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Planar Gem Mechanical Analysis

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Lina Quintic

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analysis

FEA Model (element type, material, geometry)

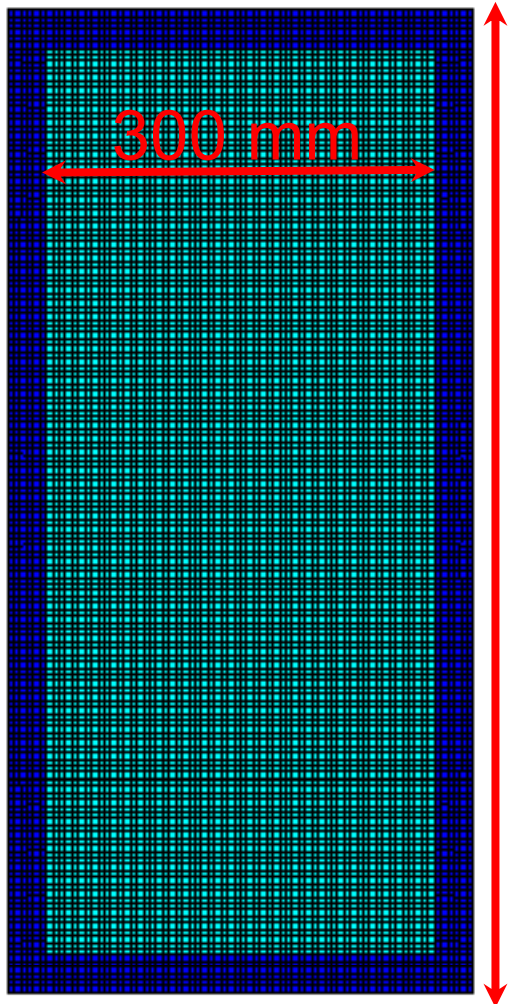
Stress and Strain Calculations (several configurations of boundary conditions and loading)

Results for static analysis

(Instability analysis in progress)



ANSYS



760 mm



FEA model

Thin Shell: 10944 elements

Non linear problems due to:

1. deformations \gg thickness

- material behaviour

- Shell63

- Shell181(keyopt0=0)

- Solid45+Shell181(keyopt1=1)



aterial data

Permoglass

densità=1850 Kg/m³

Young Modulus=2.4E+10

Poisson Coeff=0.17 [Ref1]

► Gem (kapton+copper

► densità= 2221.63 kg/m³

► Young Modulus=3.26E+9 (well-accordance
experimental-calculated stress-strain)

► Poisson Coeff=0.335**

**Ref:articolo



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Proceeding xxi ICTAM, 15-24 August 2004
Warsaw Poland

<<*Three-dimensional Thermoelastic
Analysis of plain/epoxy composite*>>

Poisson's ratio=0.17



Table 2 Elastic properties of two-layer G-11 woven glass/epoxy laminates

		Young's moduli		
		293 K	77 K	4 K
E_1^l (GPa)	3-D FEA	26.75	33.10	36.04
	Micromechanics model	26.66	32.96	35.94
	Experimental	27.9	32.7	36.9
E_2 (GPa)	3-D FEA Case 1	25.88	31.85	34.69
	Case 2	25.51	31.23	33.89
		Poisson's ratios		
		293 K	77 K	4 K
ν_{12}^l	3-D FEA	0.140	0.193	0.214
	Micromechanics model	0.13	0.19	0.21
	Experimental	0.17	0.19	0.23
ν_{13}^l	3-D FEA Case 1	0.137	0.188	0.209
	Case 2	0.132	0.182	0.202

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Boundary conditions

Shell Only

Externally Simple supported
(UZ not allowed)

Frame +Shell

Rigidly clamped
ALL degree
of freedom suppressed

Loads

Shell Only

Only gravity

gravity+traction
on external side (sf=load
per unit length)

Frame +Shell

uniform compression
on the lower surface of
the frame nodes
(f=load)

Model

Shell Only

Shell63

Shell181

Frame +Shell

Shell63+Shell63

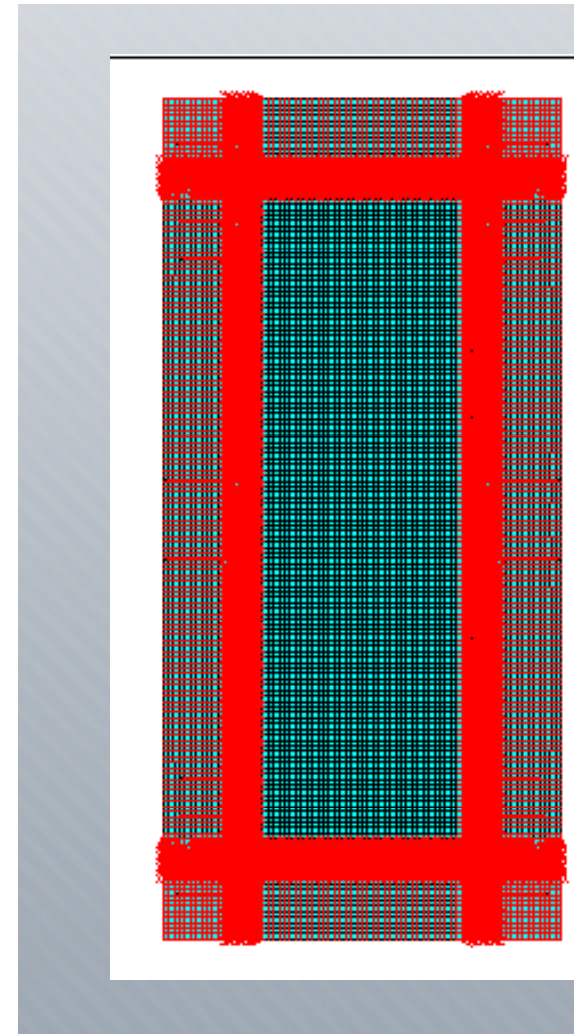
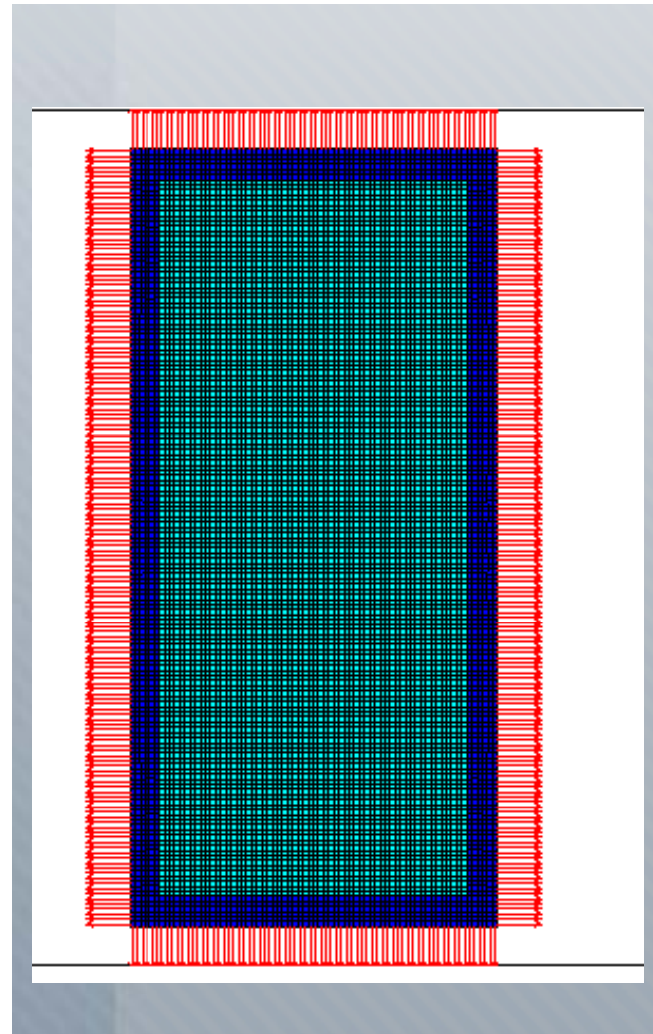
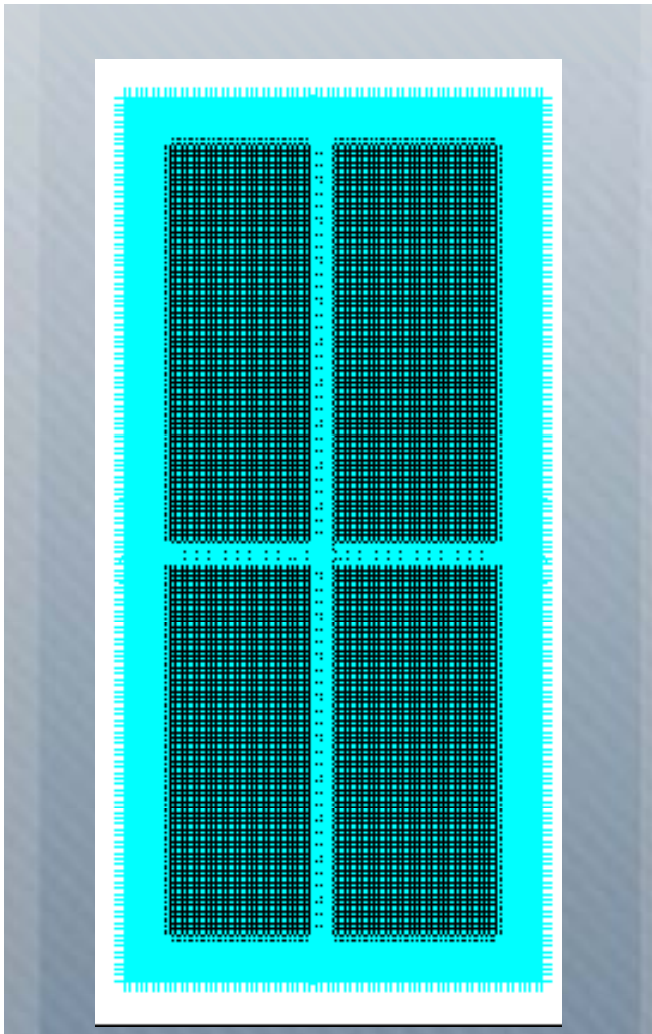
Shell181+Shell181

Shell181+solid4

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Calculations Descriptions

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only shell
with gravity

only shell
with tensile load + gravity

shell+frame with
compression load

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Boundary condition and load configuration



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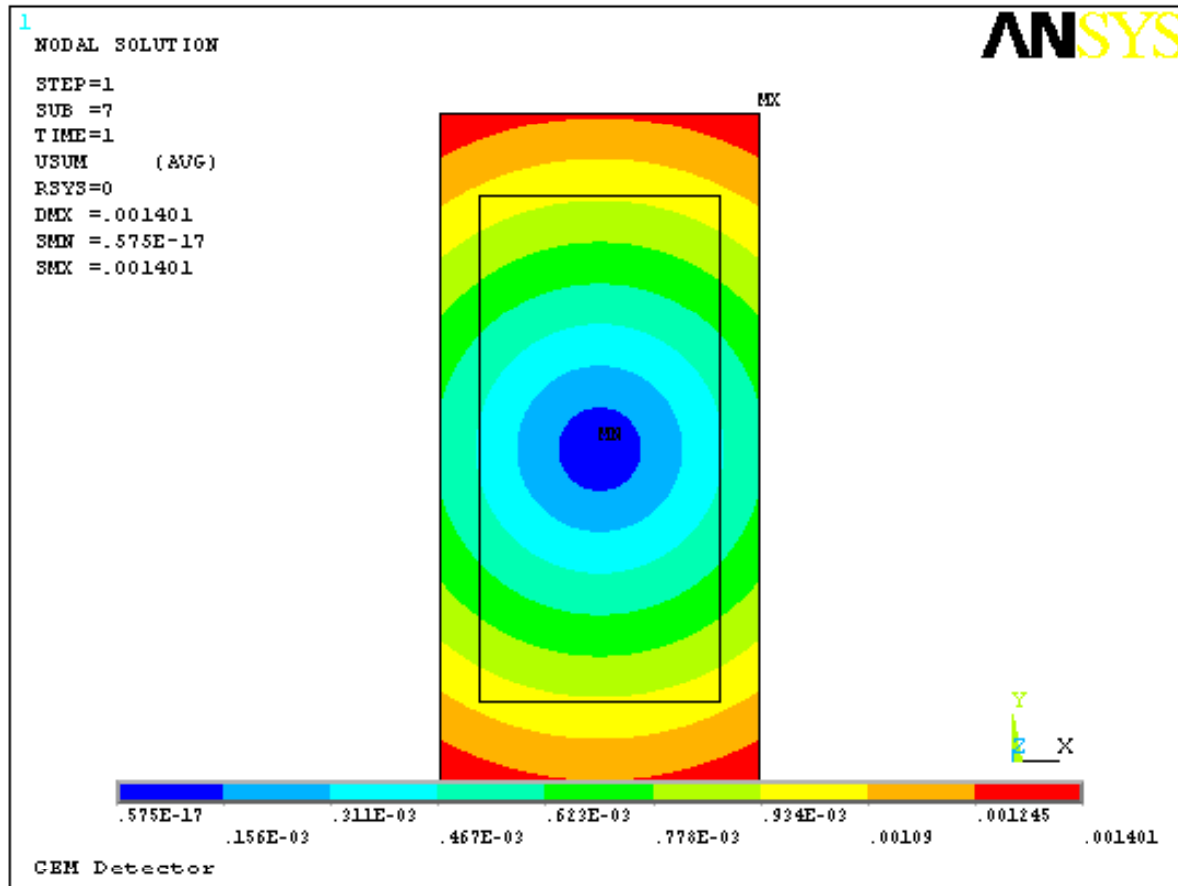
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Case 0

Simply supported shell with tensile load

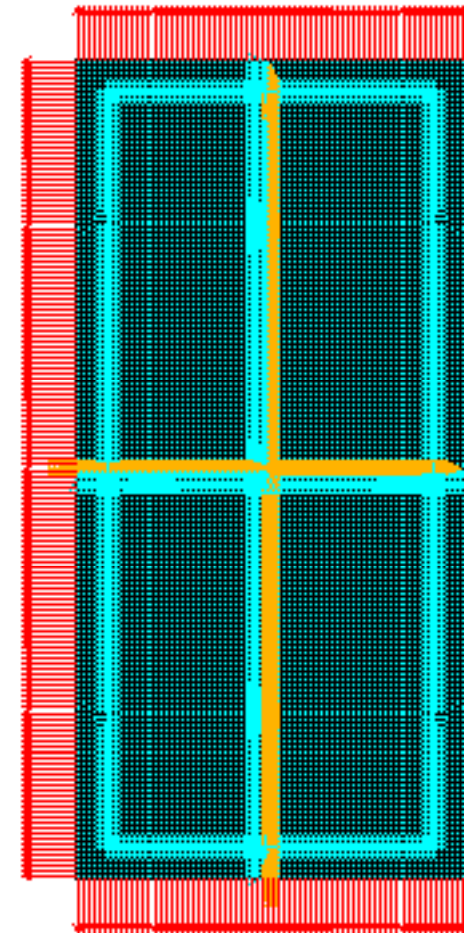
$U_{max} = 0.001407 \text{ m}$
hydrostatic stress with value
 $\sigma = 0.163E+8 \text{ Pa}$



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ELEMENTS

U
ROT





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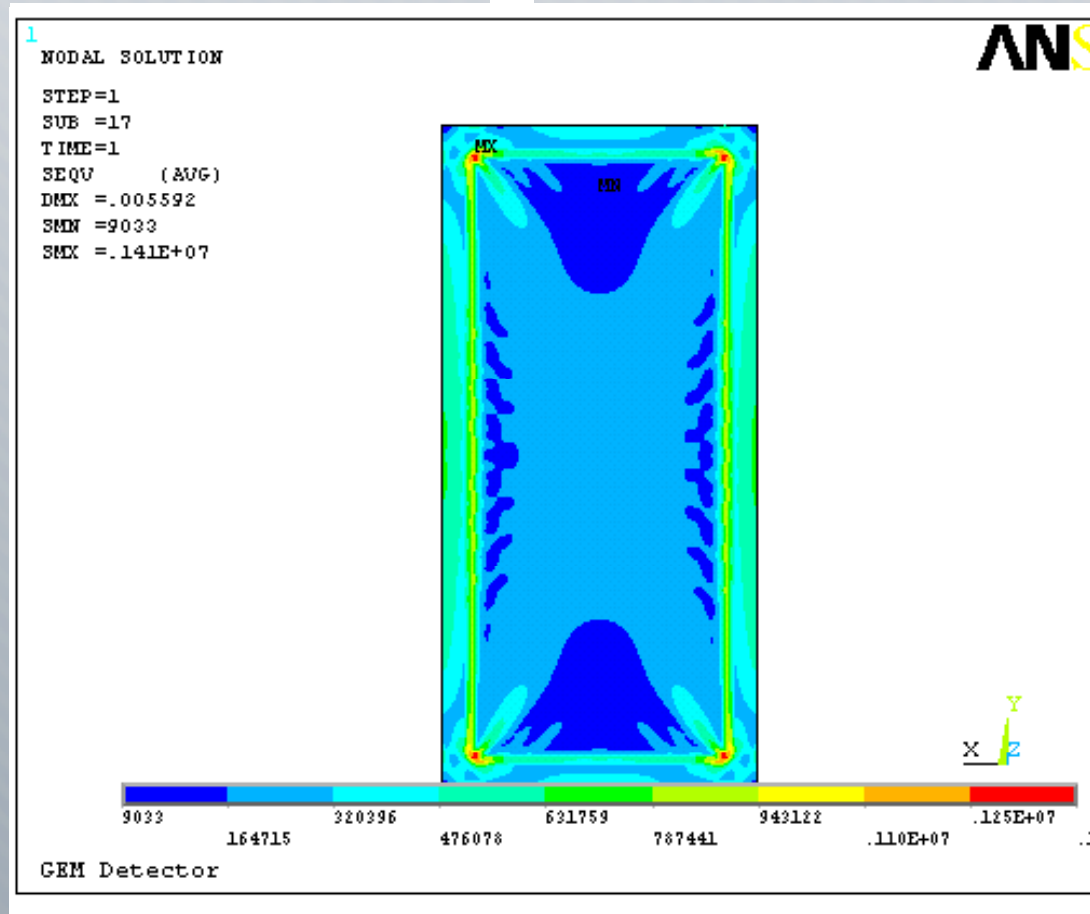
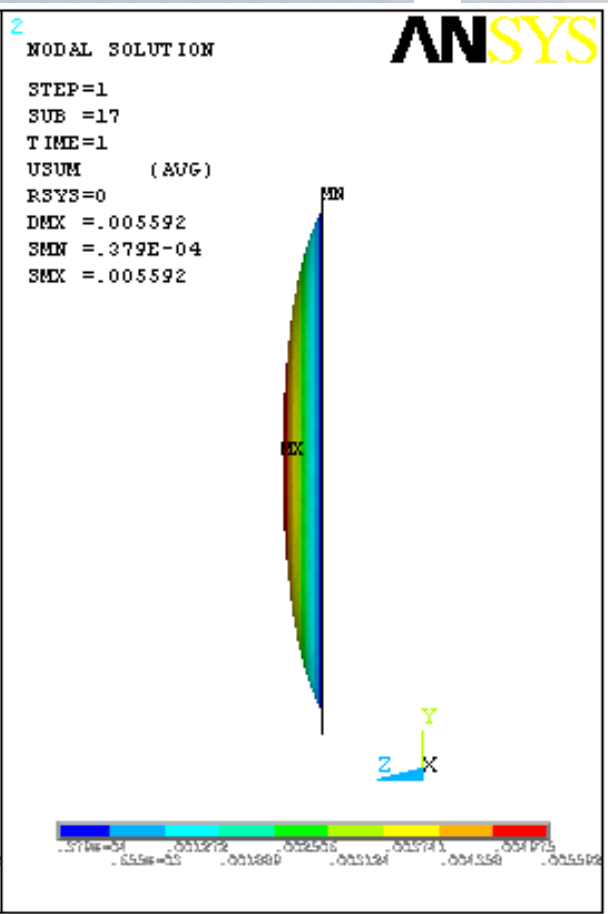
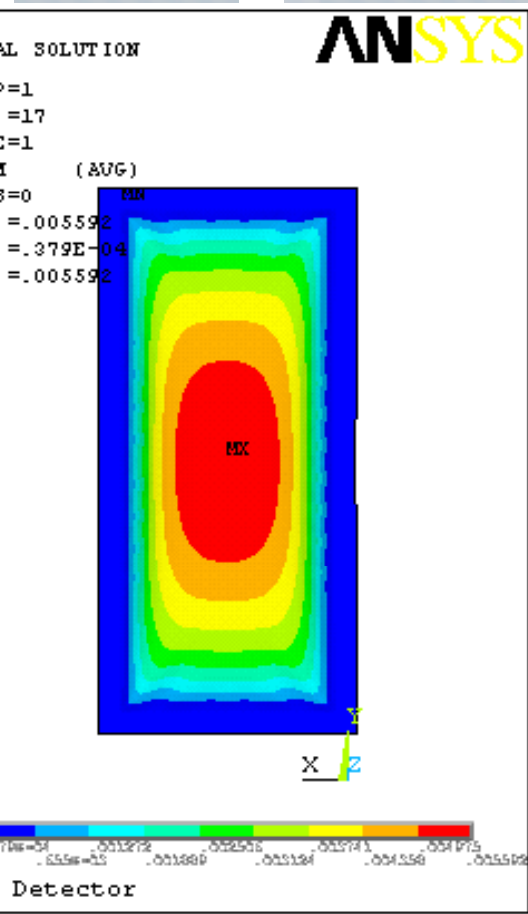
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CASE 1

SIMPLY SUPPORTED SHELL (WITHOUT FRAMING)
WITH GRAVITY



+ Displacement scale=10

BC:simply supported Sh
LOAD:only grav

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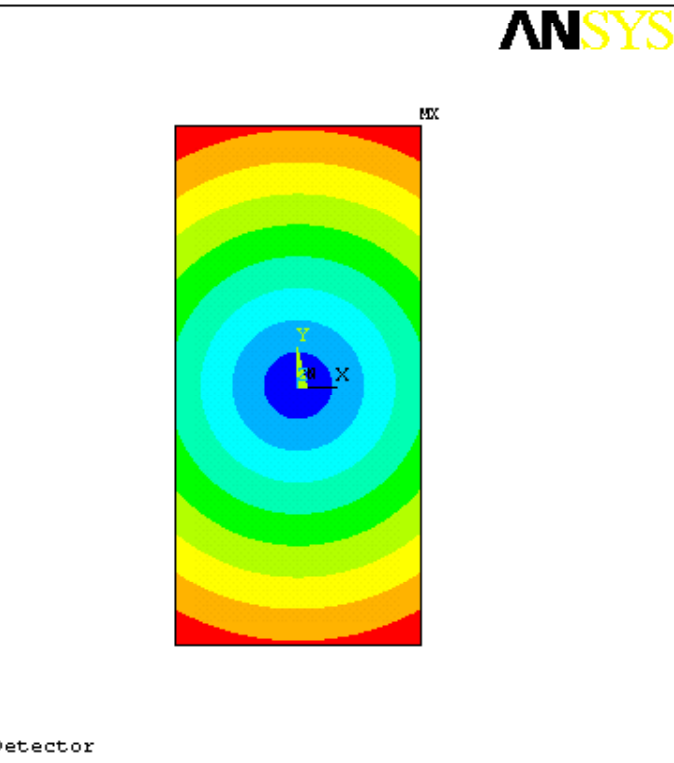
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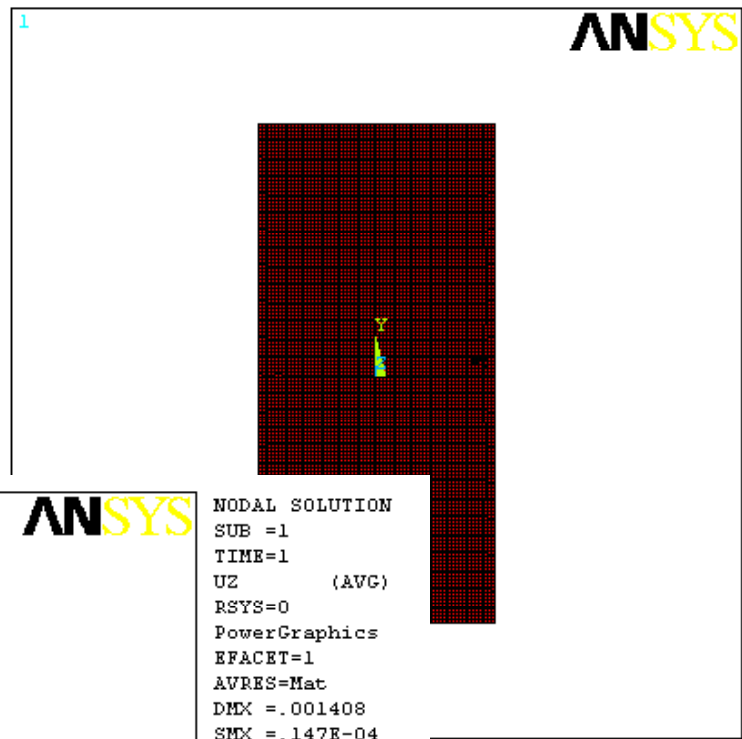
CASE 2

SIMPLY SUPPORTED SHELL (WITHOUT FRAME)
WITH GRAVITY AND TENSILE LOAD



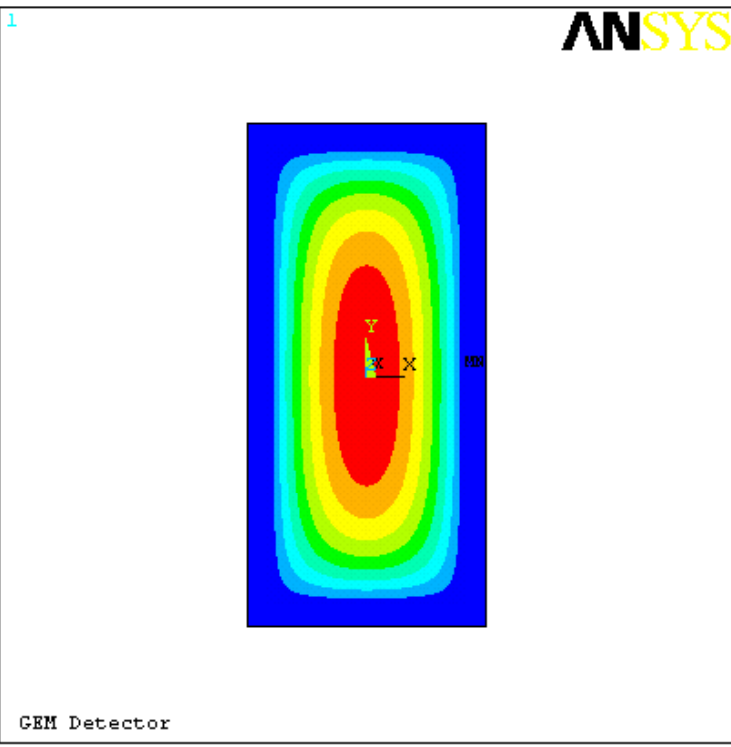
ANSYS
NODAL SOLUTION
SUB =1
TIME=1
USUM (AVG)
RSYS=0
PowerGraphics
EFACET=1
AVRES=Mat
DMX =.001408
SMN =.147E-04
SMX =.001408

ZV =1
*DIST=.744643
ZF =.735E-03
Z-BUFFER
.147E-04
.169E-03
.324E-03
.479E-03
.634E-03
.789E-03
.943E-03
.001098
.001253
.001408



ANSYS
ELEM
SUB =
TIME=
SEQV
Power
EFACE
DMX =
SMN =
SMX =

ZV =
*DIST=
ZF =
Z-BUF
[Red color swatch]



ANSYS
NODAL SOLUTION
SUB =1
TIME=1
UZ (AVG)
RSYS=0
PowerGraphics
EFACET=1
AVRES=Mat
DMX =.001408
SMX =.147E-04

ZV =1
*DIST=.744643
ZF =.735E-03
Z-BUFFER
0
.163E-05
.327E-05
.490E-05
.653E-05
.816E-05
.980E-05
.114E-04
.131E-04
.147E-04

BC:simply supported shell
LOAD:gravity+ traction load (1kgf/cm)



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CASE

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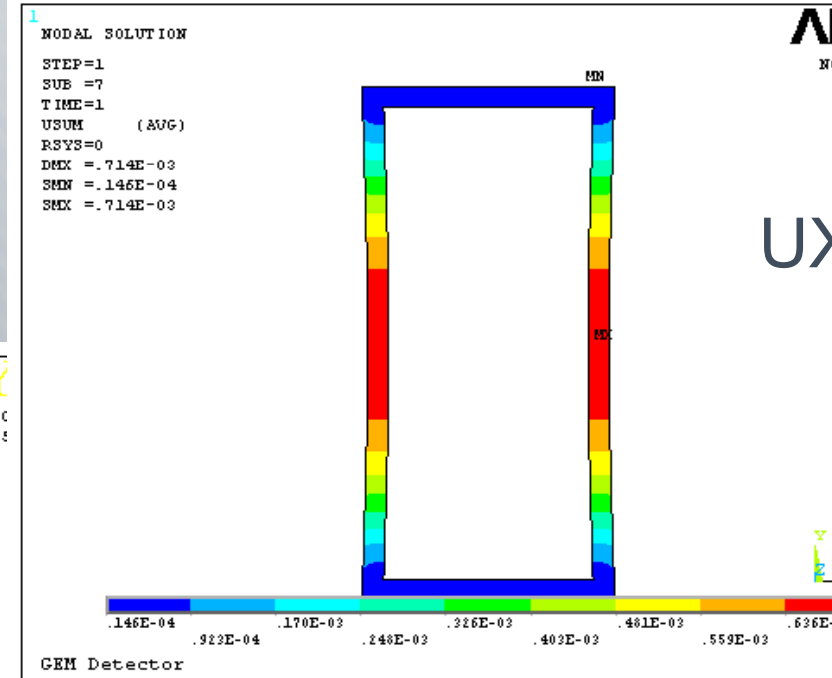
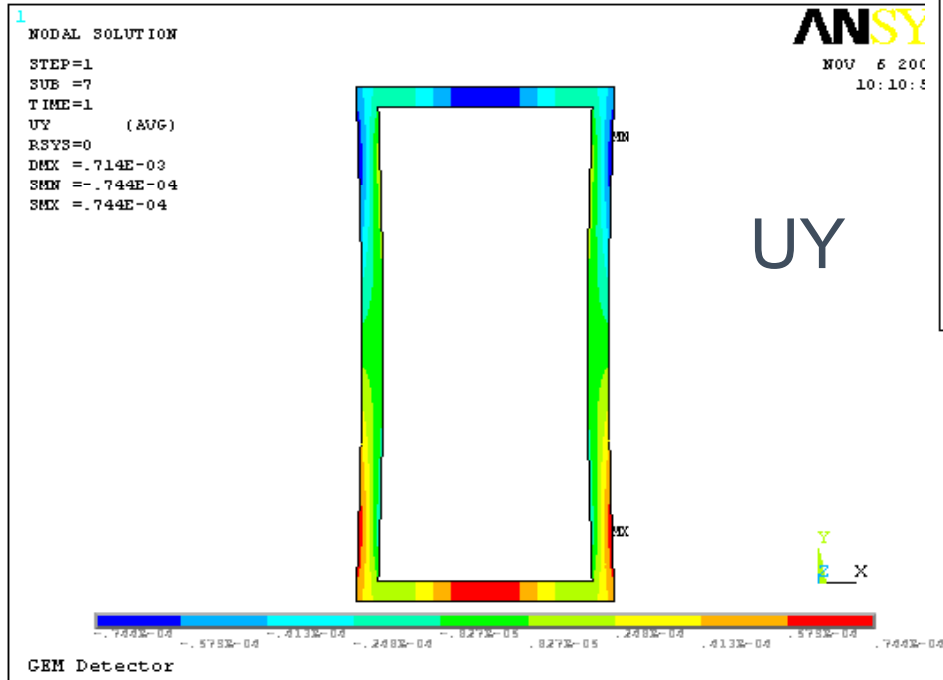
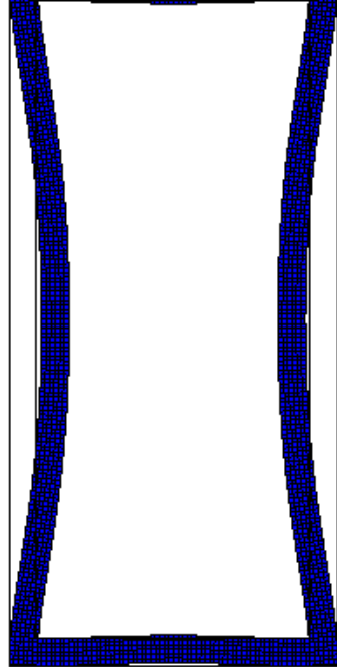
SIMPLY SUPPORTED SHELL WITH FRAME
FRAME WITH COMPRESSION LOA

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$\sigma = 0.163E+8$ Pa from CASE0 $F_x = \sigma \cdot (760 \cdot 60E-6)$ $F_y = \sigma \cdot (360 \cdot 60E-6)$

$\sigma_y(\max) = 0.152E+8$ Pa

```
STEP=1  
SUB =7  
TIME=1  
PowerGraphics  
EFACET=1  
AVERFS=Mar  
DMX =.714E-03  
  
*DSCA=50  
ZV =1  
DIST=.41738  
Z-BUFFER
```



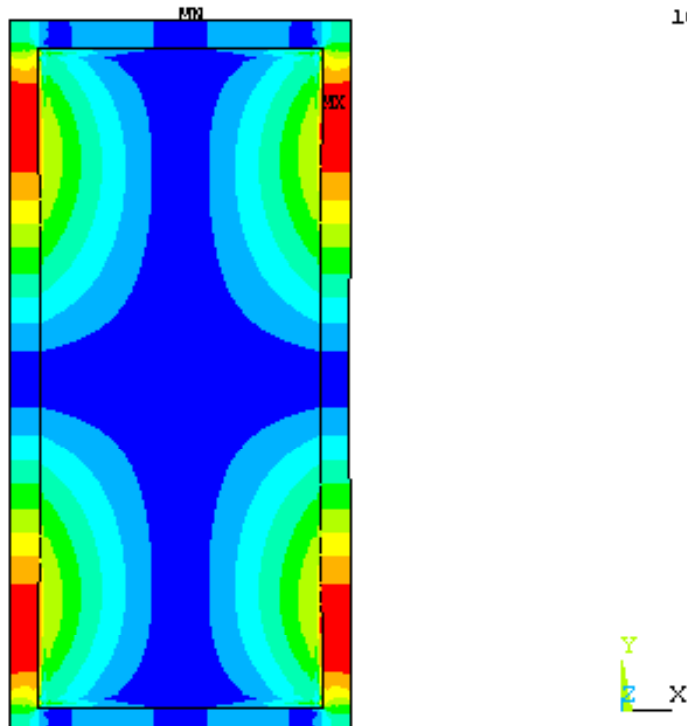
Model: Simply supported shell+frame
LOAD: only uniform compression on frame



SOLUTION

ANSYS

NOV 5 2008
10:06:48

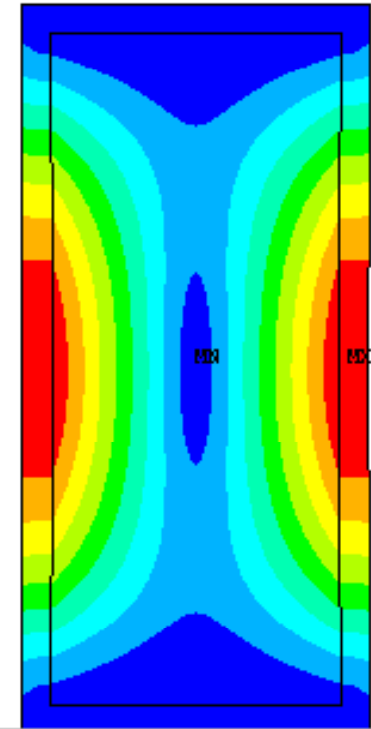


(AVG)
14E-03
37E-12
03168

1.37E-12 .352E-03 .704E-03 .001056 .001408 .00176 .002112 .002464 .002816 .003168

Detector

1
NODAL SOLUTION
STEP=1
SUB =7
TIME=1
USUM (AVG)
RSYS=0
DMX =.714E-03
SMN =.305E-12
SMX =.714E-03



.305E-12 .794E-04 .159E-03 .238E-03 .317E-03 .397E-03 .476E-03 .555E-03

GEM Detector

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Shell+frame=some comment
Conservative calculations (shell prestress not yet taken into account)
transient analysis with initial stress in the shell could be done

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Frame with different thickness: 2-4-6-8 mm

2 mm--> $u_{max}=0.714E-3$ m

4 mm--> $u_{max}=0.696E-3$ m

6 mm--> $u_{max}=0.668E-3$ m

8 mm--> $u_{max}=0.639E-3$ m

