

Introduction to cryogenics

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Contents

- Introduction
- Cryogenic fluids
- Heat transfer & thermal insulation
- Thermal screening with cold vapour
- Refrigeration & liquefaction
- Cryogen storage & transport
- Thermometry

- **κρυος, ους (το)** **1** deep cold [Arist. *Meteor.*]
 2 shiver of fear [Aeschyl. *Eumenid.*]
- **cryogenics**, that branch of physics which deals with the production of very low temperatures and their effects on matter

Oxford English Dictionary

2nd edition, Oxford University Press (1989)

- **cryogenics**, the science and technology of temperatures below 120 K

New International Dictionary of Refrigeration

4th edition, IIF-IIR Paris (2015)

Characteristic temperatures of cryogenes

Cryogen	Triple point [K]	Normal boiling point [K]	Critical point [K]
Methane	90.7	111.6	190.5
Oxygen	54.4	90.2	154.6
Argon	83.8	87.3	150.9
Nitrogen	63.1	77.3	126.2
Neon	24.6	27.1	44.4
Hydrogen	13.8	20.4	33.2
Helium	2.2 (*)	4.2	5.2

(*): λ Point

Cryogenic transport of natural gas: LNG



130 000 m³ LNG carrier
with double hull



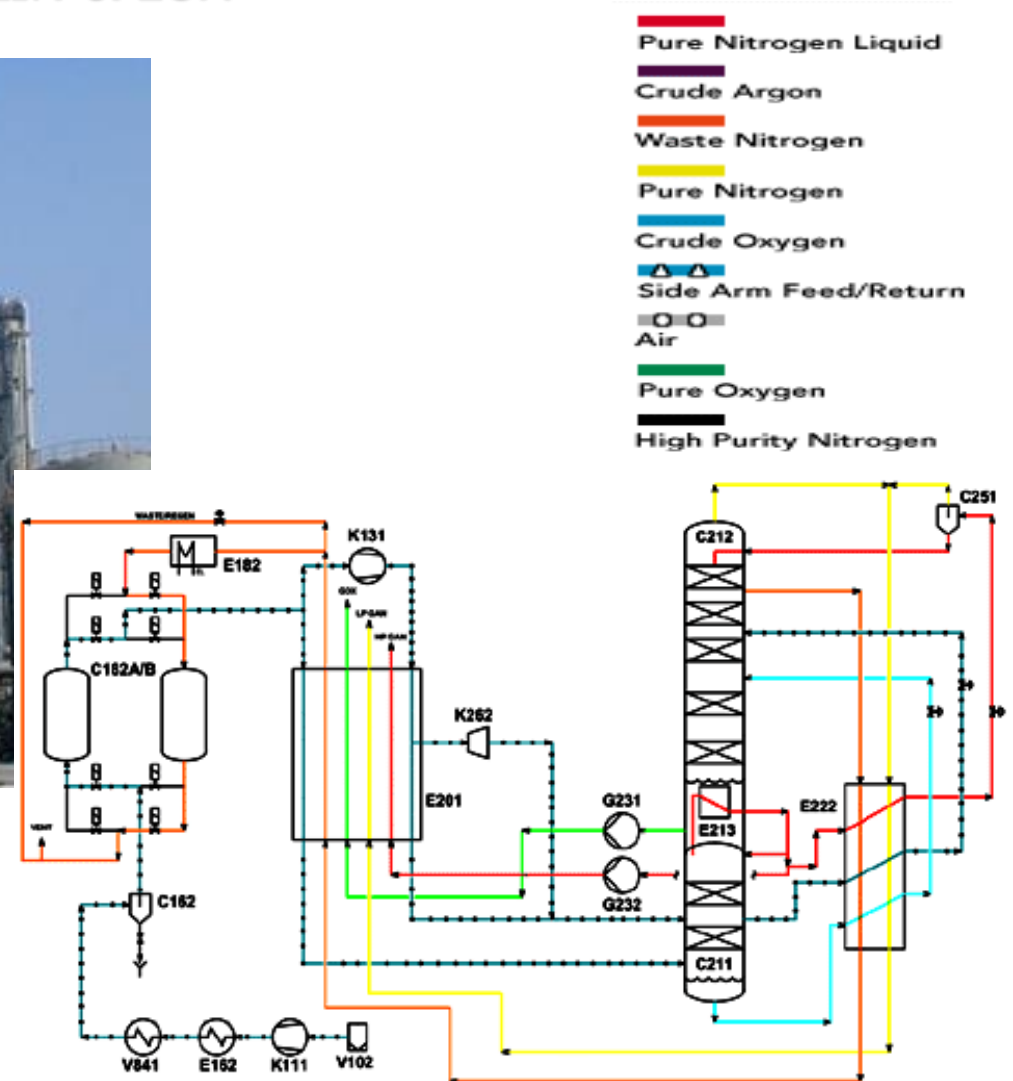
Invar[®] tanks hold LNG at ~ 110 K

Densification, liquefaction & separation of gases LIN & LOX



Air separation by cryogenic distillation
Capacity up to 4500 t/day LOX
LIN as byproduct

Ph. Lebrun



Densification, liquefaction & separation of gases

Rocket fuels

Ariane 5

25 t LHY, 130 t LOX

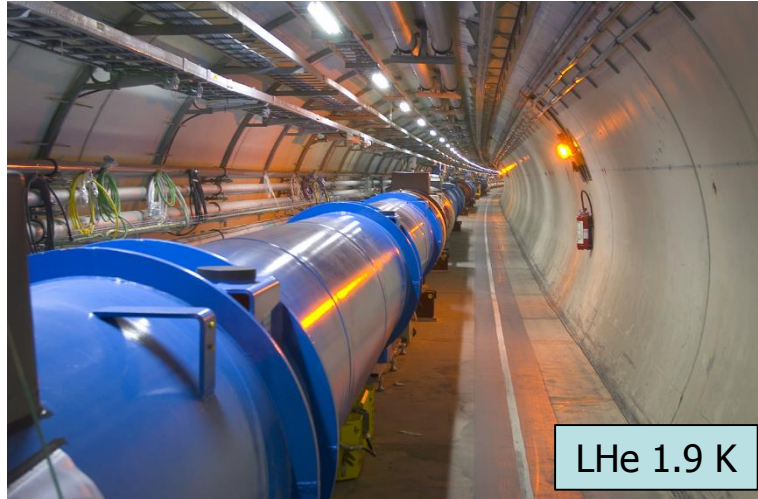


Space Shuttle

100 t LHY, 600 t LOX



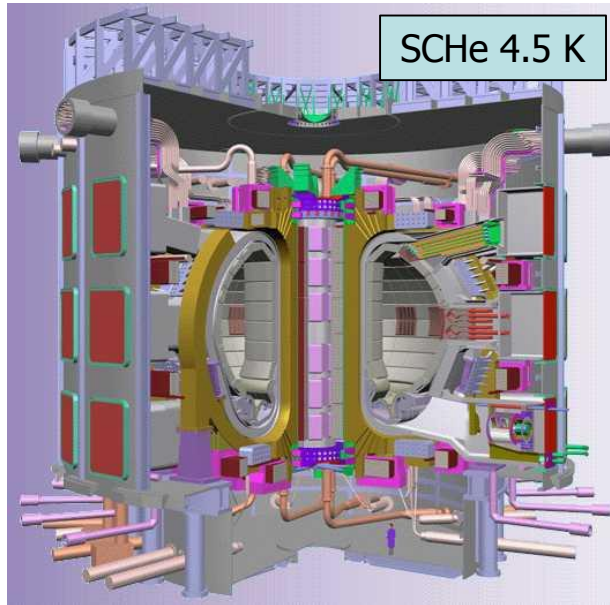
Cooling of superconducting devices



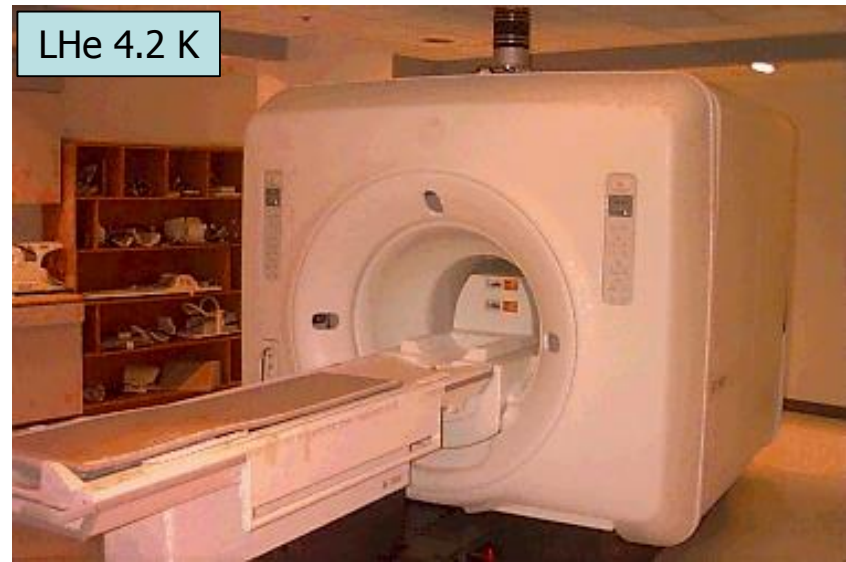
LHe 1.9 K



LIN ~70 K



SCHe 4.5 K



LHe 4.2 K

What is a low temperature?

- The entropy of a thermodynamical system in a macrostate corresponding to a multiplicity W of microstates is

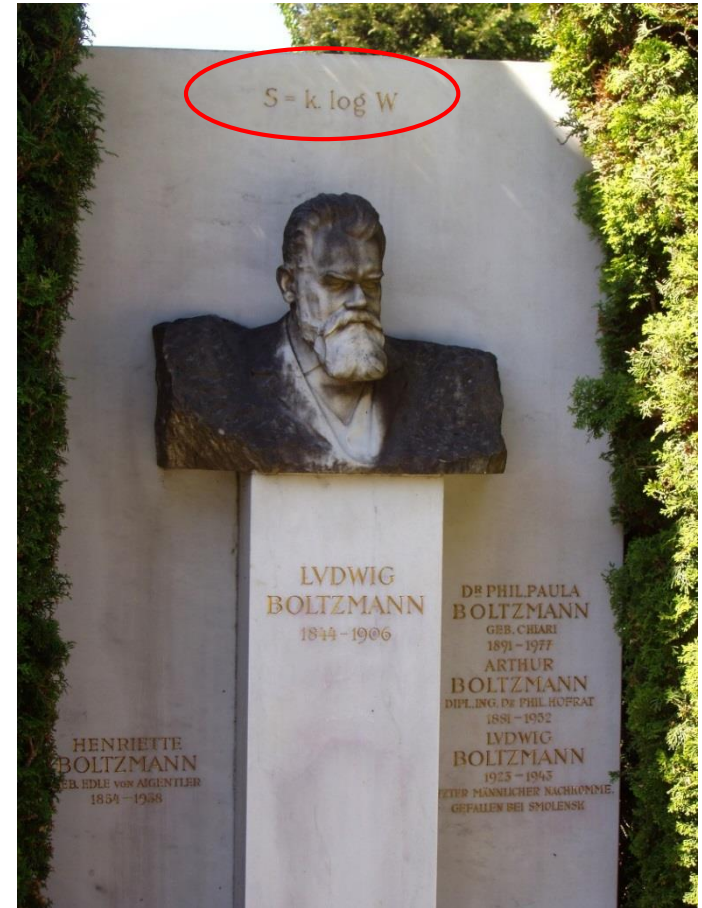
$$S = k_B \ln W$$

- Adding reversibly heat dQ to the system results in a change of its entropy dS with a proportionality factor T

$$T = dQ/dS$$

⇒ *high temperature: heating produces small entropy change*

⇒ *low temperature: heating produces large entropy change*



L. Boltzmann's grave in the Zentralfriedhof, Vienna, bearing the entropy formula

Temperature and energy

- The average thermal energy of a particle in a system in thermodynamic equilibrium at temperature T is

$$E \sim k_B T$$

$$k_B = 1.3806 \times 10^{-23} \text{ J.K}^{-1}$$

- 1 K is equivalent to $\sim 10^{-4}$ eV or $\sim 10^{-23}$ J thermal energy
 - a temperature is « low » for a given physical process when the corresponding average thermal energy $k_B T$ is small compared with the characteristic energy E of the process considered
 - cryogenic temperatures reveal phenomena with low characteristic energy and enable their study and their application

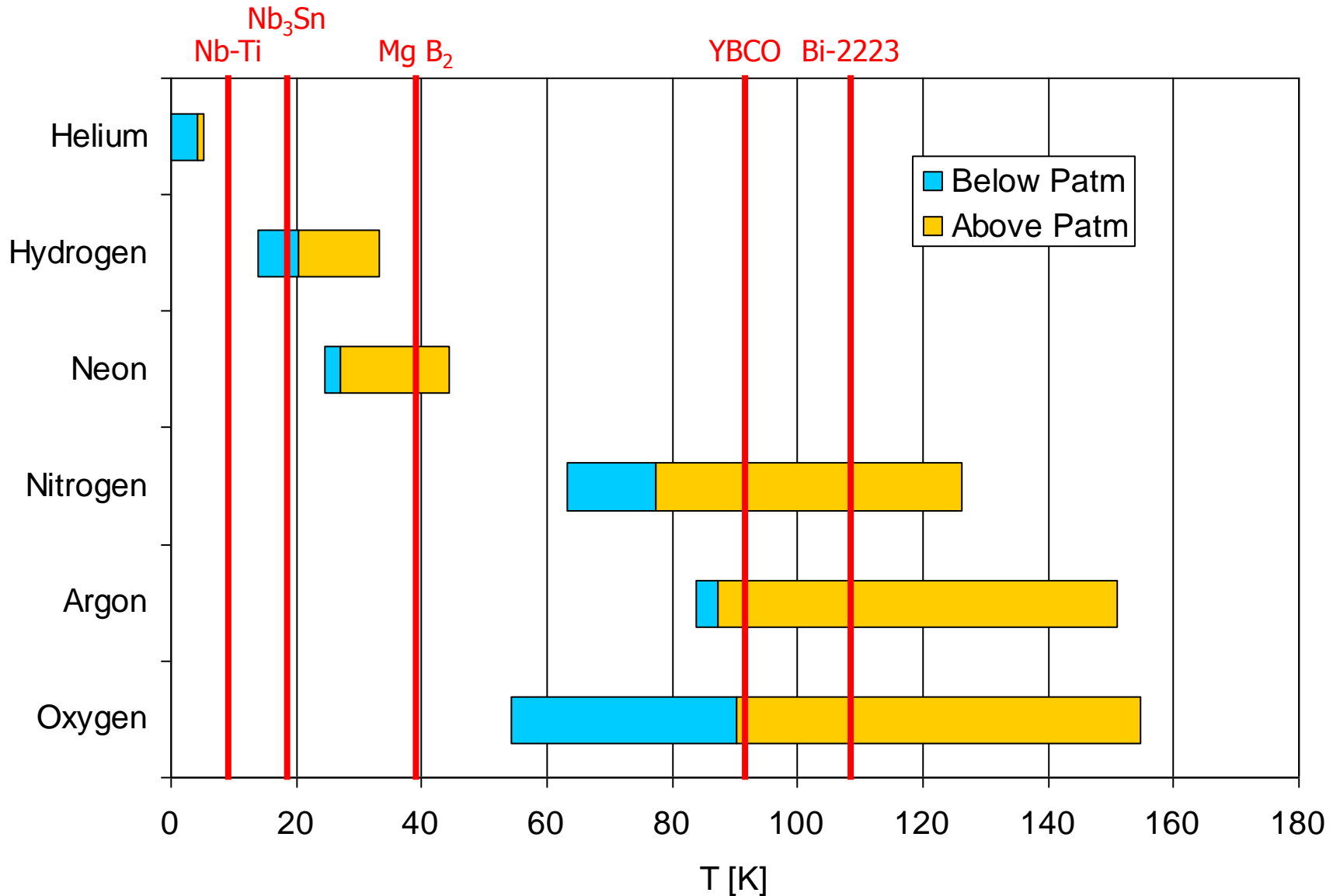
Characteristic temperatures of low-energy phenomena

Phenomenon	Temperature
Debye temperature of metals	few 100 K
High-temperature superconductivity	~ 100 K
Low-temperature superconductivity	~ 10 K
Intrinsic transport properties of metals	< 10 K
Cryopumping	few K
Cosmic microwave background	2.7 K
Superfluid helium 4	2.2 K
Bolometers for cosmic radiation	< 1 K
Low-density atomic Bose-Einstein condensates	$\sim \mu\text{K}$

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Useful range of liquid cryogenes & critical temperature of superconductors



Properties of cryogenics compared to water

Property		He	N ₂	H ₂ O
Normal boiling point	[K]	4.2	77	373
Critical temperature	[K]	5.2	126	647
Critical pressure	[bar]	2.3	34	221
Liq./Vap. density (*)		7.4	175	1600
Heat of vaporization (*)	[J.g ⁻¹]	20.4	199	2260
Liquid viscosity (*)	[μPI]	3.3	152	278

(*) at normal boiling point

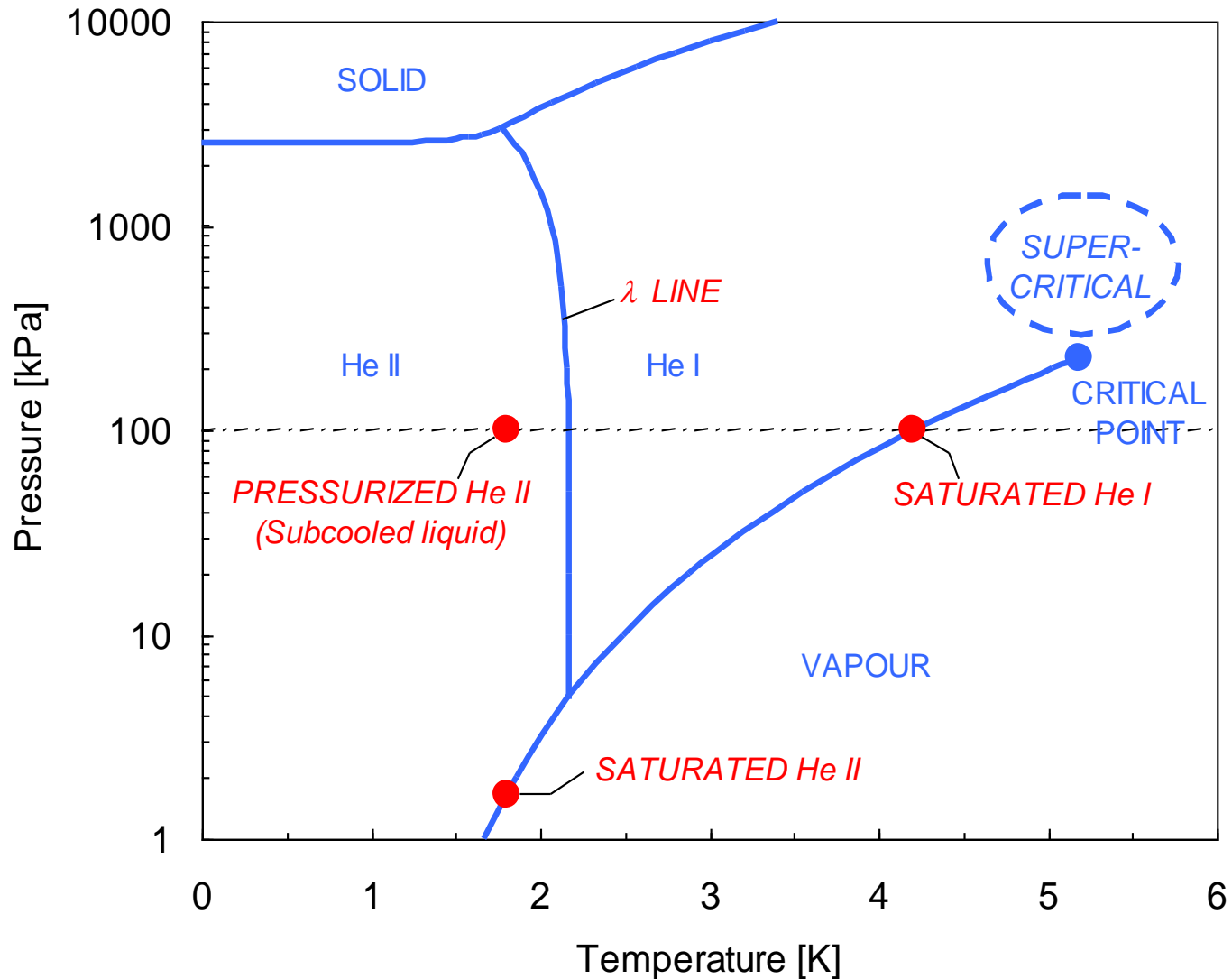
Vaporization of normal boiling cryogenes under 1 W applied heat load

Cryogen	[mg.s⁻¹]	[l.h⁻¹] (liquid)	[l.min⁻¹] (gas NTP)
Helium	48	1.38	16.4
Nitrogen	5	0.02	0.24

Amount of cryogenes required to cool down 1 kg iron

Using	Latent heat only	Latent heat and enthalpy of gas
LHe from 290 to 4.2 K	29.5 litre	0.75 liter
LHe from 77 to 4.2 K	1.46 litre	0.12 litre
LN2 from 290 to 77 K	0.45 litre	0.29 litre

Phase diagram of helium



Helium as a cooling fluid

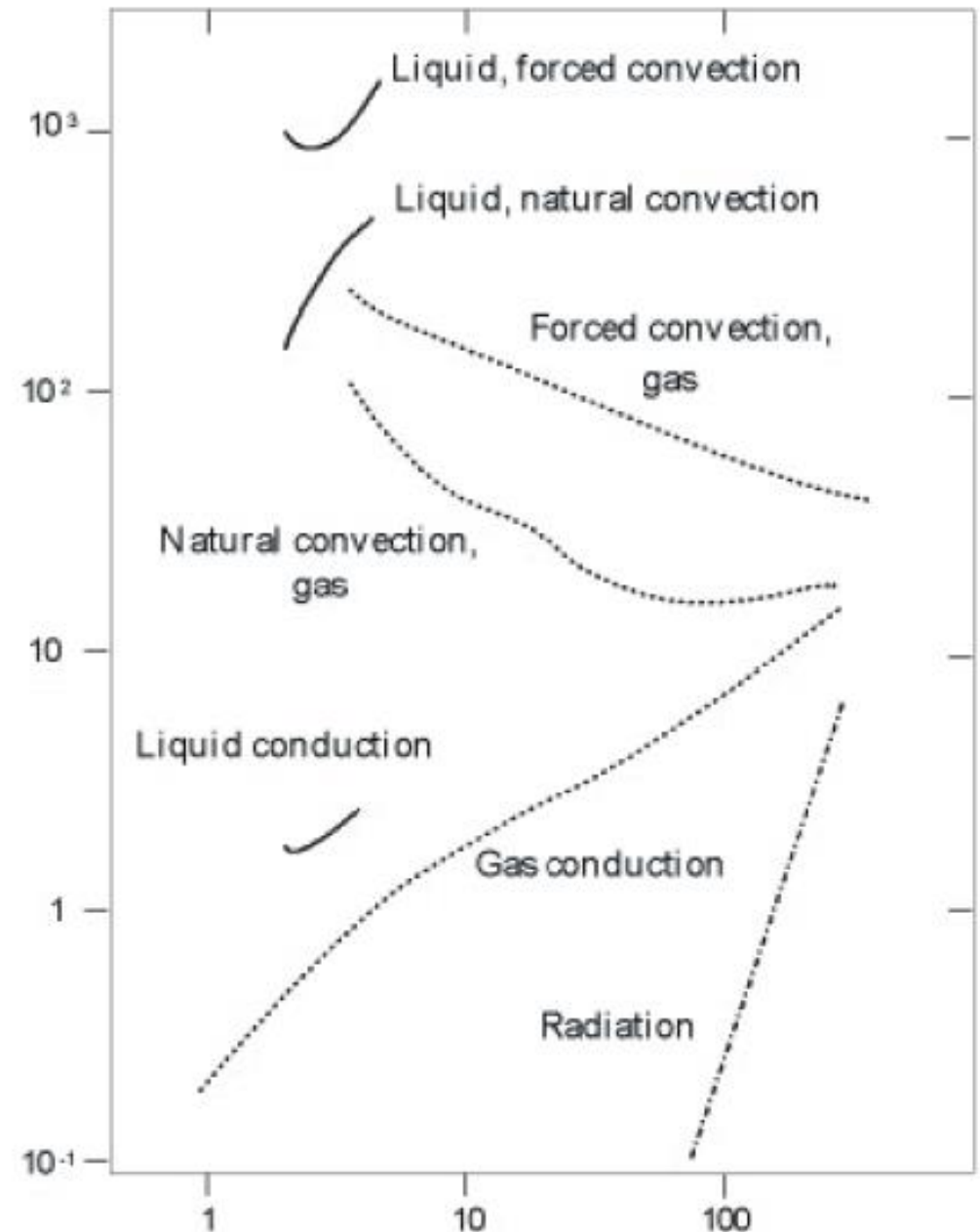
Phase domain	Advantages	Drawbacks
Saturated He I	Fixed temperature High heat transfer	Two-phase flow Boiling crisis
Supercritical	Monophase Negative J-T effect	Non-isothermal Density wave instability
He II	Low temperature High conductivity Low viscosity	Second-law cost Subatmospheric

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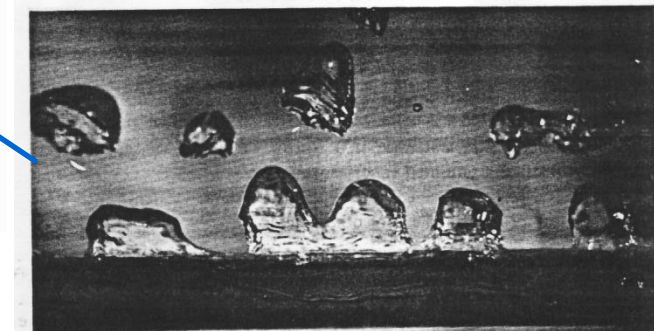
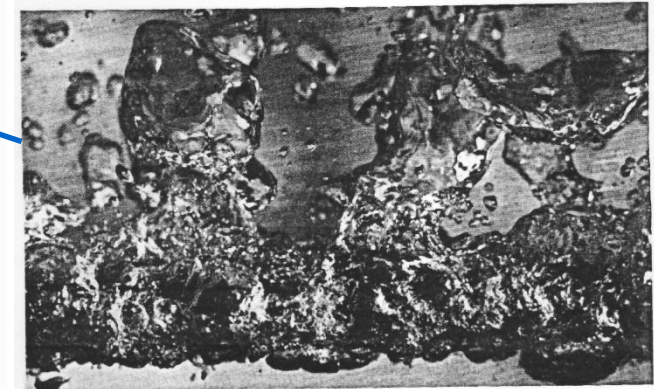
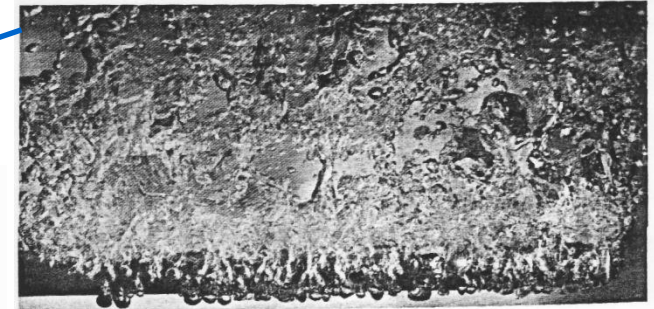
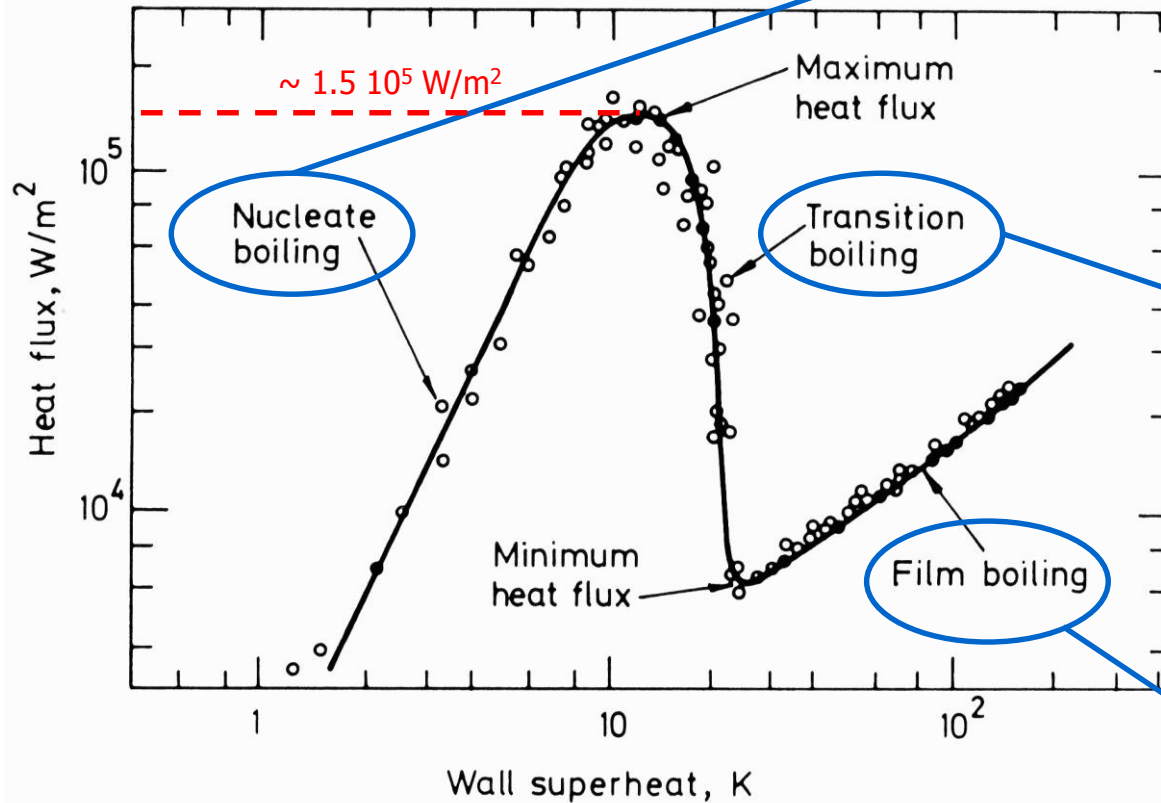
Typical heat transfer coefficients at cryogenic temperatures

- Same basic heat transfer processes as at temperatures above ambient, but large variations in orders of magnitude
- Not on this diagram:
 - Importance of two-phase heat transfer (boiling)
 - At very low temperature, solid-to-liquid interface thermal resistance (Kapitza)



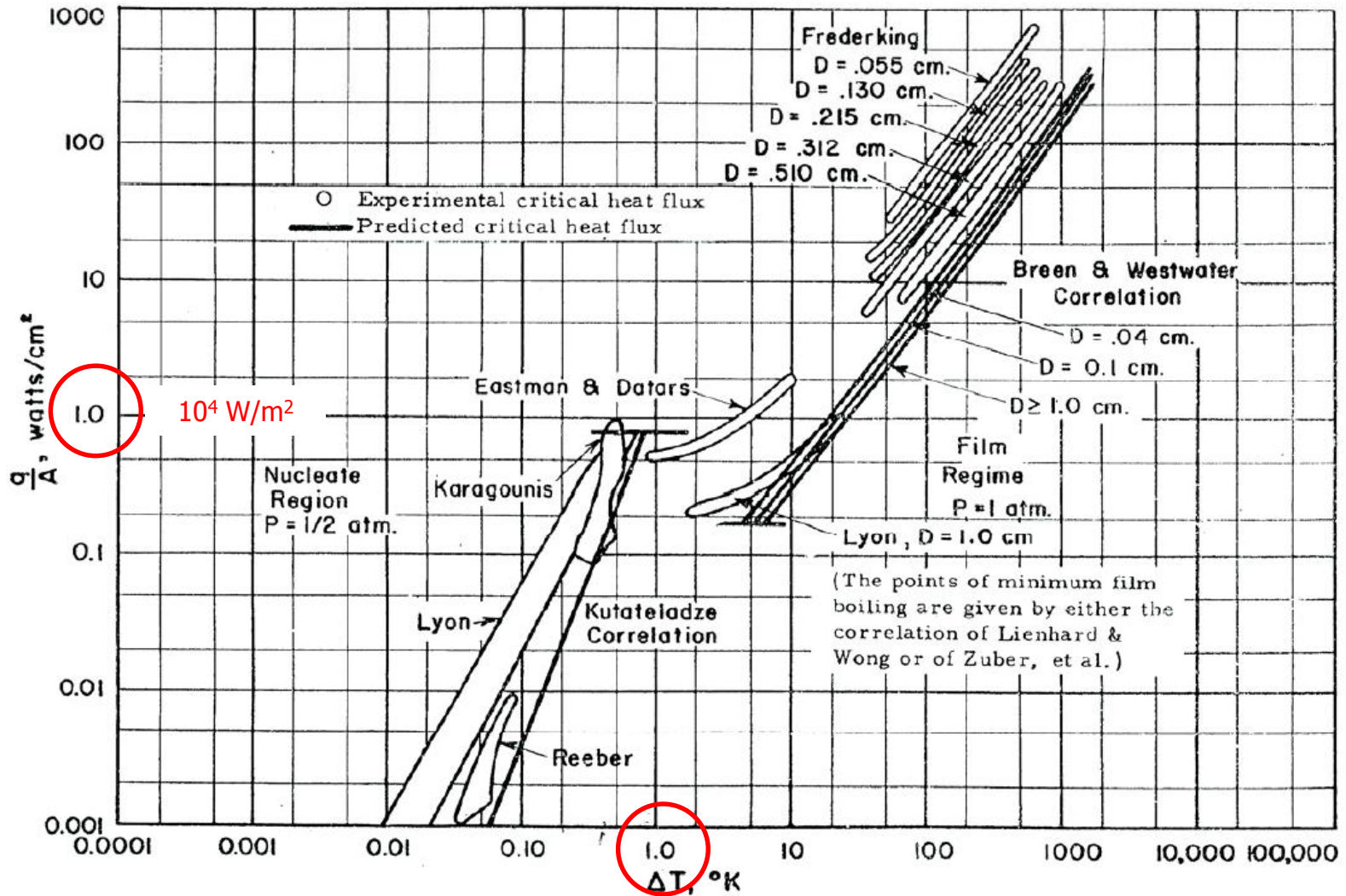
Non-linear heat transfer to liquid cryogenics

Pool boiling nitrogen

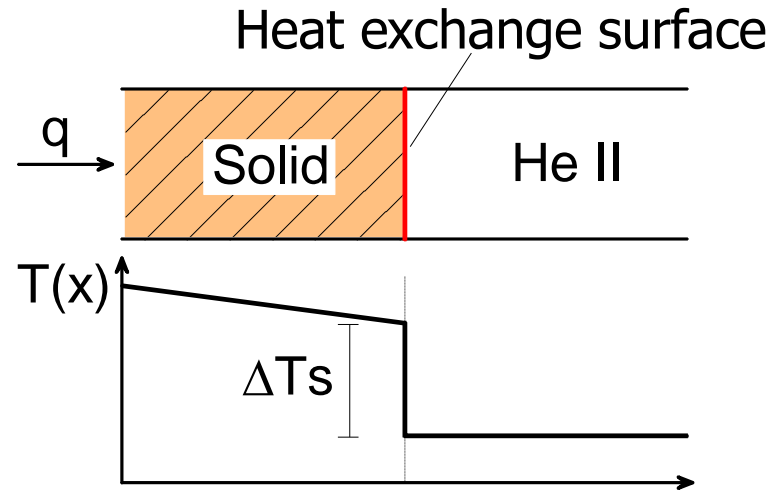
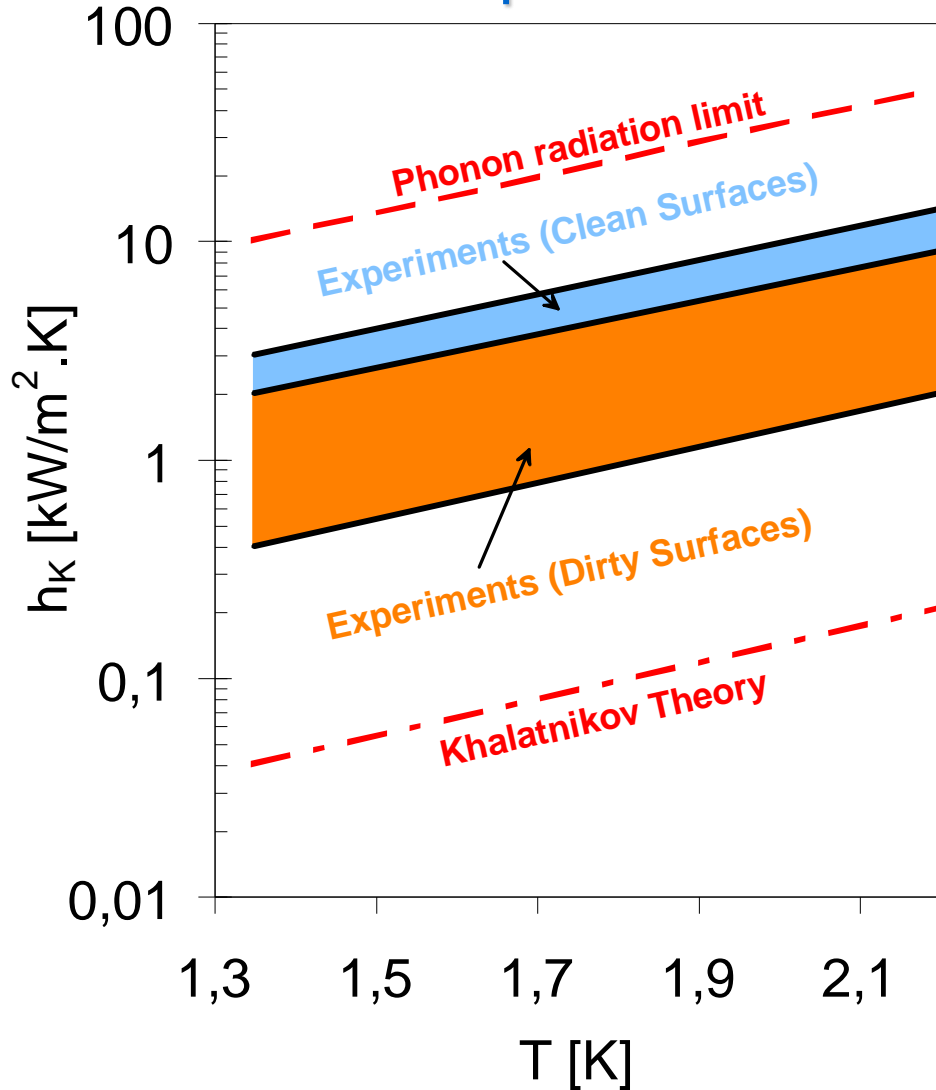


Non-linear heat transfer to liquid cryogenics

Pool boiling helium



Solid-liquid interface: Kapitza conductance

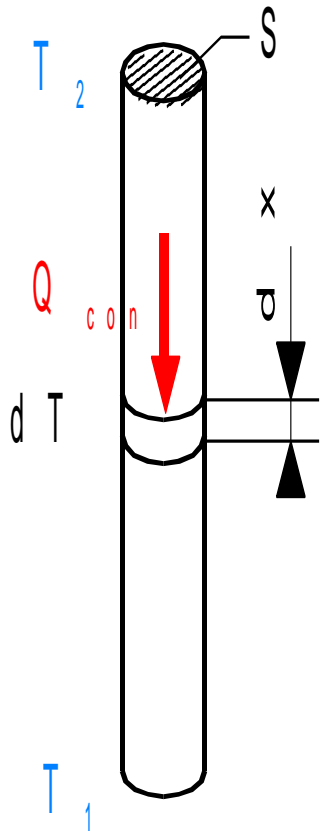


Experimental data for Copper
(S. Van Sciver, "Helium Cryogenics")

$$h_K \sim T^3$$

Valid for small heat flux (when $\Delta T \ll T$)

Heat conduction in solids



Fourier's law:

$$Q_{\text{con}} = k(T) \cdot S \cdot \frac{dT}{dx}$$

$k(T)$: thermal conductivity [W/m.K]

Integral form:

$$Q_{\text{con}} = \frac{S}{L} \cdot \int_{T_1}^{T_2} k(T) \cdot dT$$

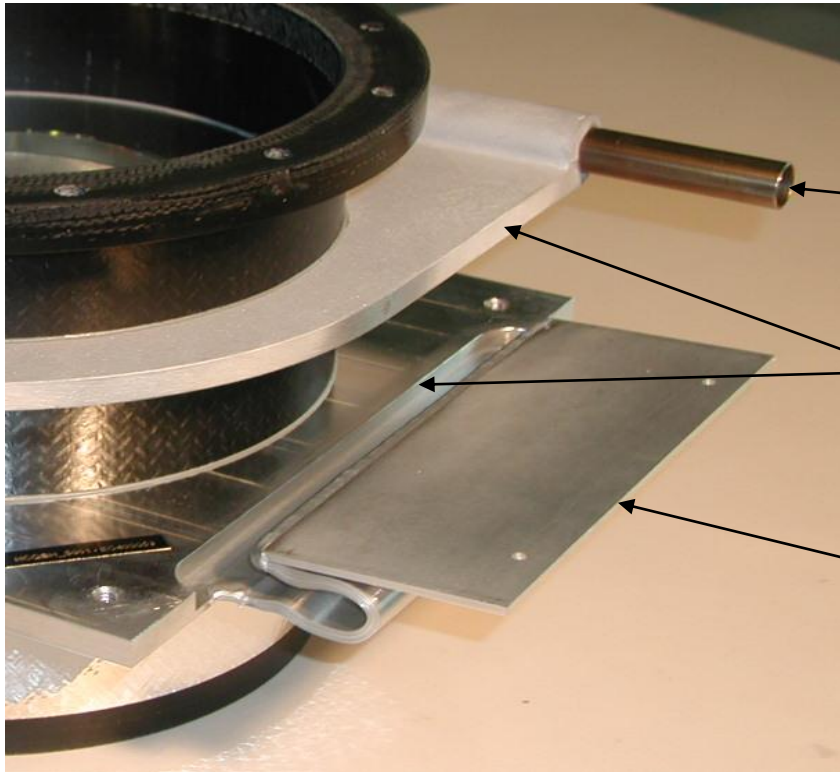
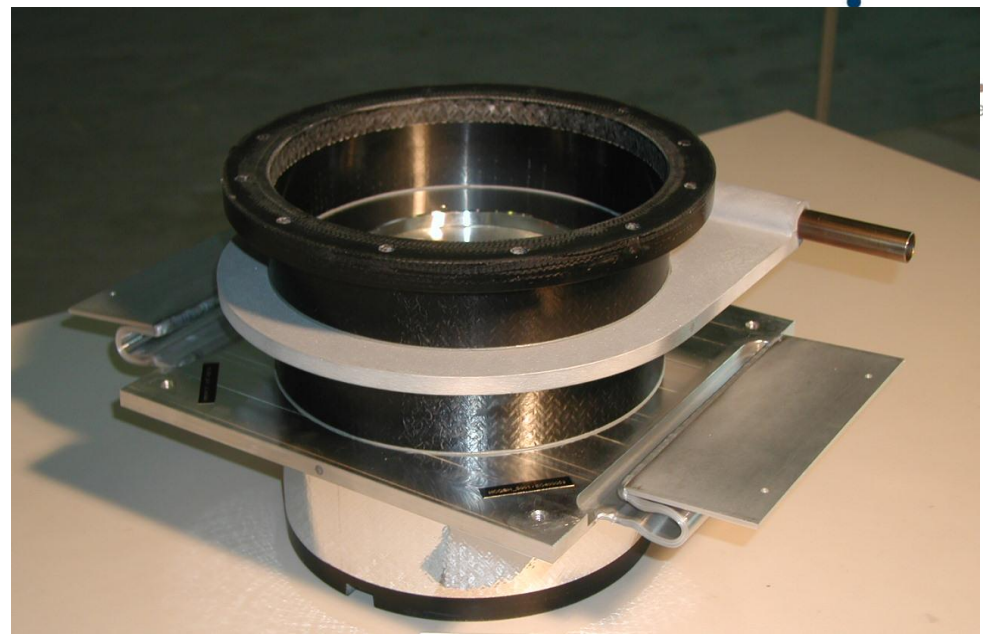
$\int k(T) \cdot dT$: thermal conductivity integral [W/m]

Thermal conductivity integrals for standard construction materials are tabulated

Thermal conductivity integrals of selected materials [W/m]

From vanishingly low temperature up to	20 K	80 K	290 K
OFHC copper	11000	60600	152000
DHP copper	395	5890	46100
1100 aluminium	2740	23300	72100
2024 aluminium alloy	160	2420	22900
AISI 304 stainless steel	16.3	349	3060
G-10 glass-epoxy composite	2	18	153

Non-metallic composite support post with heat intercepts



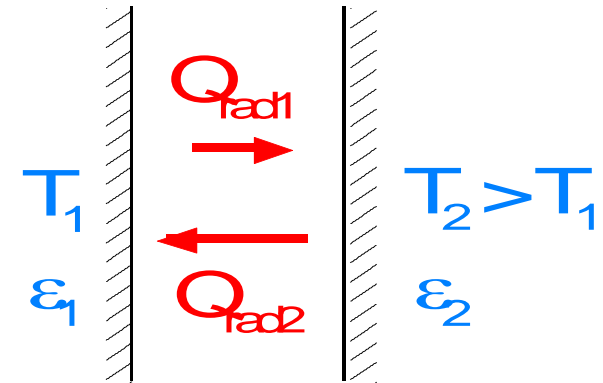
5 K cooling line (SC He)

Aluminium intercept plates
glued to G-10 column

Aluminium strips to thermal
shield at 50-75 K

Thermal radiation

- Wien's law
 - Maximum of black body power spectrum
 $\lambda_{max} T = 2898 [\mu\text{m.K}]$
- Stefan-Boltzmann's law
 - Black body
 - "Gray"body
 - "Gray" surfaces at T_1 and T_2



$$Q_{rad} = \sigma A T^4$$

$$\sigma = 5.67 \times 10^{-8} \text{ W/m}^2\cdot\text{K}^4$$

(Stefan Boltzmann's constant)

$$Q_{rad} = \epsilon \sigma A T^4$$

ϵ emissivity of surface

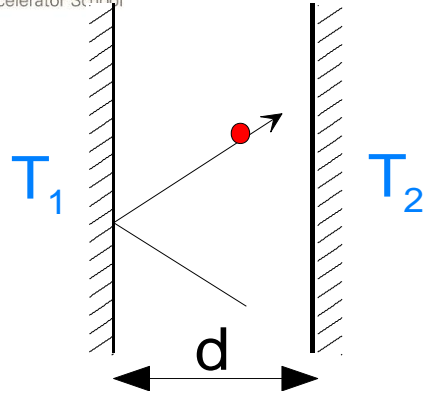
$$Q_{rad} = E \sigma A (T_1^4 - T_2^4)$$

E function of ϵ_1, ϵ_2 , geometry

Emissivity of technical materials at low temperatures

	Radiation from 290 K Surface at 77 K	Radiation from 77 K Surface at 4.2 K
Stainless steel, as found	0.34	0.12
Stainless steel, mech. polished	0.12	0.07
Stainless steel, electropolished	0.10	0.07
Stainless steel + Al foil	0.05	0.01
Aluminium, as found	0.12	0.07
Aluminium, mech. polished	0.10	0.06
Aluminium, electropolished	0.08	0.04
Copper, as found	0.12	0.06
Copper, mech. Polished	0.06	0.02

Residual gas conduction



$\lambda_{molecule}$: mean free path of gas molecules

- Viscous regime

- At high gas pressure $\lambda_{molecule} \ll d$
- Classical conduction $Q_{res} = k(T) A dT/dx$
- Thermal conductivity $k(T)$ independant of pressure

- Molecular regime

- At low gas pressure $\lambda_{molecule} \gg d$
- Kennard's law $Q_{res} = A \alpha(T) \Omega P (T_2 - T_1)$
- Conduction heat transfer proportional to pressure, independant of spacing between surfaces
 Ω depends on gas species
- Accommodation coefficient $\alpha(T)$ depends on gas species, T_1 , T_2 , and geometry of facing surfaces

Multi-layer insulation (MLI)



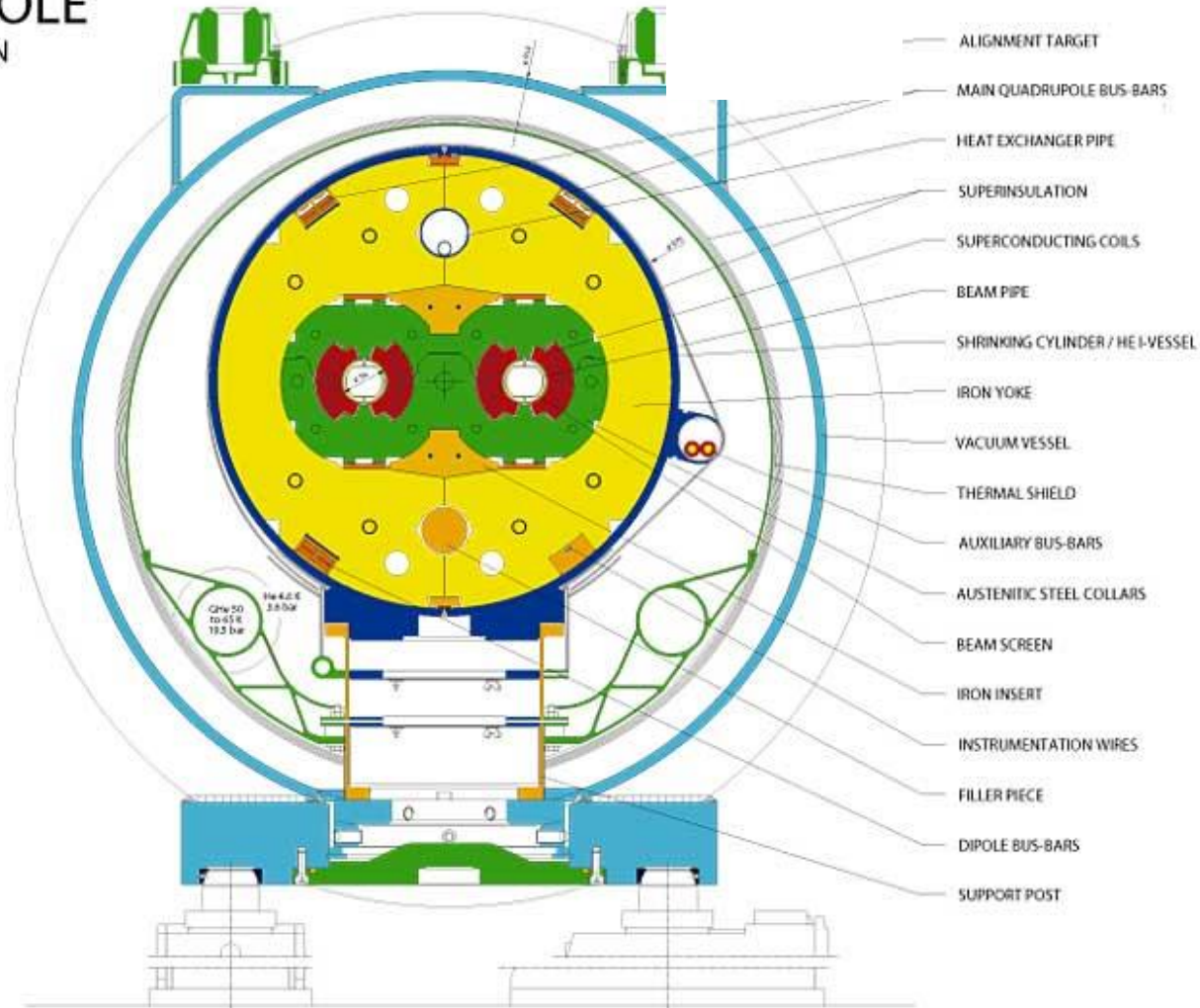
- Complex system involving three heat transfer processes
 - $Q_{MLI} = Q_{rad} + Q_{sol} + Q_{res}$
 - With n reflective layers of equal emissivity, $Q_{rad} \sim 1/(n+1)$
 - Due to parasitic contacts between layers, Q_{sol} increases with layer density
 - Q_{res} due to residual gas trapped between layers, scales as $1/n$ in molecular regime
 - Non-linear behaviour requires layer-to-layer modeling
- In practice
 - Typical data available from (abundant) literature
 - Measure performance on test samples

Typical heat fluxes at vanishingly low temperature between flat plates [W/m²]

Black-body radiation from 290 K	401
Black-body radiation from 80 K	2.3
Gas conduction (100 mPa He) from 290 K	19
Gas conduction (1 mPa He) from 290 K	0.19
Gas conduction (100 mPa He) from 80 K	6.8
Gas conduction (1 mPa He) from 80 K	0.07
MLI (30 layers) from 290 K, pressure below 1 mPa	1-1.5
MLI (10 layers) from 80 K, pressure below 1 mPa	0.05
MLI (10 layers) from 80 K, pressure 100 mPa	1-2

Cross-section of LHC dipole cryostat

LHC DIPOLE CROSS SECTION



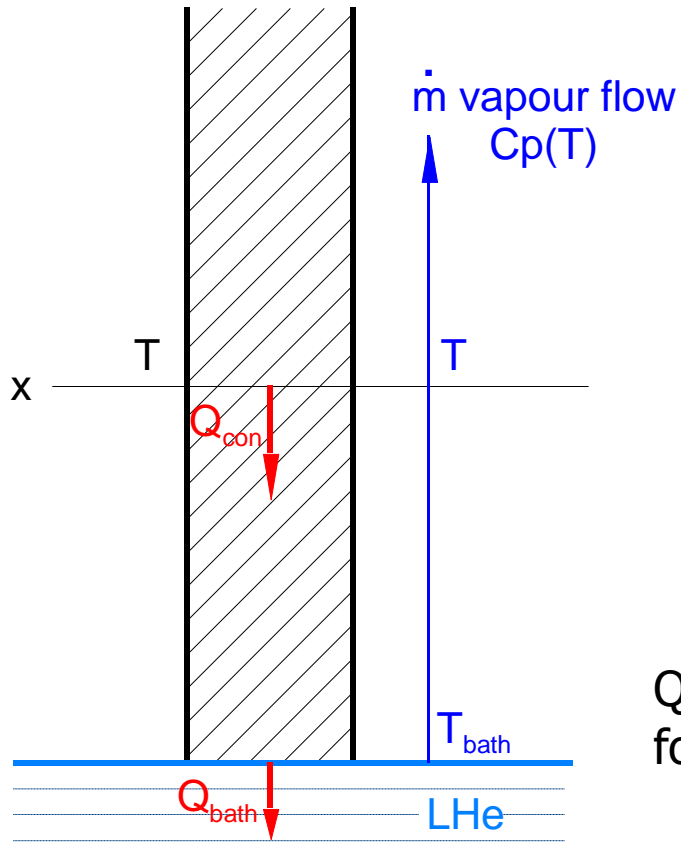
CERN AC/DI/MM — 2001/06

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Vapour cooling of necks and supports with perfect heat exchange

Cross-section A



Assuming perfect heat exchange between solid and gas, i.e. $T_{\text{sol}}(x) = T_{\text{gas}}(x) = T(x)$:

$$Q_{\text{con}} = Q_{\text{bath}} + \dot{m} \cdot C_p(T) \cdot (T - T_{\text{bath}})$$

$$k(T) \cdot A \cdot \frac{dT}{dx} = Q_{\text{bath}} + \dot{m} \cdot C_p(T) \cdot (T - T_{\text{bath}})$$

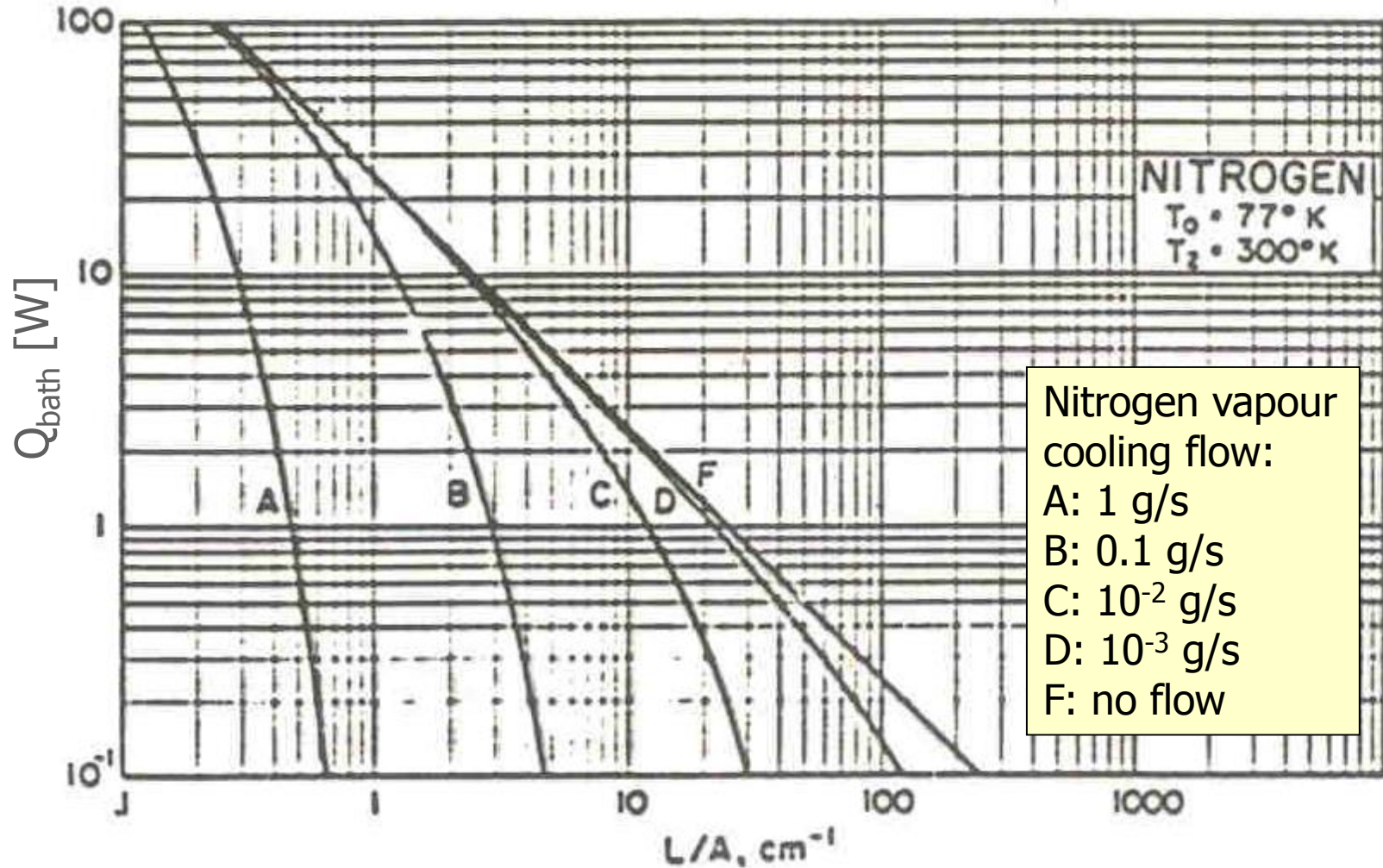
$C_p(T)$: Specific heat of vapour

$k(T)$: Thermal conductivity of the support

Q_{bath} can then be calculated by numerical integration for :

- different cryogenes,
- different values of aspect ratio L/A
- different values of vapour flow

Heat reaching the cold end of a stainless steel neck



Vapour cooling of necks and supports with perfect heat exchange in self-sustained mode

A particular case of gas cooling is the **self-sustained** mode, i.e. the vapour flow is generated only by the residual heat Q_{bath} reaching the bath. Then:

$$Q_{\text{bath}} = L_v \cdot \dot{m} \quad (\text{Lv: latent heat of vaporization})$$

Given the general equation

$$k(T) \cdot A \cdot \frac{dT}{dx} = Q_{\text{bath}} + \dot{m} \cdot C_p(T) \cdot (T - T_{\text{bath}})$$

After integration, we finally have:

$$Q_{\text{bath}} = \frac{A}{L} \cdot \int_{T_{\text{bath}}}^{T_{\text{ambient}}} \frac{K(T)}{1 + (T - T_{\text{bath}}) \cdot \frac{C_p(T)}{L_v}} \cdot dT$$

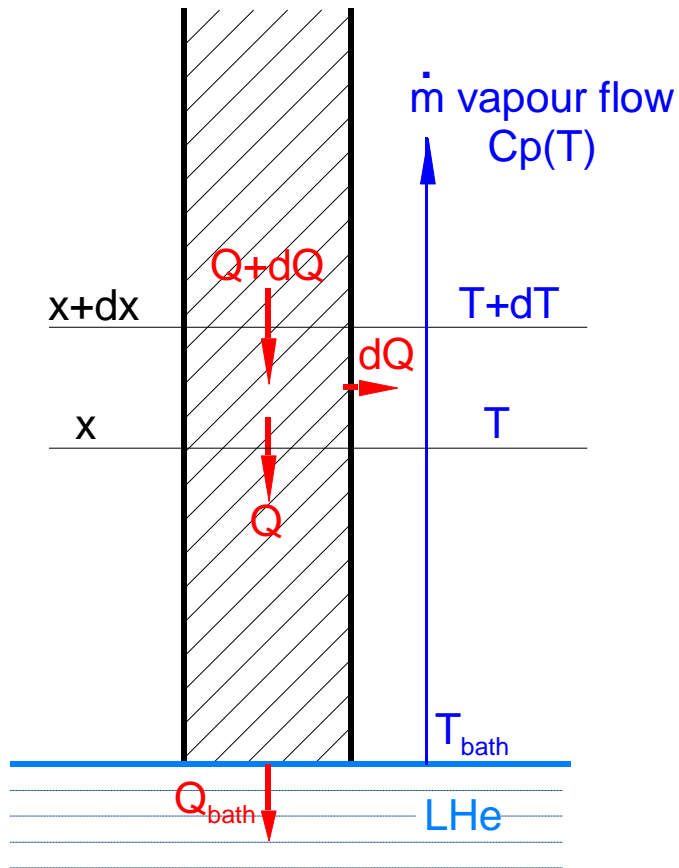
Attenuation factor w.r. to pure conduction

Reduction of heat conduction by self-sustained helium vapour cooling

Effective thermal conductivity integral from 4 to 300 K	Purely conductive regime [W.cm⁻¹]	Self-sustained vapour-cooling [W.cm⁻¹]
ETP copper	1620	128
OFHC copper	1520	110
Aluminium 1100	728	39.9
Nickel 99% pure	213	8.65
Constantan	51.6	1.94
AISI 300 stainless steel	30.6	0.92

Vapour cooling of necks and supports with imperfect heat exchange

Cross-section A



$$dQ = f \cdot \dot{m} \cdot C_p(T) \cdot dT$$

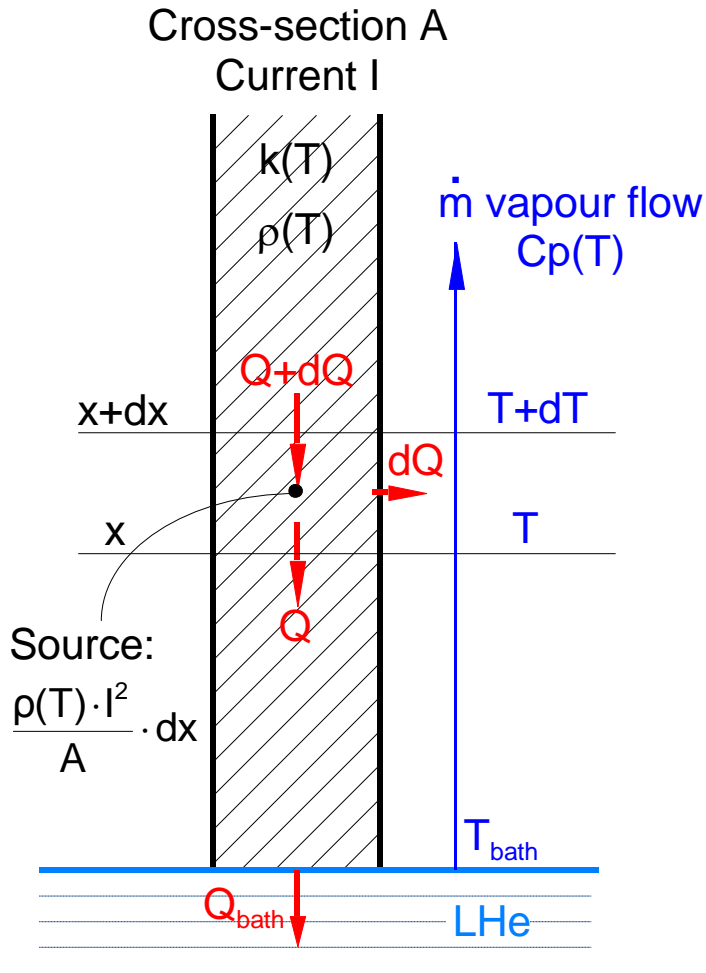
With f , the efficiency of the heat transfer

In steady state, the heat balance equation becomes:

$$\frac{d}{dx} \left[k(T) \cdot A \cdot \frac{dT}{dx} \right] = f \cdot \dot{m} \cdot C_p(T) \cdot \frac{dT}{dx}$$

→ Numerical integration for solving this equation

Vapor-cooled current leads



$\rho(T)$: electrical resistivity

$$dQ = f \cdot \dot{m} \cdot C_p(T) \cdot dT$$

In steady-state, heat balance equation:

$$\frac{d}{dx} \left[k(T) \cdot A \cdot \frac{dT}{dx} \right] - f \cdot \dot{m} \cdot C_p(T) \cdot \frac{dT}{dx} + \frac{\rho(T) \cdot I^2}{A} = 0$$

Solid conduction
Vapour cooling
Joule heating

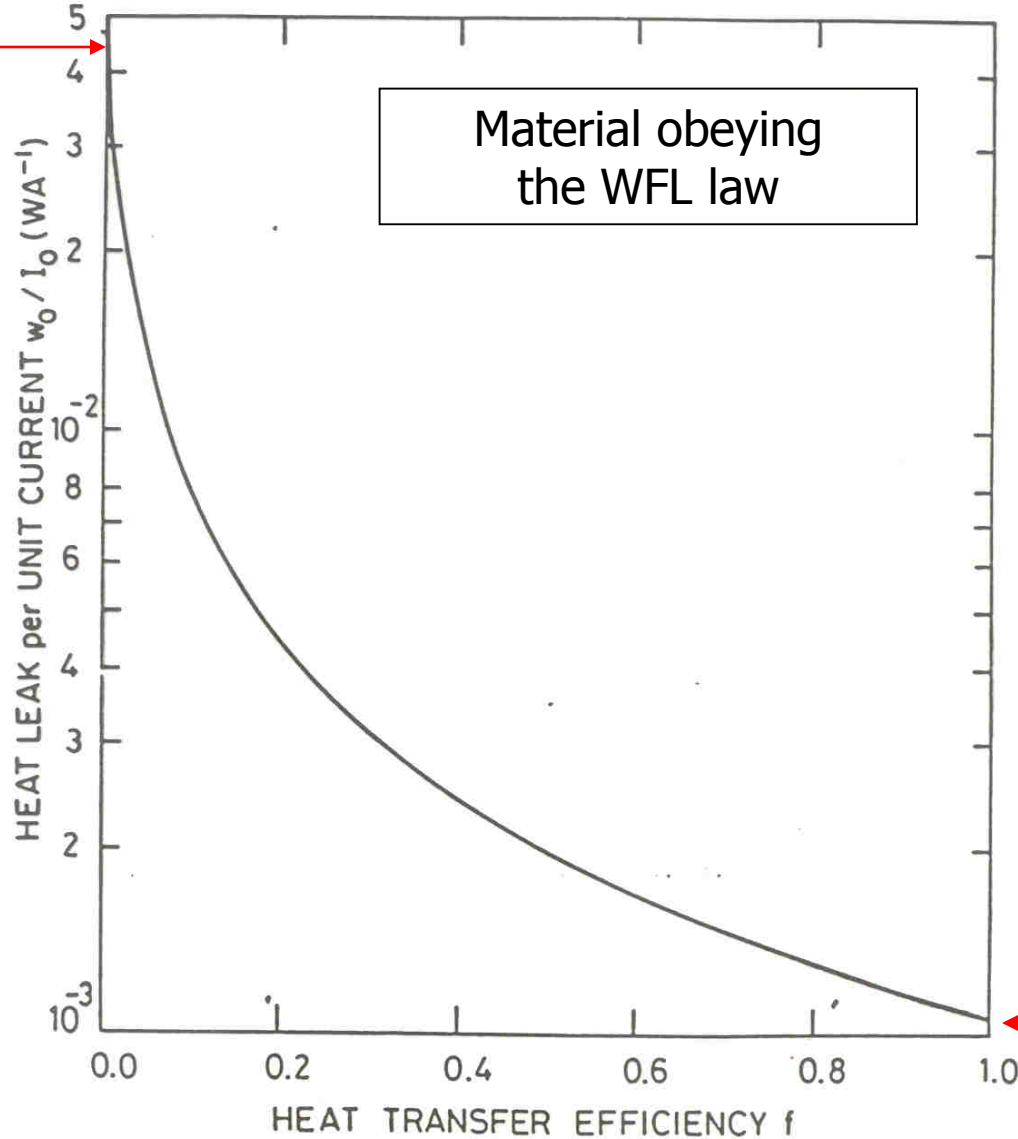
Assuming the material of the lead follows the Wiedemann-Franz-Lorenz (WFL) law:

$$k(T) \cdot \rho(T) = L_0 \cdot T$$

L_0 : Lorenz number ($2.45 \cdot 10^{-8} \text{ W} \cdot \Omega \cdot \text{K}^{-2}$)

→ Then numerical integration

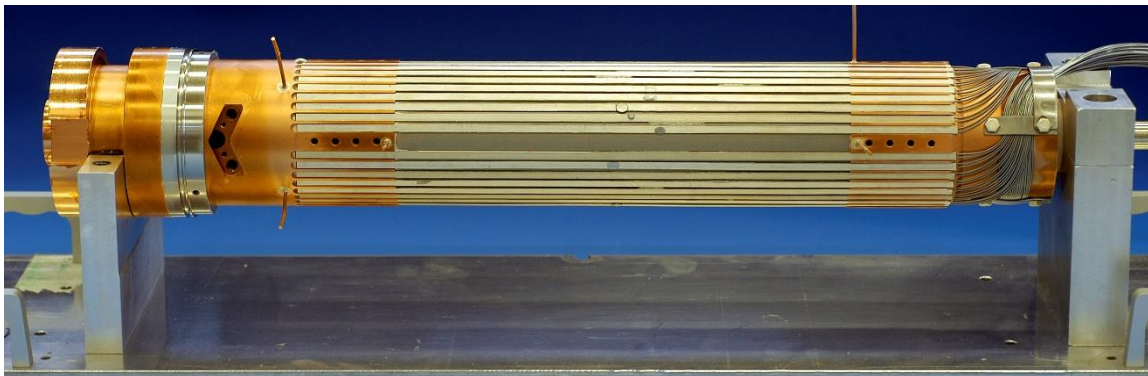
Uncooled
47 W/kA



Minimum residual
heat load
1.04 W/kA

Beating the WFL law: HTS current leads

- The WFL law essentially states that good electrical conductors are also good thermal conductors
- Current leads need good electrical conductors with low thermal conductivity
- Superconductors are bad thermal conductors with zero resistivity
- Build current lead with superconductor up to temperature as high as possible, i.e. HTS

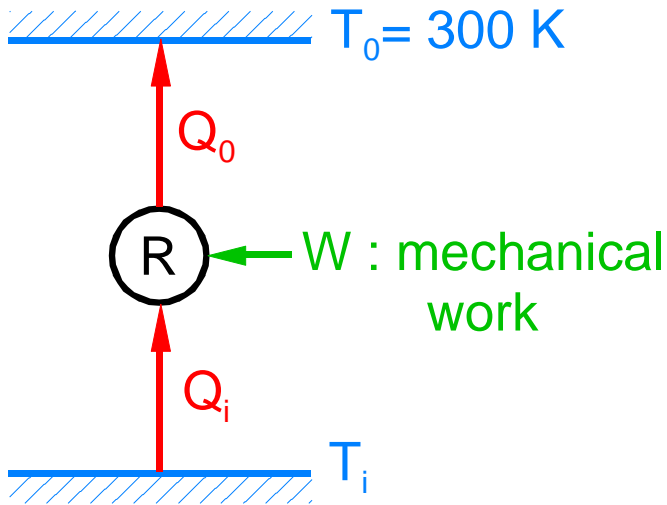


HTS vs. normal conducting current leads

Type		Resistive	HTS (4 to 50 K) Resistive (above)
Heat into LHe	[W/kA]	1.1	0.1
Total exergy consumption	[W/kA]	430	150
Electrical power from grid	[W/kA]	1430	500

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First principle [Joule]

$$Q_0 = Q_i + W$$

Second principle [Clausius]

$$\frac{Q_0}{T_0} \geq \frac{Q_i}{T_i}$$

(= for reversible process)

Hence, $W \geq T_0 \cdot \frac{Q_i}{T_i} - Q_i$ which can be written in three different ways:

① $W \geq T_0 \cdot \Delta S_i - Q_i$ introducing **entropy S** as

$$\Delta S_i = \frac{Q_i}{T_i}$$

② $W \geq Q_i \cdot \left(\frac{T_0}{T_i} - 1 \right)$ ← **Carnot factor**

③ $W \geq \Delta E_i$ introducing **exergy E** as

$$\Delta E_i = Q_i \cdot \left(\frac{T_0}{T_i} - 1 \right)$$

Minimum refrigeration work

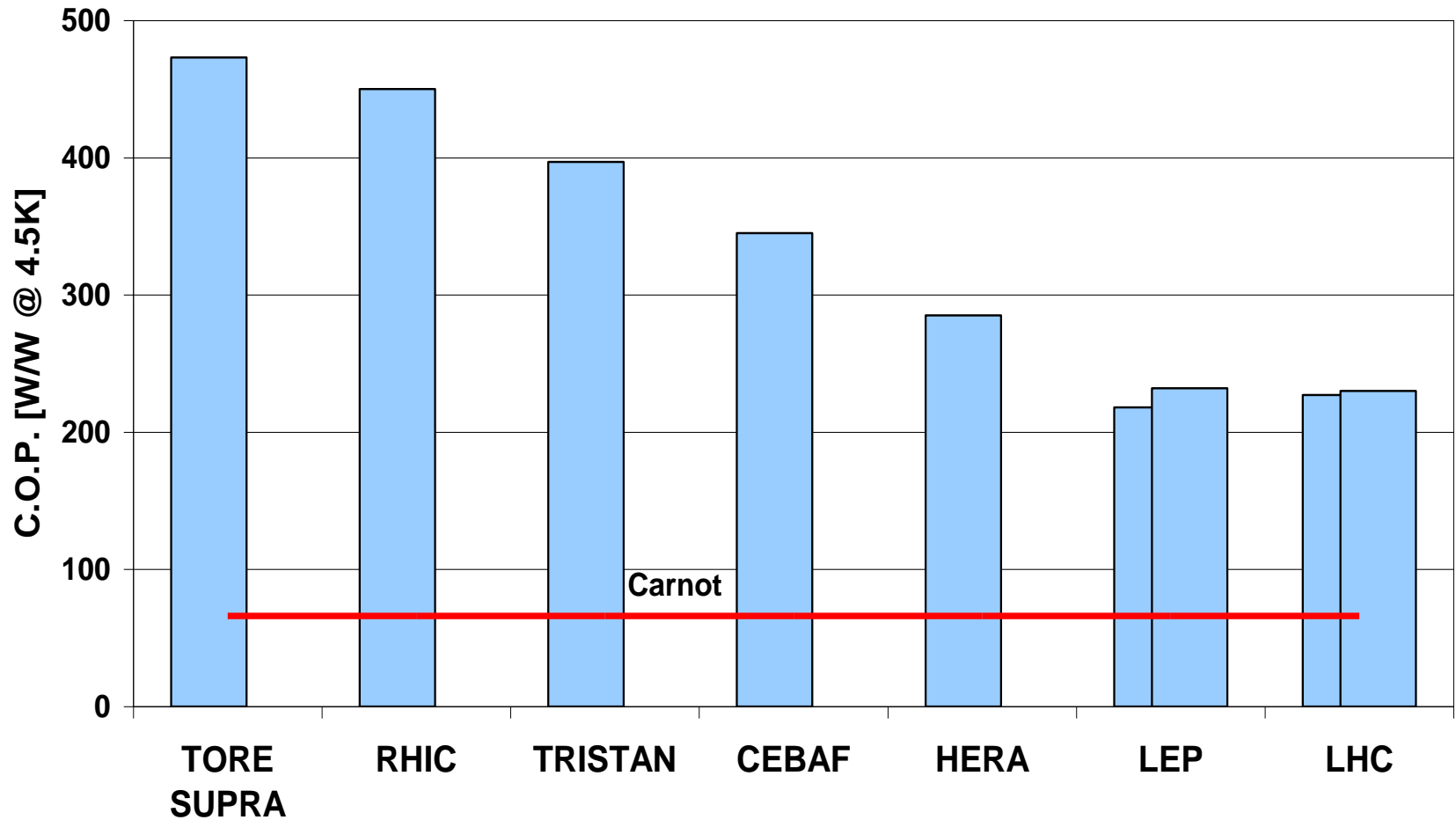
Consider the extraction of 1 W at 4.5 K, rejected at 300 K
The minimum refrigeration work (equation 2) is:

$$W_{\min} = Q_i \cdot \left(\frac{T_0}{T_i} - 1 \right) = 1 \cdot \left(\frac{300}{4.5} - 1 \right) = 65.7 \text{ W}$$

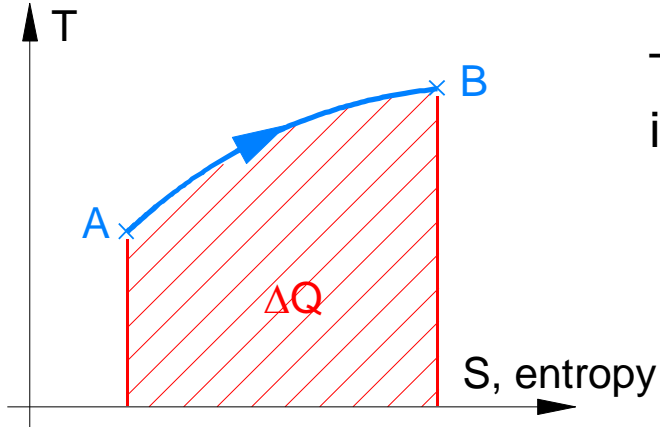
In practice, the most efficient helium refrigerators have an efficiency of about 30% w.r. to the Carnot limit

$$\Rightarrow W_{\text{real}} = \frac{W_{\min}}{\eta} = \frac{65.7}{0.3} = 220 \text{ W}$$

C.O.P. of large cryogenic helium refrigerators



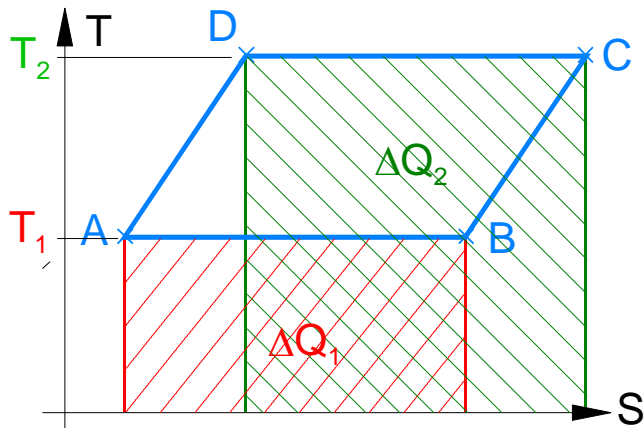
Introduction to the T-S diagram



Thermodynamic transformation from A to B, if reversible:

$$\Delta Q = \int_A^B T \cdot dS$$

To make a refrigeration cycle, need a substance, the entropy of which depends on some other **variable** than temperature



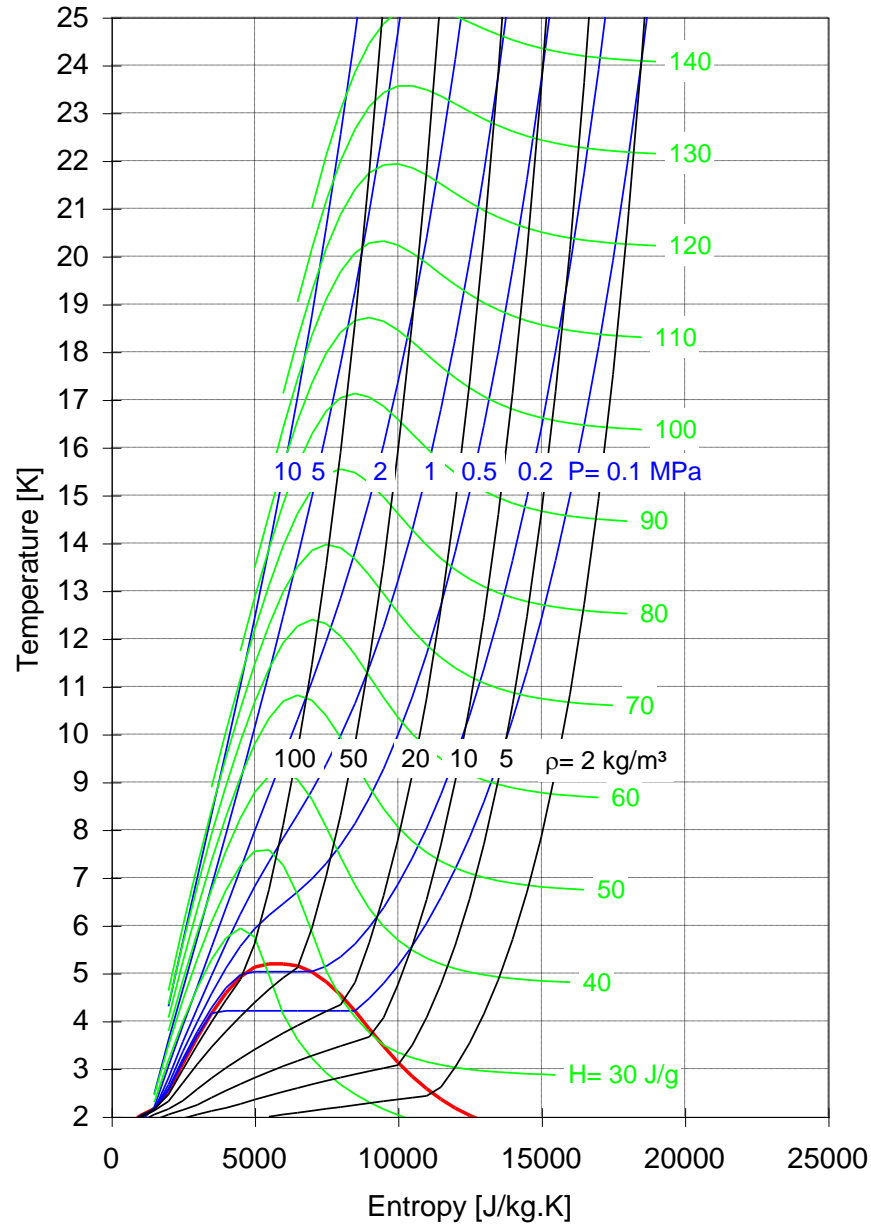
Pressure of gas: Compression/expansion cycle
Magnetization of solid: magnetic refr. cycle

ΔQ_1 : heat absorbed at T_1

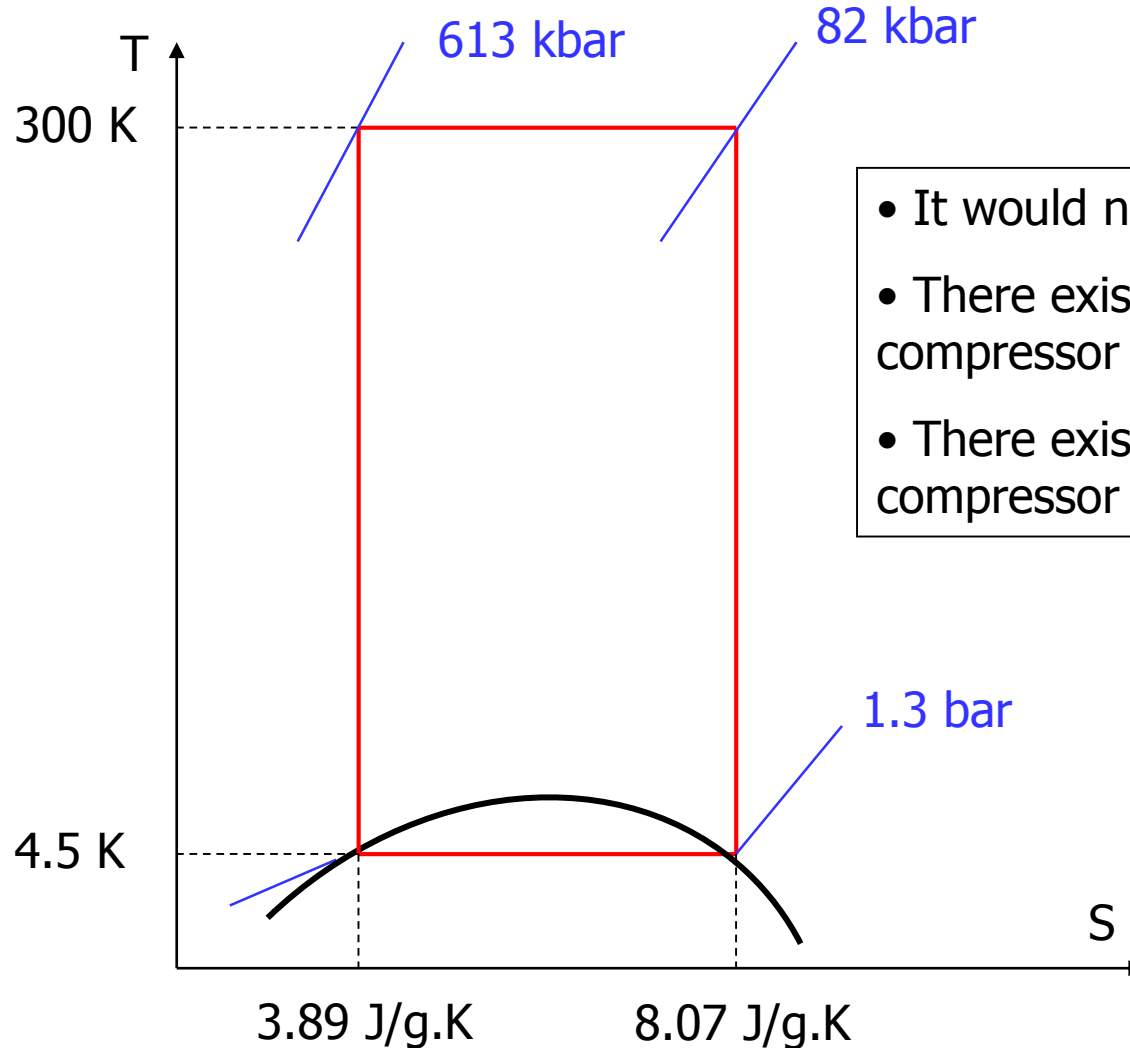
ΔQ_2 : heat rejected at T_2

→ Refrigeration cycle A B C D

T-S diagram for helium

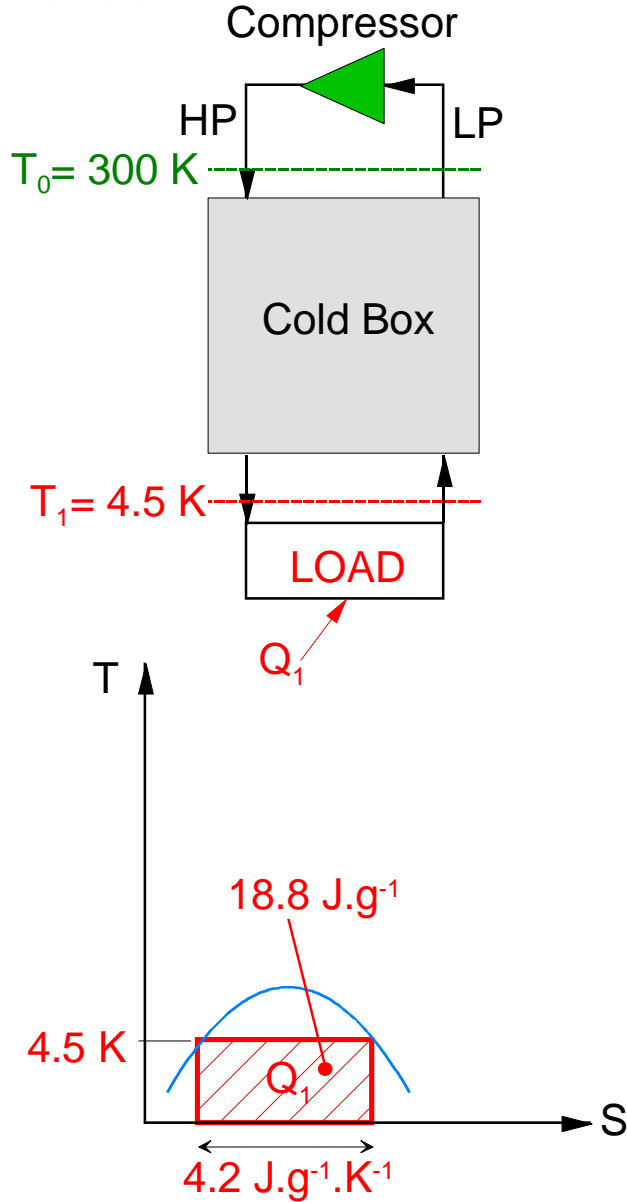


A Carnot cycle is not feasible for helium liquefaction

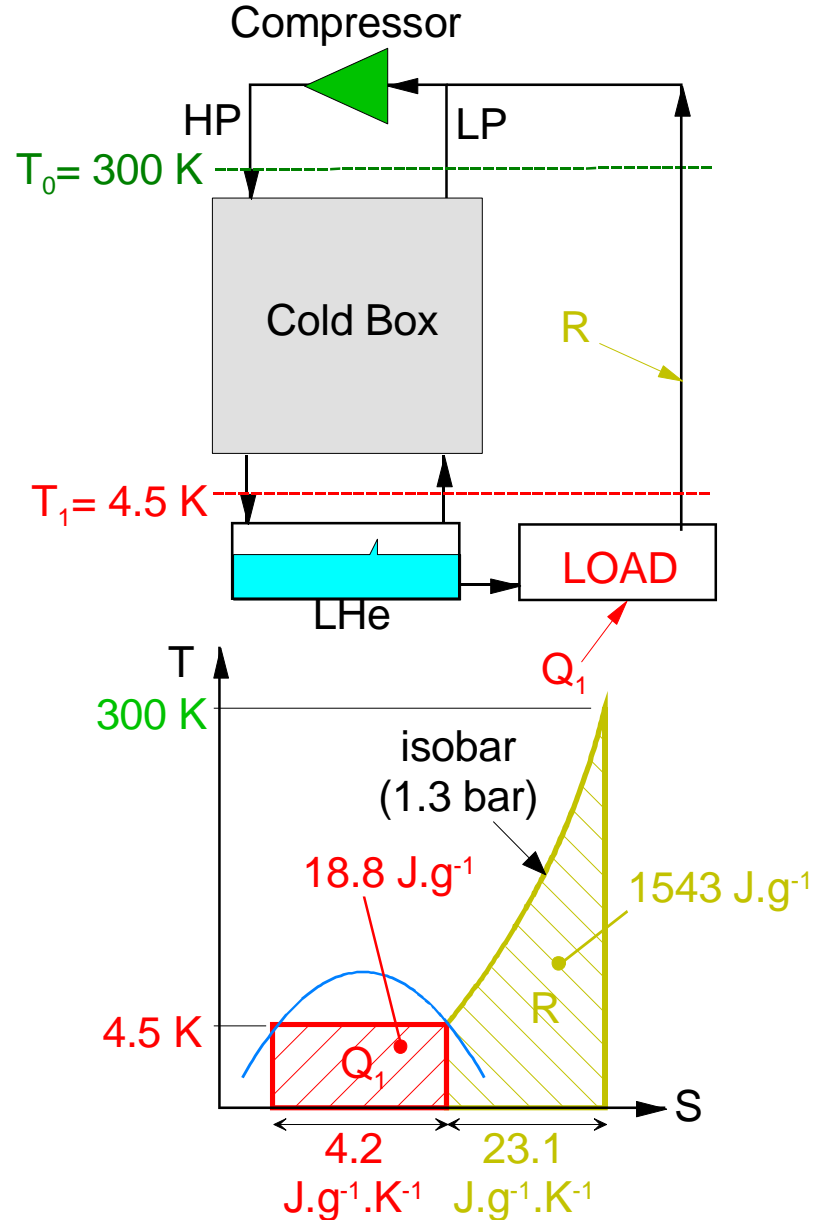


- It would need a HP of 613 kbar!
- There exists no true isothermal compressor
- There exists no true isentropic compressor or expander

Refrigerator



Liquefier



Thermodynamic equivalence between refrigeration and liquefaction

What is the isothermal 4.5 K (T_1) refrigeration equivalent to 1 g.s⁻¹ He liquefaction?

$$\dot{W}_{\text{min.lique}} = \dot{m}_{\text{lique}} \cdot (T_0 \cdot \Delta S - Q_1 - R)$$

$$\dot{m}_{\text{lique}} = 1 \text{ g.s}^{-1}, T_0 = 300 \text{ K}, \Delta S = 27.3 \text{ J.g}^{-1}.\text{K}^{-1}, Q_1 = 18.8 \text{ J.g}^{-1}, R = 1543 \text{ J.g}^{-1}$$

$$\dot{W}_{\text{min.lique}} = 6628 \text{ W}$$

Write that the same work is used to produce isothermal refrigeration at 4.5 K:

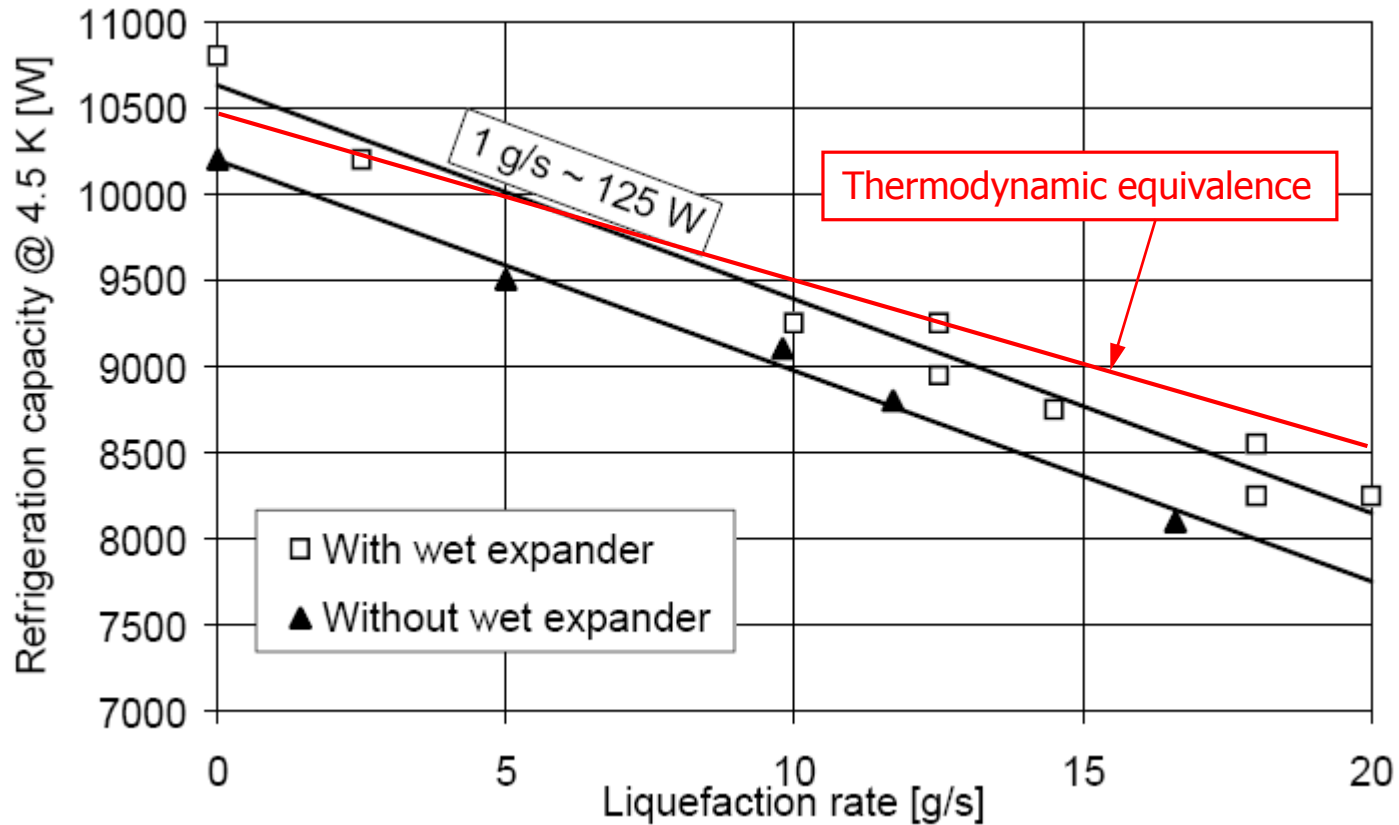
$$\dot{W}_{\text{min.refrig}} = \dot{Q}_1 \cdot \left(\frac{T_0}{T_1} - 1 \right) \quad \Rightarrow \quad \dot{Q}_1 = 100 \text{ W}$$

$$\dot{W}_{\text{min.refrig}} = \dot{W}_{\text{min.lique}} = 6628 \text{ W}$$

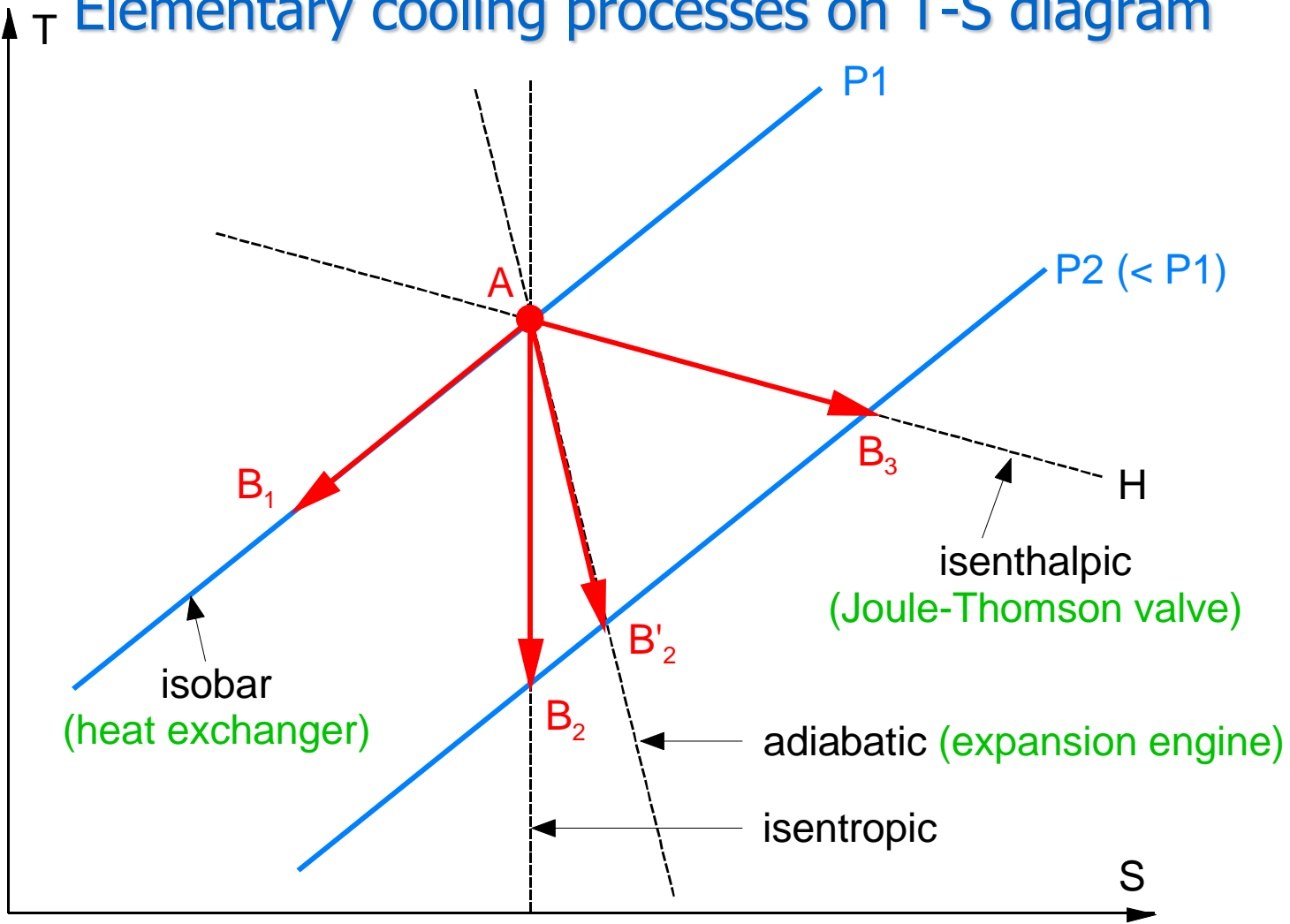
For refrigerators/liquefiers with the same efficiency:

$$1 \text{ g.s}^{-1} \text{ LHe} \equiv 100 \text{ W @ 4.5 K}$$

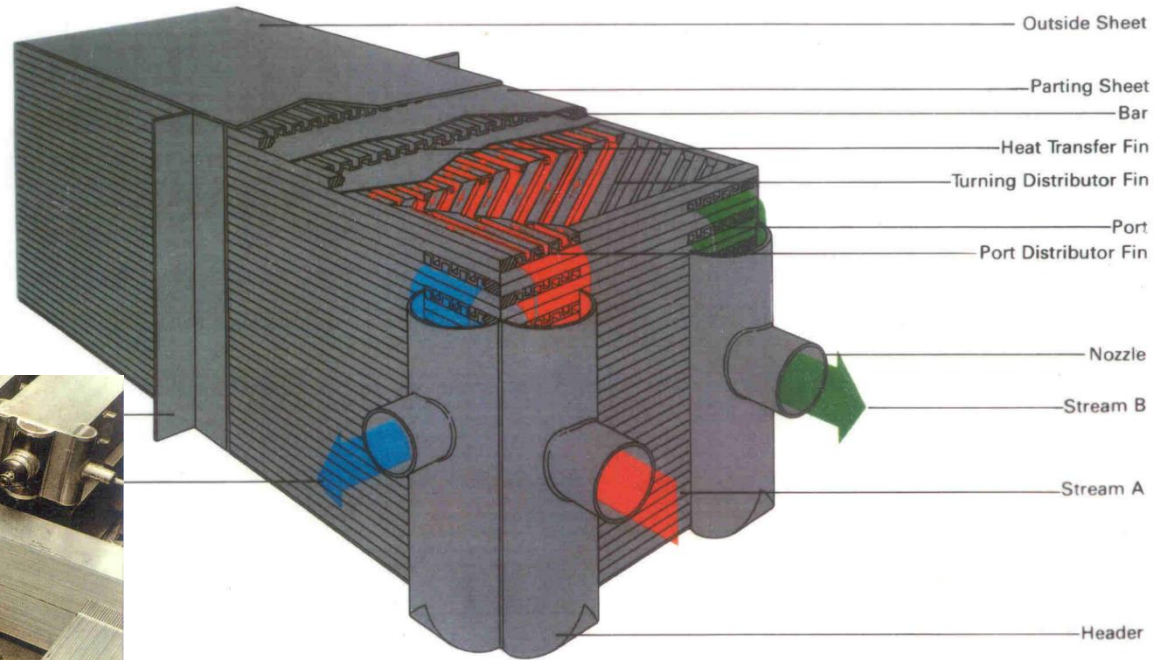
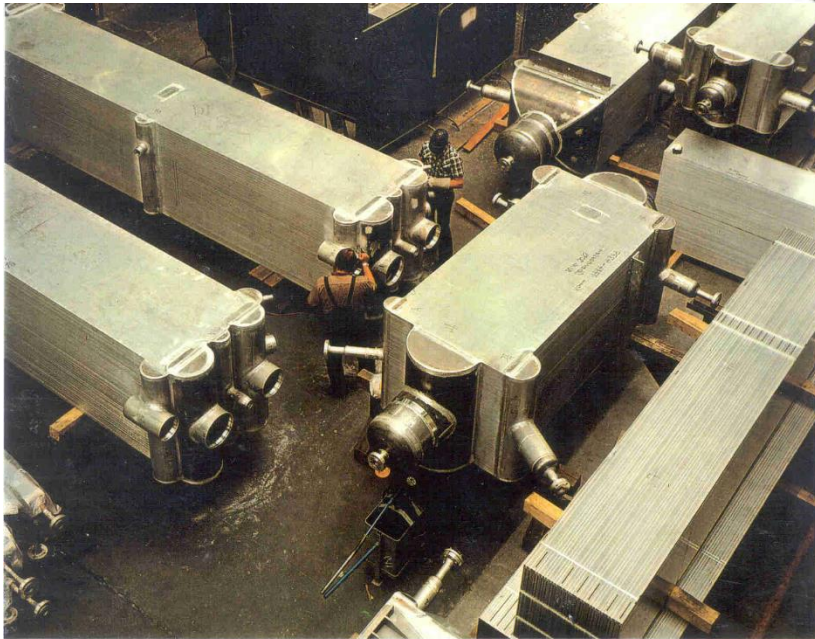
Measured refrigeration/liquefaction equivalence *12 kW @ 4.5 K helium refrigerators for LEP 2*



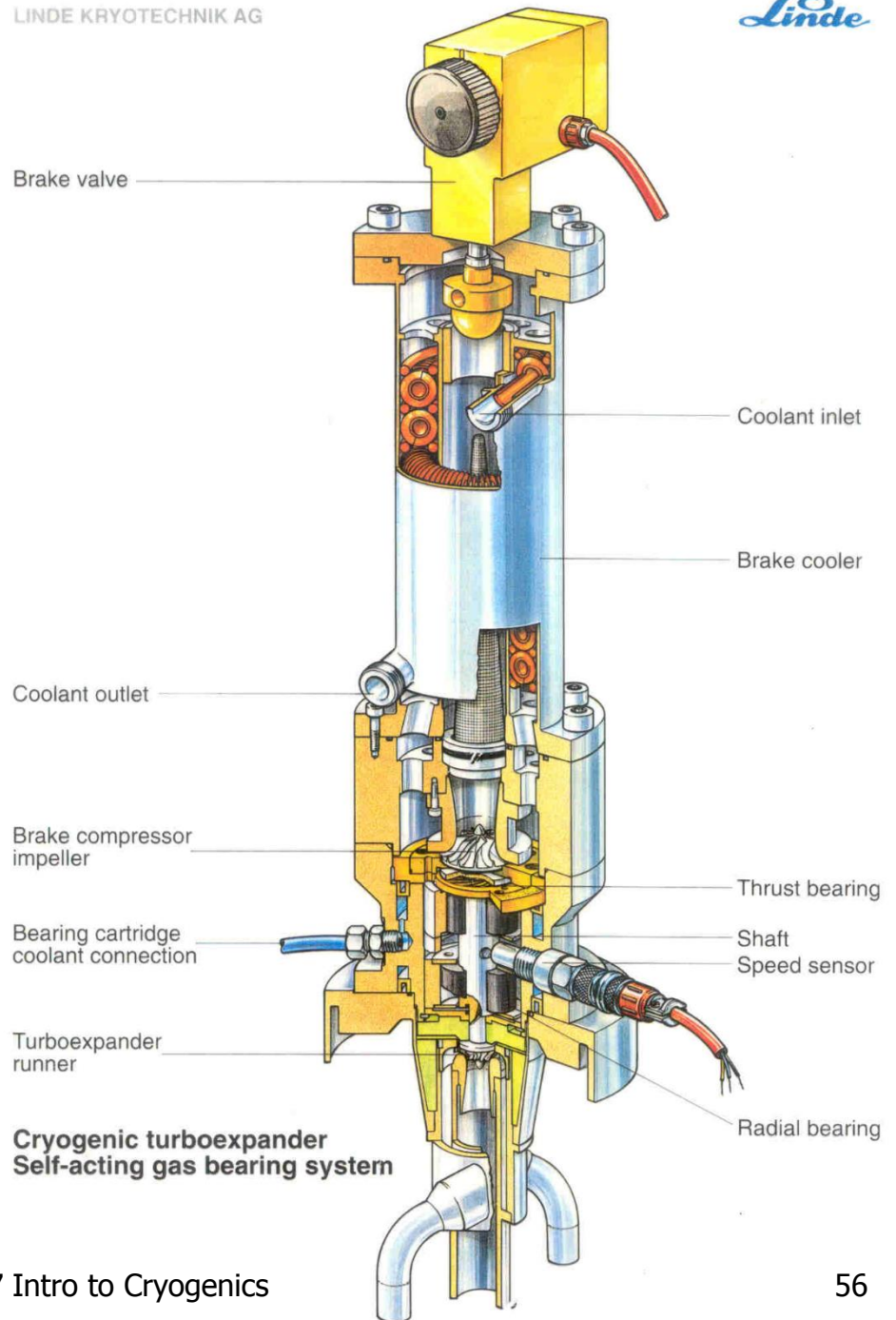
Elementary cooling processes on T-S diagram



Brazed aluminium plate heat exchanger

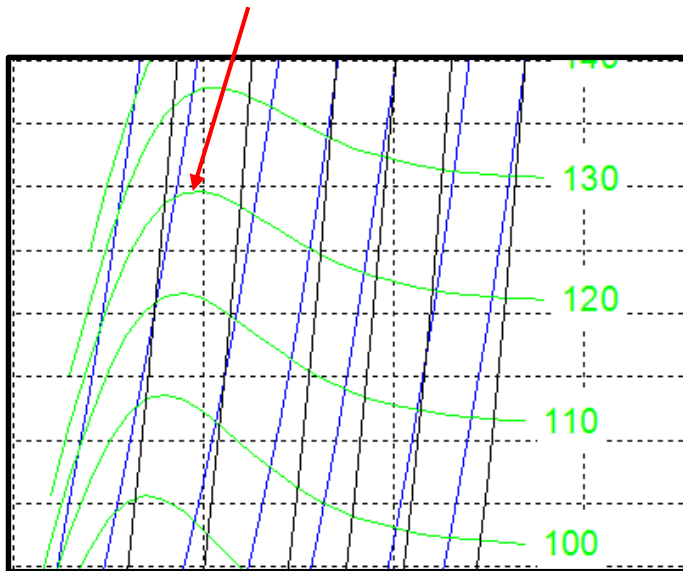


Cryogenic turbo-expander



Joule-Thomson inversion temperatures

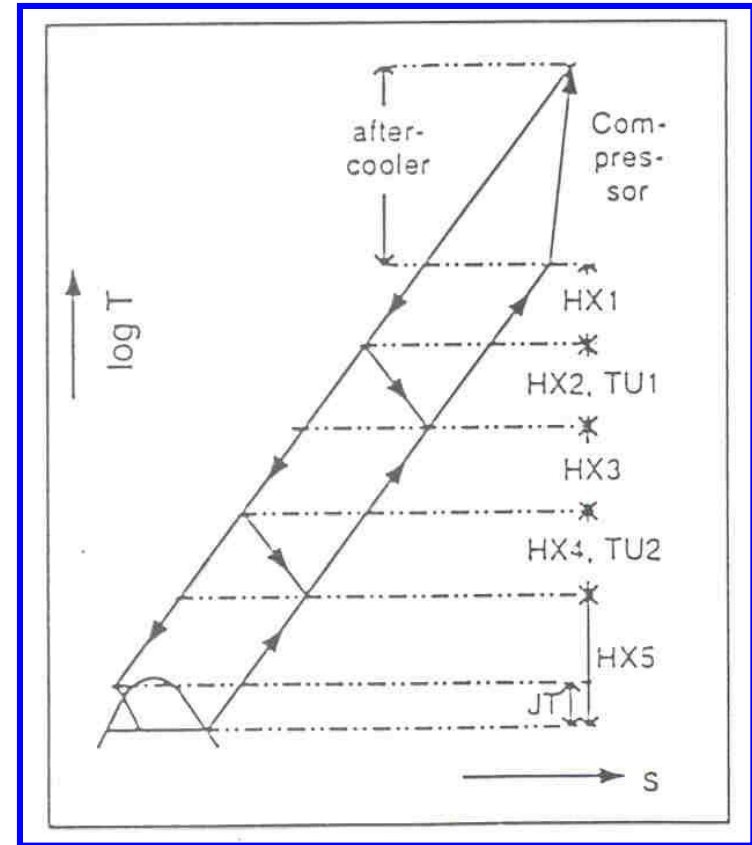
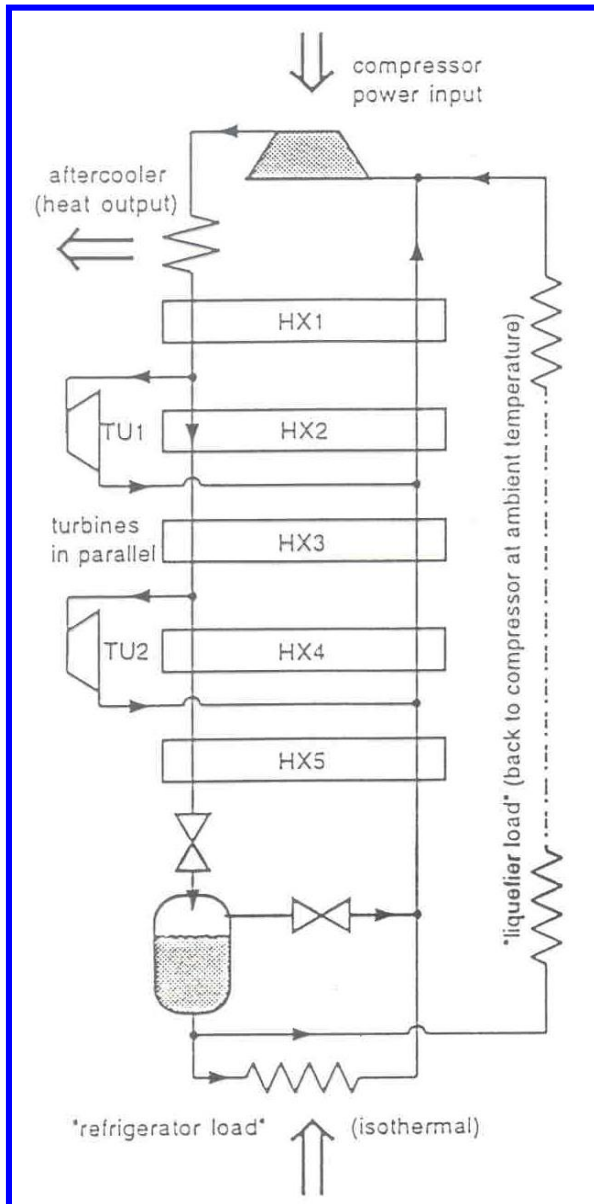
Isenthalps in T-S diagram can have positive or negative slope, i.e. isenthalpic expansion can produce warming or cooling
 ⇒ **inversion temperature**



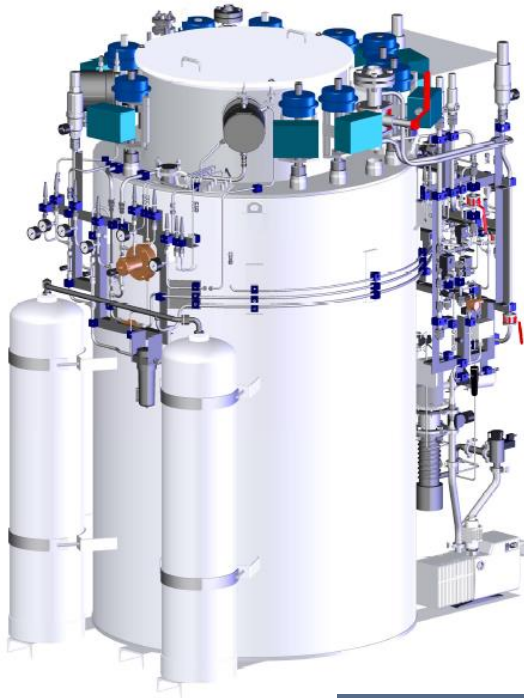
Cryogen	Maximum inversion temperature [K]
Helium	43
Hydrogen	202
Neon	260
Air	603
Nitrogen	623
Oxygen	761

While air can be cooled down and liquefied by JT expansion from room temperature, helium and hydrogen need precooling down to below inversion temperature by heat exchange or work-extracting expansion (e.g. in turbines)

Two-stage Claude cycle



Air Liquide & Linde

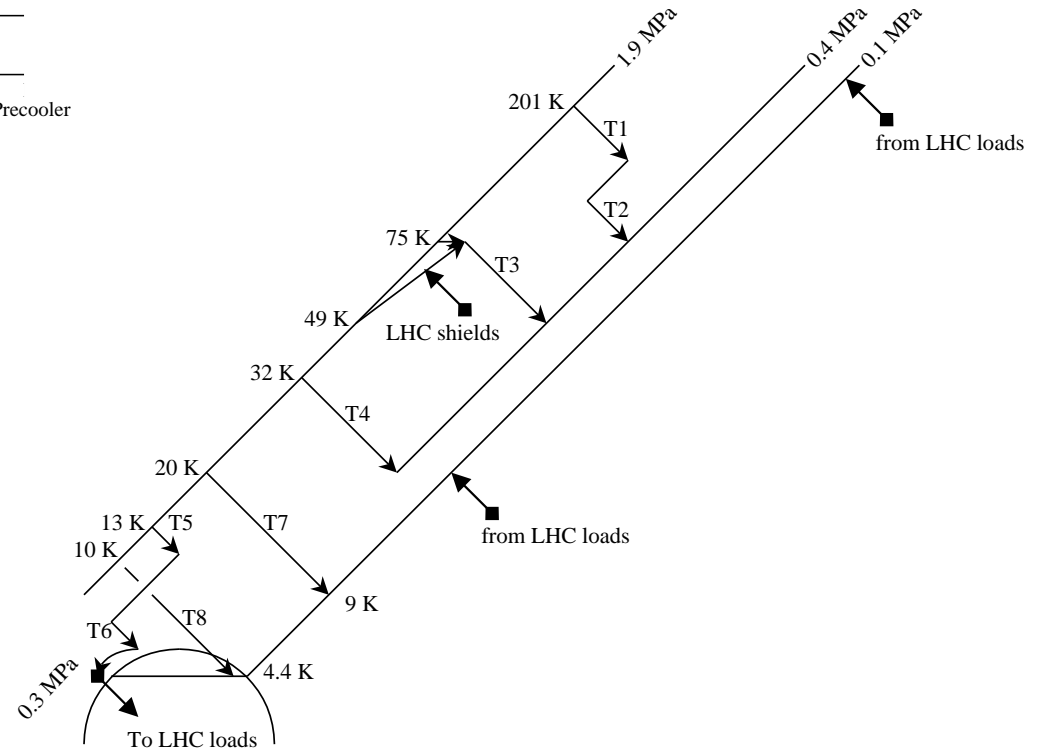
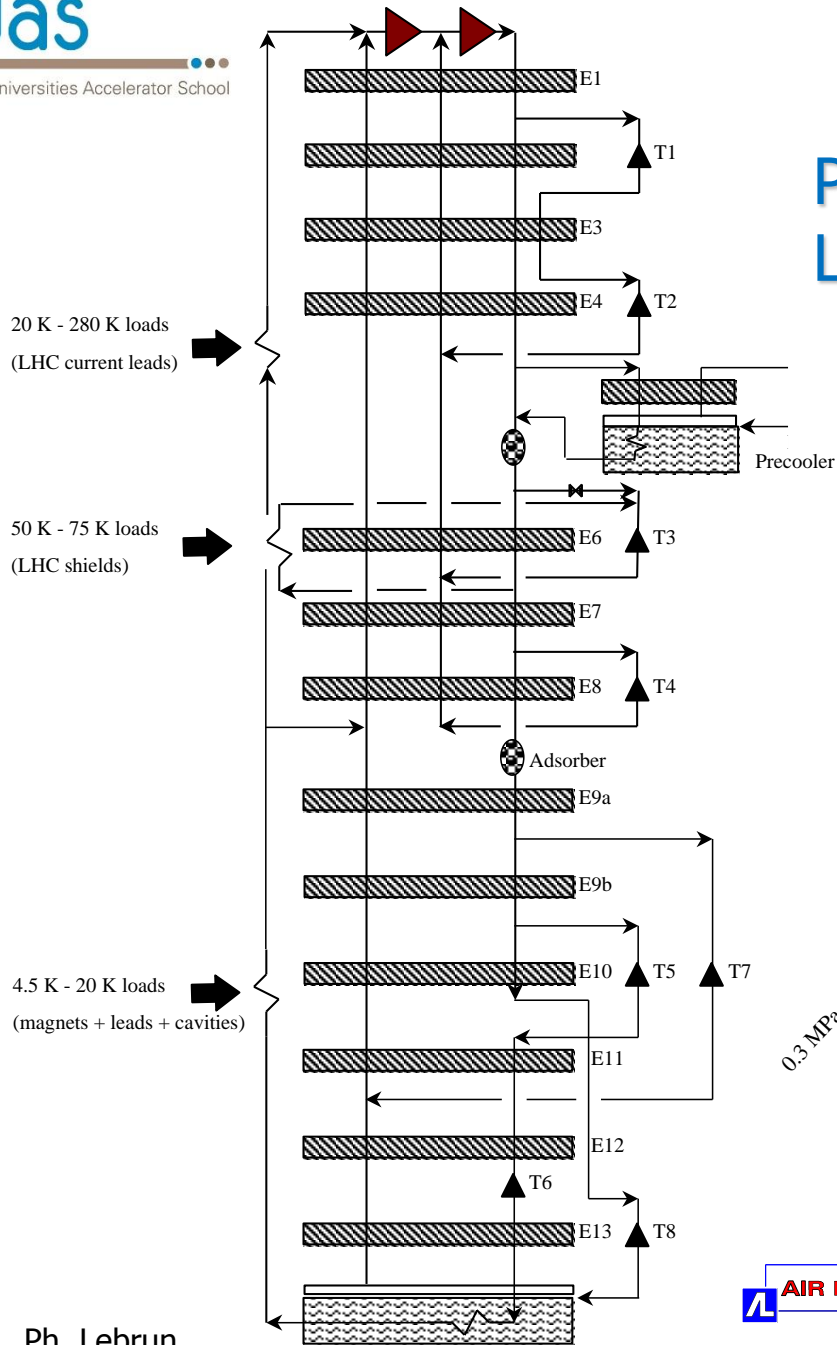


	<i>HELIAL SL</i>	<i>HELIAL ML</i>	<i>HELIAL LL</i>
Max. Liquefaction capacity without LN2	25 L/h	70 L/h	145 L/h
Max. Liquefaction capacity with LN2	50 L/h	150 L/h	330 L/h
Compressor electrical motor	55 kW	132 kW	250 kW
Specific consumption for liquefaction w/o LN2	645 W/W	552 W/W	505 W/W
% Carnot	10%	12%	13%



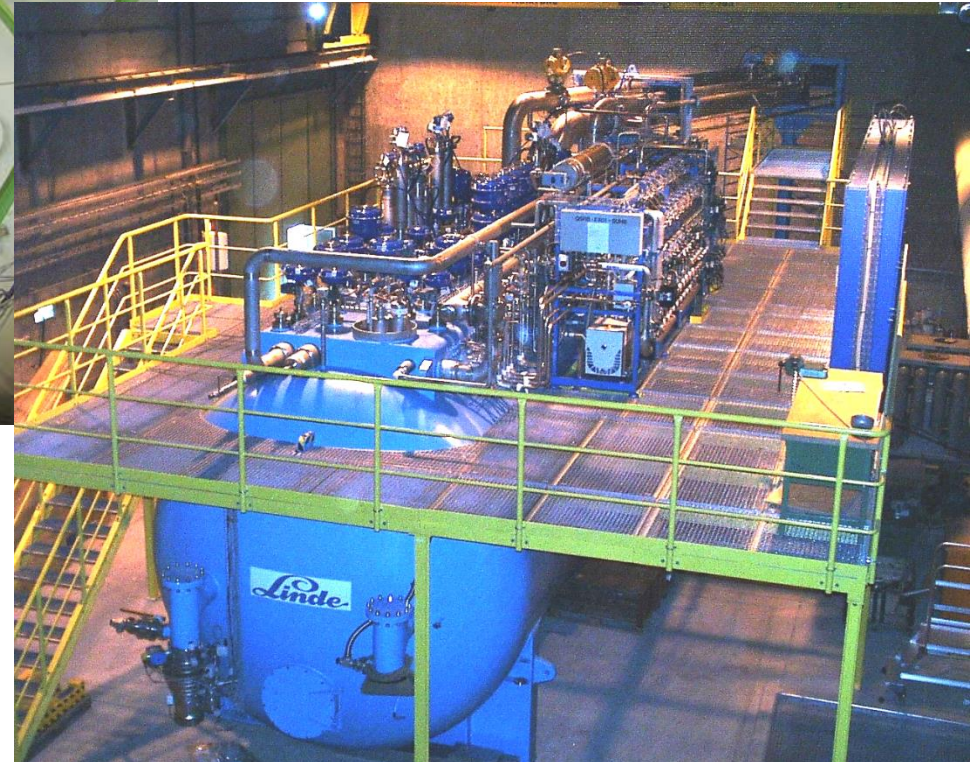
	Without LN ₂ precooling	With LN ₂ precooling
L70	20 – 35 l/h	40 – 70 l/h
L140	45 – 70 l/h	90 – 140 l/h
L280	100 – 145 l/h	200 – 290 l/h
LR70	100 – 145 Watt	130 – 190 Watt
LR140	210 – 290 Watt	255 – 400 Watt
LR280	445 – 640 Watt	560 – 900 Watt

Process cycle & T-S diagram of LHC 18 kW @ 4.5 K cryoplant

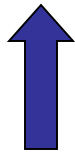


LHC 18 kW @ 4.5 K helium cryoplants

33 kW @ 50 K to 75 K
23 kW @ 4.6 K to 20 K
41 g/s liquefaction
4 MW compressor power
C.O.P. 220-230 W/W @ 4.5 K



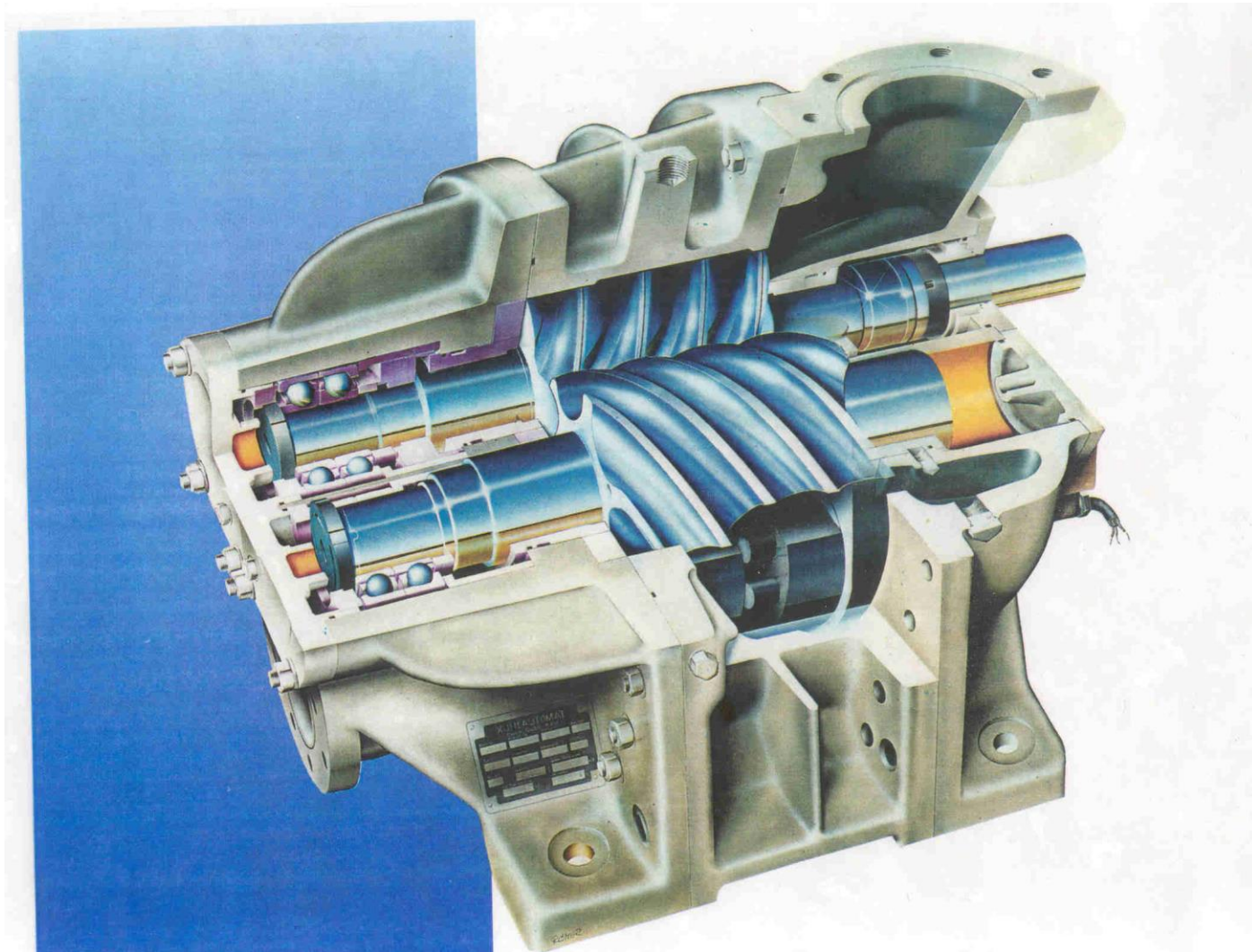
Air Liquide



Linde



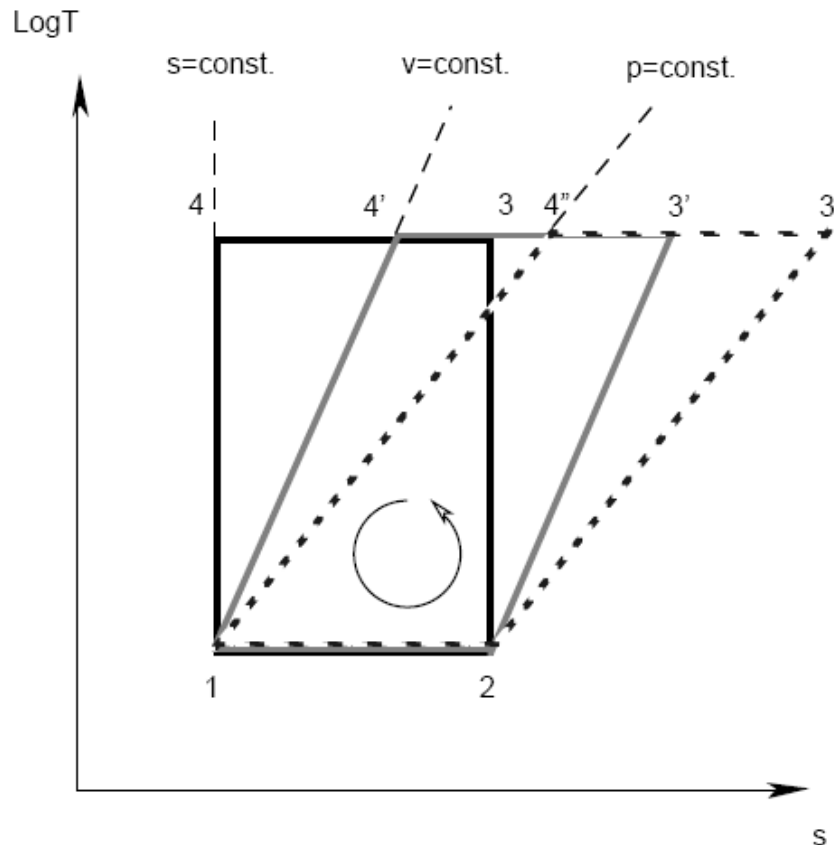
Oil-injected screw compressor



Compressor station of LHC 18 kW@ 4.5 K helium refrigerator



Carnot, Stirling and Ericsson cycles



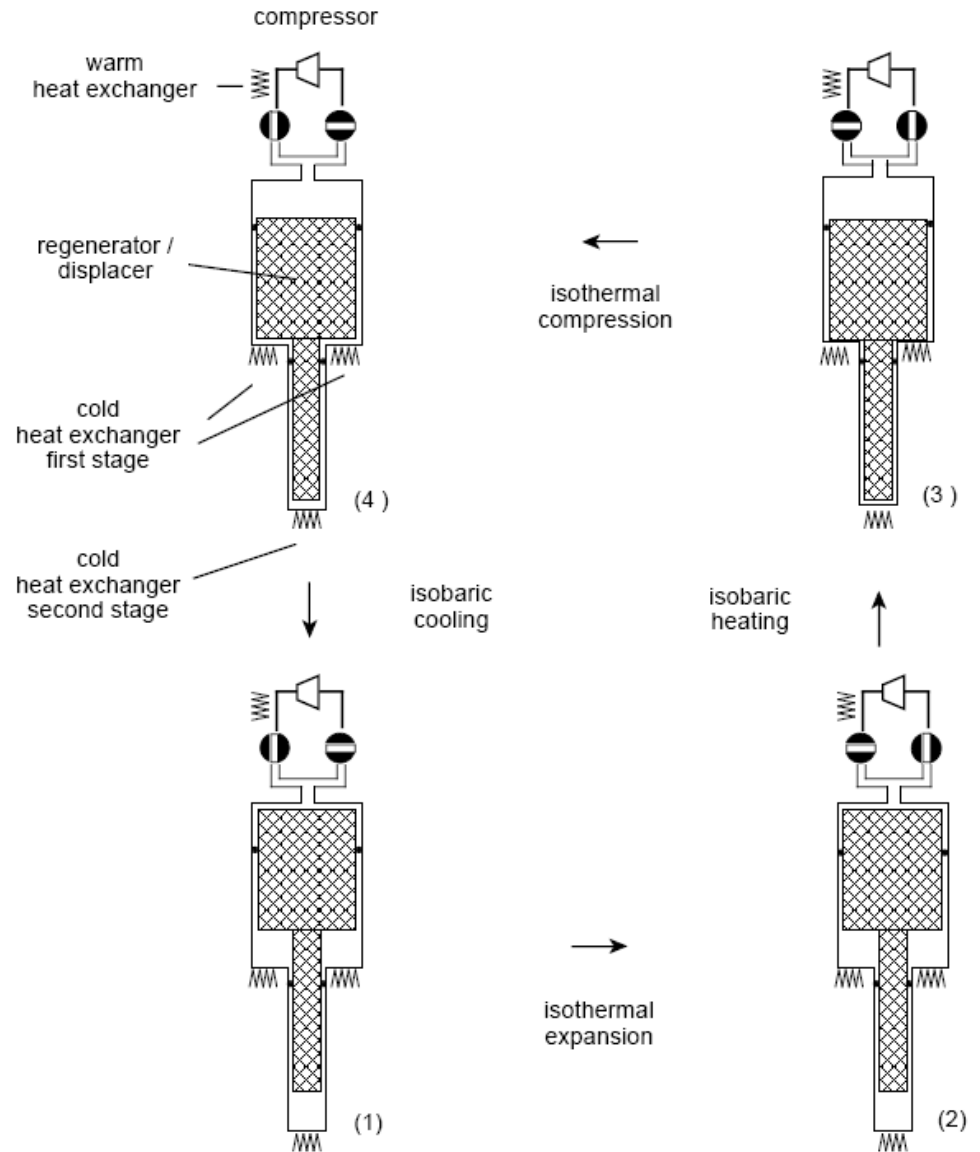
All «sloping» cycles need internal heat exchange

For small machines, this is done by regenerative, rather than recuperative heat exchangers

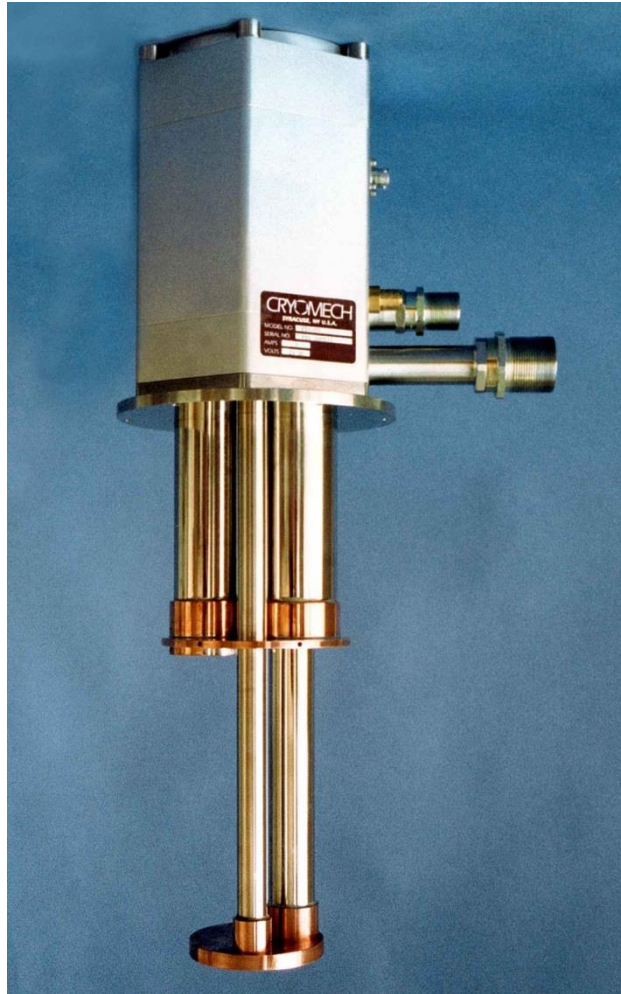
⇒ alternating rather than continuous operation

Carnot cycle (1,2,3,4), Stirling cycle (1,2,3',4') and Ericsson cycle (1,2,3'',4'')

Operation of a Gifford-McMahon cryocooler (Ericsson cycle)



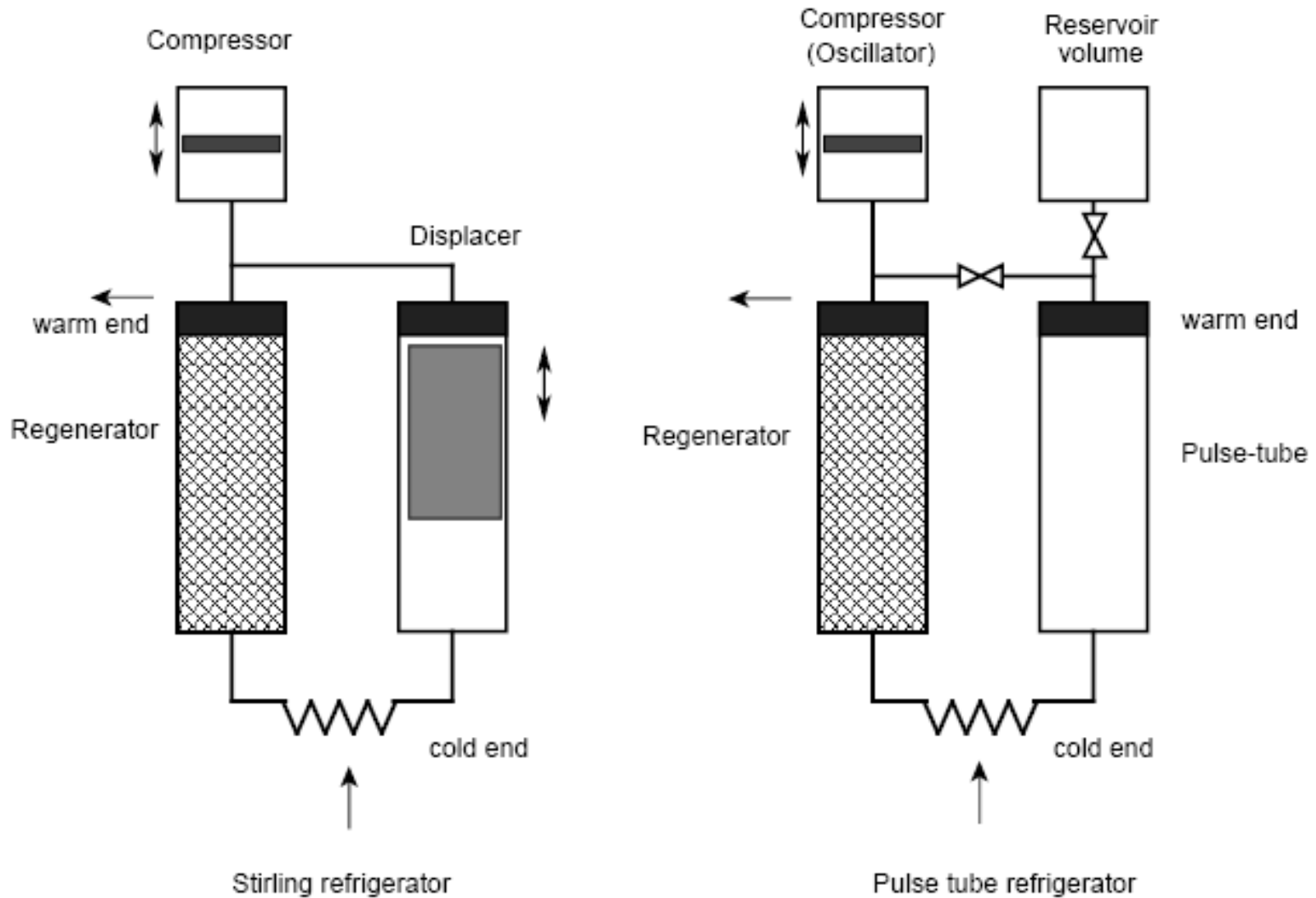
Two-stage Gifford-McMahon cryocooler



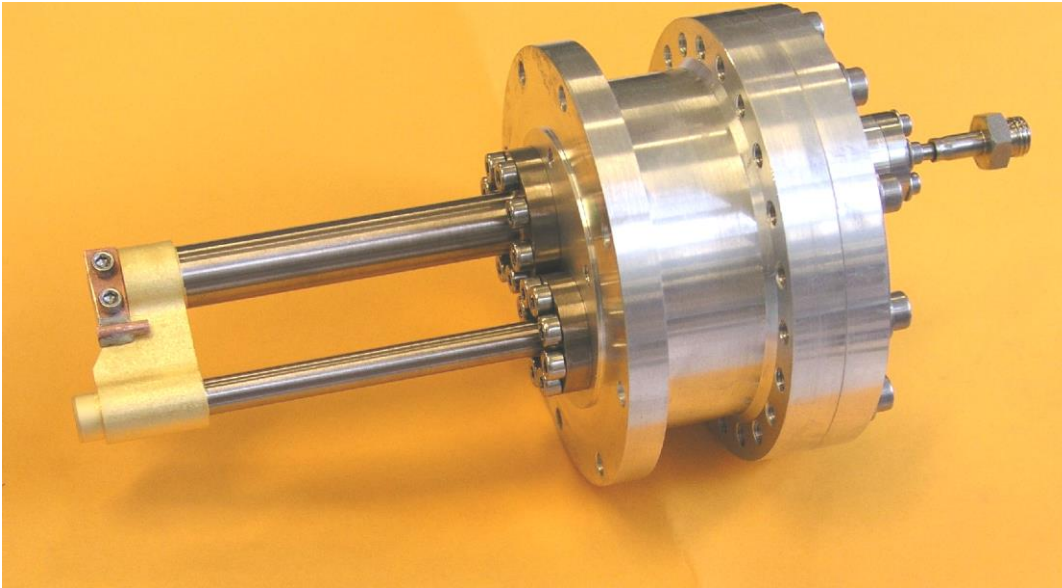
CRYOMECH PT407 & CP970 compressor
~ 0.7 W @ 4.2 K & 25 W @ 55 K



Stirling and pulse-tube cryocoolers



Mini pulse-tube cryocoolers



ESA MPTC development model – 1W @ 77K



CEA/SBT coaxial PTC– 6W @ 80K

Contents

- › Introduction
- › Cryogenic fluids
- › Heat transfer & thermal insulation
- › Thermal screening with cold vapour
- › Refrigeration & liquefaction
- **Cryogen storage & transport**
- › Thermometry

Specific cost of bulk He storage

Type	Pressure [MPa]	Density [kg/m ³]	Dead volume [%]	Cost [CHF/kg He]
Gas Bag	0.1	0.16	0	300 ⁽¹⁾
MP Vessel	2	3.18	5-25	220-450
HP Vessel	20	29.4	0.5	500 ⁽²⁾
Liquid	0.1	125	13	100-200 ⁽³⁾

(1): Purity non preserved

(2): Not including HP compressors

(3): Not including reliquefier

Bulk helium storage solutions

11000 gallon liquid container



2 MPa gas tanks

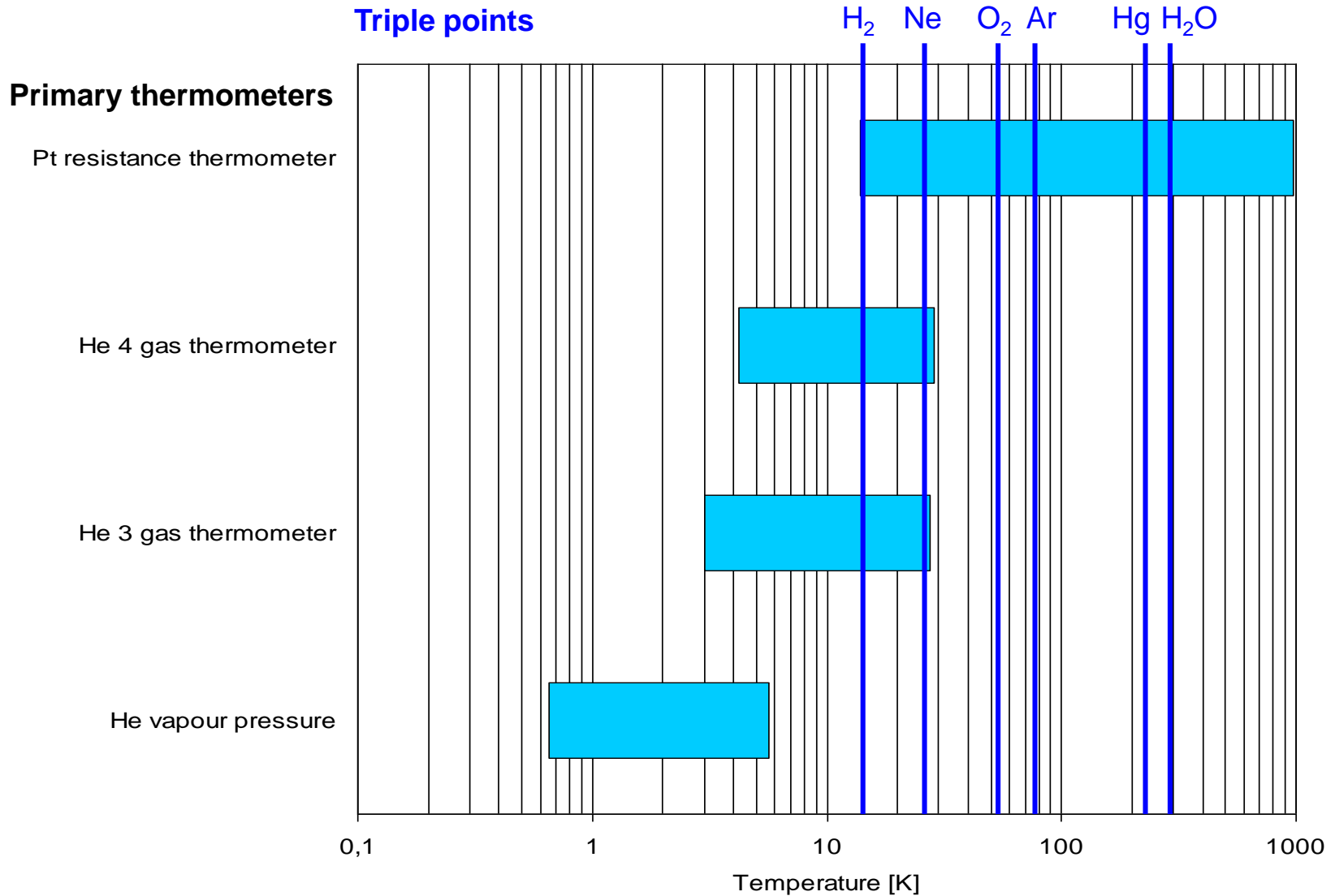


20 MPa gas cylinders

Contents

- Introduction
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Definition of ITS90 in cryogenic range

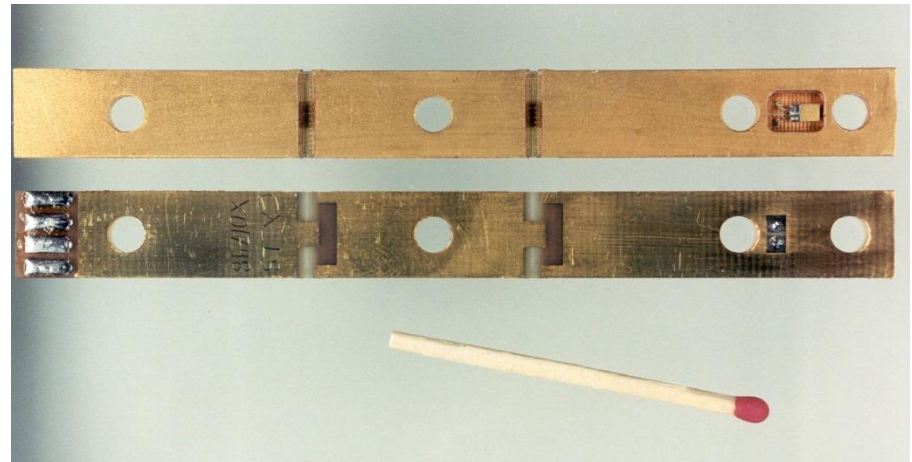
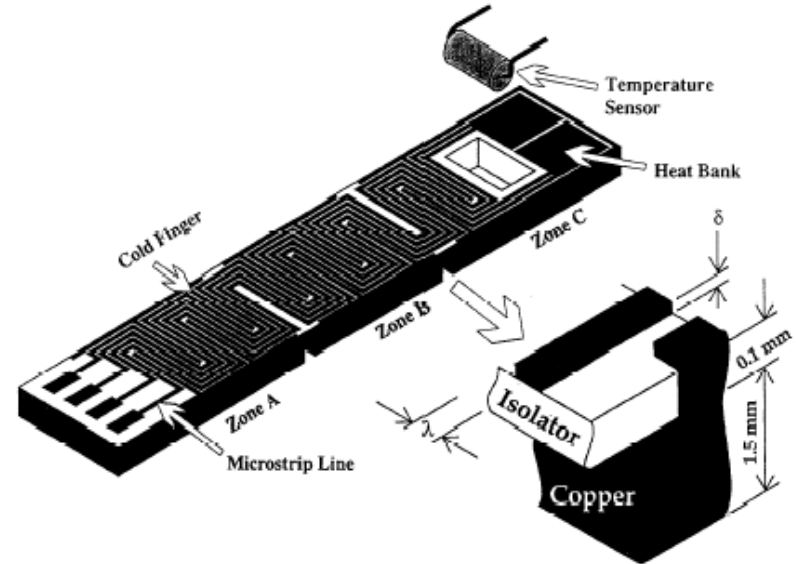
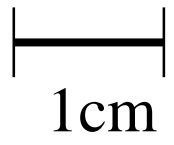
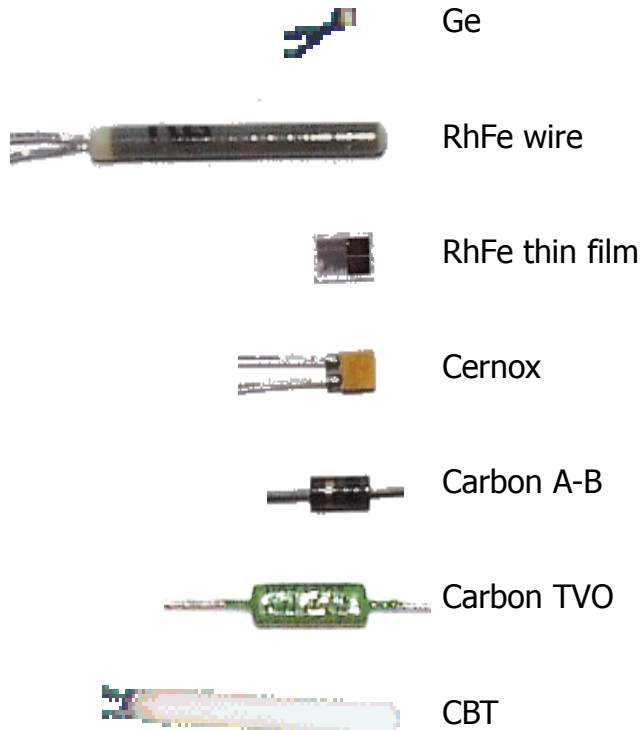


Primary fixed points of ITS90 in cryogenic range

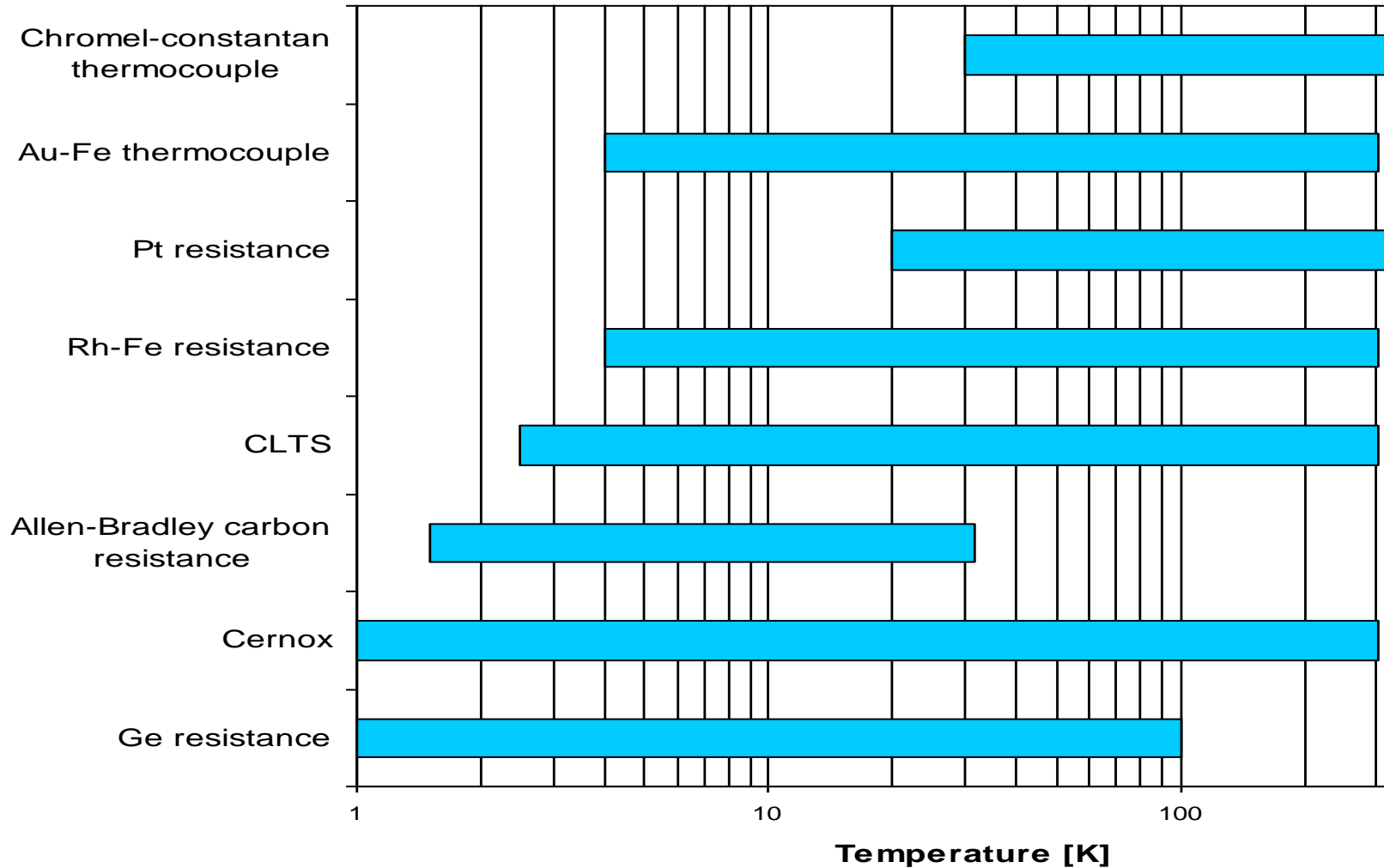
Fixed point	Temperature [K]
H ₂ triple point	13.8033
Ne triple point	24.5561
O ₂ triple point	54.3584
Ar triple point	83.8058
Hg triple point	234.3156
H ₂ O triple point	273.16 (*)

(*) *exact by definition*

From temperature sensor to practical thermometer



Practical temperature range covered by cryogenic thermometers



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