

Applications of Geant4 in space (recent developments)

Giovanni Santin*, Petteri Nieminen



*Space Environments and Effects Analysis Section
European Space Agency
ESTEC*

** on loan from RHEA Tech Ltd*



Outline



- Geant4 Space Users' community
- Highlights from 2009 workshop (Madrid)
- Other Geant4 based work
- Enabling developments

Geant4 Space Users' community



- Geant4 Space Users' web page
 - Users, publications, news
 - <http://geant4.esa.int>
 - Feedback appreciated
- Space Users' Workshops (since 2003)
- Hyper-news: “Space applications” forum



6th Geant4 Space Users' Workshop

Madrid, 19-22 May 2009

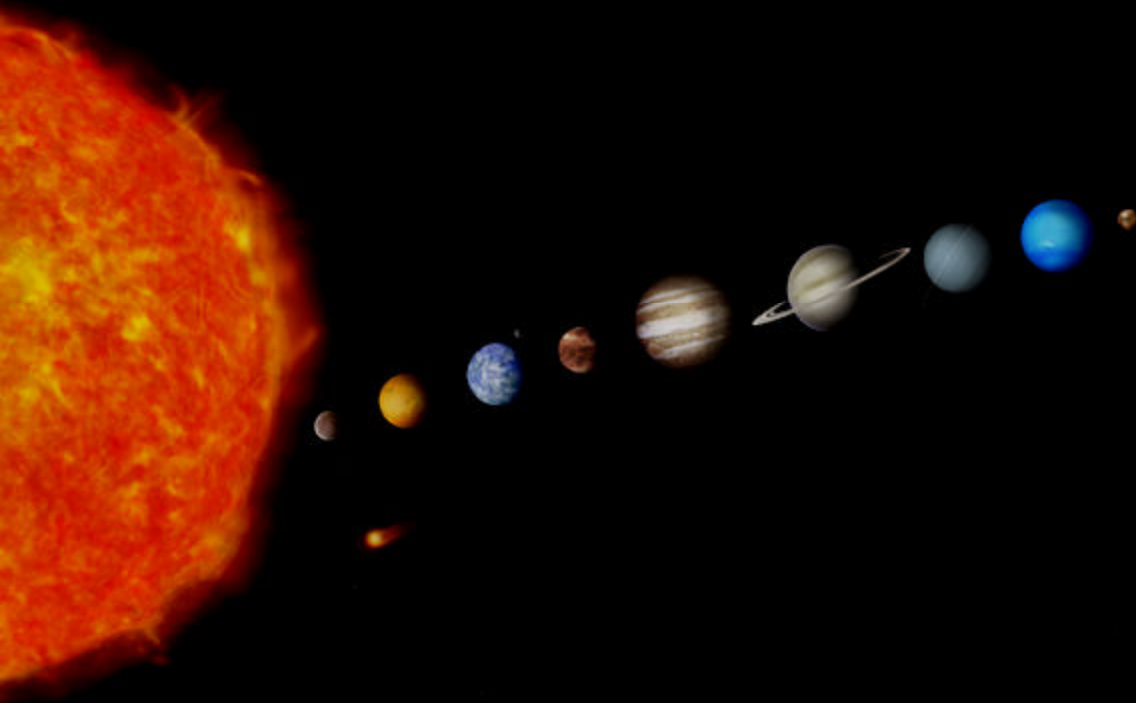


- <http://www.inta.es/g4suw2009/>
- ~60 participants
Wide range of interests and applications
- The following section contains an arbitrary, non exhaustive selection of contributions from the event



Geant4 Space Users' 2009

Planetary exploration



MarsREM



*ESA contract
(QinetiQ, LIP,
BIRA, DHC,
SpaceIT)*

- ESA exploration programme
- Physics:
 - High energy – high Z ion interactions (radiation safety)
 - Filling gaps identified in previous HSF-related studies, e.g. ESA DESIRE (T.Ersmark, KTH)
 - DPMJET2.5 interface
- Tool development
 - dMEREM
 - eMEREM – addressing engineering requirements
 - Active magnetic shielding
- SPENVIS interface
- *Presented by P.Truscott at NSREC 2008, IAC 2008*

MarsREM – DPMJET interface



ESA contract
(QinetiQ, LIP,
BIRA, DHC,
SpaceIT)

Implementation of DPMJET-II.5 model in Geant4

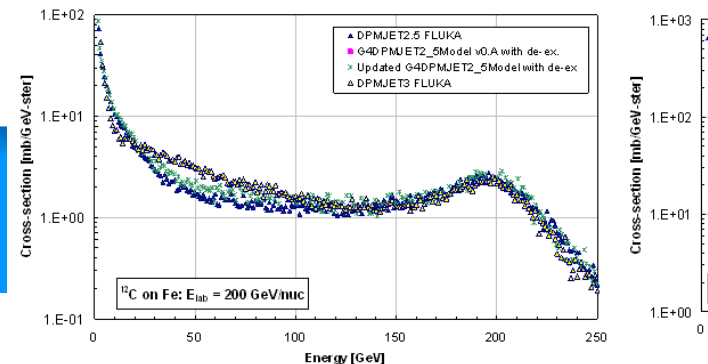
- Priority to extend the high-energy regime of Geant4 to ultra-relativistic energies
- Existing FORTRAN code DPMJET-II.5 to act as an event generator: 5GeV/nuc to $1E+11$ GeV/nuc
- Geant4 DPMJET-II.5 interface: [G4DPMJET2_5Model](#) class now developed and tested, covers projectiles from $A=2$ to $A=58$ on targets from $A=2$ to $A=58$

Currently extending this to projectiles from $A=2$ to $A=240$ on targets from $A=2$ to $A=240$

- After prompt nuclear-nuclear collisions, nuclear de-excitation treated using other Geant4 models (precompound, evaporation, Fermi break-up)
- Total inelastic cross-section class [G4DPMJET2_5CrossSection](#) also created to estimate mean-free-path between nuclear-nuclear collisions, and also covers projectiles from $A=2$ to $A=58$ on targets from $A=2$ to $A=58$

(similarly being extended to $A=240$)

- Models used on Detailed Mars Energetic Radiation Environment Model (dMEREM) to be presented



QinetiQ

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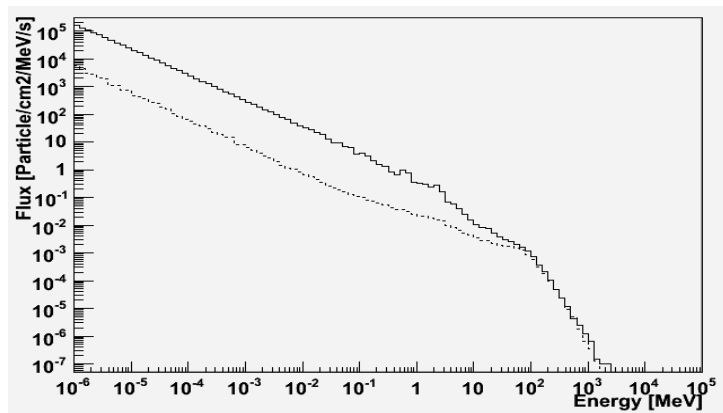
MarsREM - eMEREM



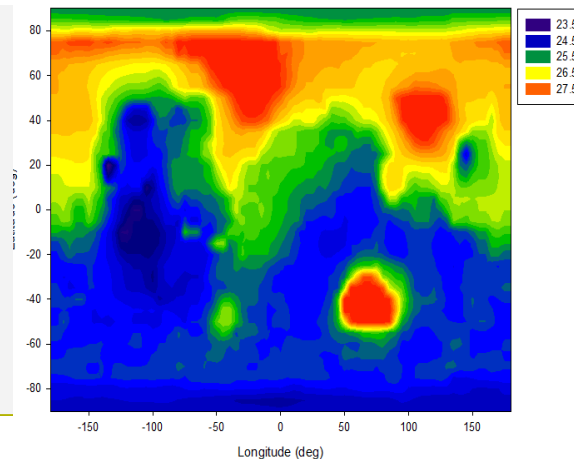
ESA contract
(QinetiQ, LIP,
BIRA, DHC,
SpaceIT)

- Fast, engineering tool
- Response matrices (note: generated with FLUKA)
- Rapid calculation of
 - radiation particle fluence and LET spectra.
 - radiobiological dose
- Available on ESA SPENVIS system
- Future work includes heavy ions contribution.

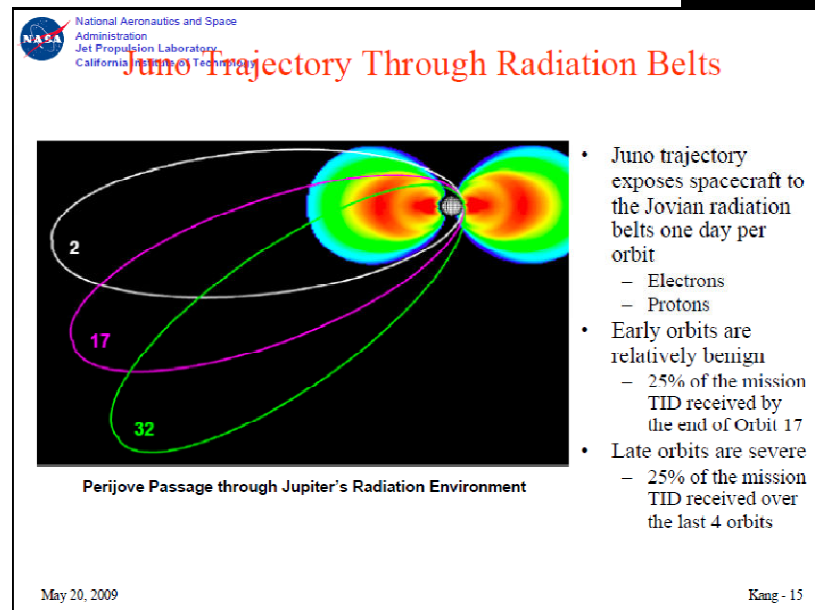
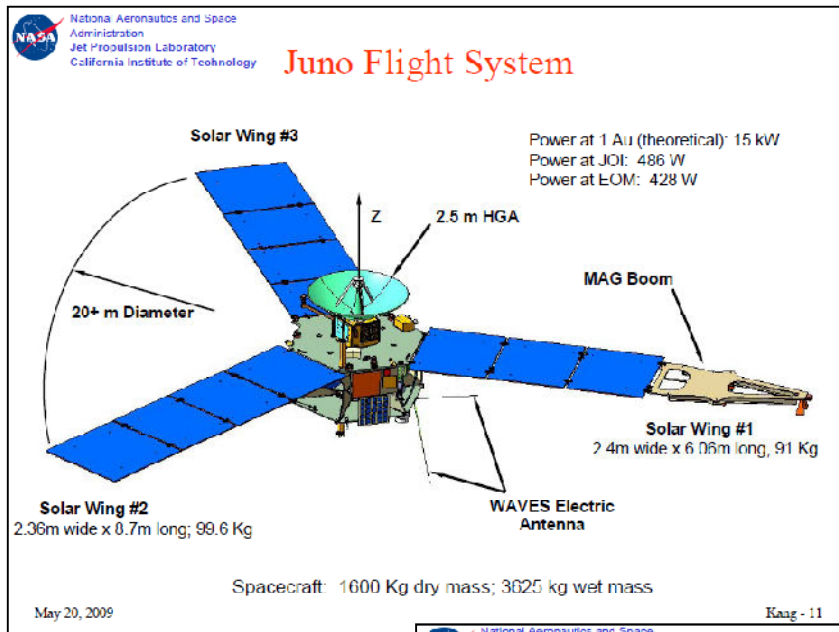
P. Truscott et al., NSREC 2008



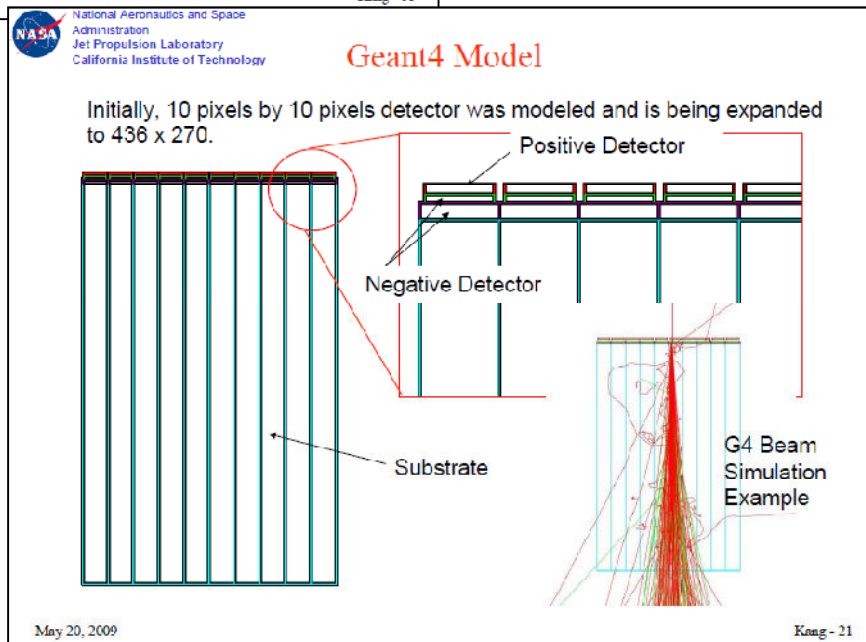
Neutron spectra (upwards):
average soil composition (solid) and water ice (dashed)



Jupiter - JUNO



- Shawn Kang,
- Insoo Jun,
- JPL



Jupiter - Galileo



NASA National Aeronautics and Space Administration
Jet Propulsion Laboratory
California Institute of Technology

Heavy Ion Counter (HIC)

- Monitor the flux of heavy ions (C to Ni) to gather information on this form of radiation.
- Compliment the science observation made by other Galileo instruments.
- Improve and extend observation made by the Voyager and Pioneer mission.

May 20, 2009 Kang - 32

NASA National Aeronautics and Space Administration
Jet Propulsion Laboratory
California Institute of Technology

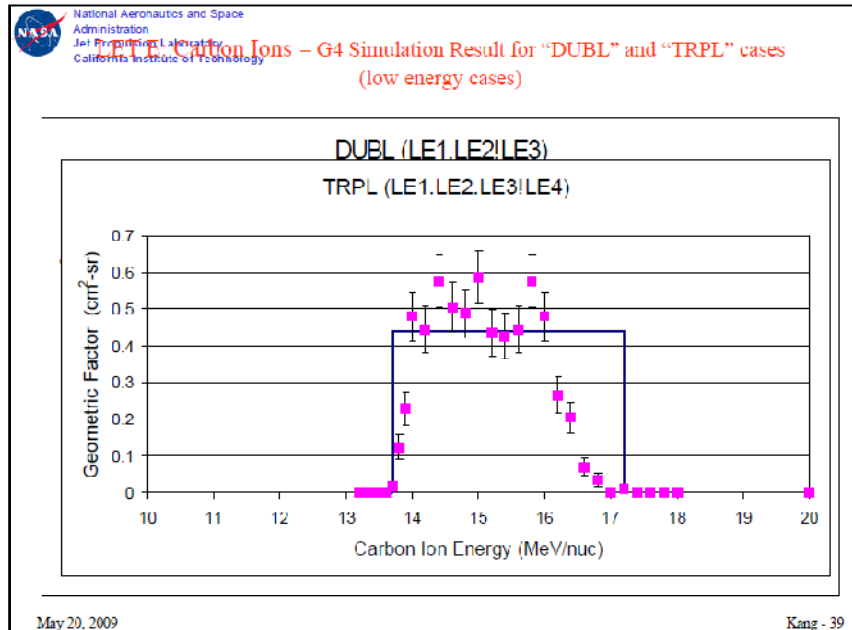
Geant4 Model

LET-B
Four Detectors
LB1, LB2, LB3, and LB4
Logic example
LB1.LB2.LB3.LB4
(Detected by LB1, LB2, LB3, but not by LB4)

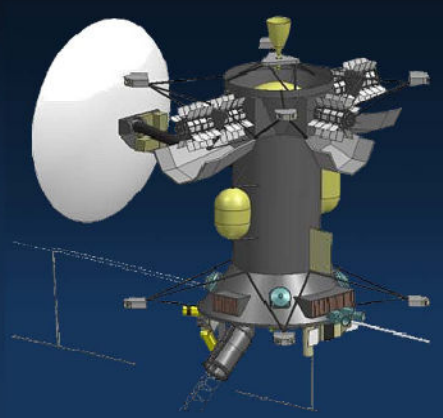
LET-E
Five Detectors
LE1, LE2, LE3, and LE5
Logic example
LE1.LE2.LE3
(Detected by LE1, LE2, but not by LE3, LE4, or LE5)

Kang - 35

■ Shawn Kang,
Insoo Jun,
JPL



Jupiter - JEO



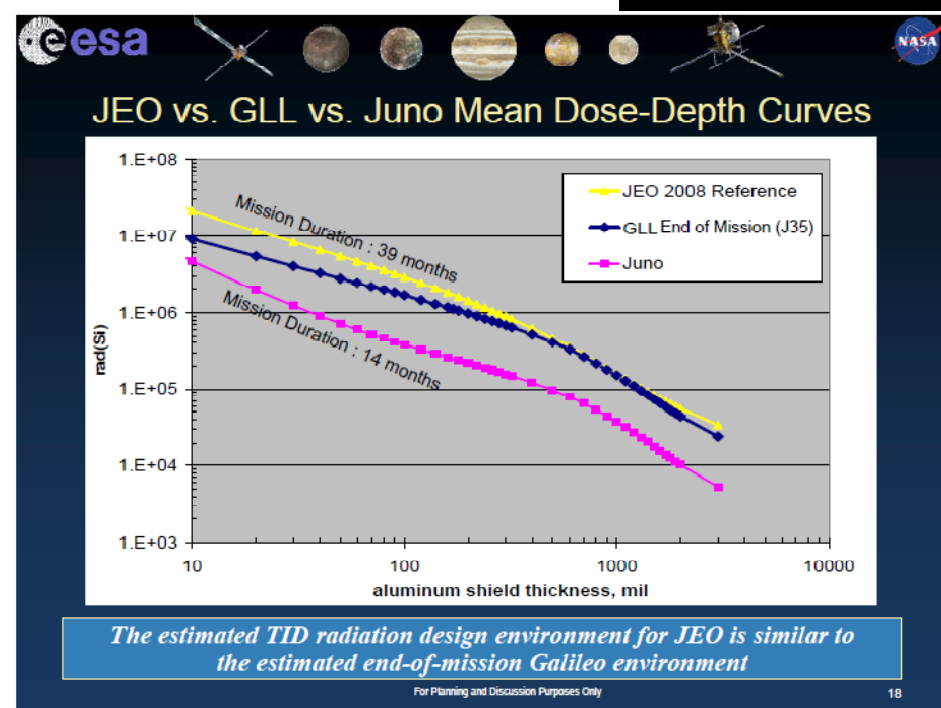
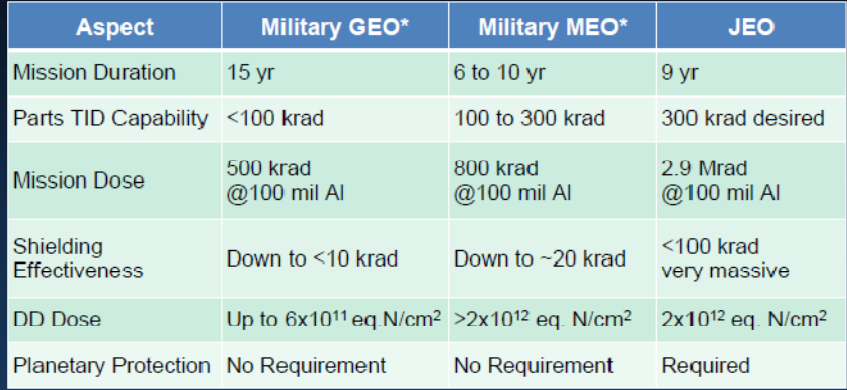
JEO Baseline Flight System

- Three-axis stabilized with instrument deck for nadir pointing
- Articulated HGA for simultaneous downlink during science observations
- Data rate of 150 kbps to DSN 34m antenna on Ka-band
- Performs 2260 m/s ΔV with 2646 kg of propellant
- Five MMRTGs would provide 540 W (EOM) with batteries for peak modes
- Rad-hardened electronics with shielding to survive 2.9 Mrad (behind 100 mil Al) environment
- 9-year lifetime
- Healthy mass and power margins (43%, >33% respectively)

JEO would be built upon minor modifications to a strong EE2007 design

For Planning and Discussion Purposes Only

- Insoo Jun, JPL
Jupiter Europa Orbiter

Parts Capabilities Radiation Requirements Comparison

Aspect	Military GEO*	Military MEO*	JEO
Mission Duration	15 yr	6 to 10 yr	9 yr
Parts TID Capability	<100 krad	100 to 300 krad	300 krad desired
Mission Dose	500 krad @100 mil Al	800 krad @100 mil Al	2.9 Mrad @100 mil Al
Shielding Effectiveness	Down to <10 krad	Down to ~20 krad	<100 krad very massive
DD Dose	Up to 6×10^{11} eq. N/cm ²	$>2 \times 10^{12}$ eq. N/cm ²	2×10^{12} eq. N/cm ²
Planetary Protection	No Requirement	No Requirement	Required

* GEO: Geostationary Earth Orbit; MEO: Medium Earth Orbit

JEO parts requirements are similar to MEO Military Satellites, though shielding limitations and planetary protection pose significant challenges

For Planning and Discussion Purposes Only

JORE²M²

Jupiter Radiation Environment & Effect Models and Mitigations



- <http://reat.space.qinetiq.com/jorem/>
- Engineering tools for the prediction of the environment and effects/mitigation analysis
- Proton and electron flux-maps in B-L* space for the complete Jovian environment
- Development of models for the energetic ion environment (helium, carbon, oxygen and sulphur)

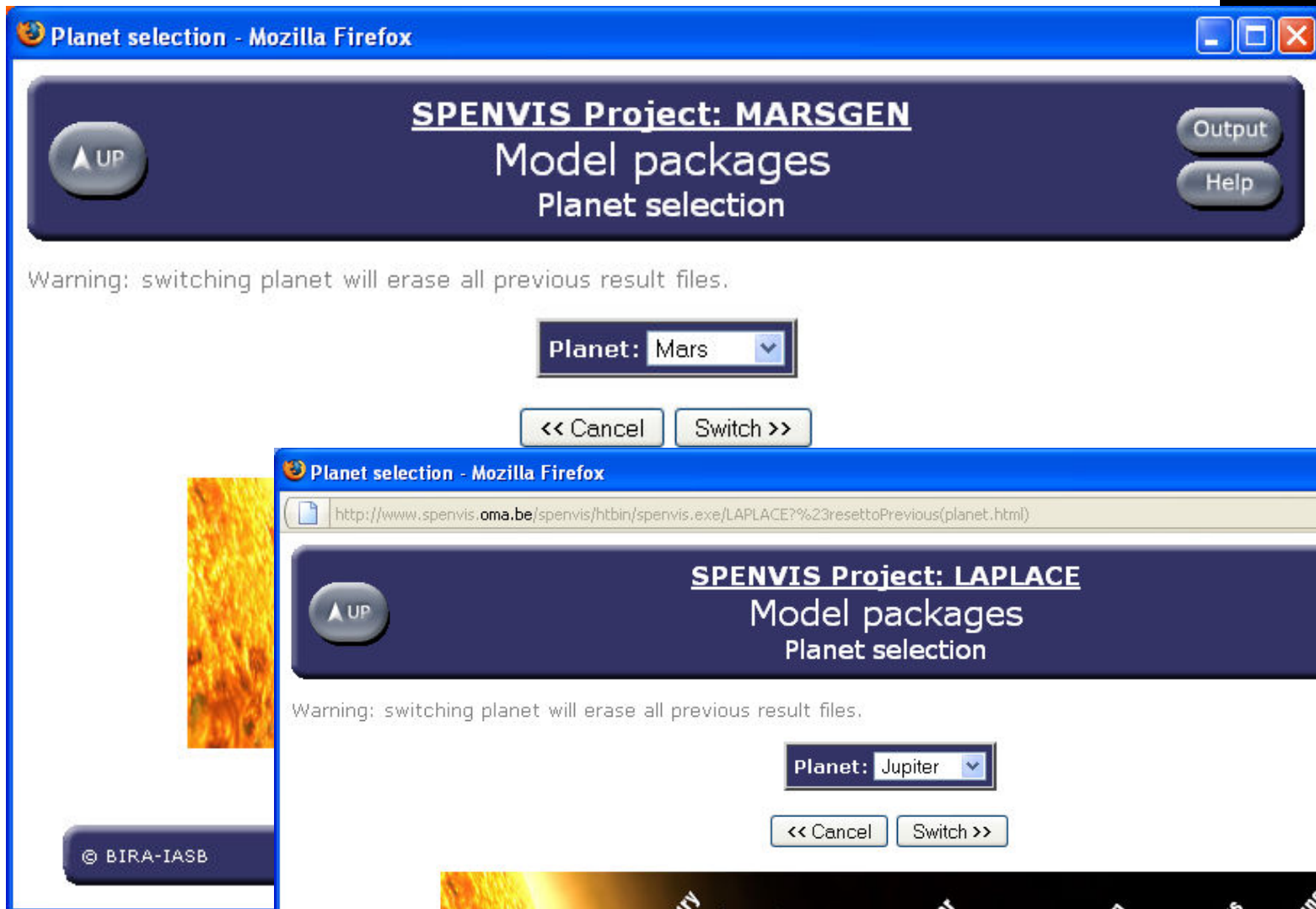
Of relevance here:

- Review of radiation effects analysis tools (not only Geant4-based)
- Implementation of updated version of PLANETOCOSMICS
 - trapped particle radiation incident upon the Galilean moons
 - including consideration of the Jovian and local fields.
- New tool based on genetic algorithms and MULASSIS
 - optimisation of radiation shields in combined e⁻ and proton environment.

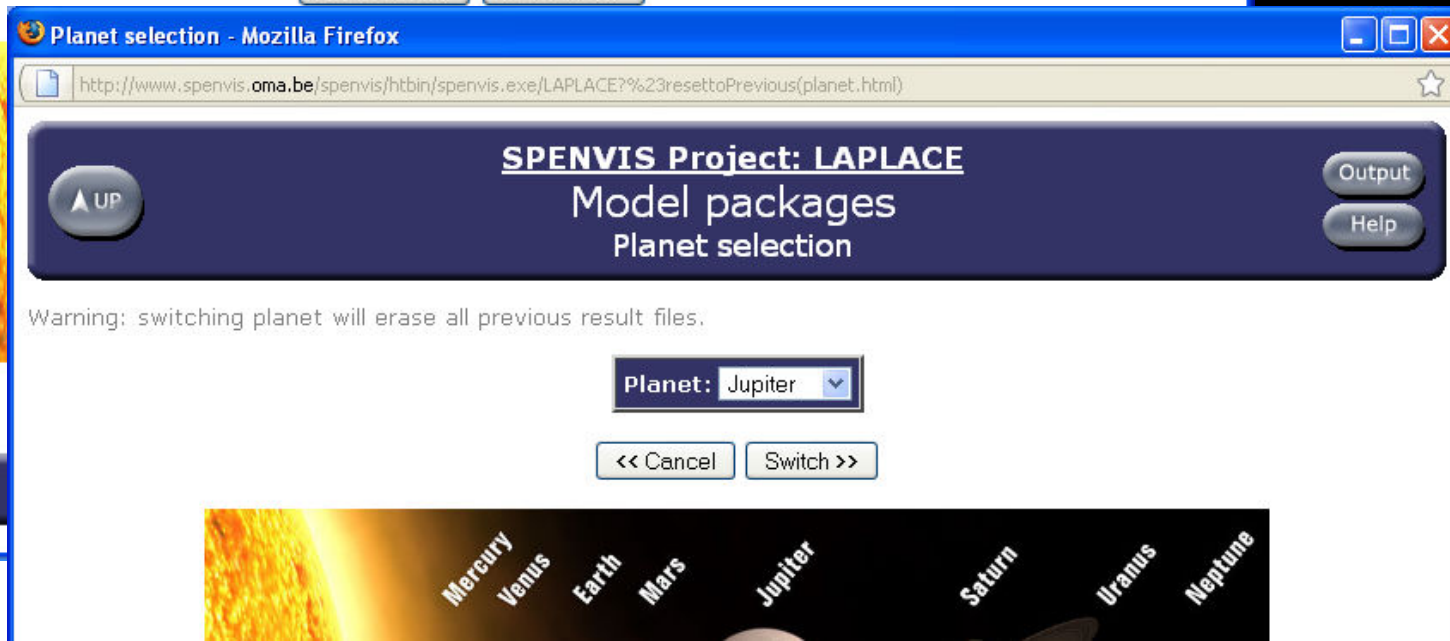
*ESA contract
(QinetiQ,
Onera, DHC)*

SPENVIS: other planets

- Mars, Jupiter



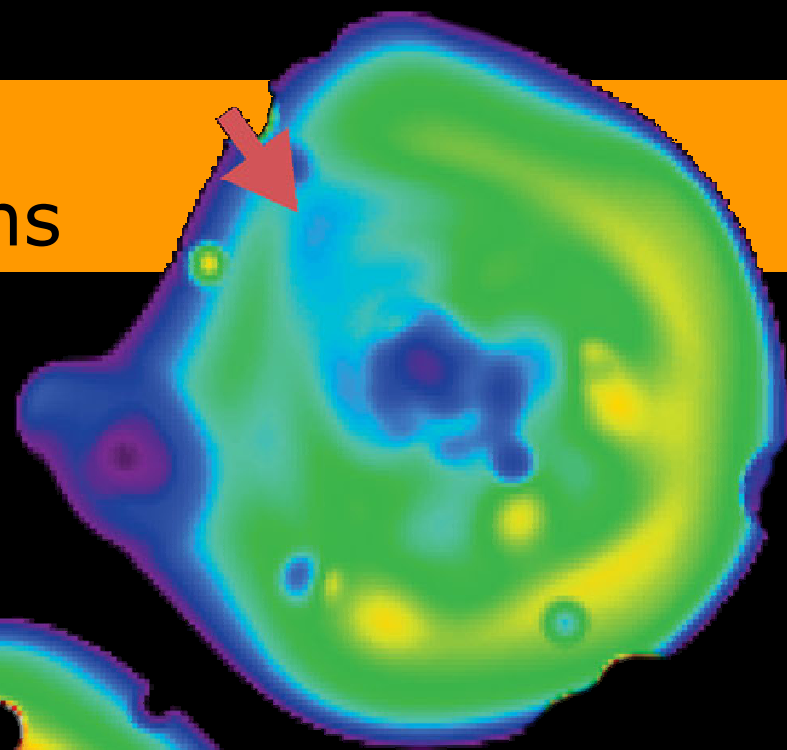
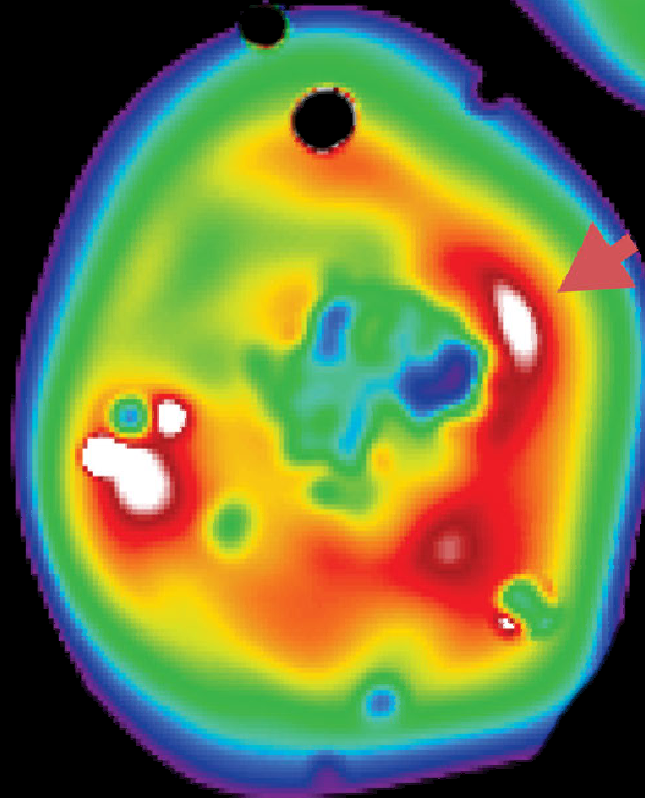
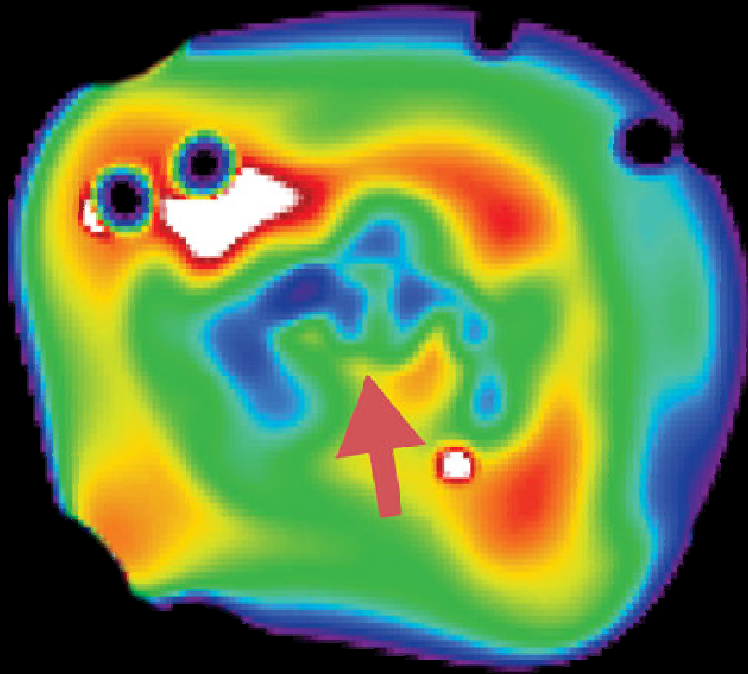
The screenshot shows the 'Planet selection - Mozilla Firefox' window for the 'SPENVIS Project: MARS GEN' model packages. The interface includes a title bar, a header with the project name and 'Model packages Planet selection', and buttons for 'Output' and 'Help'. A warning message states: 'Warning: switching planet will erase all previous result files.' Below this, a 'Planet:' dropdown menu is set to 'Mars', with '<< Cancel' and 'Switch >>' buttons. A small image of Mars is visible on the left side of the window.



The screenshot shows the 'Planet selection - Mozilla Firefox' window for the 'SPENVIS Project: LAPLACE' model packages. The interface includes a title bar, a header with the project name and 'Model packages Planet selection', and buttons for 'Output' and 'Help'. A warning message states: 'Warning: switching planet will erase all previous result files.' Below this, a 'Planet:' dropdown menu is set to 'Jupiter', with '<< Cancel' and 'Switch >>' buttons. A small image of Jupiter is visible on the left side of the window.

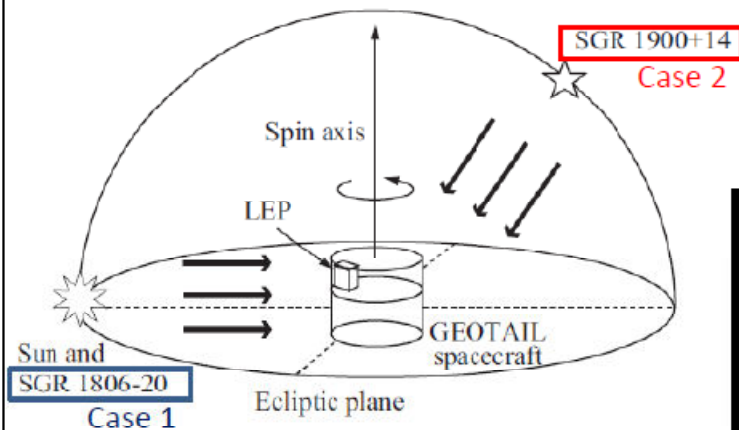
Geant4 Space Users' 2009

X- and gamma-ray missions

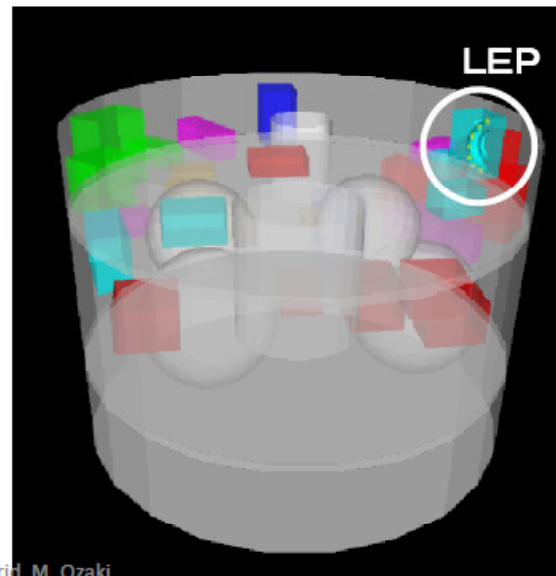


Geotail

Detection efficiency estimation by Geant4

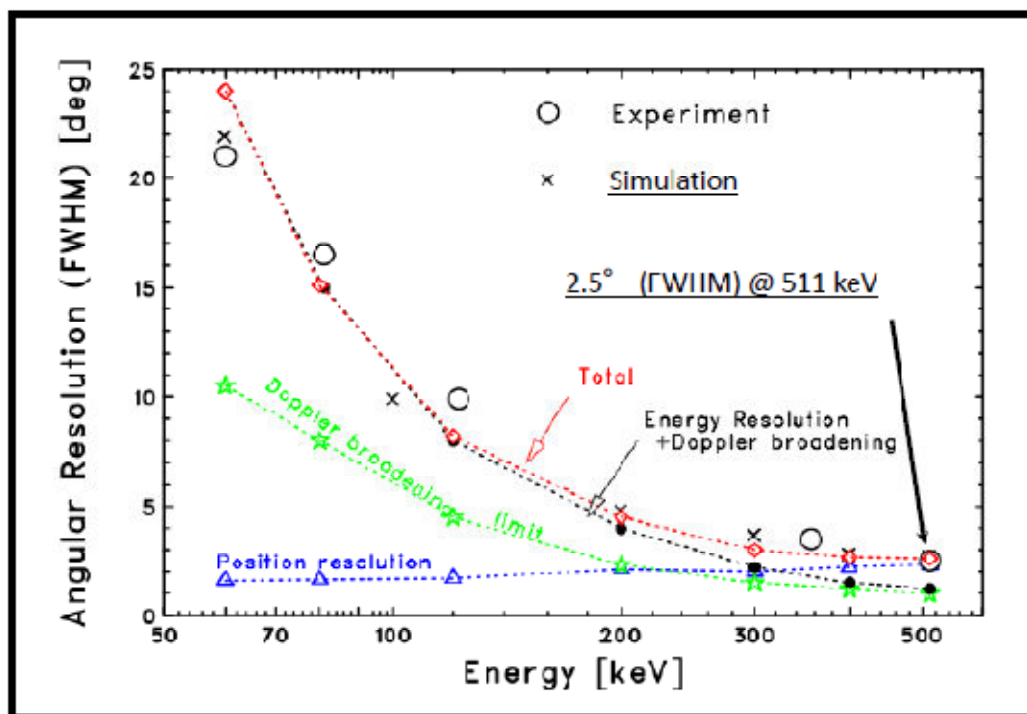


Celestial gamma-rays can
comes from any direction.



The lower energy range of MCP is
dominated by the shielding effect
of the spacecraft structure:
this changes direction by
direction.

Geant4 simulation: angular resolution

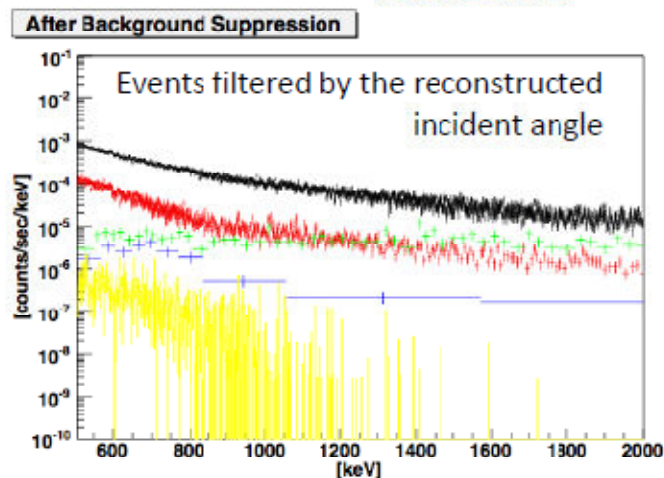
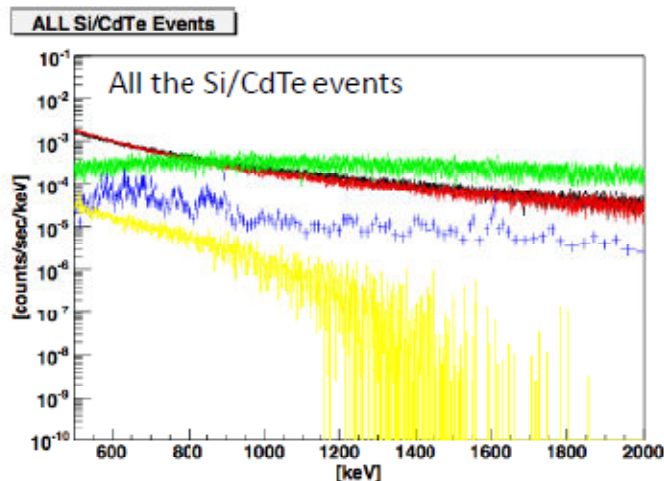
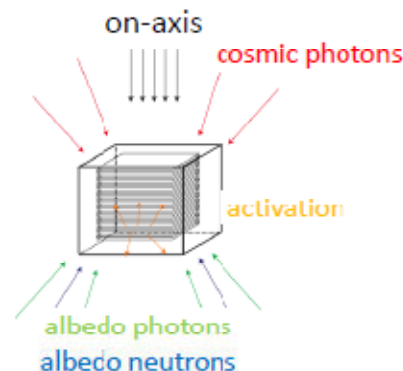


The energy dependency of the angular resolution is represented well.

- Position resolution, energy resolution and Doppler broadening are taken into account

Geant4 simulation: on-orbit BGDs

- BGD src: albedo neutrons, albedo photons, cosmic photons and activation by cosmic protons
- LEO (low activation by cosmic protons)



Geant4 Space Users' 2009

Single Event Effects

MRED

Applications of MRED

Robert A. Reed, Robert A. Weller,
Marcus H. Mendenhall

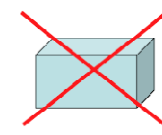
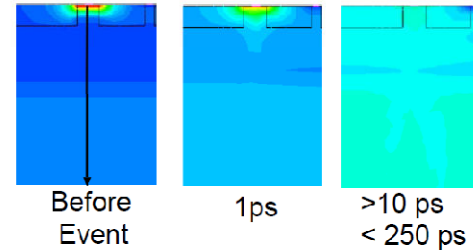


1

- Robert Reed, Vanderbilt University
- Geant4 integration with TCAD and SPICE

TCAD Simulation Results

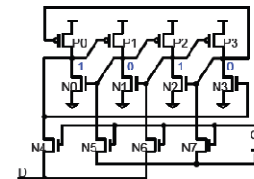
Potential Modulation: Small, Process-Isolated Diode



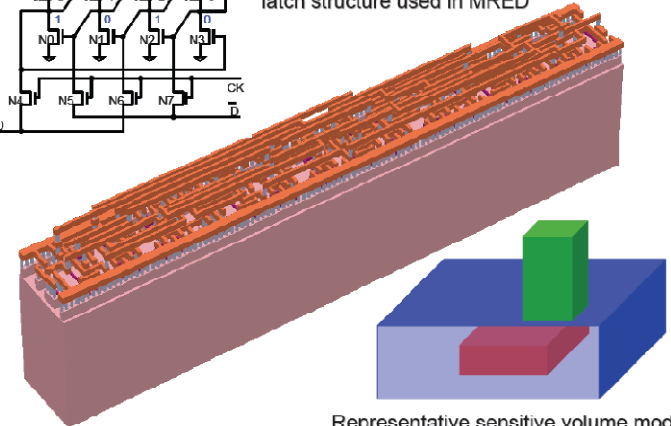
- Potential modulation controlled by process features
- Carrier density defines extent of modulation
- Carrier motion has implication on e-h cloud size

4

DICE Latch



Solid model of the latch structure used in MRED



Representative sensitive volume model for a single transistor used in MRED (each color is a different collection efficiency) 14

SEE basic mechanisms



Fundamental Aspects of Radiation Event Generation for Simulation of Space Electronics

R. A. Weller¹, M. H. Mendenhall¹, R. A. Reed¹, M. A. Clemens¹,
N. A. Dodds¹, B. D. Sierawski¹,
K. M. Warren¹, R. D. Schrimpf¹, L. W. Massengill¹,
T. Koj², D. Wright², and M. Asai²

¹Institute for Space & Defense Electronics, Vanderbilt University

²Stanford Linear Accelerator Laboratory, Stanford University
USA

Acknowledgements:

- DTRA 6.1: HDTRA1-08-1-0034 and HDTRA1-08-1-0033
- DTRA 6.2: Radiation Hardened Microelectronics Program
- NASA GSFC: NASA Electronic Parts and Packaging (NEPP) Program
- NASA MSFC: Radiation Hardened Electronics for Space Exploration (RHESE) Program

Robert A. Weller

Geant4 Space Users Workshop, Madrid, May 2009

1

General SEU Rate Prediction



The general rate expression:

$$R = - \sum_z \int_{All E} dE \int_{\hat{n} \cdot \hat{e} < 0} d\Omega \oint d\vec{A} \cdot \vec{\Phi}(z, E, \hat{e}, \vec{x}) \times P_u(z, E, \hat{e}, \vec{x})$$

Definitions: $p_c(l) = \frac{1}{\pi A} \int d^2x \int d\Omega (-\hat{e} \cdot \hat{n}(\vec{x})) UnitStep(-\hat{e} \cdot \hat{n}(\vec{x})) \delta(l - h(\hat{e}, \vec{x}))$

Assumptions:

$$\vec{\Phi}(z, E, \hat{e}, \vec{x}) = \Phi(z, E) \hat{e}$$

$$P_u(z, E, \hat{e}, \vec{x}) = \int P_d(z, E, \hat{e}, \vec{x}, E_d) \times P_u(E_d) dE_d$$

$$P_d(z, E, \hat{e}, \vec{x}, E_d) = \delta(E_d - h(\hat{e}, \vec{x}) \cdot S(z, E))$$

$$P_u(E_d) = UnitStep(E_d - E_c)$$

$$F(s) = \sum_i \sum_j \frac{\Phi(z, E_i)}{\left[\left(\frac{dS(z, F)}{dE} \right)_{E=E_i} \right]}$$

$$P_c(l) = \int_1^\infty p_c(x) dx$$

Math!

$$R(E_c) = \pi A \int_{s_{min}}^{s_{max}} F(s) \cdot P_c\left(\frac{E_c}{s}\right) ds$$

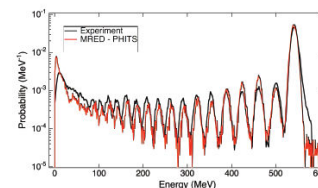
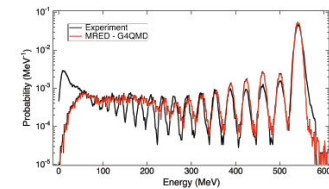
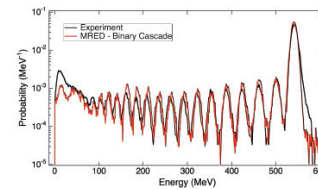
Robert A. Weller

Geant4 Space Users Workshop, Madrid, May 2009

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- later expanded for RADECS 2009 short course
- Bob Weller, Vanderbilt University

LET Spectra Comparison



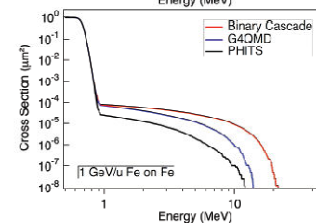
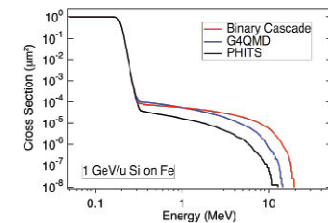
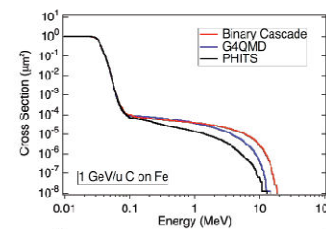
56 GeV Fe on C

Robert A. Weller

Geant4 Space Users Workshop, Madrid, May 2009

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Energy Deposition in an Fe Cube



**1 μm^3 sensitive volume
in a 5 μm cube.**

Robert A. Weller

Geant4 Space Users Workshop, Madrid, May 2009

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Geant4 Space Users' 2009

Uncertainties in standard practice and margins

Comparison and Validation of 3D Shielding Analysis Tools for Space Applications

Fan Lei, Pete Truscull
Aerospace Division, QinetiQ

Giovanni Santini and Petteri Nieminen
ESTEC, ESA

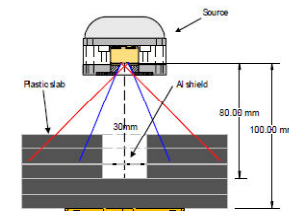
Geant4 Space User Workshop
Madrid, 21/05/2009

QinetiQ

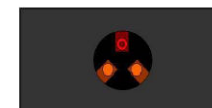


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REEF Experiment Setup



RADFET, REM TOT600 type from REM
Oxford Ltd.
TLDs: Harshaw EXTRAD type



QinetiQ



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3D Geometries & Detector locations

Shell/wall thickness: (mm)	0.1, 0.5, 1, 5, 10, 20
Overall dimensions:	
Spherical shell:	1m (Ø)
Box:	1m x 1m x 0.5m
Cylindrical shell:	1m (Ø) x 0.5m

The details of the geometry model used in the study.

Det.No.	1	2	3	4	5
X-pos (mm)	0	100	200	300	400
Distance to nearest surface	500	400	300	200	100

The detector locations in spherical shell geometries.

Det. No.	1	2	3	4	5	6	7	8	9	10	11
X-pos (mm)	0	100	200	300	400	0	0	100	200	300	400
Z-pos (mm)	0	0	0	0	0	100	200	200	200	200	200
Distance to nearest surface	250	250	250	200	100	150	50	50	50	50	50

The detector locations in box and cylindrical shell geometries.

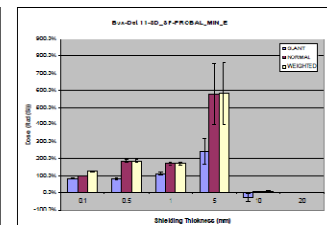
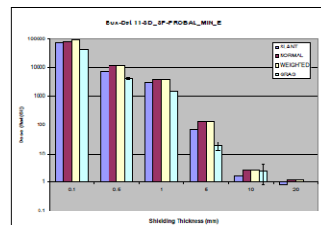
Created in GDML format,
used by both the SSAT
and GRAS tools

QinetiQ



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Results -2



QinetiQ

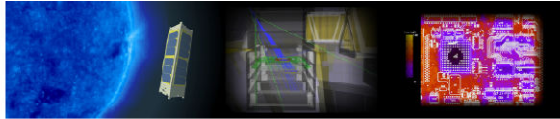


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Geant4 VS NOVICE adjoint and ray-tracing

6th Geant4 Space User's Workshop

Madrid, 19-22 May 2009

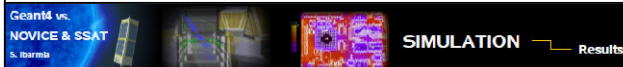


Forward G4 comparison with adjoint MC and Ray Tracing Techniques

S. Ibarmia⁽¹⁾, A. Rivera⁽¹⁾, S. Esteve⁽¹⁾, M. Vázquez⁽²⁾, J. Cueto⁽²⁾

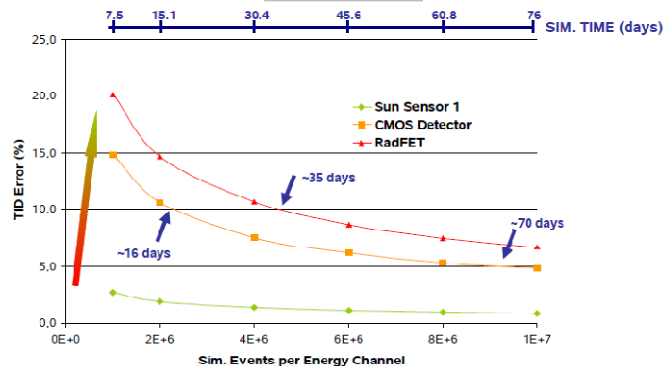
⁽¹⁾ Space Weather & Radiation Effects Lab, INTA

⁽²⁾ Radiation Analysis Section, Thales Alenia Space Spain



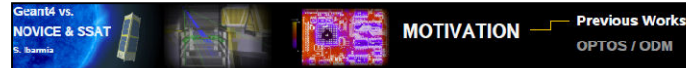
STATISTICS
EVOLUTION

CPU TIMING



6th G4 Space Users' Workshop

Madrid, 19-22 May 2009



G4 vs SSAT

R. Lindberg et Al.
(TNS Dec 2006)

NOVICE vs SSAT

P. Calvel et Al.
(NSREC 2003)

Simple
Complex

Slab, Sphere, Box
ConcXpress (CDML)

Sphere
Comm. S/C (FASTRAD)

SSAT
MC

(slant) + SHIELDOSE 2
Geant4

SSAT (slant/normal) + Dose Curve
NOVICE

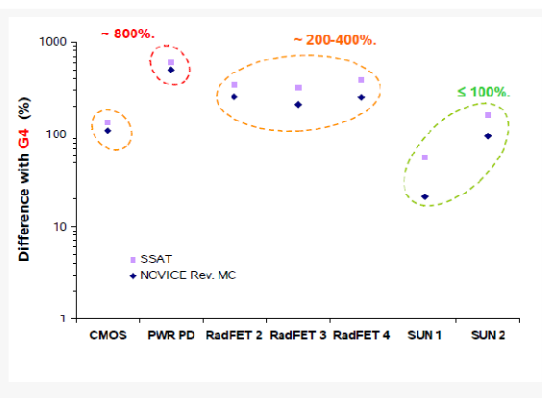
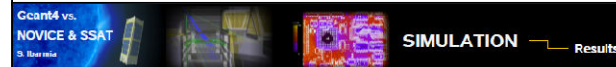
Discrepancies from
60% to ~400%
respect G4

Discrepancies from
Up to ~150%
respect NOVICE



6th G4 Space Users' Workshop

Madrid, 19-22 May 2009



6th G4 Space Users' Workshop

Madrid, 19-22 May 2009

Other published work

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File Edit View History Bookmarks Tools Help

http://geant4.esa.int/index.php/publications.html

Geant4

Space Users' Home Page

Geant4 Space Users > Publications

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Publications and reports

A selection of publications and reports related to the use of Geant4 in space is shown below. Please [contact the site administrator](#) if you would like a new item to be listed, or to suggest changes.

Title Search

10

by Title by Date by Author by Publication by URL

2009

Standard Radiation Environment Monitor - Simulation and Inner Belt Flux Anisotropy Investigation
Lulea University of Technology Master's Thesis, ISSN 1653-0187
M. Siegl

ESAF: Full Simulation of Space-Based Extensive Air Showers Detectors
arXiv:0907.5275v1 [astro-ph.IM]
C. Berat et al.

Simulation of ALTEA calibration data with PHITS, FLUKA and GEANT4
NIM B, in press
C. La Tessa et al.

Search for Cosmic-Ray Antiparticles using Balloon-borne and Space-borne Experiments
PhD thesis, RWTH Aachen University, Germany

News

2008-10-09 12:00

Reverse Monte Carlo and ion physics developments

ESA has kicked-off on 1st July 2008 a project aimed at further development of the Adjoint Monte Carlo technique...

[Read more...](#)

2008-09-12 17:00

6th Geant4 Space Users' Workshop

The 6th G4SUW will be held in Madrid, Spain, on 19-22 May 2009...

[Read more...](#)

Find: photos

Next Previous Highlight all Match case Reached end of page, continued from top

A generic X-ray tracing toolbox in Geant4

Giuseppe Vacanti, Ernst-Jan Buis, Maximilien Collon, Marco Beijersbergen, Chris Kelly
cosine, Niels Bohrweg 11, 2333 CA, Leiden, The Netherlands

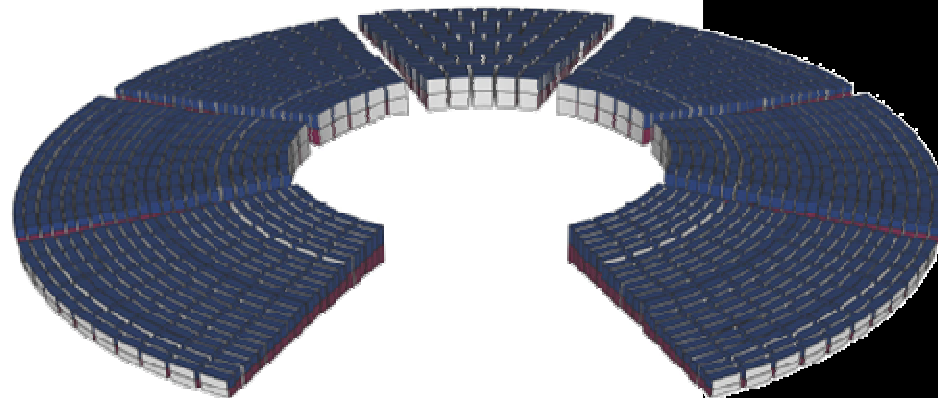
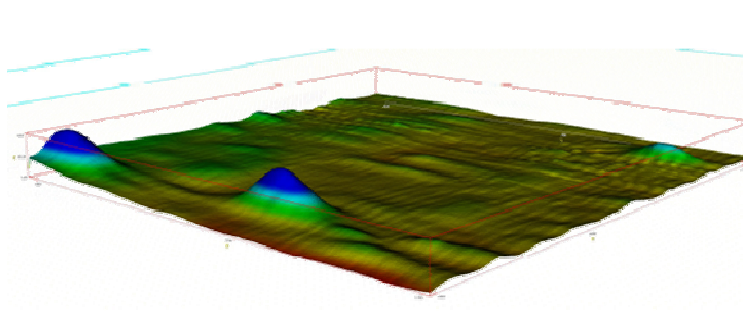
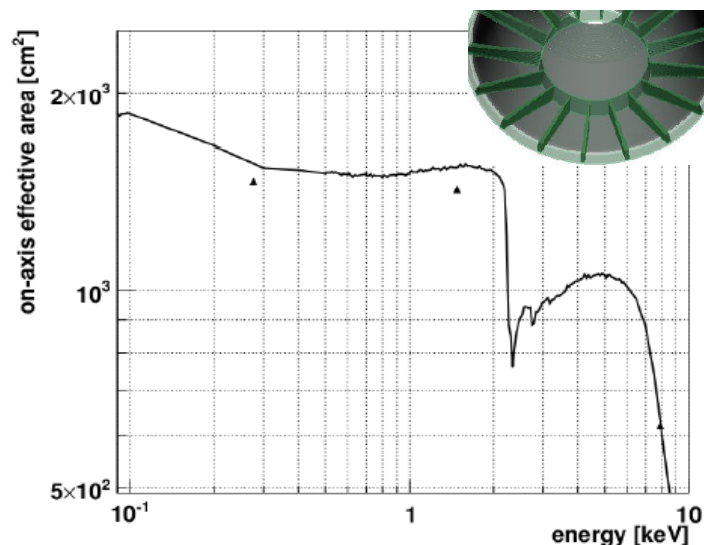
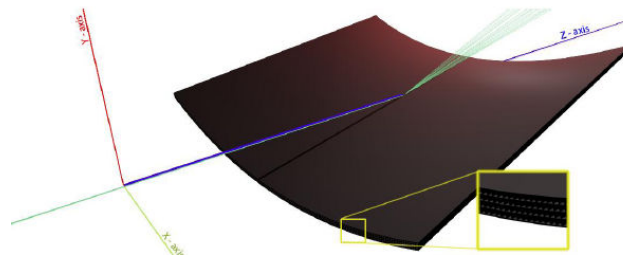
PROC. OF SPIE VOL. 7350 / 7350UC-1



Selected
publications

<http://geant4.esa.int>

- New X-ray reflection process:
G4XrayGrazingAngleScattering
- Validation with simplified XMM
 - Simulated effective area within 5% of measured one
- Metrology model input
- Applied to IXO mirror design
- Developments offered to the Geant4 collaboration

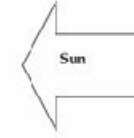
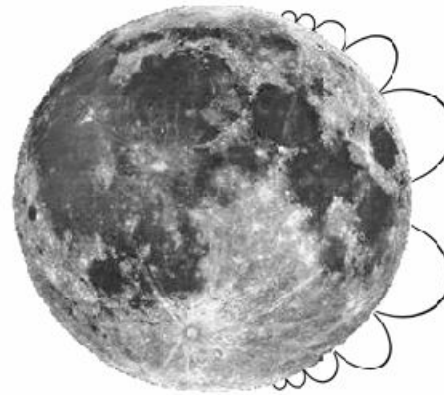


High Energy X-ray Spectrometer on Chandrayaan-1

P. Sreekumar^{1,*}, Y. B. Acharya², C. N. Umapathy¹, M. Ramakrishna Sharma¹, Shanmugam², A. Tyagi¹, Kumar¹, S. Vadawale², M. Sudhakar¹, L. Abraham¹, R. Kulkarni¹, S. Purohit², R. L. Premlatha¹, D. Banerjee², M. Bug¹ and J. N. Goswami²

¹ISRO Satellite Centre, Bangalore 560 017, India

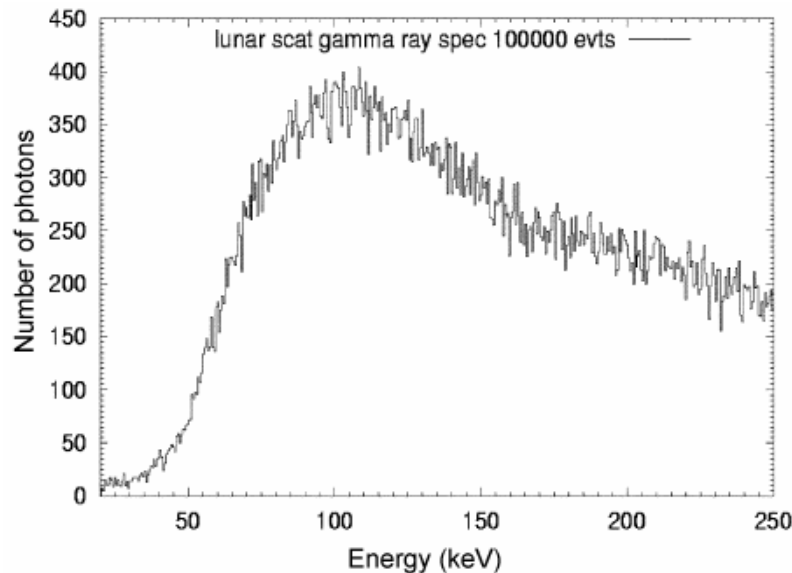
²Physical Research Laboratory, Ahmedabad 380 009, India



Selected
publications

<http://geant4.esa.int>

- India's first planetary exploration lunar mission
- High Energy X-ray spectrometer (HEX)
30–270 keV natural gamma rays from surface
- Transport of volatiles on the lunar surface
 - Detected from 46.5 keV line ^{210}Pb decay (from volatile ^{222}Rn)



- Geant4: continuum from GCR interaction in regolith
- High energy tail attenuated by Compton in soil
- Rejection of partial energy deposit events (Compton) by CsI(Tl) anticoincidence

Gamma-ray detector on board lunar mission Chang'e-1

J. CHANG^{1*}†, T. MA¹, N. ZHANG¹, M.S. CAI¹, Y.Z. GONG¹, H.S. TANG¹, R.J. ZHANG¹, N.S. WANG¹, M. YU¹, J.P. MAO¹, Y.L. ZHOU², J.Z. LIU², A.A. XU³ and L.G. LIU³

¹Purple Mountain Observatory, 2 West Beijing Road, Nanjing 210008, China

²National Astronomical Observatory, 20A Datun Road, Beijing 100012, China

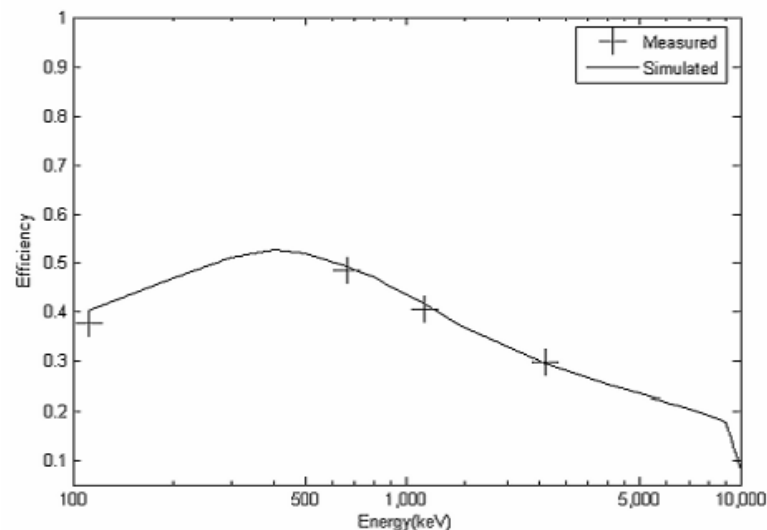
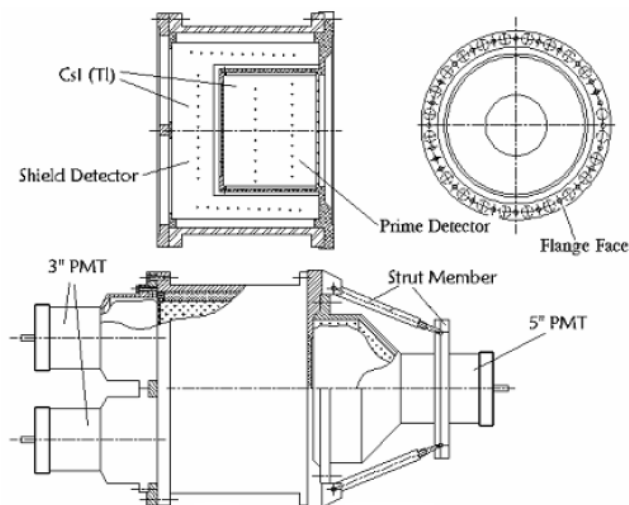
³MACAU University of Science and Technology, Avenida Wai Long, Taipa, Macau, China



- First Chinese lunar mission
Launched in 24th October 2007

GRS instrument

- Study of chemical element distribution on the Moon surface
- Experimental calibration
 - Gamma-ray sources
 - Geant4 validation

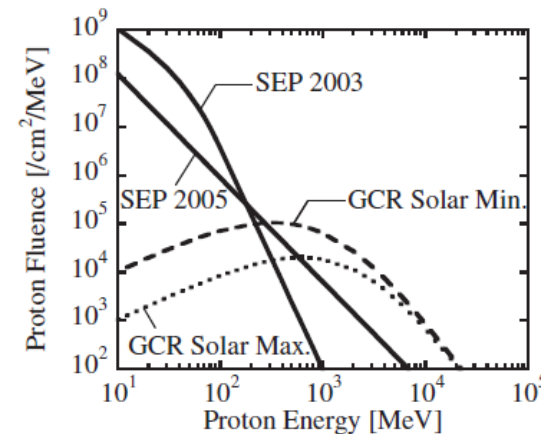


HZE Particle and Neutron Dosages from Cosmic Rays on the Lunar Surface

Kanako HAYATSU*, Makoto HAREYAMA, Shingo KOBAYASHI, Naoyuki YAMASHITA,
Kunitomo SAKURAI and Nobuyuki HASEBE

*Research Institute for Science and Engineering, Waseda University,
3-4-1 Okubo, Shinjuku, Tokyo 169-8555, Japan*

Proc. Int. Workshop Advances in Cosmic Ray Science
J. Phys. Soc. Jpn. **78** (2009) Suppl. A, pp. 149-152
©2009 The Physical Society of Japan



Selected
publications

<http://geant4.esa.int>

- Ambient dose equivalent
at Moon surface
- Galactic Cosmic Rays
 - Secondary neutrons / gammas
- SEP protons and ions
- Shielding effectiveness

	Al shield thickness [g/cm ²]	$H^*(10)_{SEP}$ [mSv]		
		H	He	O
SEP 2003/10/28	0.0	2045.7	113.0	30.8
	1.0	1178.4	53.4	2.5
	2.0	663.7	24.4	0.0
	5.0	287.4	7.0	0.0
	10.0	3.3	0.0	0.0
SEP 2005/1/20	0.0	216.7	16.8	11.0
	1.0	155.1	12.0	5.0
	2.0	113.6	8.7	0.2
	5.0	82.4	6.3	0.3
	10.0	10.2	0.8	0.1
GCR Solar Min.	0.0	39.6	17.5	110.0
	1.0	39.7	17.5	121.2
	2.0	39.7	17.4	31.4
	5.0	41.0	18.2	37.5
	10.0	32.0	11.8	22.0

	$H^*(10)$ [mSv/yr] for GCRs			
	highland region		mare region	
	Solar Min.	Solar Avg.	Solar Max.	Solar Avg.
Primary	758.5	519.8	265.8	519.8
Neutron	72.9	51.9	25.4	56.3
Gamma Ray	3.3	2.5	1.6	2.5
Total	834.7	574.2	292.8	578.6

$H^*(10)$ [mSv] for SEPs	
highland region	
2003	2005
2189.5	246.4
5.9	–
0.9	–
2196.3	246.4

More papers...

- Simulation of ALTEA calibration data with PHITS, FLUKA and GEANT4
NIM B, in press
C. La Tessa et al.
- Instrument Simulation for the Analysis of Cosmic Ray Electron with the FERMI LAT
Proc. of the 31st ICRC, Łódź
C. Sgrò et al.
- ...

Enabling developments for space applications

■ Goal

“reinforce the coordination of existing and planned space environments and effects related activities in Europe, through the implementation of a coherent European programme of activities in the domain.”

■ Roadmap on radiation analysis tools includes

- Coordination in the **development of tools for radiation effects analysis**, supporting the development of infrastructure to **support real-time analysis and/or forecast** and ultimately ensuring that users have access to well documented and **validated tools**

■ Usability, accuracy, speed are high priority requirements



Accuracy:

Low energy ion stopping power – range

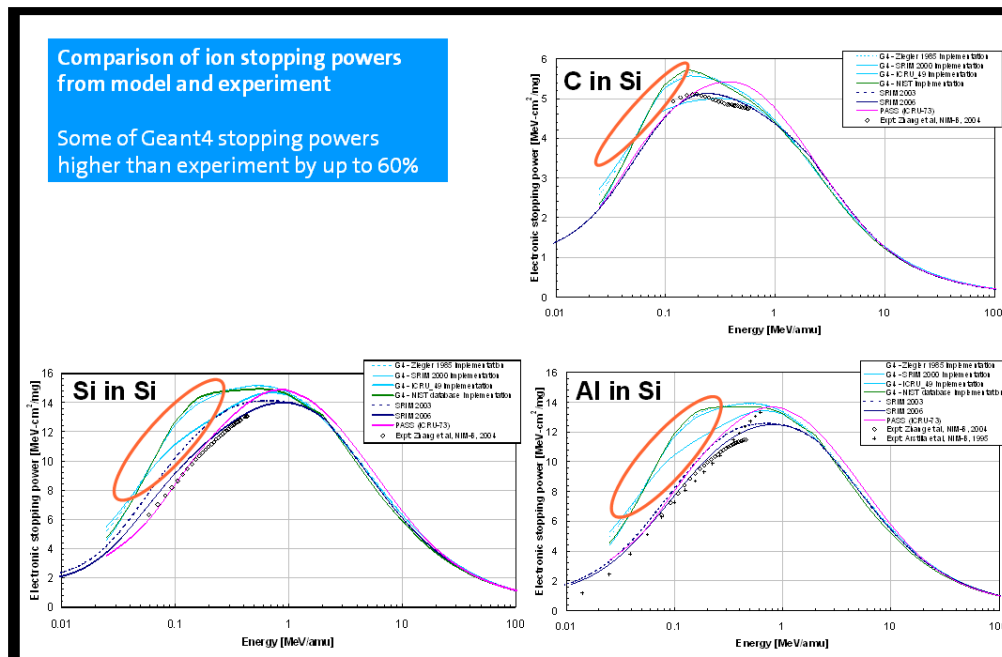


ESA contract:

RRMC

(QinetiQ,
SpaceIT,
TRAD)

- Impact e.g. on
 - SEE ground testing of EEE components
 - Recoil ion contribution to SEE prediction
- Geant4 review (April 2007)
 - “Recommendation 2: We recommend rapid integration of the ICRU 73 heavy ion stopping power model”
- Ion physics simulation requirements (Pete Truscott, QinetiQ)
 - Performance of available Geant4 models
 - Overview of other models / data
 - Significant differences



ICRU-73 implementation

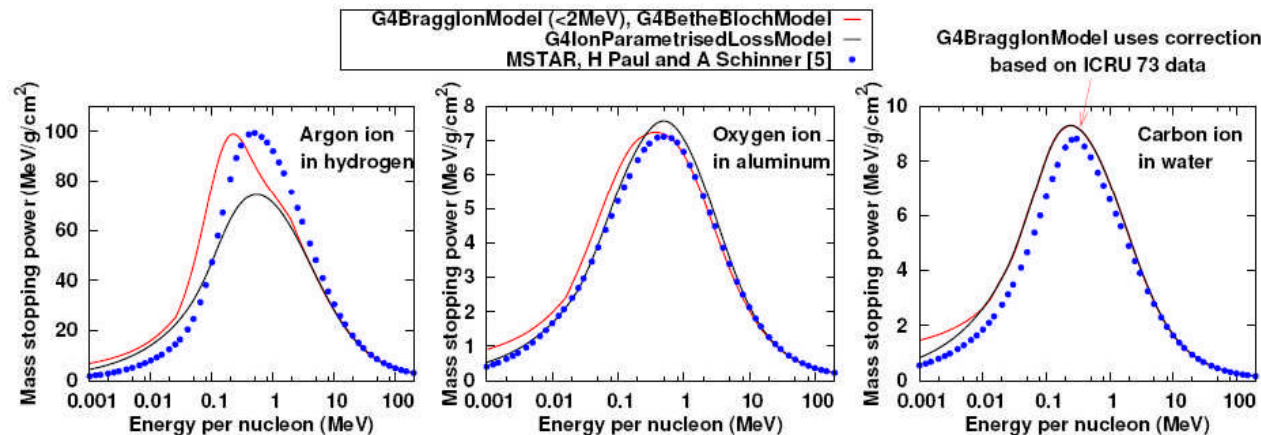


ESA contract:

RRMC.

- ICRU-73 tabulated stopping powers (PASS code results)
 - 16 incident ions $2 < Z < 19$, plus Iron $Z=26$
 - Simple and composite materials of interest for space applications
- New Geant4 “models” (V. Ivantchenko, A. Lechner)
 - Beta version released within Geant4 release 9.2
 - Poster at IEEE NSS 2008

■ First comparisons



- New models also available as option in GRAS physics list
- Future areas for improvement identified
 - Straggling models
 - Effective charge approach
 - Delta ray production at low energies
- Requirements include projectiles and materials not included in ICRU-73 publication
 - Collaboration started with Peter Sigmund (author of PASS code)

Enabling technologies: Reverse MC



Requirement from space industry

- Tallying in sub-micron SV inside macroscopic geometries

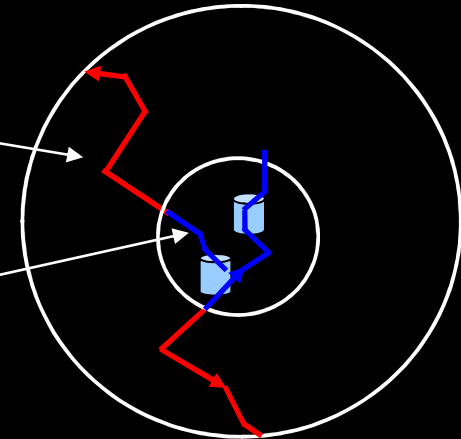
Simulation approach:

- Reverse tracking from the boundary of the sensitive region to the external source
Based on “adjoint” transport equations
- Forward tracking through the SV to compute the detector signal
Same code than in a forward MC simulation

Computing time focused on tracks that contribute to the detector signal

- Implemented Reverse Processes:
 - Ionisation for e^- , protons, ions (with delta-ray production, continuous energy loss and multiple scattering)
 - e^- Bremsstrahlung
 - Gamma: Compton scattering, Photo-electric effect

Laurent Desorgher (Space IT)

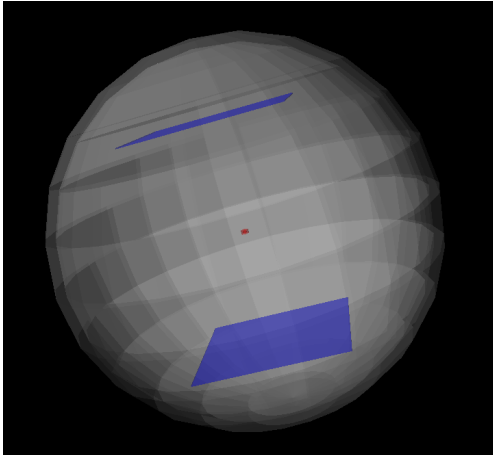


- Capability added to GRAS

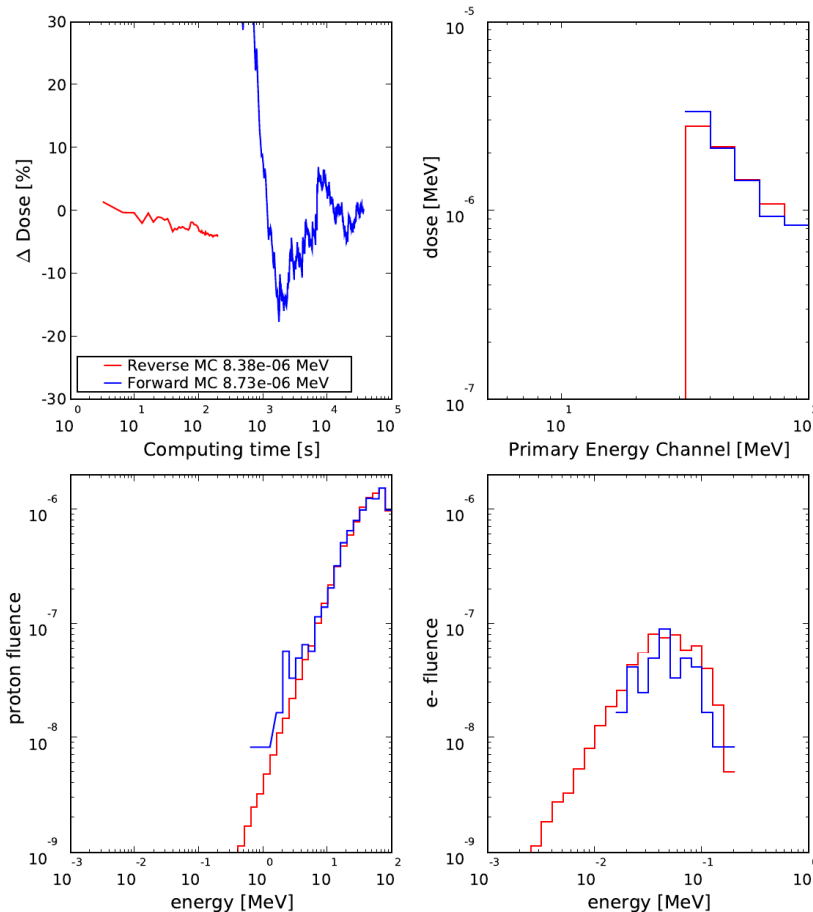
ESA REAT-MS and RRMCM contracts

Partly already included in Geant4 9.2

Reverse MC: comparison VS forward Protons, simple geometry



- Difference in total computed dose $< \sim 5\%$
- Reverse MC method is for this case 100-500 times more rapid than the forward MC method
- Realistic space component dosimetry involve sub-micron SV in $\sim 1\text{m}$ spacecraft



Enabling technologies: CAD geometry interface (and 3D modelling GUI)



ESA contract:

*REAT-MS
(QinetiQ, TRAD,
eta_max)*

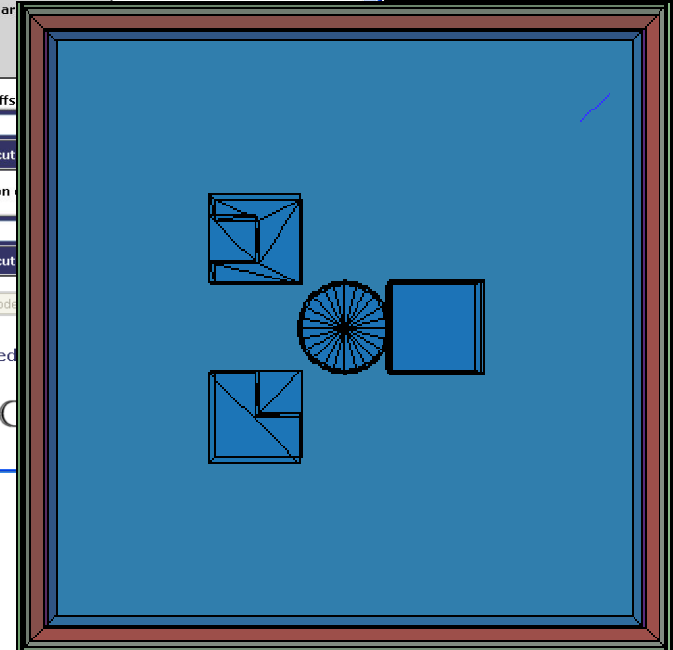
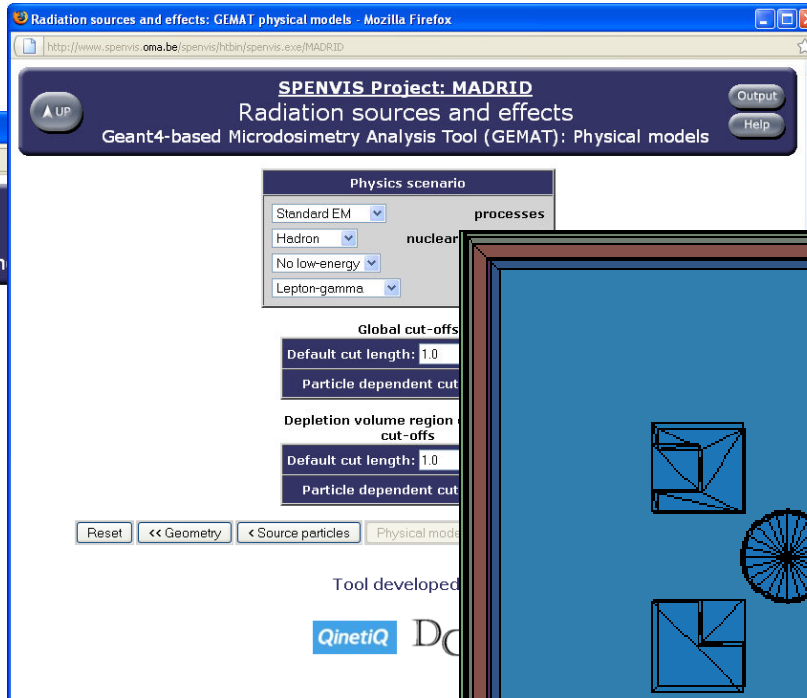
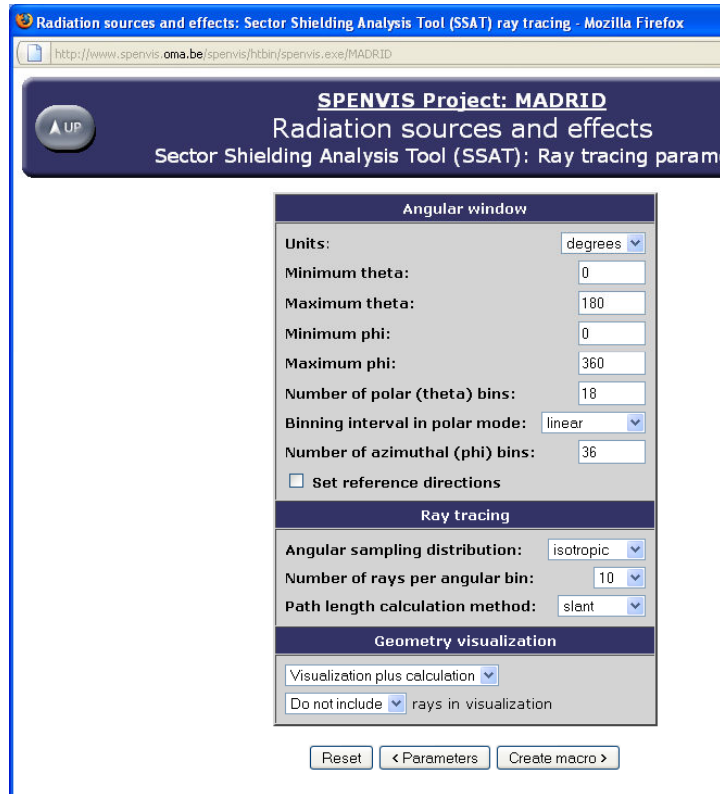
- CAD (ESA REAT-MS)
 - Using G4TessellatedSolid by P.Truscott
 - [Old prototype used to require ST-Viewer commercial S/W
 - GDML module to read ST-Viewer files]
 - New: **CAD STEP interface (and normal 3D models)**
via external 3D modelling tools tools
(ESA contract REAT-MS-2)
Direct GDML output
- STEP for Space Environment standard (STEP-SPE)
- GDML upgrade (CERN)
 - Tetrahedron and Tessellated volumes, modular models, loops
- **FASTRAD, ESABASE2**
 - GUI for 3D modelling
 - GDML output
 - Licensing policy under discussion



Usability:

SSAT, MULASSIS, GEMAT, GRAS

SPENVIS interfaces

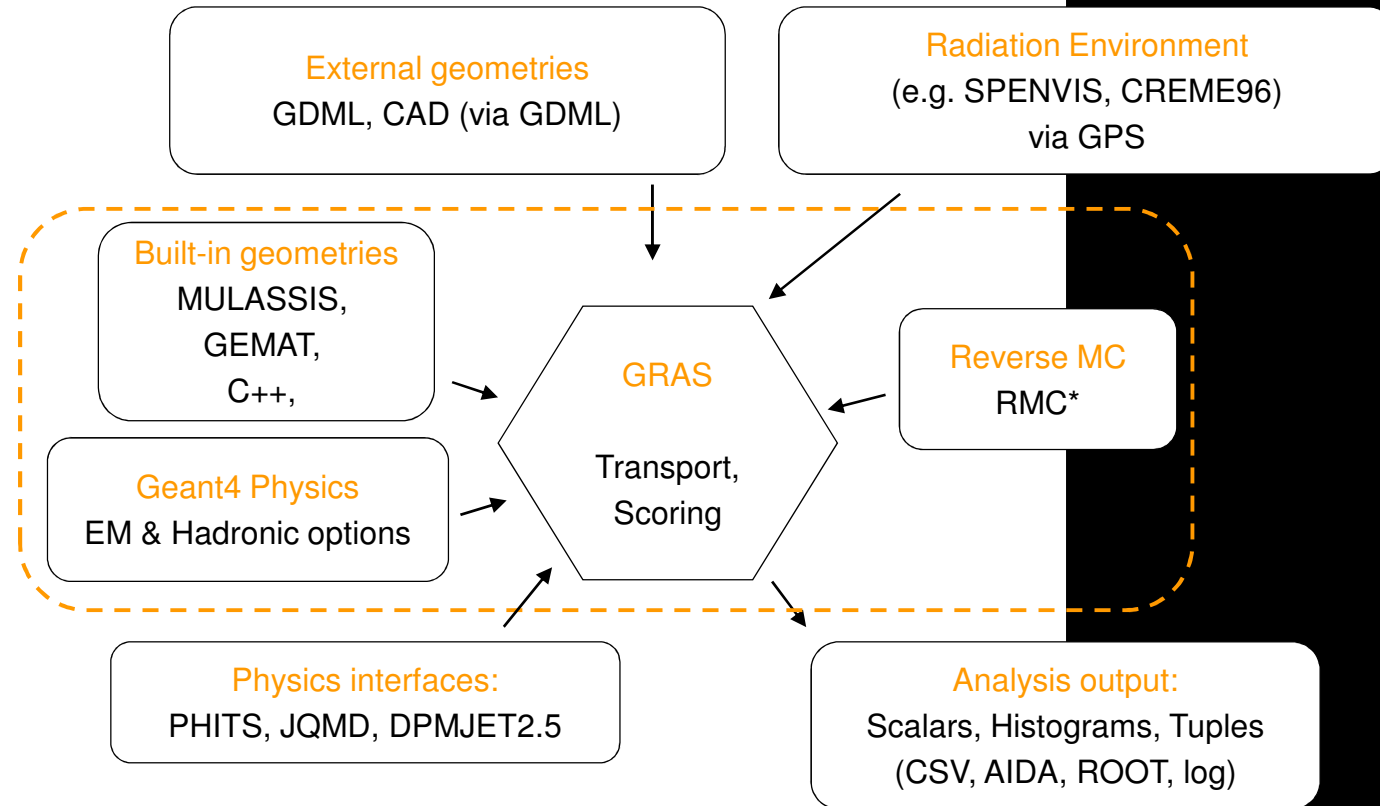


- GRAS interface under development

Usability: GRAS

Requirements:

- Ready-To-Use tool
Multi-mission approach
- Quick assessments
Ray-tracing ↔ MC
1D ↔ 3D
EM ↔ Hadronics
LET ↔ SV details



G Santin, V Ivantchenko et al, IEEE Trans. Nucl. Sci. 52, 2005

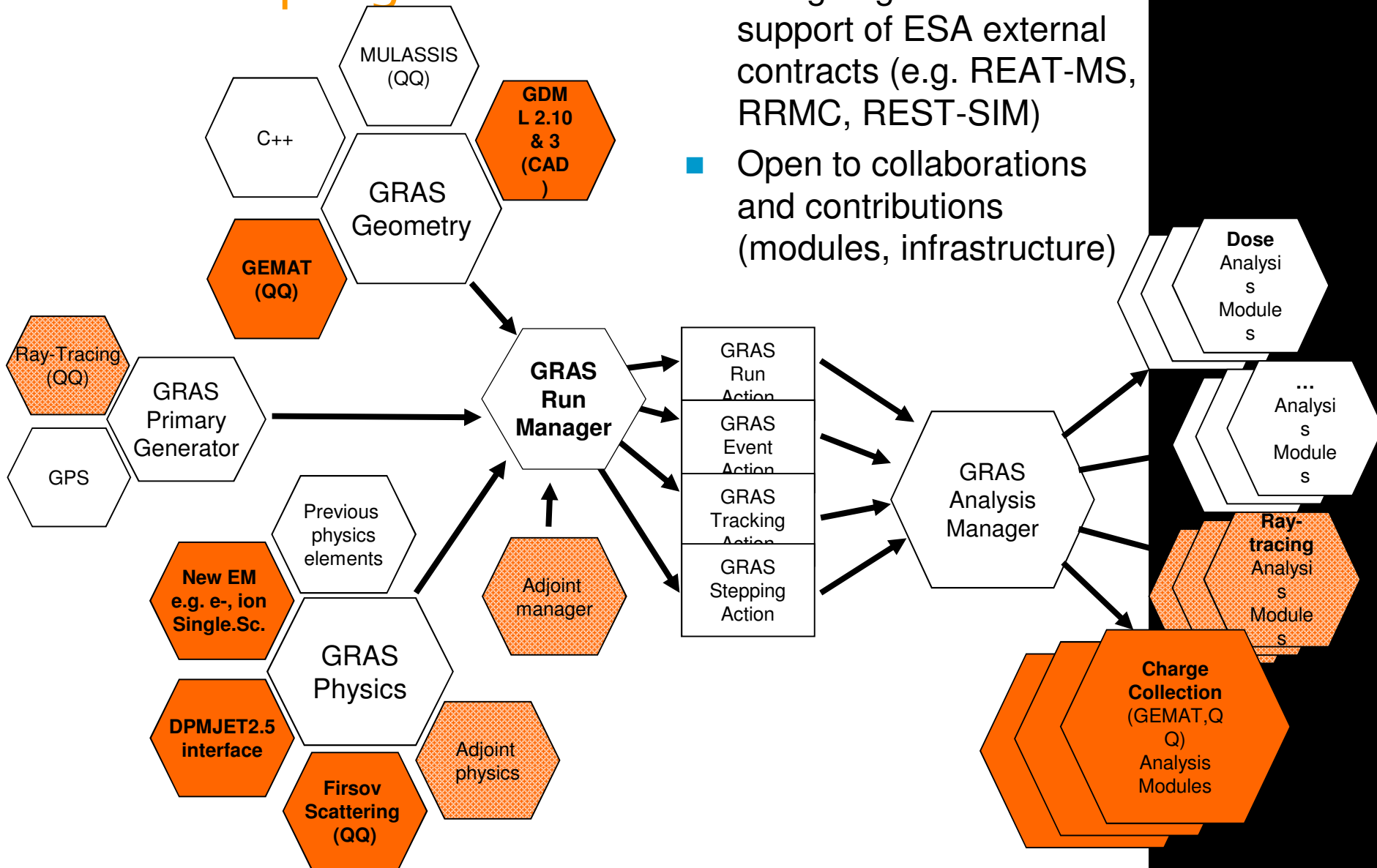
<http://space-env.esa.int/index.php/geant4-radiation-analysis-for-space.html>

** in progress*

GRAS (v2.4)

Modular progress

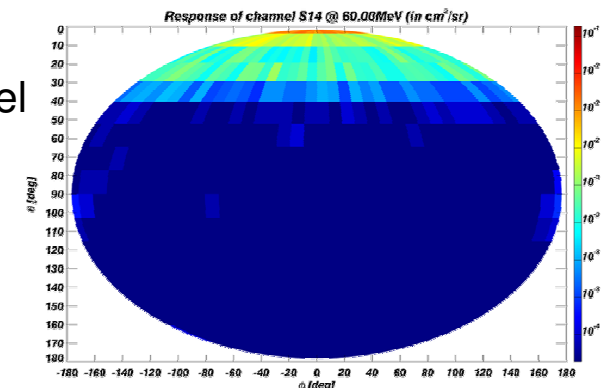
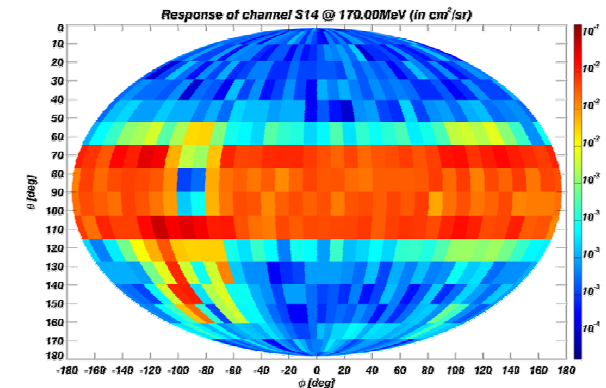
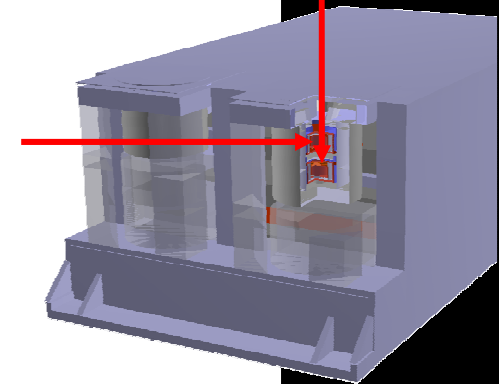
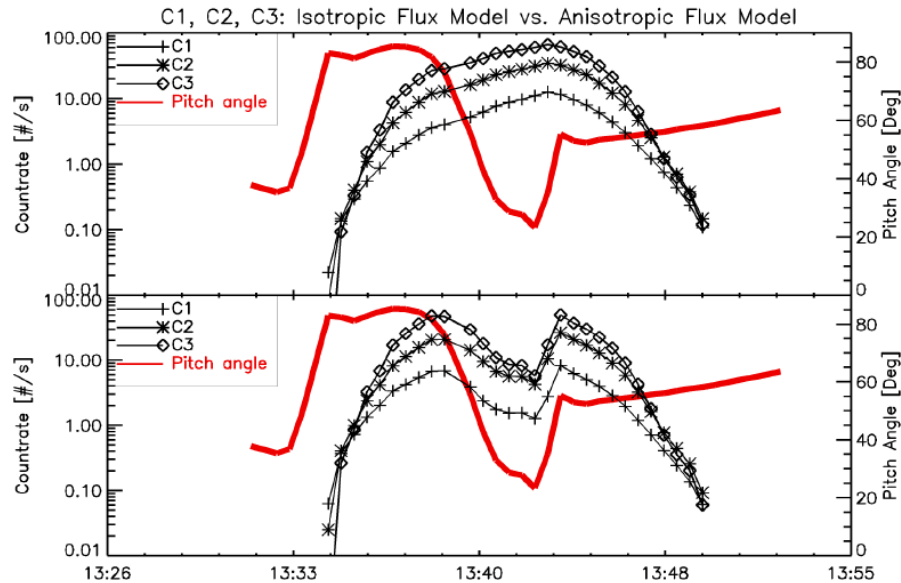
- ESA facility
- Being augmented with support of ESA external contracts (e.g. REAT-MS, RPMC, REST-SIM)
- Open to collaborations and contributions (modules, infrastructure)



SREM Response (Proba-1)

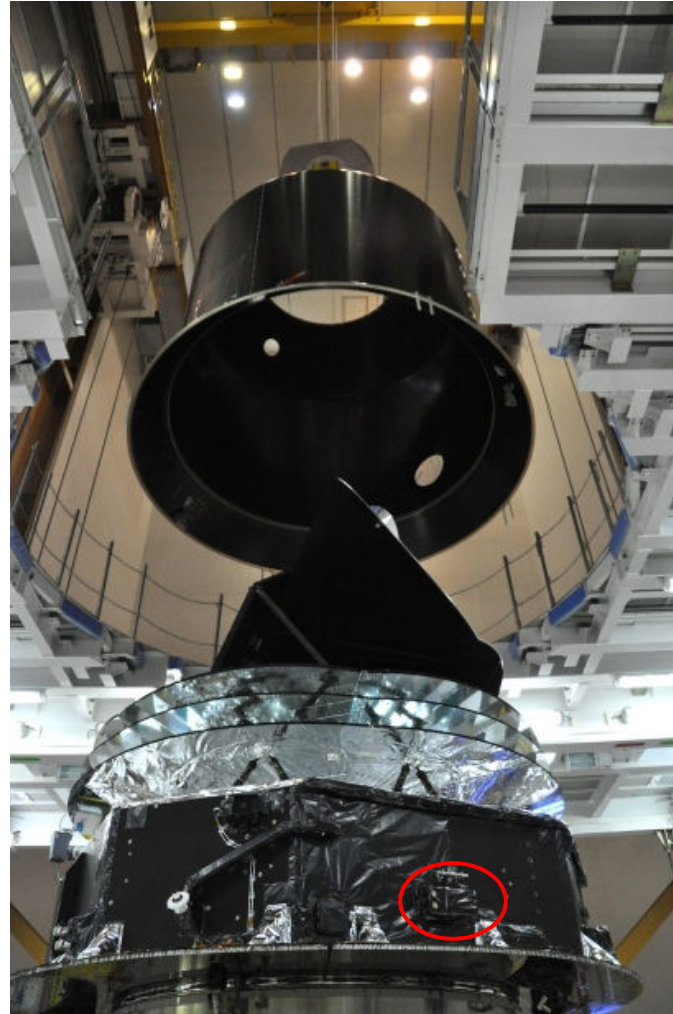


- Directional response function for all output channels
- Geant4 / GRAS simulations



- Inner Belt Anisotropy Investigations
 - AP-8 and Badhwar-Konradi pitch angle distribution model
 - Comparison with observations onboard PROBA-1
- Martin Siegl's Master's Thesis, 2009
- Presented at RADECS 09, submitted to IEEE TNS

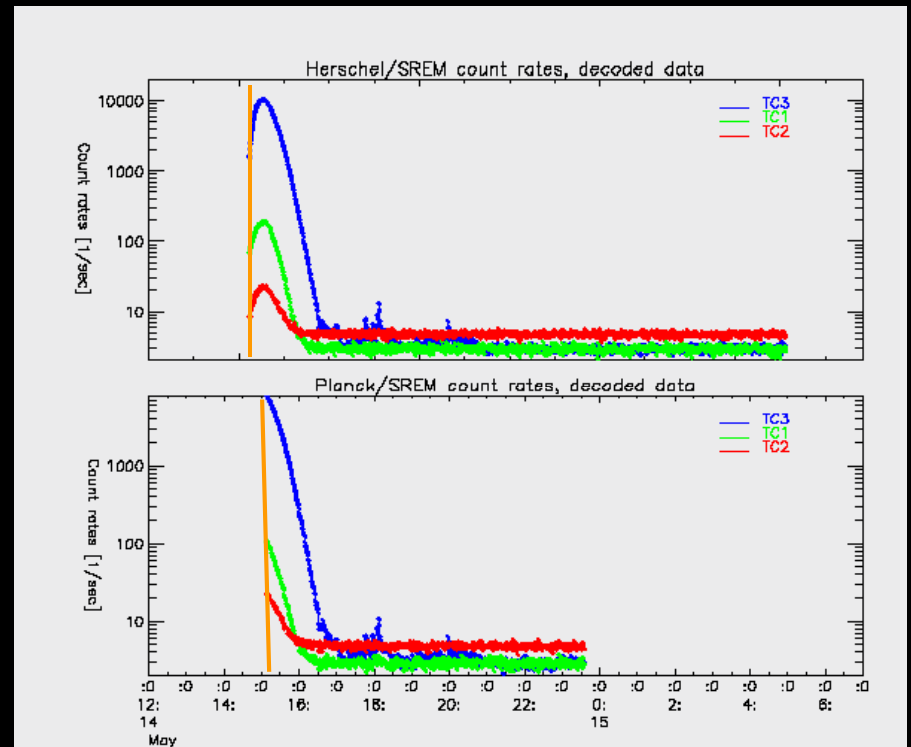
Herschel and Planck



Herschel and Planck



Launch 14.5. 2009



First SREM data 14.5. 2009

Summary

- Wide and diverse Geant4 space users' community
 - Interests ranging from cosmology, to planetary science, to spacecraft electronics engineering
 - In all these domains requirements continue to emerge for new physics, improved usability, or even new technologies for Geant4-based tools
-
- Next Geant4 Space Users' Workshop:
Boeing Co., Seattle
August 18th (Wed) - 20th (Fri), 2010

