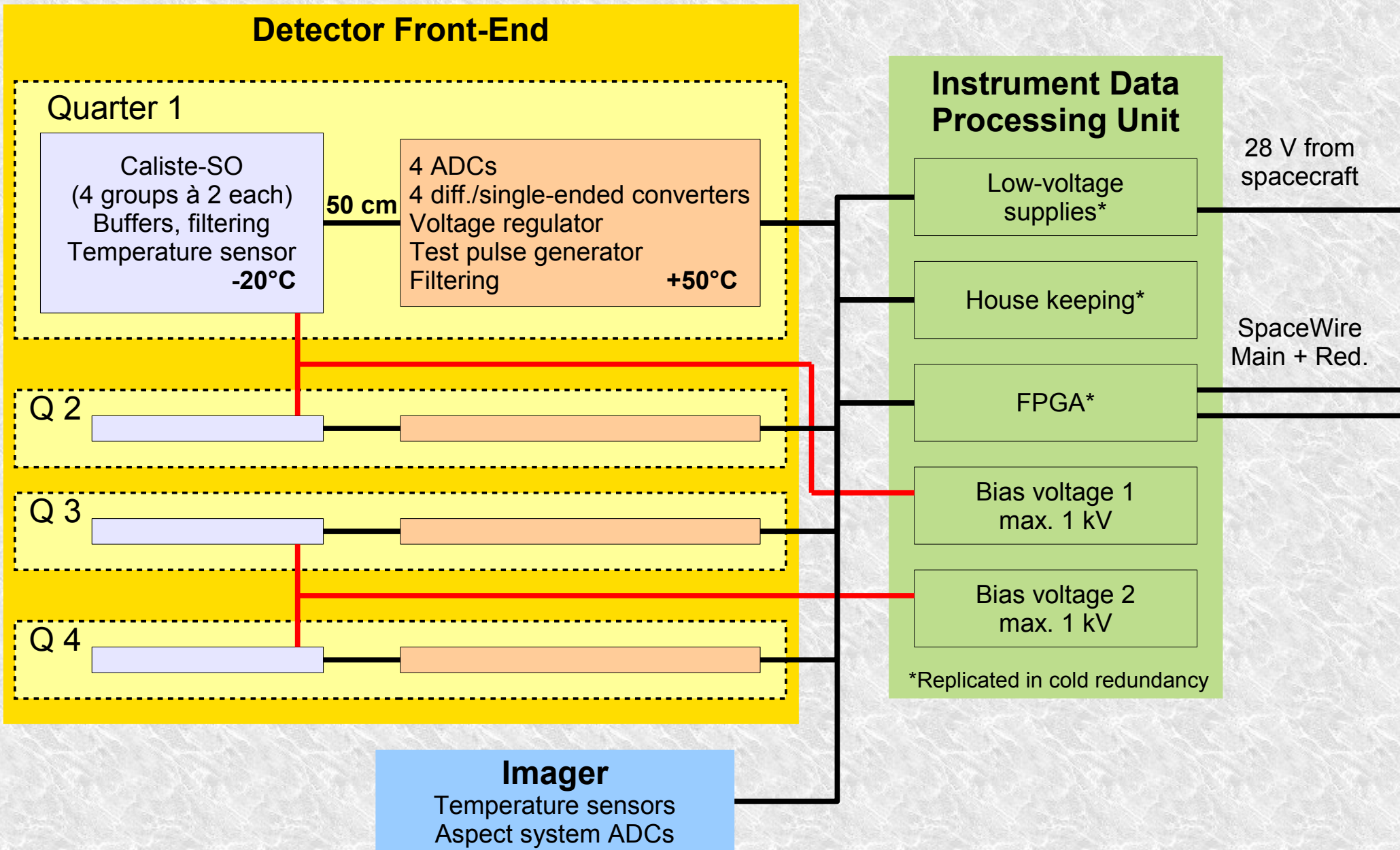
## Test unit



Model predicts heat flux to cold element of **2.5 W**

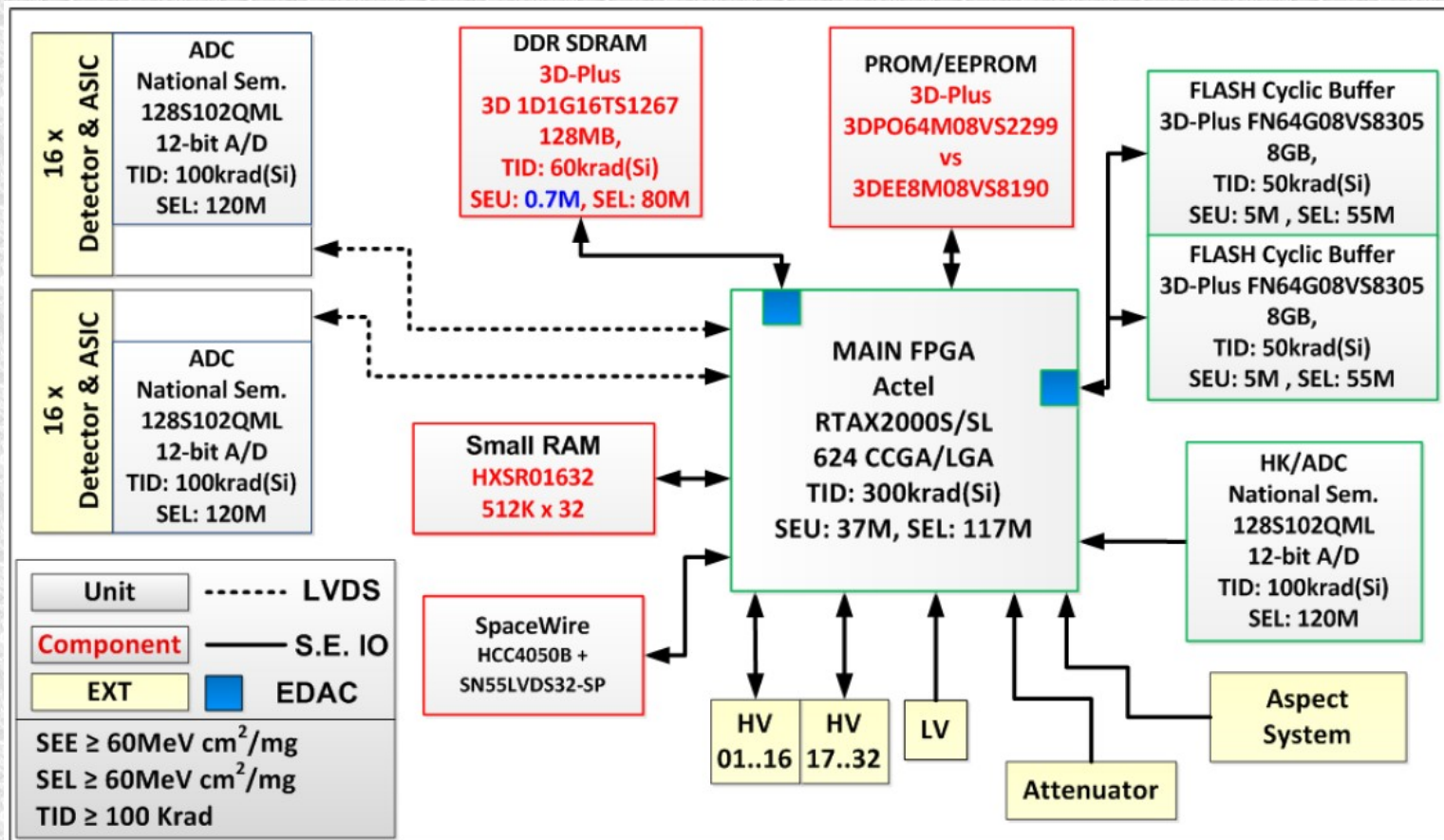
- above current space-craft allocation
- under study with a bread board

# Simplified STIX block diagram





# Instrument Data Processing Unit



Event rates up to  $10^5 \text{ s}^{-1}$  during strong Solar flares (all Caliste simultaneously)

Autonomous operation up to 80 days

Several month of science data can be stored on-board

→ provides telemetry flexibility by allowing off line data selection and downlinking

# Outlook

STIX currently in realization phase C

ESA critical design review in 2013

Industrial prime contractor will oversee instrument construction in phase D

Flight instrument delivery to ESA January 2015

Launch January 2017

# Extra slides

# In-flight energy calibration

**ADC count to x-ray energy has to be established over 10-year mission duration**

CdTe charge collection degrades due to displacement damage from protons

Changing leakage current will alter response of ASIC

ASIC gain / offset have small dependence on temperature

## Requirements on calibration

Steep spectra → 100 eV<sub>rms</sub> precision for each pixel

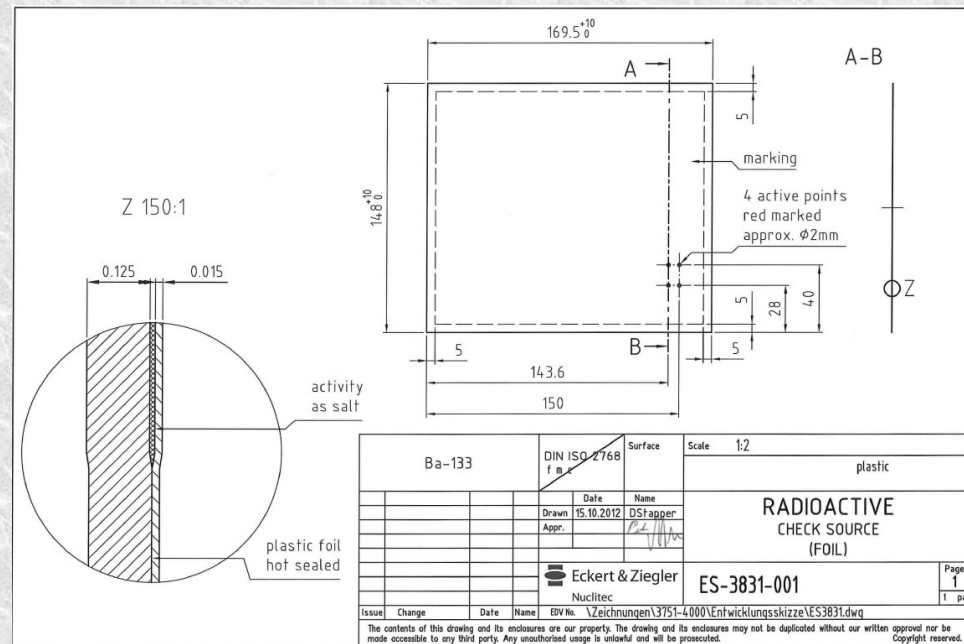
Low background → Source strength must be low

## Technical realization

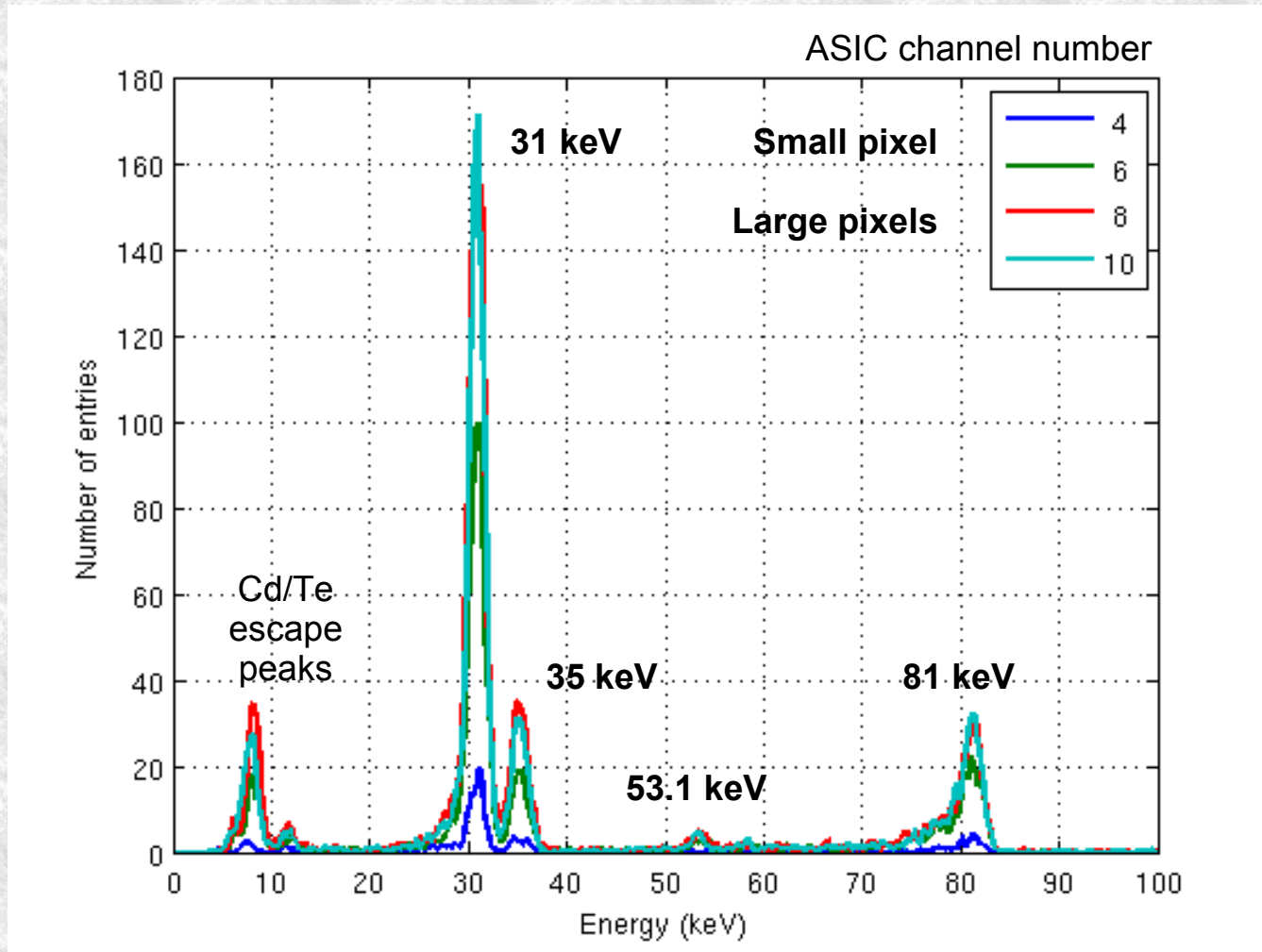
Barium-133 ( $t_{1/2}=10.5$  years)

Ø1 mm dots (<1 Bq activity)  
wedged between plastic foils

Mounted on support of  
multilayer insulation

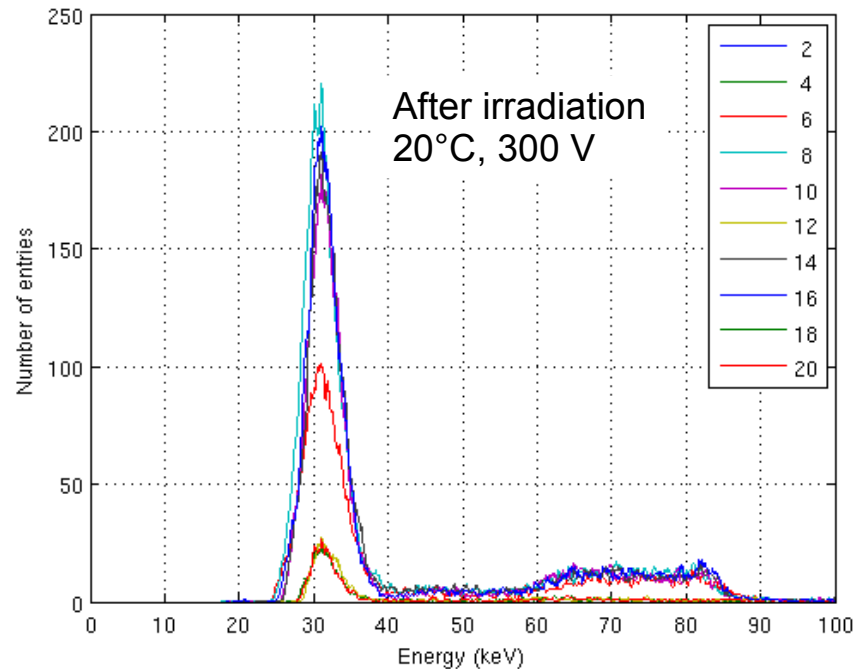
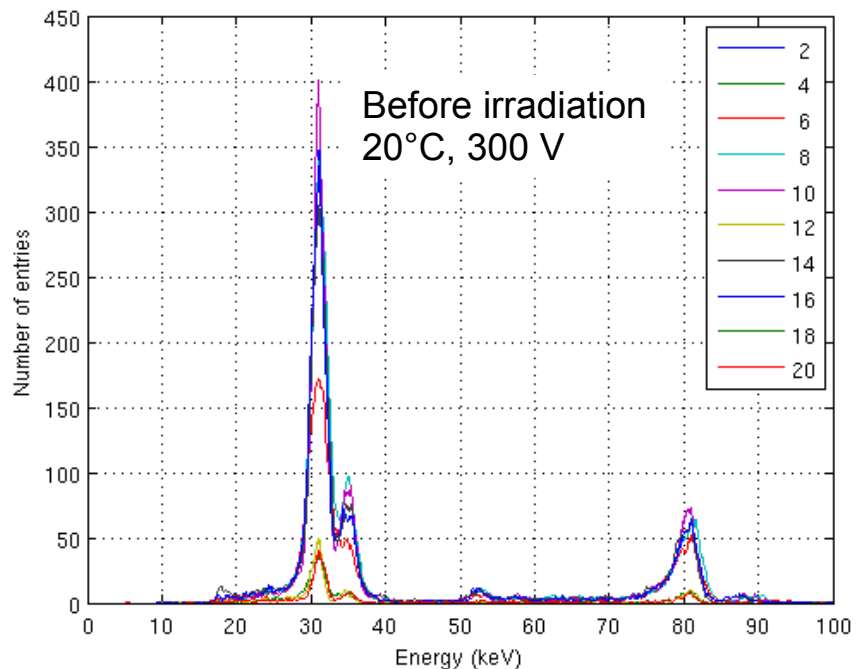


# Barium-133 spectrum



Bias **300 V**, temperature **-20°C**

# Radiation damage of CdTe



200 MeV protons

Fluence  $3.6 \cdot 10^{10}$  p/cm<sup>2</sup>  
approx. 1.4 krad  
STIX worst-case ~20 krad

Duration 10 hours

<sup>133</sup>Ba spectra

