

*Linear*

*Imperfections*

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# *Linear Imperfections*

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■ equation of motion in an accelerator

→ Hills equation

→ sine and cosine like solutions

→ closed orbit

→ sources for closed orbit perturbations

■ dipole perturbations

→ closed orbit response

→ dispersion orbit

→ integer resonances

→ BPMs & dipole correctors

■ quadrupole perturbations

→ one-turn map & tune error

→ beta-beat

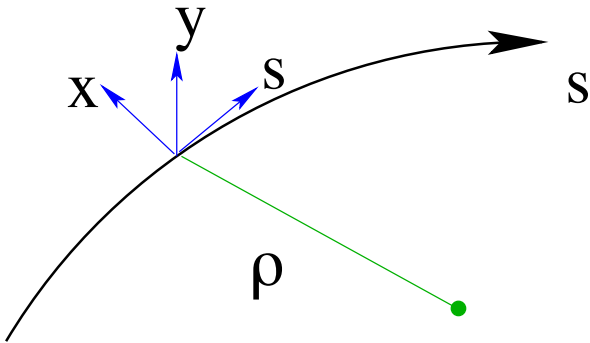
→ half-integer resonances

■ orbit correction

→ local orbit bumps

# Variable Definition

## Variables in moving coordinate system:



$$x' = \frac{d}{ds} x$$

$$\frac{d}{dt} = \frac{ds}{dt} \cdot \frac{d}{ds} \rightarrow x' = \frac{P_x}{p_0}$$

$\swarrow$   $v$

## Hill's Equation:

$$\frac{d^2 x}{ds^2} + K(s) \cdot x = 0; \quad K(s) = K(s + L);$$

$$K(s) = \begin{cases} 0 & \text{drift} \\ 1/\rho^2 & \text{dipole} \\ 0.3 \cdot \frac{B[\text{T/m}]}{p[\text{GeV}]} & \text{quadrupole} \end{cases}$$

## Perturbations:

$$\frac{d^2 x}{ds^2} + K(s) \cdot x = G(s); \quad G(s) = \frac{F(s)_{\text{Lorentz}}}{v \cdot p_0}$$

# *Sinelike and Cosinelike Solutions*

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■ system of first order linear differential equations:

$$\vec{y} = \begin{pmatrix} x \\ x' \end{pmatrix} \longrightarrow \vec{y}' + \begin{pmatrix} 0 & -1 \\ K & 0 \end{pmatrix} \cdot \vec{y} = 0$$

$$K = \text{const} \longrightarrow$$

$$\vec{Y}_1(s) = \begin{pmatrix} \sin(\sqrt{K} \cdot s) \\ \sqrt{K} \cdot \cos(\sqrt{K} \cdot s) \end{pmatrix} \quad \vec{Y}_2(s) = \begin{pmatrix} \cos(\sqrt{K} \cdot s) \\ -\sqrt{K} \cdot \sin(\sqrt{K} \cdot s) \end{pmatrix}$$

■ initial conditions:

$$\vec{Y}_1(0) = \begin{pmatrix} Y_1 \\ Y_1' \end{pmatrix} = \begin{pmatrix} 0 \\ 1 \end{pmatrix} \quad \text{and} \quad \vec{Y}_2(0) = \begin{pmatrix} Y_2 \\ Y_2' \end{pmatrix} = \begin{pmatrix} 1 \\ 0 \end{pmatrix}$$

■ general solution:  $\vec{y}(s) = a \cdot \vec{Y}_1(s) + b \cdot \vec{Y}_2(s)$

■ transport map:  $\vec{y}(s) = \underline{M}(s - s_0) \cdot \vec{y}(s_0)$

with: 
$$= \begin{pmatrix} \cos(\sqrt{K} \cdot [s-s_0]) & \sin(\sqrt{K} \cdot [s-s_0]) \\ -\sqrt{K} \cdot \sin(\sqrt{K} \cdot [s-s_0]) & \sqrt{K} \cdot \cos(\sqrt{K} \cdot [s-s_0]) \end{pmatrix}$$

# *Sineline and Cosineline Solutions*

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Floquet theorem:

$$\vec{Y}_1(s) = \begin{pmatrix} \sqrt{\beta(s)} \cdot \sin(\phi(s) + \phi_0) \\ [\cos(\phi(s) + \phi_0) + \alpha(s) \cdot \sin(\phi(s) + \phi_0)] / \sqrt{\beta(s)} \end{pmatrix}$$

$$\vec{Y}_2(s) = \begin{pmatrix} \sqrt{\beta(s)} \cdot \cos(\phi(s) + \phi_0) \\ -[\sin(\phi(s) + \phi_0) + \alpha(s) \cdot \cos(\phi(s) + \phi_0)] / \sqrt{\beta(s)} \end{pmatrix}$$

$$\beta(s) = \beta(s + L); \quad \phi(s) = \int \frac{1}{\beta} ds; \quad \alpha(s) = -\frac{1}{2} \beta'(s)$$

'sineline' and 'cosineline' solutions:

$$\vec{C}(s) = a \cdot \vec{Y}_1(s) + b \cdot \vec{Y}_2(s) \quad \vec{S}(s) = c \cdot \vec{Y}_1(s) + d \cdot \vec{Y}_2(s)$$

$$\text{with: } \vec{C}(s_0) = \begin{pmatrix} C(s_0) \\ C'(s_0) \end{pmatrix} = \begin{pmatrix} 1 \\ 0 \end{pmatrix} \quad \text{and} \quad \vec{S}(s_0) = \begin{pmatrix} S(s_0) \\ S'(s_0) \end{pmatrix} = \begin{pmatrix} 0 \\ 1 \end{pmatrix}$$

→ one can generate a transport matrix in analogy to the case with constant  $K(s)$ !

# Sineline and Cosineline Solutions

█ 'sineline' and 'cosineline' solutions:

$$\vec{S}(s) = \begin{pmatrix} \sqrt{\beta(s)\beta(s_0)} \cdot \sin(\phi(s) + \phi_0) \\ \sqrt{\beta(s_0)} \cdot [\cos(\phi(s) + \phi_0) + \alpha(s) \cdot \sin(\phi(s) + \phi_0)] / \sqrt{\beta(s)} \end{pmatrix}$$

$$\vec{C}(s) = \begin{pmatrix} \sqrt{\beta(s)} \cdot [\cos(\phi(s) + \phi_0) + \alpha(s_0) \cdot \sin(\phi(s) + \phi_0)] / \sqrt{\beta(s_0)} \\ -(1 + \alpha\alpha_0) \cdot [\sin(\phi(s) + \phi_0) + (\alpha_0 - \alpha) \cdot \cos(\phi(s) + \phi_0)] / \sqrt{\beta\beta_0} \end{pmatrix}$$

█ transport map from  $s_0$  to  $s$ :  $\vec{y}(s) = \underline{M}(s, s_0) \cdot \vec{y}(s_0)$

with: 
$$\underline{M} = \begin{pmatrix} C(s) & S(s) \\ C'(s) & S'(s) \end{pmatrix}$$

█ transport map for  $s = s_0 + L$ :

$$\underline{M} = \underline{I} \cdot \cos(2\pi Q) + \underline{J} \cdot \sin(2\pi Q)$$

$$\underline{I} = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} \quad \underline{J} = \begin{pmatrix} \alpha & \beta \\ -\gamma & -\alpha \end{pmatrix}; \quad \gamma = [1 + \alpha^2] / \beta$$



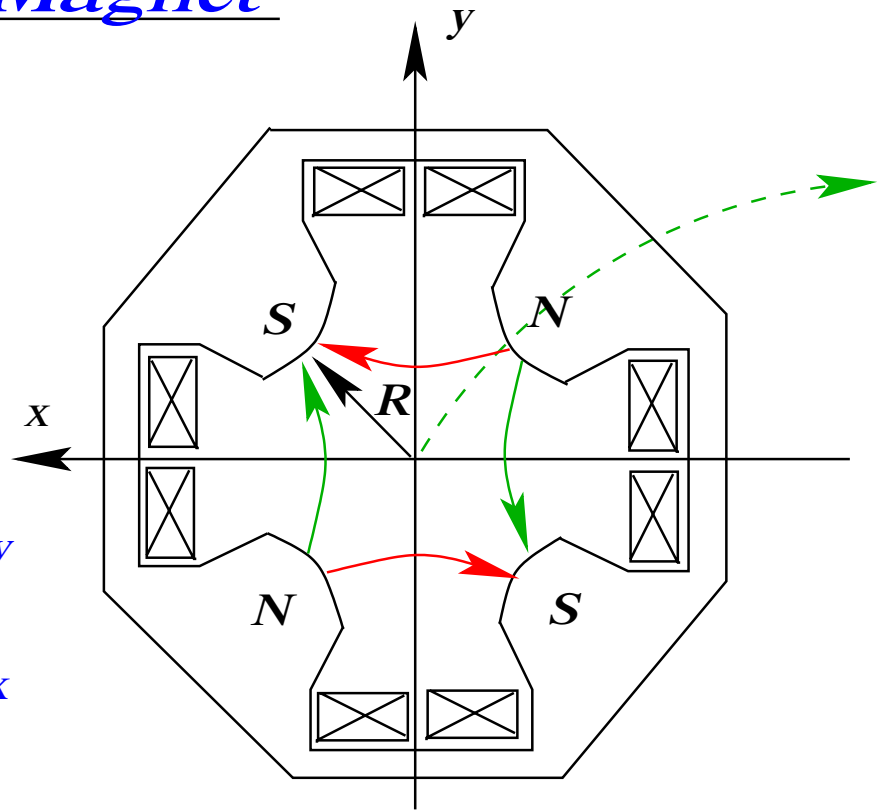
## ● Quadrupole Magnet

$$B_x = g \cdot y$$

$$B_y = g \cdot x$$

$$F_x = -q \cdot v \cdot B_y$$

$$F_y = q \cdot v \cdot B_x$$



$$\frac{d^2 x}{d s^2} + K(s) \cdot x = G(s); \quad G(s) = \frac{F(s)_{\text{Lorentz}}}{v \cdot p_0}$$

## ● normalized fields:

→ dipole:  $k_0(s) = 0.3 \cdot \frac{B_0 [\text{T}]}{p_0 [\text{GeV}]}$

quadrupole:  $k_1(s) = 0.3 \cdot \frac{g_0 [\text{T/m}]}{p_0 [\text{GeV}]}$

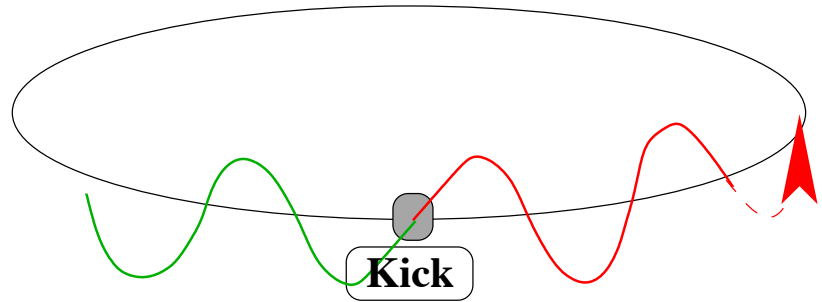
quadrupole misalignment:  $\Delta k_0(s) = 0.3 \cdot \frac{g [\text{T/m}]}{p [\text{GeV}]} \cdot x_0$



# Dipole Error and Orbit Stability

● Q: number of  $\beta$ -oscillations per turn

■ Q = N

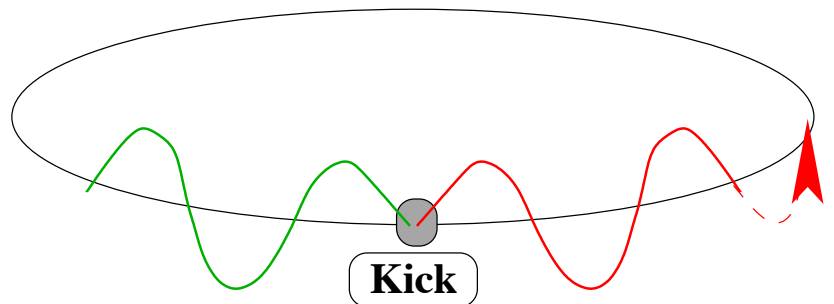


→ the perturbation adds up

→ amplitude growth and particle loss

↗ watch out for integer tunes!

■ Q = N + 0.5



→ the perturbation cancels after each turn

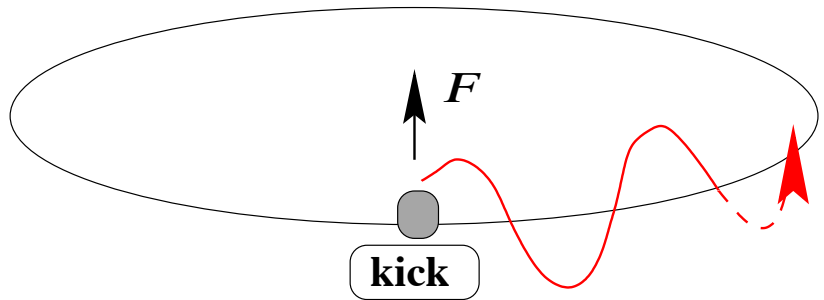
# Quadrupole Error and Orbit Stability

## ● Quadrupole Error:

→ orbit kick proportional to  
beam offset in quadrupole

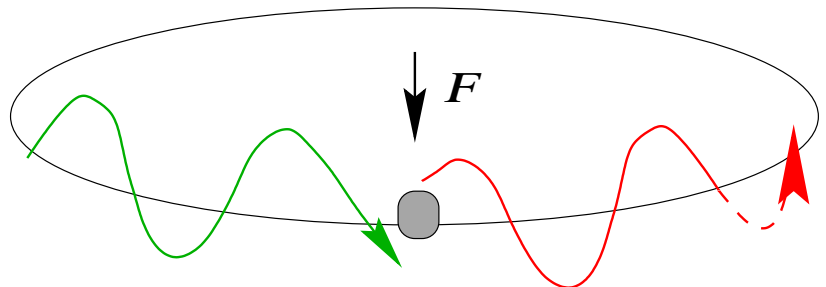
■  $Q = N + 0.5$

1. Turn:  $x > 0$



→ amplitude increase

2. Turn:  $x < 0$



→ amplitude increase

↘ watch out for half integer tunes!

# Sources for Orbit Errors

## ● *Quadrupole offset:*

■ *alignment*    *+/- 0.1 mm*

■ *ground motion*

■ *slow drift*

■ *civilisation*

■ *moon*

■ *seasons*

■ *civil engineering*

## ● *Error in dipole strength*

■ *power supplies*

■ *calibration*

## ● *Energy error of particles*

■ *injection energy (RF off)*

■ *RF frequency*

■ *momentum distribution*

# Example Quadrupole Alignment in LEP

Transversal tilt dispersion of the 3278 dipoles

$$\sigma = \pm 0.34 \text{ mrd}$$

Vertical dispersion of the 784 quadrupoles  
(with respect to the smoothing polynomial)

$$\sigma = \pm 0.65 \text{ mm}$$

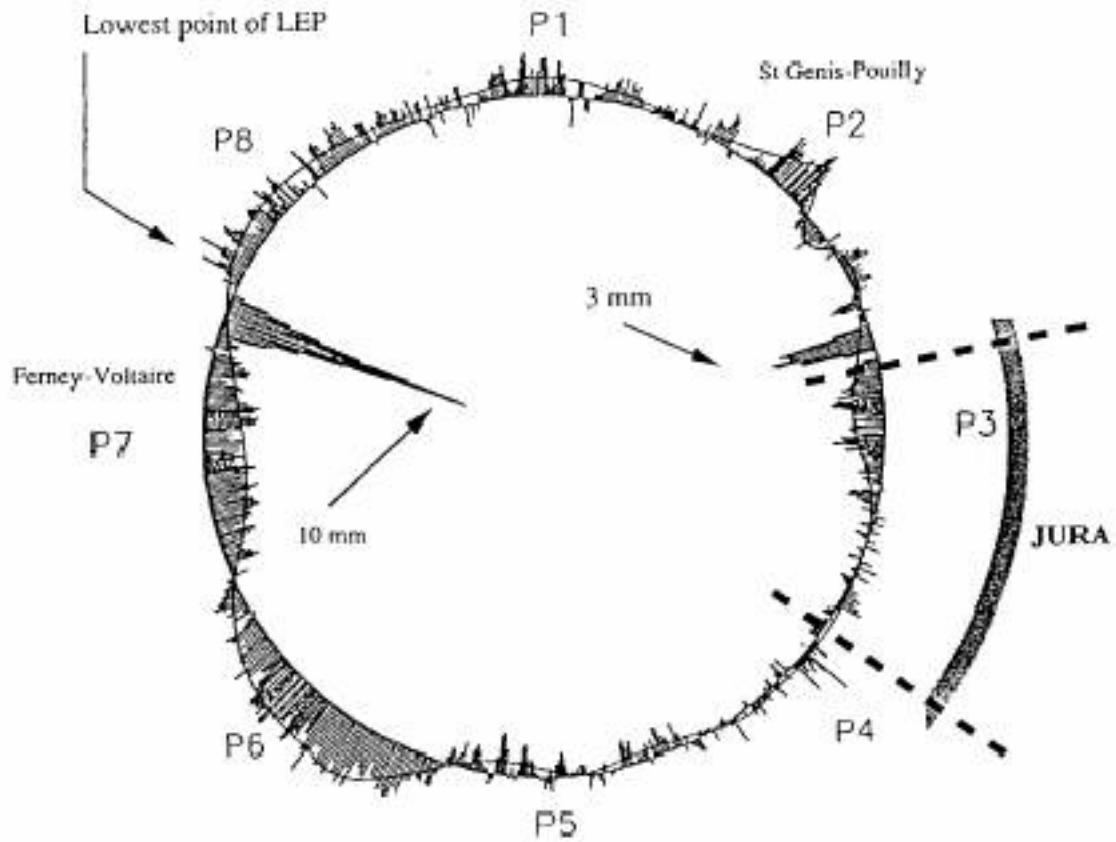


Figure 1 : observed status, end 1992

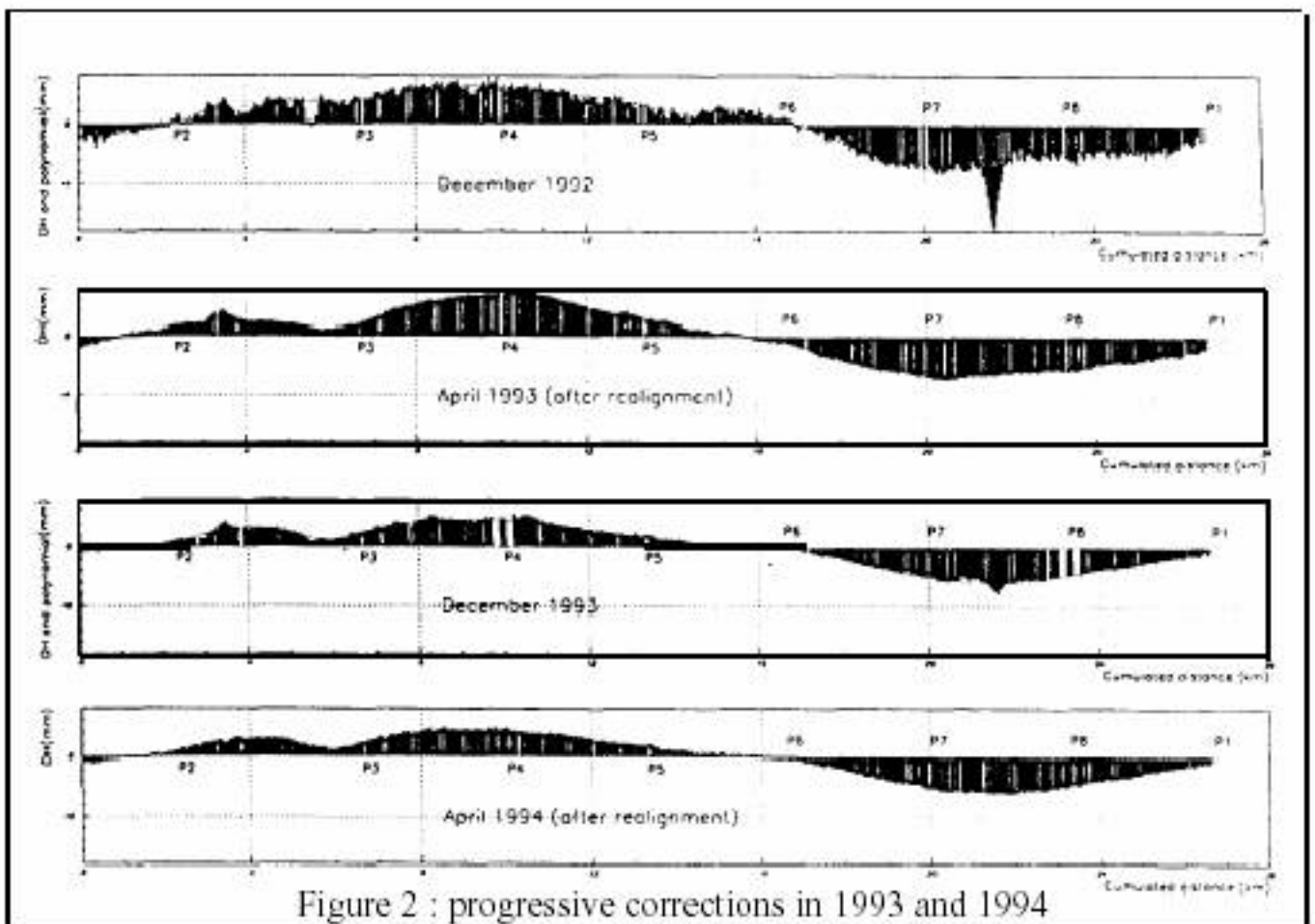


Figure 2 : progressive corrections in 1993 and 1994

# Problems Generated by Orbit Errors

## ● injection errors:

■ aperture → beam losses

■ filamentation → beam size

## ● closed orbit errors:

■ x-y coupling

■ aperture

■ energy error

■ field imperfections

■ dispersion → beam size at IP

■ beam separation

Aim:

$\Delta x, \Delta y < 4 \text{ mm}$

$rms < 0.5 \text{ mm}$

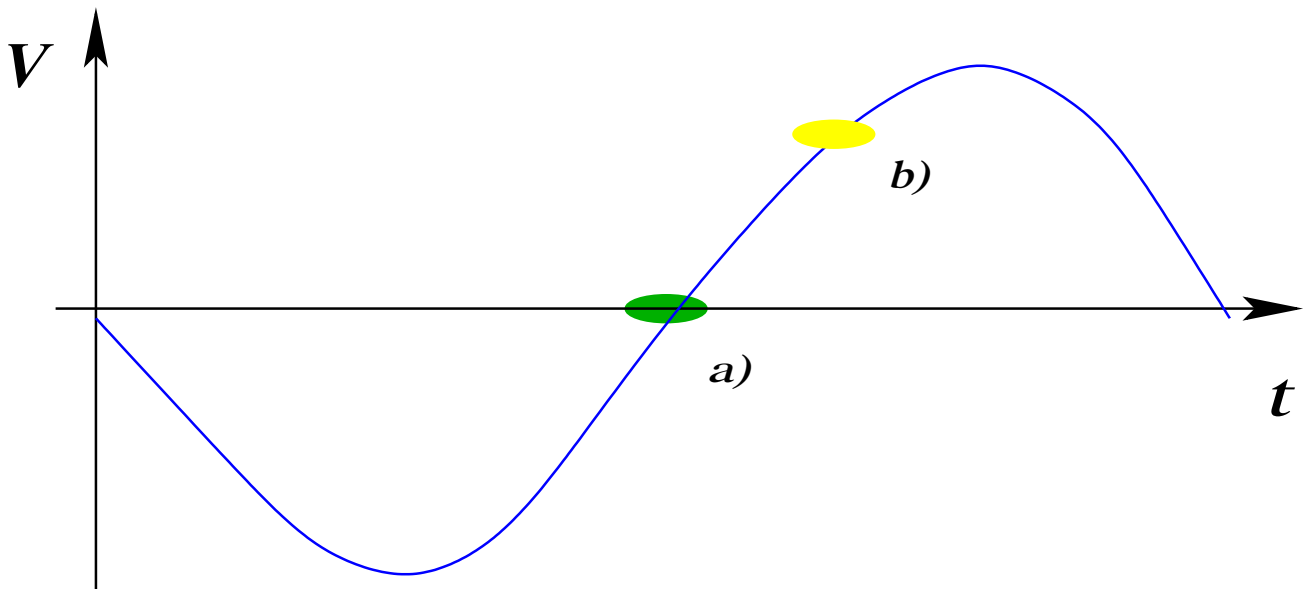


beam monitors and  
orbit correctors

Synchrotron:

→ the orbit determines the particle energy!

■ assume:  $L >$  design orbit



→ energy increase

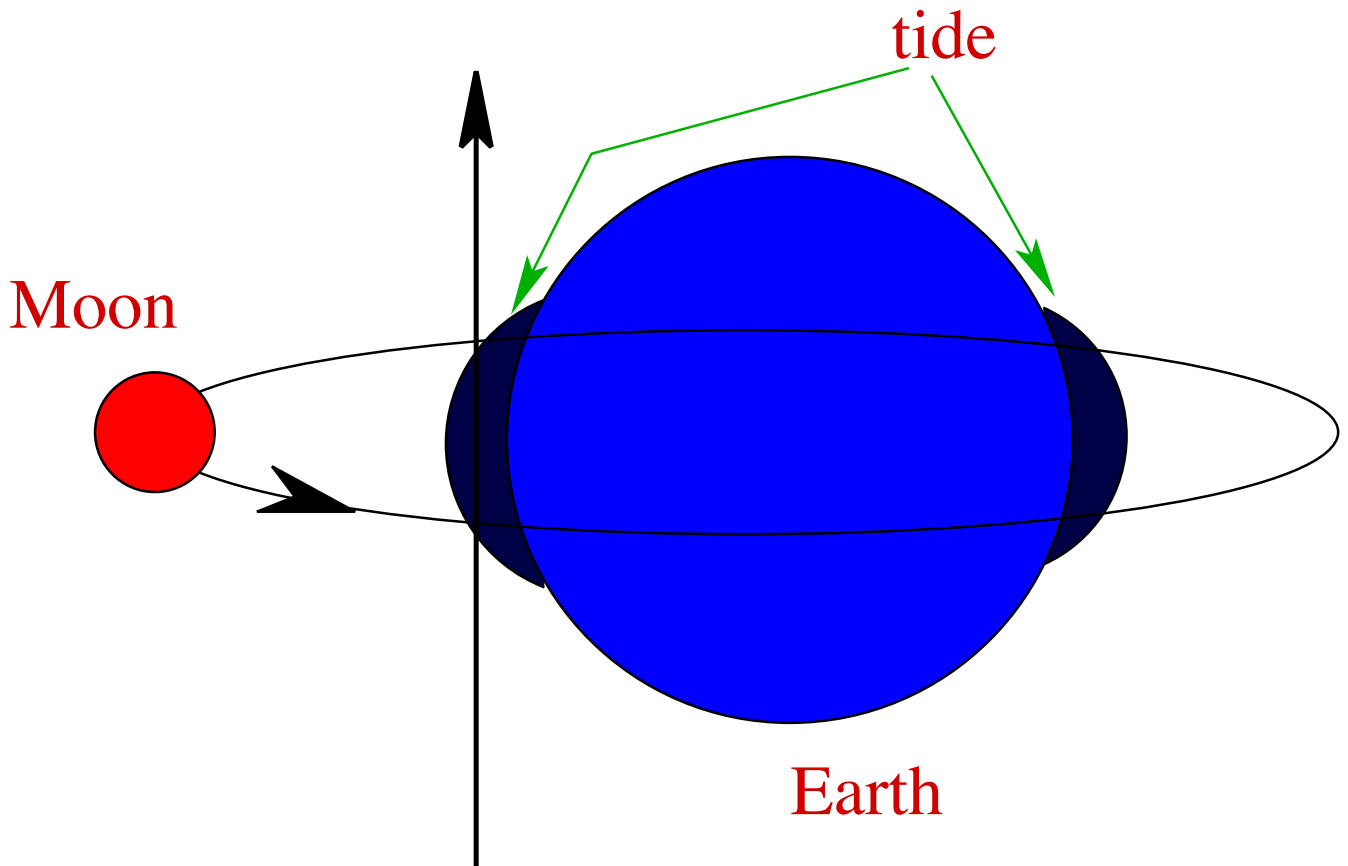
Equilibrium:

$$f_{RF} = h \cdot f_{rev}$$

$$f_{rev} = \frac{1}{2\pi} \cdot \frac{q}{m \cdot \gamma} \cdot B$$

→  $E$  depends on orbit and magnetic field!

■ *tidal motion of the earth:*



■ *orbit and beam energy modulation:*

$$f_{mod} = 24 h; 12 h$$

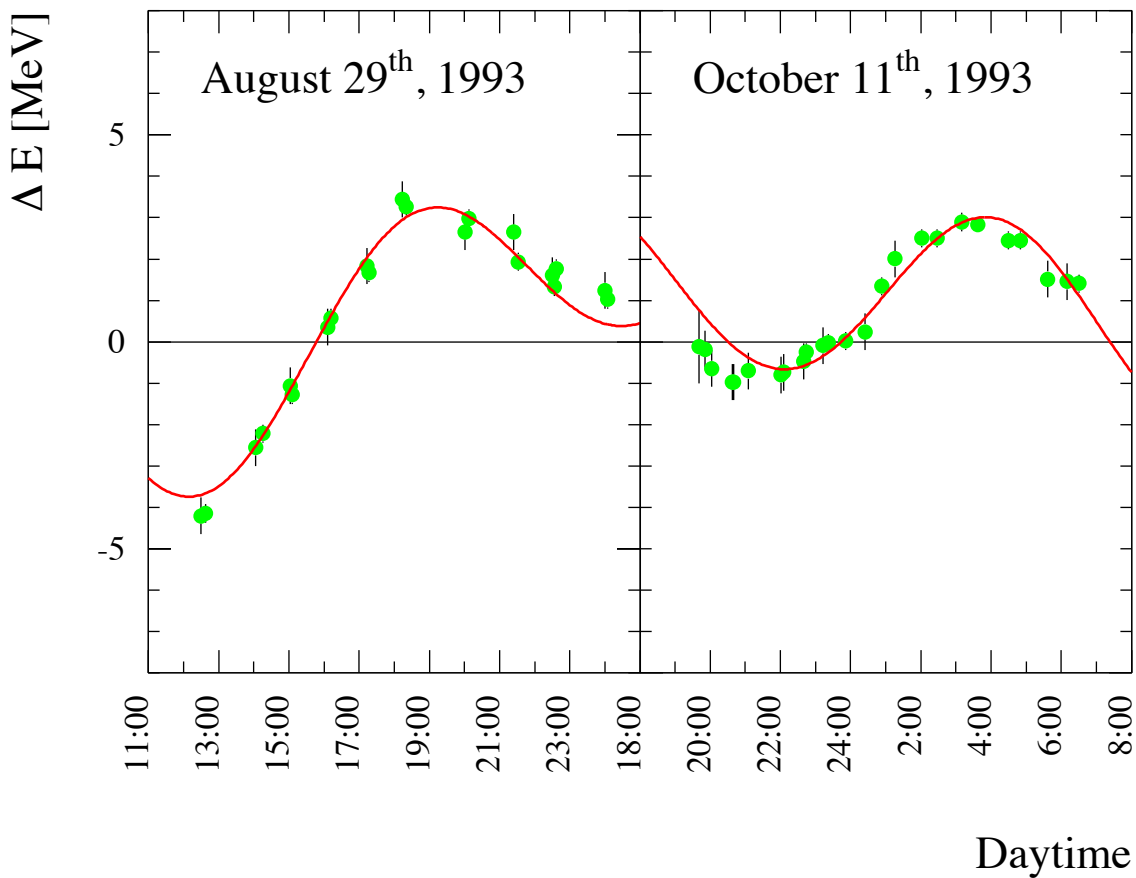
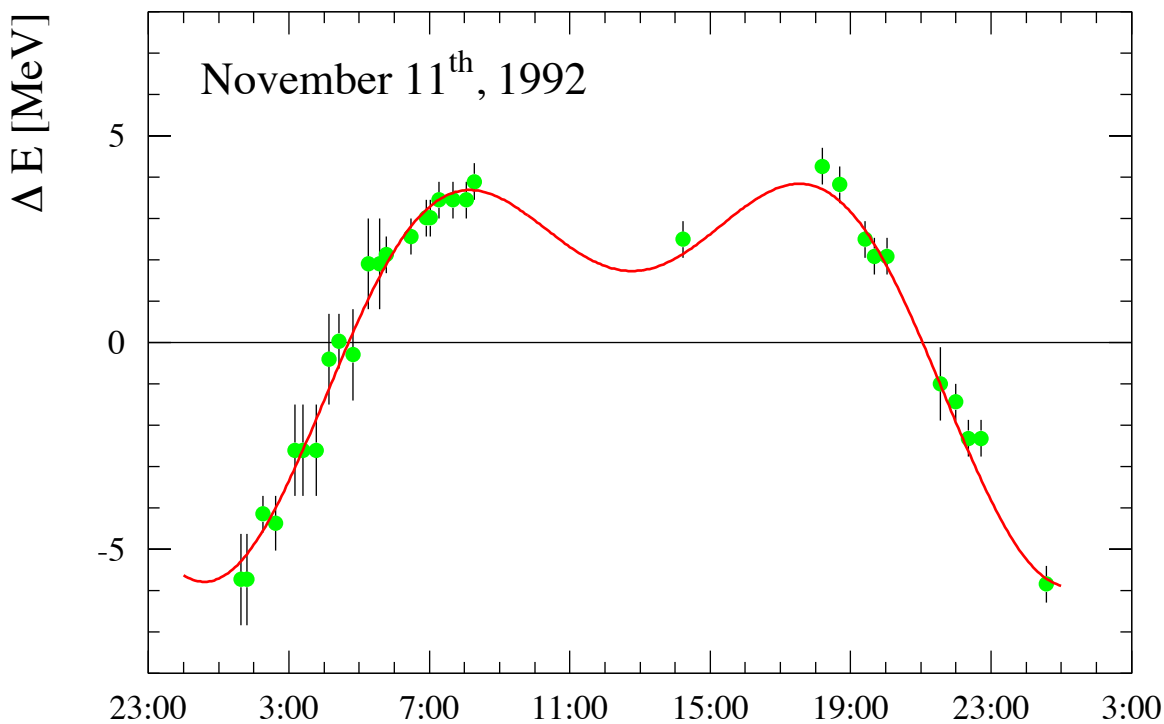
→  $\Delta E \approx 10 \text{ MeV}$

$$\approx 0.02\%$$

*aim:*  $\Delta E \lesssim 0.003\%$

→ *requires correction!*

# energy modulation due to tidal motion of earth



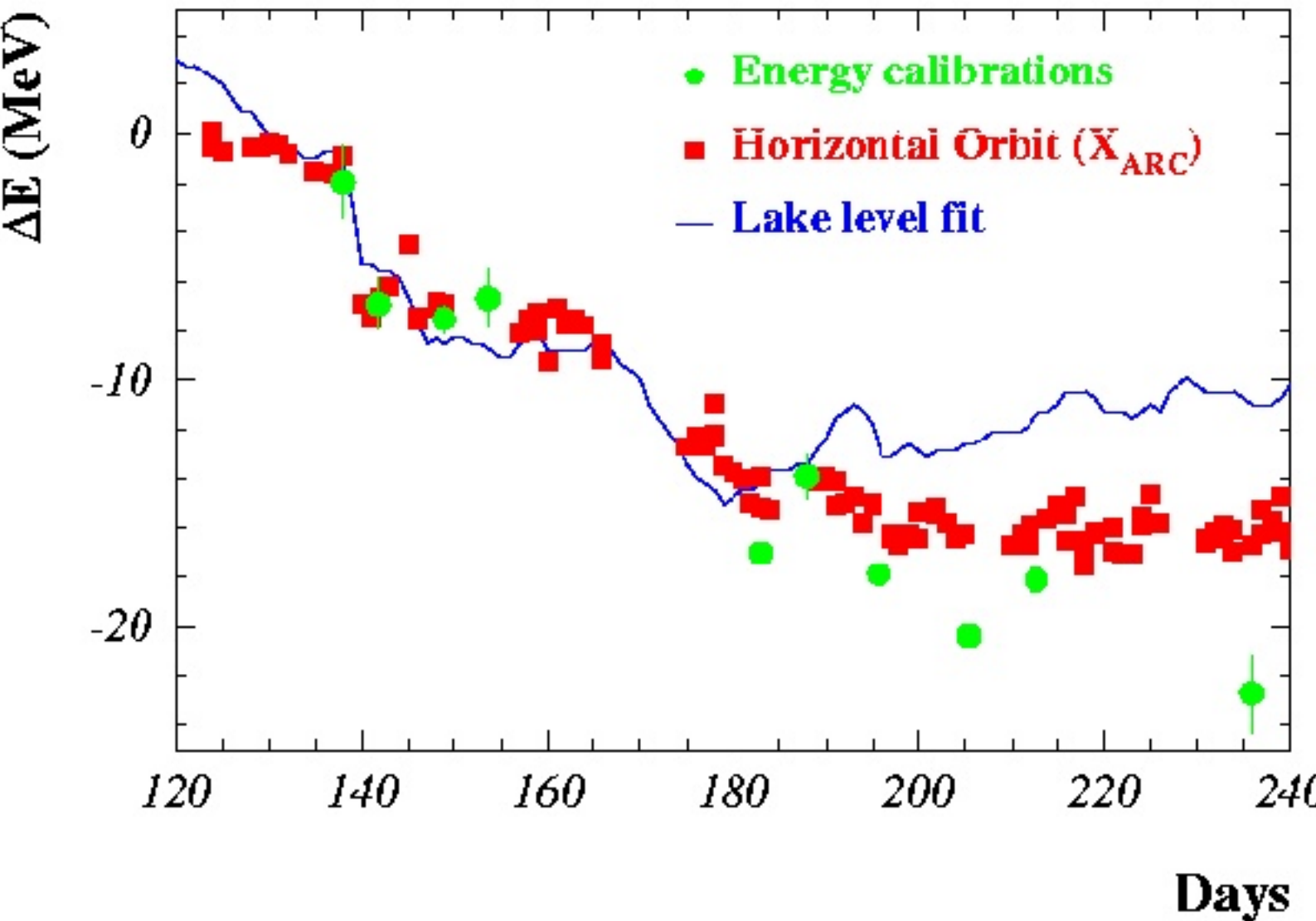
→  $\Delta E \approx 10 \text{ MeV}$



energy modulation due to lake level changes

changes in the water level of lake Geneva change the position of the LEP tunnel and thus the quadrupole positions

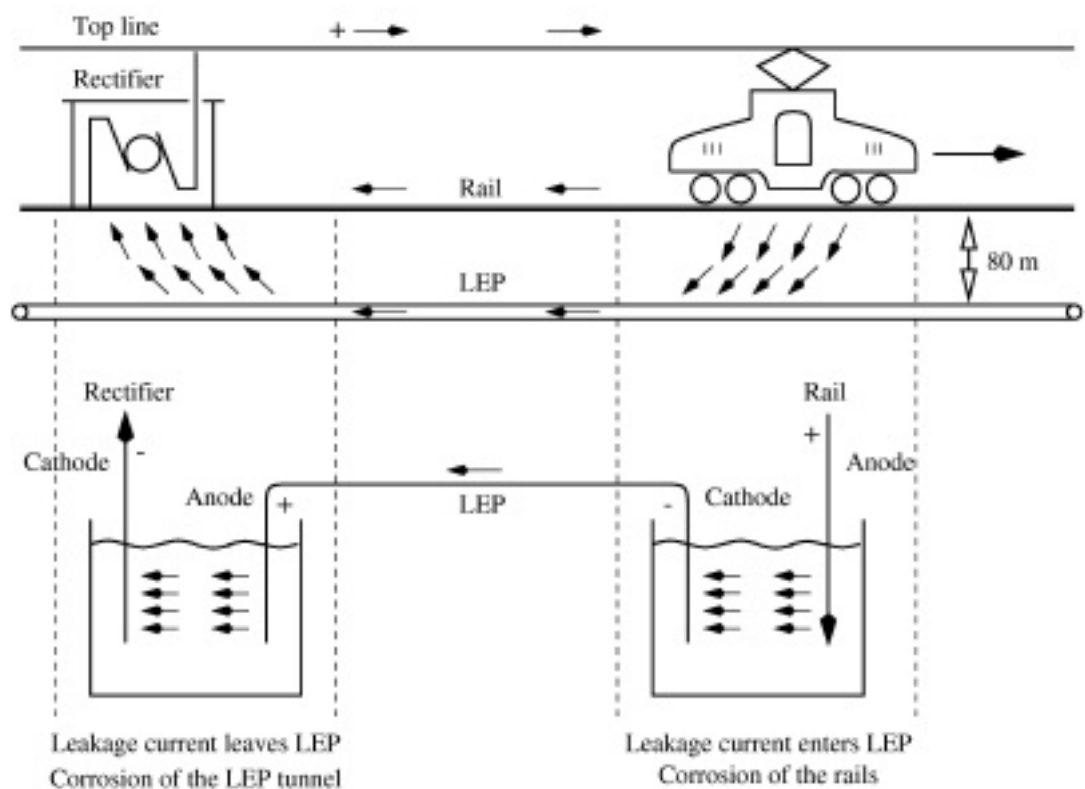
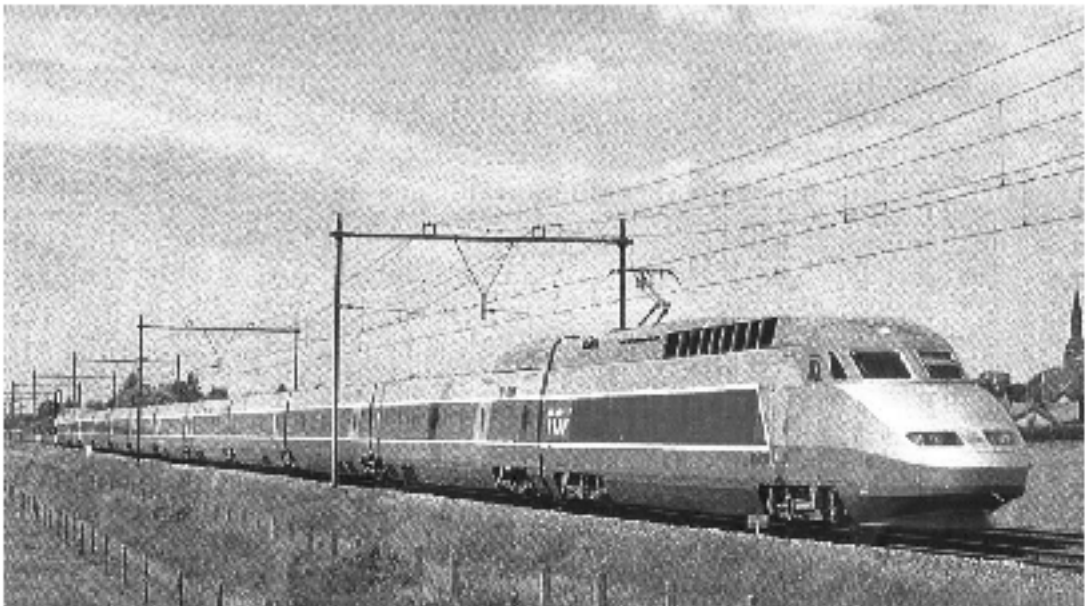
→ orbit and energy perturbations



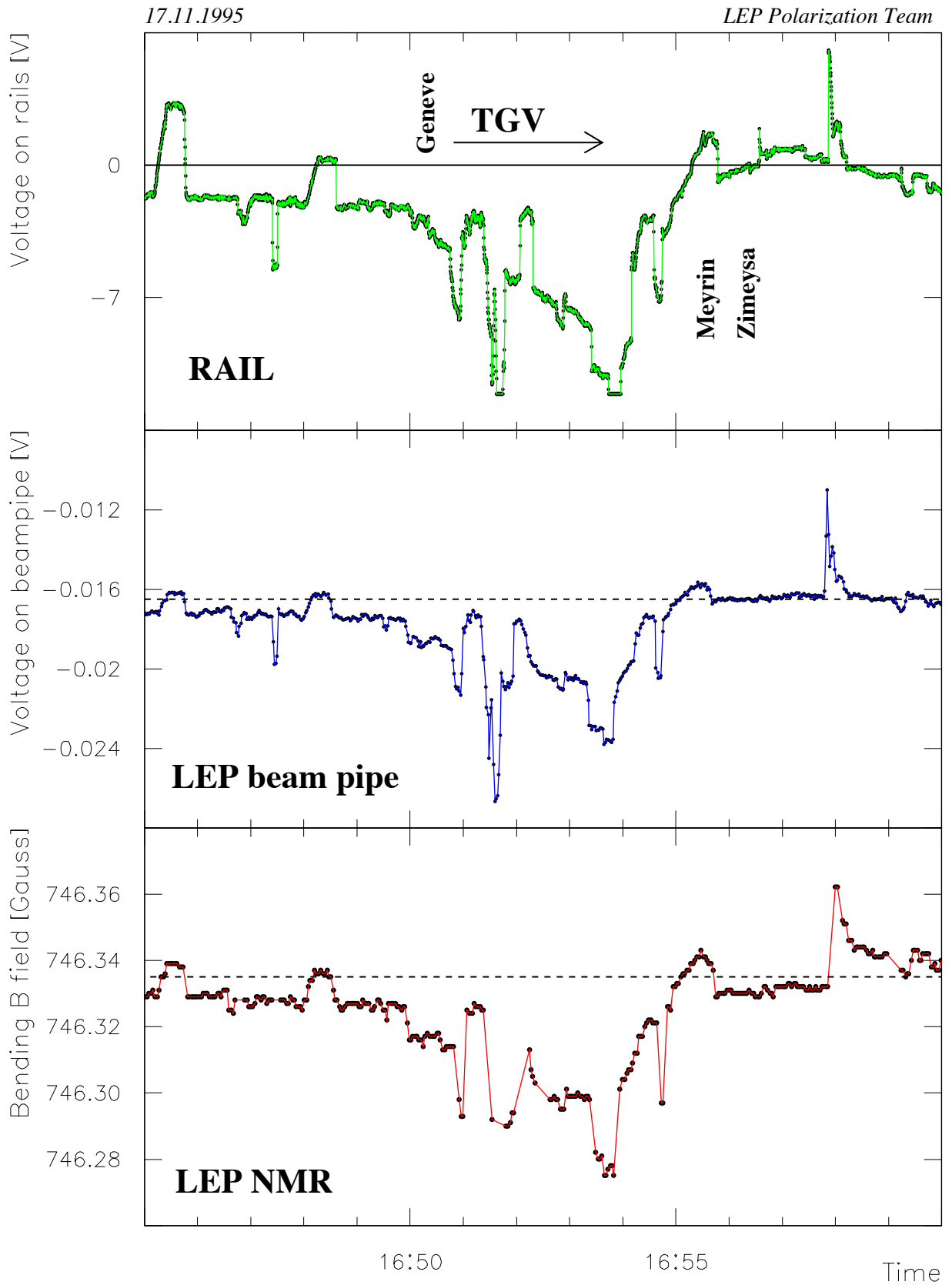
→  $\Delta E \approx 20 \text{ MeV}$

■ energy modulation due current perturbations in the main dipole magnets

■ TGV line between Geneva and Bellegarde



■ correlation of NMR dipole field measurements  
with the voltage on the TGV train tracks

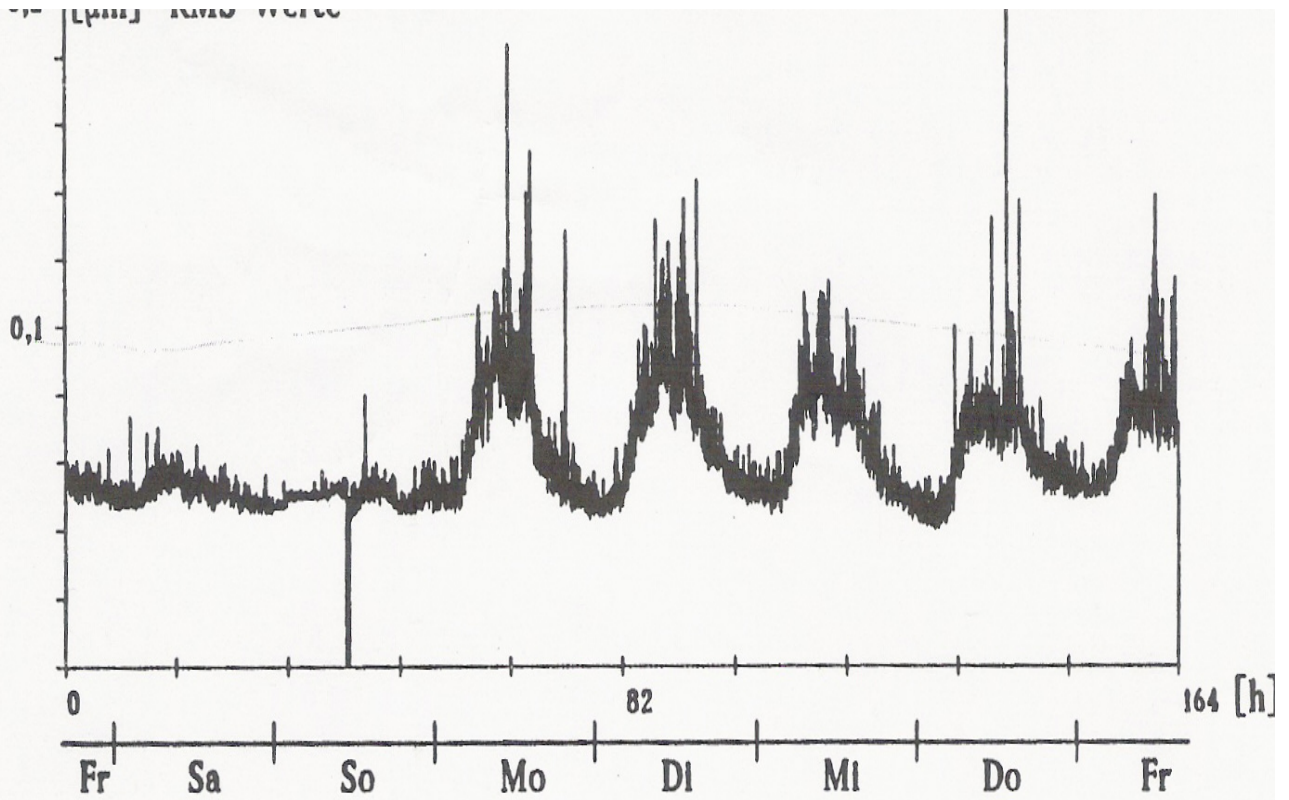


$\Delta E \approx 5 \text{ MeV}$  for LEP operation at 45 GeV

ground motion due to human activity

quadrupole motion in HERA-p (DESY Hamburg)

RMS



peak to peak

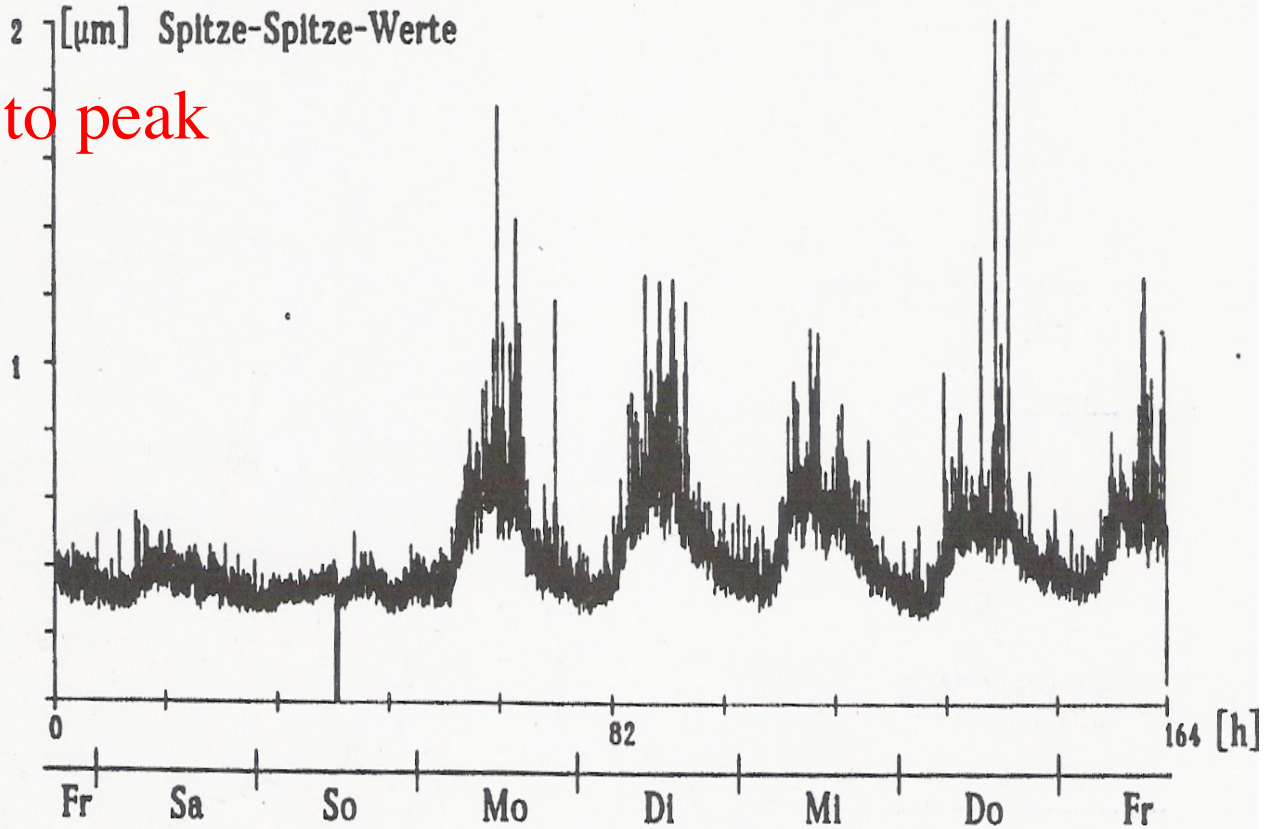


Abb. 3.13 Zeitabhängigkeit der Bodenbewegung  
oben RMS-Werte  
unten Spitze-Spitze-Werte

## Closed Orbit Response

**inhomogeneous equation:**

$$\frac{d^2 x}{d s^2} + K(s) \cdot x = G(s); \quad G(s) = \Delta k_0(s)$$

$$\vec{y}' + \begin{pmatrix} 0 & 1 \\ K & 0 \end{pmatrix} \cdot \vec{y} = \vec{G}; \quad \vec{G} = \begin{pmatrix} 0 \\ G \end{pmatrix}$$

$$\vec{y}(s) = a \cdot \vec{S}(s) + b \cdot \vec{C}(s) + \vec{\psi}(s)$$

we need to find only one solution!

**variation of the constant:**

$$\vec{\psi}(s) = c(s) \cdot \vec{S}(s) + d(s) \cdot \vec{C}(s)$$

## Closed Orbit Response

variation of the constant in matrix form:

$$\vec{\psi}(s) = \underline{\phi}(s) \cdot \vec{u}(s); \quad \text{with}$$

$$\underline{\phi}(s) = \begin{pmatrix} C(s) & S(s) \\ C'(s) & S'(s) \end{pmatrix}$$

substitute into differential equation:

$$\underline{\phi}(s) \cdot \vec{u}'(s) = \vec{G}(s)$$

$$\vec{u}(s) = \int_{s_0}^s \underline{\phi}(t)^{-1} \cdot \vec{G}(t) dt$$

$$\vec{y}(s) = a \cdot \vec{S}(s) + b \cdot \vec{C}(s) + \underline{\phi}(s) \cdot \int_{s_0}^s \underline{\phi}(t)^{-1} \cdot \vec{G}(t) dt$$

## Closed Orbit Response

**periodic boundary conditions:**

$$\vec{y}(s) = a \cdot \vec{S}(s) + b \cdot \vec{C}(s) + \underline{\phi}(s) \cdot \int_{s_0}^s \underline{\phi}(t)^{-1} \cdot \vec{G}(t) dt$$

with

$$\vec{y}(s) = \begin{pmatrix} x(s) \\ x'(s) \end{pmatrix}; \quad x(s) = x(s + L); \quad x'(s) = x'(s + L)$$



*periodic boundary conditions determine coefficients a and b*



$$x(s) = \frac{\sqrt{\beta(s)}}{2 \sin(\pi \cdot Q)} \cdot \int_{s_0}^{s_0 + \text{circ}} \sqrt{\beta(t)} \cdot G(t) \cos[\phi(t) - \phi(s) - \pi Q] dt$$

# Closed Orbit Response

■ *Example:* particle momentum error

normalized dipole strength:  $k_o(s) = 0.3 \cdot \frac{B[T]}{p[GeV]}$

$$k_o(s) = \frac{1}{\rho(t)} - \frac{1}{\rho(t)} \cdot \frac{\Delta p}{p_o} \longrightarrow G(t) = \frac{1}{\rho(t)} \cdot \frac{\Delta p}{p_o}$$

$$x(s) = \frac{\sqrt{\beta(s)}}{2 \sin(\pi \cdot Q)} \cdot \oint \sqrt{\beta(t)} \cdot G(t) \cos[|\phi(t) - \phi(s)| - \pi Q] dt$$

$$\longrightarrow x(s) = D(s) \cdot \frac{\Delta p}{p}$$

with

$$D(s) = \frac{\sqrt{\beta(s)}}{2 \sin(\pi \cdot Q)} \cdot \oint \frac{\sqrt{\beta(t)}}{\rho(t)} \cdot \cos[|\phi(t) - \phi(s)| - \pi Q] dt$$

$\longrightarrow$  *Dispersion Orbit*



# *Orbit Correction*

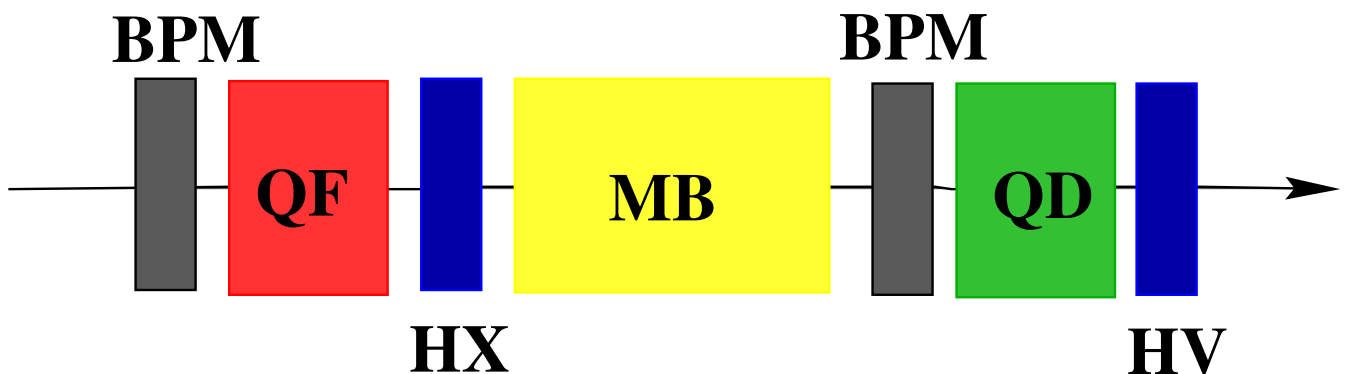
- the orbit error in a storage ring with conventional magnets is dominated by the contributions from the quadrupole alignment errors
- orbit perturbation is proportional to the local  $\beta$ -functions at the location of the dipole error
  - alignment errors at QF cause mainly horizontal orbit errors
  - alignment errors at QD causes mainly vertical orbit errors

# Orbit Correction

■ aim at a local correction of the dipole error due to the quadrupole alignment errors

→ place orbit corrector and BPM next to the main quadrupoles

→ horizontal BPM and corrector next to QF  
vertical BPM and corrector next to QD



→ orbit in the opposite plane?

relative alignment of BPM and quadrupole?

# LEP Orbit

## Horizontal Orbit:

■ *beam offset in quadrupoles:*

→ *Lake Geneva*

→ *moon*

→ *energy error*

## Vertical Orbit:

■ *beam offset in quadrupoles*

■ *beam separation*

→ *orbit deflection depends on particle energy*

→ *vertical dispersion [D(s)]*

$$\sigma_y = \sqrt{\varepsilon \cdot \beta_y + \delta_y^2 \cdot D^2}$$

→ *small vertical beam size relies on good orbit*

■ *1994: 13000 vertical orbit corrections in physics*

# Quadrupole Gradient Error

**one turn map:**

can be generated by matrix multiplication:

$$\longrightarrow \vec{z}_{n+1} = \underline{M} \cdot \vec{z}_n \quad \vec{z} = \begin{pmatrix} x \\ x' \end{pmatrix}$$

and can be expressed in terms of the C and S solutions

$$\underline{M} = \underline{I} \cdot \cos(2\pi Q) + \underline{J} \cdot \sin(2\pi Q)$$

$$\underline{I} = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} \quad \underline{J} = \begin{pmatrix} \alpha & \beta \\ -\gamma & -\alpha \end{pmatrix}; \quad \gamma = [1 + \alpha^2] / \beta$$

remember:  $\cos(2\pi Q) = \frac{1}{2} \text{trace } \underline{M}$

$\longrightarrow$  the coefficients of:  $\frac{\underline{M} - \underline{I} \cdot \cos(2\pi Q)}{\sin(2\pi Q)}$

provide the optic functions at  $s_0$

# Quadrupole Gradient Error

transfer matrix for single quadrupole:

$$m_0 = \begin{pmatrix} 1 & 0 \\ -k_1 \cdot l & 1 \end{pmatrix}$$

matrix for single quadrupole with error:

$$m = \begin{pmatrix} 1 & 0 \\ -[k_1 + \Delta k_1] \cdot l & 1 \end{pmatrix}$$

one turn matrix with quadrupole error:

$$M = m \cdot m_0^{-1} \cdot M_0$$

trace M



$$\cos(2\pi Q) = \cos(2\pi Q_0) - \frac{1}{2} \beta \cdot \Delta k_1 \cdot l \cdot \sin(2\pi Q_0)$$

# Quadrupole Gradient Error

distributed perturbation:

$$\cos(2\pi Q) = \cos(2\pi Q_0) - \frac{\sin(2\pi Q_0)}{2} \cdot \int \beta \cdot \Delta k_1 ds$$

$$\longrightarrow \Delta Q = \frac{1}{4\pi} \cdot \int \beta \cdot \Delta k_1 ds$$

chromaticity:

$$k_1 = \frac{e \cdot g}{p}$$

momentum error  $\longrightarrow \Delta k_1 = -k_1 \cdot \frac{\Delta p}{p}$

$$\Delta Q = -\frac{1}{4\pi} \cdot \int \beta \cdot k_1 \cdot ds \cdot \frac{\Delta p}{p}$$

$$= \xi \cdot \frac{\Delta p}{p}$$

# $\beta$ - Beat

■ *quadrupole error:*

$$\longrightarrow \vec{z}_{n+1} = \underline{M} \cdot \vec{z}_n \quad \underline{M} = \begin{pmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{pmatrix}$$

with

$$\underline{M} = \underline{I} \cdot \cos(2\pi Q) + \underline{J} \cdot \sin(2\pi Q)$$

$$\underline{I} = \begin{pmatrix} 1 & 0 \\ 0 & 1 \end{pmatrix} \quad \underline{J} = \begin{pmatrix} \alpha & \beta \\ -\gamma & -\alpha \end{pmatrix}; \quad \gamma = [1 + \alpha^2] / \beta$$

$$\longrightarrow \text{calculate: } \frac{m_{12}}{\sin(2\pi Q)}$$

$$\Delta\beta(s) = \frac{\beta(s)}{2 \sin(2\pi \cdot Q)} \cdot \int_{s_0}^{s_0 + \text{circ}} \beta(t) \cdot \Delta k(t) \cos[2[\phi(t) - \phi(s)] - 2\pi Q] dt$$



$\beta$  - beat oscillates with twice the betatron frequency

# Local Orbit Bumps I

■ deflection angle:

$$\theta_i = \int_{\text{dipole}} G_i(t) dt = \frac{0.3 \cdot B_i[\text{T}] \cdot l}{p[\text{GeV}]}$$

■ trajectory response:

[no periodic boundary conditions]

→  $x(s) = \sqrt{\beta_i \beta(s)} \cdot \theta_i \cdot \sin[\phi(s) - \phi_i]$

→  $x'(s) = \sqrt{\beta_i / \beta(s)} \cdot \theta_i \cdot \cos[\phi(s) - \phi_i]$



## *Local Orbit Bumps II*

### *closed orbit bump:*

compensate the trajectory perturbation with

additional corrector kicks further down stream

→ closure of the perturbation within one turn

→ local orbit excursion

→ possibility to correct orbit errors locally

→ closure with one additional corrector magnet

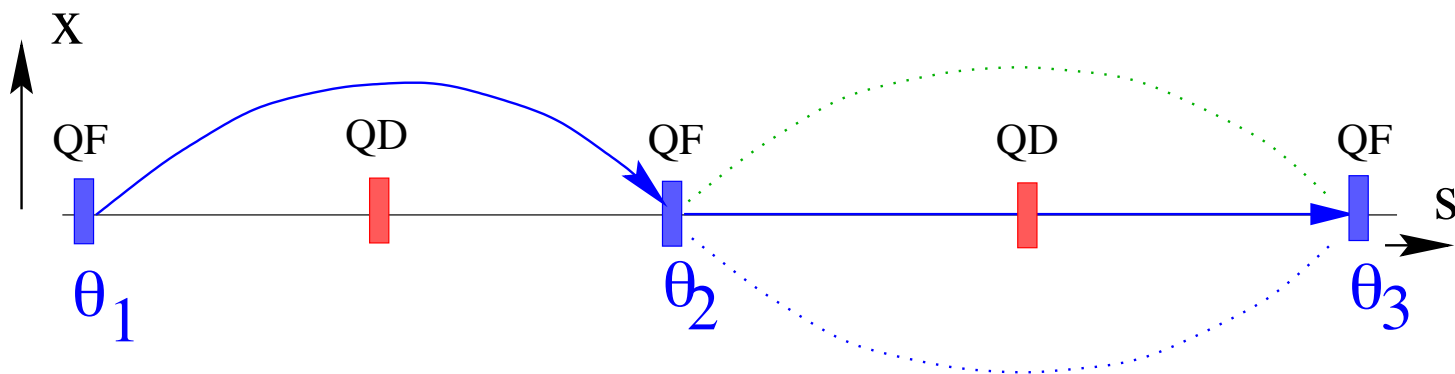
→  $\pi$  - bump

→ closure with two additional corrector magnets

→ three corrector bump

# Local Orbit Bumps III

■  $\pi$  - bump: (quasi local correction of error)



→ 
$$\theta_2 = \frac{-\sqrt{\beta_1}}{\sqrt{\beta_2}} \cdot \theta_1$$

■ limits / problems:

→ closure depends on lattice phase advance

→ requires  $90^\circ$  lattice

→ sensitive to lattice errors

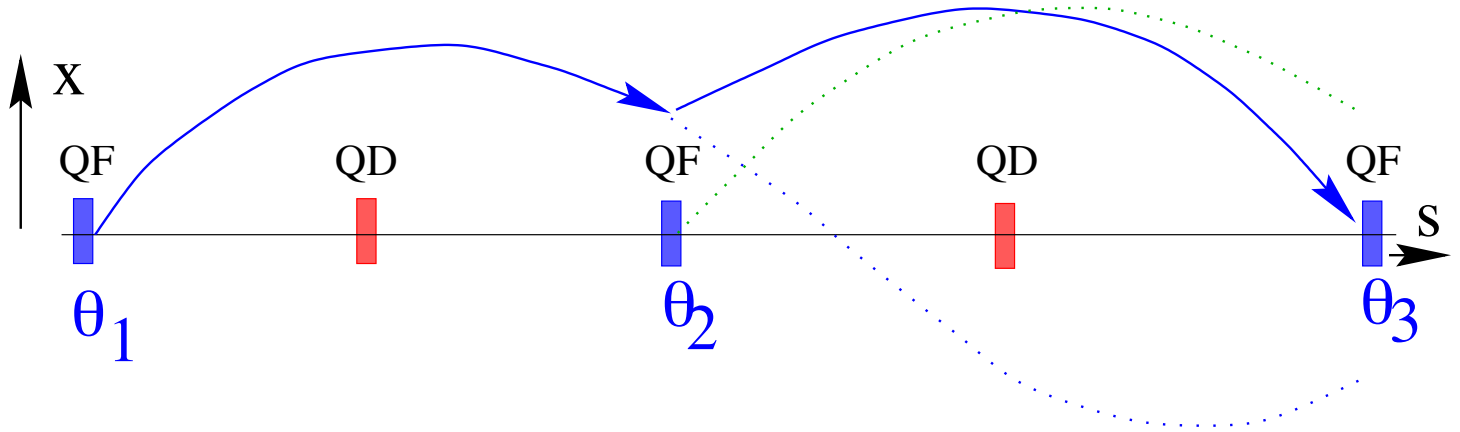
→ requires horizontal BPMs at QF and QD

→ sensitive to BPM errors

→ requires large number of correctors

# Local Orbit Bumps IV

3 corrector bump: (quasi local correction of error)



$$\theta_2 = -\frac{\sqrt{\beta_1}}{\sqrt{\beta_2}} \cdot \frac{\sin(\Delta\phi_{3-1})}{\sin(\Delta\phi_{3-2})} \cdot \theta_1$$

$$\theta_3 = \left( \frac{\sin(\Delta\phi_{3-1})}{\tan(\Delta\phi_{3-2})} - \cos(\Delta\phi_{3-1}) \right) \cdot \frac{\sqrt{\beta_1}}{\sqrt{\beta_3}} \cdot \theta_1$$

works for any lattice phase advance

requires only horizontal BPMs at QF

limits / problems:

sensitive to BPM errors

large number of correctors

can not control  $x^l$

# *Summary Linear Imperfections*

---

■ avoid machine tunes near integer resonances:

- they amplify the response to dipole field errors
- a closed orbit perturbation propagates with the betatron phase around the storage ring
- discontinuities in the derivative of the closed orbit response at the location of the perturbation

■ avoid storage ring tunes near half-integer resonances:

- they amplify the response to quadrupole field errors
- betafunction perturbations propagate with twice the betatron phase advance around the storage ring

■ integral expressions are mainly used for estimates  
numerical programs mainly rely on maps

- closed orbit = fixed point of '1-turn' map
- dispersion = eigenvector of extended '1-turn' map
- tune is given by the trace of the '1-turn' map
- twiss functions are given by the matrix elements