## LOBSTER - New Space Xray Telescopes

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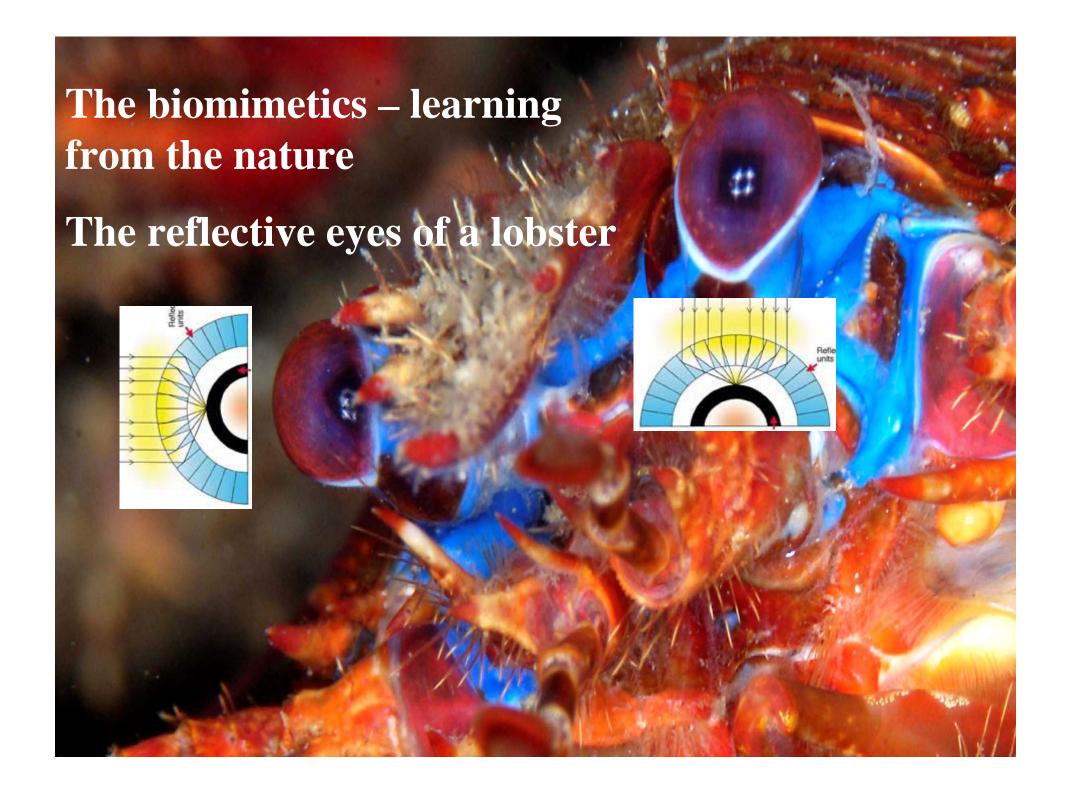
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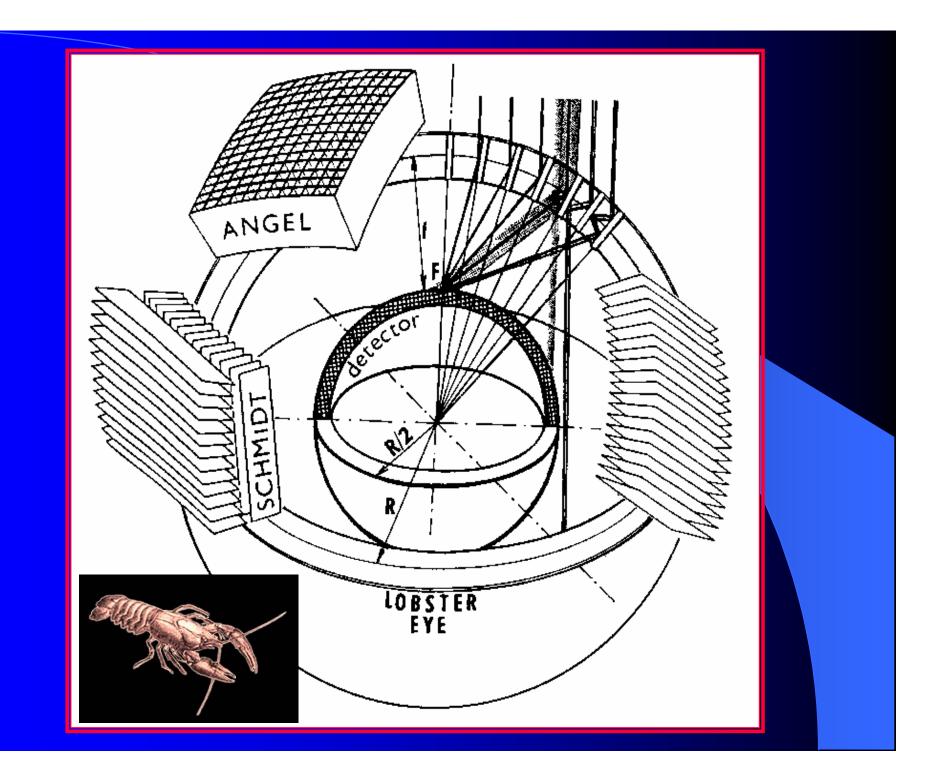
Space Part 06, Beijing, China, April 21, 2006

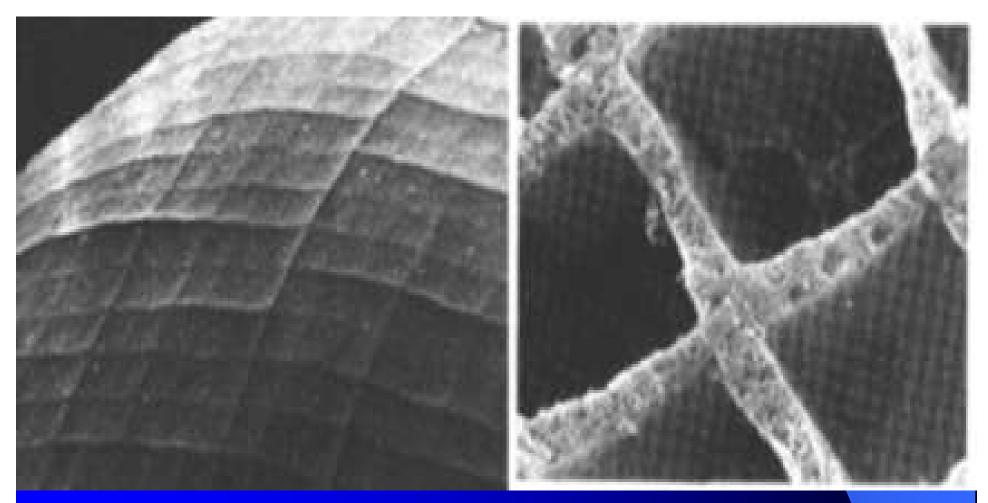


## Lobster-Eye (LE)

- Novel Wide Field X-ray Telescopes
- easily possible (classical X-ray optics only 1 deg or less)
  - Analogy with lobster eyes
- Designed for astronomy, but
   laboratory applications also possible

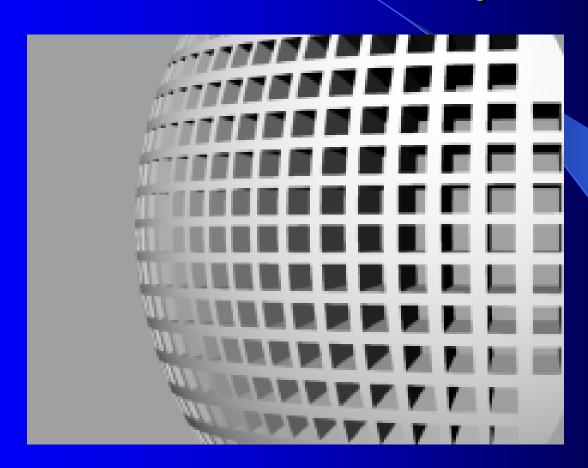






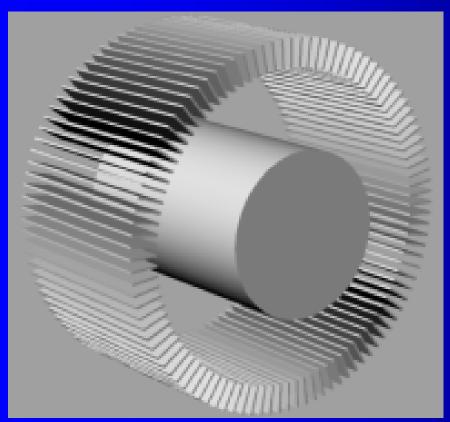
The eye of a lobster, viewed with a microscope. Right: closeup of a small area of the eye. The eye consists of millions of square "channels"; each channel measures approximately 20 microns (or two hundredths of a millimetre) across..

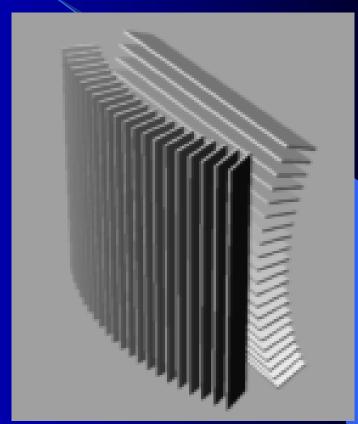
### Lobster Concept



Angel arrangement

## Lobster Concept





1D Schmidt

2D Schmidt

# LE International Space Station (ISS) like design

**MCP** design

(microchannel plate)

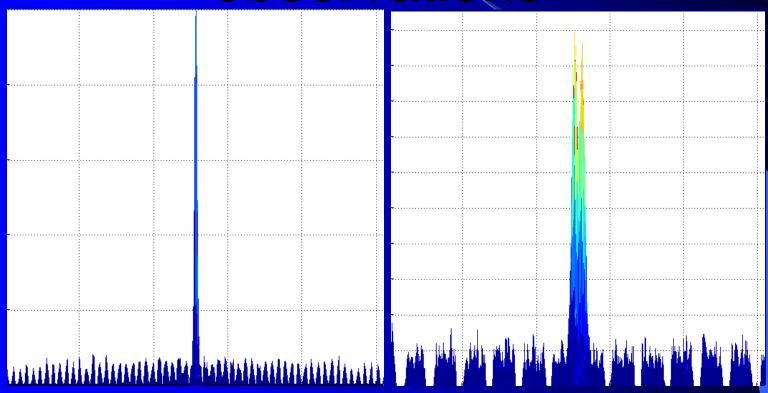
UK - Univ. of Leicester - collab.

**Angel arrangement** 



Thin foil design -**Multi Foil Optics** (MFO) (this talk) **CZ-collaboration** prototypes already available and tested high gain ~ 600 better energy range up to 10 keV **Schmidt arrangement** 

# Simulated pointed observations

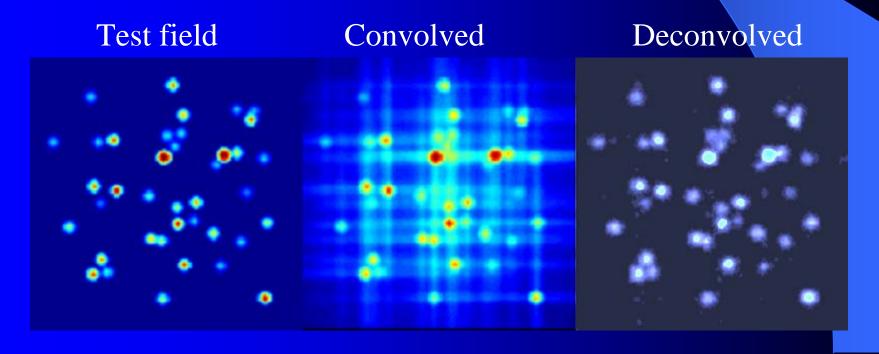


PSF – 5 keV photons, golden layered plates, 78 mm x 11.5 mm x 100 μm plates, 300 μm spacing

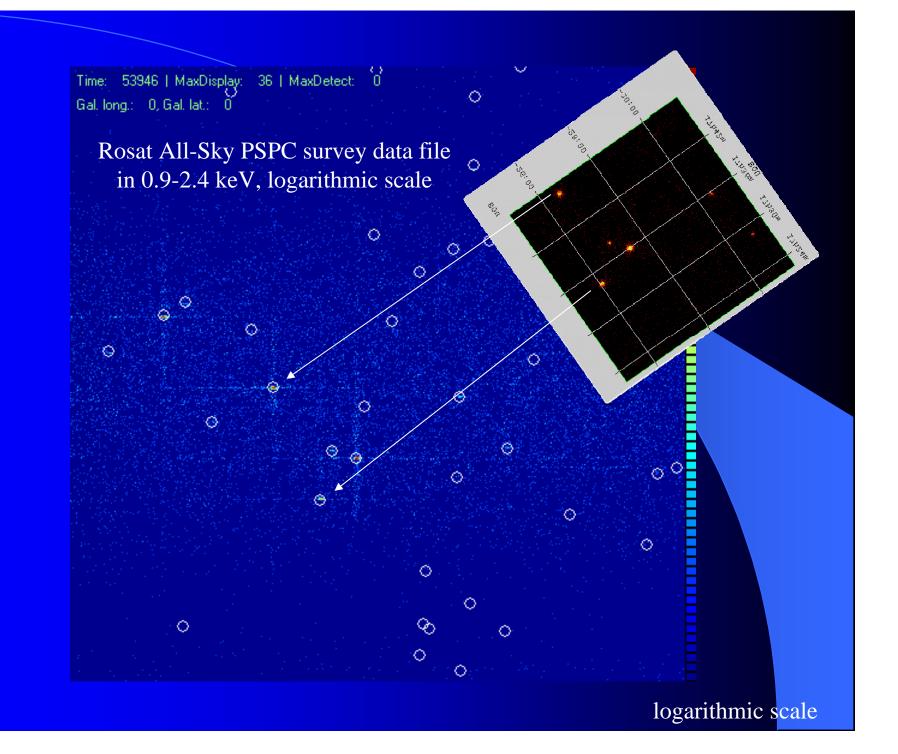


# Simulated pointed observations

- for low energies (cross faint relative to the central peak) standard deconvolution techniques can be used successfully
- for strong cross structures PSF subtraction was successfully used
- 5 keV PSF & PSF subtraction example at the figure below



0 | MaxDisplay: 0 | MaxDetect: 0 Time: Gal. long.: -0, Gal. lat.: 0 logarithmic scale



# Simulated <u>scanned</u> observations

- Data taken from The ROSAT All-Sky Survey Bright Source Catalogue (1RXS)
- Source count rate scaled to get [ct s<sup>-1</sup> cm<sup>-2</sup>] by the ROSAT effective area
- Galactic plane scans, 105 min / revolution
- 78 mm x 11.5 mm x 100 μm plates, 300 μm spacing

0 | MaxDisplay: 0 | MaxDetect: 0 Time: Gal. long.: 0, Gal. lat.: 0 logarithmic scale

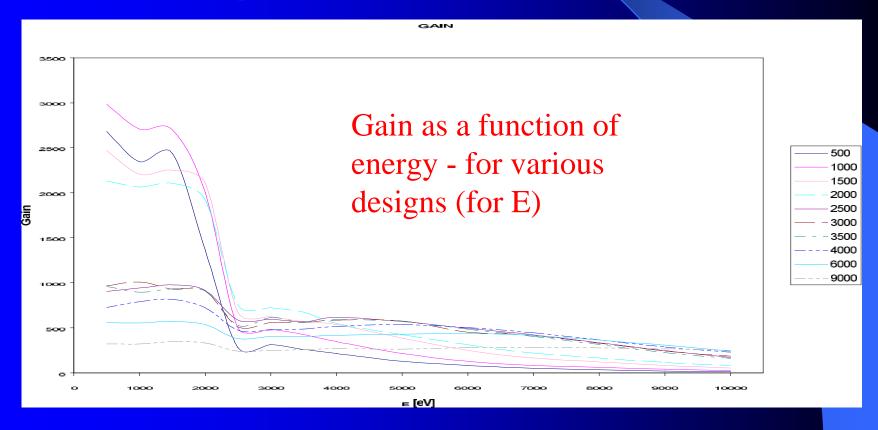


### Gain

number of photons detected inside the FWHM circle with the LE

$$G =$$

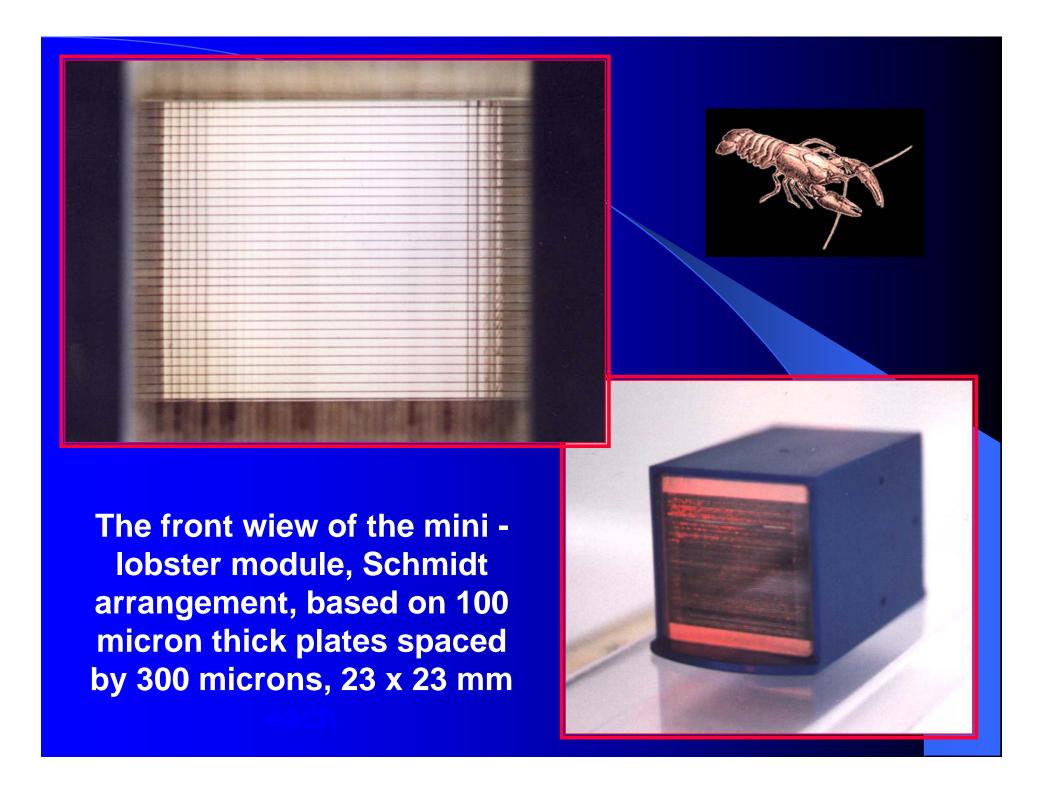
number of photons detected inside the FWHM circle without the LE



## X-ray Optics LE prototypes (Schmidt geometry) developed and tested so far

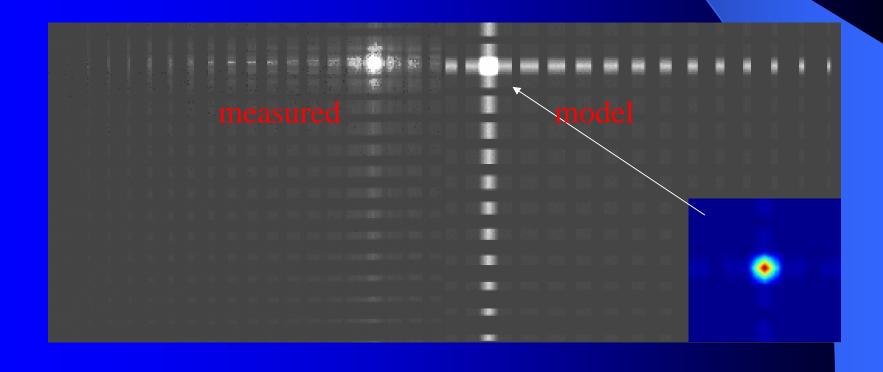
MODULE	size	plate	distance	length	eff.	focal	resolution	field	energy
		thickness			angle	length		of view	
	d(mm)	t(mm)	a(mm)	I(mm)	a/l	f(mm)	r(arcmin)	(°)	(keV)
macro	300	0.75	10.80	300	0.036	6000	7	16	3
middle	80	0.3	2	80	0.025	400	20	12	2
mini 1	24	0.1	0.3	30	0.01	900	2	5	5
mini 2	24	0.1	0.3	30	0.01	250	6	5	5
micro	3	0.03	0.07	14	0.005	80	4	3	10

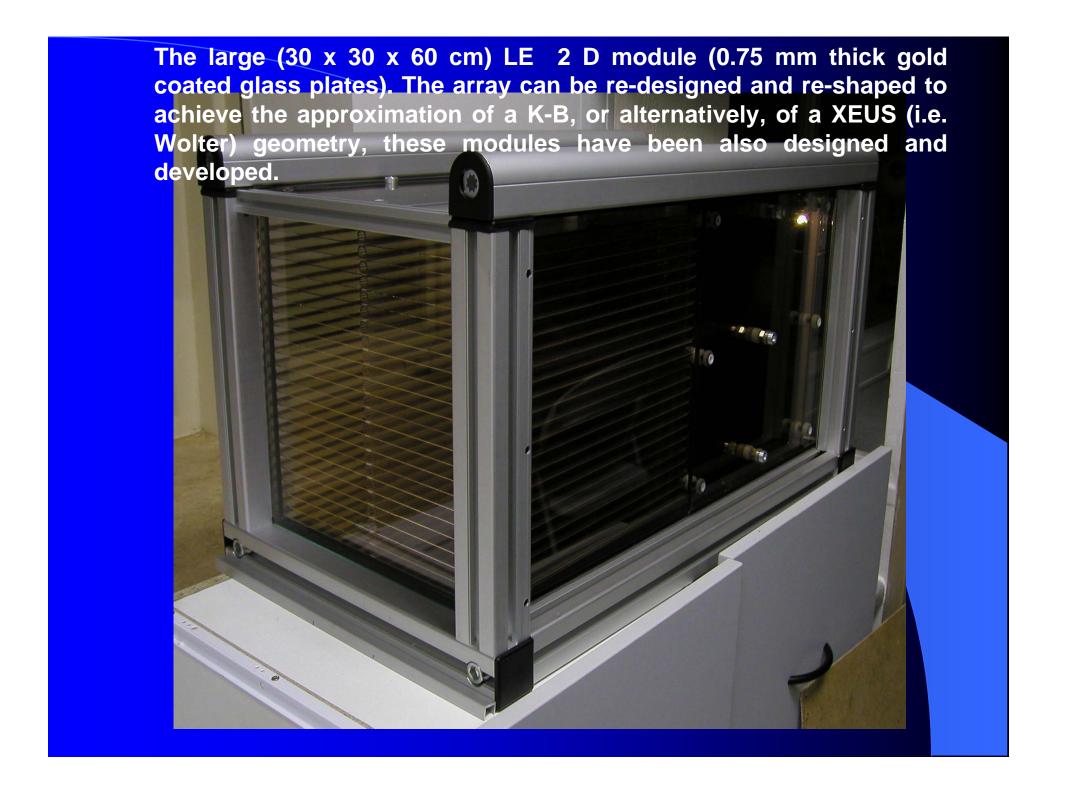
5 prototype LE telescopes have been already designed, developed, and tested

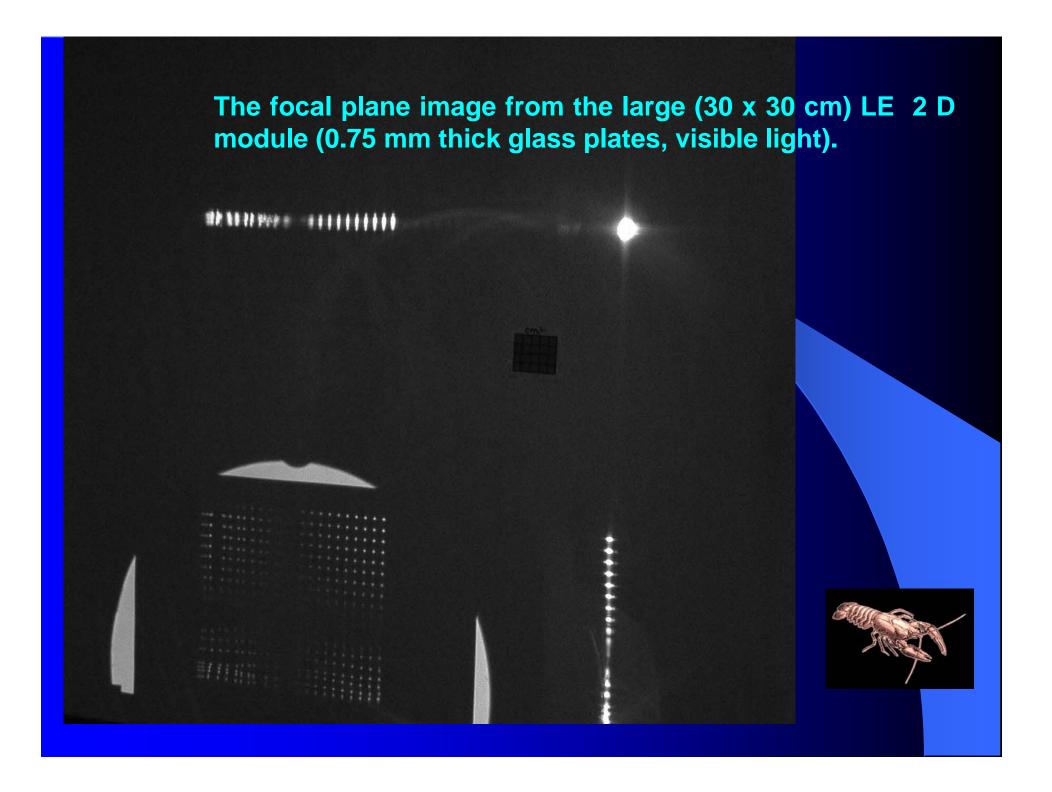


# X-ray experiment vs. simulation

- Point-to-point focusing system, LE Schmidt mini
- source-detector distance 1.2 m, 8 keV photons
- image width: 2x512 pixels, 24 µm pixel
- Gain: ~570 (measured) vs. ~584 (model)

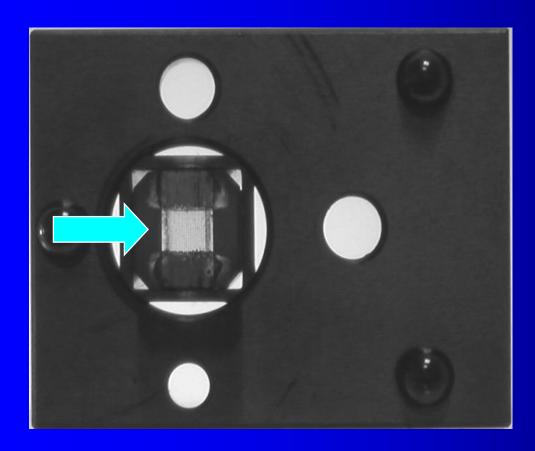






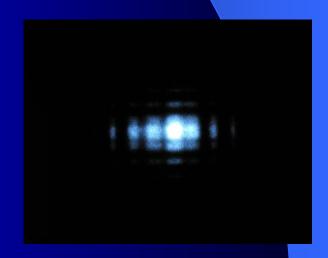
Micro LE: 3 x 3 x 14 mm module

glass foils 30 microns thick separated by 70 microns

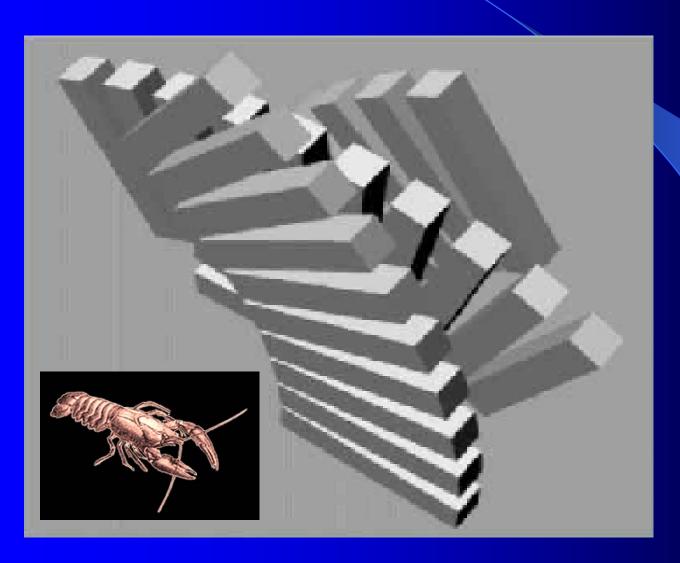




Focal image, 8 keV



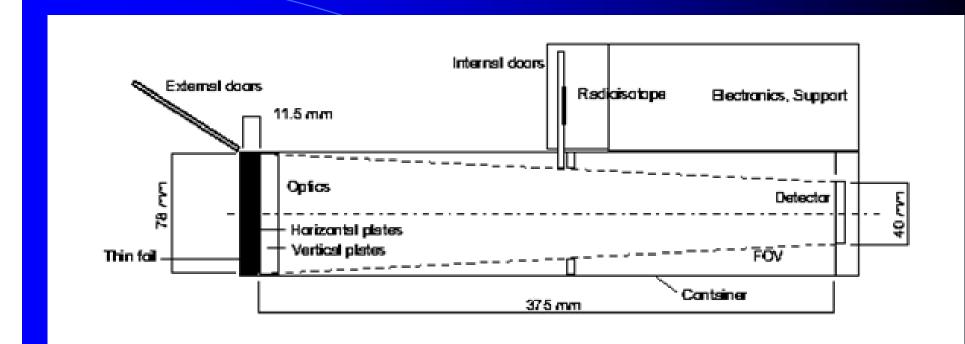
## LE All Sky Monitor - Proposal



Modular concept

Design for ISS

Easy
modification
for EXIST,
HXMT or
other
satellites



LE Modular Concept - 1 module

LE ASM ISS = 30 modules

French/Cninese XRA satellite = 3 modules



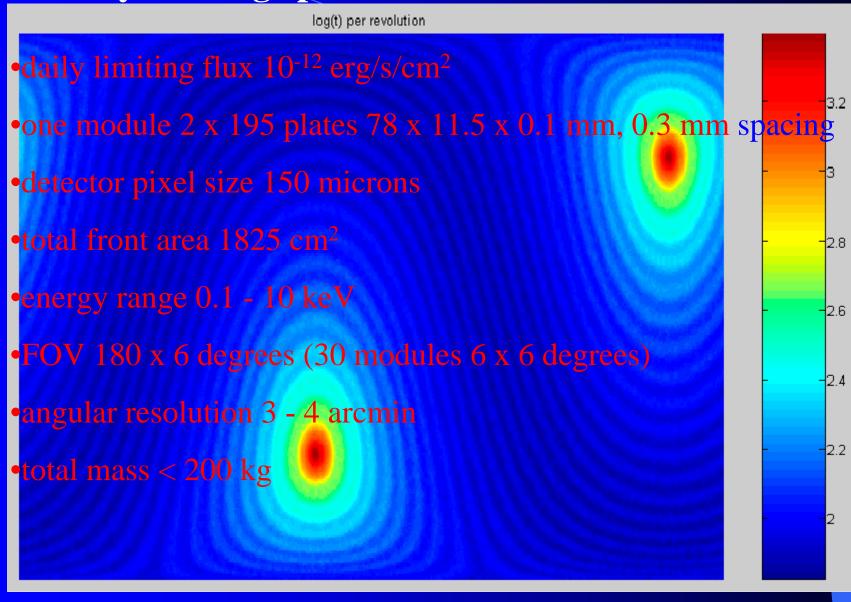
Easy modification for HXMT & other space projects

### The Detector

The detector with dimensions  $40 \times 40 \text{ mm}^2$  with pixel size  $100 \times 100 \ \mu\text{m}^2$  is needed for each of the modules. The detector has to be able to work as a photon counting device, with time resolution better than 1 second.

Fast X-Ray CCDs like Epic-PN detector on XMM-Newton or newly developing detectors based on the MOS technology seems to be the optimal choice.

#### LE Sky coverage per revolution



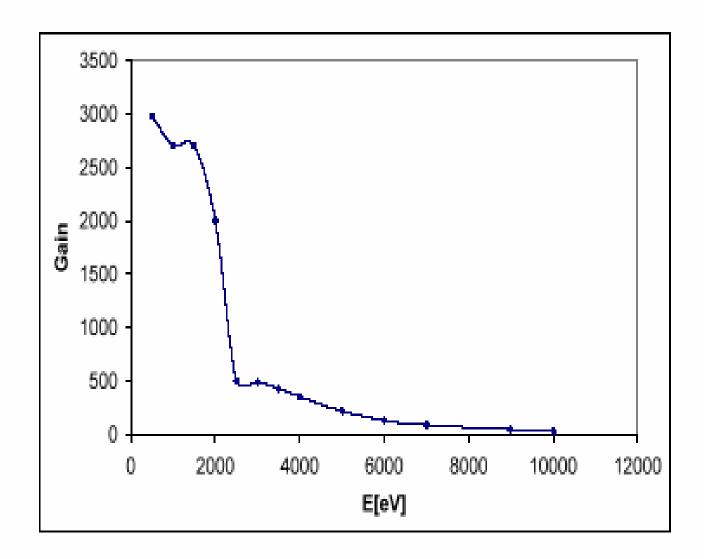


Figure 7: The gain computed from the ray-tracing simulation

### Scientific goals

#### 1. Alert System for X-ray transients



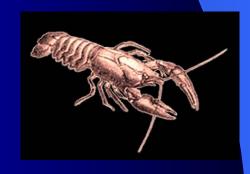
fast recognition of new X-ray sources and/or sudden changes in X-ray flux of known sources, prompt emission study, precise positioning, alert system for narrow-field instruments

- •GRB prompt and afterglow X-ray emission (20-60 triggers/year)
- •X-ray flashes (> 8 triggers/year)
- •orphan GRBs (detectable in X-rays but not in gamma)
- •SNe prompt emission (thermal flash) 10-20 triggers/year
- •X-ray binary & CVs flux changes
- •Stellar events in the Sun's vinicity

#### 2. Long-term X-ray source monitoring

long-term monitoring of large number of X-ray sources with sampling of hours to days (depending on the source flux). Light curves for all the sources together with rough spectra (continuum monitoring, strong lines, iron detectable). In the list below we assume the limiting flux of 10-12 erg/s/cm2 (but we can go deeper):

- •X-ray binaries ~ 700 triggers
- •Cataclysmic Variables ~ 200 triggers
- •stars ~ 600 triggers
- •AGN ~ 4 000 triggers
- •galaxy clusters ~ 400 triggers
- •SN remnants



### When the LOBSTER will go to space?

1. UK Leicester led collaboration (with our participation)

LOBSTER ISS, RSG with Russia

#### 2. Czech collaboration

Balloon experiment?

LE channel on Chinese HXMT satellite?

LE channel on EXIST and/or analogous satellites?

Small French-Chinese satellite (proposal delivered)

Small Czech scientific satellite?

We offer the technology to Italy and China for their space programs

### LE Balloon experiment

For celestial X-ray sources

For atmospheric X-ray sources (red sprites/above active thunderstorms/terrestrial gamma-ray flashes)







#### Positions of terrestrial gamma-ray flashes observed by RHESSI



2004 Jul 1 00:00:00

## Conclusion



The LE ASM will very significantly contribute to various regions of recent astrophysics

The necessary technical background is already available, making proposals for space project based on Lobster Eye optics possible

Applications also in other areas, e.g. atmospheric science

Space Lobster projects recently considered by ESA and by the Chinese Space Agency



Table 1: ASM main features

Energy Range	0.1 - 10.0  keV				
Angular Resolution	$\sim 3-4~\mathrm{arcmin}$				
FOV	$180 \times 6 \deg$				
Orbit	LEO				
Survey	$\sim 100\%$ sky coverage each orbit				
Daily Limitting Flux	$\sim 10^{-12} {\rm \ erg/s/cm^2}$				
Number of modules	30				
Number of detectors	30				
Detector Pixel Size	$150 \times 150 \ \mu \mathrm{m}$				
Energy Resolution	E/dE > 10				
Total Detector Area	$480~\mathrm{cm}^2$				
Total Front Area	$1825~\mathrm{cm}^2$				
Total Mass	< 200  kg				
Data Transfer Rate	$\sim 30 \; \mathrm{kbps}$				
Mission Lifetime	> 2  years				