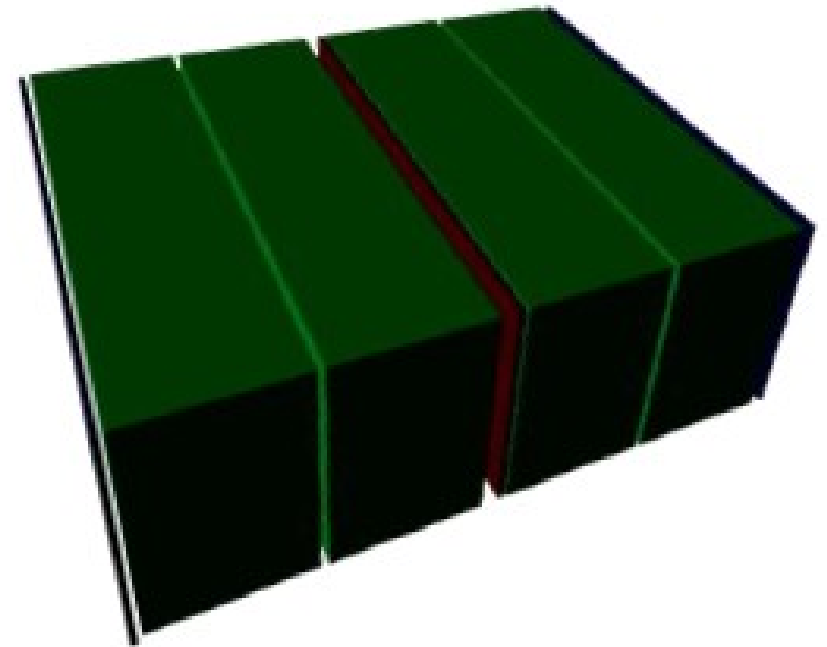


TPC field cage design and prototype - status report

Overview

1. TPC **design concepts**
2. **Prototype** for testing
issues of the building process



IFAE: T.Lux, J.Mundet, C.J.Valls

INFN Bari: L.Magaletti, E.Radicioni

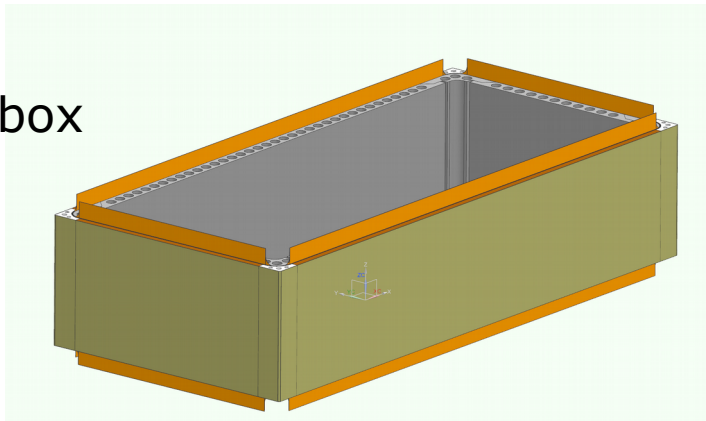
INFN Legnaro Labs: F.Gamegna, T.Marchi

INFN Padova: G.Cogo, G.Collazuol, A.Longhin, A.Pepato, M.Romanato

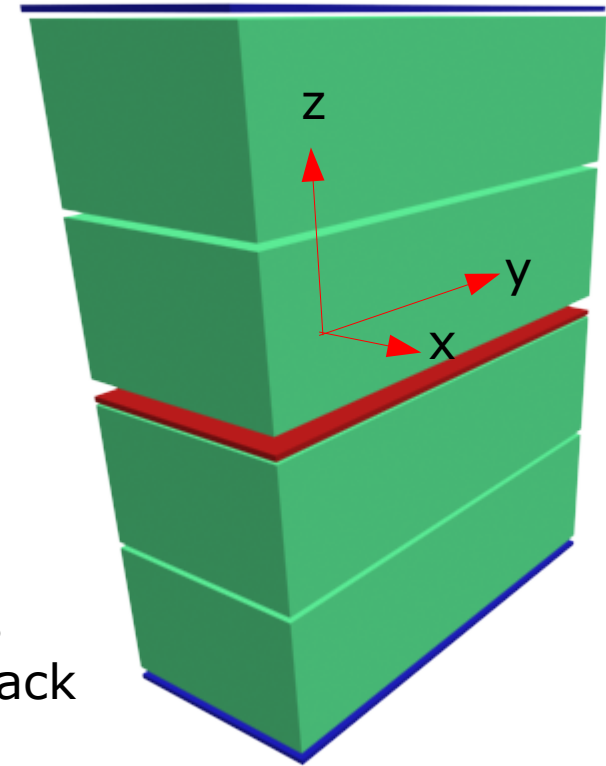
1 - TPC design concepts – Concept and specs

Basic design idea is to build the new thin wall TPCs by **piling up boxes made of composite material walls**

single box



boxes
in a stack



Walls composite structure to define active gas volume and to provide mechanical support to electrodes. Main specs:

- **gas tightness**, for keeping O_2 level below $O(1\text{ppm})$
- **electric insulation** against breakdown with cathode V max of $O(40\text{kV})$
- internal **surfaces (E field shaping strips) flatness** better than $O(100\mu\text{m})$ considering $O(5\text{mbar})$ overpressure (and gravity)
- **alignment of strips plane (x-y) || end-plates || cathode** better than $O(100\mu\text{m})$

1 - TPC design concepts – Box structure

- Rectangular TPC of 200 (B field / drift dir.) x 180 (beam dir.) x 85 (height) cm³
- Aiming at **thin wall** (~3cm) made of **low Z composite material** (few % X₀)

Carbon Fiber based layer stack

Material	Thickss (mm)
<i>Copper coated polyimide film</i>	~ 0.15
<i>Carbon Fiber</i>	1.20
<i>Aramide HoneyComb panel</i>	15.00
<i>Carbon Fiber</i>	1.20
<i>Aramide HoneyComb panel (insulation)</i>	10.00
<i>Polymide film (insulation)</i>	~ 0.10
<i>Strips (double later) on Kapton foil</i>	~ 0.15
TOTAL RADIATION LENGHT ~ 2% X₀	~27.5

Aramide Fiber fabric based layer stack

Material	Thickss (mm)
<i>Copper coated polyimide film</i>	~ 0.15
<i>Aramid Fiber Fabric (Kevlar)</i>	4.00
<i>Aramide HoneyComb panel</i>	15.00
<i>Aramid Fiber Fabric (Kevlar)</i>	4.00
<i>Polymide film (insulation)</i>	~ 0.10
<i>Strips (double later) on Kapton foil</i>	~ 0.15
TOTAL RADIATION LENGHT ~ 4% X₀	~23.5

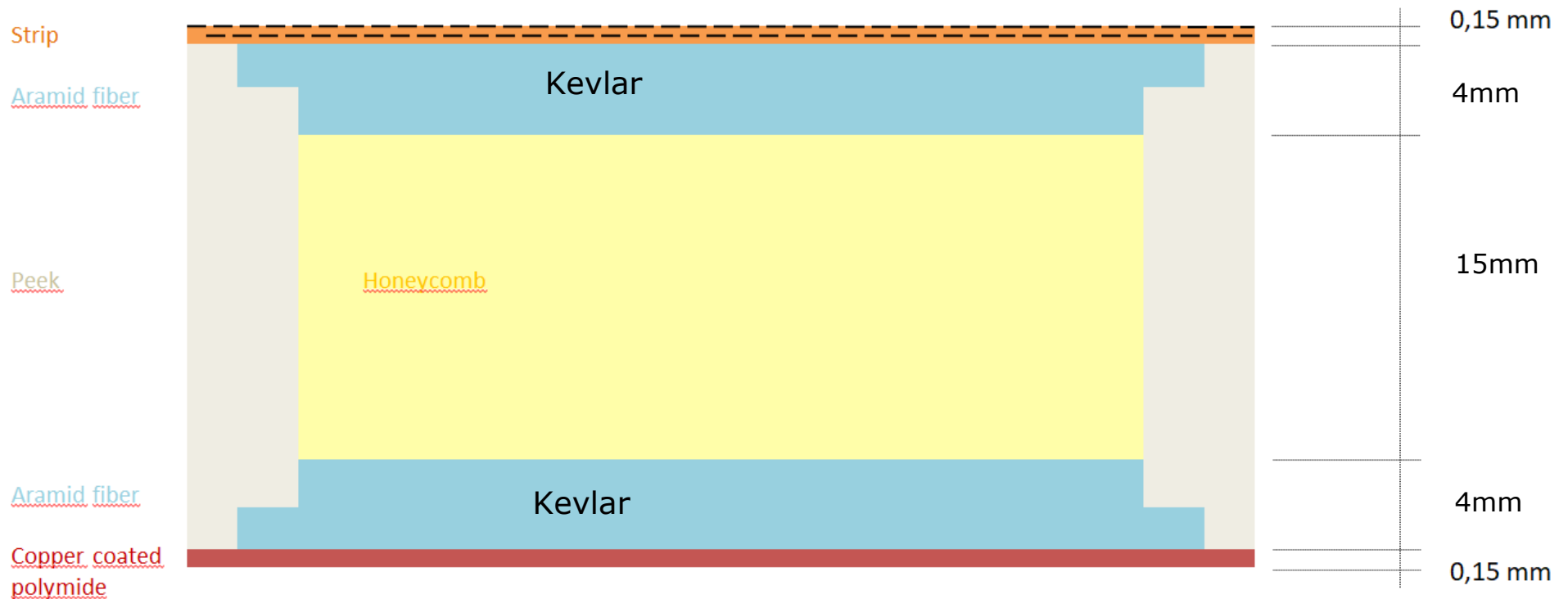
Note: aiming at a **thin** and **low Z** composite wall:

- peels symmetry about Honeycomb (HC) core to minimize deformations (overpressure 5mbar)
- Carbon Fiber (CF) provides best stability, simulations tells: max deform. < O(150μm)
 - CF is low Z material (radiation length ~20cm)
 - CF has low resistivity → additional electric insulaiton required
- Aramid Fiber (or Kevlar) (AF) allows much simplified stack:
 - AF has larger rad. length (~ 28cm)
 - AF provides very good insulation and stability: simulation tells deformation < O(150μm)

(*) mechanics simulation
by J.Mundet

1 – Box wall cross section - example

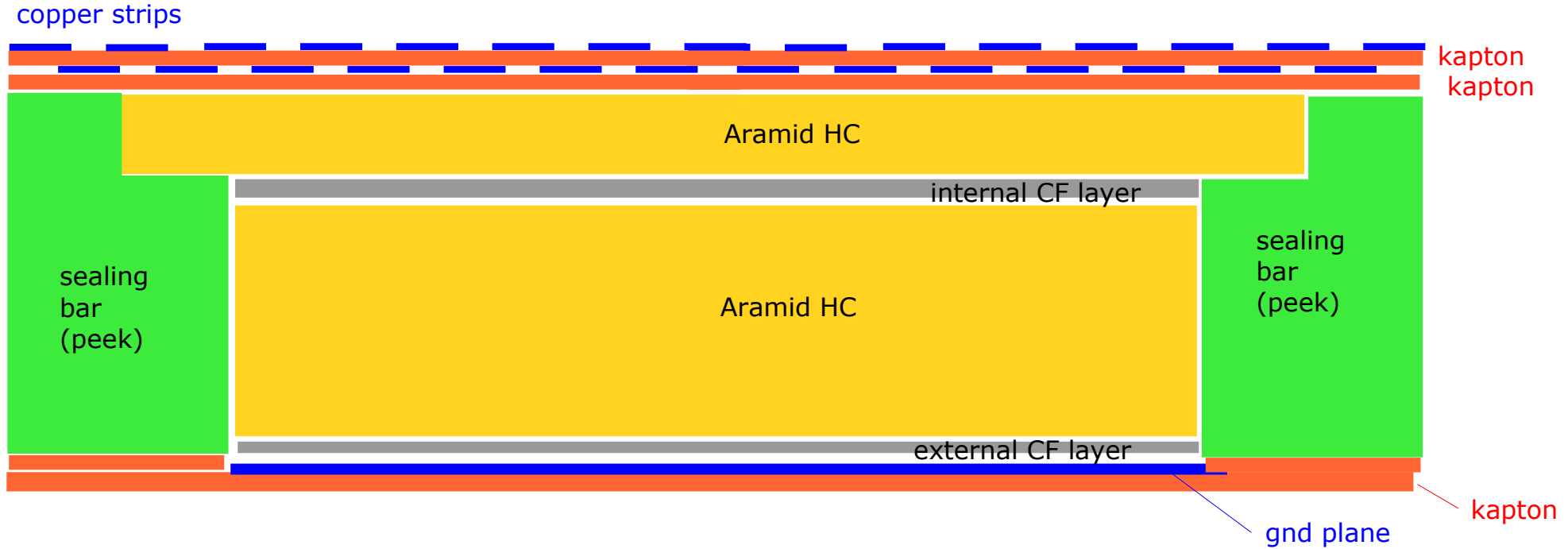
Option 1 → **full aramide** wall structure



M.Romanato – G.Cogo (INFN-PD)

1 – Box wall cross section - example

Option 2 → **carbon fiber** based wall structure



1 – Box wall cross section – impact on physics

César Jesús Valls (IFAE)



ND280-Upgrade TPC-FC Simulation Status

Objectives:

Three possible Models for the TPC FC: [Nexus](#), [GF](#), [Solid](#).

I worked on a Geant4 simulation to study:

- The relative momentum loss.
- The angle spreading.

Simulation overview:

The simulation is made using different logical volumes (one per layer). Each volume has a precise thickness, and material.

To simulate particles I used Particle Gun (Either Muon or Electron).

Energy for the Particle Gun in the range of 0.2 - 2.0 GeV in steps of 0.2 GeV.

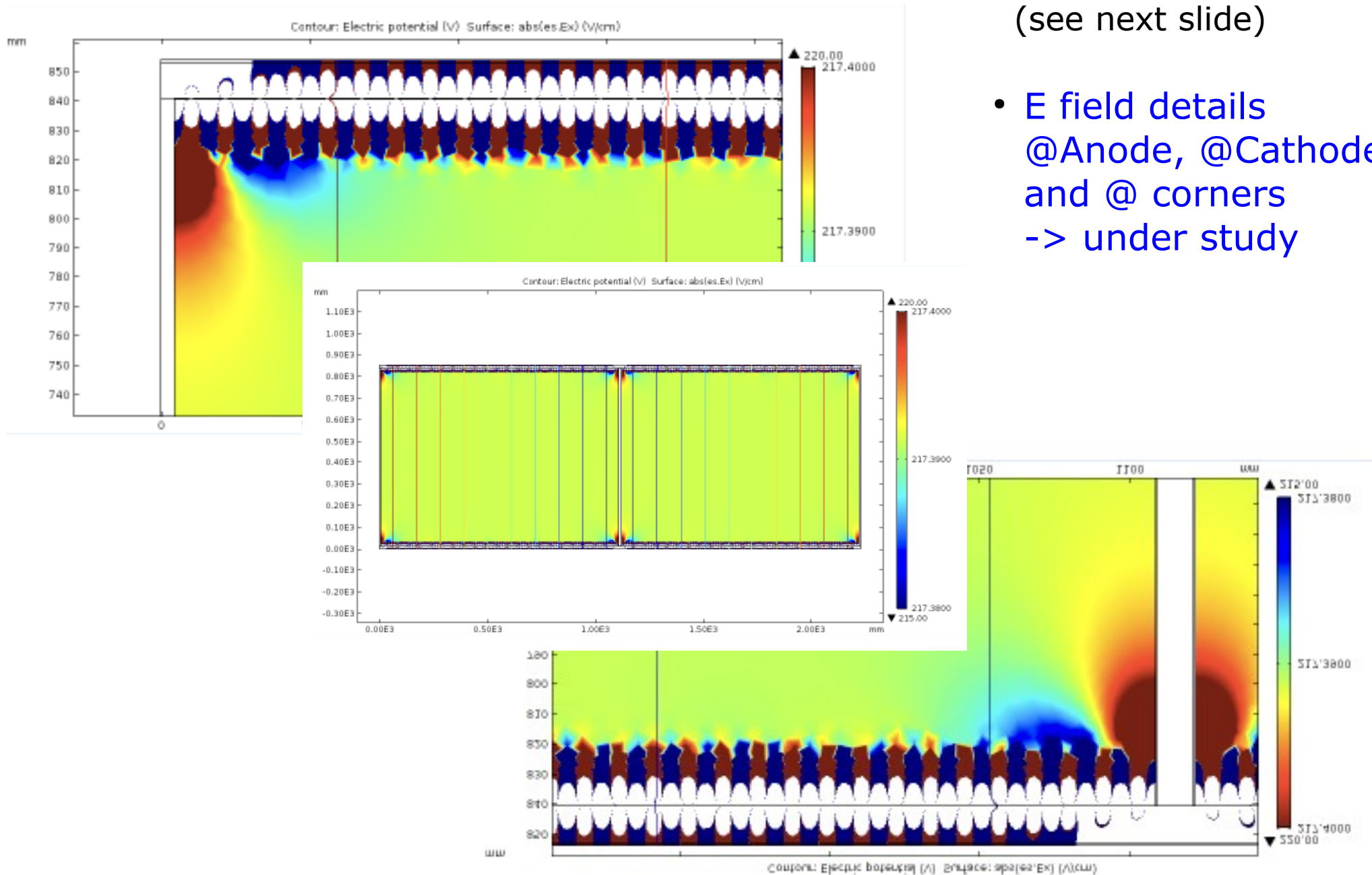
The angle of the Particle Gun has been tuned in from 0° to 75° in steps of 15° . (0° is \perp).

1 - TPC design concepts – Simulation work

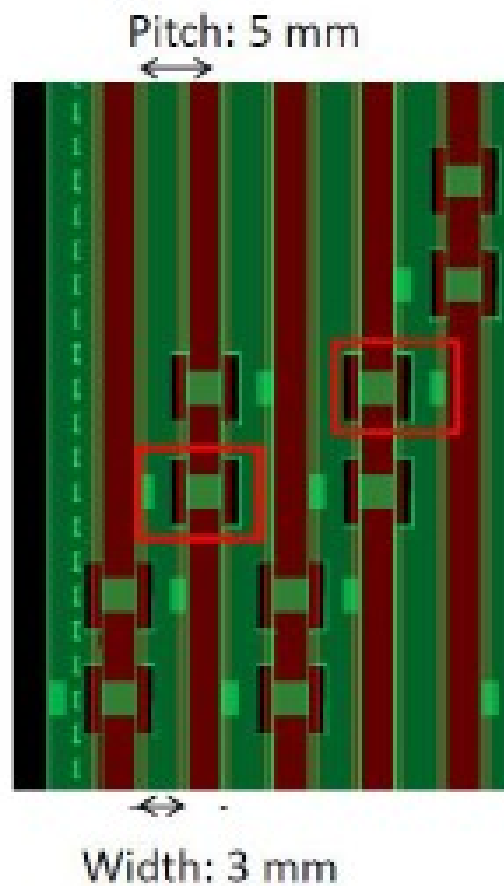
Thanks to simulation work by Lorenzo (COMSOL)

- strip geometry defined (see next slide)

- E field details @Anode, @Cathode and @ corners -> under study



1 - TPC design concepts – Strip design



- double sided
- mirror strips
- all resistors on inner side
- cut marks all 5 cm on inner side
- cross marks for alignment on mirror strip
- foil dimensions currently: ~55x220 cm

ILC TPC Design

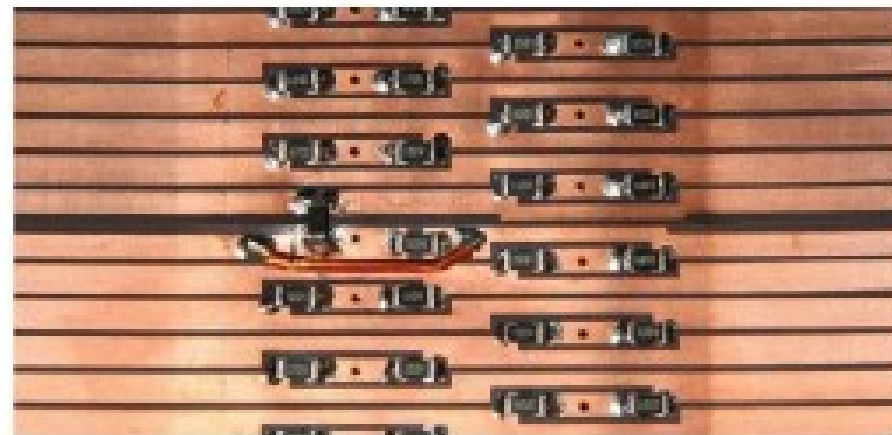
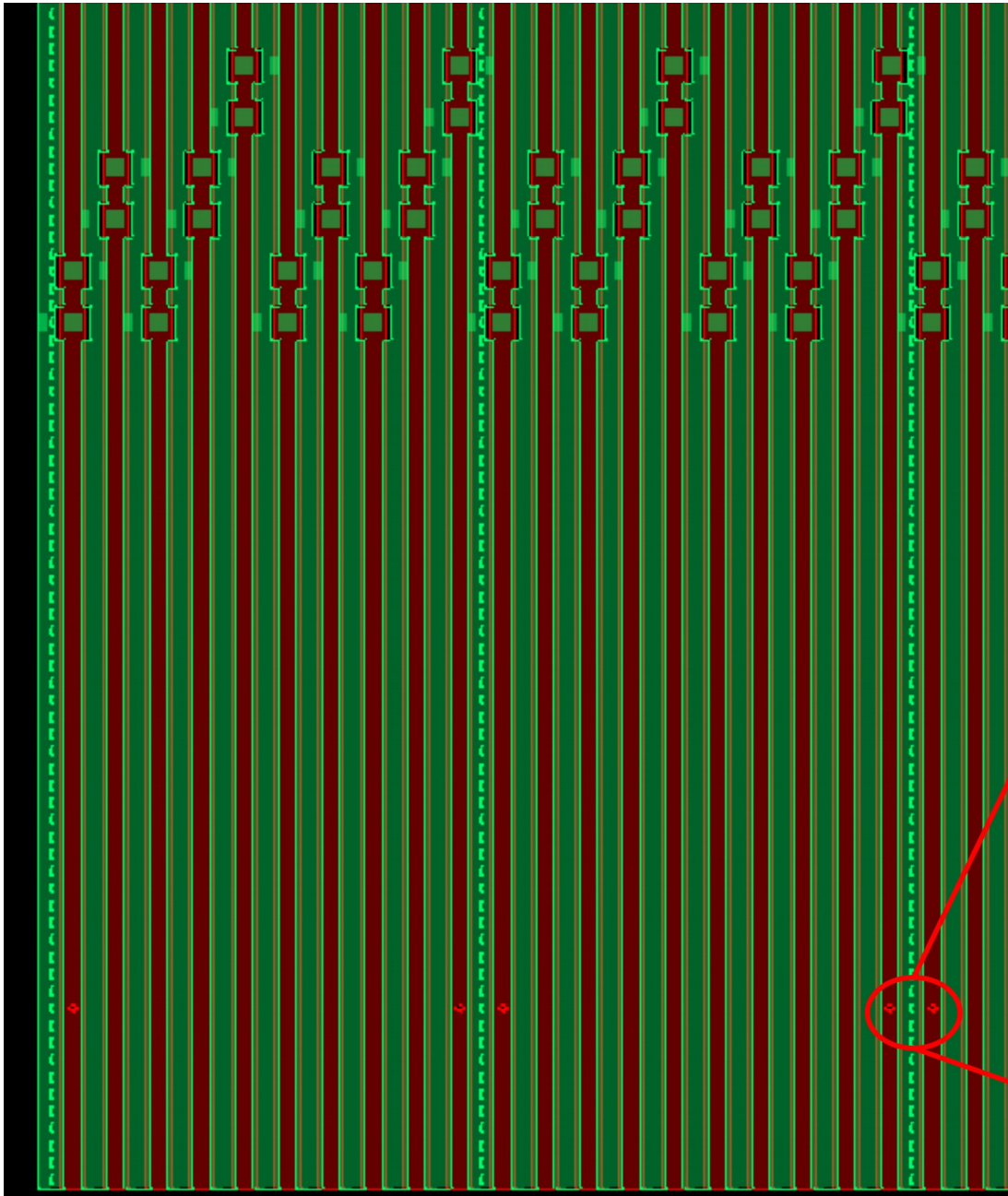
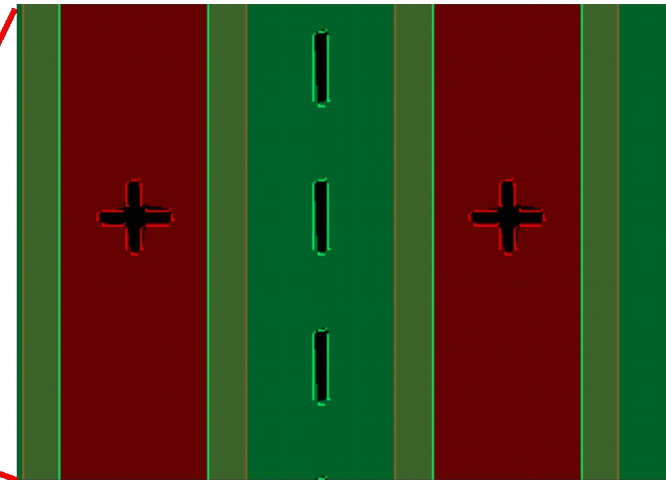


Figure 5.13: Resistor chain in the field cage at the central connection between the half boards

1 - TPC design concepts – Strip design



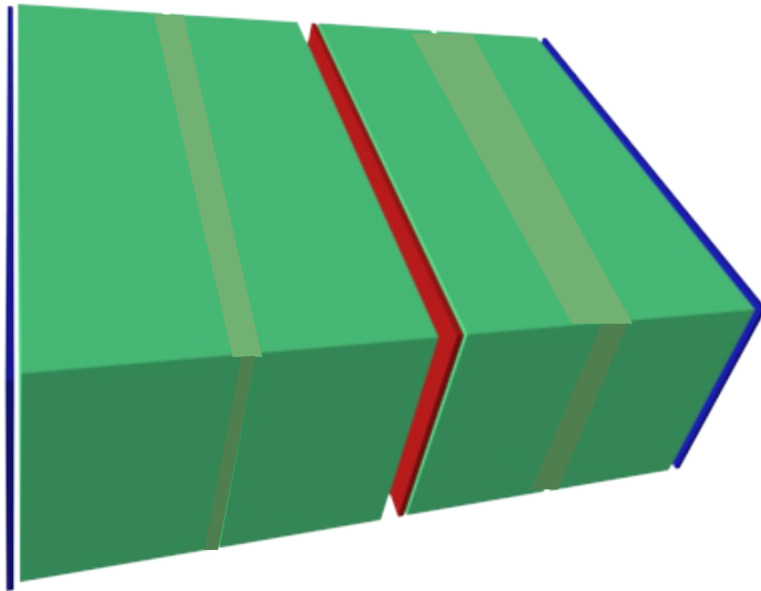
- designed at IFAE
- 4 foils ordered at CERN for tests on prototype (delivery date → early June)



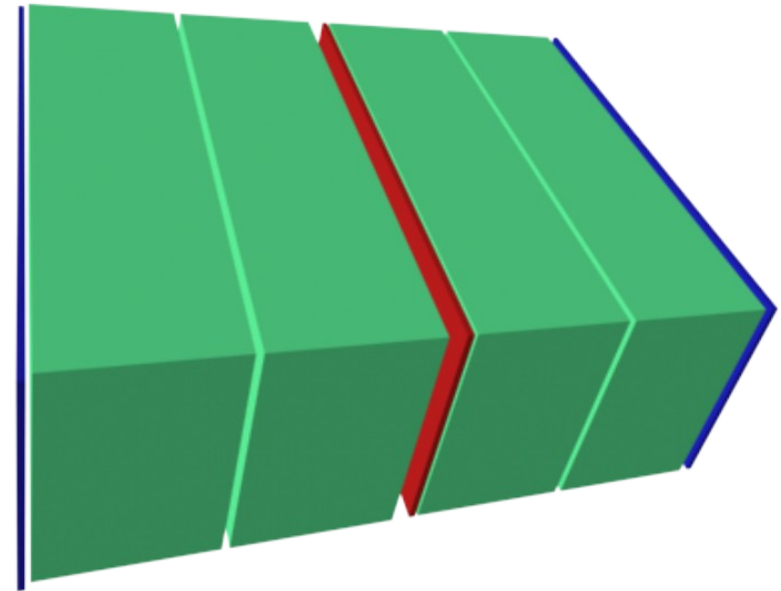
1 - TPC design concepts – Overview

Rectangular TPC assembled either in 2 or 4 boxes:

- 2 boxes (100cm in height) solution apparently simpler than 4 (50 cm) but...
 - composite material process for boxes 50cm in height might be simpler
 - cost estimations will play a role in choice of the final concept design



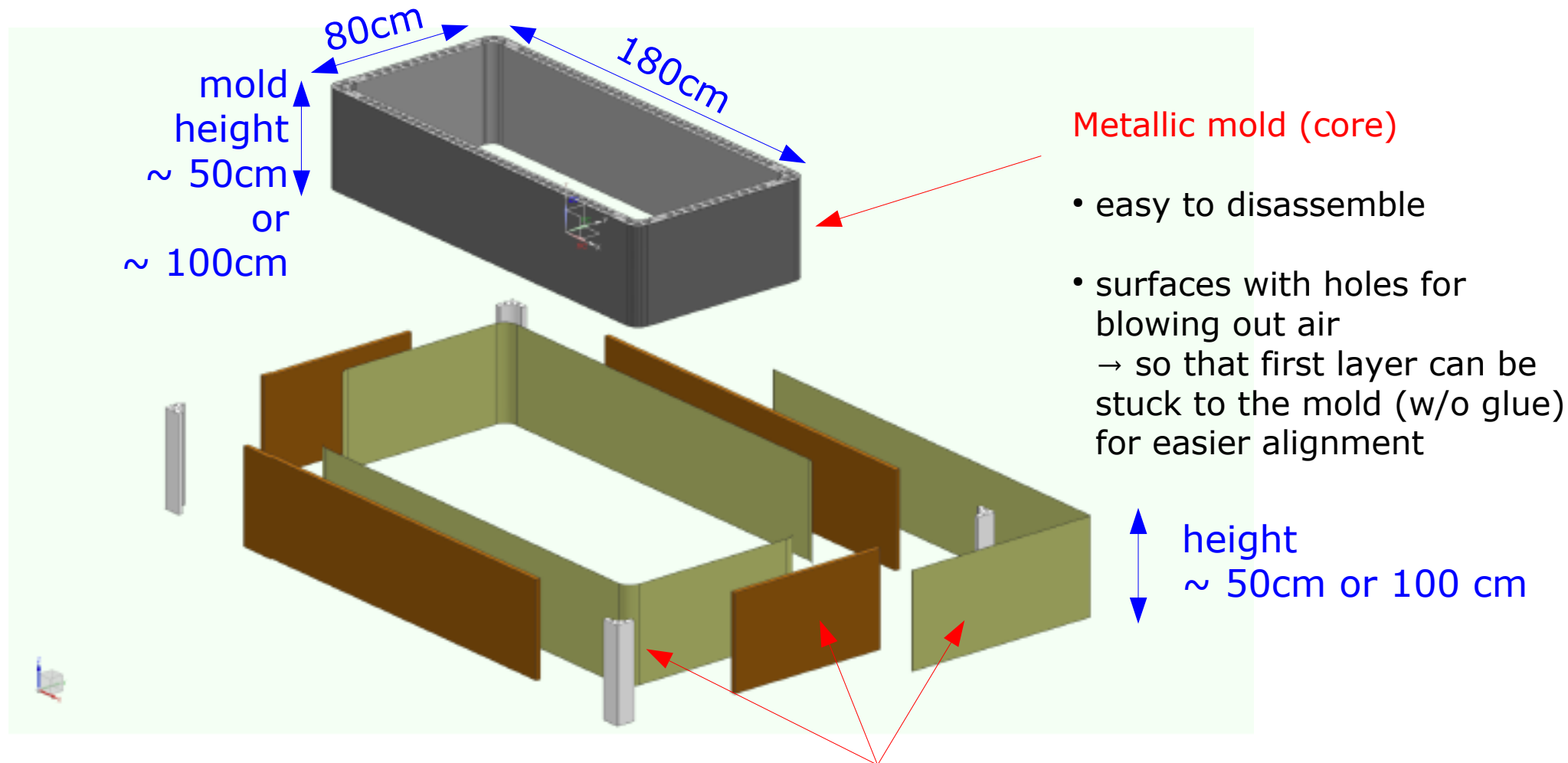
either 2 boxes
each 180 x 85 x **100** cm³



or 4 boxes
each 180 x 85 x **50** cm³

→ prototype useful for **testing box composite material building procedure**

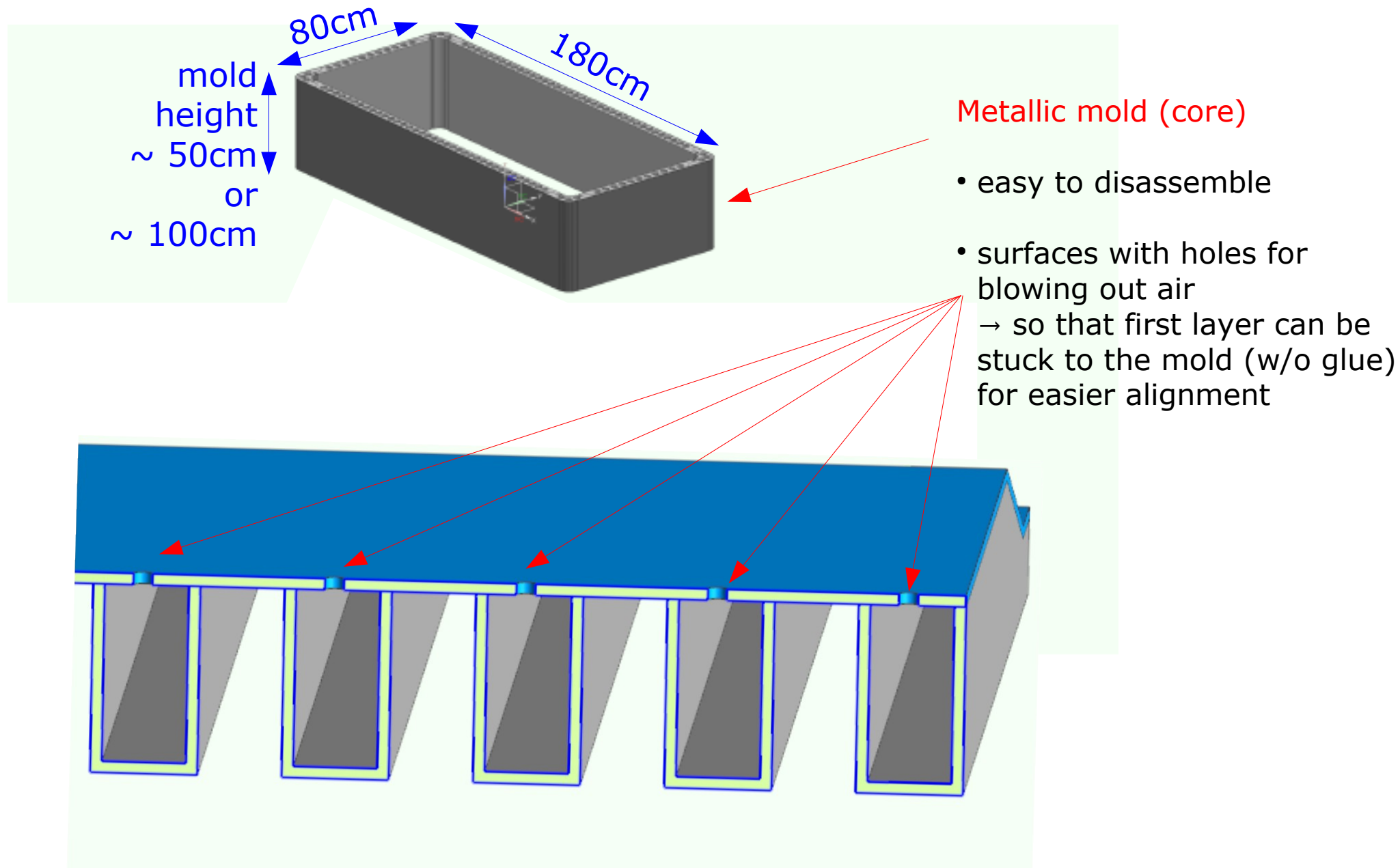
1 - ... details about how to **build a box**



Various **layers to be wrapped** around the mold one by one

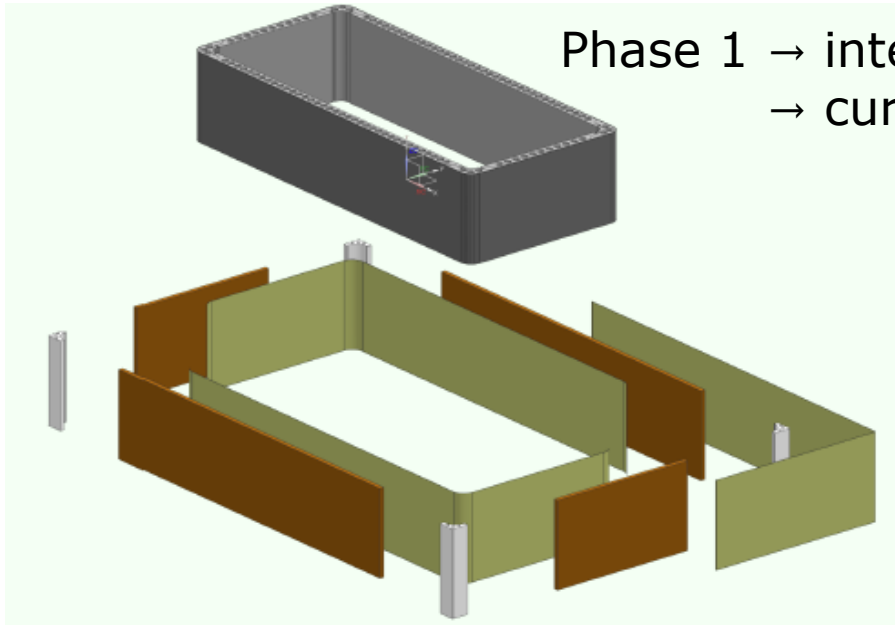
- the following layers should be continuous
 - foils w/ strips → minimizing strip junctions
 - kaptan foils → better insulation but ok w/ joints
 - structural peels (carbon-fiber or kevlar) → mechanical properties
- discontinuous layers
 - panels (HC)
 - corners (PEEK) + bars at edges (PEEK)

1 - ... details about the mandrel/mold

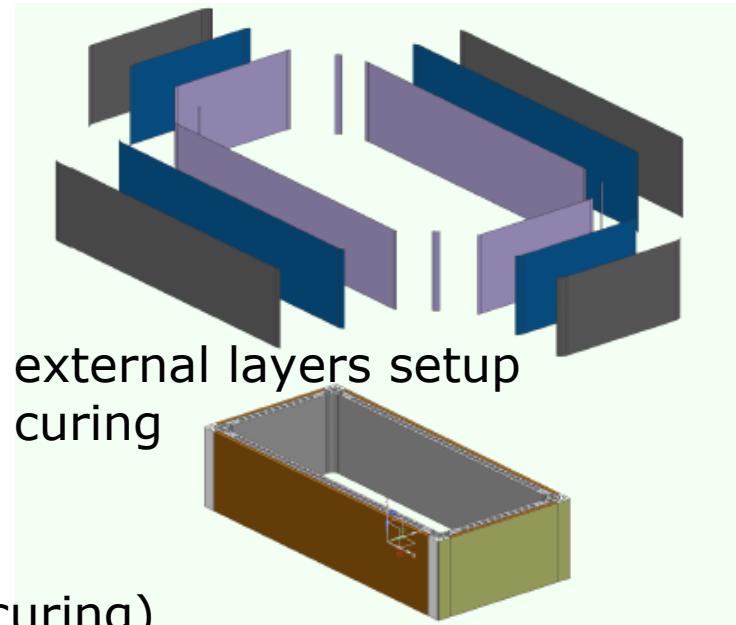


Mandrel concept and design → M.Romanato and G.Cogo (INFN-PD)
(technical design – work in progress → end by mid May)

1 - ... details about how to **build a box**



Phase 1 → internal layers setup
→ curing (autoclave)



Phase 2 → external layers setup
→ curing

Note: about **alignment** and **positioning of layers**
(before curing)

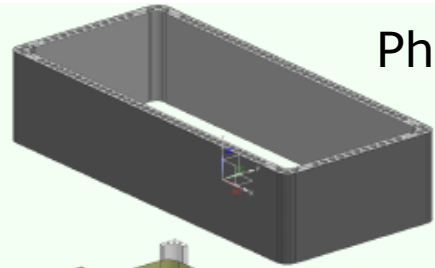
1) use of holes for blowing air trough mold walls
→ positioning of first (internal) layer (strips)

2) use of 3M AF163 type adhesive film
→ layers where no glue is possible/needed

3) use of prepreg Carbon Fiber or Aramid (Kevlar) layers



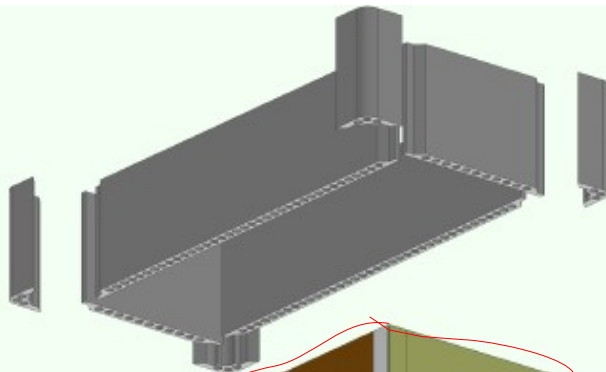
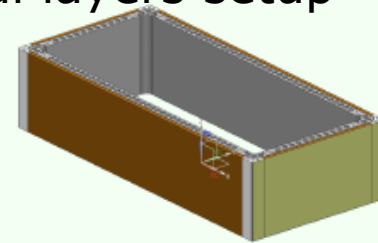
1 - ... details about how to **build a box**



Phase 1 → internal layers setup
→ curing (autoclave)



Phase 2 → external layers setup
→ curing



Phase 3 (Mold dismantled)

→ box edges must be "terminated" by bars to provide:

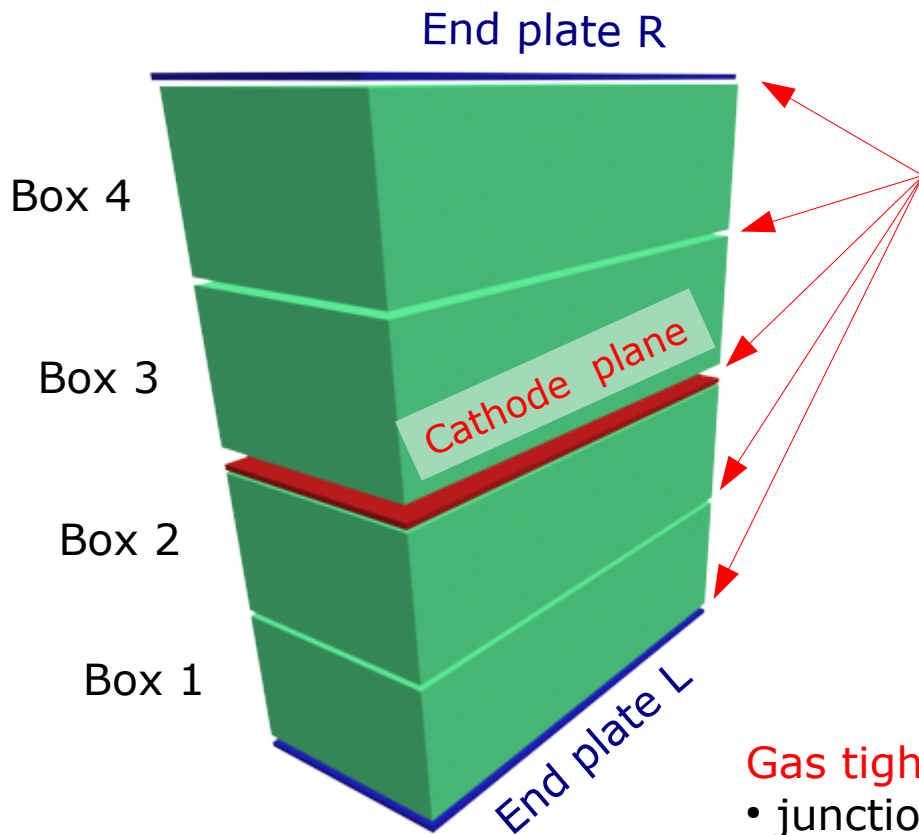
- precision machining
- sealing
- mechanical

top edge

bottom edge

1 - TPC design concepts –

mechanical stiffness
gas tightness
electrical insulation



5 junctions at box edges...

Mechanical point of view

- junctions B1-B2 and B3-B4
- Mechanical stiffness given by proper joints on the **corner structures** (made by ceramic material – Peek)
→ not critical

Gas tightness point of view

- junctions B1-B2, B2-CP-B3 and B3-B4 externally sealed by **gluing a kapton band** wrapped all around
→ must be tested (but not expected to be critical)

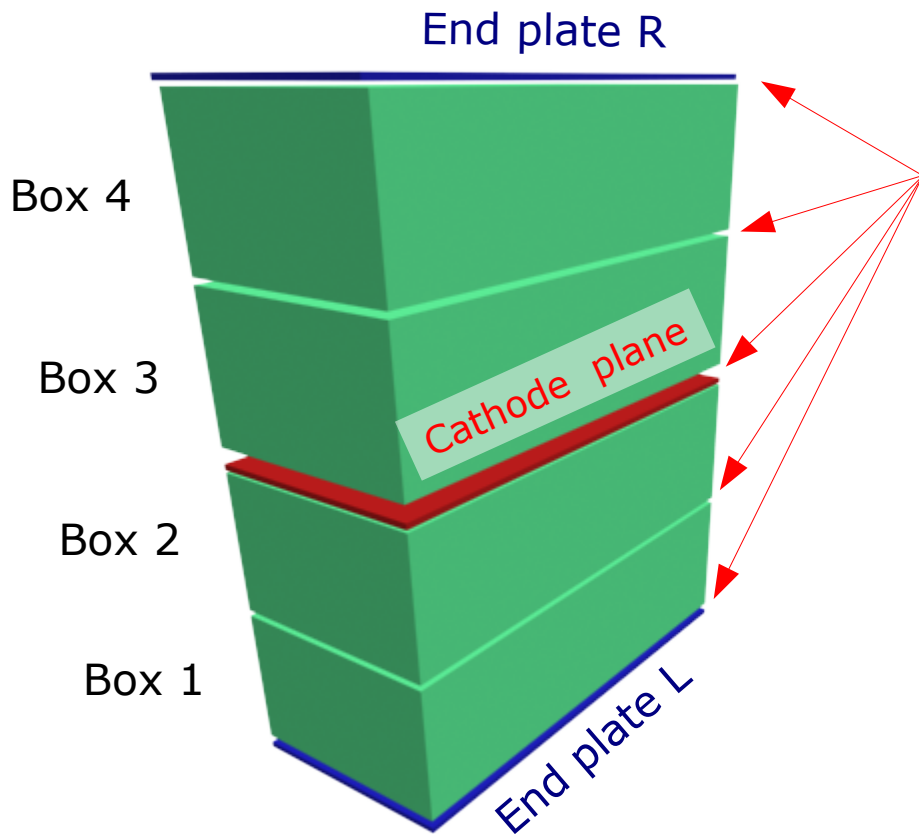
(C) Electrical point of view

- **box edges should be “sealed” by dielectric material** against dangerous current paths to conductive layers (in case of Carbon Fiber option)

Note: presence of internal Carbon Fiber layer might be critical
(should think to set its potential at which level)

1 - TPC design concepts –

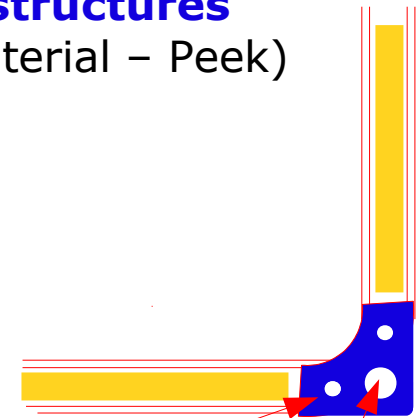
mechanical stiffness
gas tightness
electrical insulation



5 junctions at box edges...

Mechanical point of view

- junctions B1-B2 and B3-B4
- Mechanical stiffness given by proper joints on the **corner structures** (made by keramic material – Peek)
→ not critical

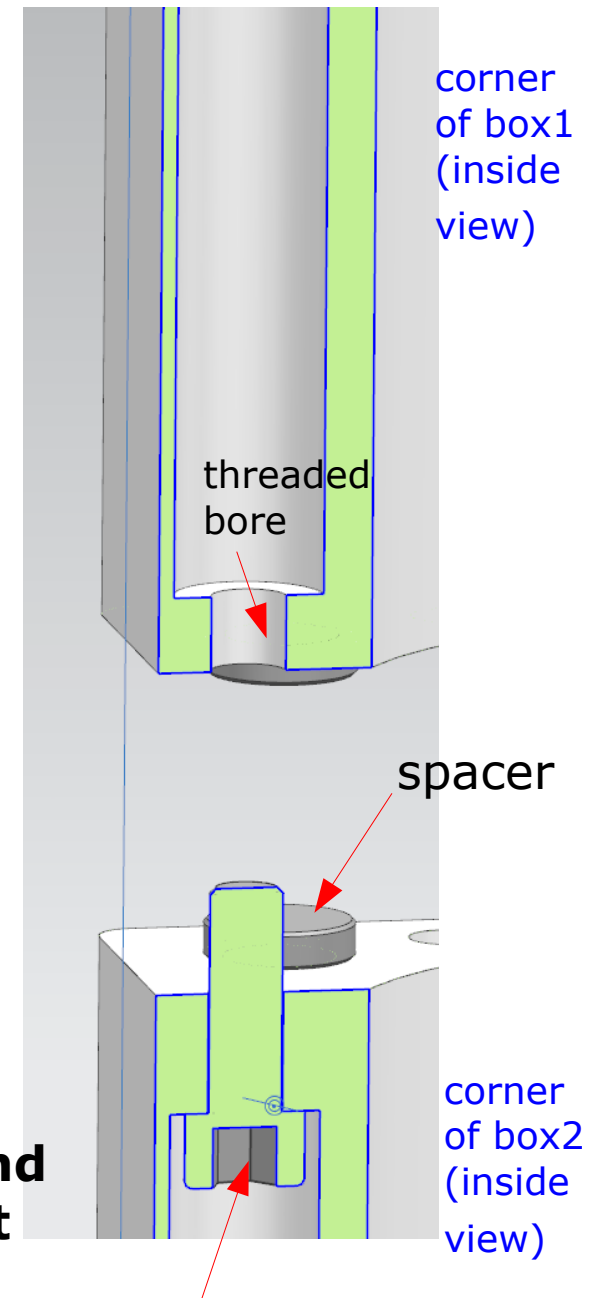
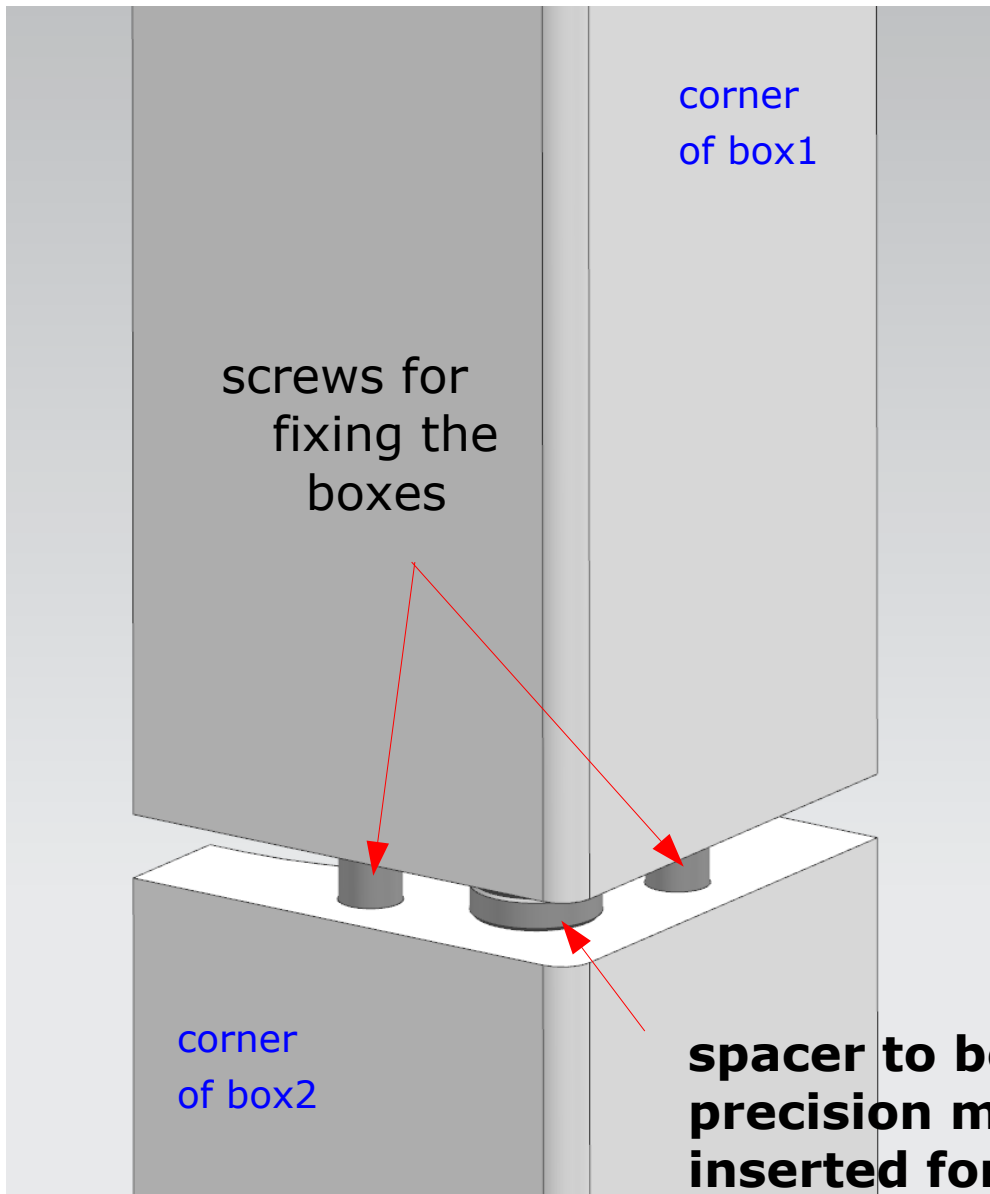


Note: inserts here for alignment
Indeed due to solution below the edges
will not be machinable → we should rely
only on corner structures for alignment

bores for
joining boxes

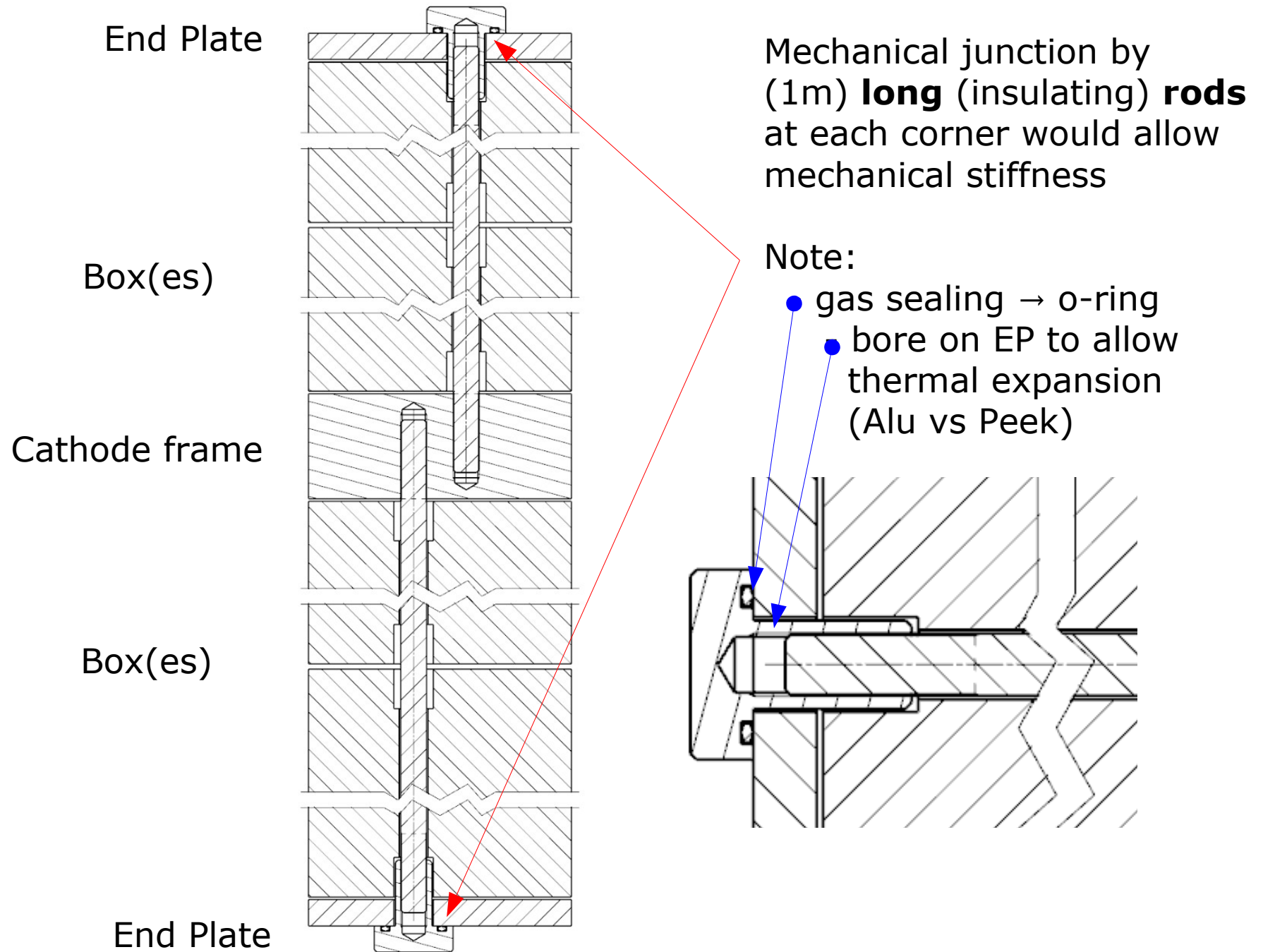
Note: corner structure to be easility
machined → Peek

Joining 2 boxes – by the angular structures



access to screw with tool from below

Joining 2 boxes – by the angular structures

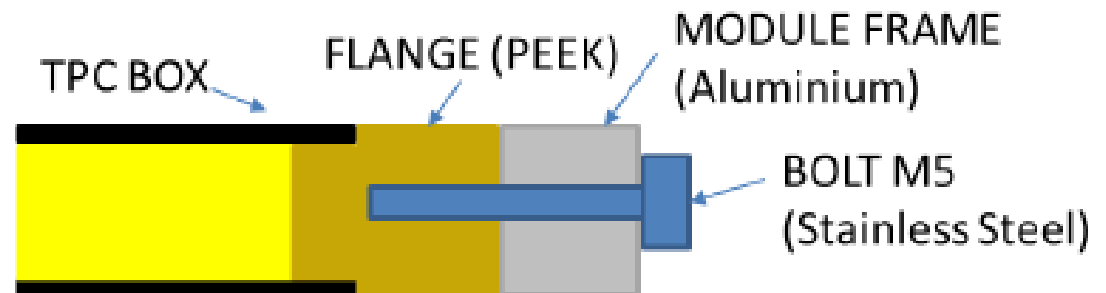


Note, BTW: no relevant problem in joining End-Plate Al frame to Field-Cage edge/corner Peek parts

MODULE FRAME (ALUMINIUM) / TPC BOX FLANGE (PEEK) - Thermal loads

$$\Delta t_{\max} = 5^{\circ}\text{C}$$

$$\Delta T_{\min} = -5^{\circ}\text{C}$$



DEFORMATION DUE TO THERMAL LOAD

		Module Frame	Flange	Assembly	Screw	Difference	
Material		Al7075-T6	PEEK	-	SS	-	
Coefficient thermal expansion	CTE [1/K]	2,14E-05	4,68E-05	-	1,80E-05	-	
Length	L [mm]	15	15	30	30	-	
Deformation due to ΔT_{\max}	xmax [mm]	-0,0016	-0,0035	-0,0051	-0,0027	0,0024	INCREASE
Deformation due to ΔT_{\min}	xmin [mm]	0,0016	0,0035	0,0051	0,0027	-0,0024	DECREASE

LOAD DUE TO THERMAL LOAD

Screw Young's Modulus	E [Mpa]	200000
Screw Equivalent section	As [mm ²]	14,18
Screw Length	L [mm]	30
Screw Rigidity	k [N/mm]	94552
Deformation due to ΔT_{\max}	xmax [mm]	0,0024
Deformation due to ΔT_{\min}	xmin [mm]	-0,0024
Force in the screw due to ΔT_{\max}	Pdtmax [N]	228
Force in the screw due to ΔT_{\min}	Pdtmin [N]	-228

Preload on M5 = 5000 N

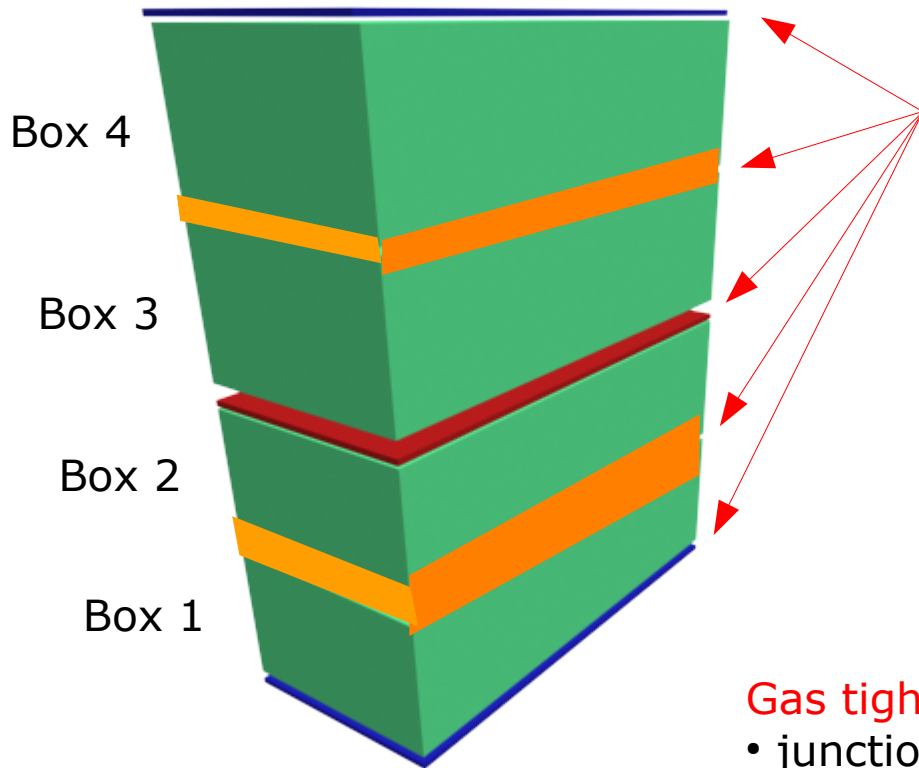
Increase of 4% preload (OK)

Loss of 4% preload (OK)

Calculations by Juli; OK also by Adriano

1 - TPC design concepts –

mechanical stiffness
gas tightness
electrical insulation



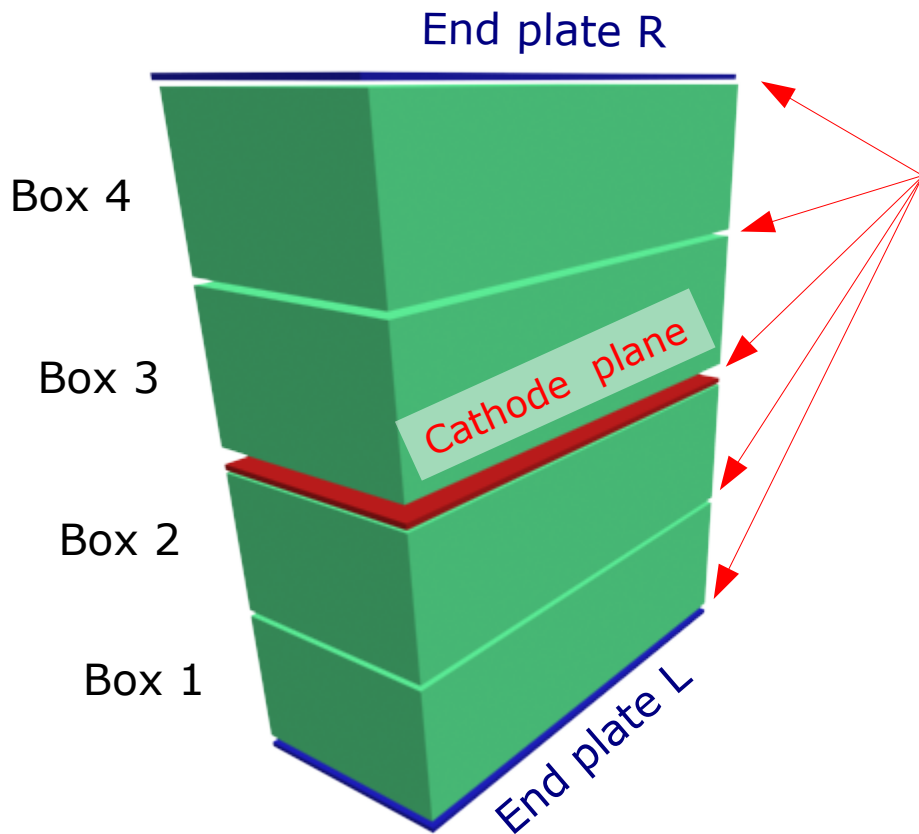
5 junctions at box edges...

Gas tightness point of view

- junctions B1-B2, B2-CP-B3 and B3-B4 externally sealed by **gluing a kapton band** wrapped all around
→ must be tested (but not expected to be critical)

1 - TPC design concepts –

mechanical stiffness
gas tightness
electrical insulation



5 junctions at box edges...

(C) Electrical point of view

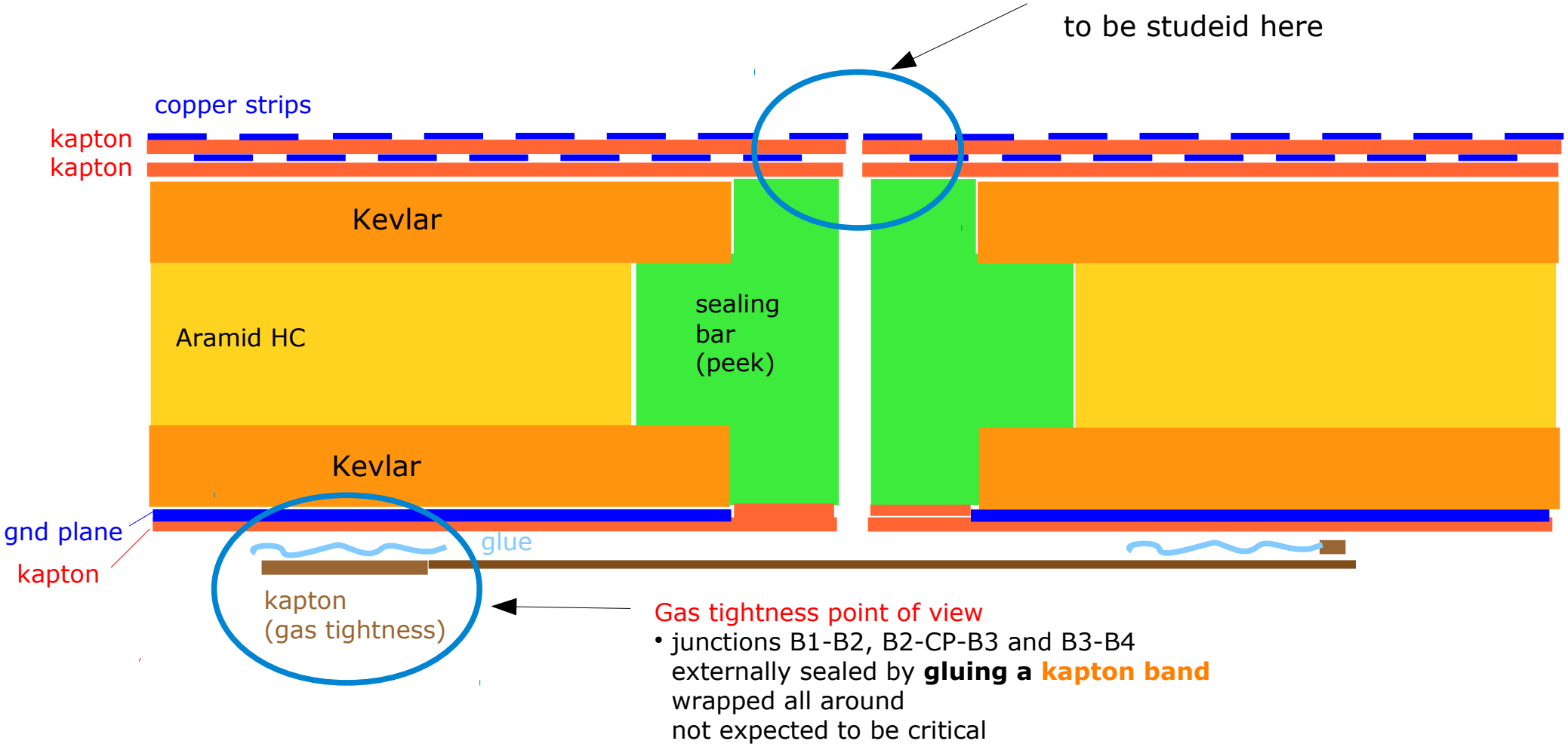
- **box edges should be "sealed" by dielectric material** against dangerous current paths to conductive layers

Note: presence of internal Carbon Fiber layer might be critical (should think to set its potential at which level)

Dielectric and gas "sealing" at Box edges

Option 1 → **full aramide** wall structure

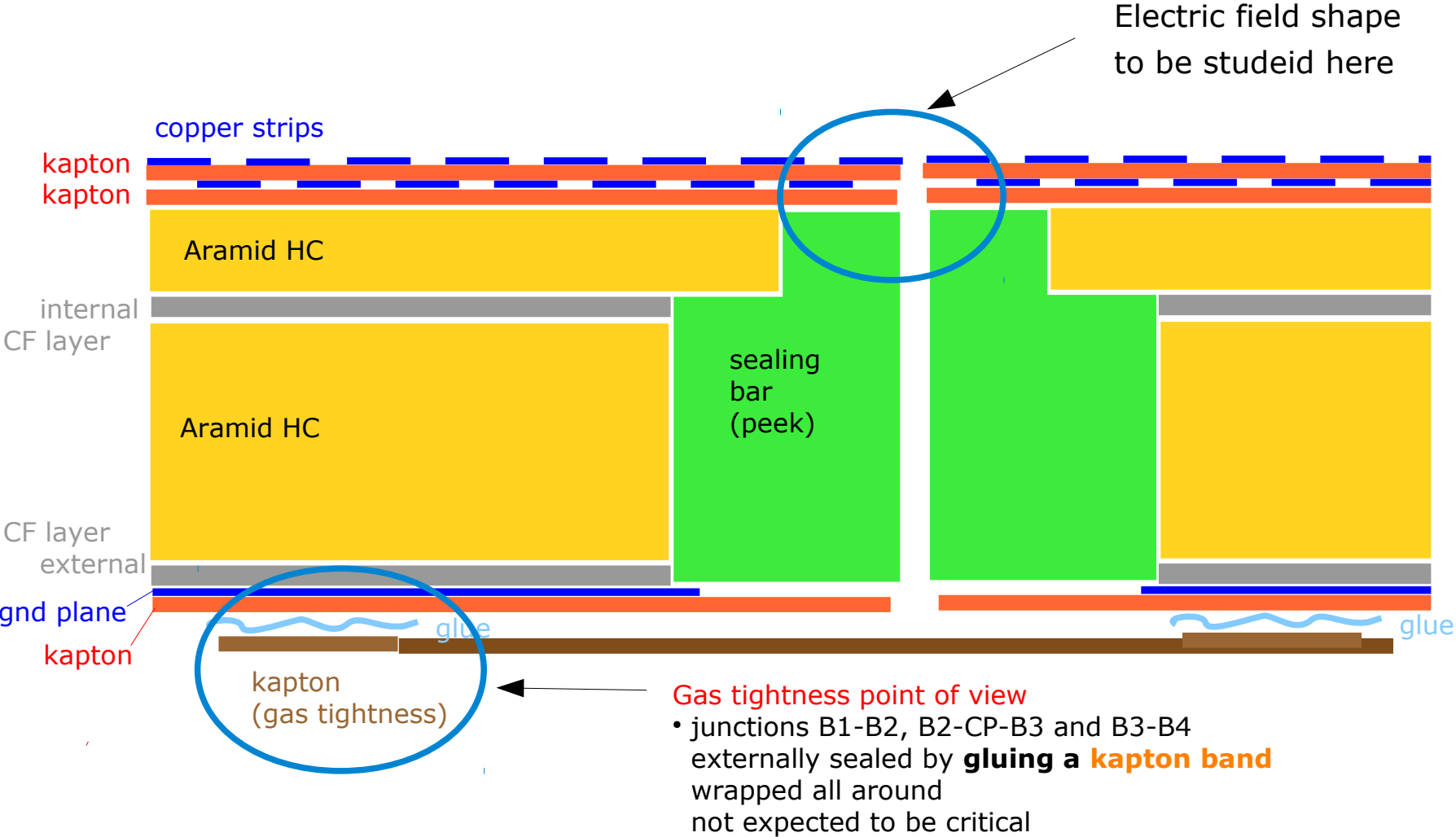
Electric field shape to be studied here



→ **will be proved with mockup structure**

Dielectric and gas "sealing" at Box edges

Option 2 → carbon fiber based wall structure



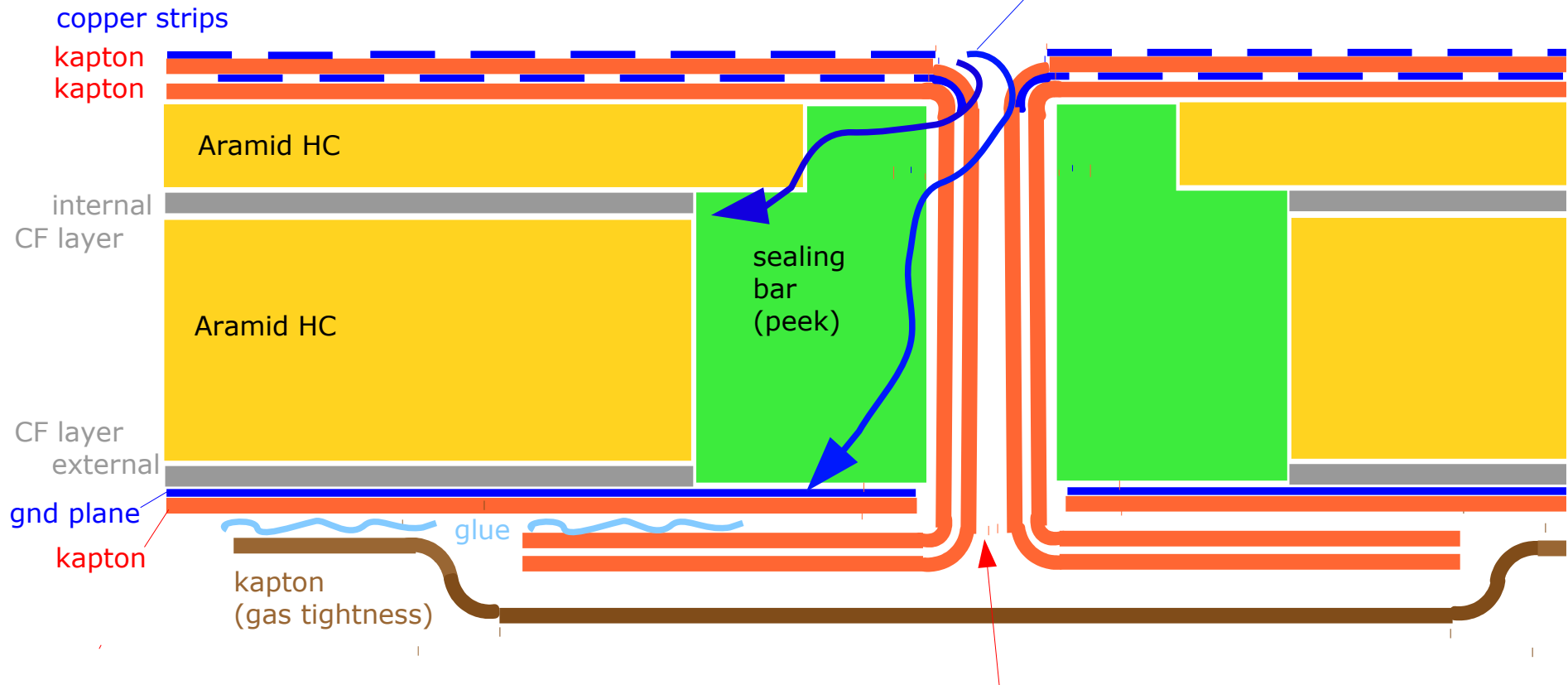
Dielectric and gas sealing at box edges – CF problems

Carbon fiber seems to add more complications due enhanced probability of breakdown

Electrical point of view

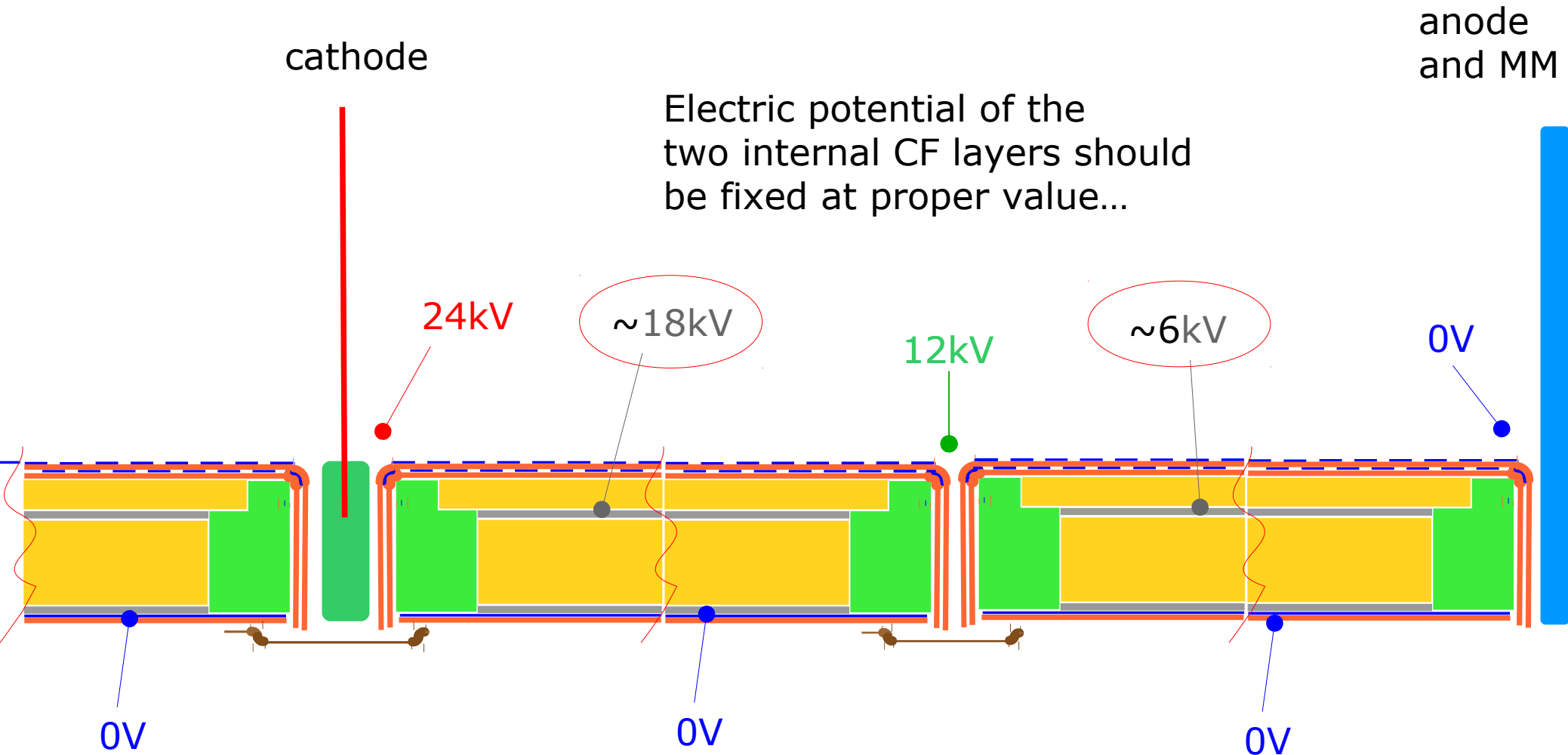
will be tested with sample structure ←

- box edges should be “sealed” by dielectric material against dangerous **current paths** to conductive layers



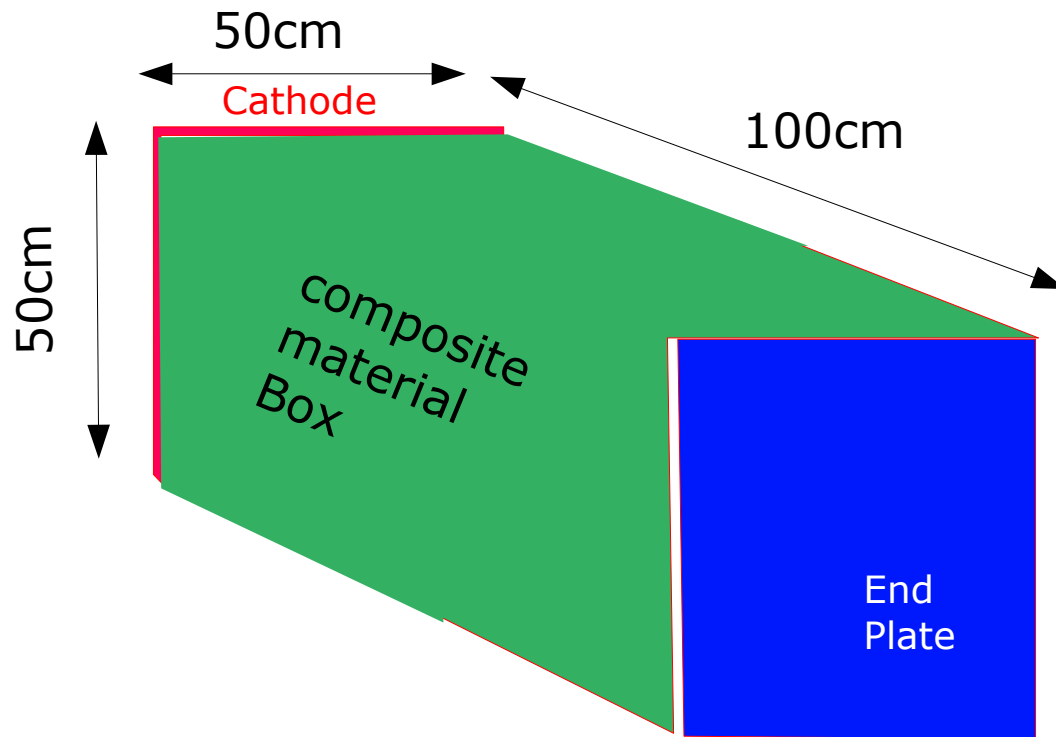
This wall structure might require additional insulation (by Kapton later folded outward)

Dielectric and gas sealing at box edges - CF potential



Note: presence of internal Carbon Fiber layer might be critical (should think to set its potential at which level)

2. Prototype and tests



Composite samples to be produced at 2 facilities:

1. NEXUS company (Barcelona)
2. CERN composite material lab

TPC concept design tests by building the prototype

1. building the box
→ 100cm in height (baseline)
2. mechanical properties of walls
→ deformation
→ planarity of internal wall plane
→ bubbles affecting internal wall side
→ building box edges
3. electrical breakdown at edges
→ edge termination structure
4. gas tightness solution (O_2)
→ gas sealing by kapton band gluing
(in vacuum w/ mass spectrometer)

2. Prototype Tentative Time schedule

Test structures

- test structures at CERN: by end of May
 - mechanical and electrical tests (Padova) by end of June
 - gas sealing and outgassing test (Padova) by end of June
- test structures by NEXUS: by mid May

Prototype:

- mandrel design (INFN Padova): by mid May
 - mandrel building (INFN Bari): by mid July
 - mandrel at NEXUES (Barcellona): by end July
 - building of the boxes: still to be agreed with NEXUS
- ... dead line for prototype assembly at CERN → late October ?

Conclusions

- Design of the new thin TPCs by exploiting **composite material techniques**
- Two main options for composite structure under evaluation:
Carbon Fiber vs Full Aramid
- Building by Boxes approach: **box length (drift direction) under evaluation**
- Electric field: **details at the Anode, Cathode corners under study**
- **Mechanical and Electrical breakdown tests**: samples in preparation
- **Prototype not ready for August TB**, but plans are still to have it **ready for tests with particles at CERN in Fall 2018**

Additional material

1 - TPC design concepts – Box structure

- Rectangular TPC of 200 (B field / drift dir.) x 180 (beam dir.) x 85 (height) cm³
- Aiming at **thin wall** (~3cm) made of **low Z composite material** (few % X_0)

Carbon Fiber based layer stack

Material	Thickss (mm)
<i>Copper coated polyimide film</i>	<i>~ 0.15</i>
<i>Carbon Fiber</i>	<i>1.20</i>
<i>Aramide HoneyComb panel</i>	<i>15.00</i>
<i>Carbon Fiber</i>	<i>1.20</i>
<i>Aramide HoneyComb panel (insulation)</i>	<i>10.00</i>
<i>Polyimide film (insulation)</i>	<i>~ 0.10</i>
<i>Strips (double later) on Kapton foil</i>	<i>~ 0.15</i>
TOTAL RADIATION LENGHT ~ 4,0 % X_0	~27.5

Glass Fiber based layer stack

Material	Thickss (mm)
<i>Copper coated polyimide film</i>	<i>~ 0.15</i>
<i>Glass Fiber</i>	<i>0.30</i>
<i>Aramide HoneyComb panel</i>	<i>20.00</i>
<i>Glass Fiber</i>	<i>0.30</i>
<i>Polyimide film (insulation)</i>	<i>~ 0.10</i>
<i>Strips (double later) on Kapton foil</i>	<i>~ 0.15</i>
TOTAL RADIATION LENGHT ~ 6.5 % X_0	~21.5

Note: aiming at a **thin** and **low Z** composite wall:

- peels symmetry about Honeycomb (HC) core to minimize deformations (overpressure 5mbar)
- Carbon Fiber (CF) provides best stability, simulations (*) tells: baseline structure max deform. < O(150 μ m)
 - CF is low Z material (radiation length ~20cm)
 - CF has low resistivity → additional electric insulaiton required
- Fiber Glass (GF) allows much simplified stack;
 - GF has lower rad. length (~ 13cm)
 - GF provides worser stability: simulation tells deformation < O(400 μ m)

(*) mechanics simulation
by J.Mundet

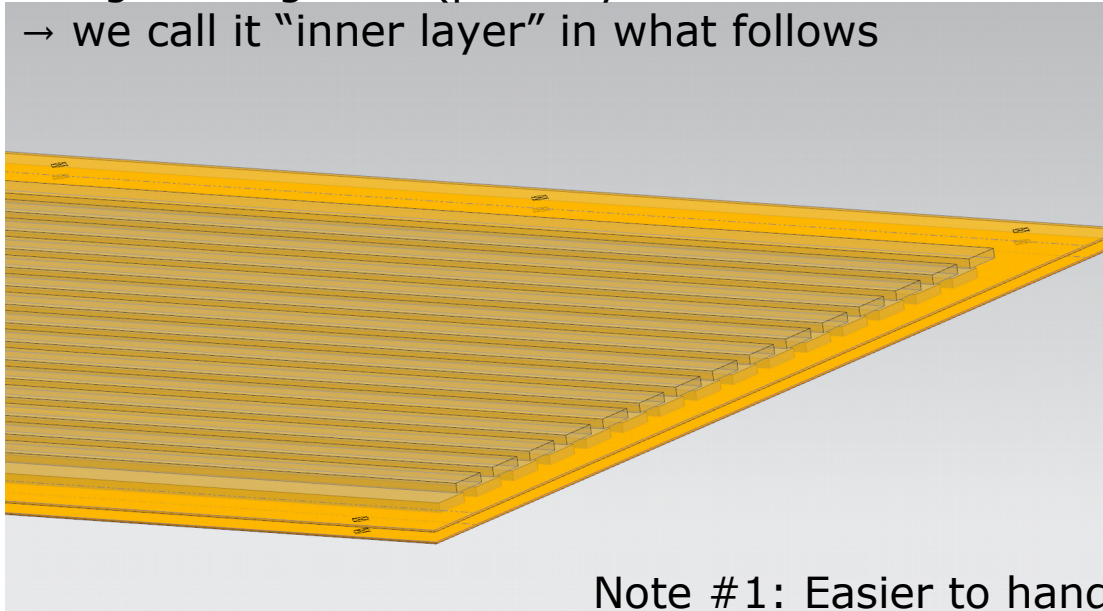
Possible procedure for

- bulding the boxes in 3 phases and
- assembling them

Phase I – preparing the inner layer on a flat table

On top of a flat table the strip double layer + the kapton layer are glued together (possibly with a no flaw adhesive thin film)

→ we call it "inner layer" in what follows



Note #2: extra foil length beyond strips region for two reasons:
1) alignment (markers should be there)
2) to be folded for sealing edges of the box

Note #1: Easier to handle two foils corresponding to half of the total TPC perimeter (180+80cm)

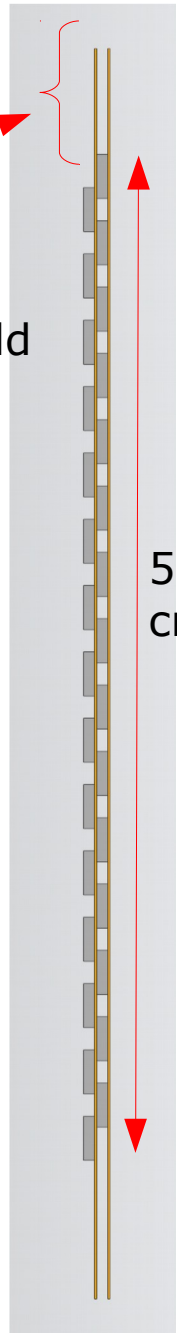
180cm

80cm

50 cm

strips geometry not in scale

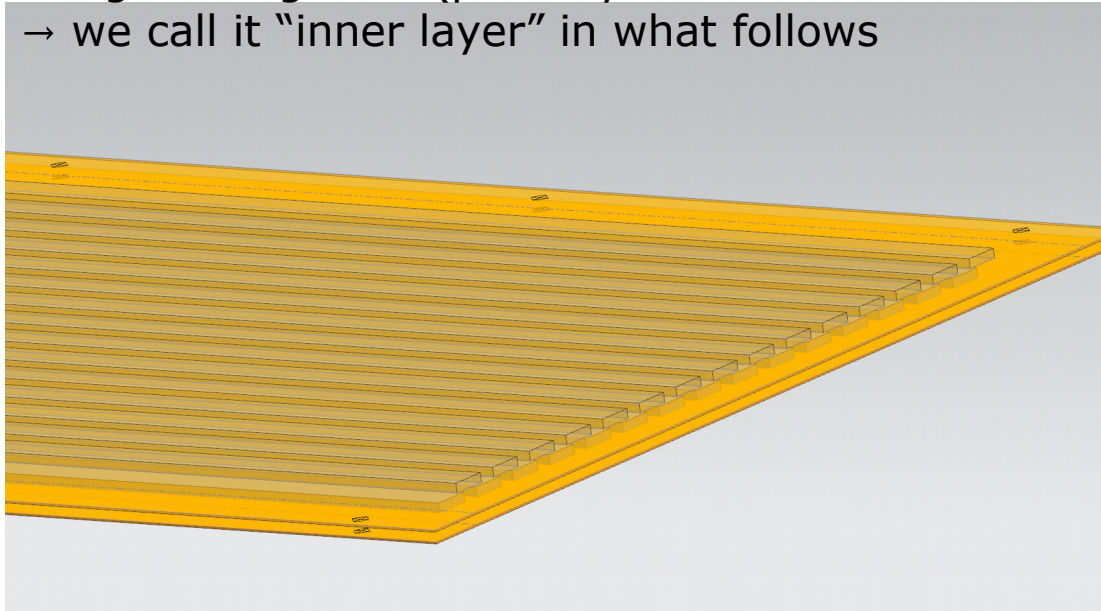
50 cm



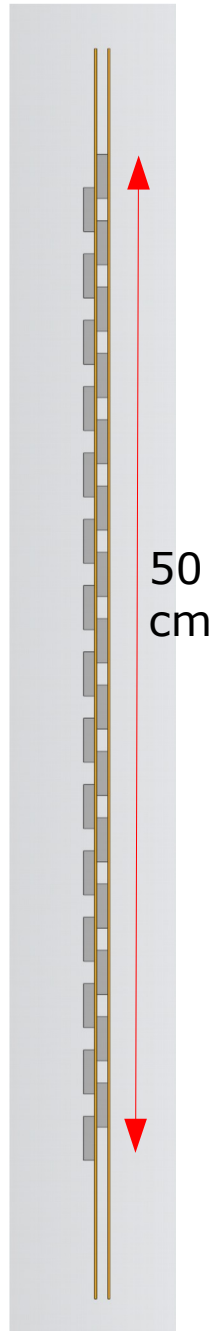
Phase I – preparing the “inner layer” on a flat table

On top of a flat table the strip double layer + the kapton layer are glued together (possibly with a no flaw adhesive thin film)

→ we call it “inner layer” in what follows



Note #3: the dip here is for allowing proper folding outward of the extra region of the foil (see 3D view next slide)

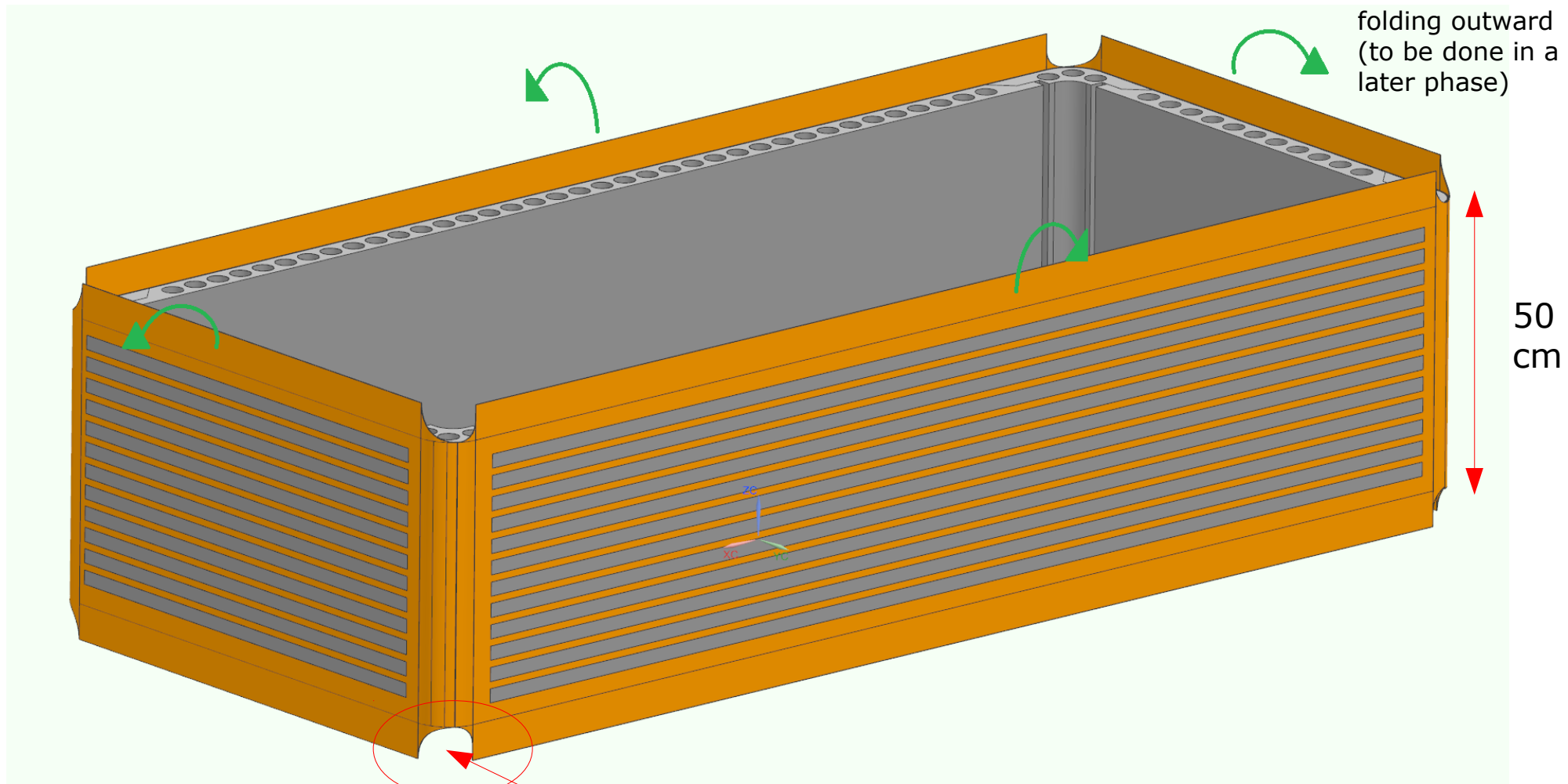


strips geometry not in scale

50
cm

Phase II – the “inner layer” is wrapped around the mold

The Inner layer is aligned (with markers markers) on the mold
Blowing out air simplifies a lot the procedure



folding outward
(to be done in a
later phase)

50
cm

Note: the dip here is for allowing
proper folding outward of the
extra region of the foil

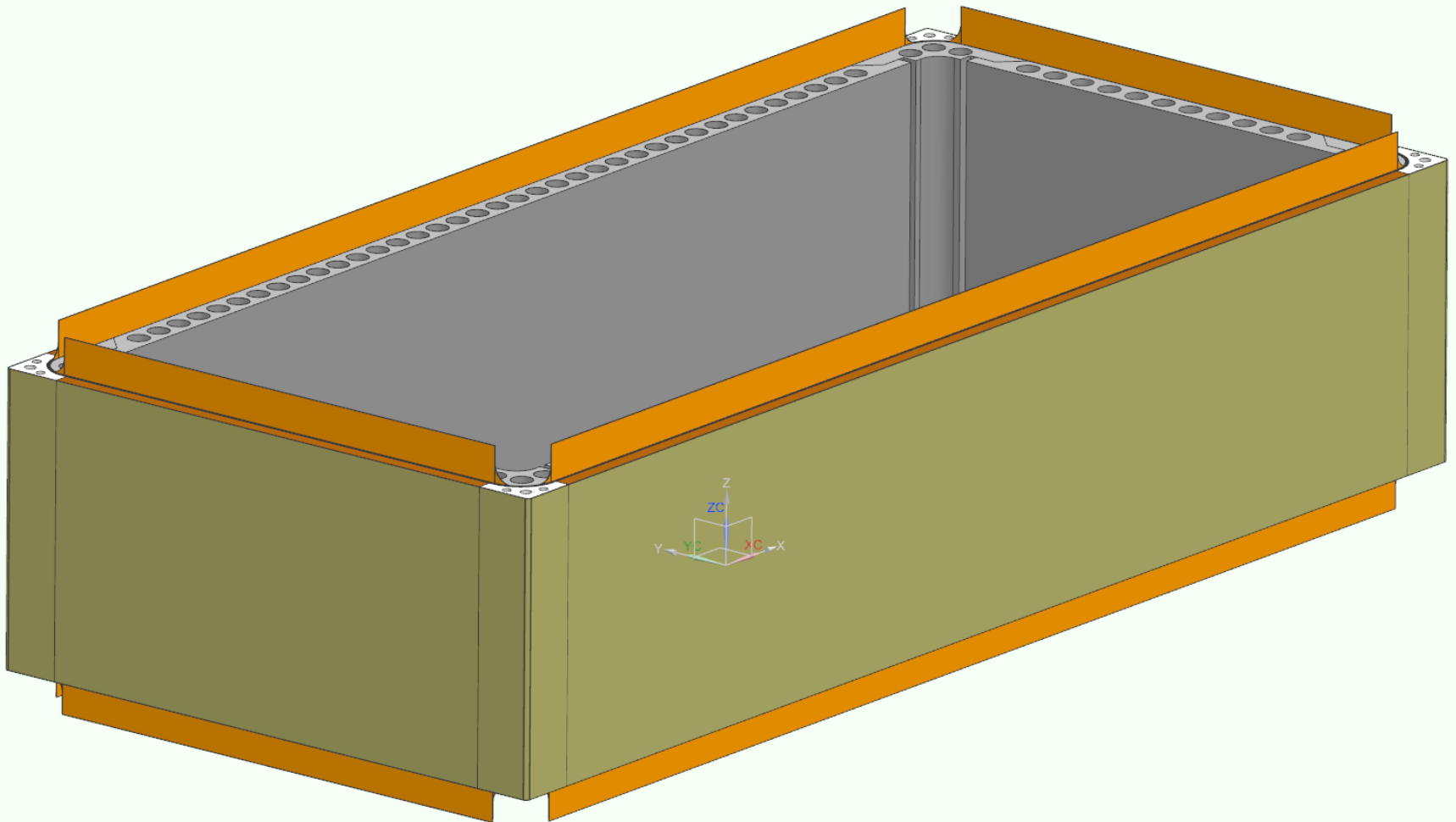
Phase II – the other layers are applied around the mold

Layers are applied according to the stack sequence (going outward):

1. inner layer – 2. HC (6mm) – 3. CF – 4. HC (16mm) – 5. CF – 6. (outer layer)

Note #1: the use of thin (160um) adhesive film layers (no flow) simplify the stack preparation

Note #2: intermediate curing phases might be needed



Phase II – the other layers are applied around the mold

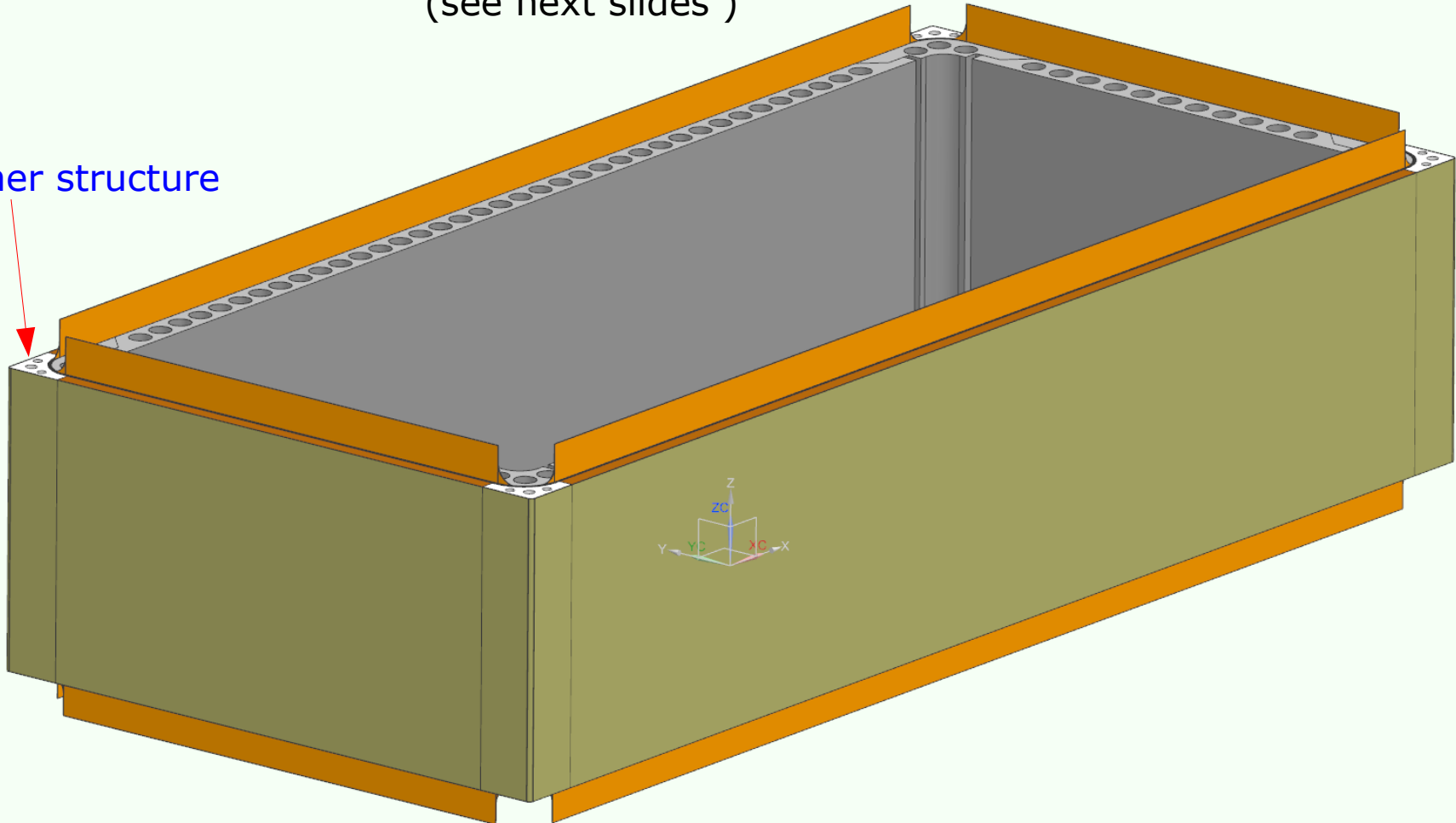
Layers are applied according to the stack sequence (going outward):

1. inner layer – 2. HC (6mm) – 3. CF – 4. HC (16mm) – 5. CF – 6. (outer layer)

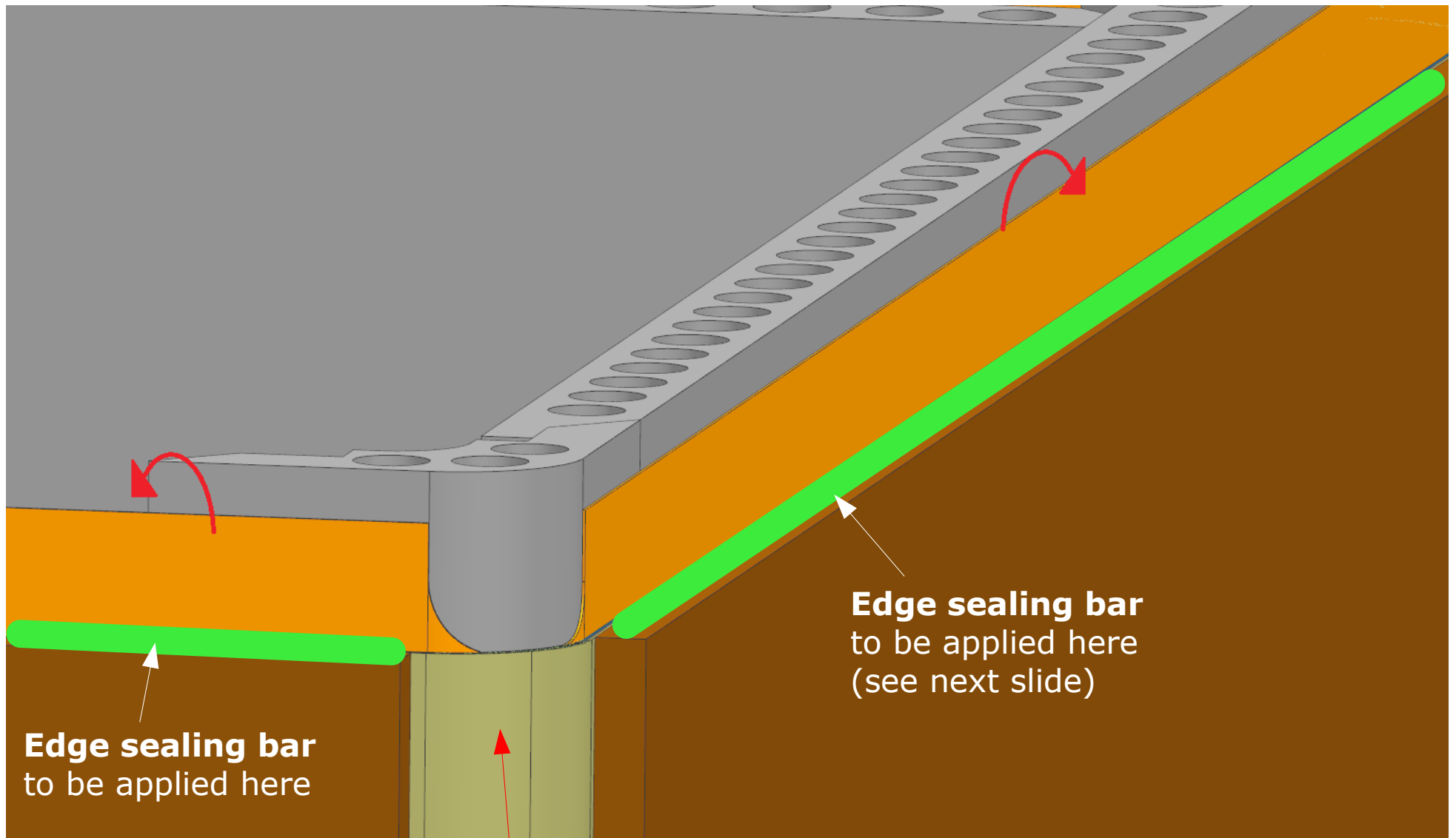
Note #3: Before subphases 5. and 6. (wrapping 5.CF and 6.outter_layer)

Corner structures and **Edge Sealing Bars** will be applied
(see next slides)

Corner structure



Phase II – the other layers are applied around the mold



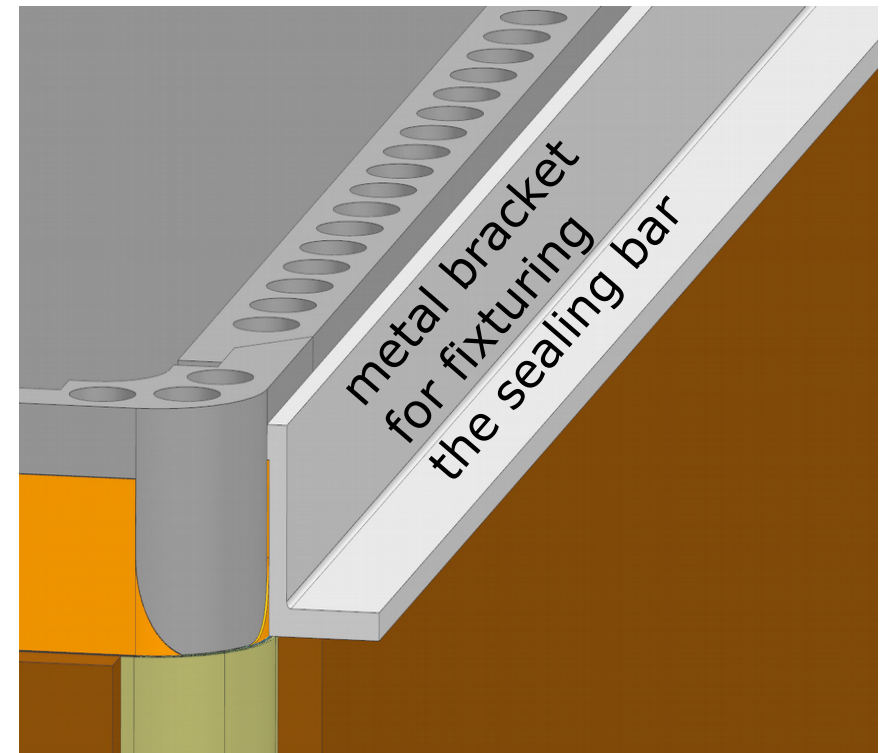
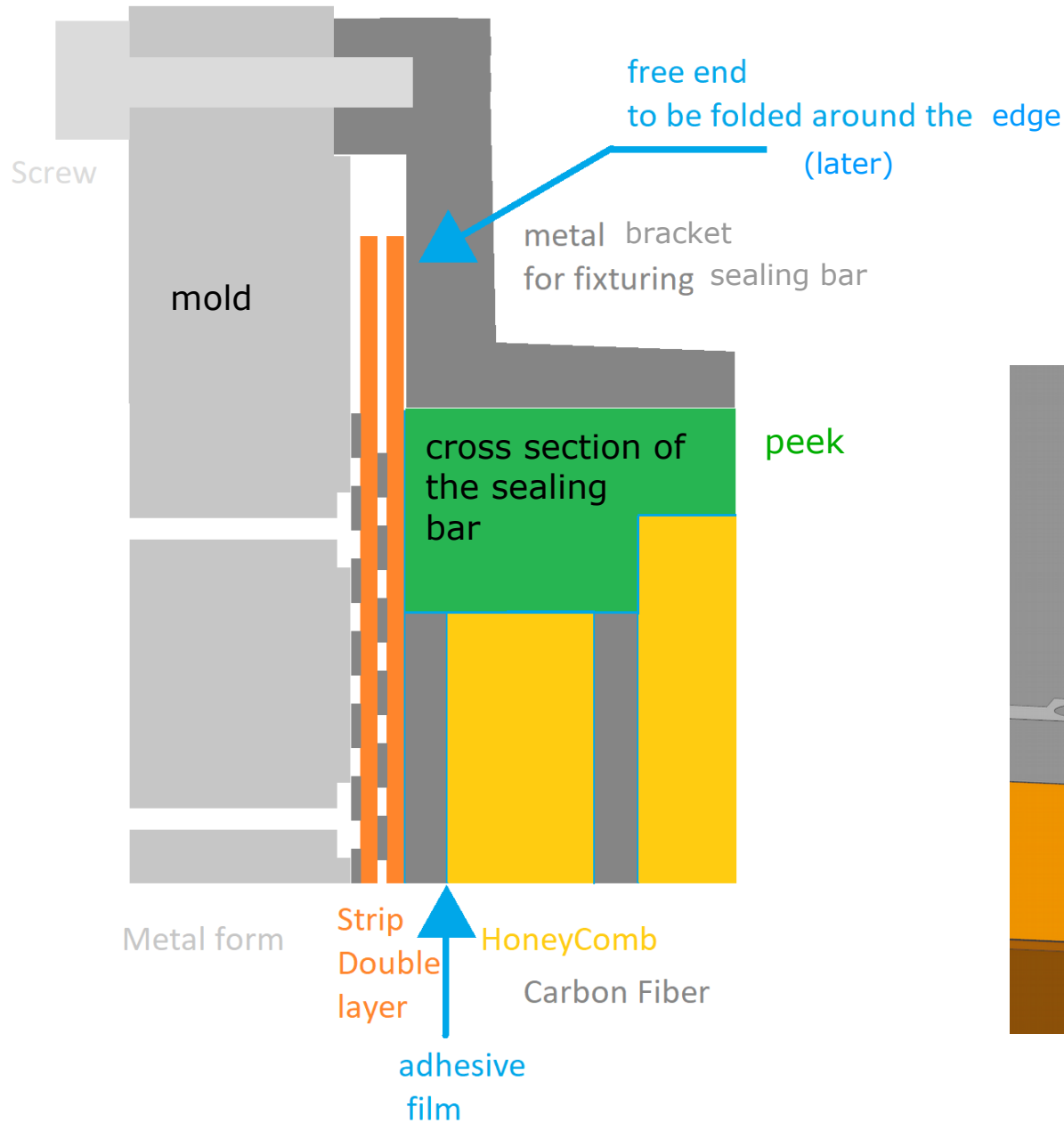
Edge sealing bar
to be applied here

Edge sealing bar
to be applied here
(see next slide)

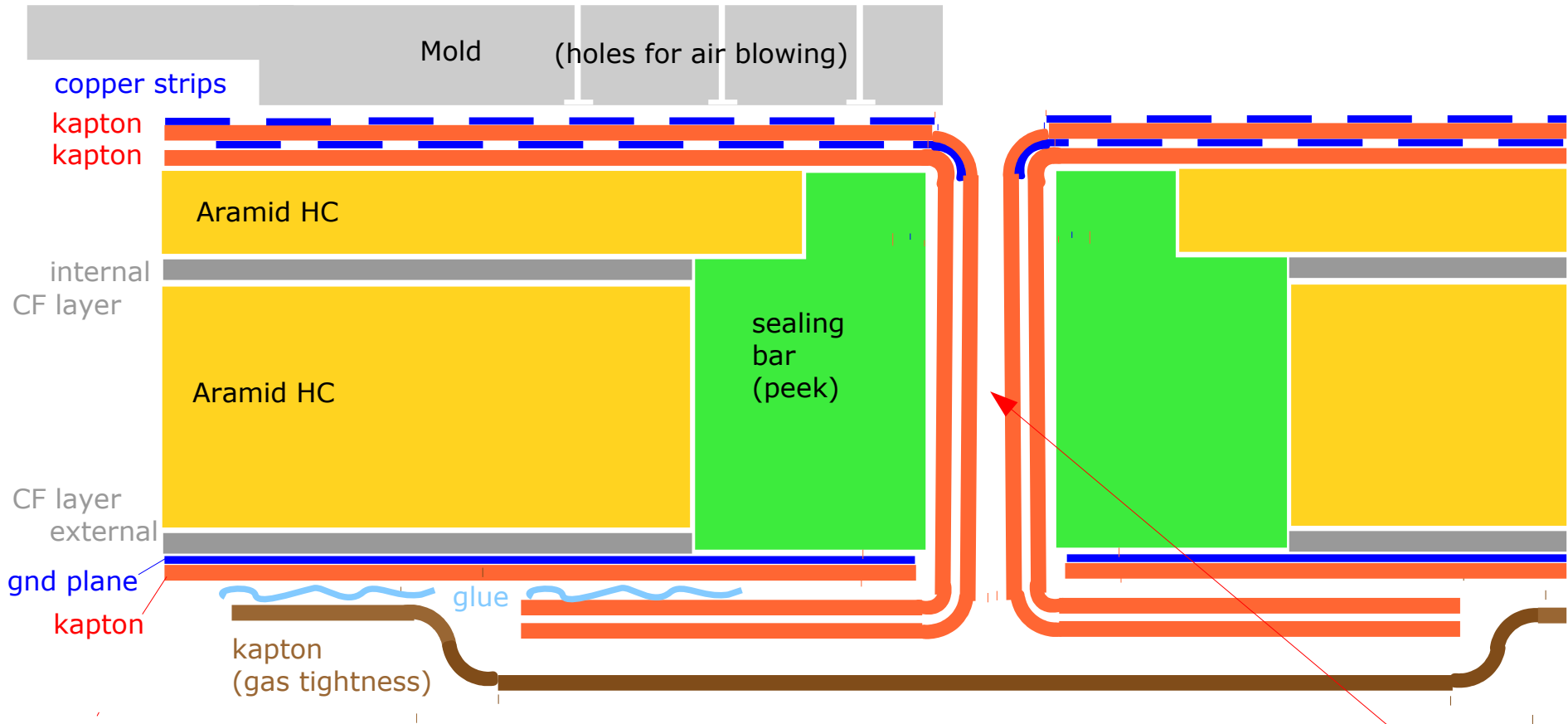
Corner structure
to be applied here

Phase III – completion of the composite layers stack including edge sealing bar

Vertical cross section of the stack @ box edge



Dielectric and gas sealing at box edges – Note #1



“soft” gas sealing (illustrated in picture) vs “hard” sealing (by gluing here)

To be discussed... tested

- “soft” sealing appears more convenient for various reasons (including it allows to rework the chamber
- if “soft” sealing OK → it might be applied also at Box-cathode and Box-end_Plate junctions !!!

Dielectric and gas sealing at box edges – “soft” sealing

cathode

anode
and MM

