

# Error Studies for Low-Emittance Rings with Application to APS-U



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Accelerator Systems Division  
Argonne National Laboratory

**1st Workshop on Low Emittance Lattice Design**

April, 23-24 2015

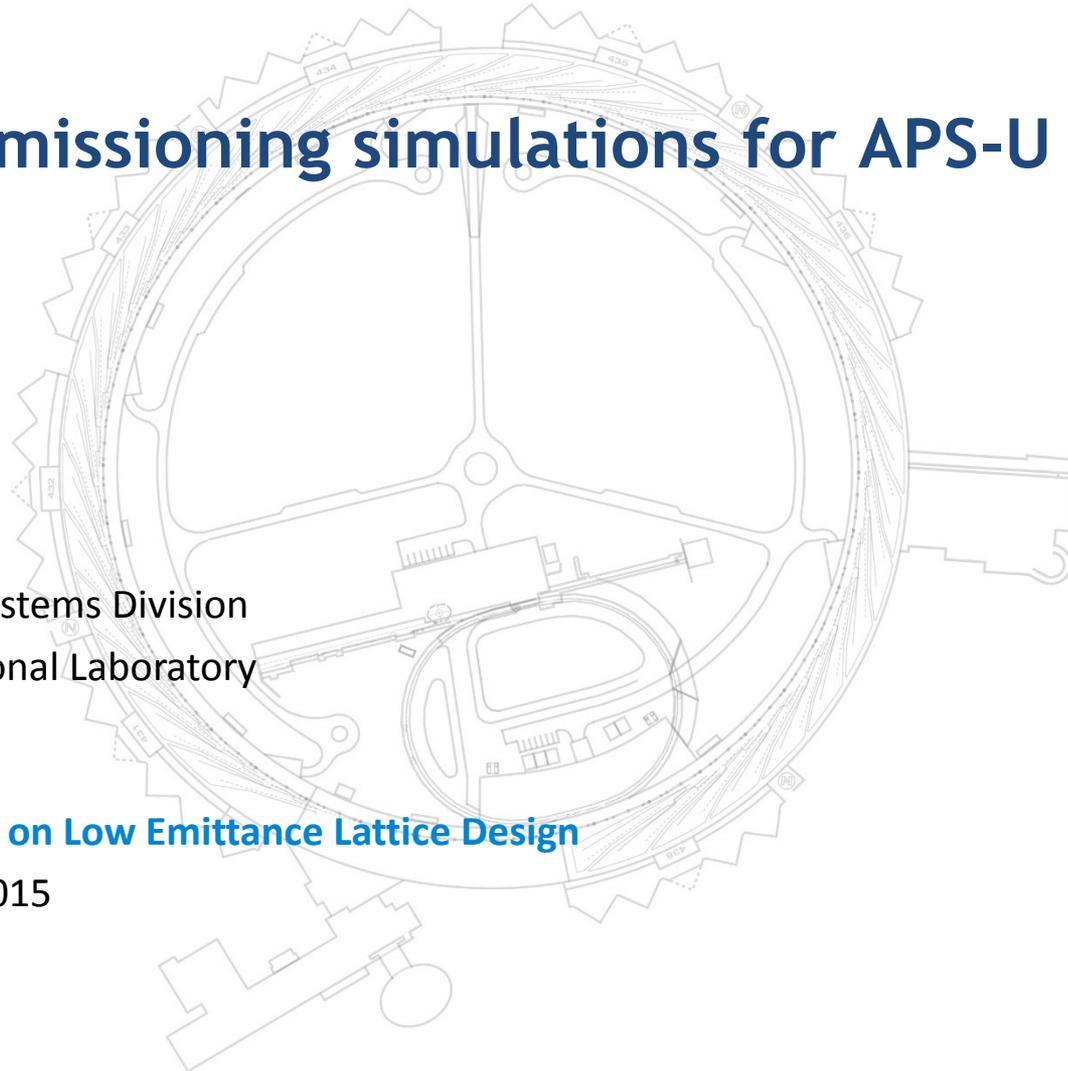
# Commissioning simulations for APS-U lattice

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# Motivation

- Several important questions are being asked from the beginning of the project:
  - With such strongly-focusing lattice and such small-gap vacuum chamber, will you be able to commission the ring (in a short amount of time)?
  - What are the right errors to use for lattice evaluation
  - Will you be able to keep the orbit stable down to a fraction of the beam size with such small emittances?
- In addition, we needed to make a decision on maximum strength of dipole and coupling correctors
  - Minimize corrector strength
- Commissioning simulation is the best way to answer these questions (though not the orbit stability)
  - Automated commissioning procedure



# Automated Commissioning Procedure

- The procedure closely follows steps that will be made during commissioning
  - The intent is to develop a real automated commissioning program
- First, the lattice is detuned to move tunes away from integer and coupling resonances
  - The original fractional tunes are 0.12, moved to 0.18 and 0.24
- The commissioning simulation consists of 4 main steps
  - Error generation
  - Trajectory correction until closed orbit is found
  - Orbit correction down to an acceptable level
  - Lattice and coupling correction
- All simulations are done with elegant



# Error values

- The following errors were simulated (all rms, distribution is limited at  $2\sigma$ ):

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|                                      |                   |
|--------------------------------------|-------------------|
| Girder misalignment                  | $100 \mu\text{m}$ |
| Elements within girder               | $30 \mu\text{m}$  |
| Dipole fractional strength error     | $1 \cdot 10^{-3}$ |
| Quadrupole fractional strength error | $1 \cdot 10^{-3}$ |
| Dipole tilt                          | 0.4 mrad          |
| Quadrupole tilt                      | 0.4 mrad          |
| Sextupole tilt                       | 0.4 mrad          |
| Initial BPM offset error             | $500 \mu\text{m}$ |
| BPM gain error                       | 5%                |
| BPM orbit measurement noise          | $1 \mu\text{m}$   |
| Corrector calibration error          | 5%                |

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- The correction procedure described below is the result of extensive experimentation with various correction approaches



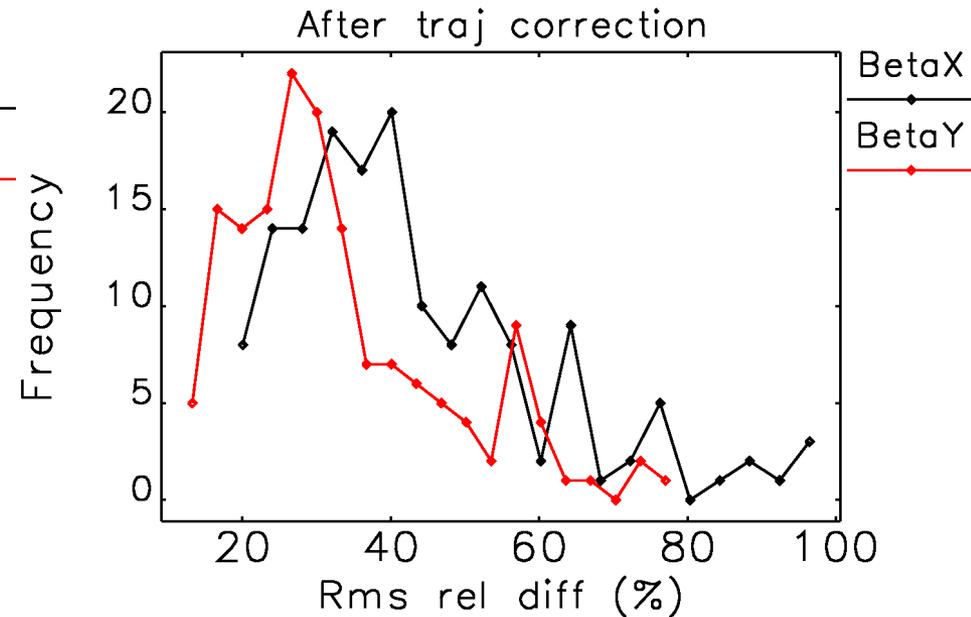
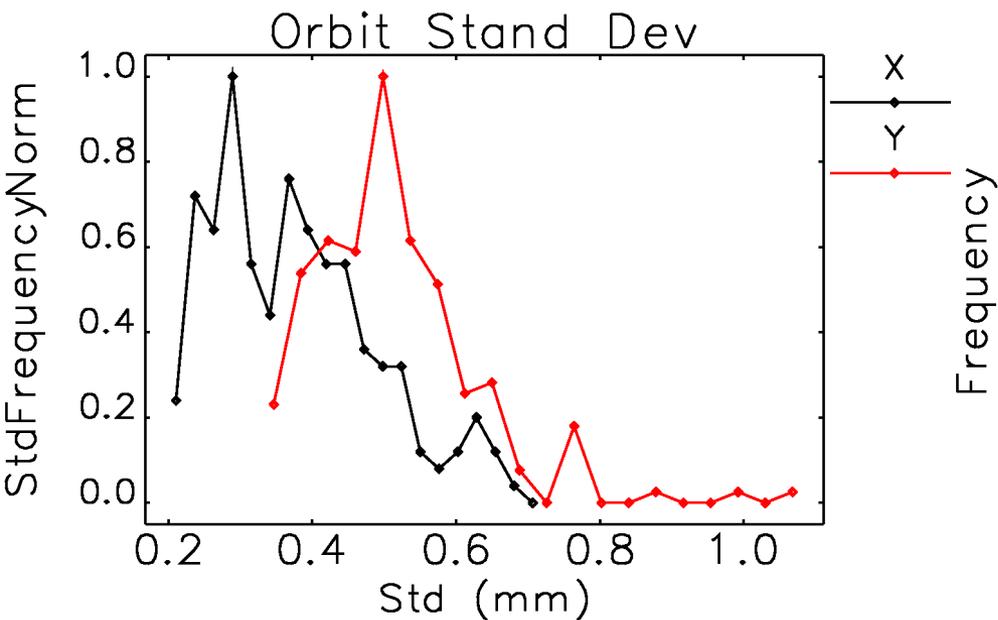
# Trajectory correction procedure

- If we want to expect the closed orbit to be inside the vacuum chamber, the alignment should be at least 3 times better than assumed
  - Too expensive
  - Trajectory correction is required
- First step of trajectory correction is to make the beam complete the first full turn
  - Uses elegant's one-to-best method and only 4 correctors per sector (out of 12) per plane
  - Works well but tends to drive some correctors quickly to their limit
  - To reduce corrector strength, first limit correctors to 1/4 of their maximum strength
  - If first-turn transmission is not achieved, the correction is repeated with increased limits
- Then, global trajectory correction is performed using ideal response matrix
  - Goal – to find the closed orbit
  - Start with small number of singular values, slowly increase SV number, if closed orbit exists, record it
  - When SV number limit is achieved, return to the best orbit
  - If any corrector reaches a limit, reduce it by a certain fraction



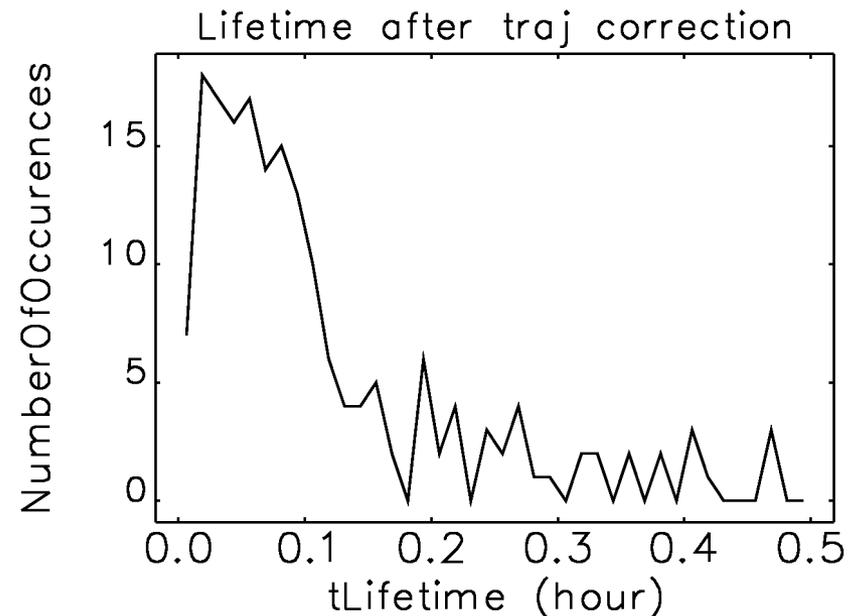
# Trajectory correction results

- The procedure finds closed orbit in 100% of cases
- Beta function errors (right plot) are quite large



# Lifetime after trajectory correction

- Lifetime is calculated based on Local Momentum Aperture for 1-mA bunch
- We have found that many cases have zero lifetime (even with closed orbit)
- To get non-zero lifetime, run transmission optimization
  - Simplex optimization varies betatron tunes and lowest beta function correction harmonics using predefined quadrupole knobs
  - In real life, we will be maximizing the number of completed turns, here we maximize transmission of 5 particles with  $0$ ,  $\pm 0.5\%$ , and  $\pm 1\%$  momentum errors
- With 90% certainty, lifetime is longer than 1 minute; median lifetime is 5 minutes
- Lifetime is adequate to start orbit correction



# Orbit correction

- The orbit correction starts with small number of correctors per sector and small SV number and consists of 2 loops
- The inner loop increases SV number
  - If at any iteration a corrector gets close to its limit, it is reduced by a certain fraction and the iteration is repeated until all correctors are away from limits
  - Tunes are adjusted after every iteration to keep them away from integer resonances
  - The loop is interrupted if the orbit target or SV number limit is reached
- The outer loop increases the number of correctors per sector
  - Correction starts with 2 correctors, increases to 10 correctors in five steps
  - After two iterations, it is assumed the BPM offset measurement is performed
    - Offset measurement is not simulated, offsets are simply reduced by a factor of 10
  - The loop is interrupted if the orbit target or SV number limit is reached



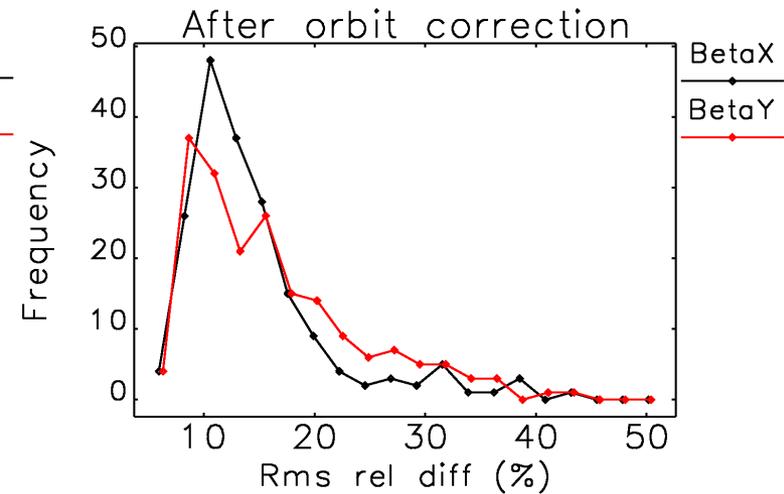
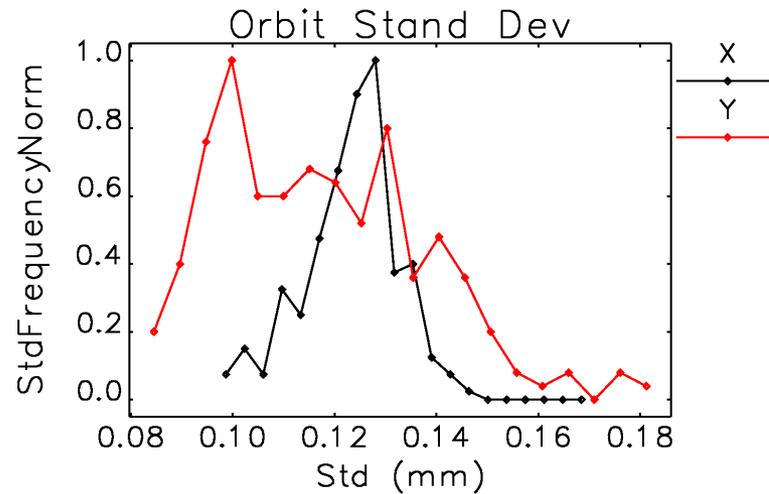
## Orbit correction (2)

- If any iteration fails to improve orbit, a coarse fast kick-based optics correction is performed
  - A 0.1 mrad kick in horizontal plane is used
  - Coupling is so strong, the motion couples into vertical plane immediately
  - Fast enough even for 1-minute-long lifetime
- Ignore BPM gain errors since we are trying to correct  $\sim 100\%$  modulation of beta functions
  - Only interested in BPM-to-BPM oscillation amplitude modulation
  - A few tens of turns is enough, decoherence should not be a problem
  - Simulated as a single particle tracking for 50 turns
- Oscillation is proportional to  $\sqrt{\beta}$ , but proportionality coefficient is unknown
  - Since the tunes are close to ideal, it is assumed that average inverse beta functions are equal to those of the ideal lattice
- Only used one BPM and one quadrupole per sector for simplicity
- Works nicely
  - Will try to apply it during transmission optimization

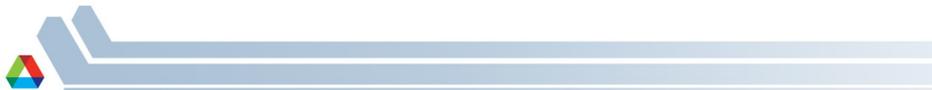
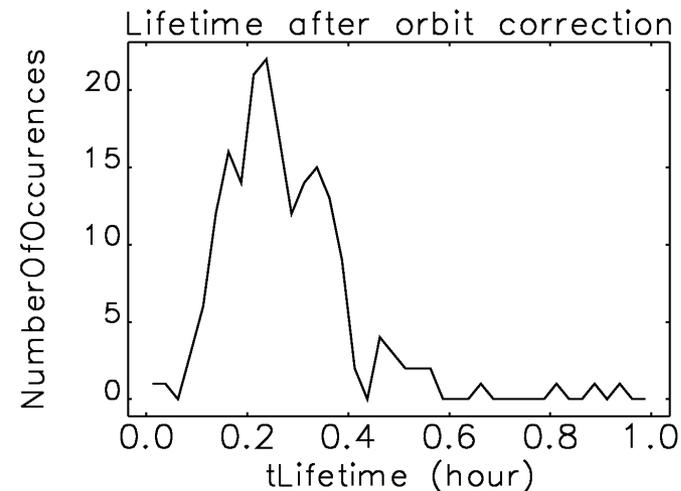


# Orbit correction results

- Target of 0.5mm maximum orbit error is achieved in 98% of cases
- Orbit rms is about 100  $\mu\text{m}$



- With 90% certainty, lifetime is above 8 minutes, median lifetime is 15 minutes



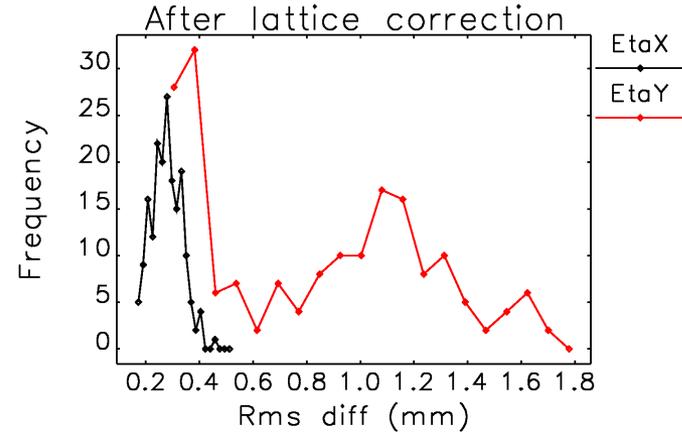
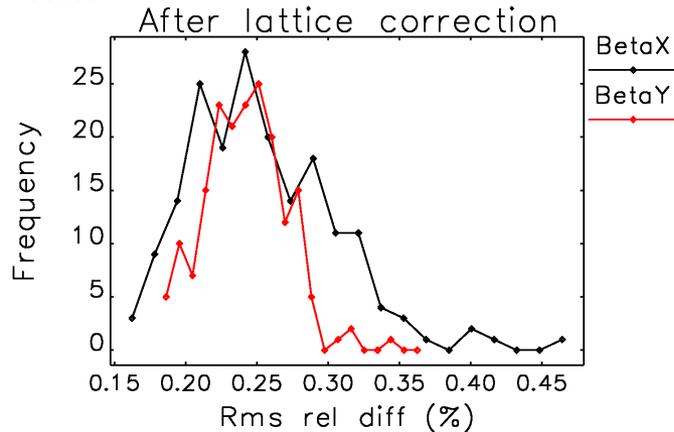
# Lattice correction

- Now the lifetime is considered long enough to start lattice correction
- A standard correction procedure that uses response matrix (RM) fit and was developed for APS operations is used
- “Measured” response matrix is generated with BPM noise, BPM gain, and correction calibration errors
  - For measurement speed, only 8 correctors per plane are used (measurement time should be below 5 minutes)
- After response matrix fit is run, the beta functions and horizontal dispersion are corrected using ideal beta function response matrix
- Coupling is corrected by minimizing cross-plane orbit response together with vertical dispersion
- Correction is performed in several iterations with increasing SV number
  - All quads and 4 skew quads per sector are used for correction
  - Orbit is corrected after every iteration
- After minimization, coupling is adjusted to the target of 10% by exciting nearest difference resonance using skew quads

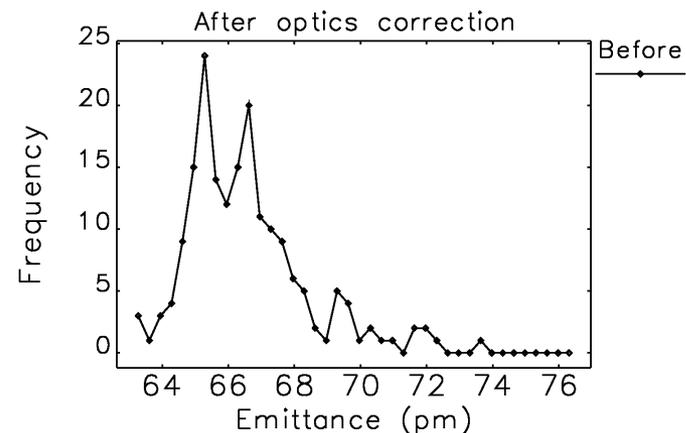
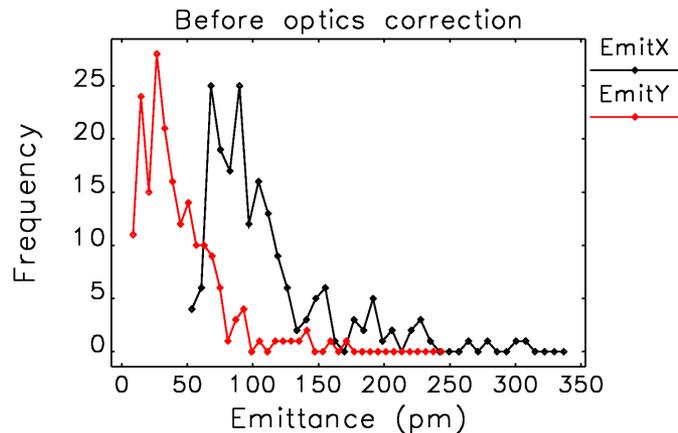


# Lattice correction results

- Beta functions are corrected to below 1% relative rms, dispersion is below 1mm rms

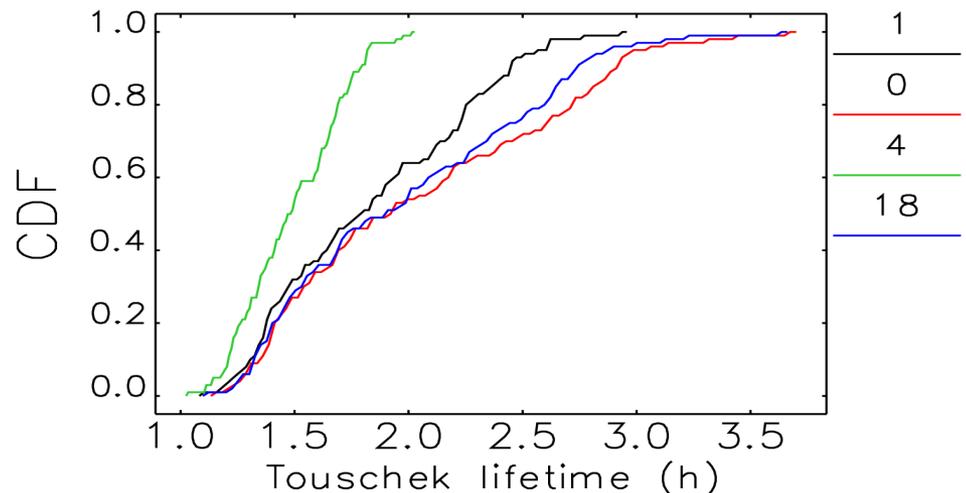


- Emittances before (left) and after (right) lattice correction:



# Nonlinear evaluation of the lattice<sup>1</sup>

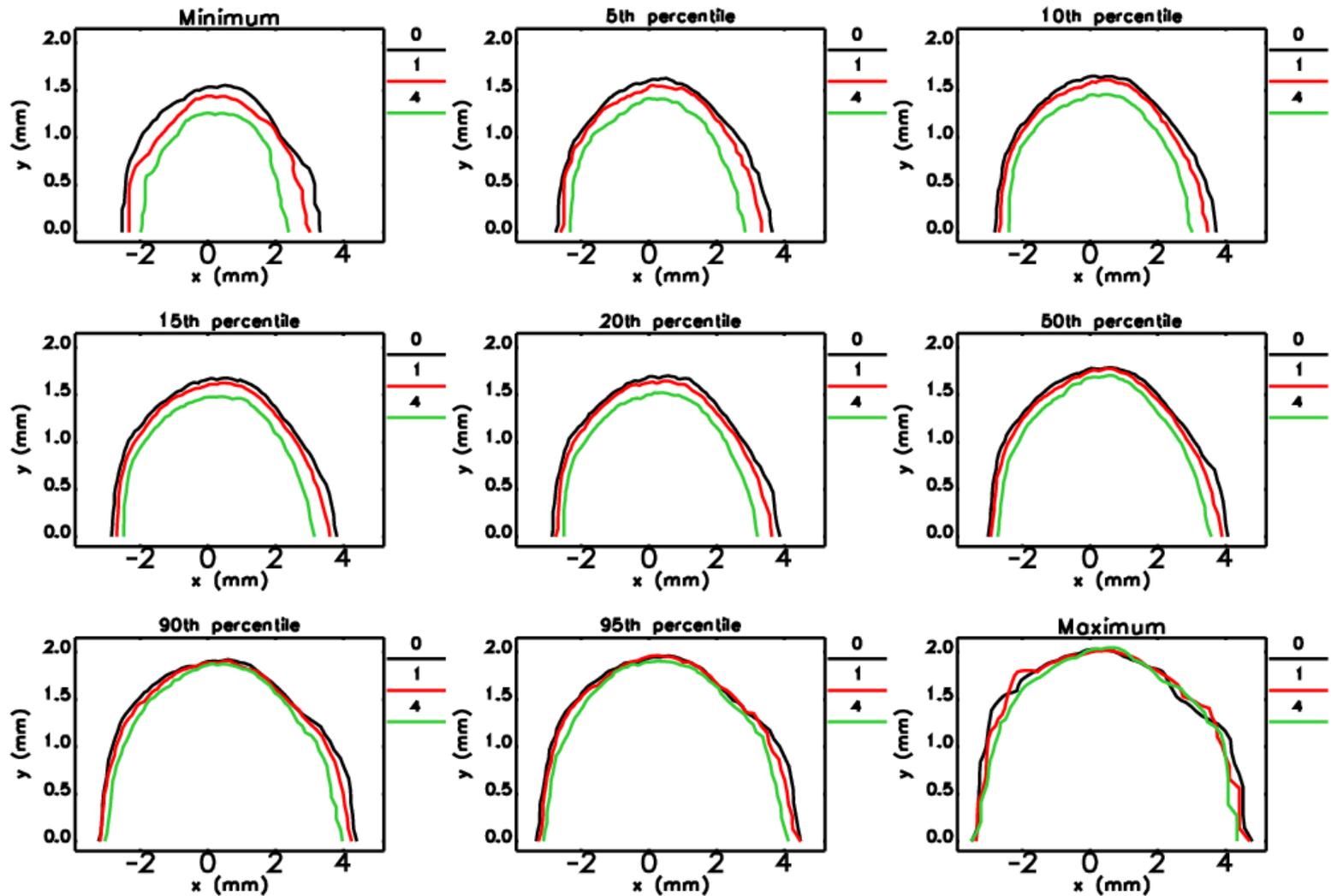
- Corrected lattice files were used to calculate dynamic and momentum aperture
- Multipole errors were added to all magnets
  - Systematic multipoles were taken from magnetic calculations
  - Random multipoles were generated based on NSLS-II experience
- Many multipole configurations were simulated, random multipoles in quadrupoles were found to have the most effect on lifetime:
  - Case 0: No multipoles
  - Case 1: Nominal multipoles
  - Case 4: Double multipoles in quads
  - Case 18: Half multipoles in quads
- Lifetime was calculated for 48-bunch pattern without high-harmonic cavity



<sup>1</sup>Calculations performed by M. Borland

# Nonlinear evaluation of the lattice (2)

- Multipoles have less effect on dynamic aperture



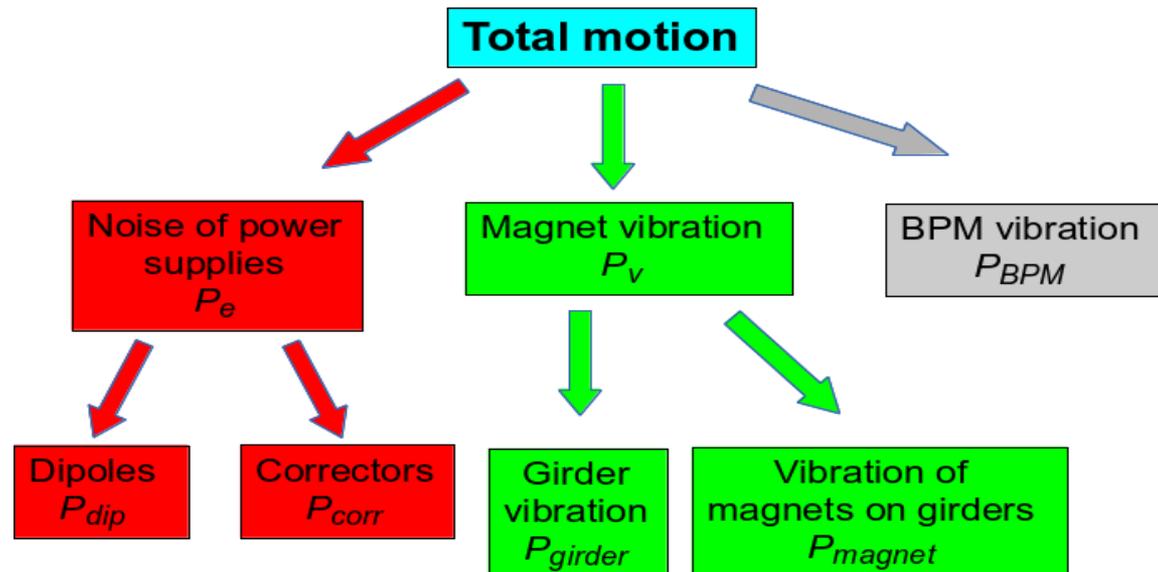
# Conclusions

- Automated commissioning simulation procedure was written
- Commissioning was simulated for 200 sets of random errors, succeeded in 98% of cases
- Corrected lattices have lifetime and dynamic aperture that is adequate for on-axis swap-out injection and 7-second topup
- Commissioning simulation gave requirements on maximum dipole and skew quadrupole corrector strength
- High accuracy correction of orbit and lattice is possible, and the design low emittance is achievable
- Commissioning simulation procedure will serve as a guide for real automated commissioning procedure



# Orbit stability

- Orbit motion is produced by electrical noise in magnet power supplies and by variable magnetic fields generated by vibrating quadrupoles through frequency-independent amplification factors
- Orbit stability in 0.01Hz-1kHz region is a very stringent requirement (0.4  $\mu\text{m}$  in Y)
  - Cannot be achieved without orbit correction
- Orbit correction reduces orbit motion with frequency-dependent attenuation factor
- Mechanical motion of BPMs affects orbit via orbit feedback
- In some cases, stability without orbit correction is important too



$P$  is portion of mean square motion

# Assumptions about Spectral Characteristics

- Electrical noise PSD is  $1/f$
- Ground vibrational noise PSD is  $1/f^3$ , lower frequency is limited to 0.1 Hz (wavelength of 500 m or longer)
  - Girders/supports and individual magnets follow the same spectrum
- BPM motion follows the same vibrational spectrum, orbit correction makes orbit directly follow BPM motion
  - BPM electronic noise of 100 nm is added to this motion
- Orbit correction is an integral controller with bandwidth  $f_{bw}$ , its attenuating effect on noise power is  $f^2/f_{bw}^2$ 
  - Only global orbit correction is considered
- Attenuation of electrical noise due to solid core magnets and vacuum chamber was simulated<sup>1</sup>
- A single orbit measurement (for response matrix) is one-second averaging
  - Only frequencies above 1 Hz contribute to noise

<sup>1</sup>S. Kim, private communication

# Apportioning of motion budget

- Allowable contribution of different motion sources varies depending on whether orbit correction is running or not
- Orbit correction is very efficient at attenuating motion due to vibration, but it emphasizes motion due to BPM vibration
- During beam studies, only frequencies above 1 Hz contribute, where motion due to power supply noise has the most power

|  |                  | User operation,<br>orbit correction on                                     | Beam studies,<br>orbit correction off |
|--|------------------|--|---------------------------------------|
| Total motion budget                      |                  | $0.1\sigma$<br>( $1.7 \mu\text{m}$ , $0.4 \mu\text{m}$ )<br>0.01 - 1000 Hz | $1 \mu\text{m}$<br>1 - 1000 Hz        |
| Portion of mean square motion<br>due to: |                  |  |                                       |
| power supply noise                       | $P_e$            | 0.1  | 0.89                                  |
| BPM motion                               | $P_{\text{BPM}}$ | 0.89   | 0.01                                  |
| vibrational noise                        | $P_v$            | 0.01   | 0.1                                   |



# Summary of tolerances

- Electrical noise tolerances (fractional strength error):

| Magnet type   | Requirement         | Based on                             |
|---------------|---------------------|--------------------------------------|
| Correctors    | $1.9 \cdot 10^{-4}$ | Orbit stability w/o orbit correction |
| Dipoles M3-M4 | $2.2 \cdot 10^{-5}$ | Orbit stability w/o orbit correction |
| Dipoles M1-M2 | $1.4 \cdot 10^{-5}$ | Tune stability                       |
| Quadrupoles   | $2.3 \cdot 10^{-5}$ | Tune stability                       |
| Sextupoles    | $1.8 \cdot 10^{-3}$ | Tune stability                       |

- Vibrational tolerances:

|              | X<br>(rms)<br>1-100 Hz | Y<br>(rms) | X<br>(rms)<br>0.1-1000 Hz | Y<br>(rms) |
|--------------|------------------------|------------|---------------------------|------------|
| $u_{girder}$ | 32 nm                  | 40 nm      | 320 nm                    | 400 nm     |
| $u_{quad}$   | 13 nm                  | 9 nm       | 130 nm                    | 90 nm      |

