



Frequency Scanning Interferometry and its application for the internal monitoring of the cold mass position inside the cryostat

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HL-LHC MQXF Workshop 2019.10.19

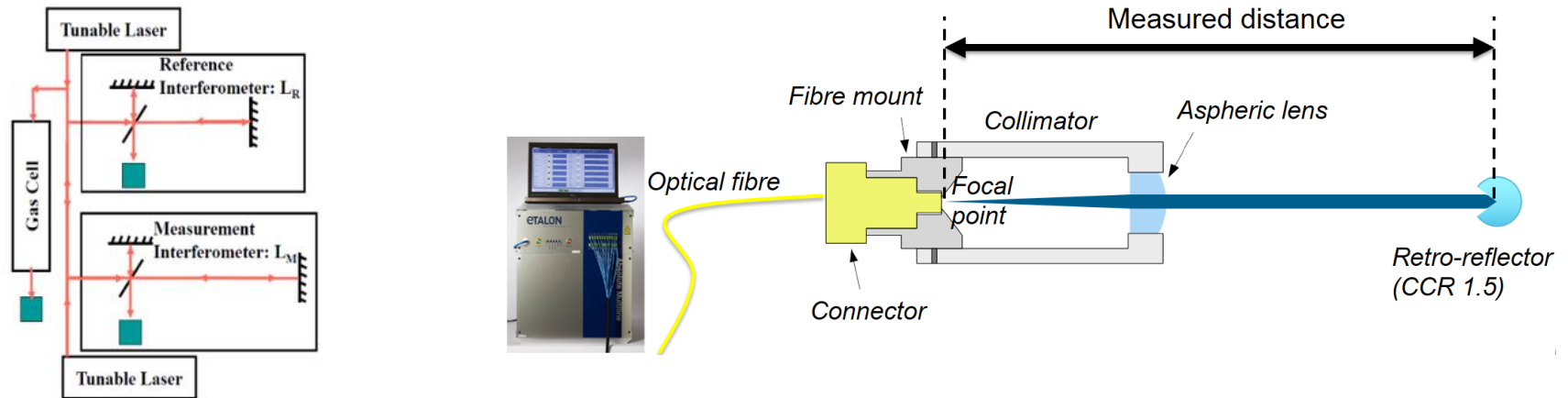


Outline

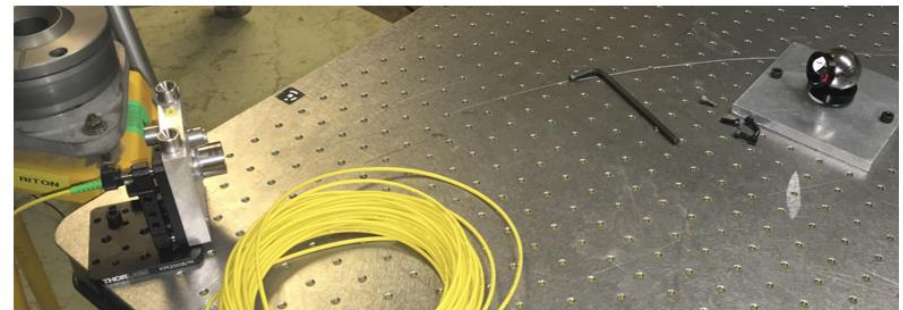
- FSI ETALON system to monitor HL-LHC cold mass position
 - FSI – short reminder
 - Crab-cavity FSI system and performance
 - Inner-triplet FSI system
- Dipole test
 - Integration of FSI instrumentation
 - Initial cool down tests
 - Survey results
 - Insulated prism concept, initial tests
 - Heat in-leak optimization, insulated prism design, test results
- FSI interface design, HL-LHC triplet integration
- Other tests and validations
- Summary

Reminder – FSI ETALON system

FSI – absolute interferometric distance measurement

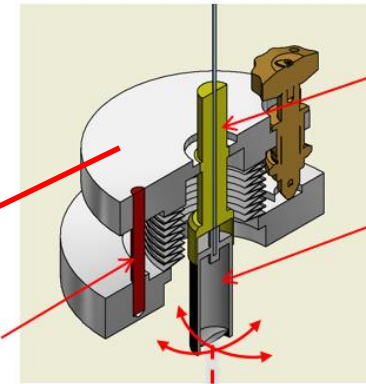
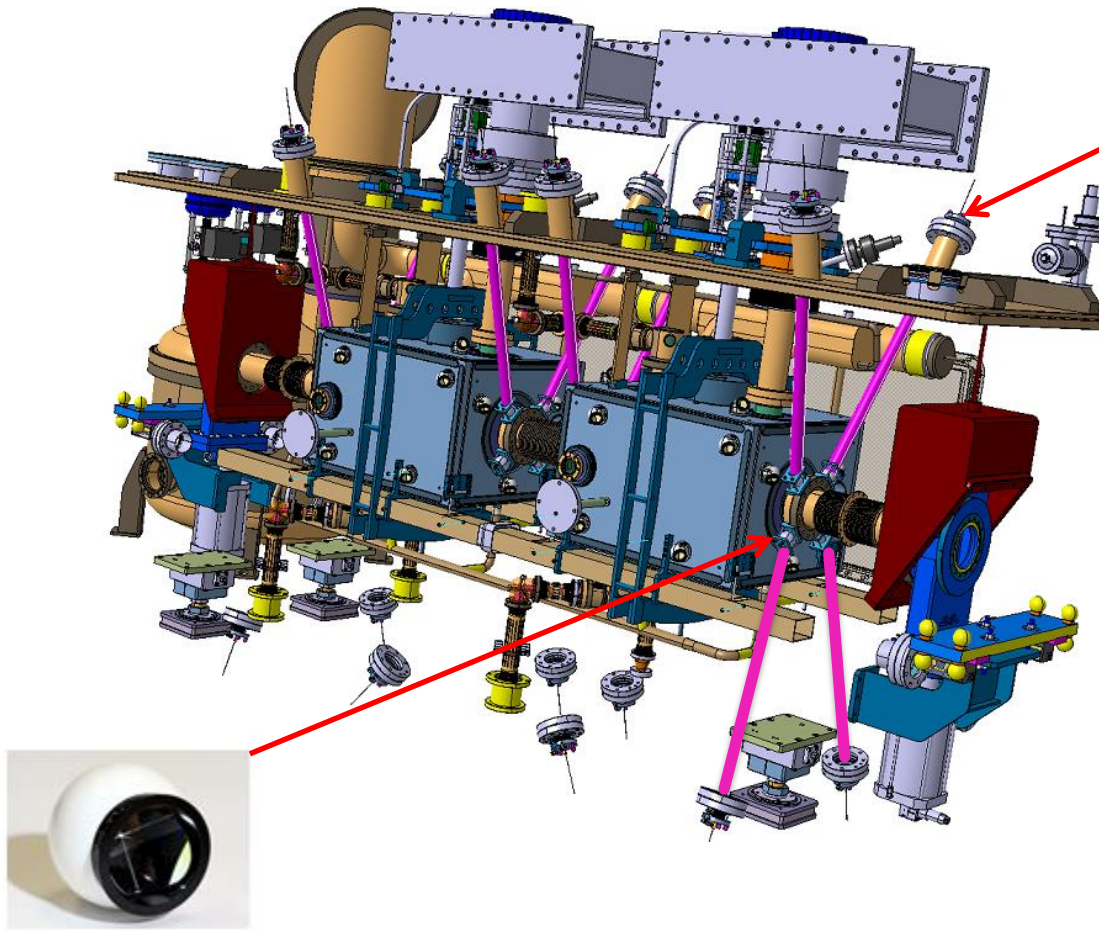


- Absolute distance measurement
- 8 channels expandable to 100
- Uncertainty (95%) = $0.5 \mu\text{m/m}$
- Traceable
- Measurement distance: 0.2 – 20 m
- Fiber length – up to several hundreds meters
- Length given by the ratio of measurement interferometer to reference interferometer fringes



Crab cavity FSI monitoring system

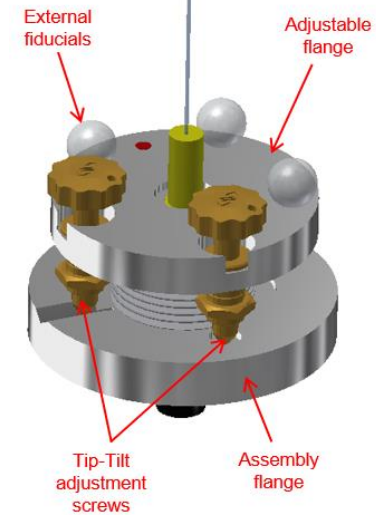
FSI solution – cryo-module implementation (DQW)



Optical fiber vacuum feedthrough

FSI head/collimator (tip-tilt) adjustment

Flexural support

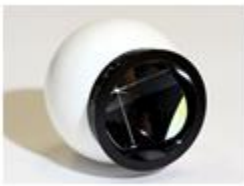


External fiducials

Adjustable flange

Tip-Tilt adjustment screws

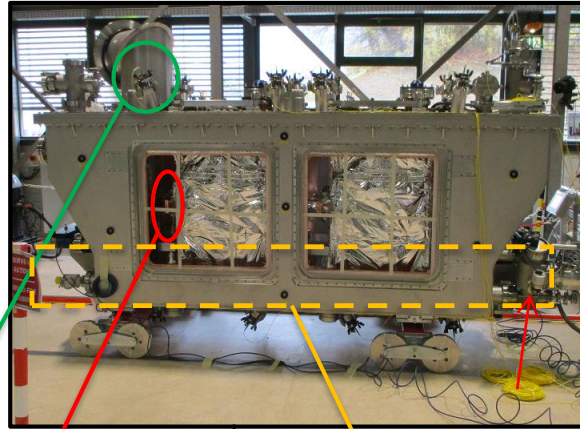
Assembly flange



Crab cavity position monitoring systems for SPS test

Frequency Scanning Interferometry (FSI) – main system

- Cavity position/orientation known thanks to absolute distance measurements between cryostat **FSI heads** and cavity **Corner Cube Reflectors (CCR)**
- System for HL-LHC use (cold, vacuum and radiation compatible)

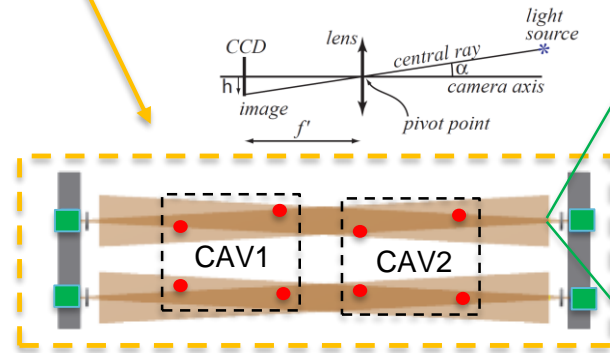
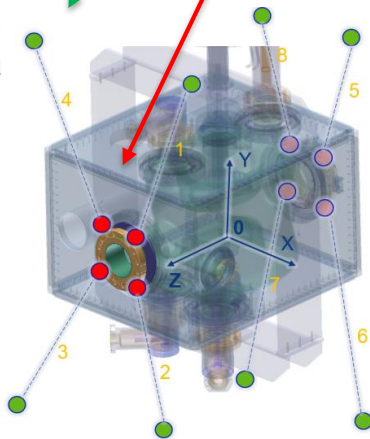


Brandeis Camera Angle Monitoring (BCAM)

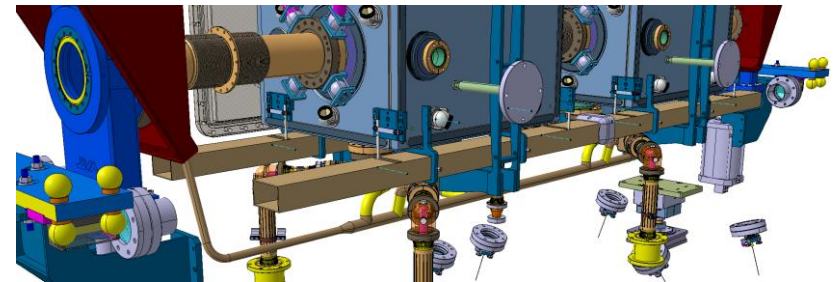
- Cavity position/orientation known thanks to **reflective targets** angular position measured by **BCAM cameras** (triangulation method)
- System used for FSI measurements crosscheck - only for SPS prototype alignment validation



- FSI
- CCR



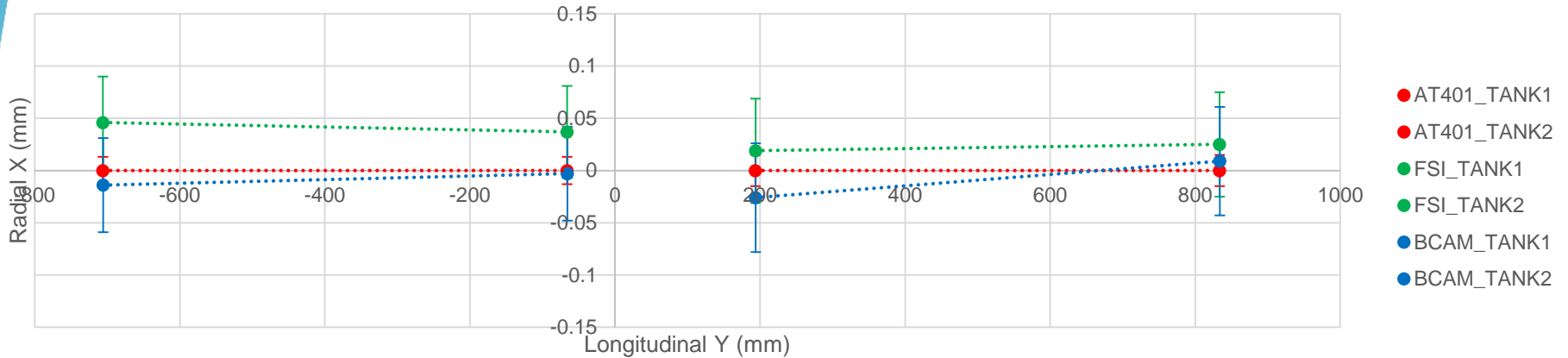
BCAM



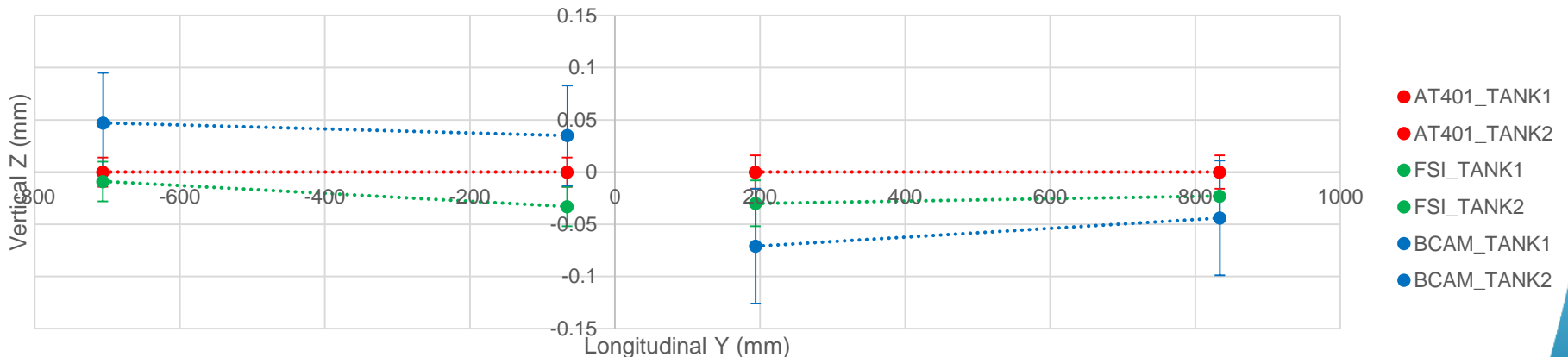
Crab cavity FSI performance at warm

- FSI, BCAM systems precision better than $50\mu\text{m}$ (1σ), crosschecked with AT401 laser tracker measurements

Radial position (relatif with respected to AT401 measurement)

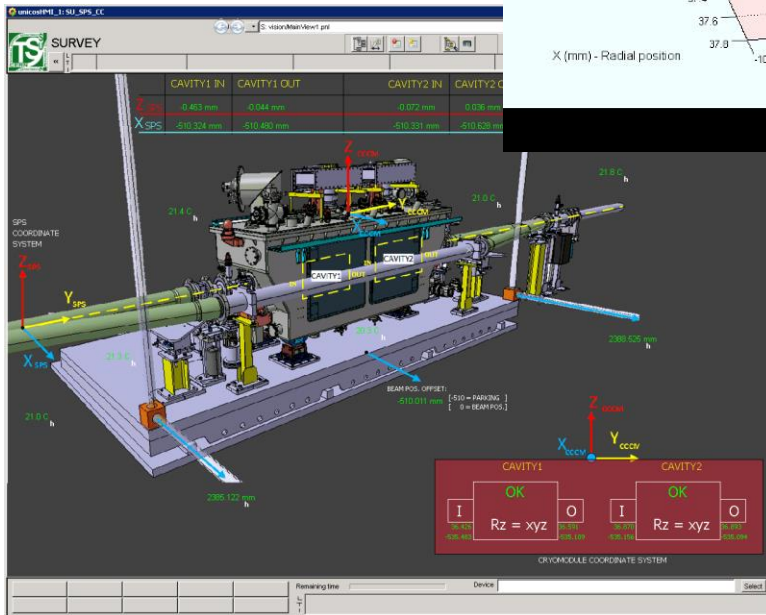
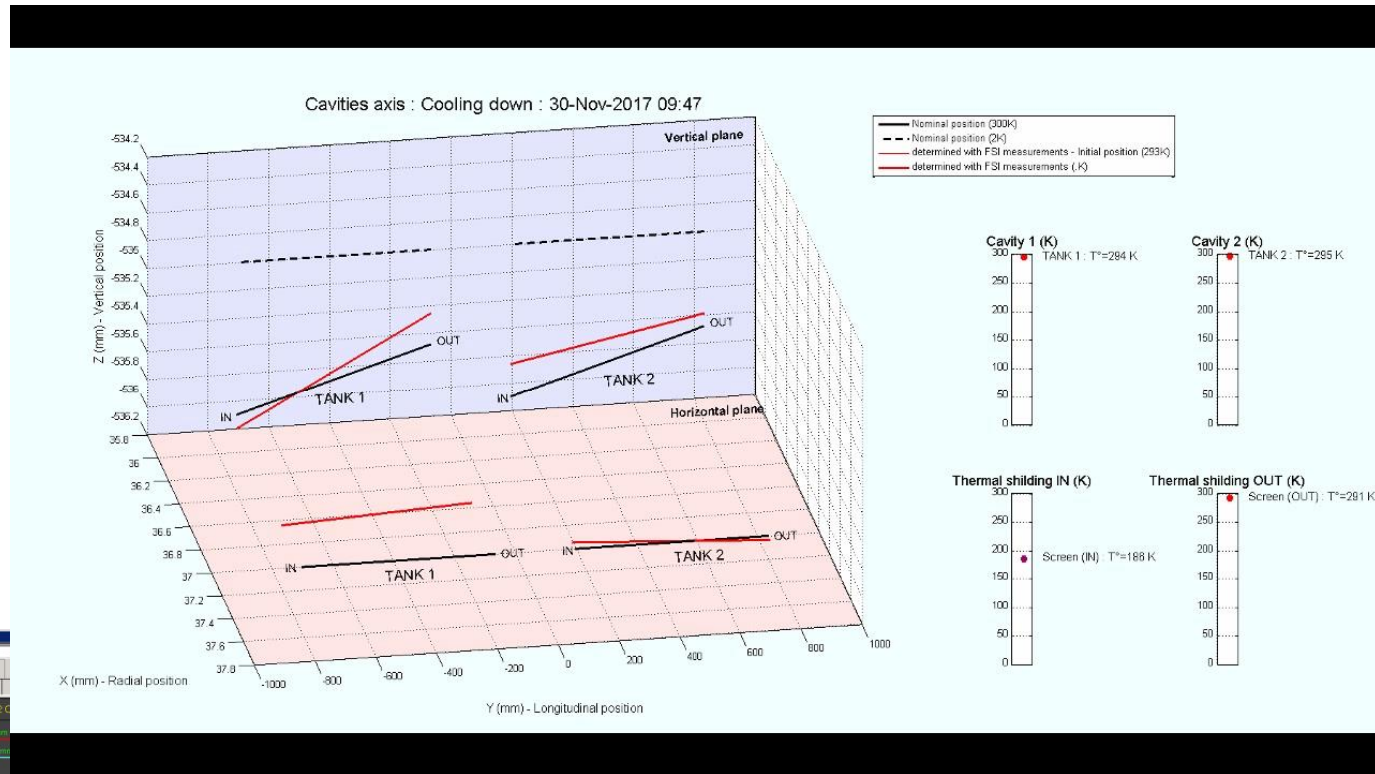


Vertical position (relatif with respected to AT401 measurement)



Crab cavity FSI performance – cool down

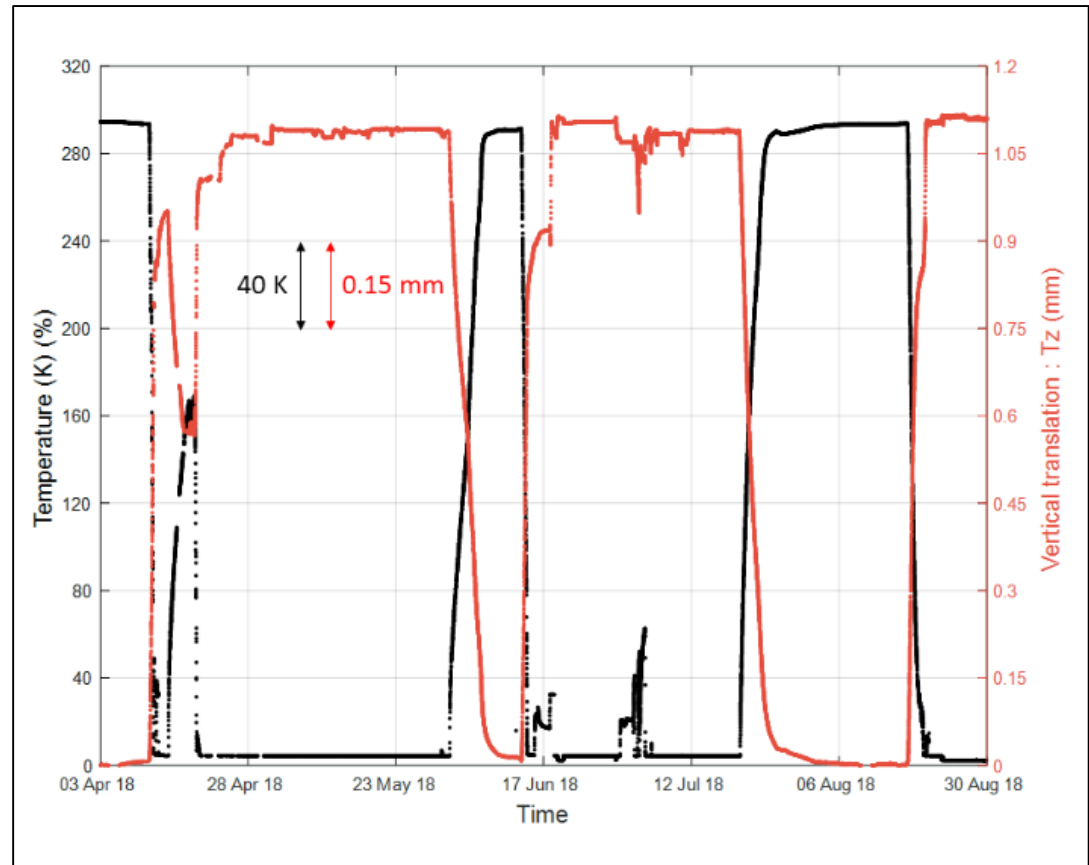
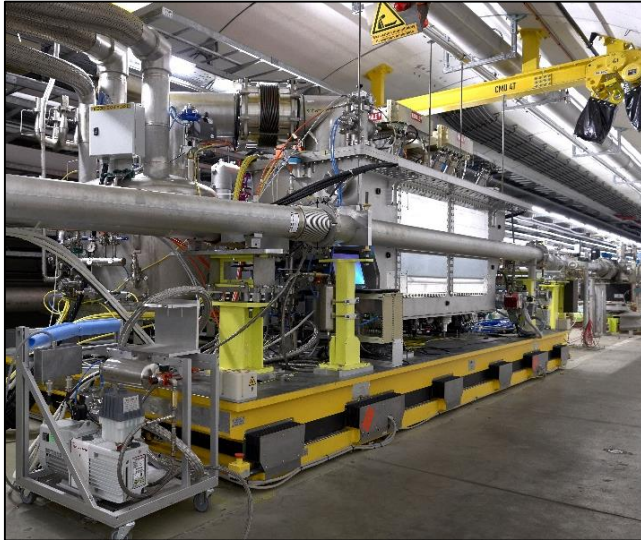
- System allows to follow the cooled crab cavity position/orientation
- Precision better than 50µm (1σ)



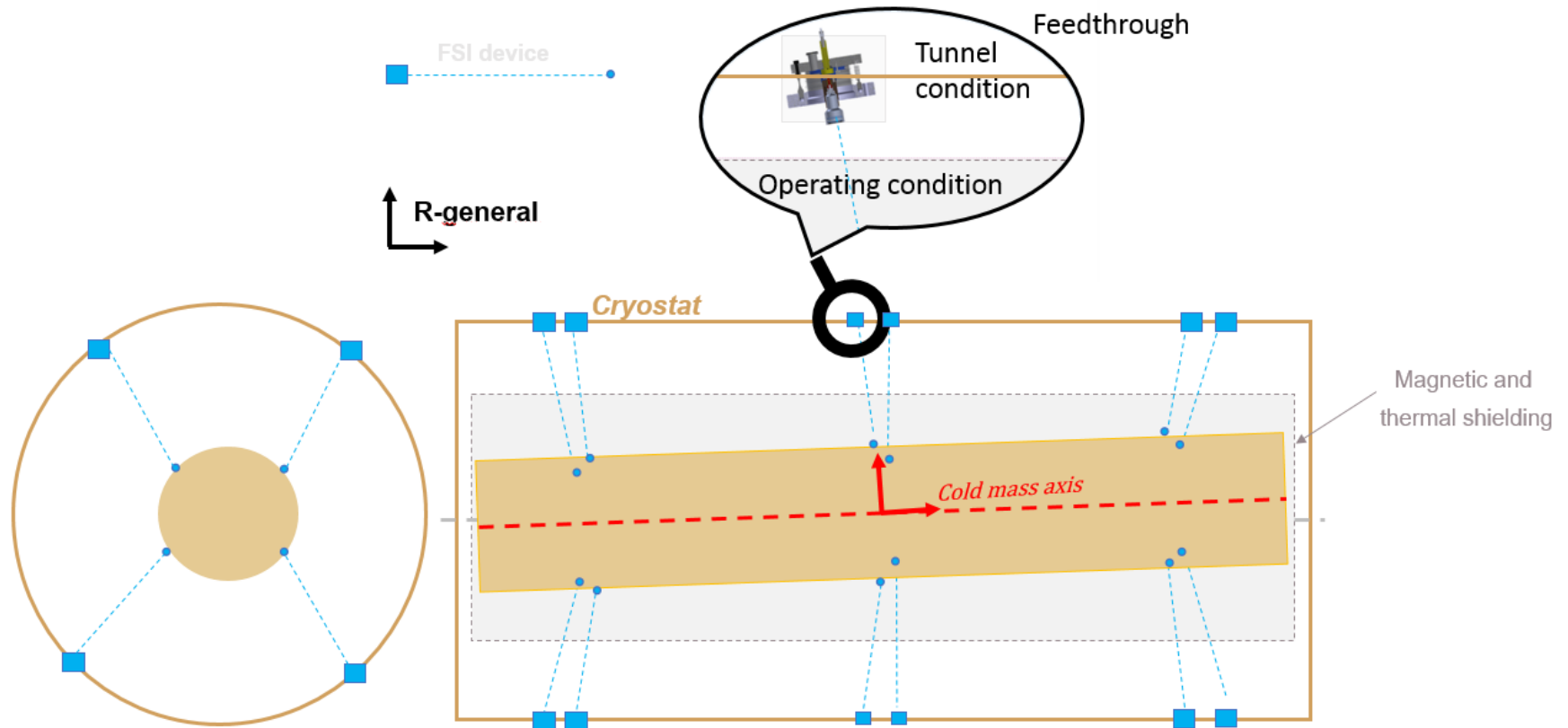
- Several cool-down cycles performed
- Micrometric resolution of objects moves observation (relative monitoring)
- FSI deployed in SPS crab cavity test stand

V. Rude

Crab cavities – long term monitoring

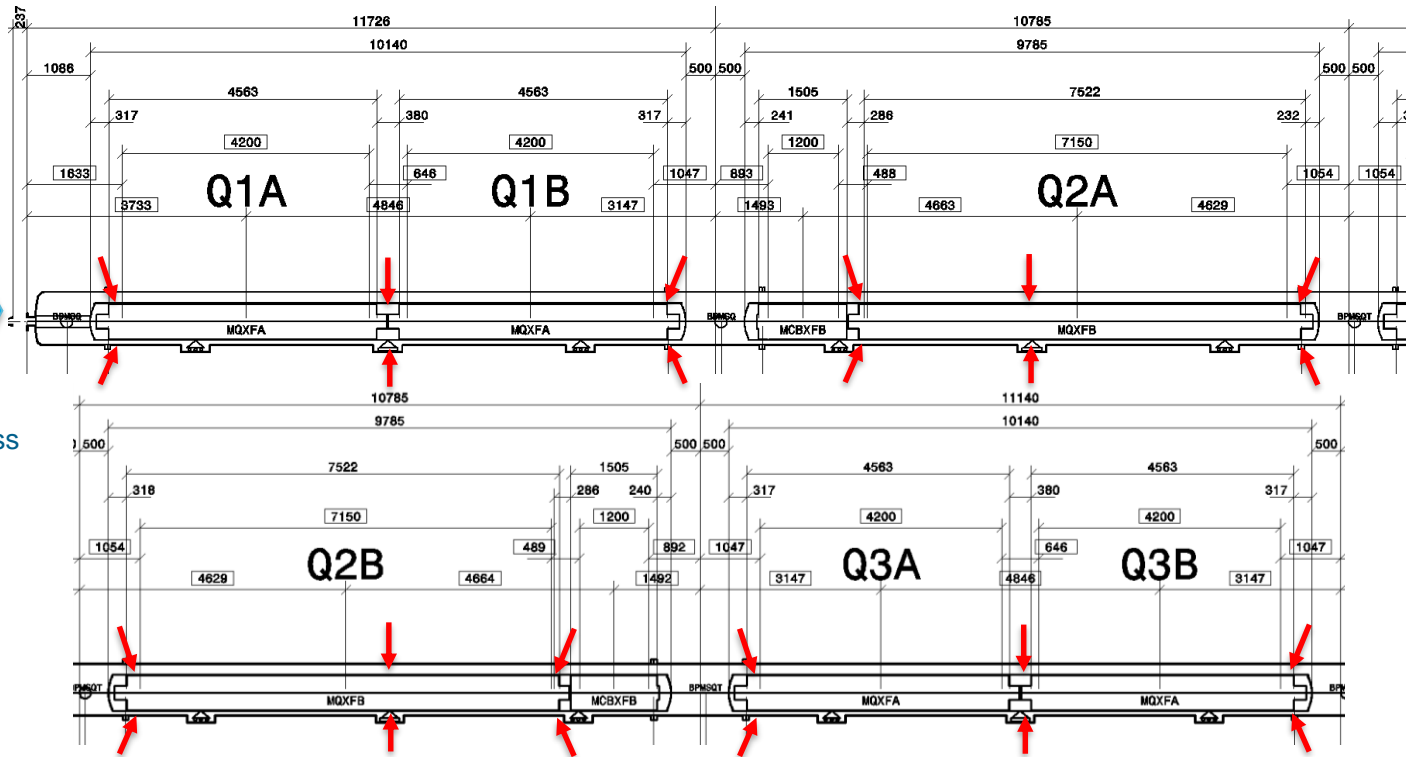


HL-LHC Inner triplet – FSI integration layout



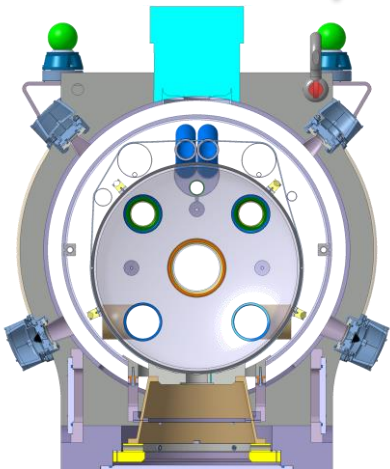
HL-LHC Inner triplet – FSI integration layout

IR5 right example:
LHCLSXH_0010

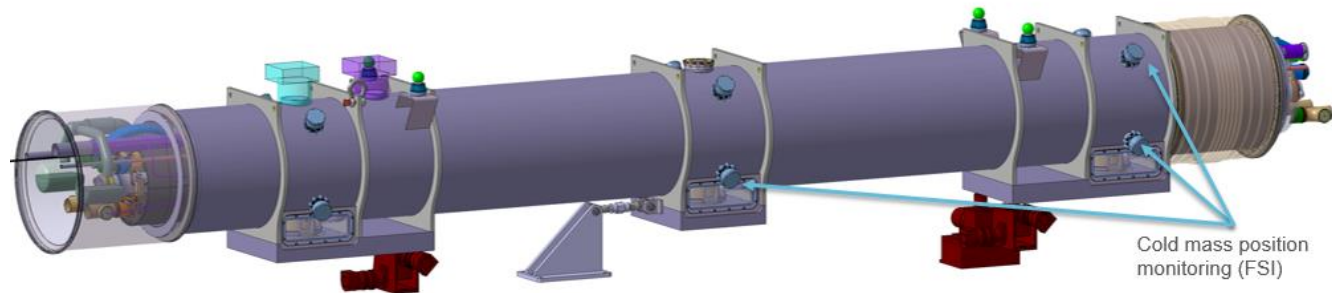


3 x 4 FSI heads
for each cold mass

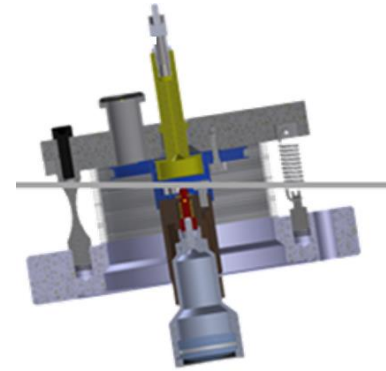
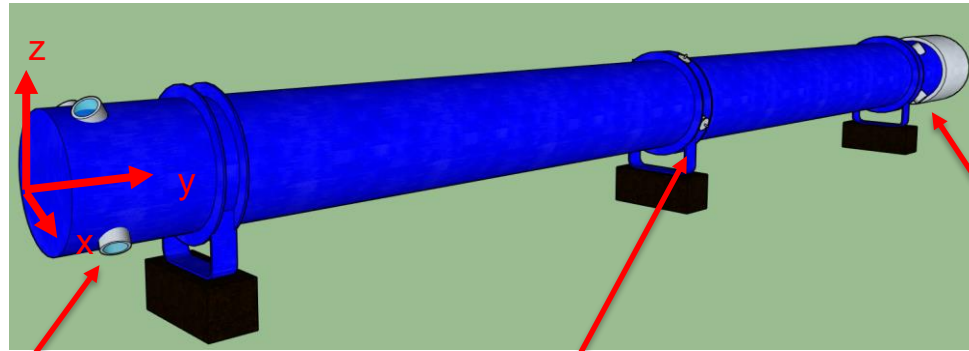
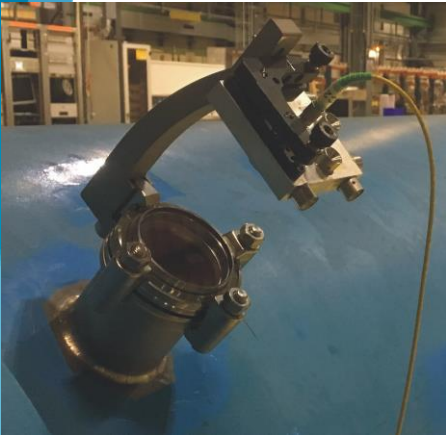
48 FSI heads per triplet needed



D. Ramos



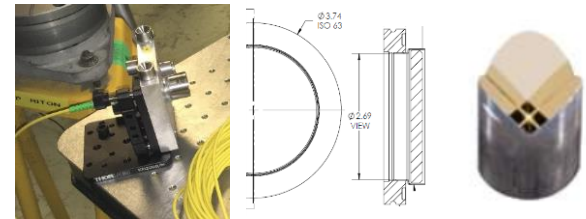
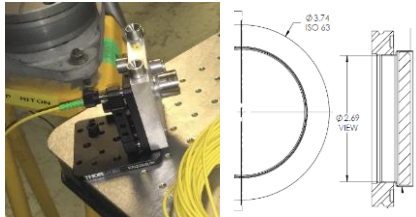
Dipole test



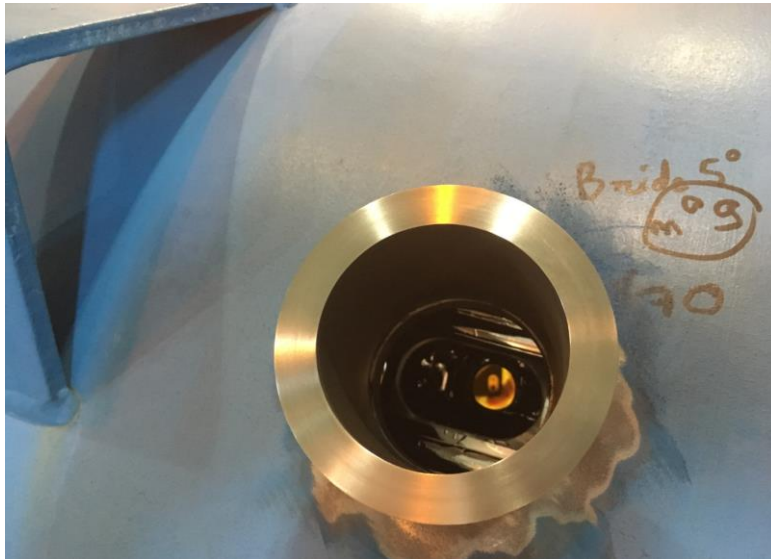
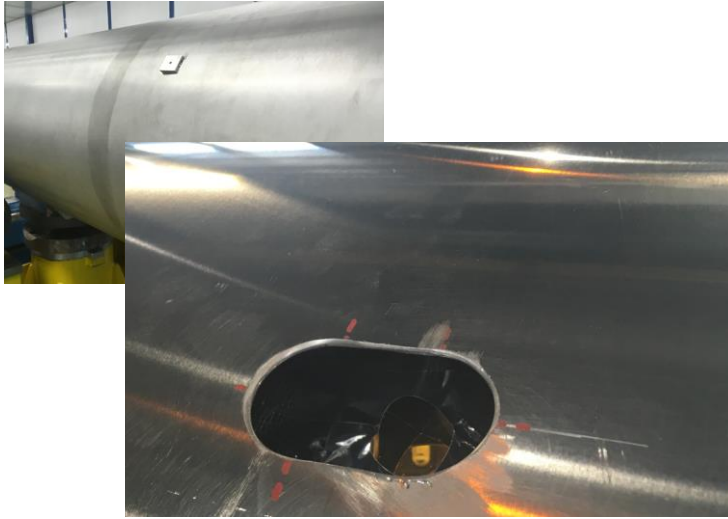
4 FSI mounts + collimators
+ 4 Viewports ISOK DN63
+ 4 Low-cost Alu reflectors

4 Crab Cavity Feedthroughs
+ 4 CCR targets

4 FSI mounts + collimators
+ 4 Viewports ISOK DN63
+ 4 Low-cost Alu reflectors



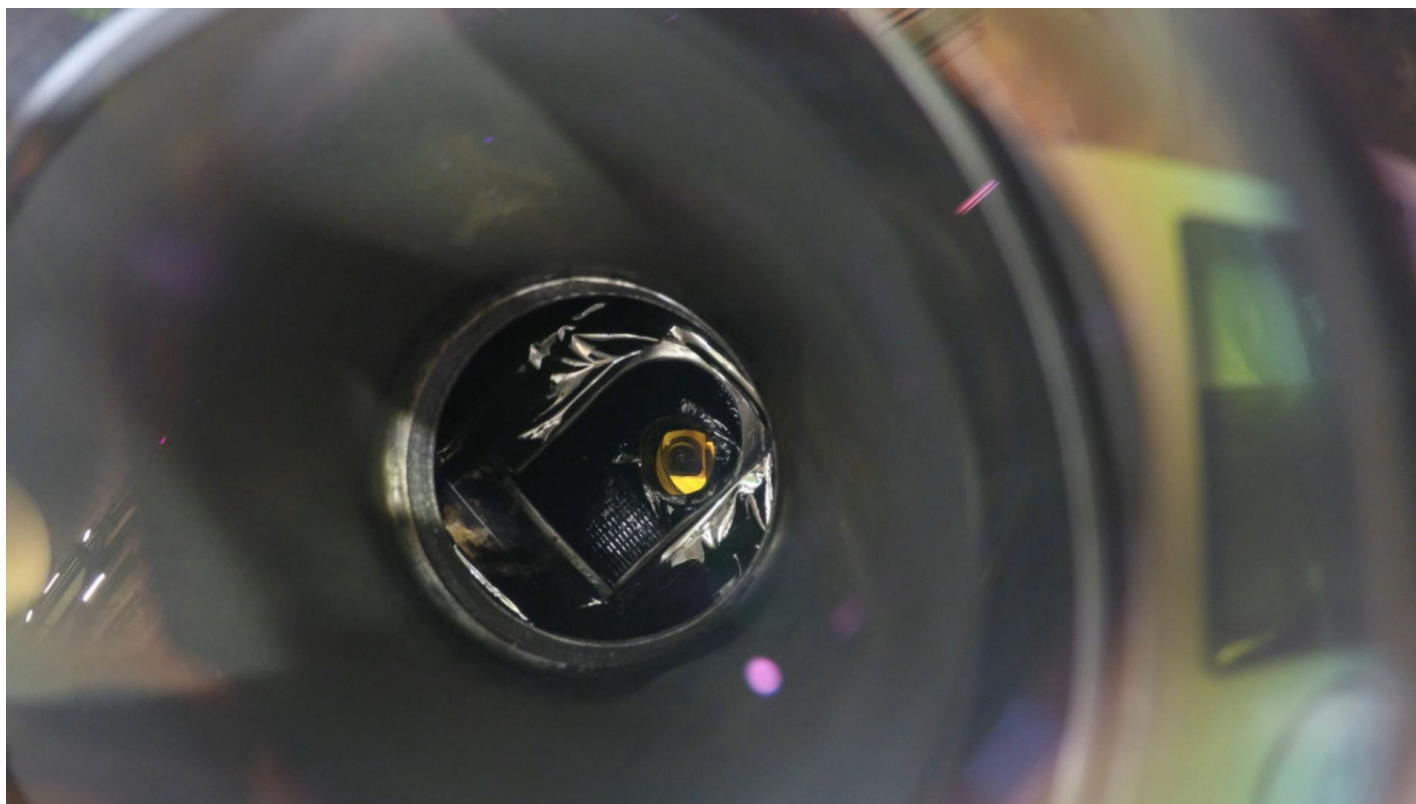
Dipole test - instrumentation integration



Dipole initial tests (Phase 1)

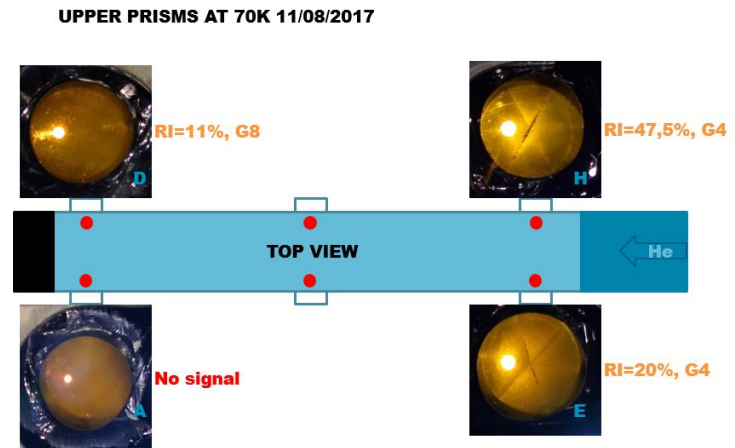
- Cooldown 1 (70K): 4-8/08/2017
- Cooldown 2 (70K): 10-15/08/2017
- Cooldown 3 (70K, 20K): 12/09/2017-11/10/2017

Several different approach verified:
- Nitrogen rinsing of cryomodule
- Cold mass/thermal shield -> different cooling order



Dipole test – Aug-Oct 2017- conclusions

- FSI Etalon system use very sensitive to cryo-condensation effect
- Cryo-condensation strongly dependent on cryostat cleanliness/vacuum quality and cooling scheme



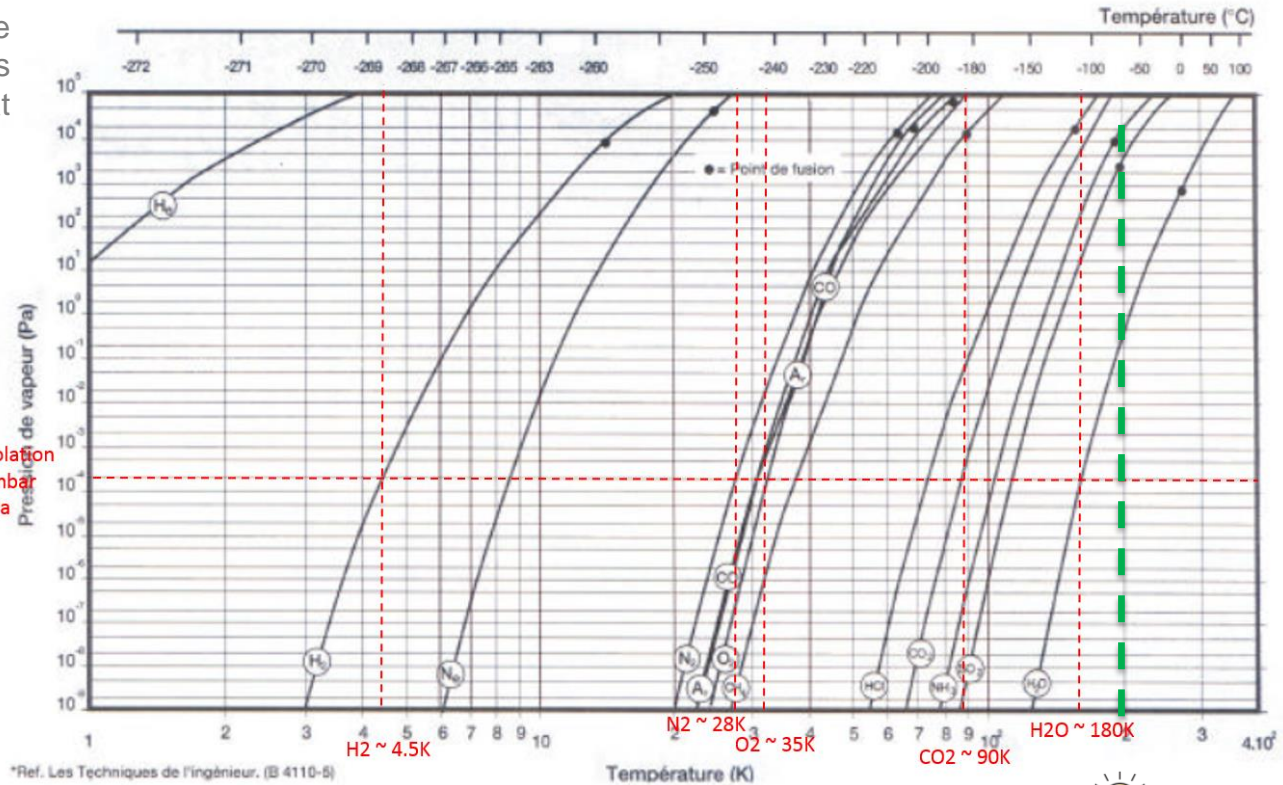
Cryo-condensation - how to cope

At the residual pressure of the insulation vacuum every species naturally present in air condense at or below 180K.

→ If we manage to keep the reflectors at ~200K, no permanent cryo-condensation should occur! ☺

However the reflector should stay rigidly attached to a cold mass cooled down to 1.9K...

Vide isolation
~1E-6 mbar
~1E-4 Pa



Hyp.: we assume that the vacuum vessel pressure is made of 100% of each specie to get the corresponding condensation temperature (i.e. Vacuum pressure = vapor pressure). This is conservative.



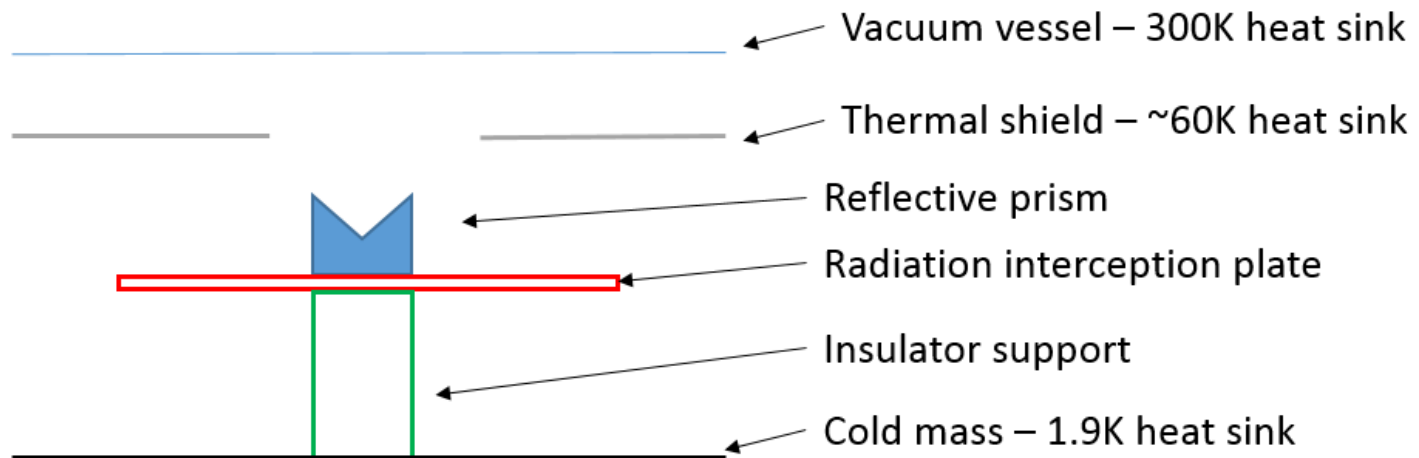
Cryo-condensation - how to cope

How can we achieve a “heating” of the probes up to $\sim 200\text{K}$:

□ Temporary heating – by heating the probe from the outside of the view port (with adequate wavelength) one could in principle temporarily “evaporate” the layer of condensed crystals.

→ technical complexity – not the preferred solution for now.

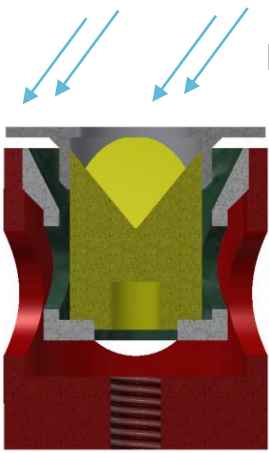
□ Permanent heating – by making sure that the probe stays at $\geq 200\text{K}$, no cryo-condensation should ever take place in principle. This could be achieved using the power radiated from the vacuum vessel (which is 300K “hot”).



Dipole test – initial validation of insulated prism concept (Phase 2)

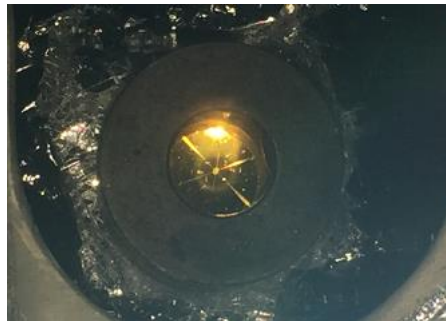
- Exchange of upper prism to prototype ones:
 - Thermally insulated prism with radiation interception surface
 - Prism shielded from the top (to check if particles are freezing due to gravity falling)

INSULATED REFLECTOR



- Cooldown 4 (80K): 27/11/2017-8/12/2017
 - Screen cold, the cold mass heated and cooled down afterwards
 - After several days reference prism and top shielded prism signal lost while measuring using ETALON FSI
 - No problem with insulated prism intensity during whole time of test (ETALON FSI measurement)

INSULATED REFLECTOR

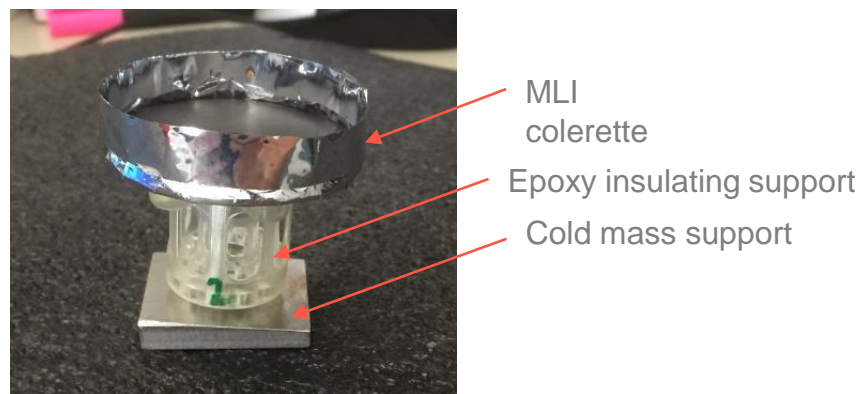
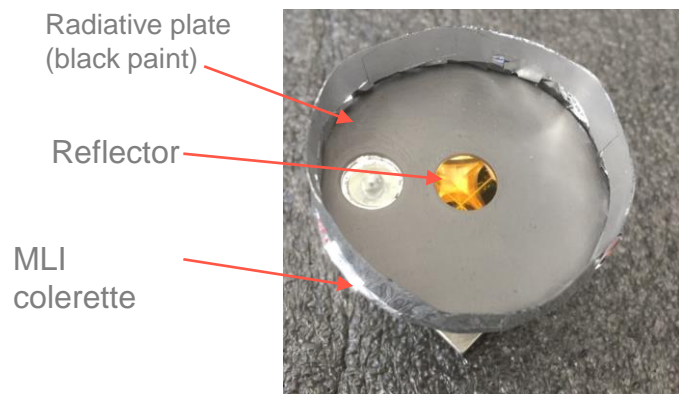
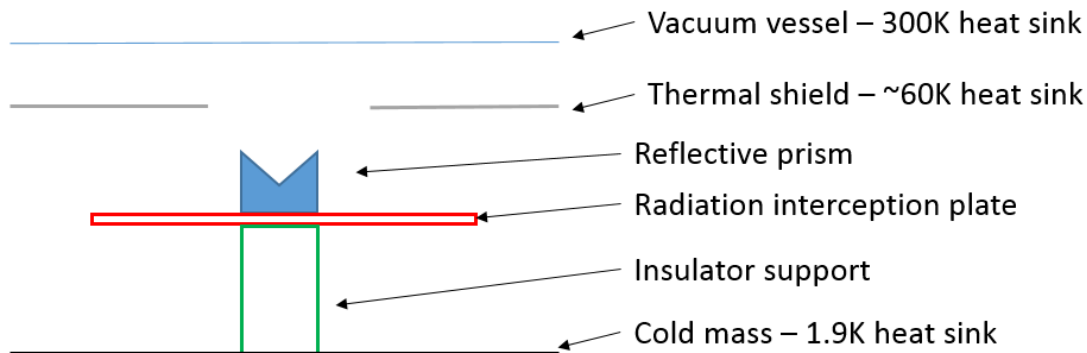


REFERENCE REFLECTOR



Dipole case in-leak on the cold mass

On the dipole test, cryo-condensation issues lead us to use an isolated target support to keep an adequate temperature. A radiative plate is added to intercept radiative in-leak from the V. Vessel to heat up the reflector.



Dipole case in-leak on the cold mass

2 & 3) Flux conducted by the insulating support & flux self-radiated by the “hot” insulating support (FE simulations)

Probe temperature (K)	Power conducted to CM (W)	Power self radiated (W)	Power in-leak to CM (W)
150	0.02	0.01	0.022
200	0.026	0.03	0.032
250	0.044	0.062	0.056

Power to CM = power conducted + 1/5*power radiated

Cold mass thermal in-leaks summary

	Power (mW)
Residual opening	60
Heat conducted by support	56
Total heat to CM	116

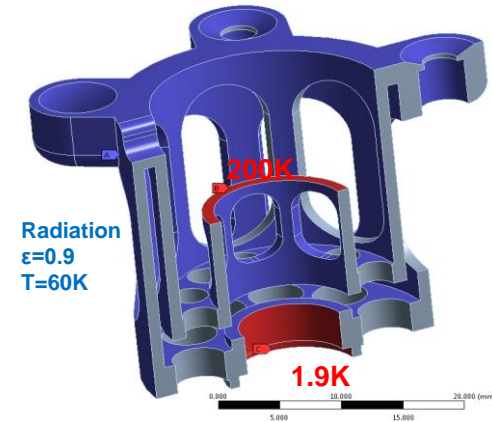
The FSI system is fitted with 12 holes per magnet – 48 holes per triplet

→ $48 \times 0.116 = 5.57W$ (note: this is valid as long as all TS holes are $\Phi 40mm$)

By taking the approx. operating costs issued by TE/CRG
[1W@1.8K](#) ~ 5kCHF for 10 years operation

→ Thermal in-leaks additional operating cost is 28kCHF for 10 years operation per triplet string

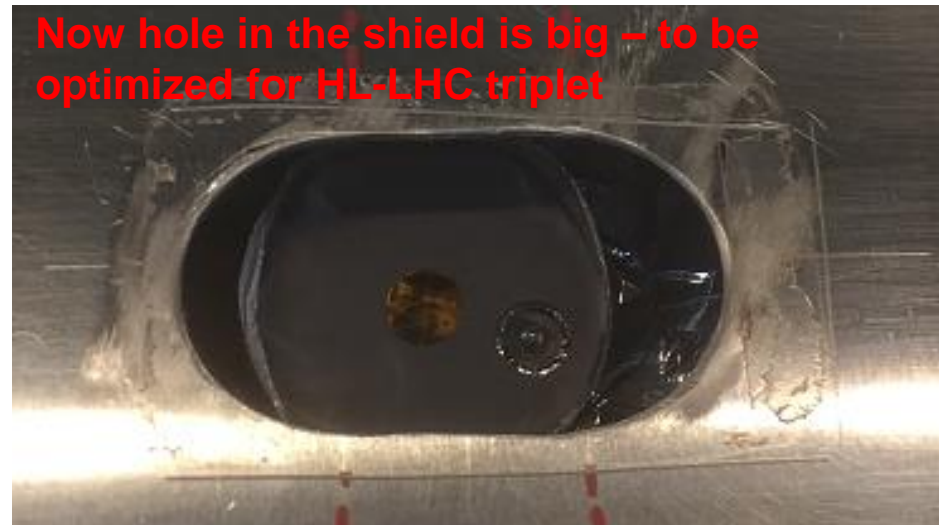
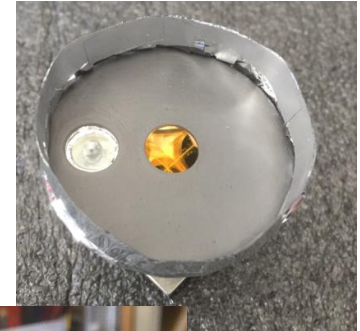
Steady-State Thermal
 Steady-State Thermal
 Time: 1 s
 09/02/2018 14:53
 Radiation: 60 K, 0.9, 1
 Temperature: 200 K
 Temperature 2: 1.9 K



F. Micolon

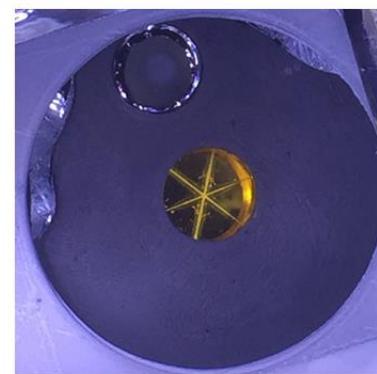
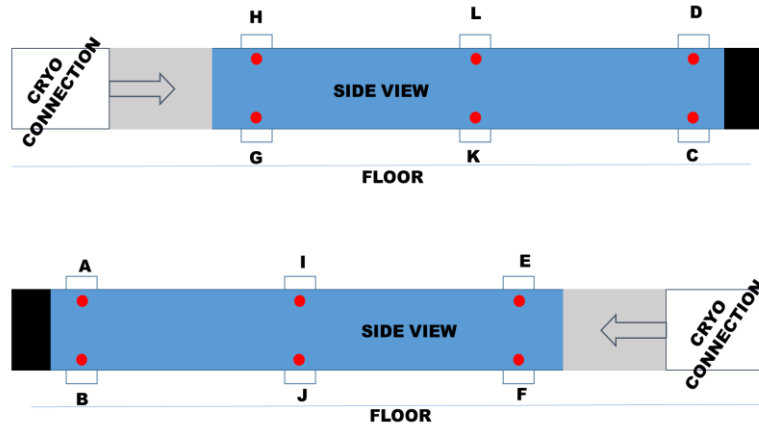
FSI thermal load budget discussed with S. Claudet 20.03.2018 and assessed as acceptable

200K insulated prism prototype design and integration (Phase 3)

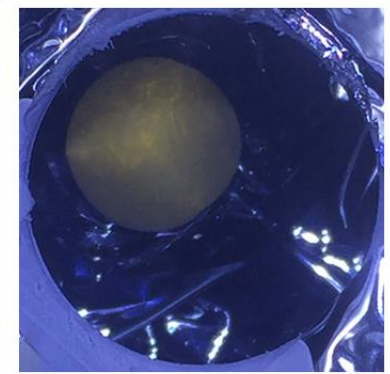


8 insulated targets were then attached to the magnet for testing in February 2018.

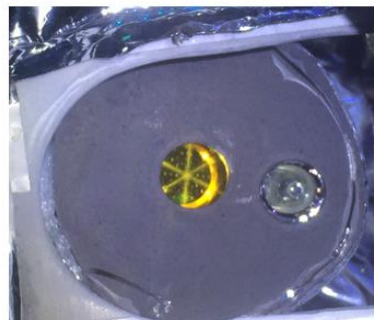
200K insulated prism test results after 3 weeks of cool-down (Phase 3)



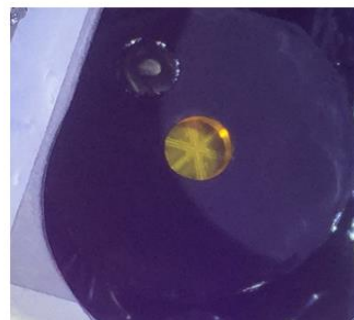
Target L – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
Condensation on S-LAH79, no condensation on CCR.



Target K – no thermal insulation (1" CCR + Φ 4mm S-LAH79 ball).
Thick cryo-condensed 'ice' layer visible on the reflector and ball surface.



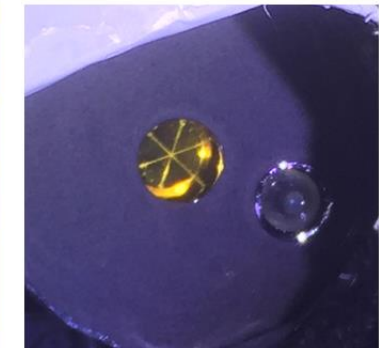
Target H – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
No condensation visible



Target G – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
Light condensation on CCR, no condensation on S-LAH79 ball

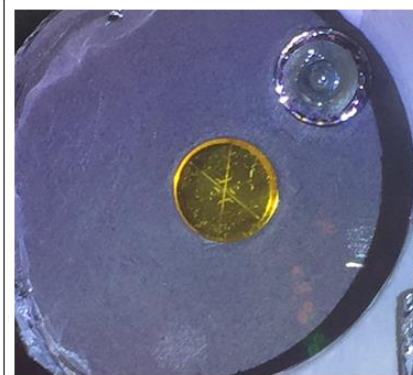
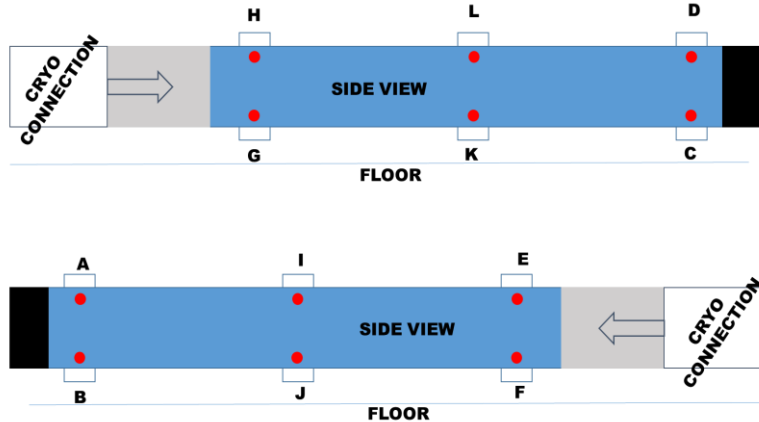


Target D – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
No condensation visible



Target C – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
Light condensation on S-LAH79, no condensation on CCR.

200K insulated prism test results after 3 weeks of cool-down



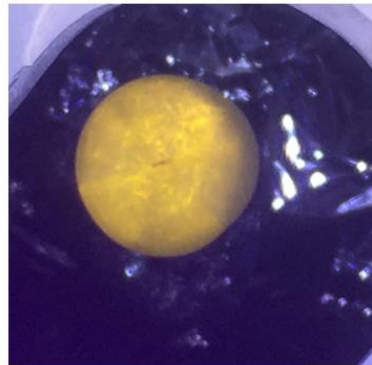
Target A – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
No condensation visible



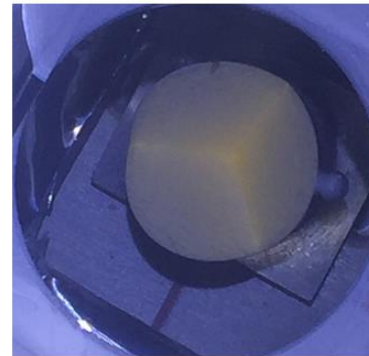
Target B – no thermal insulation (1" CCR).
Thick cryo-condensed 'ice' layer visible on the reflector surface



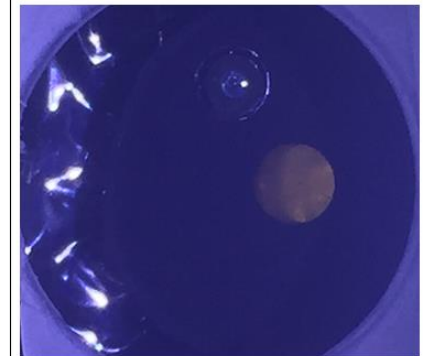
Target E – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
No condensation visible



Target F – no thermal insulation (1" CCR + Φ 4mm S-LAH79 ball).
Thick cryo-condensed 'ice' layer visible on the reflector and ball surface



Target I – no thermal insulation (1" CCR + Φ 4mm S-LAH79 ball).
Thick cryo-condensed 'ice' layer visible on the reflector and ball surface



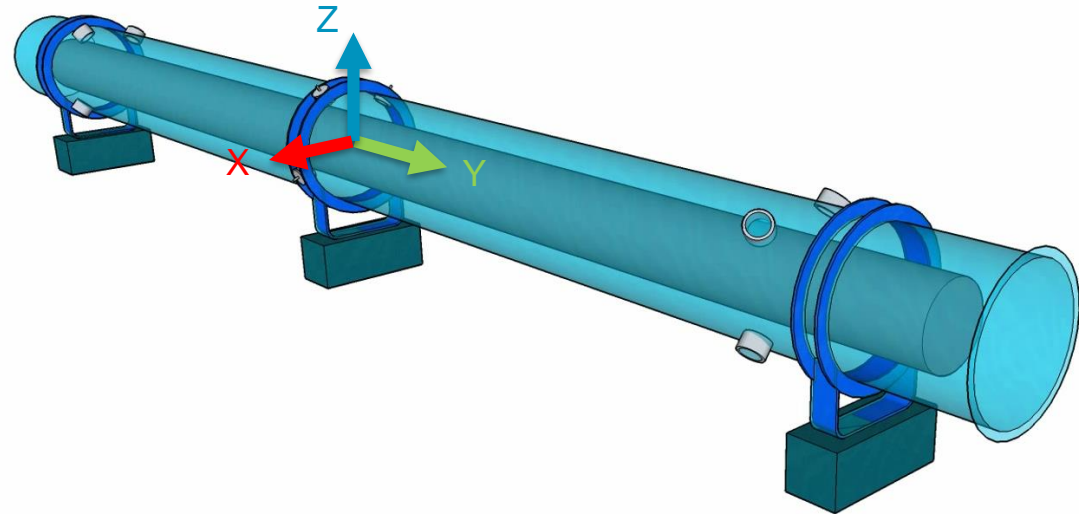
Target J – thermally insulated (0.5"CCR + Φ 4mm S-LAH79 ball).
Light condensation on CCR, no condensation on S-LAH79 ball.

Phase 3 – measurements results

Results of cooling down (phase 3)

- 0 / 4 : CCR or optical retroreflectors
- 7(8)/8 : isolating targets visible (one out of fitting tolerance)

→ Max residual : 7 μ m

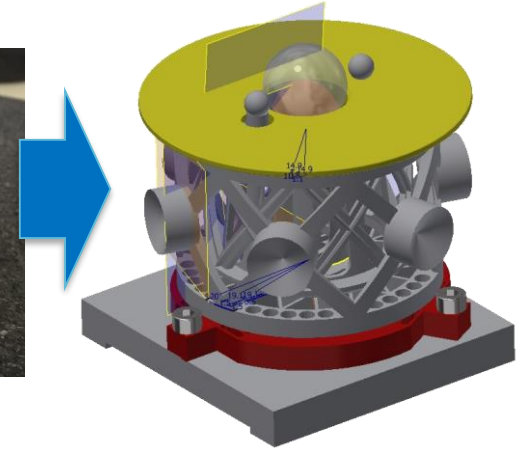


	293 K → 4 K	
Tx : radial (mm)	-0.001 mm	+/- 0.066 mm
Ty : longitudinal (mm)	0.825 mm	
Tz : vertical (mm)	-1.078 mm	+/- 0.023 mm
Rx : pitch (rad)	0.000003 rad	+/- 0.000004 rad
Ry : yaw (rad)	-0.001050 rad	+/- 0.001282 rad
Rz : roll (rad)	-0.000022 rad	+/- 0.000003 rad
F : scale factor	-0.002889	+/- 0.000015

} **Coherent with the simulation**

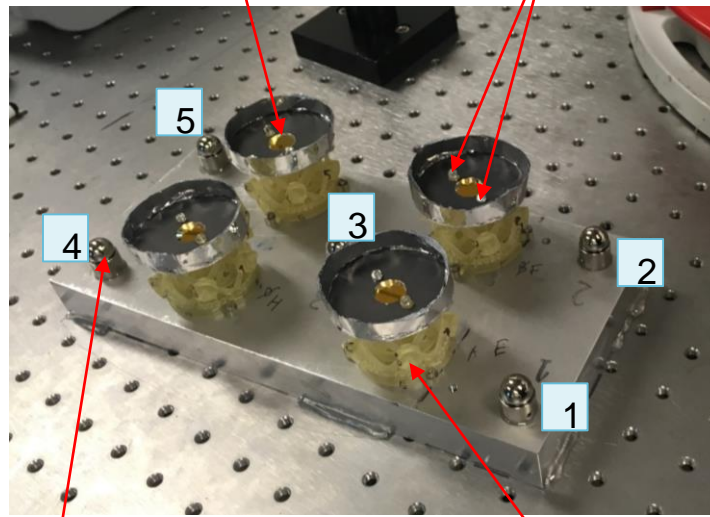
Insulated prism – rigid prototype design and integration (Phase 4)

- Phase 4 to fully validate metrological strategy of IT cold mass FSI measurements
 - Use of new rigid insulated target support
 - Check of cost optimized reflectors (n=2 glass balls)
 - Check of new technology of multi-target FSI



Newport reflector

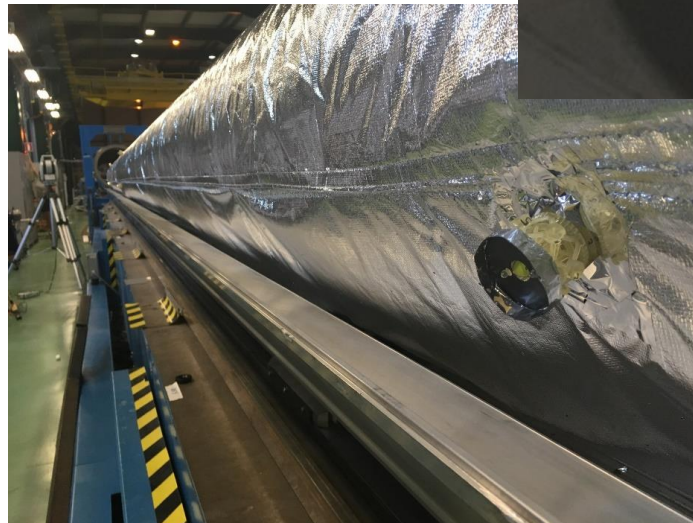
Glass spheres



Fiducials on the plate Fiducials around the mount

Insulated prism – rigid prototype design and integration (Phase 4)

- Phase 4 current status
 - Fiducialisation of the dipole cold mass, cryostat performer mid of September
 - Cool-down started Begin of October
 - For now the dipole is at 4K (no cryo-condensation)



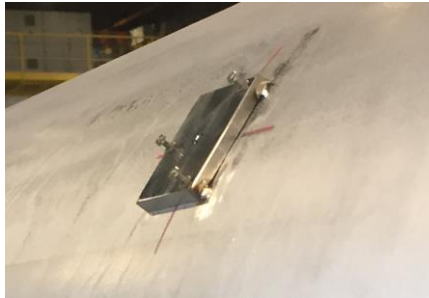
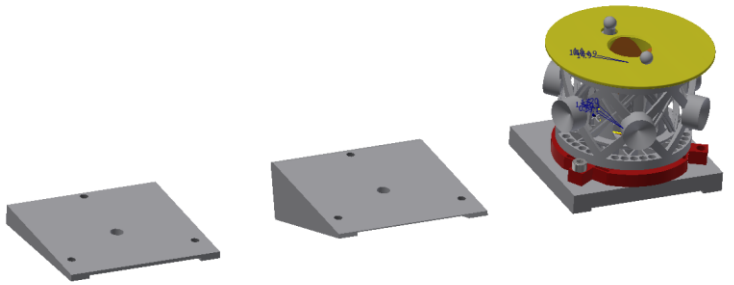
Latest results – Phase 4 cool down



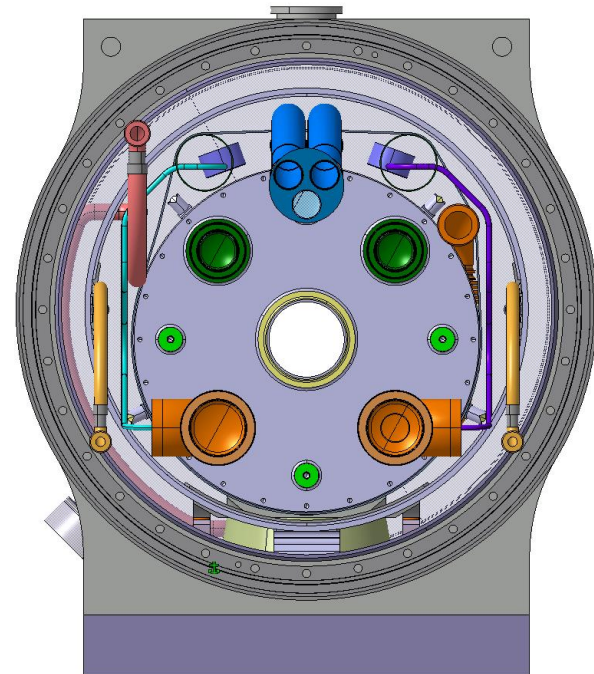
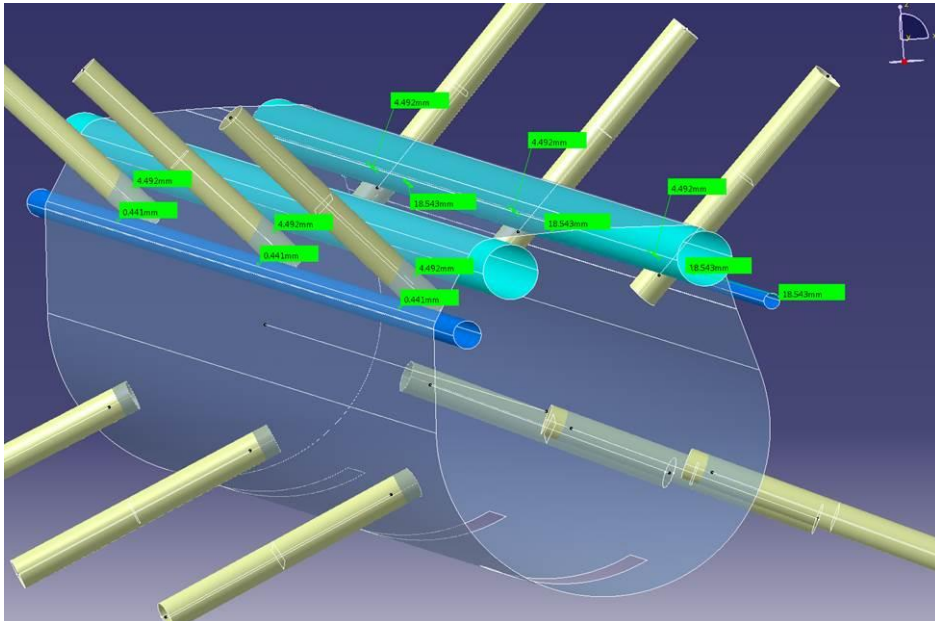
	October 2018 PHASE 4 293 K → 4 K	
Tx : radial (mm)	0.016 mm	+/- 0.138 mm
Ty : longitudinal (mm)	0.266 mm	+/- 0.126 mm
Tz : vertical (mm)	-0.975 mm	+/-0.045 mm
Rx : pitch (rad)	0.000012 rad	+/- 0.000009 rad
Ry : roll (rad)	0.000781 rad	+/- 0.002861 rad
Rz : yaw (rad)	0.000020 rad	+/- 0.000005 rad
F : scale factor	-3109 ppm	+/- 17ppm
Sag (radial)	-0.069 mm	+/- 0.024 mm
Sag (vertical)	0.034 mm	+/- 0.042 mm

Successful test at 4 K of the new support of targets!
Cryo-condensation issue solved!

HL-LHC triplet FSI integration

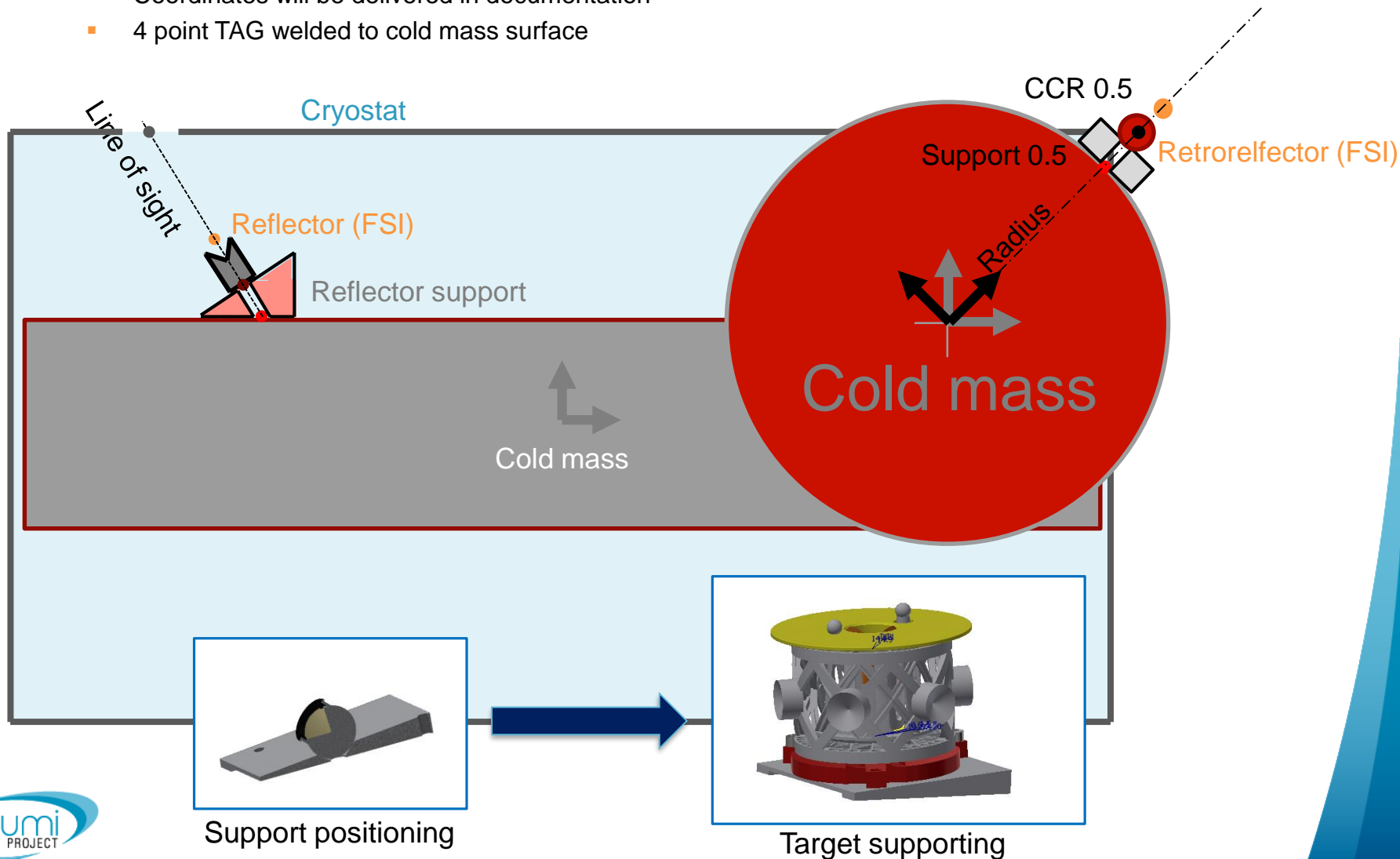


- Final design of HL-LHC IT FSI target after finalisation of Phase 4 test (begin 2019 and onwards)
- Cold mass interface is a 316L machined plate, to be welded on cold-mass surface
- Integration within cryostat ongoing (space reserved, cryostat flanges done, MLI/thermal shield to be finalized)



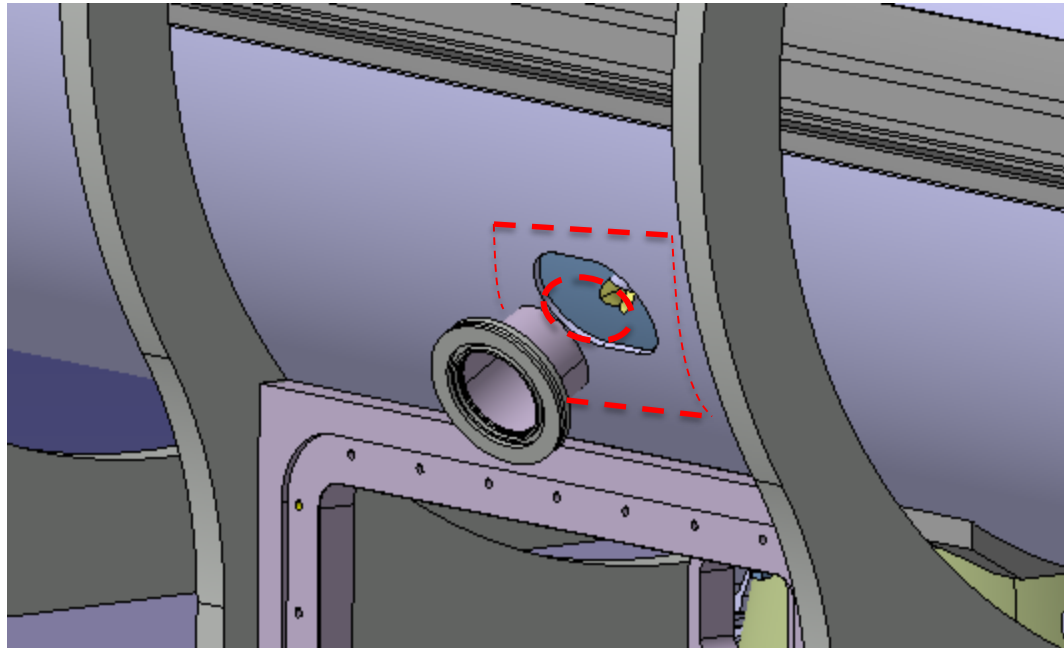
HL-LHC triplet FSI assembly

- Target support welding
 - Targets support positioning with help of integrated fiducial for 0.5" CCR (or CCR adapter)
 - Coordinates will be delivered in documentation
 - 4 point TAG welded to cold mass surface



HL-LHC triplet FSI assembly

- Thermal shield hole positioning approach – under design and integration
 - Bigger hole in the thermal shield to allow line of sight considering lower machining tolerances of thermal shield
 - Fine adjustable plate point welded (or riveted) during cryostating
 - Plate positioning with help of delivered reference tooling or with use of fiducials integrated on plate (under discussion)

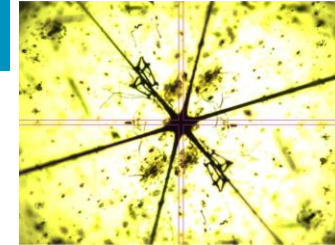


Pre-test achieved on FSI

- Validation of targets through irradiation & vacuum and cold tests



10 MGy



Liquid nitrogen test:

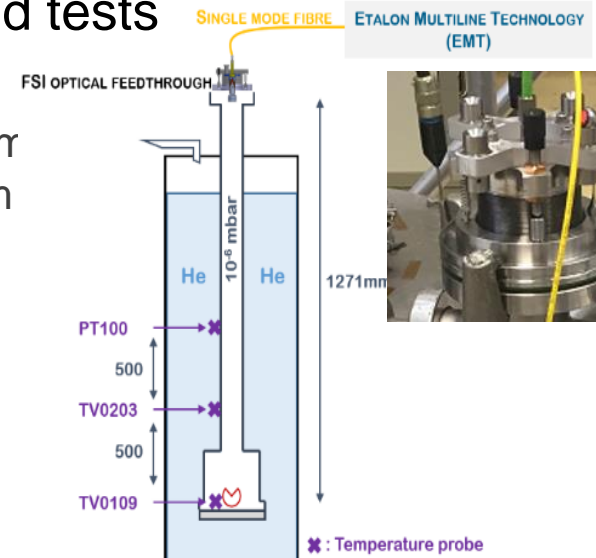
- No damage of targets
- No loss of performance

Radiation tests of BMRs :

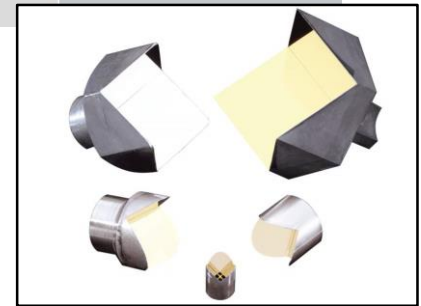
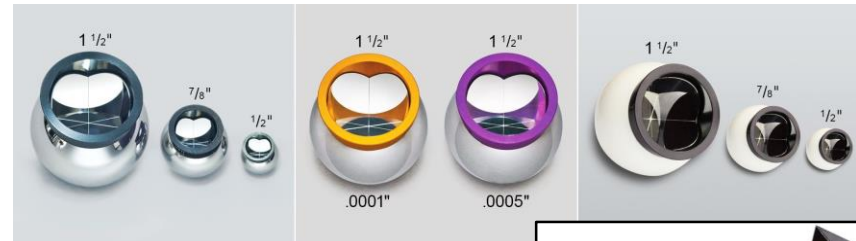
- Ceramic BMRs and collimators validated with TID of 10MGy
- BMR mirror centricity lost $\sim 20\mu\text{m}$

- Validation of measurement chain through cold tests

- No visible deformation of the feethrough
- Decrease of intensity but no impact on the measurer
- Comparison with AT401 measurements within $20\ \mu\text{m}$



Reflectors validation and optimisation



- Cheap, hollow retroreflector already used in dipole test
- Refractive index ≈ 2 glass ball as a alternative to hollow retroreflectors or replicated reflectors (~40€ vs. ~2k€ vs. 300 €)
- Balls glasses behaviour under investigation (impact on sweeping interferometry, laser tracker measurements, ...)
- Radiation tests of hollow and glass reflectors started in August 2018

Summary

- The special design of targets (temperature optimized) allows for use ETALON FSI without cryo-condensation effect, with thermal load at reasonable level
- Dipole measurement results confirm
- Further works on optimization of the insulated targets and their integration within HL-LHC triplet cryostat ongoing
 - Final HL-LHC target design after Phase 4 dipole test



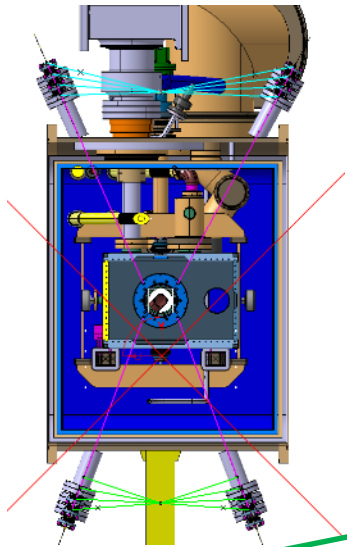
Thank you!



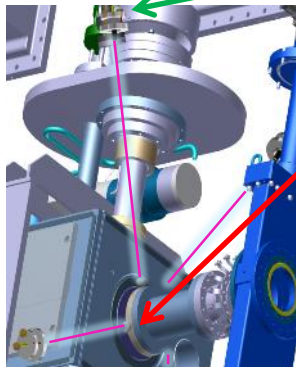
Spare slides

Crab cavity FSI monitoring system

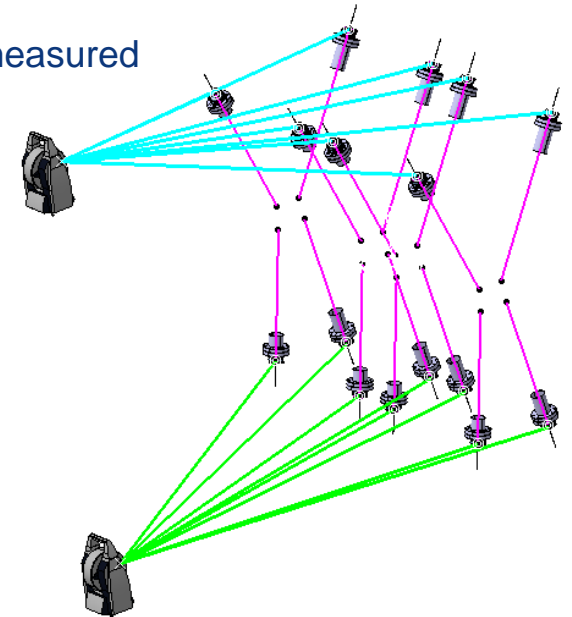
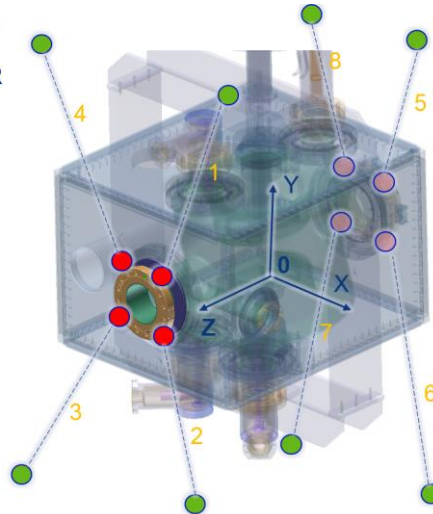
FSI solution – system configuration



- 6 targets per cavity is required to calculate cavity position and orientation (least mean square method).
Cavity fiducialisation data and measured distances between all FSI heads and centres of CCR targets are used
- Current design assumes 4 targets per flange (8 Corner Cube Retroreflectors [CCR] per cavity) to provide measurements redundancy
- Dressed cavity have to be fiducialized (known cavity geometry w.r.t. reference targets)
- Positions of the FSI heads have to be measured



● FSI
● CCR



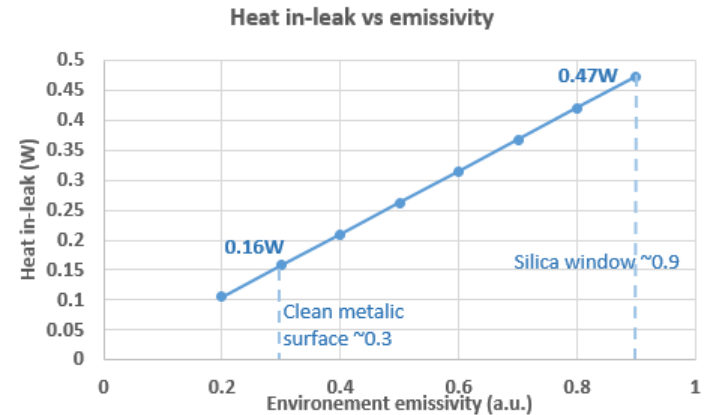
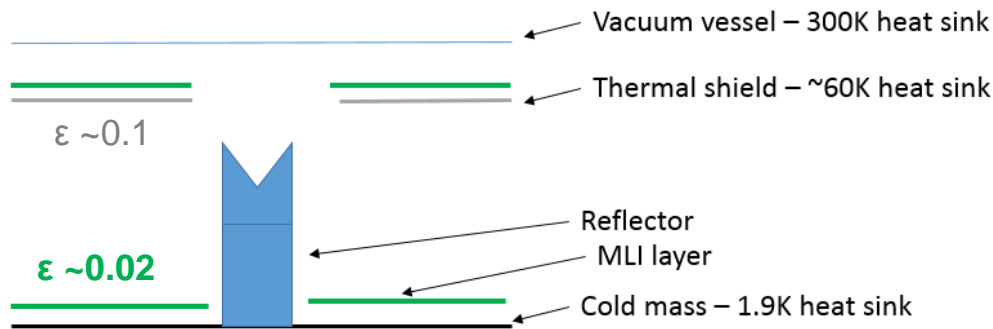
200K insulated prism design, heat in-leak from the thermal shield openings optimization

Hyp. For the dipole – the hole are $\Phi 40\text{mm}$ on the center

The maximum heat radiated from the environment :

$$P_{in} = \sigma \cdot \epsilon_E \cdot S_{opening} \cdot (T_{ambient}^4 - T_{rad_surface}^4)$$

Assuming $\epsilon_E=0.9$, the power inlet is $\sim 0.5\text{W}$ per $\Phi 40\text{mm}$ hole ($\sim 400\text{W/m}^2$)



The ratio of emissivity between the thermal shield inner surface and the MLI covering the CM is ~ 5 .

Thus we assume that the flux leaked in the hole will be absorbed at :
 80% by the thermal shield inner surface
 20% by the cold mass

→ $\sim 0.1\text{W}$ per $\Phi 40$ hole on the cold mass ($\sim 80\text{W/m}^2$)

Dipole case in-leak on the cold mass



The MLI *colerette* attempts to :

- focus as much radiation as possible on the heat interception plate.
- close the direct view between the cold mass and the vacuum vessel to limit the in leak

The remaining cold mass in-leaks are :

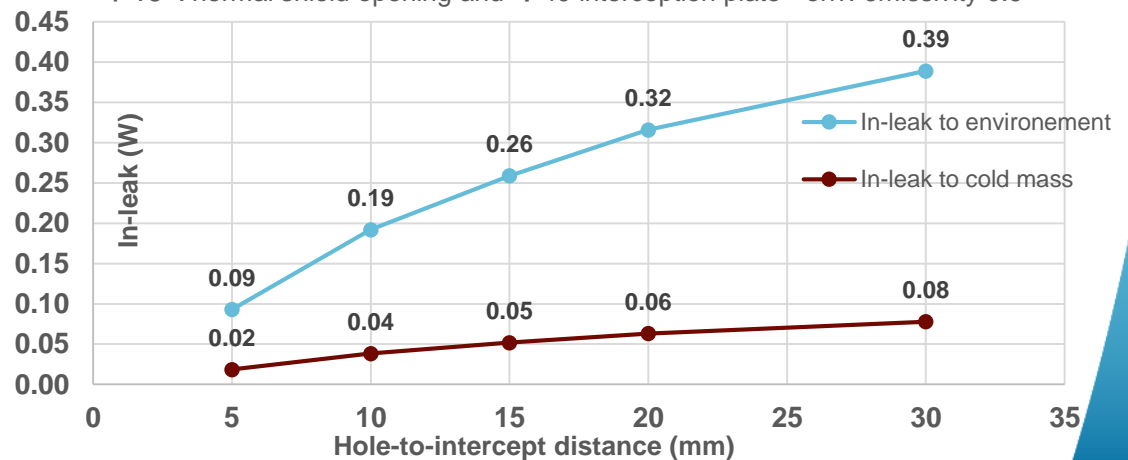
- 1) Residual opening between TS and heat intercept
- 2) Flux conducted by the insulating support
- 3) Flux self-radiated by the “hot” insulating support

Residual opening between TS and heat intercept (With a $\Phi 40$ TS opening and $\Phi 40$ interception plate):

Thermal in-leak vs. hole-to-intercept distance

→ The closer the colerette is from the thermal shield the better
...but we should make sure to avoid permanent contact between the colerette and the inner thermal shield however.

$\Phi 40$ Thermal shield opening and $\Phi 40$ interception plate - env. emissivity 0.9



Latest results – Phase 4 cool down

	Parameter	Value
IN	X (mm)	0.569
	Y (mm)	-6200.536
	Z (mm)	71.602

	Parameter	Value
CENTER	X (mm)	2.286
	Y (mm)	0.306
	Z (mm)	73.991

	Parameter	Value
OUT	X (mm)	4.053
	Y (mm)	4984.820
	Z (mm)	76.003

Residual	Value (μm)
Col A	39
Col B	-19
Col C	36
Col D	-31

Scale factor : 0.996920

Residual	Value (μm)
Col I	6
Col J	-5
Col K	4
Col L	-4

Scale factor : 0.997129

Residual	Value (μm)
Col E	12
Col F	-5
Col G	11
Col H	-9

Scale factor : 0.996903

