



**THE EUROPEAN SPACE AGENCY**

# **Irradiation characterization of EEE components for space application**

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ESA-ESTEC*

**RADSAGA Initial Training Event  
5 October 2017, Geneva**



RADECS 2017 - CERN, Geneva



European Space Agency  
[www.esa.int](http://www.esa.int)

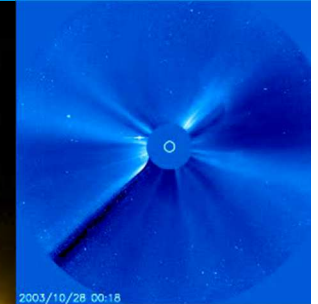
RADSAGA Initial Training Event

## Cosmic rays

- Ions (up to 300MeV/n)

## Solar flares

- Protons (1keV - 500 MeV)
- Ions (1 – 10 MeV/n)



SOHO – Nov 2003

PROBA2/SWAP 17.4nm 2017-08-21 17:08:03

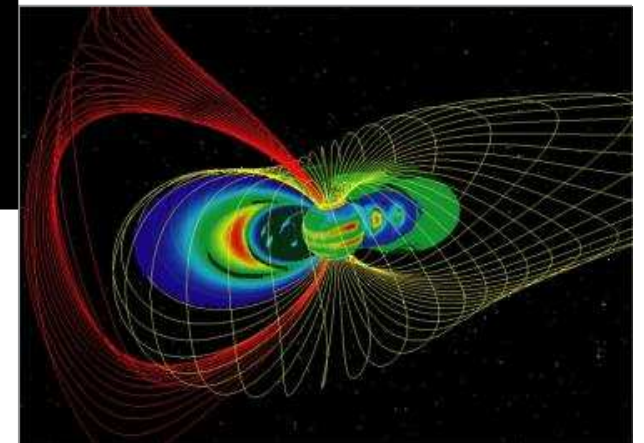


**Trapped in the earth magnetic field:**

- **Protons < 400MeV**
- **Electrons < 7 MeV**

[www.esa.int](http://www.esa.int)

**Van Allen belts: the trapped radiation environment**





## The effects of radiation on electronic devices and materials depend on :

- *Type of radiation (photon, electron, proton...)*
- *Rate of interaction*
- *Type of material (Silicon, GaAs..)*
- *Component characteristics (process, structure, etc.)*

## Interaction mechanisms :

- *Ionization*
- *Internal Charging*
- *Displacement Damage*

## Consequences :

- *Performance Degradation*
- *Transient errors/malfunctions*
- *Permanent Damages*

PROBA2/SWAP 17.4nm 2017-08-21 17:08:03

Radiation hardness assurance methodology is followed to ensure that the radiation environment does not compromise the functionality and performances of electronics during the system life.

**ECSS European Cooperation for Space Standardization**  
**ESCC European Space Components Coordination** ([escies.org](http://escies.org))

**RHA:** [ECSS-Q-ST-60-15C](#), [ECSS-E-ST-10-12C](#), [ECSS-E-HB-10-12](#), [ECSS-Q-HB-60-02A](#)

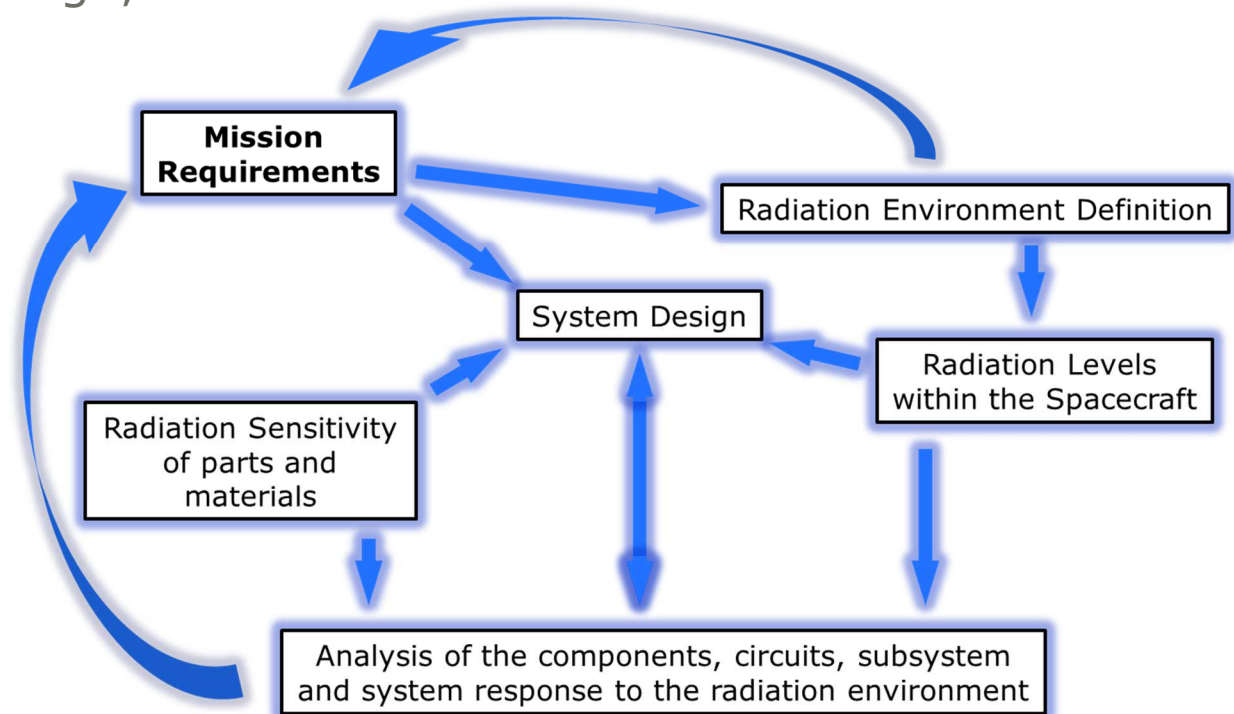
**TID :** [ESCC 22900](#)

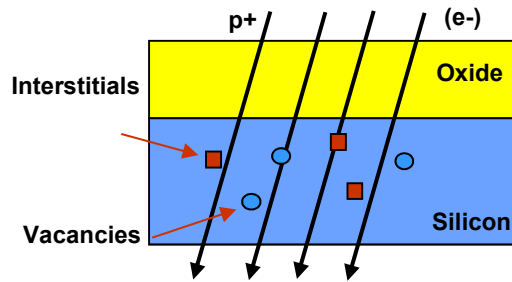
**SEE:** [ESCC 25100](#)

<http://ecss.nl/>

Radiation Hardness Assurance (RHA) deals with:

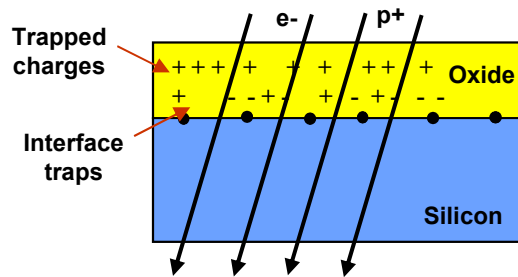
- system requirements,
- space environment definition,
- radiation tolerant design,
- part selection,
- **part testing**,
- shielding
- etc...





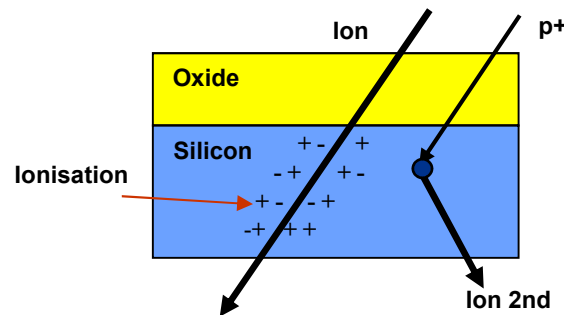
## Displacement Damage (DD)

Cumulative effect result of particle energy deposition in the bulk of the semiconductor. It is also referred to as Total Non-ionizing Dose (TNID) damage



## Total Ionising Dose (TID)

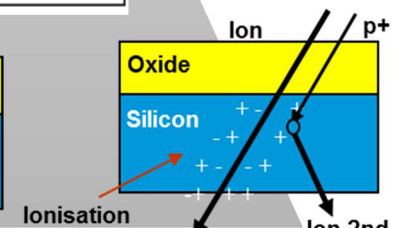
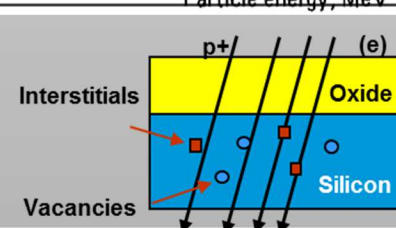
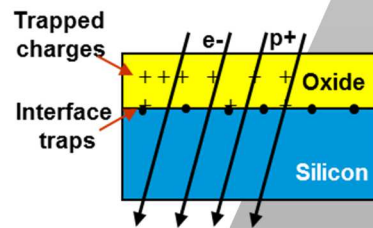
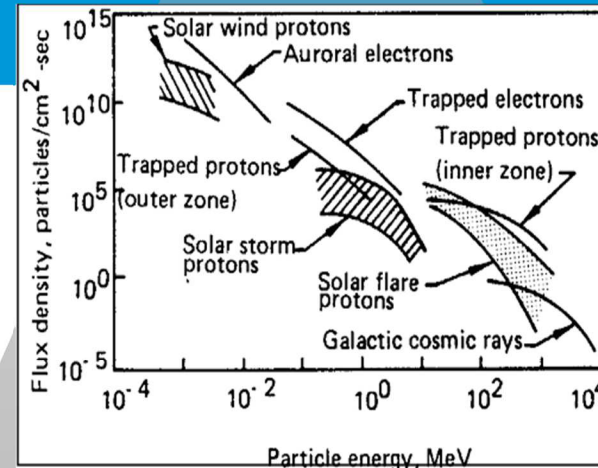
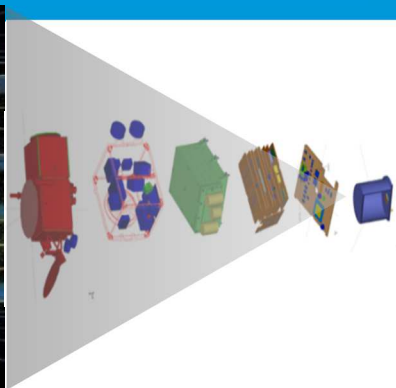
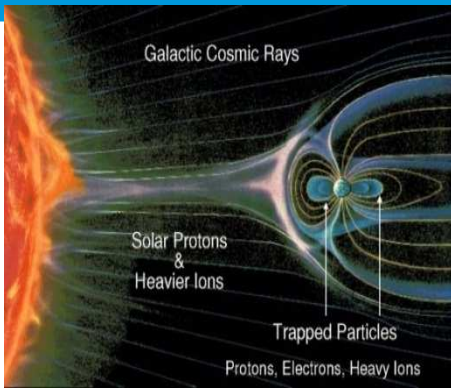
The creation of electron-holes pair in the material cause long term effects in the oxide (charge trapping) and at oxide-semiconductor interface (interface states)



## Single Event Effects (SEE)

Ionization is created by direct energy deposition along ion tracks or the energy is deposited by fragments of nuclei from inelastic collision between a proton and a lattice nucleus. The fragments usually deposit in turn all their energy along their track or produce a cascade of secondary particles.

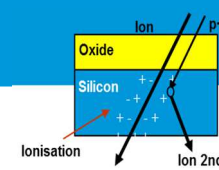
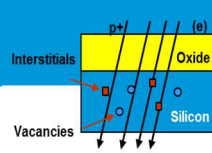
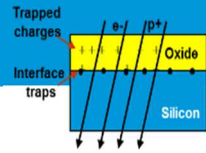
# From space environment to EEE components testing



Radiation/Matter interaction	Ionising process	Non ionizing process	Charge deposition
Radiation effects on semiconductors: 3 Mechanisms Model	Total Ionising Dose	Displacement Damage	Single Event Effects
Standard source to create these effects in LABORATORY	High energy <b>Photons</b>	High energy <b>protons</b>	High energy <b>ions</b>



# What to expect from testing



## Total Ionising Dose

## Displacement Damage

## Single Event Effects

### Cumulative effect:

Gradual global degradation of device parameter

### Probabilistic effect:

transient, permanent or static errors.

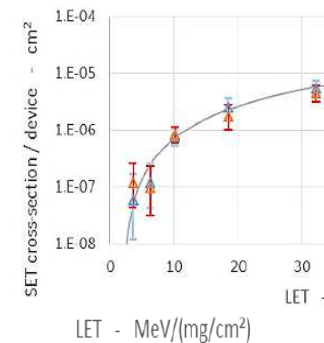
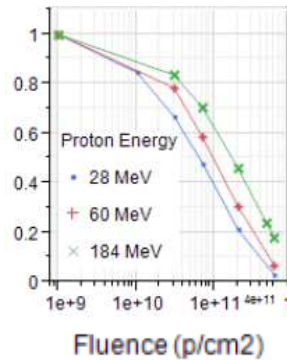
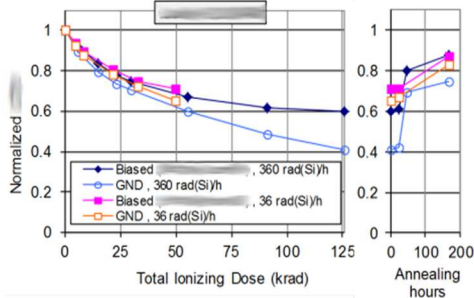
PERFORMANCE DEGRADATION  
as function of :

CROSS SECTION  
(=Probability of SEE occurrence)  
as function of:

Total dose [rad(Si)] [Gy(Si)]

Fluence [p/cm<sup>2</sup>]

Linear energy transfer [MeV/mg/cm<sup>2</sup>]



Other parameters:  
Dose rate [rad(Si)/s] [Gy(Si)]

Other irradiation test parameter:  
Proton Energy [MeV],  
Flux [p/cm<sup>2</sup>/s]

Other irradiation test parameter:  
Flux[p/cm<sup>2</sup>/s] , particle range [um]

# TID testing

# ESA $^{60}\text{Co}$ facility for TID testing at ESTEC



ESA Co60 Facility  
European Space and Technology  
Research Center (ESTEC)  
The Netherlands

Info on ESCIES :

<https://escies.org/webdocument/showArticle?id=227&groupid=6>

Beamtime reservation and current schedule:

<https://estecco60facility.esa.int/>

E-mail

[Co60admin@esa.int](mailto:Co60admin@esa.int)

ESA's Space Research and Technology Centre  
Noordwijk – The Netherlands

The **ESTEC GAMMABEAM 150C Co-60** facility was first installed in **1988** and has been in use ever since.

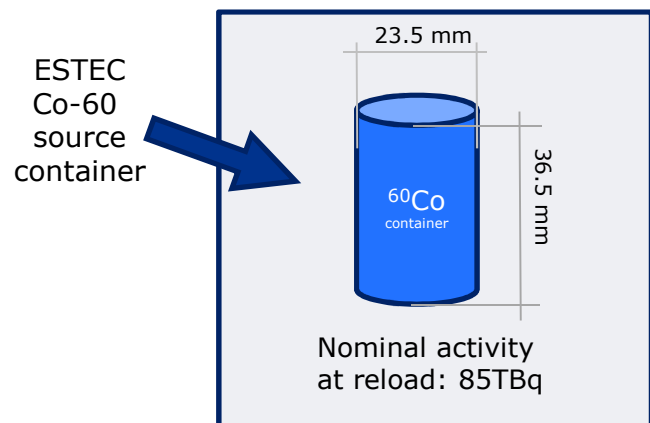
The source housing has **collimated layout and allows DUTs to be exposed to different dose rates** by varying their distance over a range up to 8 meters away from the source.

The initial activity at each reload is  $\sim 85$  TBq with a half life of 5.3 years, the source has been re-loaded several times. The last reload was in May 2016



European Space Agency

# The $^{60}\text{Co}$ facility – source characteristics

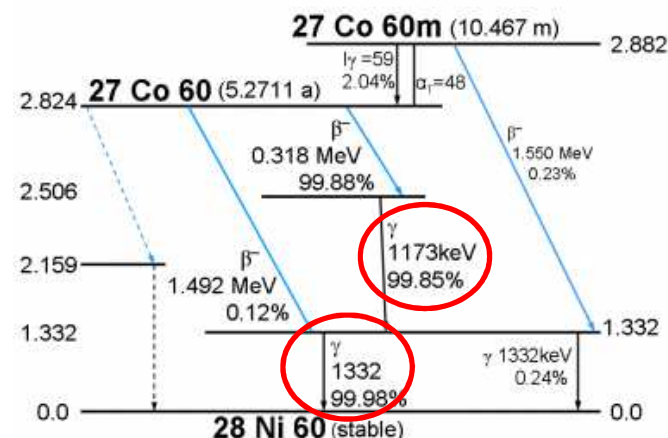
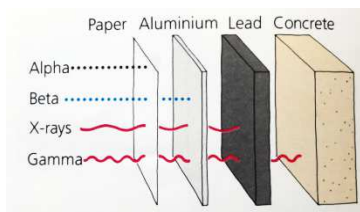


The  $^{60}\text{Co}$  isotope undergoes beta decay with a half-life of 5.272 years producing the nuclear excited state of  $^{60}\text{Ni}$  from which it emits either one or two gamma ray photons to reach the Nickel ground state.

- $^{60}\text{Co}$  Decay products => photons 1.173 MeV, 1.332 MeV
- $^{60}\text{Co}$  Half life = 5.27 years

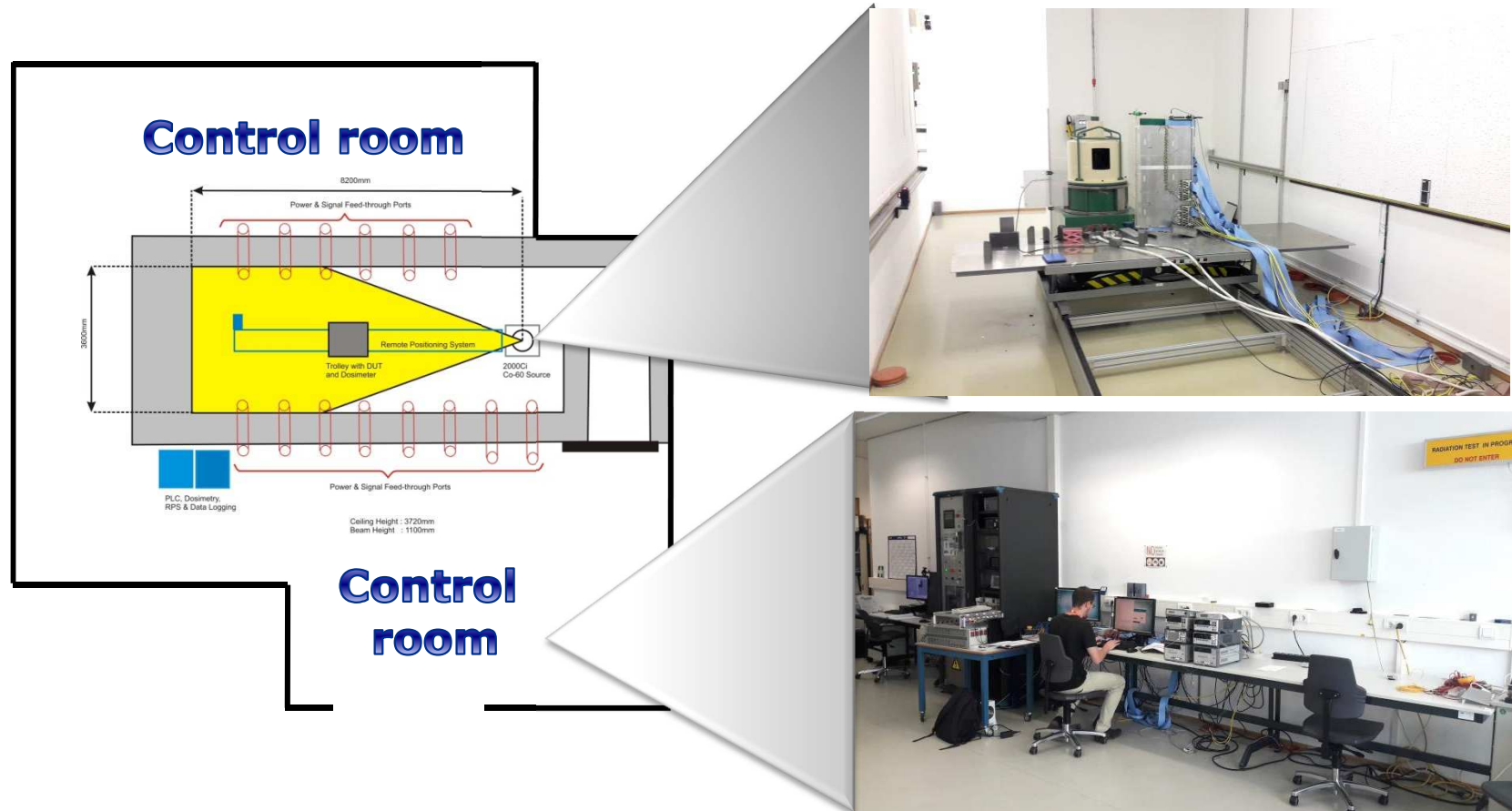
Advantages of using gammas:

- **No material activation.**
- Irradiation in **air**
- **No de-lidding necessary**

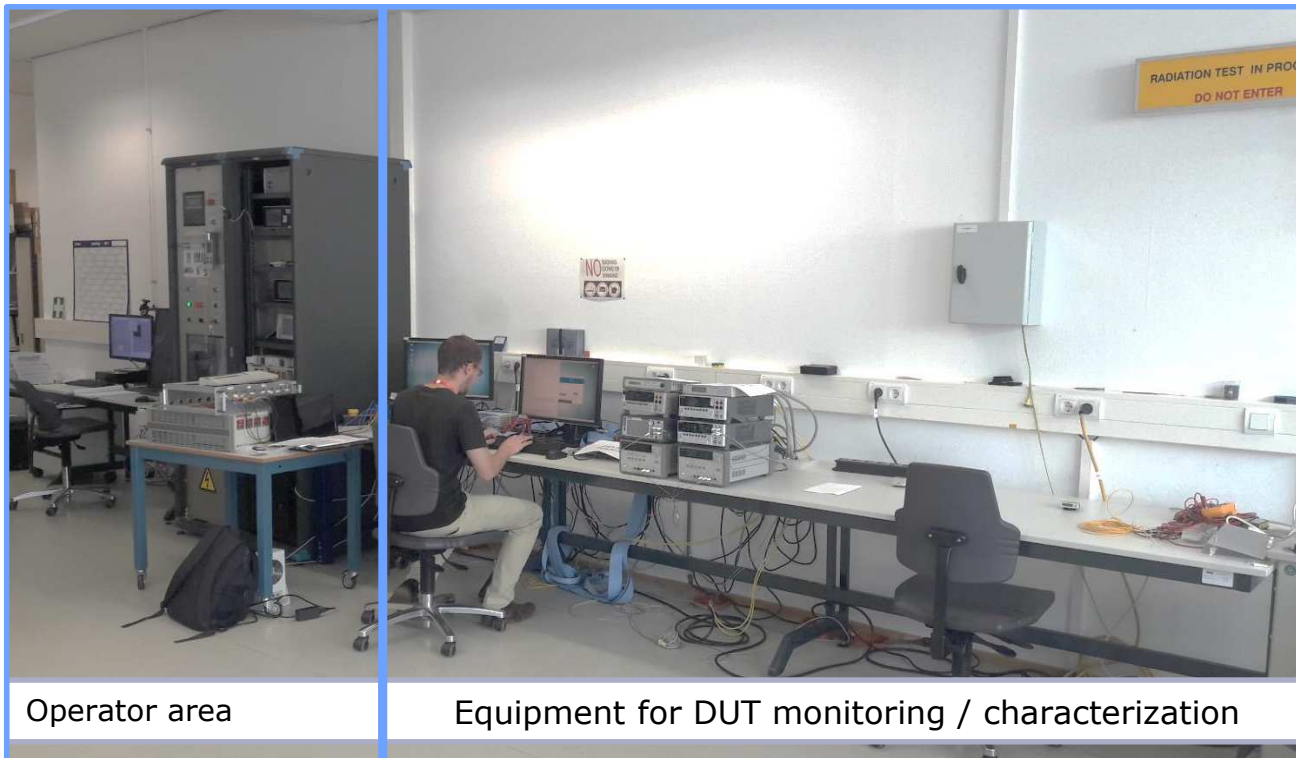




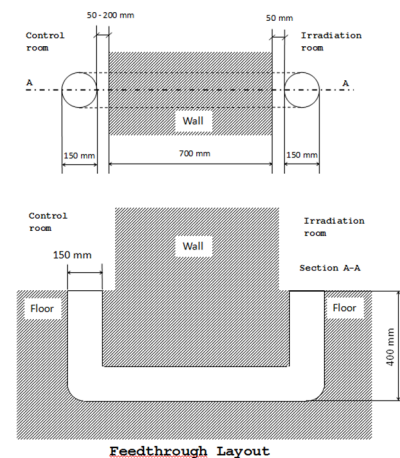
# The ESA-ESTEC $^{60}\text{Co}$ facility



# $^{60}\text{Co}$ facility – The control room



## Feedthrough layout:

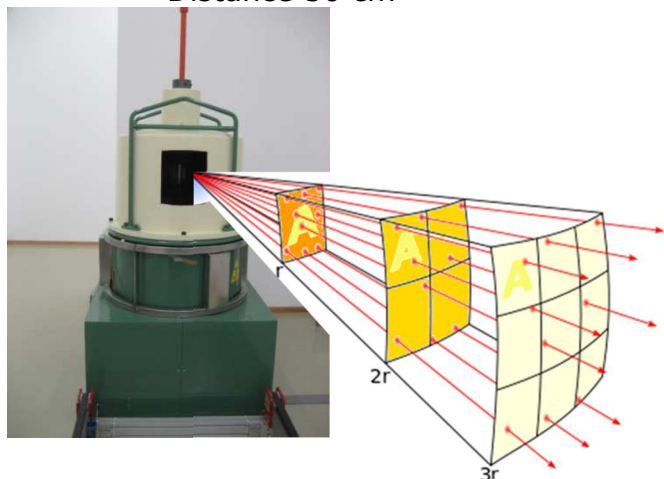


Recommended  
~ **6m** cable length  
to connect DUT in the  
irradiation room to  
equipment in the control  
room

# <sup>60</sup>Co facility – The irradiation room



**Min distance Max dose rate**  
 Irradiation area 20cm x 20 cm  
 Dose rate 10-5krad (Si)/h  
 Distance 50 cm



**Source Irradiator:**  
 Lead container with an opening window to expose the source and collimate the beam

**Max distance Min dose rate**  
 Irradiation area 3.6 m x 3.6 m  
 Dose rate 40-20 rad (Si)/h  
 Distance 7.5m

**Dosimetry based on:**

- Farmer type 0.6 cc ionization chamber



**Table for DUT's installation**  
 Variable height, remote controlled distance

# $^{60}\text{Co}$ facility – Live footage





# $^{60}\text{Co}$ facility – how to request beam time



1) Compile the on-line beamtime request form  
<https://estecco60facility.esa.int>

2) Send a **Test plan** for review and approval

The screenshot displays a web-based form for requesting beamtime at the ESTEC Co60 Facility. The form is divided into several sections:

- Contact Details of Test requester:** Includes fields for Company Name, Name, Surname, Phone number, e-mail, Technical officer or contact person in ESA, Cost Code, and Project. A note states: "The ESTEC Co60 Facility is available only to com projects and activities. Please try to complete all the fields. Once the t will be contacted by the facility administrator v please contact [Co60admin@esa.int](mailto:Co60admin@esa.int)."
- Parts and setup description:** Contains "Devices under test" and "Info on DUTs" (Part type, Description, Manufacturer, Date code). It also includes "Setup information" with "Overall dimensions of the setup to be irradiated:" (Width, height (cm), Other details) and "Special Handling Precautions" (checkboxes for "The DUT or the setup present risks for Health or Safety" and "Special handling is needed (fragile, ESD sensitive..)", each with a "Please specify:" field).
- Irradiation test requirements:** Includes "Irradiation Requirements:" (Indicate the semiconductor material (or material) under test:), "Total dose required on DUT (krad)", "Requested Dose Rate (rad/min)", and "Scheduling:" (Requested Start Date, Requested End Date). It also has checkboxes for "Please indicate if the dose rate" (can be lowered, can be increased, shall be as high as possible, shall be as low as possible) and "Please indicate if the start date" (can be earlier, can be later) and "Please indicate if the end date" (can be earlier, can be later).
- Supervision during the irradiation test phases:** A table with columns for "Customer on site" and "Action delegated to Co60 Facility personnel".

	Customer on site	Action delegated to Co60 Facility personnel
Setup installation	<input type="checkbox"/>	<input type="checkbox"/>
Intermediate measurements	<input type="checkbox"/>	<input type="checkbox"/>
Setup removal	<input type="checkbox"/>	<input type="checkbox"/>

# Why using a Co60 source for TID testing



TID testing employing  $^{60}\text{Co}$  facilities (gamma source) has, over the years, become the de-facto standard test method.

This is due to the high yield of the Co60 gamma ray beam, long heritage of Co60 use in industry and medical application, availability of the isotope...

- In some cases electrons, protons and x-rays are also used to perform TID testing.
- Care has to be shown when employing other sources than  $^{60}\text{Co}$  (range in the material, dosimetry ...)

**In space high energy photons are not a concern compared to the other sources of radiation. However, it has been empirically shown that  $^{60}\text{Co}$  testing is conservative compared to testing with protons or electrons.**

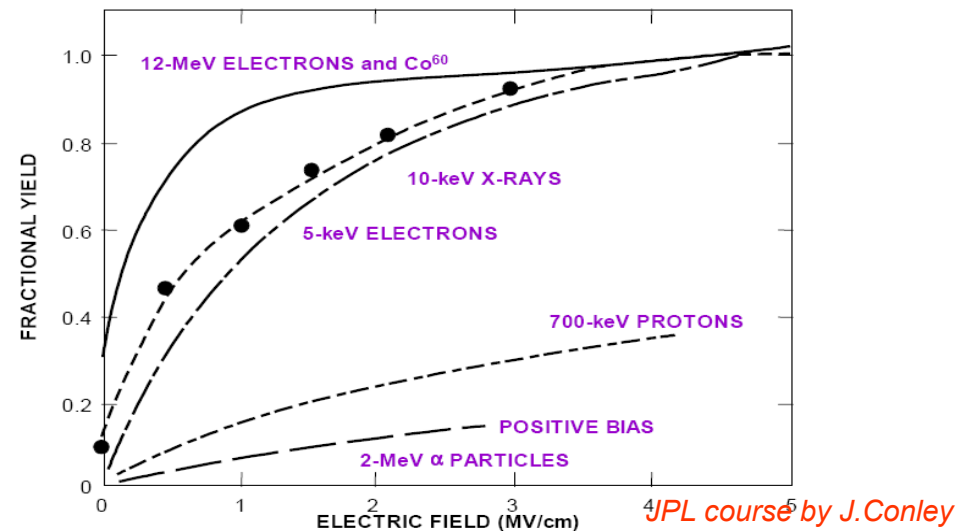


Fig: electron-hole pair generation (ionisation) yield in matter for various particle species, energy and field across oxide.

JPL course by J. Conley

# Total Ionizing Dose (TID) effects

The total dose deposited by electrons or protons alters the electrical characteristics of electronic components

Some effects in MOS transistors:

- Gate voltage threshold shifts
- Leakage increases, flat band shift
- Channel resistivity increase

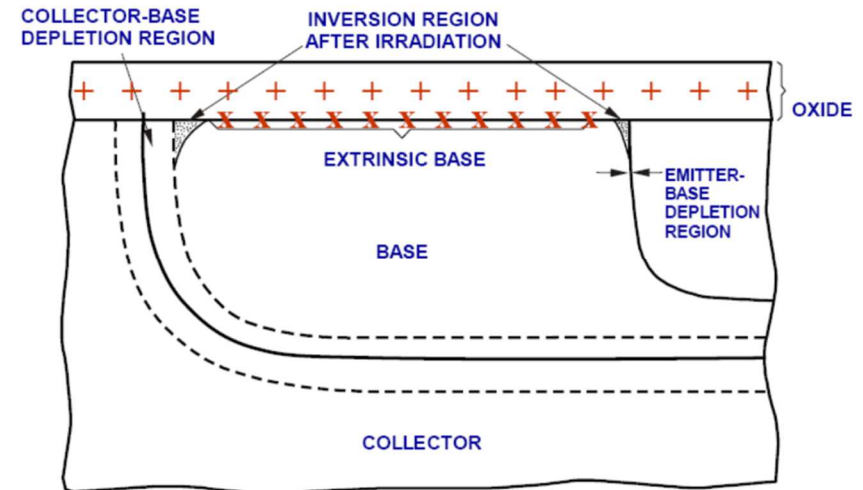
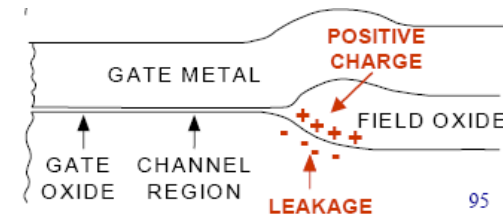
Caused by oxide and interface traps

In Bipolar transistors:

- leakage increase
- Reduction of the gain
- Noise figure worsening

A variety of factors influence the TID effects on the devices:

- Temperature
- Electric fields over oxides (magnitude and polarity)
- Dose rates
- Technology
- Process (including lot to lot variation)
- Design rules
- ...



After Space Radiation Effects on  
Microelectronics course from JPL, J.F. Conley

## ESCC Basic Specification 22900 : *Total Dose Steady-State Irradiation Test Method*

**Specification** -> <https://escies.org/download/specdraftappub?id=3413>

**Test plan template** -> <https://escies.org/download/webDocumentFile?id=62995>

### SCOPE

The specification defines the basic requirements applicable to the **steady-state irradiation testing** of **integrated circuits** and **discrete semiconductors** suitable for space applications.

### PURPOSE

1. The specification addresses three cases:

- The **evaluation** testing procedure: main objectives to establish worst case conditions for TID qualification (dose level, dose rate effects, bias dependency, critical parameters, annealing effects, etc..)
- the **qualification and lot acceptance** testing procedures: test conditions identified in the evaluation phase
- the testing **outside ESCC context**: no hard requirements in terms of sample size, dose rates, pass/fail criteria (can be project and/or application specific not just related to the datasheet).

1.ESCC22900

Other standards and references:

- MIL-STD883 Method 1019.9 "Ionizing Radiation (Total Dose) Test Procedure"
- MIL-STD750 Method 1019.5 "Steady-state Total Dose Irradiation Procedure"
- ASTM F 1892-06 "Standard Guide for Ionizing Radiation (Total Dose) Effects Testing of Semiconductor"



# TID Test: dose rate ranges comparison



## Dose rate (DR = dD/dt)

the quantity of radiation absorbed per unit time

## Absorbed dose (D)

the **mean energy imparted to matter** per unit mass by ionizing radiation

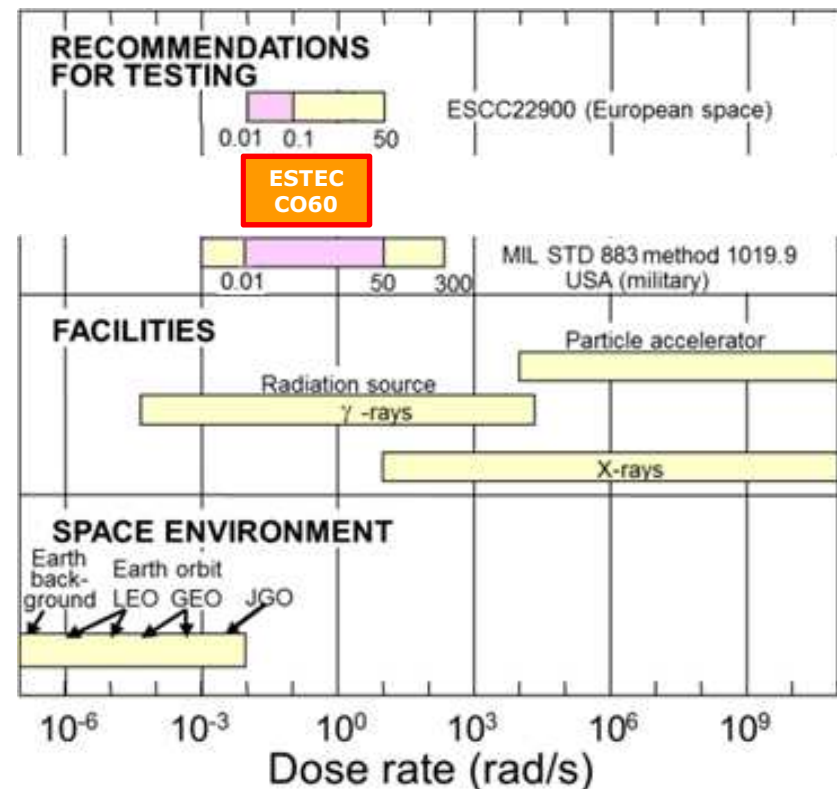
The SI unit of measure is **gray [Gy]**

1Gy equals 1 joules per kilogram

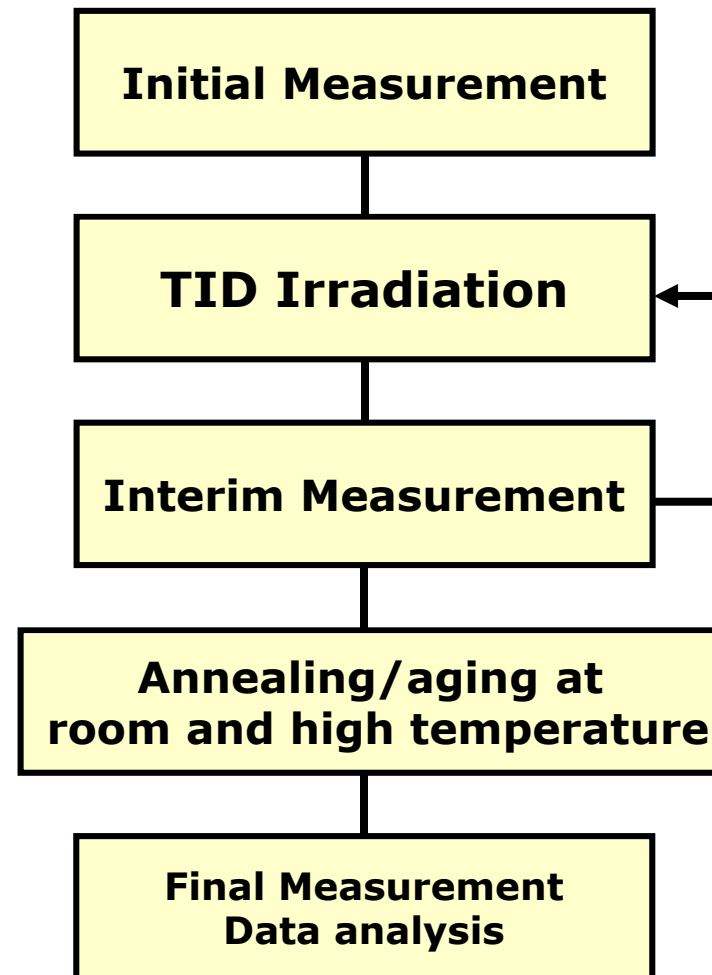
The historical, and most used unit is **[rad]**

**1 rad =100 Gy**

The dose depends on the matter considered => specify dose in matter ( eg. rad(Si), Gy(SiO2)...) )



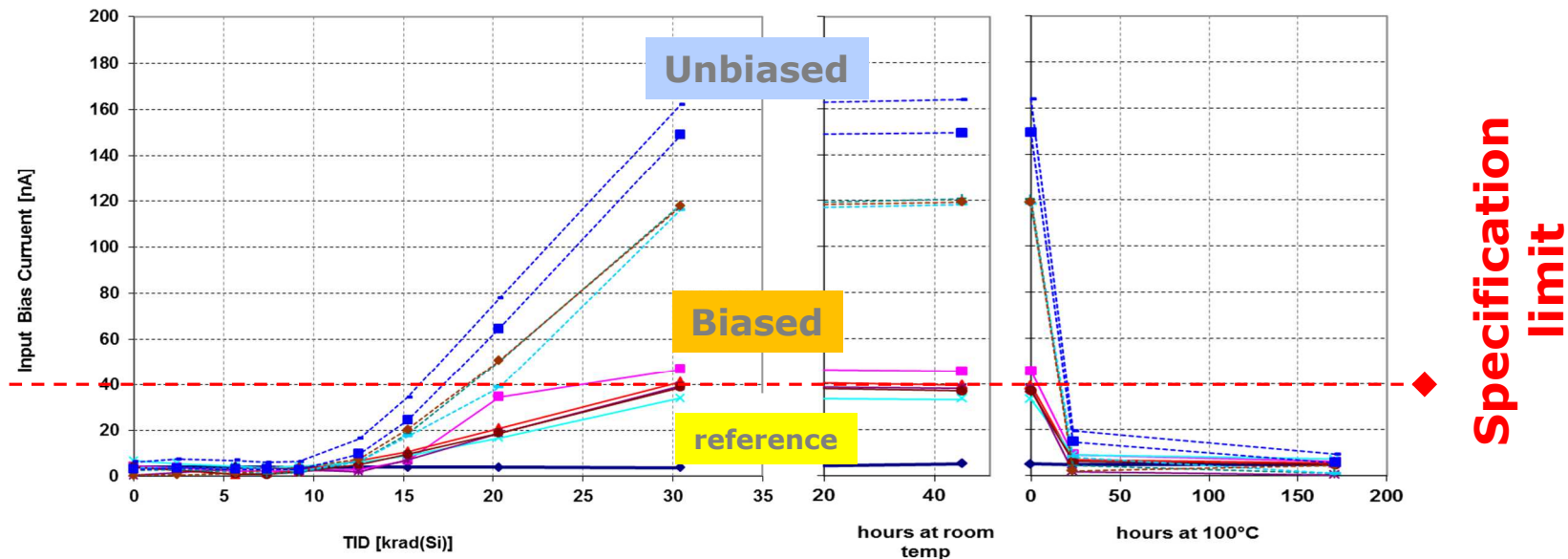
Irradiation testing is performed to identify whether a component meets specification/application requirements when exposed to desired/application radiation doses.



# TID TEST RESULTS

# TID effects - TEST CASE 1

Device under test: Low noise precision Operational Amplifier  
**100krad rad hard qualified has failed at 20krad(Si)**



Why so far from qualification data?

- a. dose rate – *this test* :  $\sim 10\text{mrad(Si)} \text{ s}^{-1}$  *qualification*:  $50\text{-}300\text{rad(Si)} \text{ s}^{-1}$
- b. biasing condition – qualification requests only one biasing condition ( $V=+-15$ )

*In this case the worst case is unbiased*

*Unbiased case is important for cold redundancy and off period during transfer orbit*

# Ionizing Radiation induced charge trapping and interface state generation in MOS Structure



Applicable limits: 2'000    4'500    [mV]

Detailed results - Measurement data in [mV]

s/n	0	6'197	15'000	22'500	30'062	55'000	[70'067	110'500]	Annealing 6hrs RT	Annealing 21hrs RT	Annealing 139hrs RT	Ageing 168hrs 100°C	Applied Bias Condition
001	3,770.31	3,556.56	3,313.38	3,117.63	2,937.19	2,486.56	2,269.63	<b>1,860.38</b>	<b>1,890.06</b>	<b>1,890.50</b>	<b>1,879.75</b>	2,904.63	(V <sub>DS</sub> 0V, V <sub>GS</sub> 0V)
002	3,809.38	3,605.44	3,362.94	3,174.31	2,999.63	2,549.06	2,334.88	<b>1,913.00</b>	<b>1,943.94</b>	<b>1,932.63</b>	<b>1,923.31</b>	2,944.75	(V <sub>DS</sub> 0V, V <sub>GS</sub> 0V)
003	3,808.81	3,601.00	3,359.25	3,157.69	2,989.63	2,537.06	2,302.69	<b>1,899.00</b>	<b>1,930.06</b>	<b>1,916.88</b>	<b>1,910.56</b>	2,937.63	(V <sub>DS</sub> 0V, V <sub>GS</sub> 0V)
004	3,858.31	3,650.63	3,414.50	3,214.06	3,042.19	2,589.56	2,362.75	<b>1,938.31</b>	<b>1,962.75</b>	<b>1,958.38</b>	<b>1,946.50</b>	3,021.31	(V <sub>DS</sub> 0V, V <sub>GS</sub> 0V)
005	3,778.06	3,569.81	3,342.50	3,133.88	2,968.88	2,520.69	2,297.75	<b>1,885.06</b>	<b>1,910.56</b>	<b>1,907.69</b>	<b>1,888.25</b>	2,934.75	(V <sub>DS</sub> 0V, V <sub>GS</sub> 0V)
006	3,800.25	3,704.38	3,607.56	3,495.06	3,400.69	3,141.63	2,972.88	2,632.81	2,625.25	2,626.00	2,696.25	3,169.00	(V <sub>DS</sub> +100V, V <sub>GS</sub> -20V)
007	3,708.31	3,595.25	3,500.69	3,373.44	3,273.88	3,016.50	2,828.69	2,475.50	2,464.75	2,489.69	2,550.81	3,025.63	(V <sub>DS</sub> +100V, V <sub>GS</sub> -20V)
008	3,792.56	3,575.56	3,351.88	3,151.94	2,973.19	2,512.69	2,261.06	<b>1,852.38</b>	<b>1,867.50</b>	<b>1,864.94</b>	<b>1,860.88</b>	2,918.00	(V <sub>DS</sub> +80V, V <sub>GS</sub> 0V)
009	3,843.56	3,623.63	3,396.75	3,189.88	3,002.31	2,546.13	2,296.75	<b>1,870.00</b>	<b>1,891.19</b>	<b>1,888.63</b>	<b>1,876.69</b>	2,971.19	(V <sub>DS</sub> +80V, V <sub>GS</sub> 0V)
010	3,739.88	3,527.38	3,298.06	3,095.25	2,913.81	2,460.25	2,200.44	<b>1,792.81</b>	<b>1,810.44</b>	<b>1,817.56</b>	<b>1,806.56</b>	2,904.94	(V <sub>DS</sub> +80V, V <sub>GS</sub> 0V)
011	3,847.31	3,635.69	3,402.44	3,192.00	3,011.13	2,555.94	2,309.69	<b>1,882.31</b>	<b>1,898.94</b>	<b>1,909.94</b>	<b>1,888.63</b>	2,966.56	(V <sub>DS</sub> +80V, V <sub>GS</sub> 0V)
012	3,818.50	3,608.81	3,385.31	3,185.31	3,000.44	2,549.94	2,309.75	<b>1,884.13</b>	<b>1,891.00</b>	<b>1,900.44</b>	<b>1,890.25</b>	2,923.25	(V <sub>DS</sub> +80V, V <sub>GS</sub> 0V)
013	3,807.44	3,547.44	3,325.19	3,136.63	2,953.25	2,612.50	2,370.31	<b>1,997.63</b>	<b>1,926.50</b>	2,082.13	2,325.13	<b>4,600.56</b>	(V <sub>DS</sub> 0V, V <sub>GS</sub> +15V)
014	3,772.69	3,510.81	3,288.44	3,108.38	2,925.38	2,576.75	2,345.94	<b>1,968.44</b>	<b>1,923.25</b>	2,027.88	2,296.63	<b>4,521.50</b>	(V <sub>DS</sub> 0V, V <sub>GS</sub> +15V)
015	3,756.69	3,490.94	3,272.19	3,089.19	2,904.38	2,566.00	2,326.00	<b>1,960.38</b>	<b>1,905.69</b>	2,024.44	2,279.44	<b>4,522.81</b>	(V <sub>DS</sub> 0V, V <sub>GS</sub> +15V)
016	3,775.31	3,513.38	3,291.63	3,112.56	2,938.38	2,586.88	2,351.81	<b>1,973.75</b>	<b>1,944.38</b>	2,047.00	2,312.13	<b>4,503.75</b>	(V <sub>DS</sub> 0V, V <sub>GS</sub> +15V)
017	3,847.75	3,579.38	3,358.69	3,169.69	2,990.56	2,616.19	2,397.38	2,014.44	<b>1,984.44</b>	2,092.38	2,374.31	<b>4,610.88</b>	(V <sub>DS</sub> 0V, V <sub>GS</sub> +15V)
018	3,779.06	3,480.13	3,231.19	3,010.56	2,825.63	2,419.56	2,123.31	<b>1,725.81</b>	<b>1,713.63</b>	<b>1,810.75</b>	2,091.56	<b>4,554.88</b>	(V <sub>DS</sub> 0V, V <sub>GS</sub> +12V)
019	3,844.25	3,532.75	3,281.25	3,016.19	2,872.44	2,419.38	2,155.75	<b>1,725.50</b>	<b>1,714.31</b>	<b>1,817.50</b>	2,091.00	<b>4,557.44</b>	(V <sub>DS</sub> 0V, V <sub>GS</sub> +12V)
034	3,817.81	3,796.69	3,823.81	3,800.44	3,789.50	3,811.56	3,800.00	3,821.56	3,811.25	3,818.63	3,809.44	3,833.56	<a href="#">Reference device</a>

Reference device    Mean value: **3,811.19**    Estimated uncertainty: **± 0.29 % ( ± 11.05 mV )**

*Red values: greater than max limit*  
*Dark red Values: lower than min limits*

## REFERENCE SAMPLE

- Setup measurement long term stability over time
- Statistical contribution to uncertainty
- Environmental effects
- Setup functionality

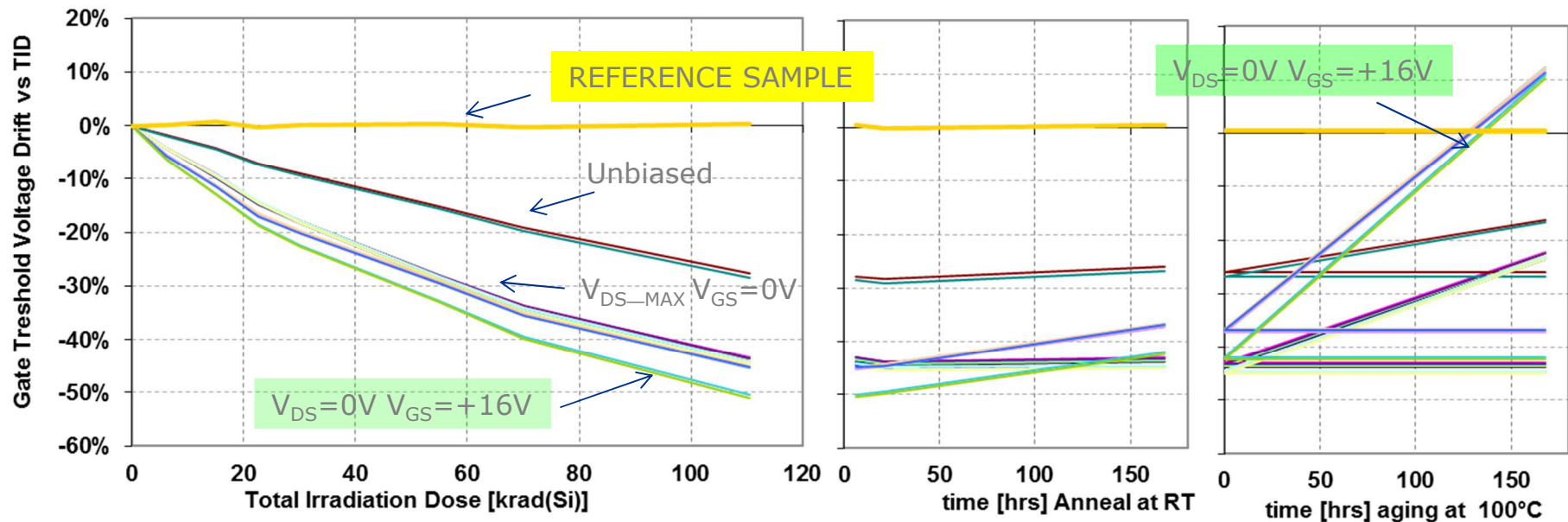


# TID effects – TEST CASE 2



Device under test: Power Mosfet

Dose rate:  $\sim 100 \text{ mrad(Si) s}^{-1}$  worst case for  $V_{DS}=0V$   $V_{GS}=V_{GSmax}$



Irradiation: negative shift due to the oxide traps prevails on the interface traps effect. The worst case for  $V_{GS}$  max

Ageing: positive shift due to the interface traps prevails. The worst case for  $V_{GS}$  max

# TID effects – TEST CASE 2



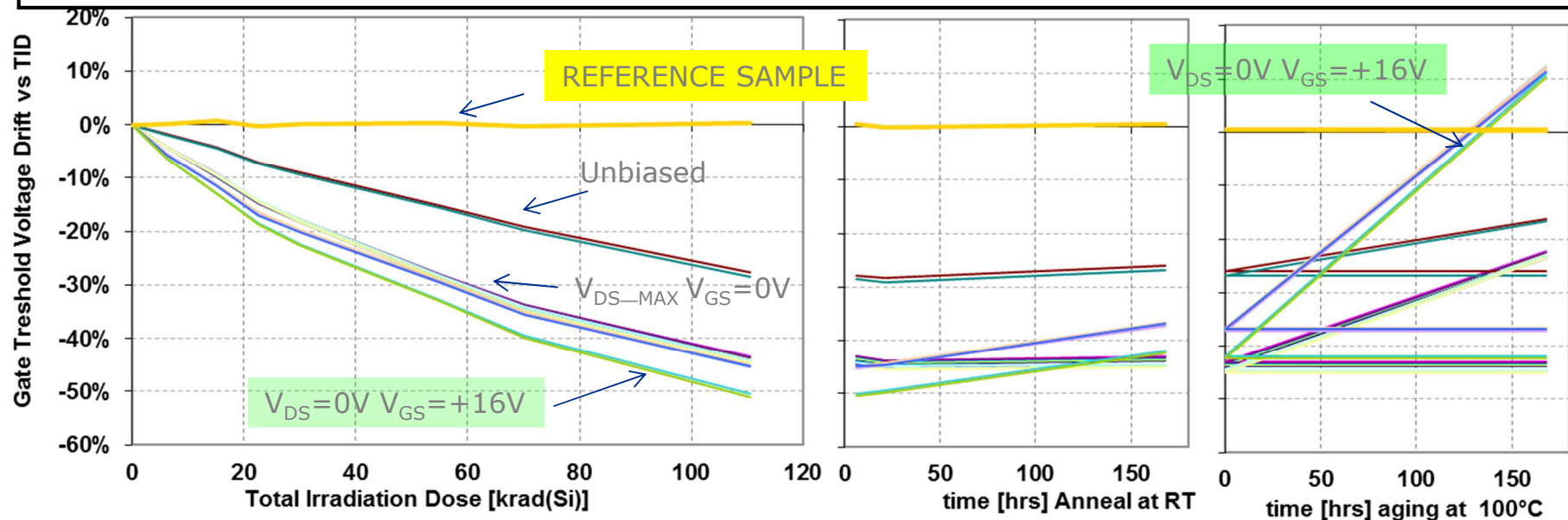
Device under test: Power Mosfet

Test condition : Dose rate:  $\sim 100 \text{ mrad(Si) s}^{-1}$

**worst case** always for  $V_{DS}=0V$   $V_{GS}=+16V$

After the aging: minimum drift ( $\approx 10\%$ ) on the bias condition  $V_{DS}=0V$   $V_{GS}=+16V$

**the “always worst case condition” is actually the better one?**

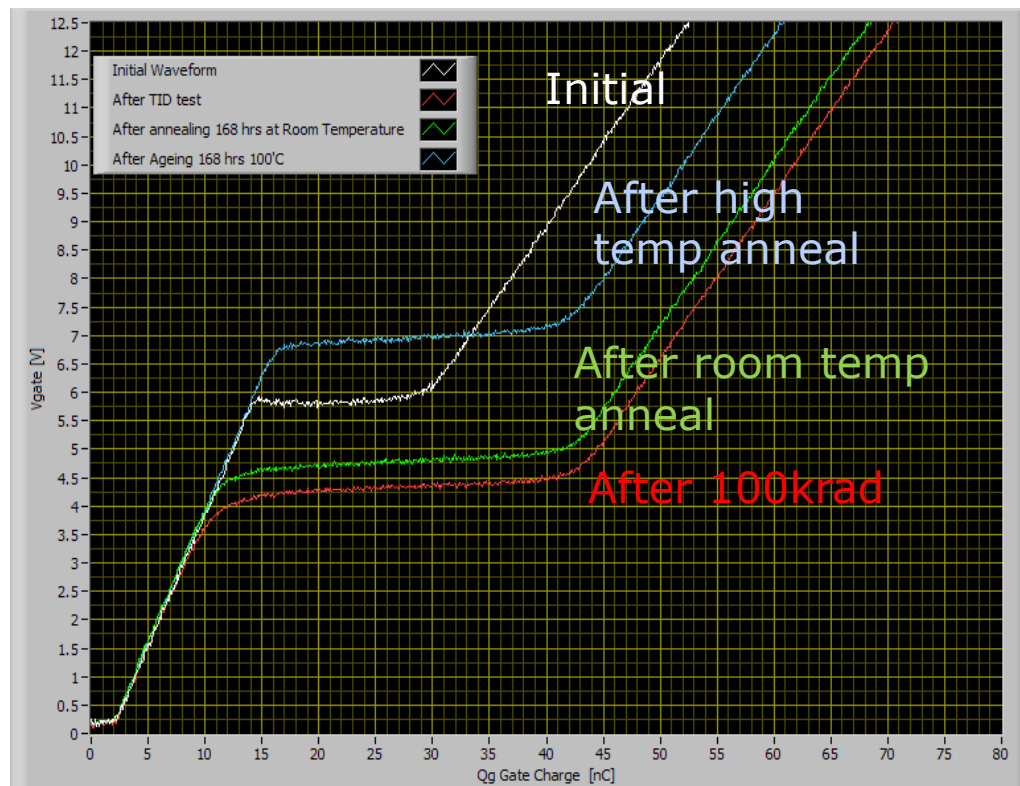


For oxide and interface trap dynamics, see D.M.Fleetwood short course lecture

# TID effects – TEST CASE 3

Device under test: MOSFET

Result: at the end of the test static parameters seems still “OK”



However, the charge needed to **switch on/off** the MOSFET is about 30% higher than the initial value.

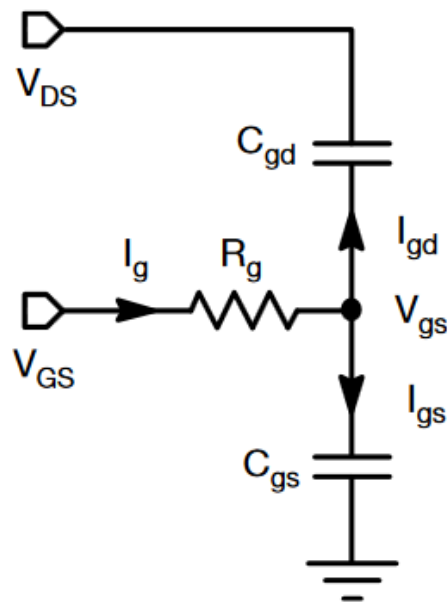
The MOSFET switching characteristics will be significantly degraded!

Ref: MIL STD 750 method 3471

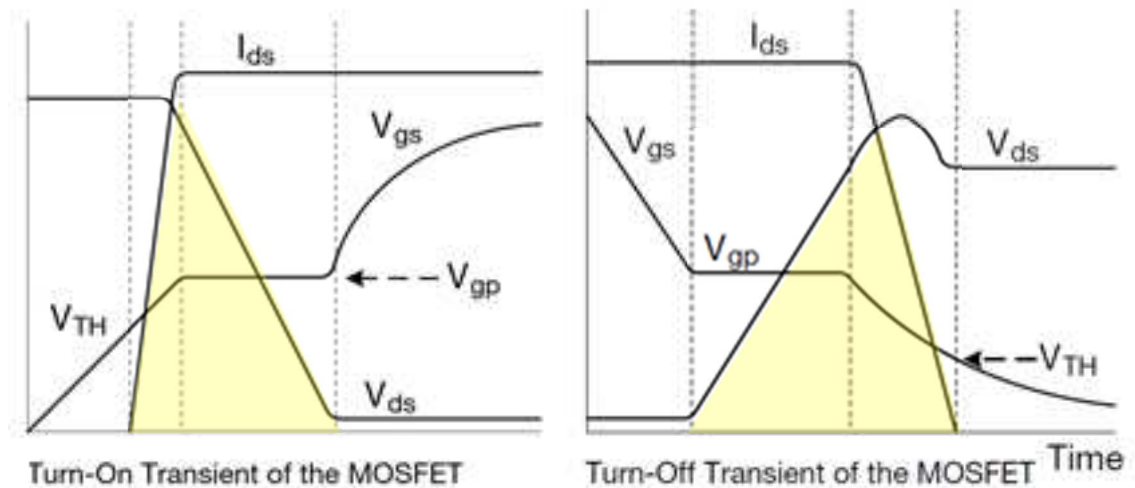
# TID effects – TEST CASE 3

Device under test: MOSFET

Result: at the end of the test, the trapped charges at the interface will still affect the switching characteristic of the MOSFET



([www.vishay.com](http://www.vishay.com) AN608A)



# TID effects – to keep in mind



- TID predominantly affects the device oxide resulting in device parameter degradation
- Co-60 gamma ray most common used radiation source
- Sensitivity to TID is strongly technology and manufacturing process dependent.
- ELDRS can be an issue for bipolar based devices
- Lot-non-uniformity is an issue for TID specifically with respect to Enhanced Low Dose Rate Sensitivity (ELDRS) issue
- New Deep Sub Micron (DSM) technologies **seem** quite TID tolerant...  
... keep testing! (see D.M. Fleetwood, RADECS 2017 short course)



## DD testing

The Displacement Damage is not an oxide effect but is a result of the particle energy deposition in the bulk of the material that creates defects in the semiconductor lattice.

As for TID, DD (or TNID) is a cumulative effect and produces parametric degradation of the device performances.

The ECSS-E-HB-10-12A standards provide detailed guidelines For DD testing of electronic devices.

# Displacement Damage – some effects

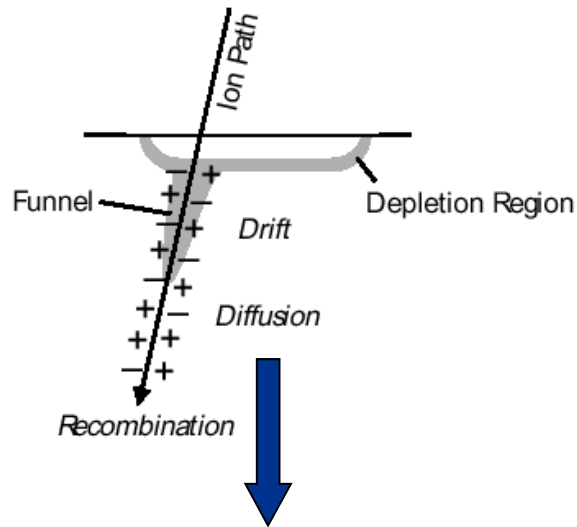


Technology category	sub-category	Effects
General bipolar	BJT	hFE degradation in BJTs, particularly for low-current conditions (PNP devices more sensitive to DD than NPN)
	diodes	Increased leakage current increased forward voltage drop
Electro-optic sensors	CCDs	CTE degradation, Increased dark current, Increased hot spots, Increased bright columns Random telegraph signals
	APS	Increased dark current, Increased hot spots, Random telegraph signals Reduced responsivity
	Photo diodes	Reduced photocurrents Increased dark currents
	Photo transistors	hFE degradation Reduced responsivity Increased dark currents
Light-emitting diodes	LEDs (general)	Reduced light power output
	Laser diodes	Reduced light power output Increased threshold current
Opto-couplers		Reduced current transfer ratio
Solar cells	Silicon GaAs, InP etc	Reduced short-circuit current Reduced open-circuit voltage Reduced maximum power
Optical materials	Alkali halides Silica	Reduced transmission

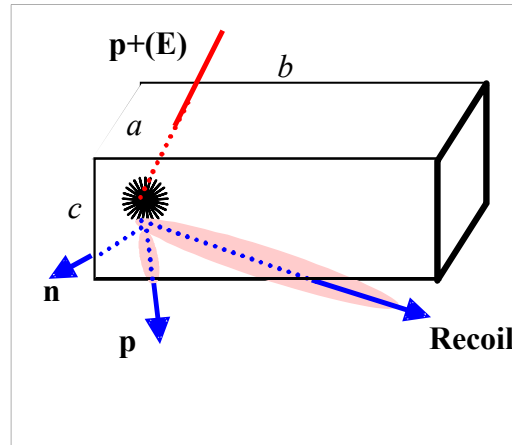
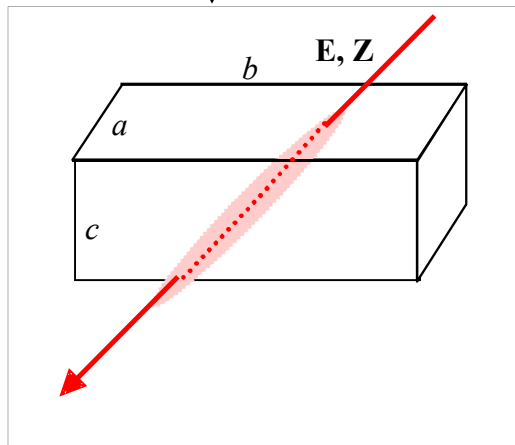
see the ECSS-E-HB-10-12A standard, paragraph 7 for more details

# SEE testing

# Single Event Effect (SEE) basic mechanism



The volume responsible for charge collection for a SEE (in  $\mu\text{m}^3$ )

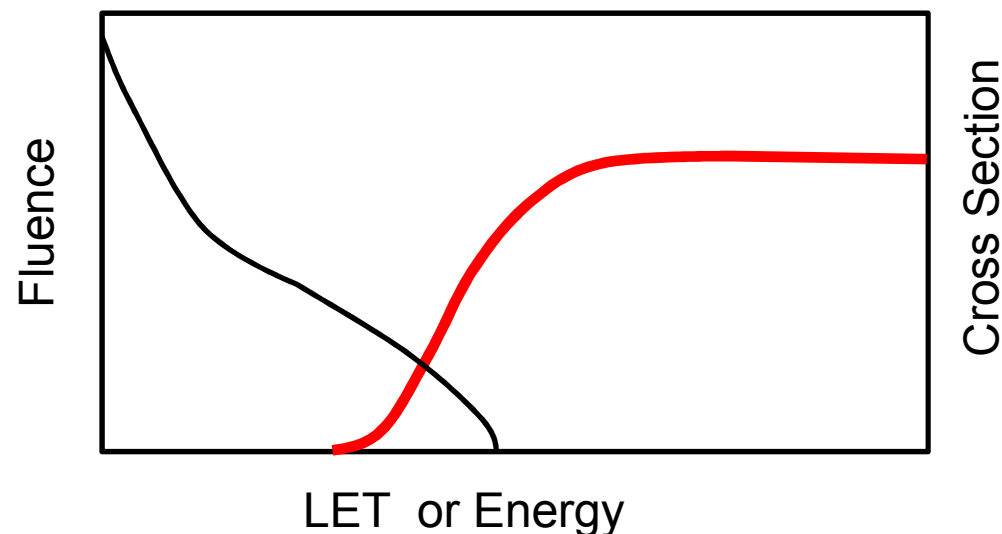


Real sensitive volumes do not really have regular shapes and therefore, not only incidence angle but also pitch and tilt may have an impact in the total amount of charge collected triggering the SEE.

Estimate the sensitivity to SEE in order to predict the SEE rate during the system life in space.

Need to know:

- a. Space Environment: Integral flux as a function of LET or energy
- b. Cross-section vs. ion LET or proton energy**

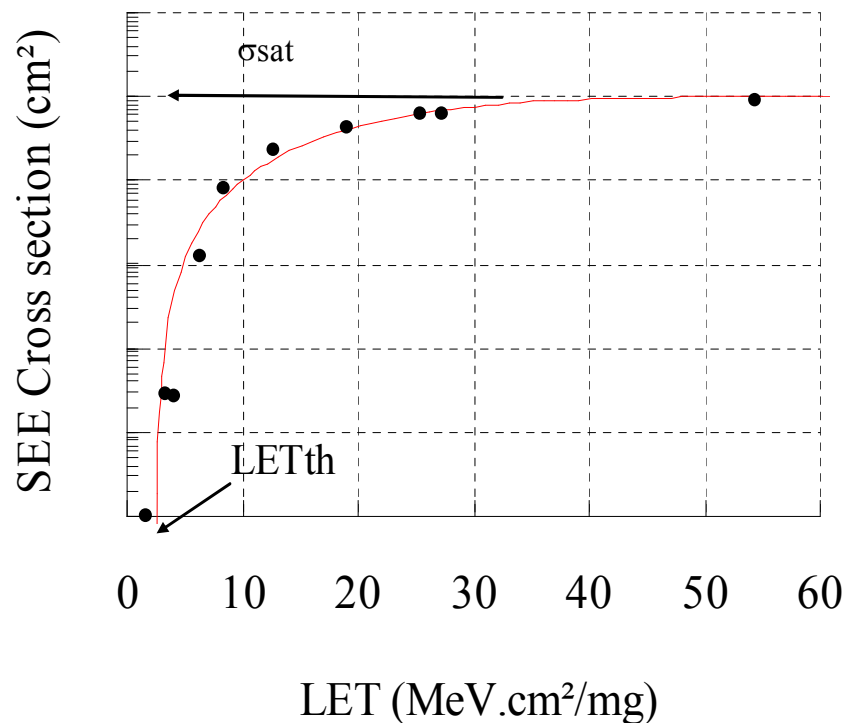




# SEE Cross Section Curve vs LET



## ESCC 25100 Single Event Effects Test Method and Guideline



The SEE cross section measures the probability for a SEE to occur

$$[\text{cm}^2] \rightarrow \sigma = \frac{N_{\text{events}}}{\text{Fluence}} \leftarrow [N_{\text{particles}}/\text{cm}^2]$$

Fit with Weibull (integral form)

$$\sigma = \sigma_{\text{sat}} \left( 1 - \exp\left( -\left( \frac{\text{LET} - \text{LET}_{\text{th}}}{W} \right)^S \right) \right)$$

W and S are fitting parameters

SEE cross-section is a crucial input for in-orbit SEE rate prediction.

## Soft Errors (no permanent damage)

- SEU Single-Event Upset
- MBU, MCU Multiple Bit (or Cell) Upset
- ASET Analog Single Event Transient
- DSET Digital Single Event Transient
- SEFI Single Event Functional Interrupt
- (\*) Others...

***(\*) the one that always occurs  
when you do not expect it !***

## Hard Errors (permanent damage of device)

- SEL Single-Event Latchup (non destructive test still possible...)
- SEDR Single-Event Dielectric Rupture
- SEB Single-Event Burnout
- SEGR Single-Event Gate Rupture
- (\*) Others...

} When possible, identify  
the Safe operating area

# SEE testing – TEST CASE 4

## Safe Operation Area (SEGR - SEB)



	$V_{GS} = 0\text{ V}$	$V_{GS} = 6\text{ V}$	$V_{GS} = 12\text{ V}$	$V_{GS} = 15\text{ V}$
$V_{DS} = 0\text{ V}$	not tested	pass	pass	pass
$V_{DS} = 15\text{ V}$	pass	pass	pass	
$V_{DS} = 22\text{ V}$	pass	pass	(FAIL)	
$V_{DS} = 28\text{ V}$	pass	pass		
$V_{DS} = 30\text{ V}$	not tested	FAIL		
$V_{DS} = 34\text{ V}$	pass			

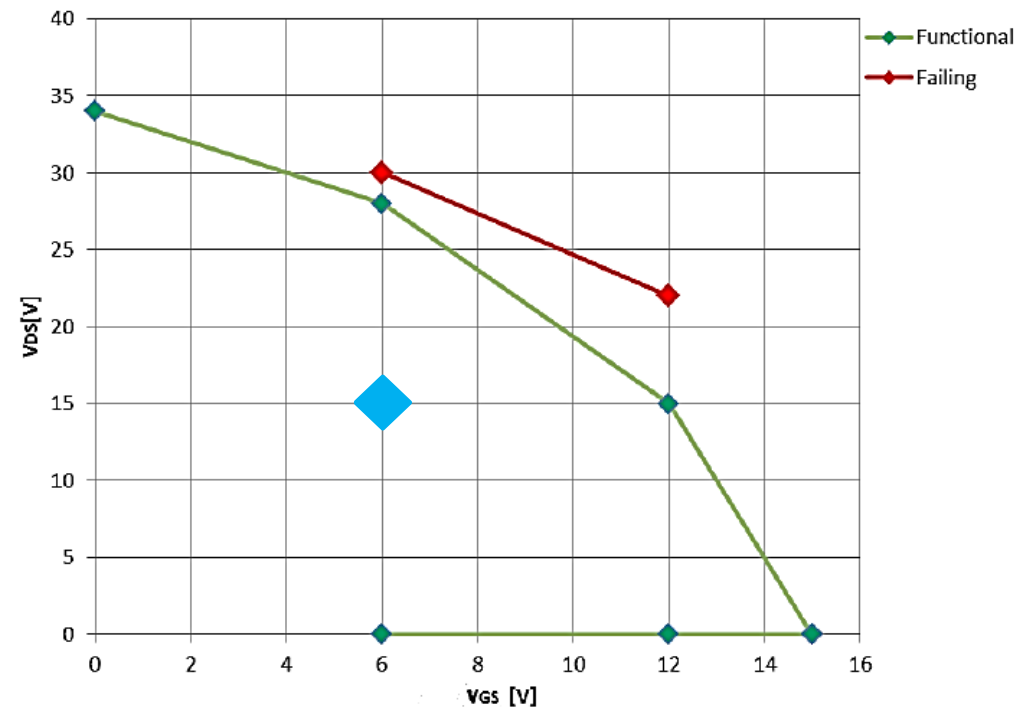
The application requirements are appropriate since they fall within the SOA

Nominal Device characteristics:

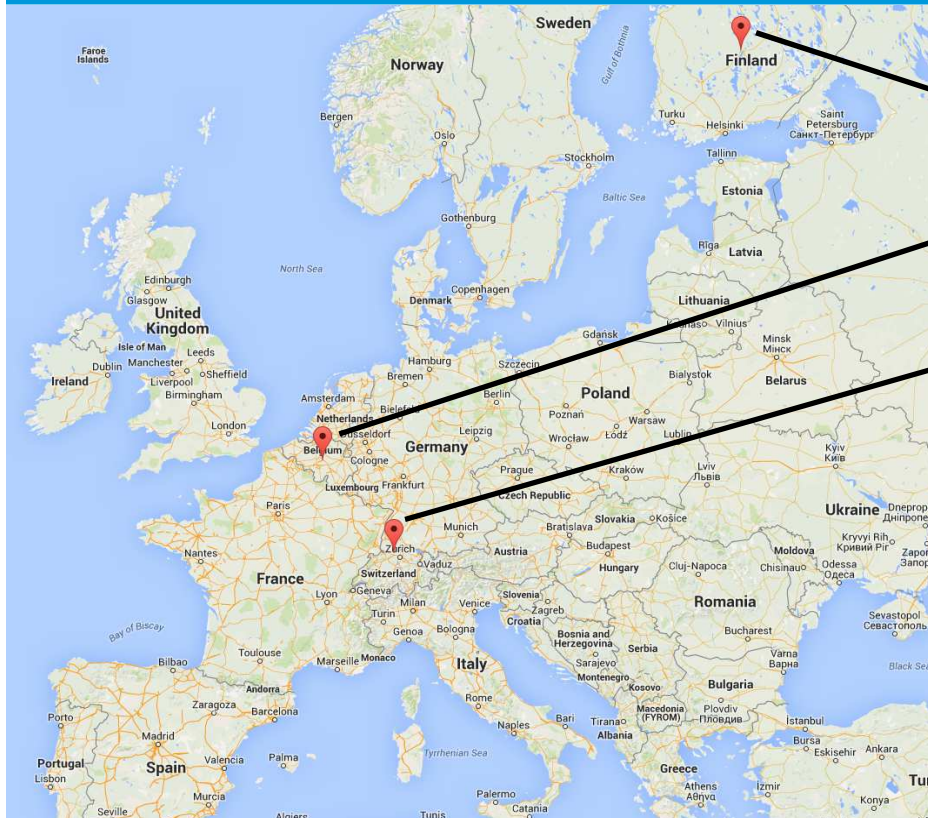
- $V_{DSmax} = +60\text{ V}$
- $V_{GSmax} = \pm 20\text{ V}$

◆ Application requirements:

- $0\text{ V} \leq V_{DS} \leq +15\text{ V}$
- $0\text{ V} \leq V_{GS} \leq +6\text{ V}$



# External irradiation test facilities – supported by



## **RADEF, JYFL**

Jyväskylä,  
Finland

*Heavy ions, protons, electrons*

## **UCL**

Louvain-la-Neuve  
Belgium

*Heavy ions, protons*

## **PSI**

Villigen  
Switzerland

*Protons, electrons*

ESA has been collaborating with the three external test facilities for more than 25-years.

Aiming at continuous improvement of the quality of the beam, dosimetry and testing infrastructure

( Stable flux and energy levels, high particle selectivity, accurate dosimetry, electrical/optical interfaces for cabling...)

The support from the cooperative funding programmes has been crucial for the success of this collaboration.

The network has been supporting most of the space programs worldwide becoming a well established reference for the scientific and industrial community in the field of radiation effects on electronic components.

# External Facilities factsheet



	PSI (Switzerland)	RADEF (Finland)	UCL (Belgium)
Proton Beam	74 - 200 MeV	0.5 - 52 MeV	14 - 62 MeV
Heavy ion beam		7.16 - 69.2 MeV/cm <sup>2</sup> /s	1.3 - 62.5 MeV/cm <sup>2</sup> /s








RADECS 2017 - CERN Geneva

RADSAGA Initial Training Event



# Practical aspects of Testing



	<b>TID test</b>	<b>DD test</b>	<b>SEE test</b>
Source type	Co60 gamma rays Decay of radioactive source	Proton Irradiation Particle accelerator (cyclotron)	Heavy Ions Irradiation Particle accelerator (cyclotron)
Irradiation in	Air	Air	Vacuum
Die need to be exposed?	 No	No	<b>YES</b>
Activation of irradiated DUT?	 No	<b>YES</b>	NO
Test duration	 Days / Weeks	Some hours (8 h)	Some hours (8 h)
Irradiation execution	 Irradiation 24/7 Access to the facility: office hours	Irradiation in shifts <b>(24/7)</b> Typically 8/12h	Irradiation in shifts <b>(24/7)</b> Typically 8/12h
How in advance to book	1 Month	<b>3 Months</b>	<b>3 Months</b>
Beam sharing	Yes (up to 3 tests at different dose rates)	No	No
Beam diameter	 From 20 cm to 3 m	Approx <b>8 cm</b>	Approx <b>3 cm</b>
Cable length from control to DUT	6 m	<b>10-20 m</b>	3m
	€	€€€	€€€

# What we would like to achieve when we do testing ?



... just do our best to have satellites performing “simple tasks” in space



according to the plan... after 12 years from the launch.

# Thank you !



Undergraduate programs  
Stages

Graduate programs  
Young Graduate Training

PhD programs  
Networking Partner Initiative

Staff positions  
Contractor positions

[http://www.esa.int/About\\_Us/Careers\\_at\\_ESA](http://www.esa.int/About_Us/Careers_at_ESA)