







Development of a Cryogenic System for the FCC-hh Inner Triplets Cold Mass Cooling



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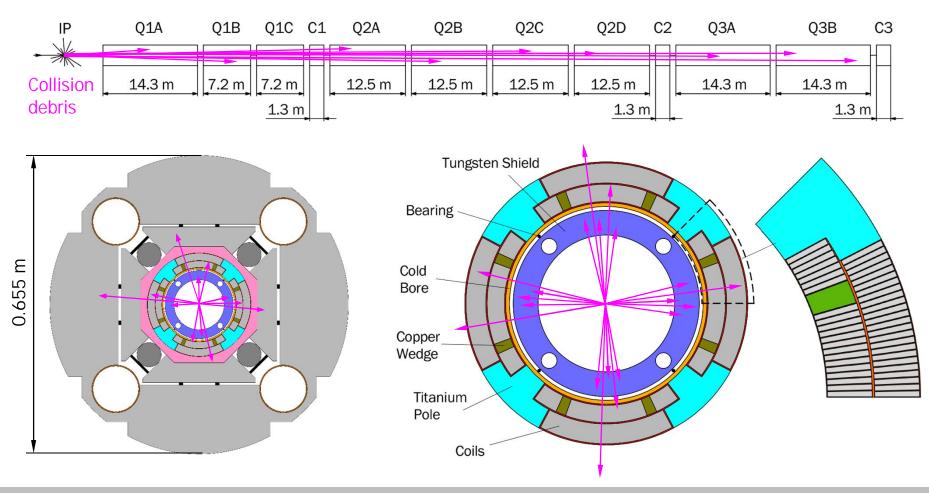
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### FCC Inner Triplets – Structure & Heat Source



### The FCC-hh Inner Triplets

Final focussing magnets before Interaction Points (8 quadrupoles in 3 groups)



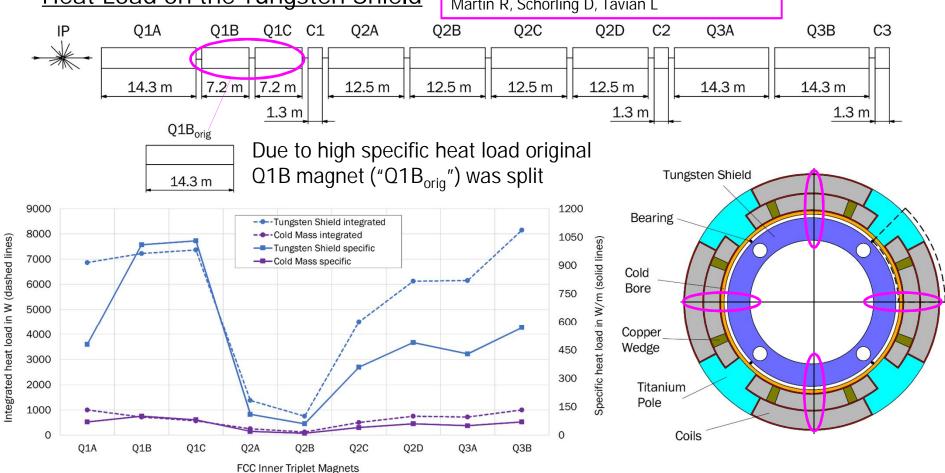
### FCC Inner Triplets – Heat Load on Tungsten Shield



### Heat Load on the Tungsten Shield

Rapid Reaction Team

Besana MI, Cerutti F, Delikaris D, Humann B, Martin R, Schörling D, Tavian L



- Large cooling channels in tungsten shield possible (due to impact distribution)
- Tungsten Shield cooling between 40 60 K (compared to 1.8 K of Cold Mass)

### FCC Inner Triplets – Cold Mass Cryogenics



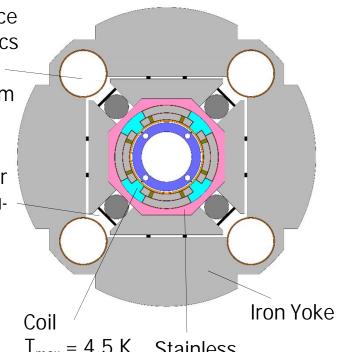
### Heat Load on the Cold Mass

- 1. Available space for cryogenics installations is limited
- 2. Available driving temperature range for heat extraction is limited
- 3. The range of specific heat loads varies strongly for the single magnets
- → Small, reliable and possibly uniformly designed cryogenic system needed

Available space for cryogenics installations  $d_{max} \approx 100 \text{ mm}$ 

Free space for pressurized superfluid helium

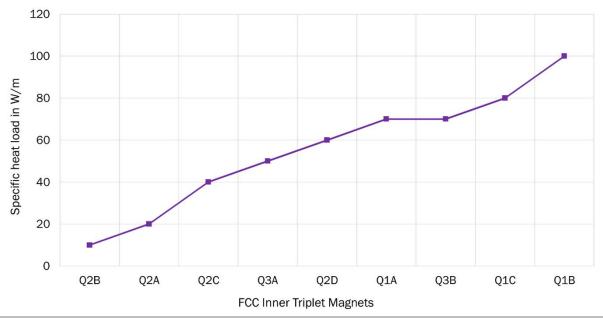
 $T_{max} = 1.9 K$ 



 $T_{max} = 4.5 \text{ K}$  Stainless Steel Collar

Two approved cooling concepts with superfluid He:

- 1. Two-phase cooling with bayonet heat exchanger pipe
- 2. (Pure) Conduction cooling



### FCC Inner Triplets – Two-phase cooling



The two-phase cooling scheme

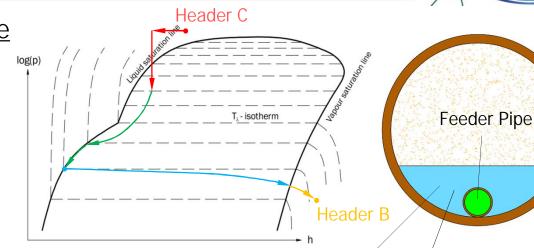
Helium supplied by header C:

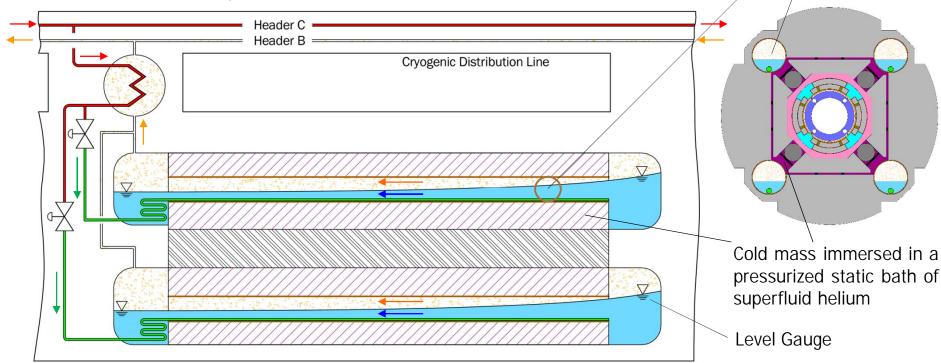
$$T_{\rm C} = 4.6 \, \text{K}$$
  $p_{\rm C} = 3 \, \text{bar}$ 

Helium discharged into header B:

$$p_B = 16 \text{ mbar}$$

→ Determines stratified flow temperature (= T | p<sub>Sat</sub>=17 mbar)



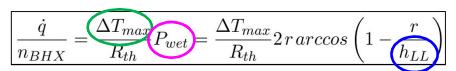


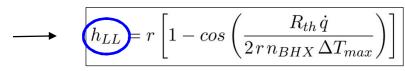
### Two-phase cooling - The two main constraints



### **Heat extraction**

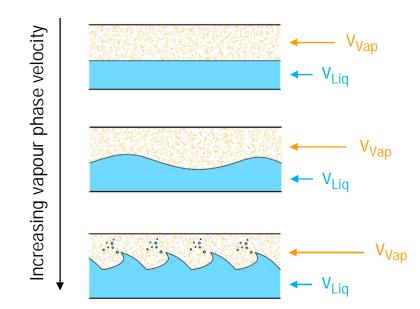
# $A_{Vap}$ Bad heat transfer via vapour phase $\begin{array}{|c|c|} \hline \textbf{DT}_{\text{max}} & \hline & T_{\text{max}} = 1.9 \text{ K} \\ \hline & T_{\text{2F}} \approx 1.81 \text{ K} \\ \hline \end{array}$ $h_{II}$ Good heat transfer via liquid phase





Vapour velocity

Relative movement between liquid and vapour phase determines shape of the surface:



Minimal for sufficient heat transfer

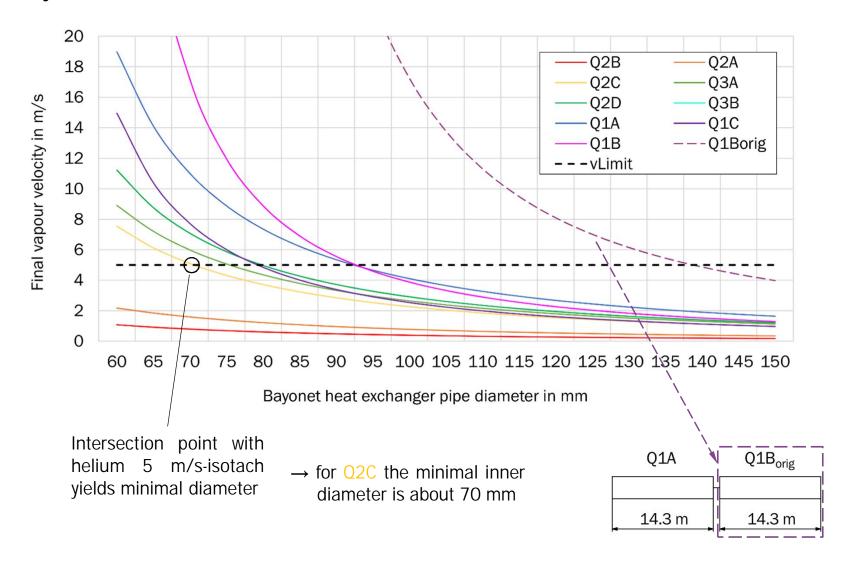
Maximal V<sub>Vap</sub> determined in experimental setup

$$v_{Vap} = \frac{\dot{m}}{\rho_{Vap} A_{Vap}} = \frac{\dot{q}L}{n_{BHX} \rho_{Vap} \left(h'' - h_{JT}\right) \left[r^2 \arccos\left(1 - \frac{h_{LL}}{r}\right) - \left(r - h_{LL}\right) \sqrt{2rh_{LL} - h_{LL}^2}\right]} \leq 5 \, m/s$$

### Two-phase cooling – Analytical estimations



### **Analytical estimation**



### Two-phase cooling – Mathematical Model



### **Steady-State Calculation**

#### Conditions:

- I. Longitudinal pressure drop in the two phases equal
- II. Solving for minimal HX pipe diameter
- III. Smooth stratified flow regime in HX pipe

#### Modelling 1D-equations:

Mass balance: 1)  $\pm \dot{m} \frac{d\xi}{dz} = \frac{d}{dz} \left( A_{\Phi} \rho_{\Phi} v_{\Phi} \right)$ 

Momentum balance:

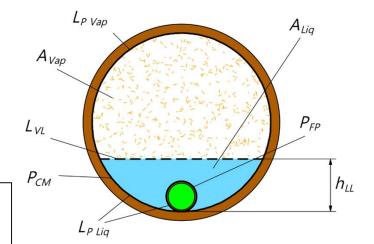
2) 
$$\frac{d}{dz} \left( A_{\Phi} \rho_{\Phi} v_{\Phi}^2 \right) = -A_{\Phi} \frac{dp_{IF}}{dz} \pm$$

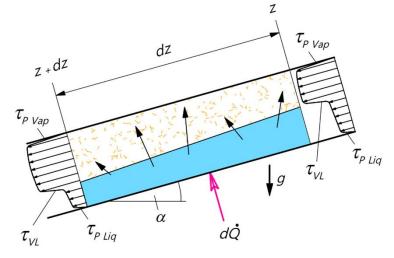
$$\pm \tau_{VL} L_{VL} - \tau_{P\Phi} L_{P\Phi} + g A_{\Phi} \rho_{\Phi} \left( sin\alpha \pm cos\alpha \frac{dh_{LL}}{dz} \right) - \dot{m} \frac{d\xi}{dz} \Delta v_{VL}$$

Thermal energy 3)  $\left(h'' - h'\right) d\xi = \frac{\dot{q}_{CM} P_{CM} + \dot{q}_{FP} P_{FP}}{\dot{m}} dz$  balance:

Thermal conduction in superfluid helium:

$$4) \frac{dT}{dz} = -f_k \dot{q}^m$$

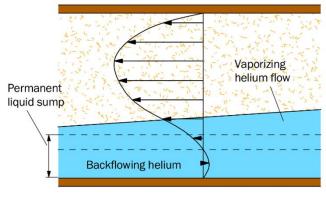


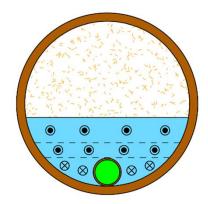


### Two-phase cooling – Numerical results



### **Numerical solutions**

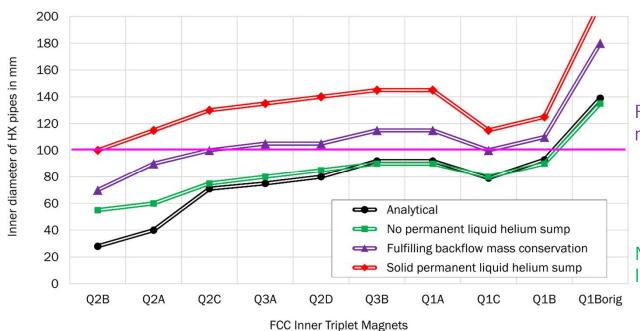




Comparisons with former experimental results indicate an overestimation of the wave generation of the model ( $v_{Vap} < 5m/s$ )

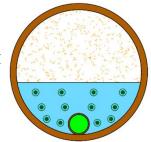
→ The requested maximal diameter of 100 mm seems to be feasible

Solid permanent liquid helium sump



Fullfilling backflow mass conservation





### FCC Inner Triplets – Conduction Cooling



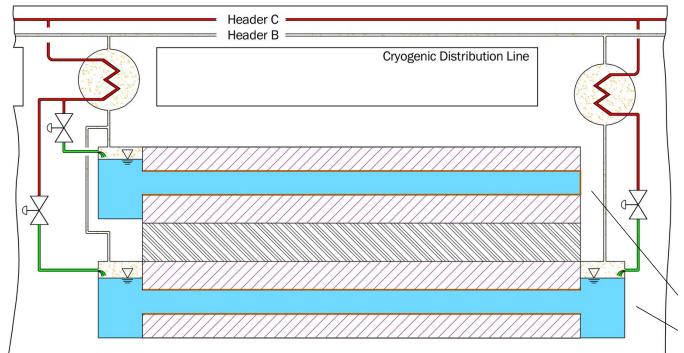
### The conduction scheme

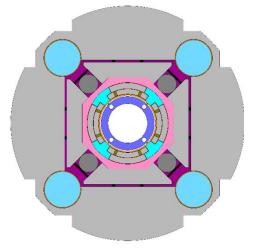
$$\Delta T_{max} = \int_{0}^{L_{Mag}} \frac{(\dot{q}x)^{m} f_{k}}{A^{m}} dx = \frac{\dot{q}^{m} L_{Mag}^{m+1} f_{k}}{A^{m} (m+1)}$$

Needed cross section area for heat conduction  $\alpha$  L<sup>1.3</sup>  $\searrow$ 

→ Conduction cooling is advantegeous for short magnets (like in the Inner Triplets)

$$A^{m} = \frac{\dot{q}^{m} L_{Mag}^{m+1} f_{k}}{\Delta T_{max} (m+1)} \to d = \sqrt{\frac{4 \dot{q}}{\pi n_{BHX}}} \sqrt[m]{\frac{L_{Mag}^{m+1} f_{k}}{\Delta T_{max} (m+1)}} \propto L_{Mag}^{\frac{m+1}{2m}} \sqrt{\dot{q}} \approx L_{Mag}^{0.65} e^{0.5}$$





### Two possible designs:

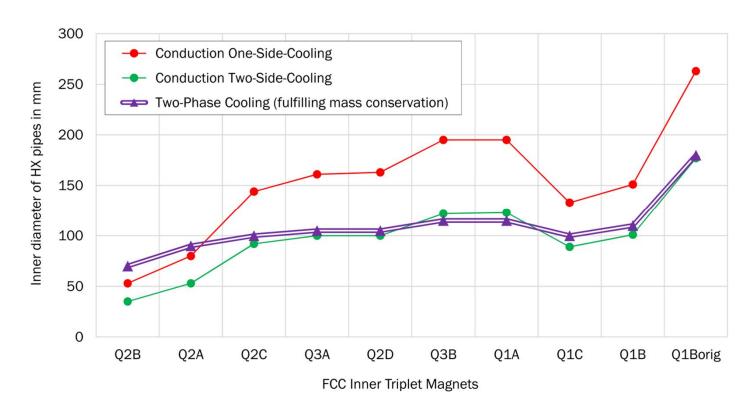
Option 1: One-Side-Cooling

Option 2: Two-Sides-Cooling

### Conduction cooling – Numerical results



### **Numerical solutions**



- The Two-Side Conduction cooling option and the Two-Phase cooling need similar space in the cold mass for cryogenics - both concepts seem to be in the feasibility's range
- The choice could be made by different aspects (required space between adjacent magnets, transient behavior, controlling effort, reliability, ...)

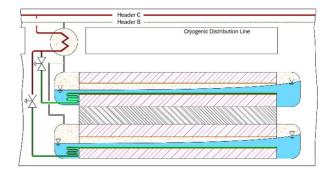
### Summary & Outlook



### **Summary**

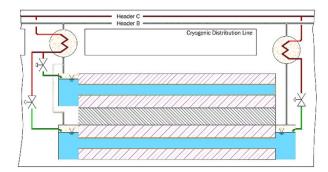
- Very high, but strongly non-uniform heat loads on the FCC Inner Triplet Magnets challenge the cryogenics design (structure was changed to be able to design a reliable cooling system for the available space)
- Two well-established cooling concepts were investigated

#### Two-Phase Flow Cooling



+ Less space needed for cryogenics

#### Pure Conduction Cooling



- + Robust concept and simple controlling
- With both cooling concepts the space requirements are in the range of feasibility the
  possibility of choosing between different designs provides freedom of choice for taking
  into account other aspects as well

### The end









Thank you very much for your attention!

"What starts out as science fiction today may wind up being finished tomorrow as a report."

PULL STUDEN STUD

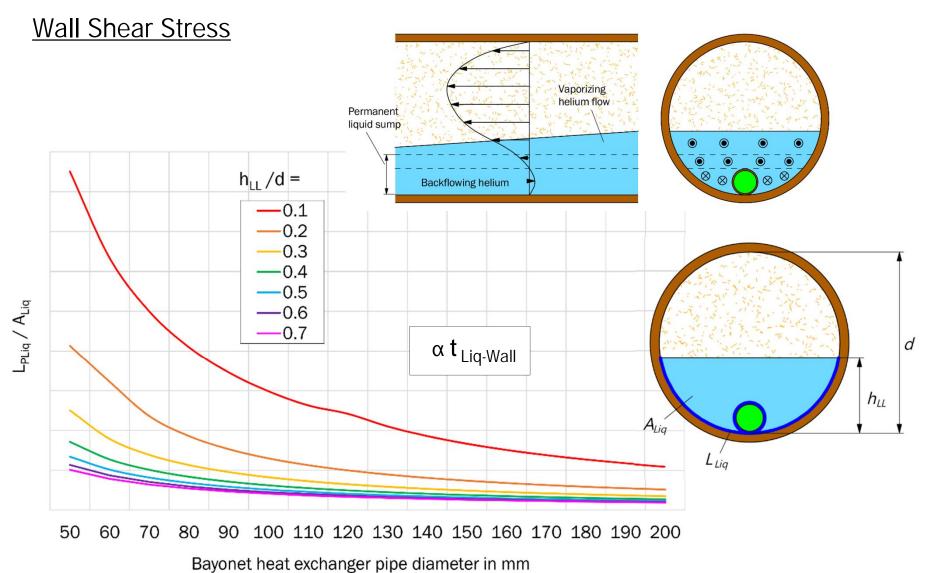
Norman Mailer



http://fccweek2019.web.cern.ch,

## Supplemental – Wall Shear Stress (Liquid)



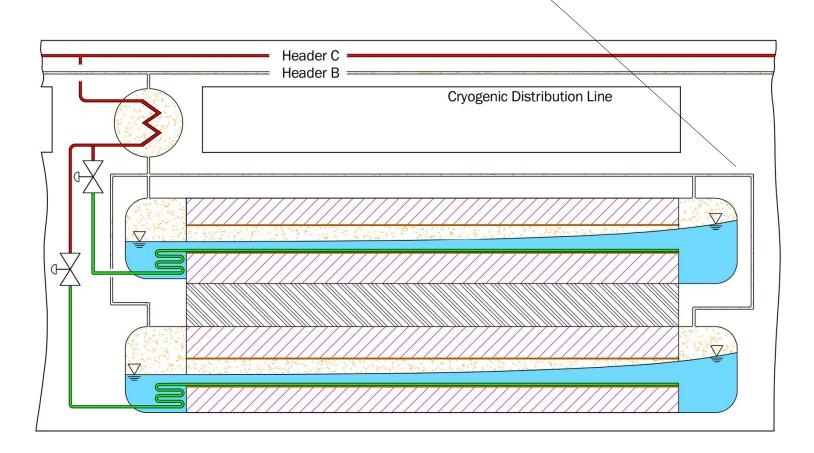


### Supplemental – Two-Sides vapour extraction



### Two-Sides vapour extraction

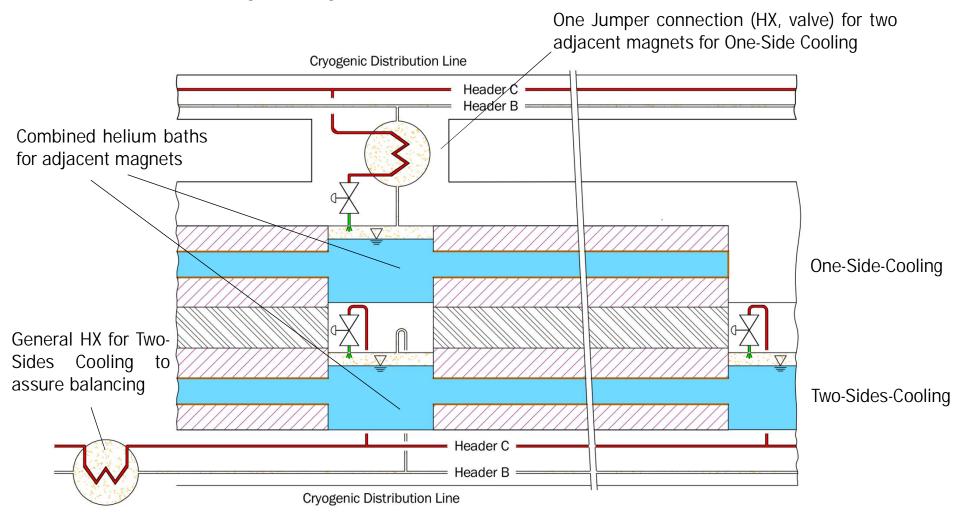
Two-Sides vapour extraction could decrease bayonet HX diameter size requirement



### Supplemental – Conduction Cooling Interconnections



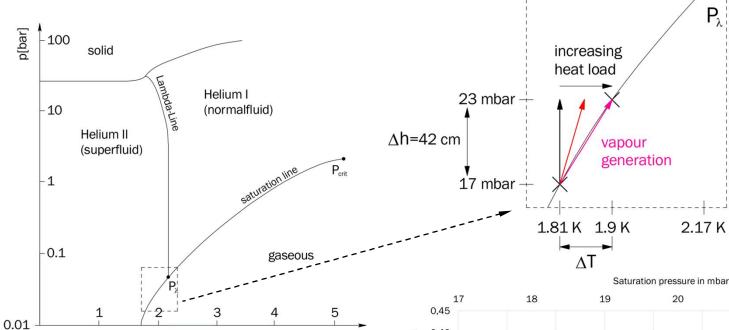
### **Conduction Cooling Arrangements**



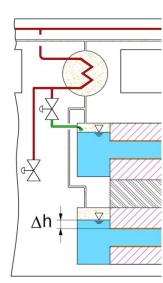
### Supplemental – Conduction Cooling Vapour Generation (1/2)



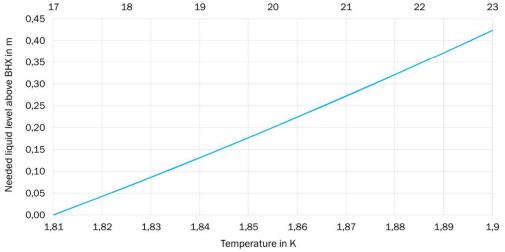
Conduction Cooling Vapour Generation (1/2)



T[K]



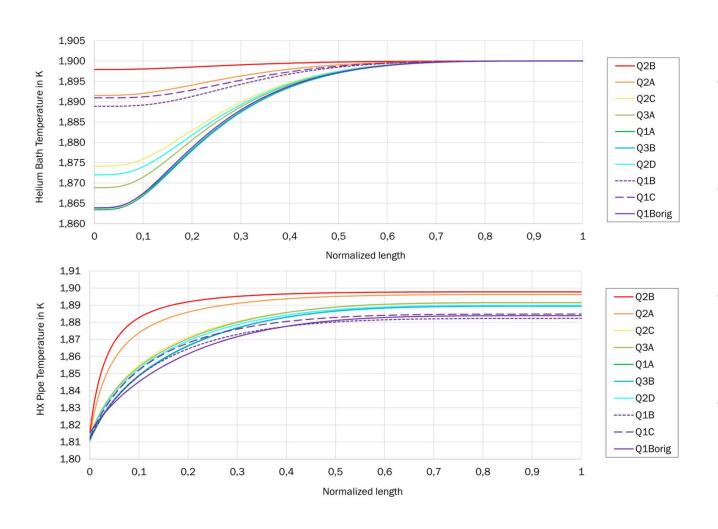
- $\rho_{Liq} = 145 \text{ kg/m}^3$
- HX pipe diameters calculated to reach a temperature of  $\approx 1.9 \text{ K}$
- Minimal pressure head needed to avoid saturation line (≈ 42 cm)



### Supplemental – Conduction Cooling Vapour Generation (2/2)



### Conduction Cooling Vapour Generation (2/2)



- Liquid helium temperature in HX pipe
   < 1.9 K for radial heat transfer
- Two parallel longitudinal heat fluxes (in HX pipe and static helium bath)
- Radial driving temperature difference increases towards outlet
- Limit determined by the magnets Q1A and Q3B (T = 1.887 K → 30 – 40 cm head)

### Supplemental-Mixed Cooling Concepts



### **Mixed Cooling**

High-loaded magnets: Two-Sides Conduction Cooling Low-loaded magnets: Combined bayonet heat exchanger cooling

High-loaded magnets: Two-Sides Conduction Cooling

