

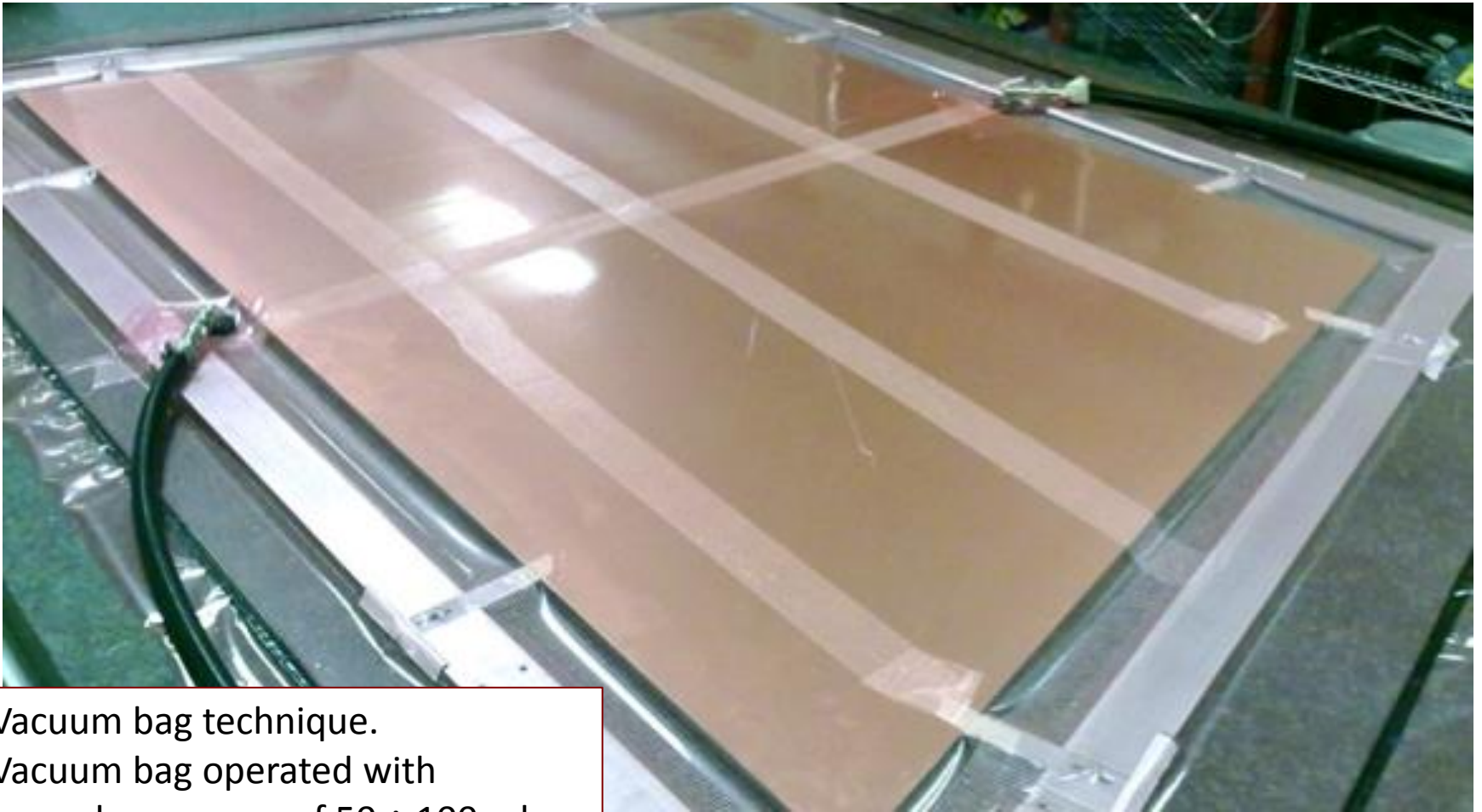
Status Update on Mechanical Prototype in Rome

November 6, 2013

Introduction

- We have built the mechanical ***M2 quadruplet*** prototype:
 - Assembly of the **5 panels** with the “*vacuum bag*” technique (with some variants);
 - Assembly of the **quadruplet**.
- Outline of the talk:
 - Description of the method used for the assembly of the last 2 panels (**two-steps gluing technique**);
 - M2 panels: structure and components
 - Results of **planarity measurements** of the last 2 panels;
 - Quadruplet assembly;
 - Conclusions.

Method of construction - I



Vacuum bag technique.
Vacuum bag operated with
an underpressure of $50 \div 100$ mbar.

Method of construction - II

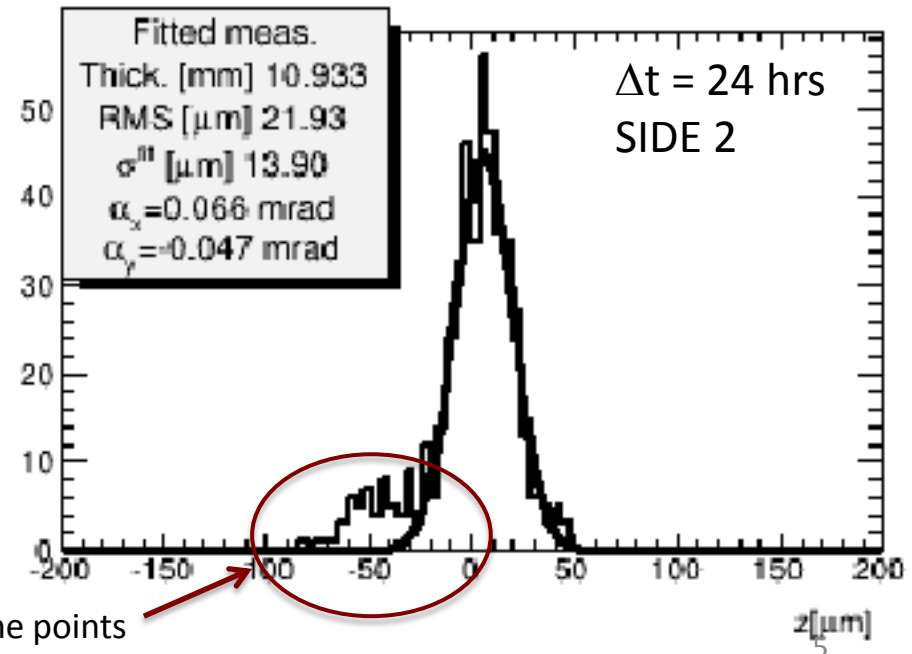
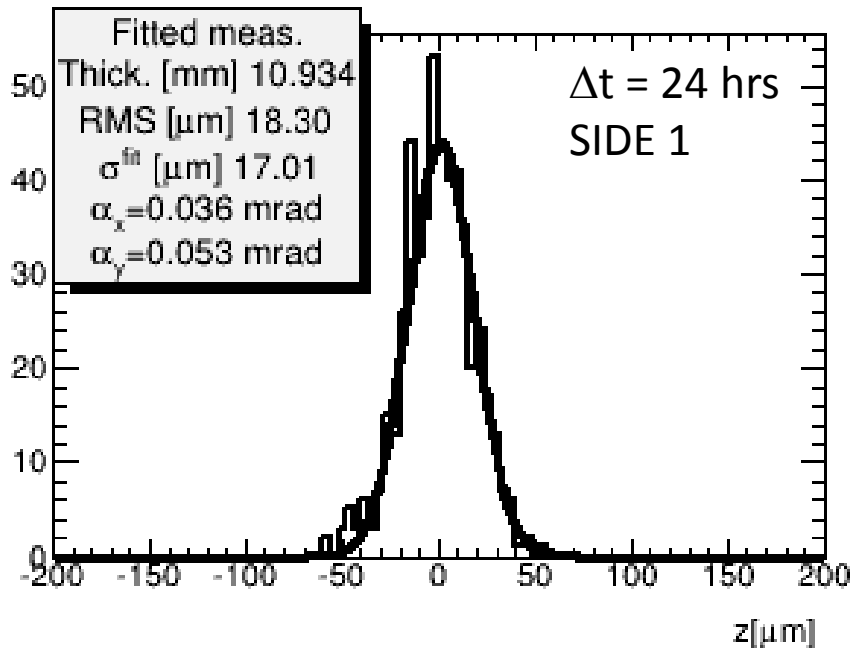
- **First 3 panels** (see july workshop presentation):
 - **Single-step** vacuum bag gluing:
PCB-1 on the granite table, PCB-2 on the honeycomb “pushed” by the bag. Glue is put on both PCBs;
→ asymmetric planarities obtained (*)
≈ 10 ÷ 20 μm (bottom side)
≈ 30 ÷ 50 μm (top side)
- **Last 2 panels:**
 - **Two-steps** vacuum bag gluing:
PCB-1 glued as above. After a time Δt the panel is turned and glued to PCB-2 standing on the granite table.
 - $\Delta t = 24$ or 5 hrs (glue curing time ≈ 15 hrs @ 20°)

(*) Module 2 built with segmented honeycomb has larger RMSs

Test of two-step gluing method

Aim of the method: reach a good planarity on both sides of the panel.
We have done several tests to validate the two-step method, both with $\Delta t=5$ and $\Delta t=24$ hrs .

100x60 cm² panel → planarity below 20 μm on both sides
→ thickness value -26 μm wrt nominal



frame points
frames were not rectified in this test

M2 panels - components

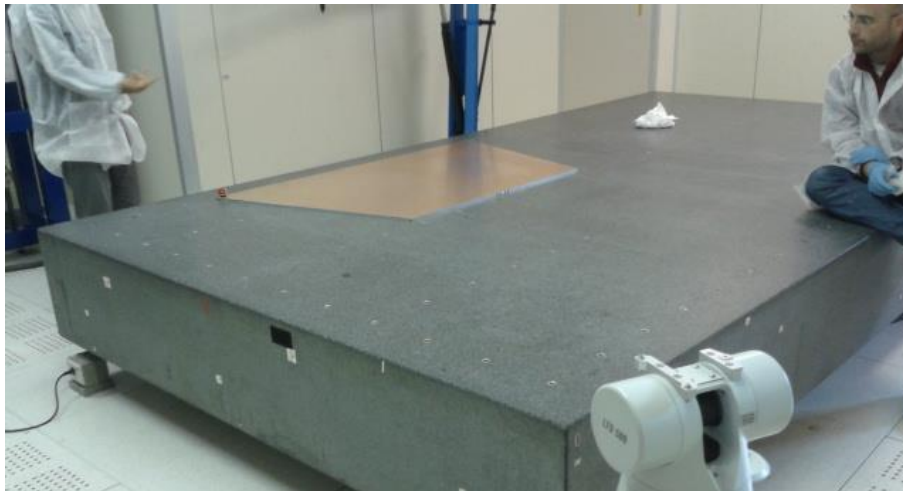
	PCB foils	Honeycomb NOMEX	Aluminum frame
nominal thickness (mm)	0.5	10	10
effective average thickness (mm)	0.40 0.56 (G10 foils)	10.040	10
planarity RMS (μm)	$\approx 20 \div 30$	≈ 25	≈ 25

- **PCB** measurements done by PV group on sucking plane.
- **Honeycomb** measurements done using the LNF laser tracker with a gauge block above it.
- **Frame** thickness and planarity checked with Mitutoyo.
- **Glue** Araldite 2011: the glue is rolled to give a uniform layer of $\approx 60 \mu\text{m}$ thickness.
- **Nominal thickness**= $2 \times 0.40 + 2 \times 0.06 + 10.040 = 10.960$ (**11.120** panel 2) mm

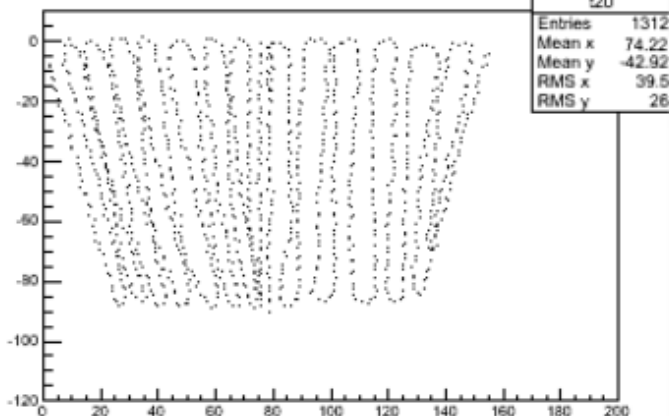
Methods of planarity measurement - I

Laser tracker @ LNF

≈2500 points with few μm accuracy

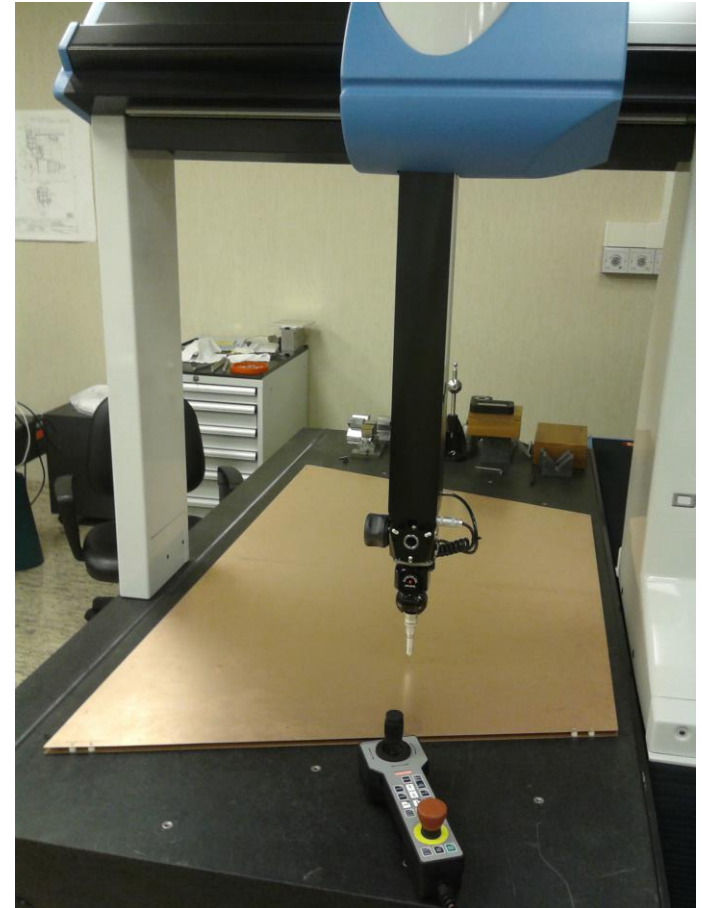
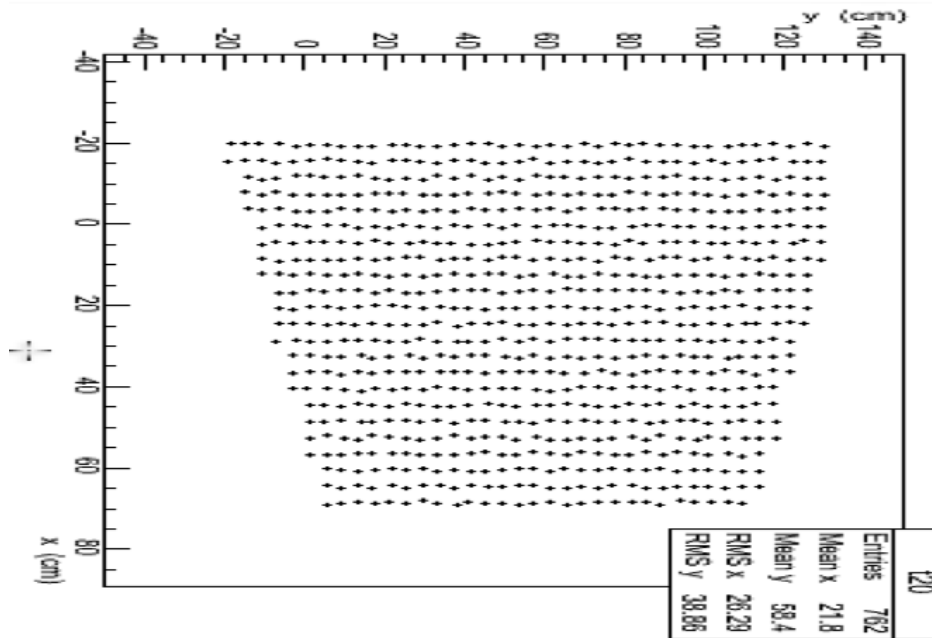


x y point raw



Methods of planarity measurement - II

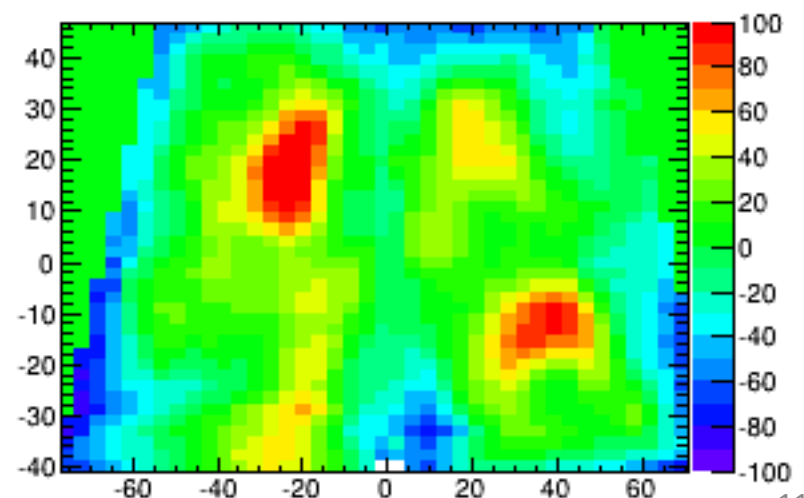
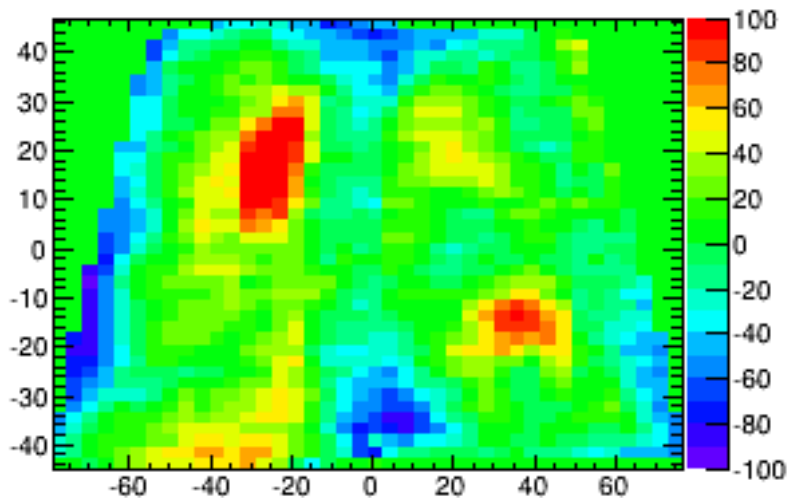
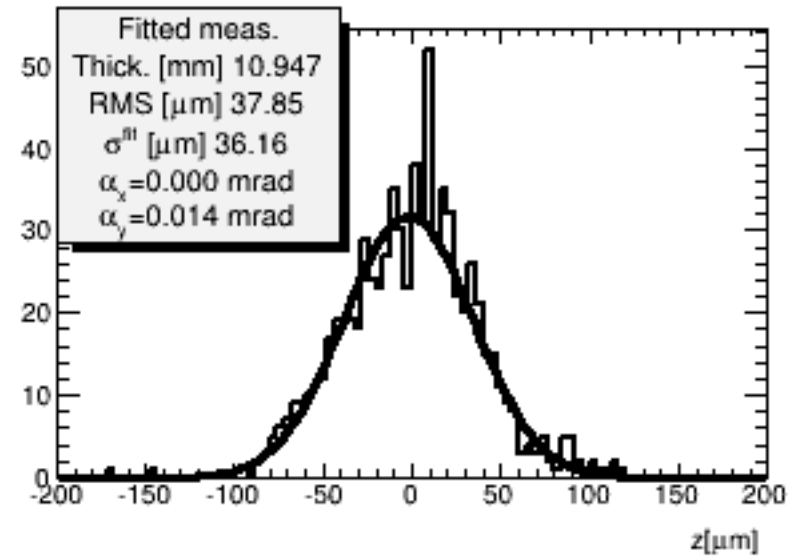
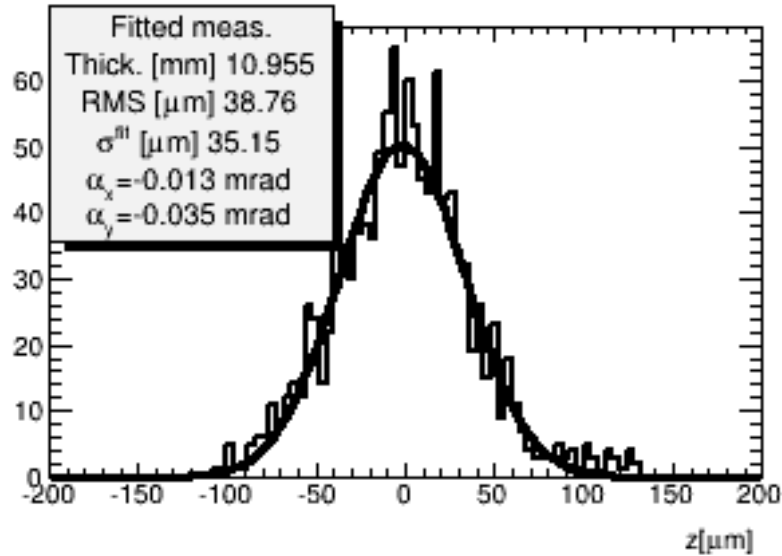
CMM Machine @ INFN Pisa
(points are in a 4 cm spacing lattice)
 ≈ 700 points with $2\ \mu\text{m}$ nominal precision



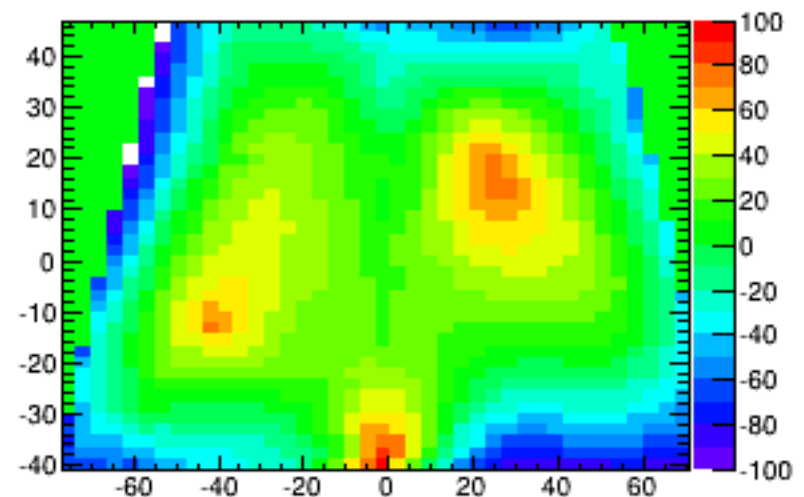
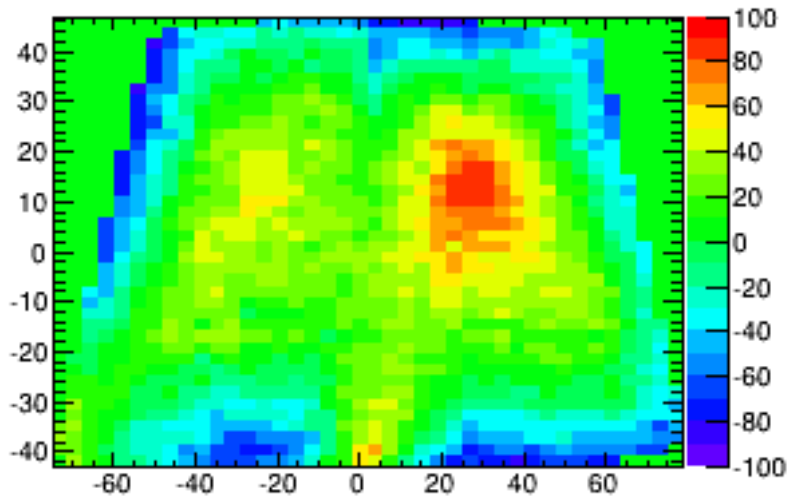
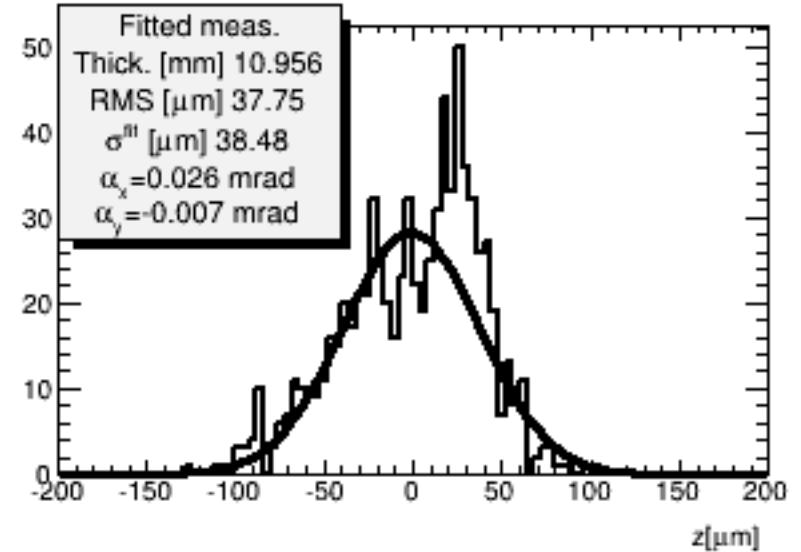
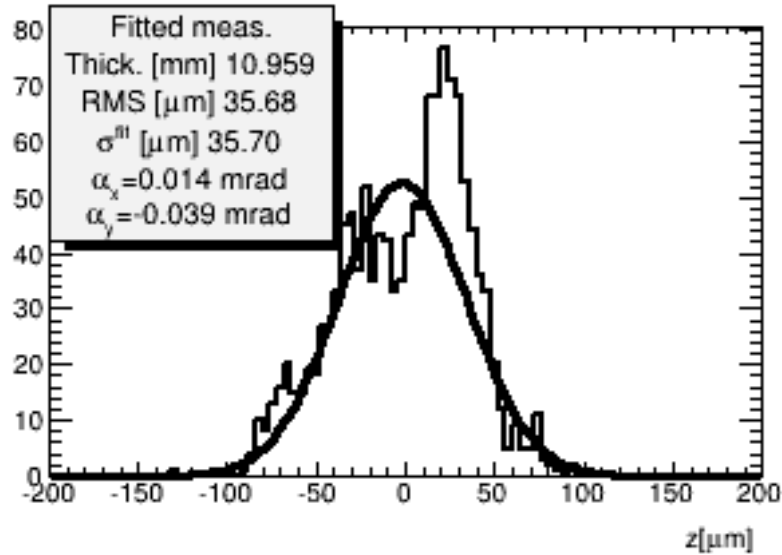
Analysis of planarity measurements

- → A list of **3D points** wrt a plane defined by the machine itself (obtained by measuring the granite table).
- Our analysis:
 - **Raw distribution** of heights of the points → σ_{RAW}
 - **Fit of the “best plane”** → **<thickness>**, α_x , α_y
 - Distribution of heights wrt fitted plane → σ_{FIT}
- All these infos are given in the following.
- LNF laser tracker is our baseline measurement, comparison with CMM will be shown.

Comparison between planarity measurements: LNF (laser tracker) vs. Pisa (CMM machine) – panel 5 side 1



Comparison between planarity measurements: LNF (laser tracker) vs. Pisa (CMM machine) – panel 5 side 2



Angles of fitted planes

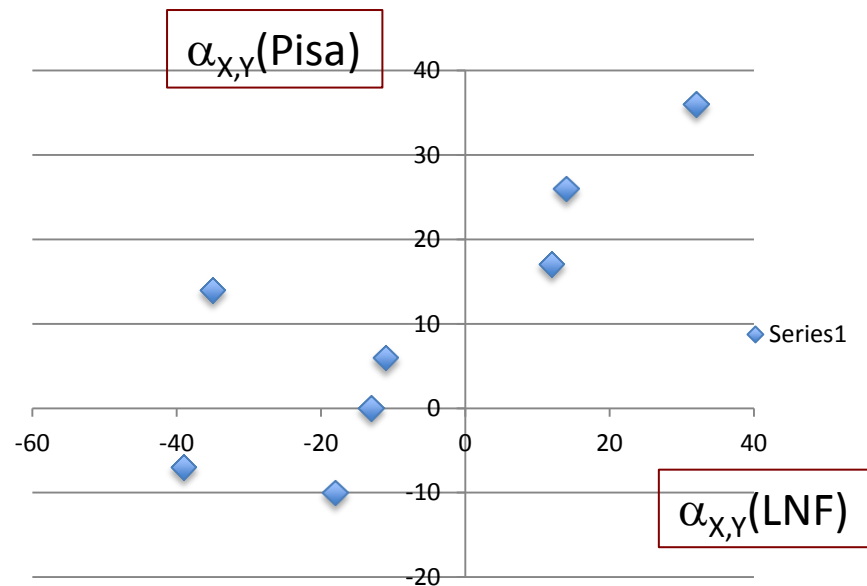
Panel	$\alpha_x(\text{LNF})$	$\alpha_x(\text{Pisa})$	$\alpha_y(\text{LNF})$	$\alpha_y(\text{Pisa})$
4_1	32	36	-11	6
4_2	12	17	-18	-10
5_1	14	26	-39	-7
5_2	-13	0	-35	14

RMS(LNF) = 25 μrad

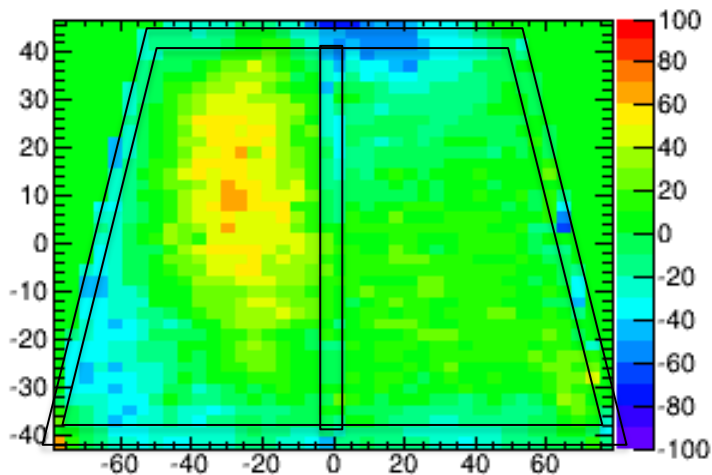
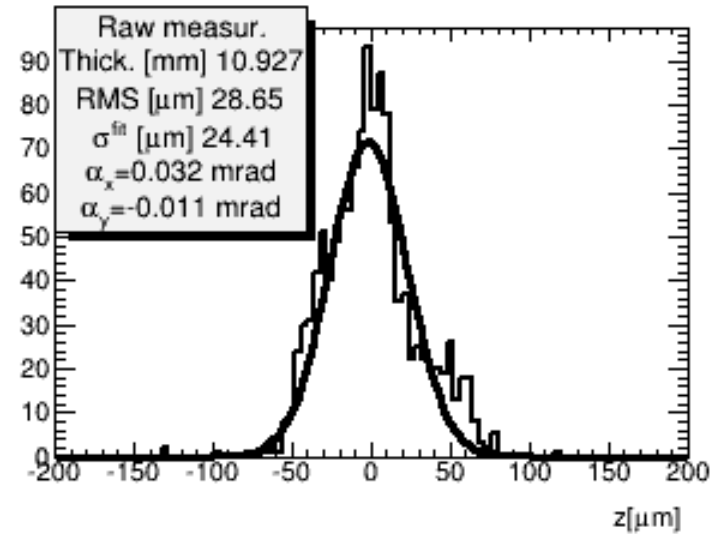
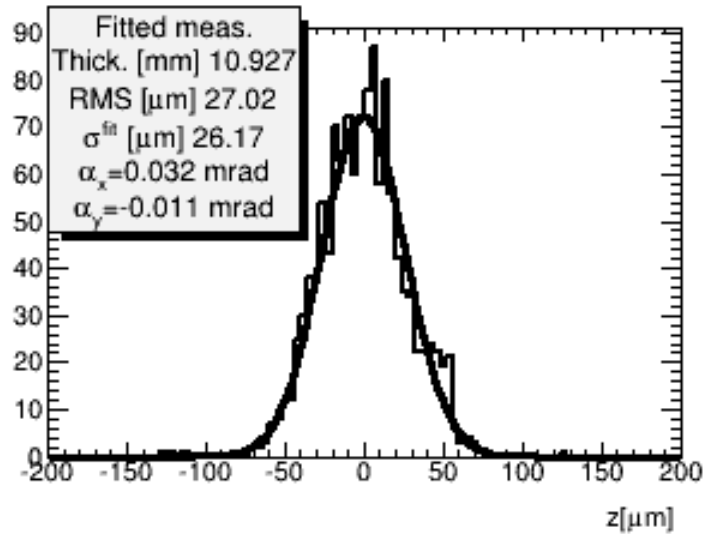
RMS(Pisa) = 16 μrad

Correlation LNF-Pisa:

Some correlation is present.



Results: panel 4 ($\Delta t = 24$ hrs) side 1



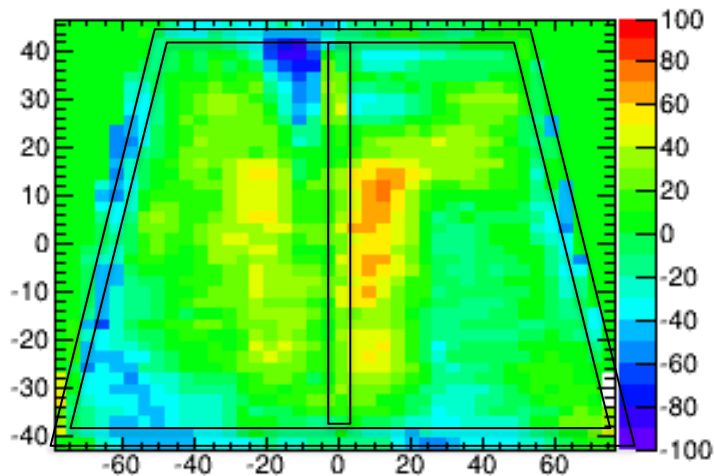
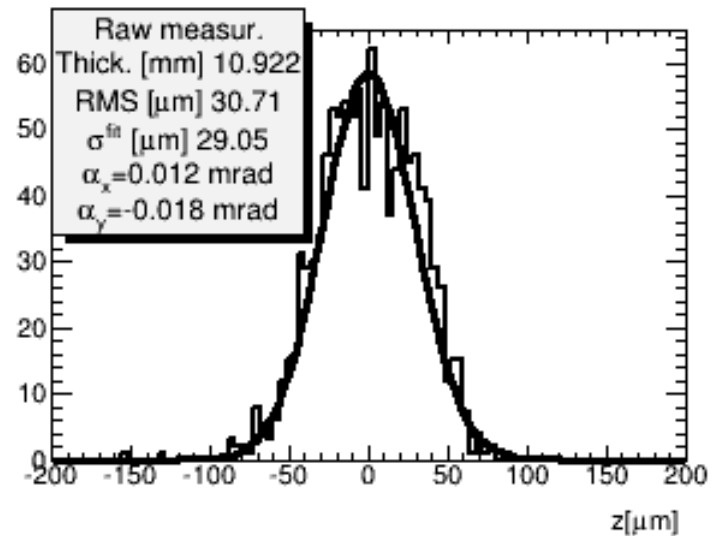
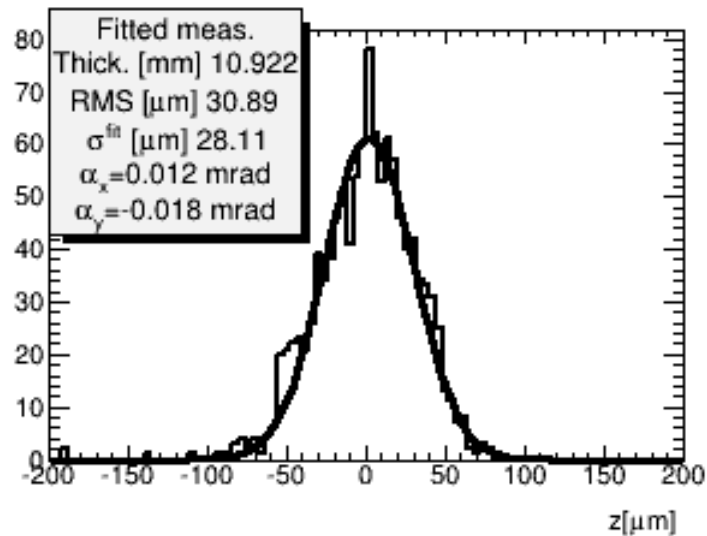
$$\sigma_{\text{RAW}} (\text{RMS}) = 24.4 (28.6) \mu\text{m}$$

$$\sigma_{\text{FIT}} (\text{RMS}) = 26.2 (27.0) \mu\text{m}$$

removing the frames:

$$\sigma_{\text{FIT}} = 21.7 \mu\text{m}$$

Results: panel 4 ($\Delta t = 24$ hrs) side 2

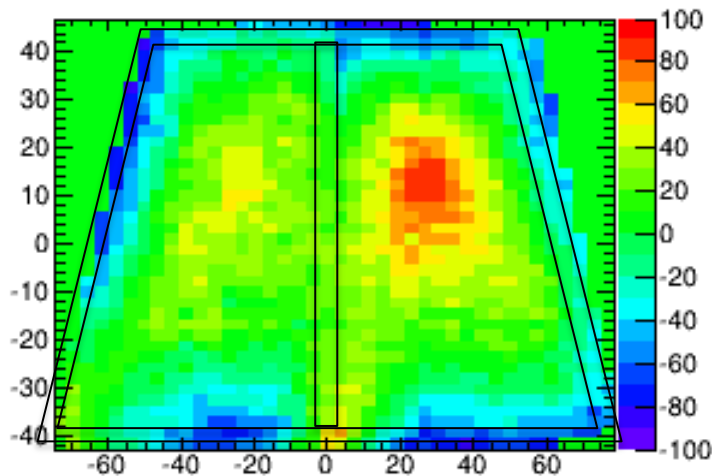
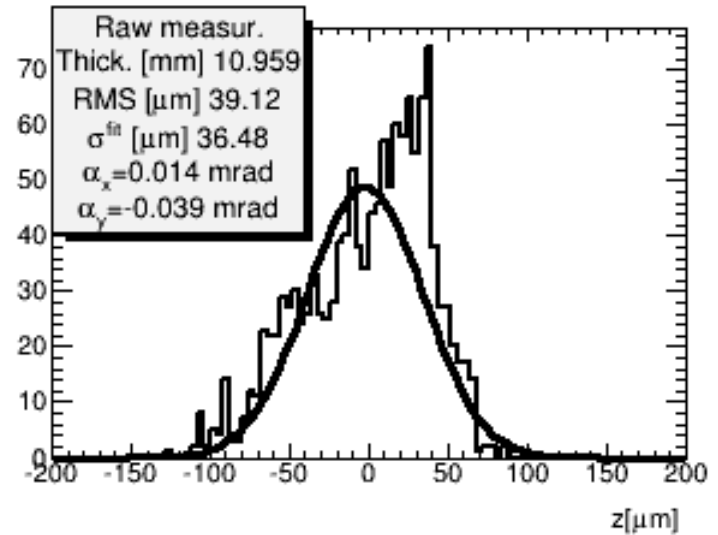
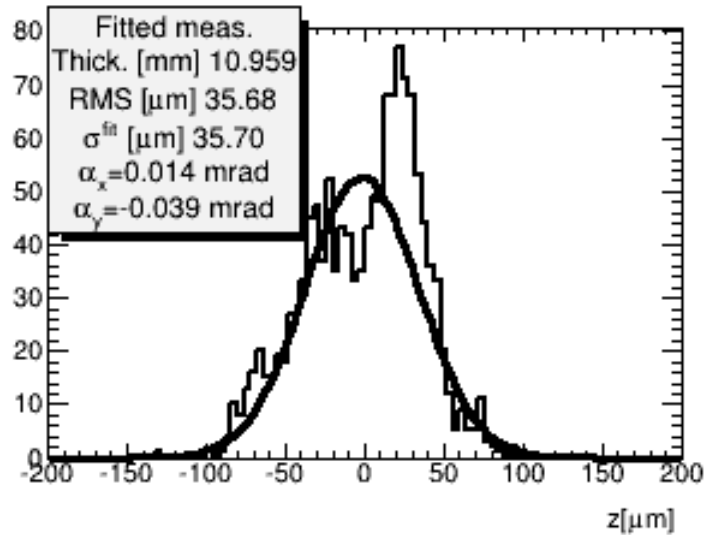


$$\sigma_{\text{RAW}} (\text{RMS}) = 29.0 (30.7) \mu\text{m}$$
$$\sigma_{\text{FIT}} (\text{RMS}) = 28.1 (27.9) \mu\text{m}$$

removing the frames:

$$\sigma_{\text{FIT}} = 22.9 \mu\text{m}$$

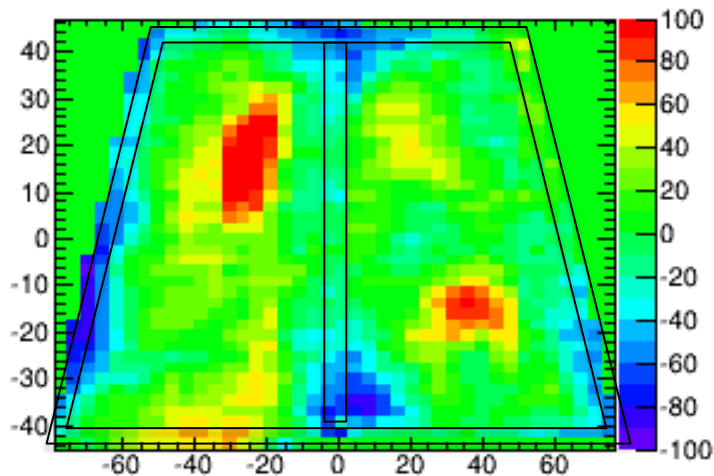
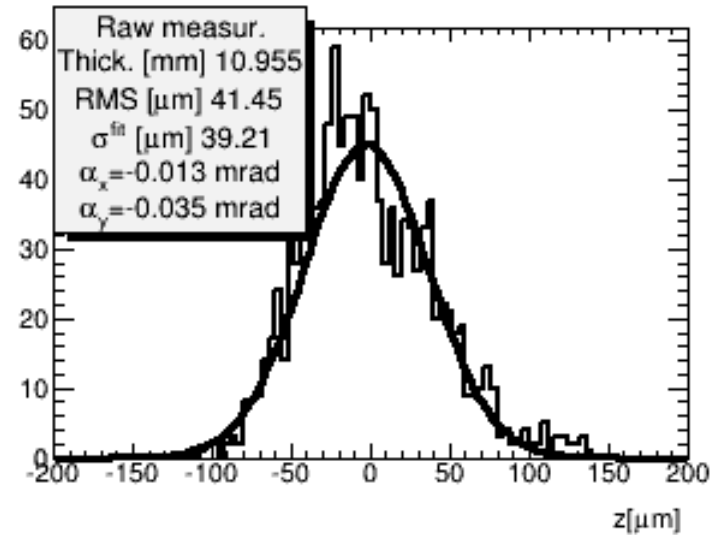
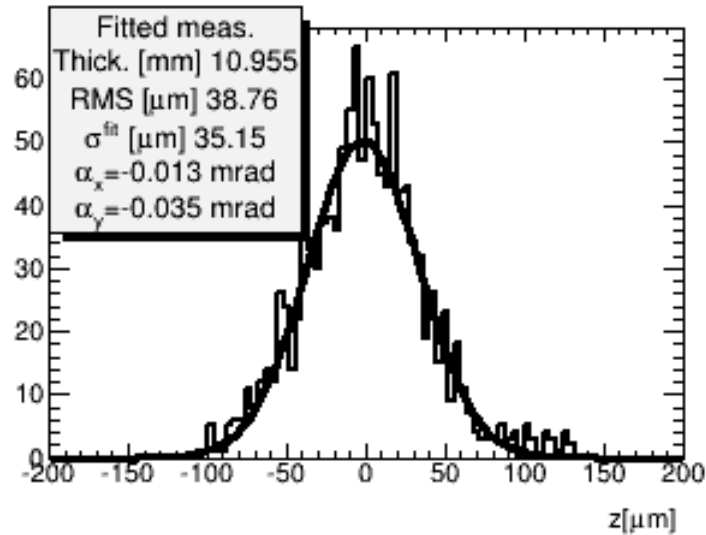
Results: panel 5 ($\Delta t = 5$ hrs) side 1



σ_{RAW} (RMS) = 36.5 (39.1) μm
 σ_{FIT} (RMS) = 35.7 (35.7) μm
“depression” on external frame

removing the frames:
 $\sigma_{\text{FIT}} = 22.8 \mu\text{m}$

Results: panel 5 ($\Delta t = 5$ hrs) side 2



$$\sigma_{\text{RAW}} \text{ (RMS)} = 39.2 \text{ (41.2)} \mu\text{m}$$

$$\sigma_{\text{FIT}} \text{ (RMS)} = 35.2 \text{ (38.8)} \mu\text{m}$$

“depression” on external frame

removing the frames:

$$\sigma_{\text{FIT}} = 21.5 \mu\text{m}$$

Summary - I

Panel #	construction	$\sigma(\mu\text{m})^{(1)}$ overall	$\sigma(\mu\text{m})^{(2)}$ no frame	Thickness ⁽³⁾ (mm)	
1	One-step gluing	16 47	11 34	10.948 (-12)	CMM machine
2	One-step gluing (segmented hc)	85 76	73 54	11.154 (-34)	CMM machine (not to be considered)
3	One-step gluing	15 41	12 38	10.963 (+3)	CMM machine
4	Two-step gluing ($\Delta t = 24$ hrs)	26 28	22 23	10.927 (-33)	Laser tracker
5	Two-step gluing ($\Delta t = 5$ hrs)	36 35	23 21	10.959 (-1)	Laser tracker

(1) Sigma obtained after plane fitting σ_{FIT}

(2) The measurement is performed removing all points within 10 cm from the frame

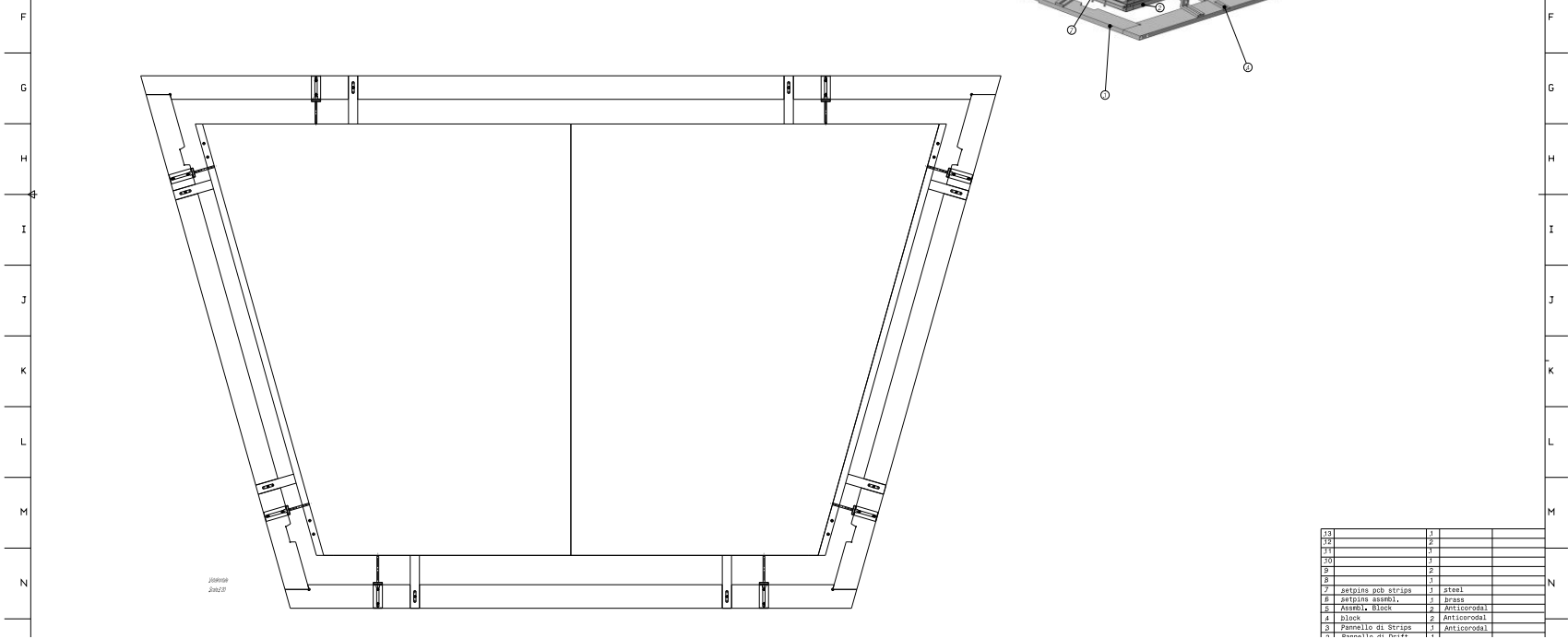
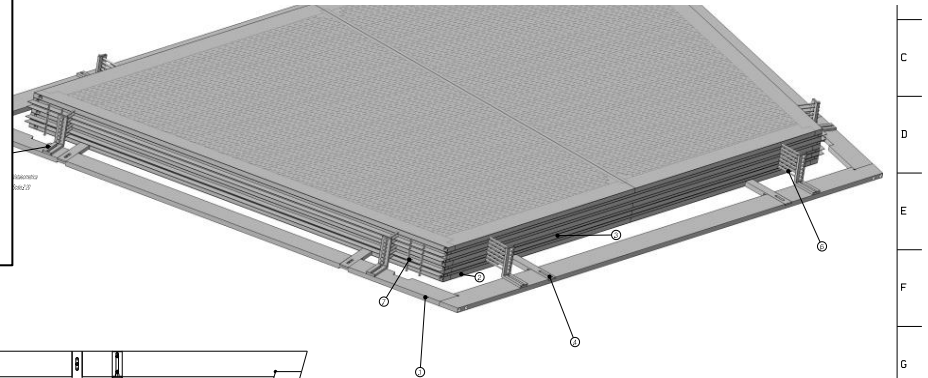
(3) Thickness measured with side 1 in the table. In () difference in μm wrt nominal.

Summary - II

- **Planarities:**
 - Two-steps method with $\Delta t = 24$ hrs turns out to be the best option with overall planarities RMS well below $30 \mu\text{m}$ on both faces.
 - Improvements expected by a better “definition” of the frame → $20 \mu\text{m}$ planarities can be reached.
- **Panel thickness wrt nominal** (assuming $2 \times 60 \mu\text{m}$ glue thickness):
 - In average $-15 \mu\text{m}$ with an RMS of $17 \mu\text{m}$.
 - Improvements expected by a better control of the glue amount.
- **Methods of measurement**
 - CMM machine and Laser-tracker give results in good agreement.

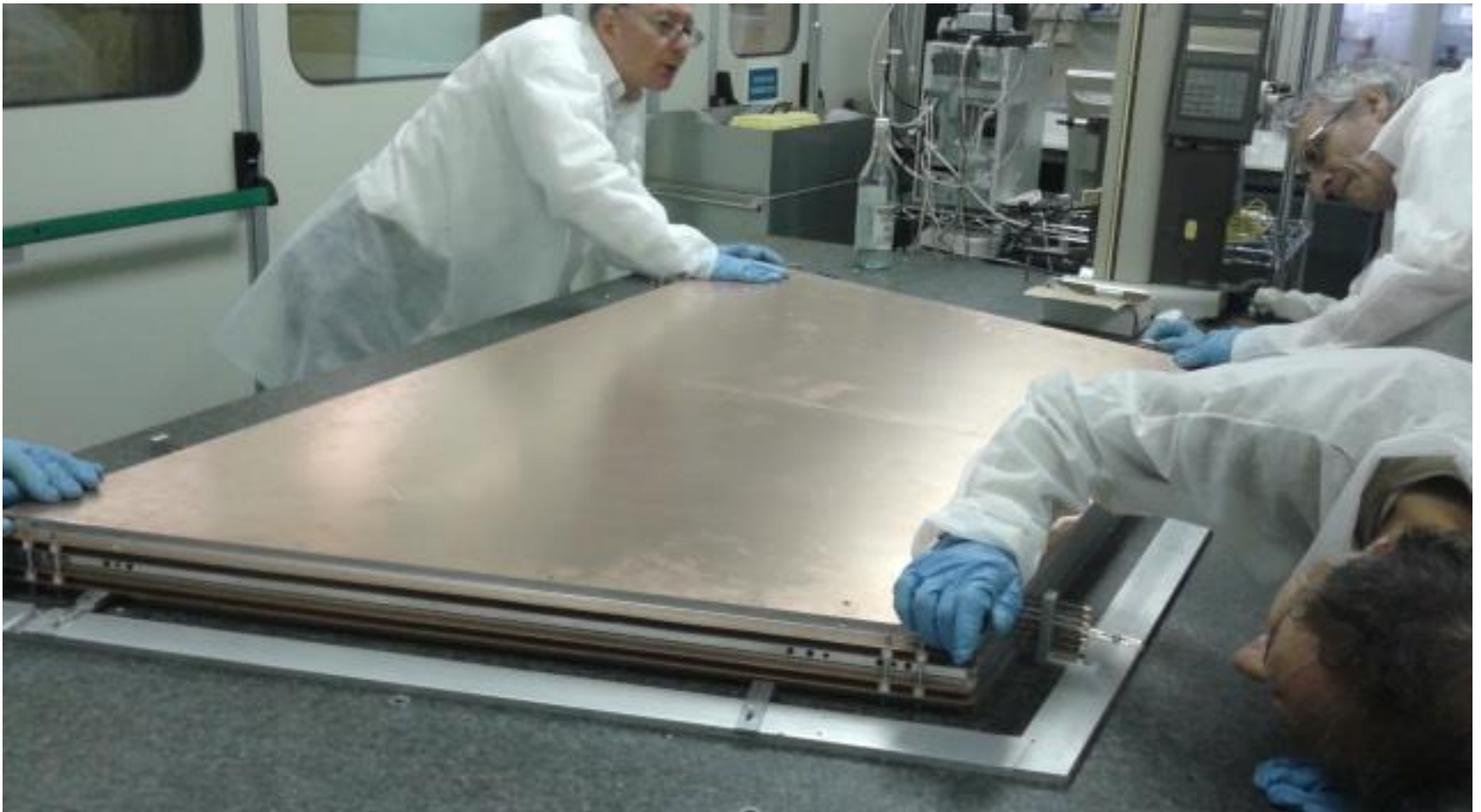
Quadruplet assembly - I

Assembly scheme:
 Panels are glued to drift gap frames
 (5 mm thick Al not rectified, 20 mm wide)
 No vacuum bag used, gluing by gravity
 all panels in a single step.

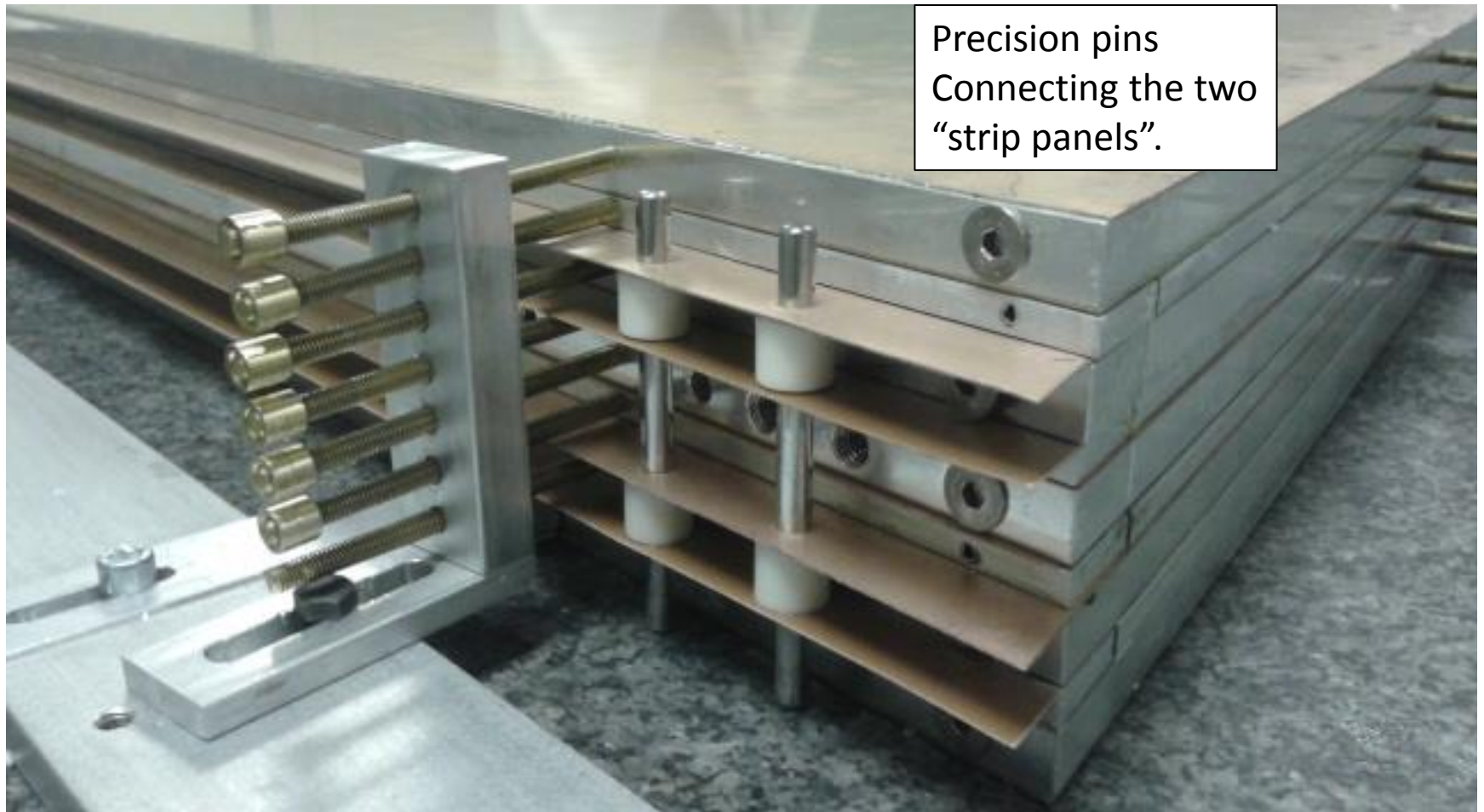


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Quadruplet assembly - II



Quadruplet assembly - III



Precision pins
Connecting the two
“strip panels”.

Status of quadruplet

- Mechanical properties still to be measured. Probably a “sag” effect on the top side, to be quantified.
- We plan to make holes along the frame to allow mounting on the spacer.

Conclusions

- **M2 dummy quadruplet ready to be mounted** (only holes are still missing).
- **Vacuum bag technique** (with two-steps gluing option) allows to get:
 - *Overall planarities below 30 μm on both sides;*
 - *Average thickness reproducibility better than 20 μm ;*
 - Space for improvements from better frame structure and glue amount control.
- Plan to build a **full-size SM1 drift panel** using the two-step method with $\Delta t = 24$ hrs

Backup

