

Experiences from the connector development for the current LHCb-Velo and AMS-Tracker with extreme high leak tightness requirements.

Workshop on pipe joining techniques for the ATLAS and CMS Tracker Upgrades

https://indico.cern.ch/event/721360/overview

18 May 2018

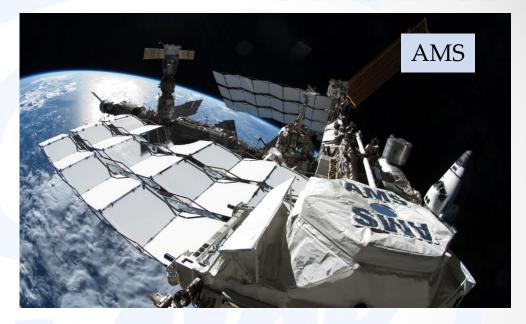
Bart Verlaat



CO₂ cooling developments for AMS and LHCb

LHCb-Velo





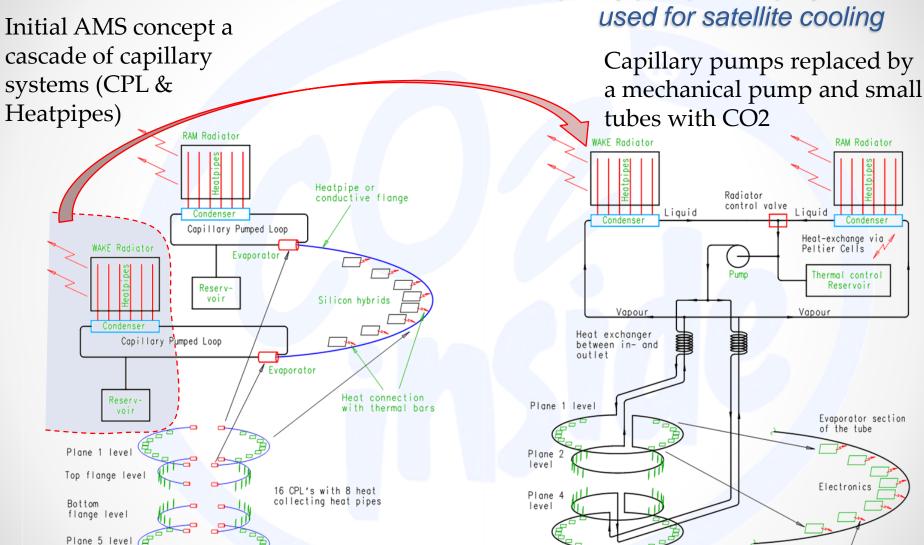
- CO₂ cooling was 1st considered in 1998 at Nikhef for the LHCb-Velo cooling to reduce the evaporator tube size.
- CO₂ cooling was adopted by AMS-Tracker in 2000 for the same reason
- For AMS a new circulation concept was developed based on capillary system used in satellite thermal control => 2PACL
- This 2PACL from AMS has become the baseline for all CO₂ systems at CERN so far.



The birth of the 2PACL cycle for AMS

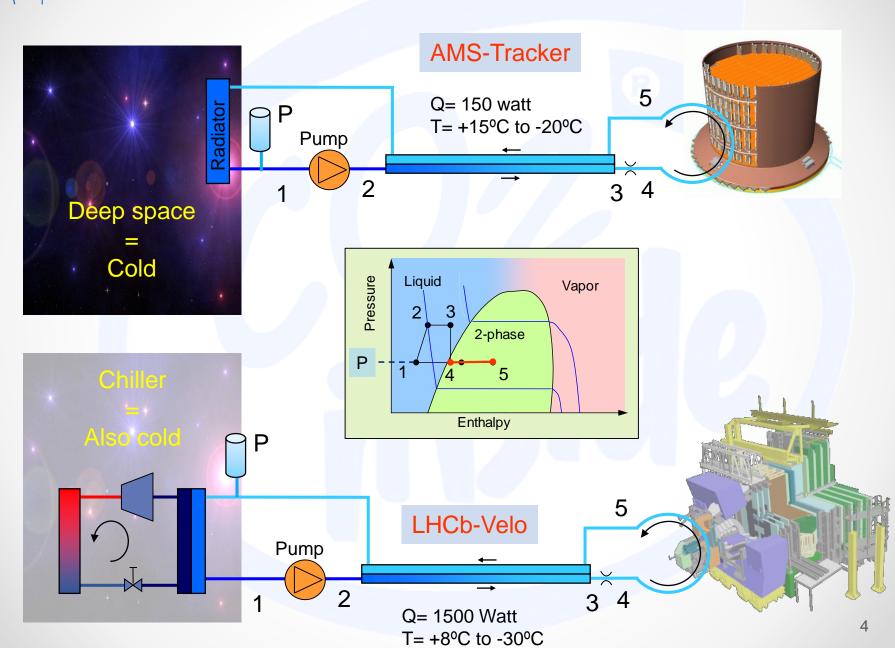
Plane 5 level

Derived from the capillary pumped loop system used for satellite cooling



Thermal bar connection

From AMS to LHCb





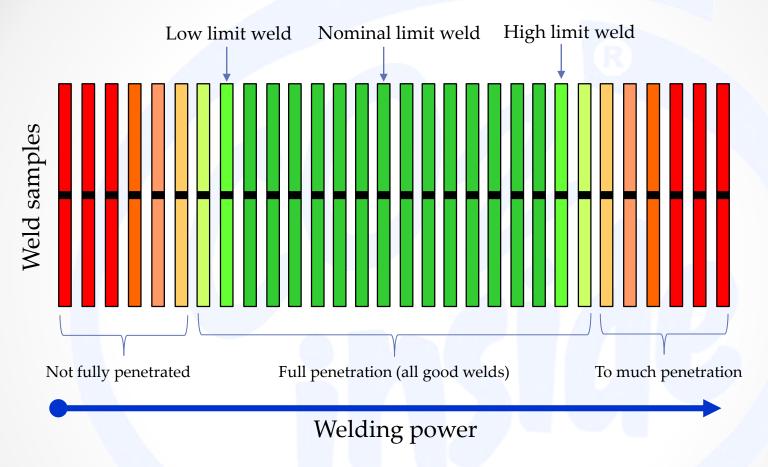
Requirements for tube and connection design

- Both AMS and LHCb CO₂ piping had to be designed for extreme circumstances:
 - Vacuum environment
 - In accessible
 - Vibration due to launch (AMS)
- Extreme leak tightness requirements
 - AMS is a small volume and no possibility to refill
 - LHCb inside the LHC vacuum system, a small leak would be catastrophic
- Only known and robust joining technologies were chosen
 - Orbital welding of standard 316L tubing
 - Capillary tubing: Vacuum brazing (LHCb) / Laser welding (AMS)
 - Vacuum brazing was not an option for AMS as the process requires more quality control at NASA than welding
 - Connectors: Swagelok VCR (LHCb) / Dynatube (AMS)
- We were obliged to use the NASA welding method and certification.
 - This method was easy and reliable and was also adapted for the construction of the LHCb-Velo cooling



NASA welding method

Selecting the right welding parameters for automatic welding



 The Low-Limit / High limit method is a good method to select the nominal welding parameters and assure that possible fluctuations do not bring the series welds out of spec.

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NASA welding method

Certification procedure

PRC-0010, Rev. A

Process Specification for Automatic and Machine Arc Welding of Steel and Nickel Alloy Flight Hardware

Engineering Directorate

Manufacturing, Materials, and Process Technology Division



July 1999



National Aeronautics and Space Administration

Lyndon B. Johnson Space Center Houston, Texas

Verify correct version before use.

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- Each welding type is pre-qualified with weld samples
 - 5 Low limit welds, 5 nominal welds and 5 high limit welds.
 - From each weld group the following tests are done to qualify the quality
 - All visual inspection (in and outside), all should look similar
 - 1 burst test (>4x design pressure)
 - 1 longitudinal cut and microscopic analyses
 - 3 samples for further research if needed
- Production of a series of welds
 - 3 high, 3 low and 3 nominal pre welds samples.
 - All process welds done at nominal settings
 - 3 high, 3 low and 3 nominal post welds samples.
 - o The pre and post weld samples are send to NASA for inspection
 - 1 burst, 1 microscopic and 1 spare for each type
 - All process welds are inspected visually and compared with post and pre weld samples.
 - Dye penetrant tests on process welds
- Due to the automatic welding and the similarity of all welds, no further complex inspection of the welds is needed
- Welding requirements are described in document PRC-0010, Rev A, attached to the indico

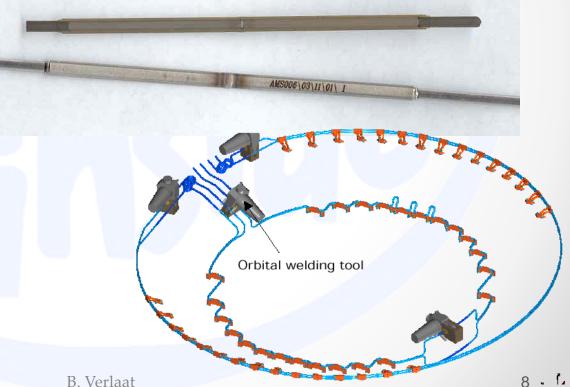




The evaporators were designed for welding

Supporting PQR no.(s) WELD 18 We	mpany / OrganizationTIC/AEI4
	elding Process(es) Automatic Orbital Tube Gas Tungsten Arc
BASE and FILLER METAL :	
Material number 45376832600 Group	to Material number 45376832600 Group
Material spec., type, and grade 14404 DIN17458 D4/T3	
Base metal thickness range	GAS:
	Torch/Head gas(es) ARGON 4.6
Filler metal F No. X AWS Class & Spec.	% Composition 4.6 Flow Rate 6.LPM
Consumable Insert, AWS Class & Spec	Prepurge Time 15 SEC Postpurge Time 15 SEC
	Backing gas(es) ARGON 4.6
WELDING SET-UP:	% Composition 4.6 Flow rate 1.77 MMWC
Power Supply (Model)SWAGELOK M100-2 \ 1728	Prepurge Time 15 SEC Postpurge Time 15 SEC
Weld Head(s)5H	
Joint Position(s) Tungsten type CWS,040-,705 Diameter Arc gap 0,64	PRE and POSTWELD HEAT
Tip diameter 1MM Tip angle	Preheat temperature minimum
Weld direction ROTATION HORZ Pulse Mode	Preheat temperatue maximum
1000000	Interpass temperature minimum
	Interpass temperature maximum
WELD SETTINGS :	Postweld HeatTreatment
Start current (amps) 13.3 Upslope (sec.)	
Level Slope Time (sec.) 10+10 Downslope (sec.) 5	JOINT DESIGN
Start Delay (sec.) 0,3 Override (%)	
Finish Current (amp) 8,6 Fixture Speed (RPM) 12,01	Joint type Radius Land
Weld Timer (on/off) Step Mode (on/off)	Root opening Size of fillet
Wire Mode (on/off)	Socket weld pull-back
	SETUP SKETCH -
NOMINAL HEAT INPUT CONDITIONS:	SEIOP SKEICH -
Weid Allowable Current (amps) Settings Pulse Pulse Level Time HIGH I LOW Rate Width	II .
Number (sec.) +5% Nominal -5% Nominal (pps) Nominal	.1. ♥
2	A
3	
4	
	€ II (100%
TECHNIQUE:	/- !!
	∠
Joint cleaning ETHANOL	
Joint cleaning ETHANOL Other	
Other	
	coordance with the requirements of NASA / JSC PRC-0010.
Other	

- Thin wall tubes in straight length were laser welded to a thick wall tube weld connector
 - 3mmx0.2 to 4mmx0.7 (both 2.6mm ID)
- After bending and assembly the sections were welded together with orbital welding.
- Tube layout designed for orbital weld head access.





AMS laser weld

Welds were done by the Shell Technology Centre in Amsterdam (Shell-STCA)

Laser welded with a Haas™ Neodinium-YAG pulsed Laser

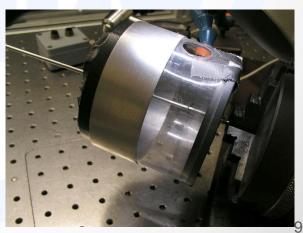
- Laser spot welds
- Argon backing gas
- No material addition













Laser weld pre-qualification

Laser Welds for AMS 02 TTCS Evaporator

Base material: 316L

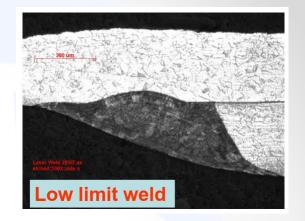
Specimen Traceability Numbers: ASM006\04\04\28\7,

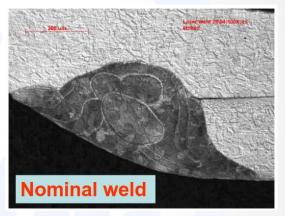
ASM006\04\04\28\4, ASM006\04\04\28\1

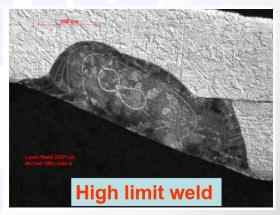
Three cross-sections were taken transverse to the laser weld, mounted, ground and polished. Then, the six weld locations on the cross-sections were electrolytically etched with 10% oxalic acid and examined on the metallograph.

The laser welds were judged to be sound and of high quality. No metallurgical anomalies were detected at magnifications up to and including 1000X. There was no notches, no lapped material, no pores, no voids, and no cracks detected. Penetration and fusion were proper for this type of socket weld.

John Figert Metallurgical Engineer, P.E. Lockheed Martin Space Operations









Connectors in AMS

Dynatube[®] Fittings

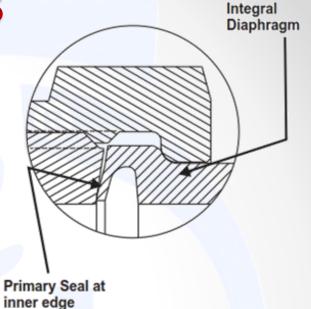


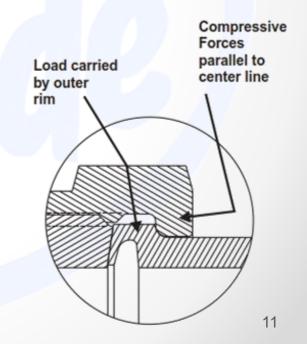
Material	Designation	Material Code Specification		Operating Temp.		
Titanium	6AL-4V STA	Т	AMS 4965	-65°F to 650°F		
Stainless Steel	17-4PH H1075	Р	AMS 5643	-110°F to 650°F		
Stainless Steel	15-5PH H1075	S	AMS 5659	-110°F to 650°F		
Inconel*	Alloy 718	N	AMS 5663	-320°F to 1300°F		

*Trademark



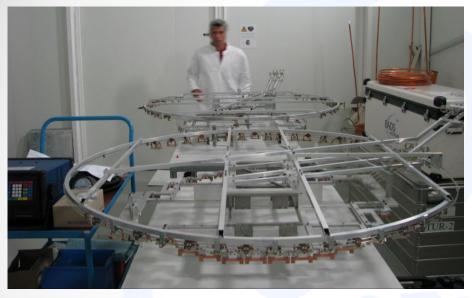
Parker Hannifin Corporation Stratoflex Products Division Jacksonville, Florida





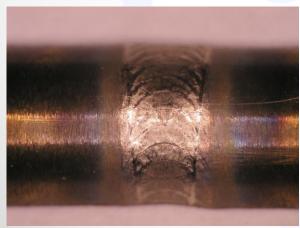


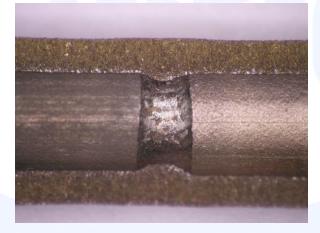
AMS Orbital welding



- CRES 316L 4x0.7 mm orbital weld
- Welded with a Swagelok/Cajon[™] orbital welding system
- Welding head: CajonTM CWS-5H-B
- Fixture block: CajonTM CWS-5TFB
- Collets: CajonTM CWS-5UCI-04mm



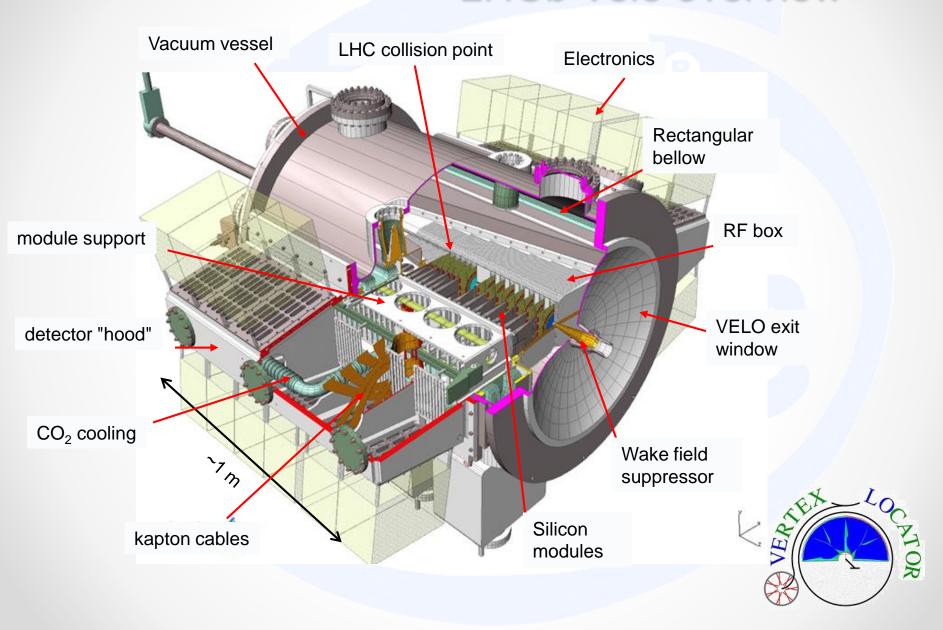


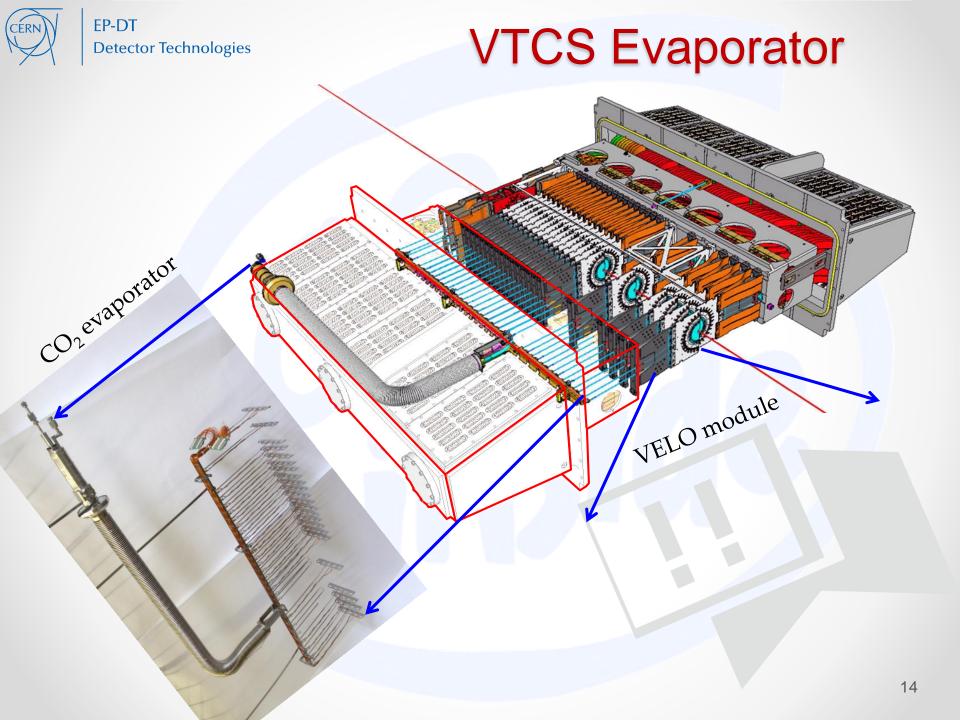






LHCb Velo overview



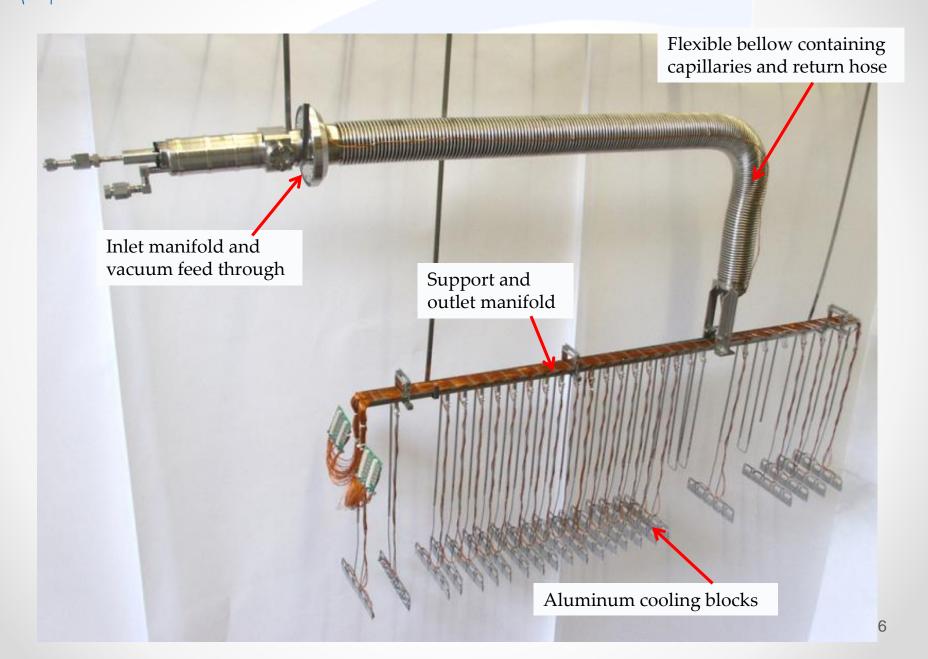




VTCS evaporator

- The VELO evaporator is inside the LHC vacuum system
- Only very reliable joining techniques allowed
- No connectors inside the vacuum system!
- Designed for orbital welding and vacuum brazing only
 - Vacuum brazing Nicrobraz EL-36
 - Swagelok Series 5, 4 and 8 orbital welding tools
 - Welding tool space and access taken into account in 3D model
 - Models of weld heads can be downloaded at:
 https://www.nikhef.nl/pub/departments/mt/projects/ams/SiTracker/ModelExchange/Orbitaal/
- Sub elements are brazed first and all orbital welded together
- The rest of the system (plant etc) made with orbital welding and VCR connectors (1/4" VCR and 1/4" HVCR)

VTCS evaporator





EP-DT Detector Technologies

LHCb Velo evaporator joint overview

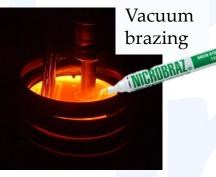
		Welding method	Number weldings per unit	Total number weldings	Part number	Material	Fixing weld	Support tool	Remarks
Α	Balg assembly								
A1	Ø4*0.7mm orbital weld connctors to Ø1x0.25mm capilairies (ca 2m)	Nicro-Braz EL-36	27	54	16-14	316Ti-321	Y	Υ	High frequency oven
A2	3/8"x0.035" return bellow flange	3/8" Orbital weld	2	4	08-10-13	316L-316L-316L	N	N	Orbital head 8 series incl. Fixing block
A3-1	Capilairies (A1) en 3/8"(A2) feedthrough in flange	Nicro-Braz EL-36	1	2	14-13-12	321-316L-316L	Y-12-13	Υ	High frequency oven
A3-2	Capilairies (A1) en 3/8"(A2) feedthrough in flange	Nicro-Braz EL-36	1	2	14-08-01	321-316L-316L	Y-01-08	Υ	High frequency oven
A4	Outer below to A3-1	TIG-welding	1	2	04-12	316L-316L	N	N	
A5	Base flange to A4	TIG-welding	1	2	03-04	316L-316L	N	N	
A6	Tube on base flange	TIG-welding	1	2	03-?	316L-316L	N	N	
A7	Llength determination								
A8	Heatshield to capillary flange	TIG-welding	1	2	01-02	316L-316L	N	N	
A9	Heatshield A8 to A6	TIG-welding	1	2	02-?	316L-316L	N	N	
A10	3/8" cajon connector with weld elbows to return	3/8" Orbital weld	3	6	06-07-08	316L-316L-316L	N	N	Orbital head 8 series incl. Fixing block
	Orbital			10					
	TIG			10					
	Nicro-Braz EL-36			58					
	Laser Fixing weld			56					
	Total number weldings			134					
В	Manifold assembly								
B1	27 Ø4mm pipes in ca 1m manifold with lid	Brazing	1	2					Vacuum oven
С	Cooling block assembly								
C1	2x Ø4*0.7mm orbital weld connections to Ø1.5x0.25mm capilairies	Brazen	27	54					High frequency oven
D	Total assembly								
D1	3/8" Return tube welding to manifold	3/8" Orbital weld	1	2					Retal of serie 8 weldhead with 3/8" fix block
D2	Spot weld of weld connectors (A1) to manifold (B)	TIG fixing	27	54					Hand weld
D3 = /o	Orbital weld of coolingblock (C) assemblies to manifold (B)		81	162					Retal of serie 4 weldhead with 4mm fix block

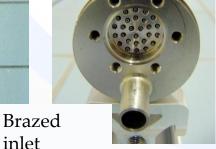
VTCS assembly

Flexible in and outlet below

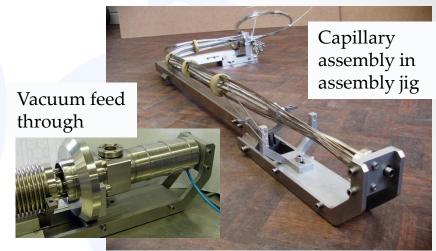


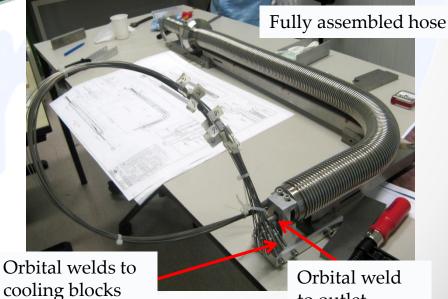






manifold



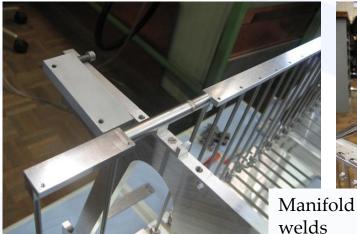




VTCS assembly Support manifold



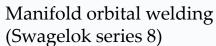














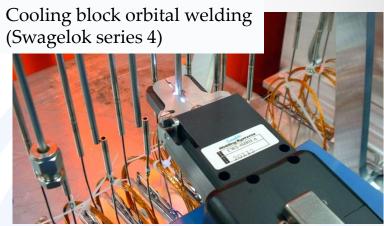


VTCS assembly Evaporator block assembly

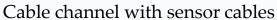




cooling block











Summary & Conclusions

- AMS and LHCB-VELO cooling was designed using professional joining technologies.
- No leaks have ever been observed in detector cooling lines
 - 7 years of development experience and 10 years of underground operation
- The NASA welding method is a helpful method also for non-space hardware
- Philosophy of making weld connectors for orbital welding to thin wall piping before production worked really well
- The LHCb Velo system will be taken out of service in LS2.
 Inspection of joints and pipes with 10 years of operation with CO₂ is foreseen to see the effects of long term exposure

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