

Gas expansion and nozzle shape

Barbara Kędzierska BGC Collaboration Meeting, 27.11.2018

Outline

- •Gas expansion in a nozzle
 - slightly underexpanded jet
 - highly underexpanded jet
 - overexpanded jet
 - supersonic gas jet expansion into the molecular vacuum
- •Nozzle shapes
 - different possibilities
 - design methodology

Conclusions

medium vacuum conditions

Gas jet

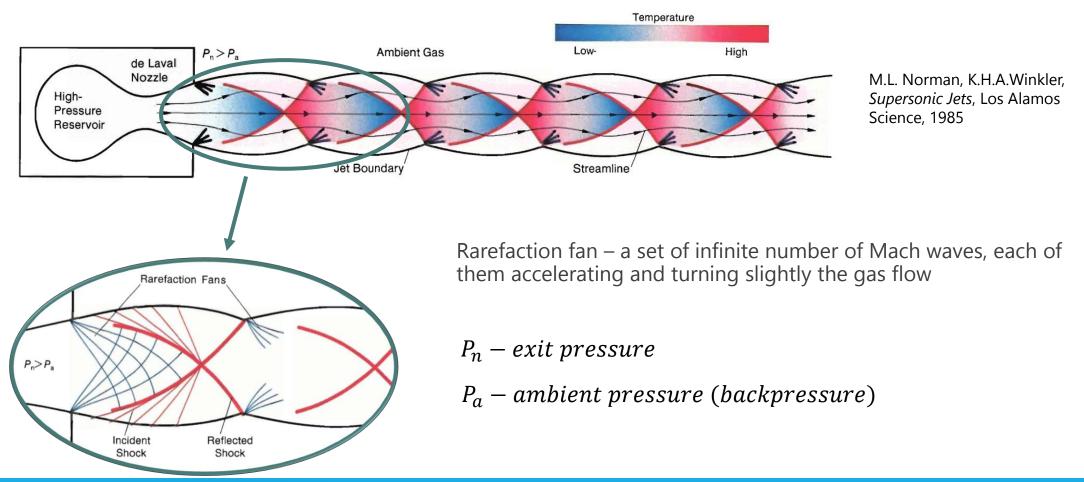
GAS EXPANSION AND NOZZLE SHAPE

BARBARA KĘDZIERSKA

Theory

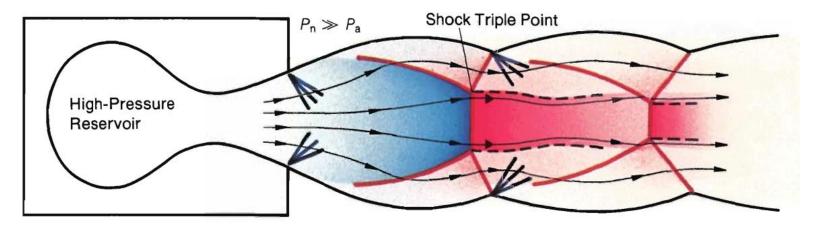
- 1) underexpanded gas jet the gas leaves the nozzle at pressure higher than ambient (backpressure)
 - 1) slightly underexpanded
 - 2) highly underexpanded
- overexpanded gas jet the gas leaves the nozzle at pressure lower than ambient (backpressure)
- Pressure in the container must be obviously always higher than backpressure.





GAS EXPANSION AND NOZZLE SHAPE

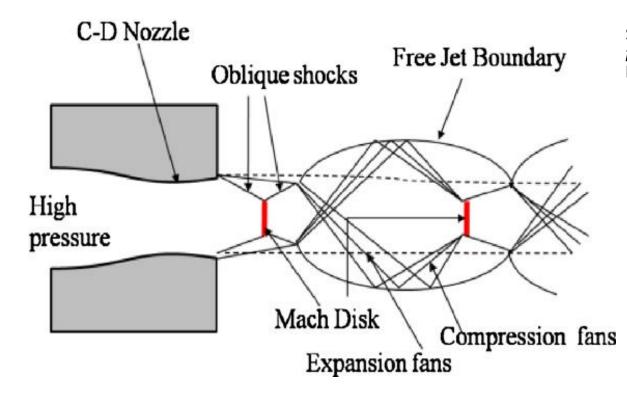




M.L. Norman, K.H.A.Winkler, *Supersonic Jets*, Los Alamos Science, 1985

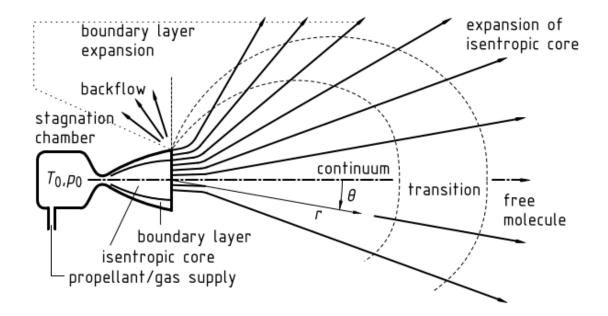
$$\frac{L}{D} = 0.65 \cdot \left(\frac{p_0}{p_{inf}}\right)^2 \qquad \text{for } PR = \frac{p_0}{p_{inf}} = 20 \div 200$$

Overexpanded jet

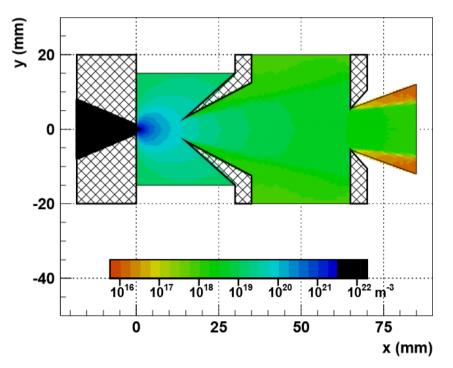


S.M.V. Rao, J. Gopalan, Acetone planar laser-induced fluorescence for supersonic flow visualization in air and nitrogen gas, https://www.researchgate.net/publication/270762073





M. Grabe, *Numerical simulation of nitrogen nozzle expansion using kinetic and continuum approaches*, conference paper, September 2008, https://www.researchgate.net/publication/224999345



A. Naβ, E. Steffens, *Direct Simulation of Low-Pressure Supersonic Gas Expansions and its Experimental Verification*, Preprint submitted to Elsevier Preprint, 28 May 2018

Conclusions for BGC

- We are in molecular flow regime. Backpressure is very low.
- There is no force which can push the gas back towards the centre.
- Normal shock wave cannot be formed.
- There are no Mach disks.

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Nozzle shape

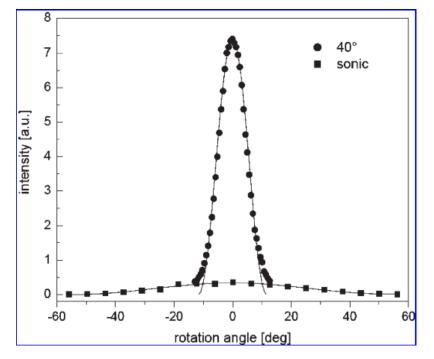
BGC nozzle design

- For BGC purposes we want to have high local gas density (through the third skimmer)
- We are dealing with vacuum conditions → satellites analogy was obvious
 - NASA reports on this issue are difficult to be found
 - reports from aerospace research centers have been studied
 - original Rao's publications
 - Aerospace Testing Alliance

Possible designs (1) – simplest options

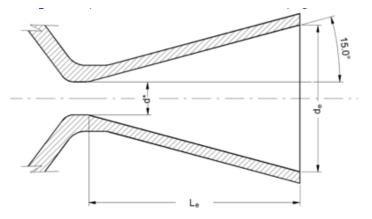
SIMPLE HOLE (SONIC NOZZLE)

gives a subsonic flow



CONICAL NOZZLE

- gives supersonic flow
- much bigger density



Helium density after 200 mm for 20-mm-nozzles at 50 bar

K. Luria, W. Christen, U. Even, *Generation and propagation of intense supersonic beams*, The Journal of Physical Chemistry, A 2011, 115, 7362-7367



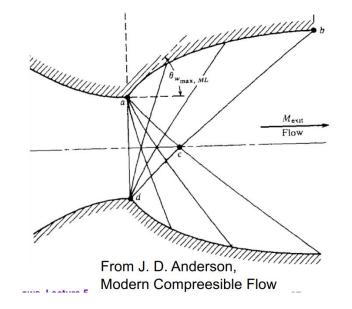
NOZZLE WITH EXPANSION SECTION

- better quality flow
- preferred for wind tunnels

From J. D. Anderson, Modern Compreesible Flow

MINIMUM LENGTH NOZZLE

- no expansion section
- preferred for rocket engines

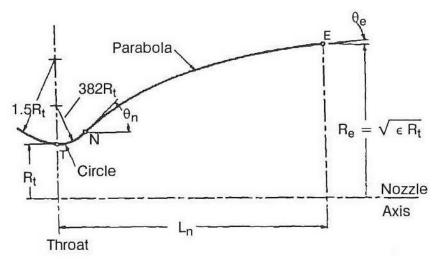


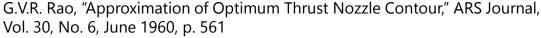
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Possible designs (3)

RAO DESIGN

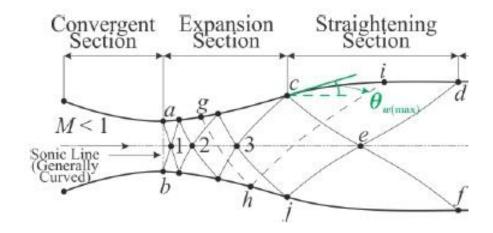
- simplified shape of Thrust Optimized Contour
- ready-made formula





IDEAL CONTOUR

- produces uniform axial flow
- expansion is slow



B. Vallabh, B.W. Skews, *Approximation of nozzle contours in the CSIR supersonic wind tunnel*, R&D Journal of the South African Institution of Mechanical Engineering 2017, 33, 32-41

Conclusions

- Simple hole (subsonic) nozzle gives the lowest density.
- Conical nozzle requires the smallest attachment angle and thus greater length is needed to expand to the same pressure.
- Optimized shapes can be used, further research will be conducted to choose the best one.
- Suggestions about sources for space nozzles?