

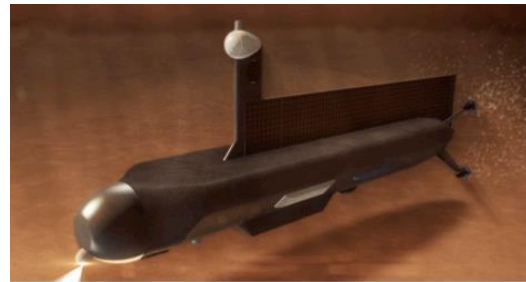
Pulse Tube Coolers for Earth Observation & Science



Introduction – why Space cryogenics ?

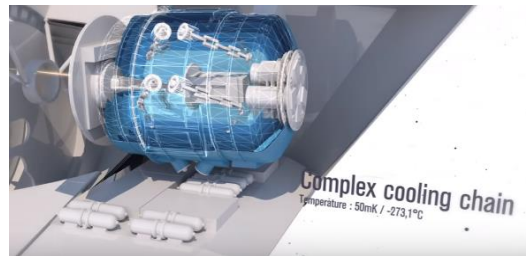
- Imposed by the environment

- ❖ Example : explore Saturn's moon Titan



- For the purpose of observation performance

- ❖ Detector cooling for IR in earth observation
- ❖ Detector cooling for science payloads
- ❖ Storage of biological samples



- For the purpose of propulsion optimization

- ❖ Cryogenic propellant for launchers
- ❖ Cryogenic propellant for deep space exploration



Air Liquide : A world leader in Space Cryogenics

25 years experience



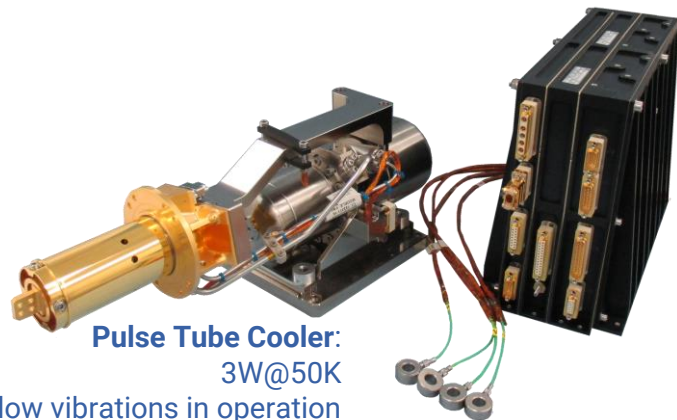
MELFI:

Turbo Brayton cooler 100W at -80°C for ISS
3 models operating in orbit with >80000 hour cumulated operation



HERSCHEL Superfluid Helium tank:

2400 L superfluid He
3.5 year autonomy
1.6K during all the mission



Pulse Tube Cooler:

3W@50K

Very low vibrations in operation
Lifetime = 10 years in orbit
24 flight coolers for 4 instruments

Planck Dilution Cooler:

From 1,6 K to 0,1 K

Launch lock mechanism using
shape memory alloy
2.5 year autonomy (extended)
20,000 hours continuous
operation : no failure!



Acknowledgements

Contributors to the work presented here :

ALAT team : René GARY, Diogo LOPES, Guillaume DARQUE, Jean-Michel NIOT, Simon CARPENTIER, Ian PENNEC, Guillaume BODOVILLE, Pascal BARBIER,...

Customer dream team : Thales Alenia Space, Airbus Defence & Space, OHB, CNES, ESA

Airbus CRISA, SITAEL : development of the electronics



Thales Cryogenics b.v. : development of the compressors



CEA SBT : cold fingers license



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Space coolers for Earth Observation projects

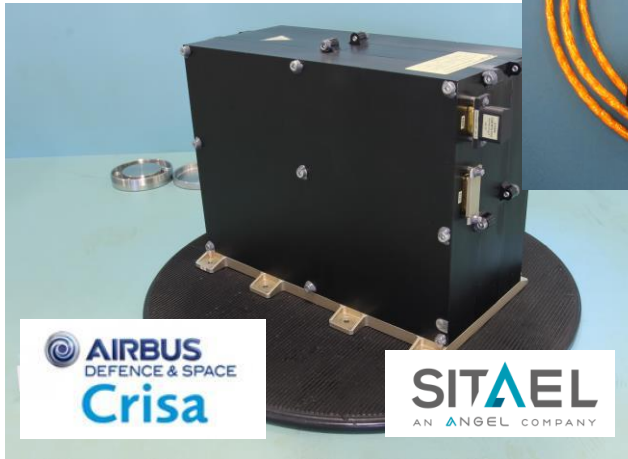
Air Liquide and Space Coolers

- Starting from 2009, The R&D effort started with the support of ESA and CNES concretised with orders from several programs:
 - ❖ French national : 6 flight coolers for Earth observation (delivered)
 - ❖ MTG: 12 flight coolers on two meteorological instruments (FCI and IRS)
 - ❖ METOP-SG: 6 flight coolers on IASI-NG instrument
- Altogether, 24 flight coolers are to be manufactured, assembled, acceptance tested and delivered in a very short timeframe (3 years)
- To date, 10 flight coolers have been delivered, 2 are flying.

Air Liquide and Space Coolers MTG Project & IASI-NG project

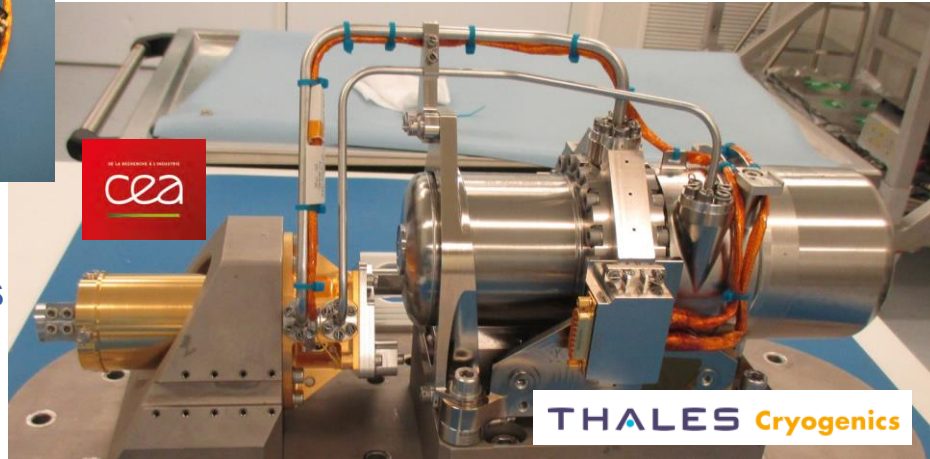
- A Large Pulse Tube Cooler (LPTC) is composed of

Cooler Control
Electronics (CCE)



Telemetry and
Power harness
(TPH)

Cooler Mechanical
assembly (CMA)

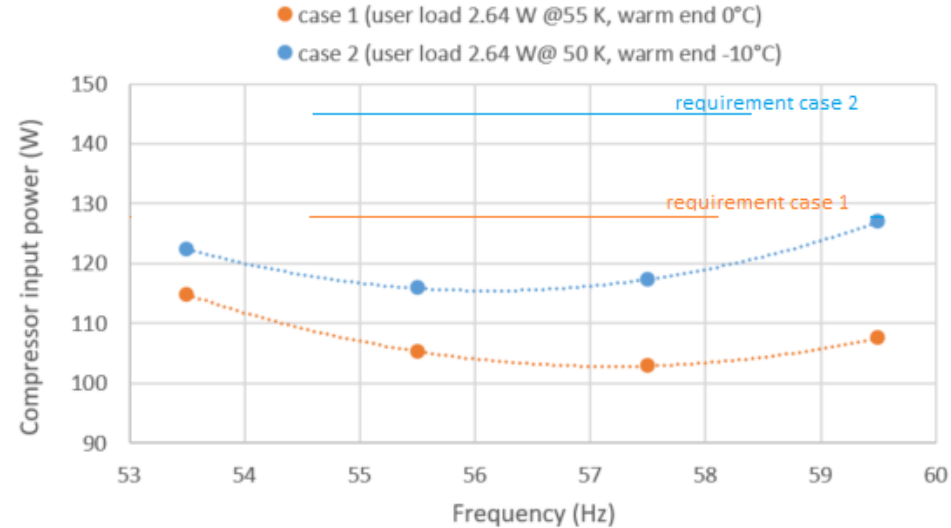
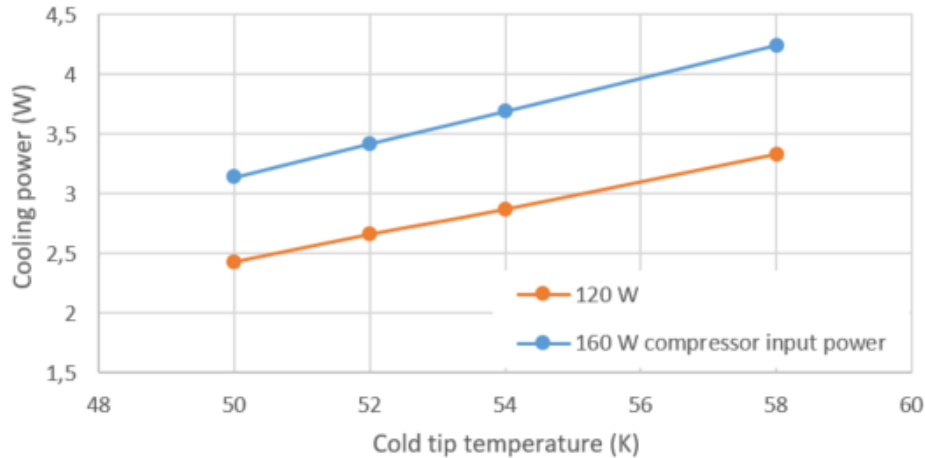


Air Liquide is responsible for the cooler system

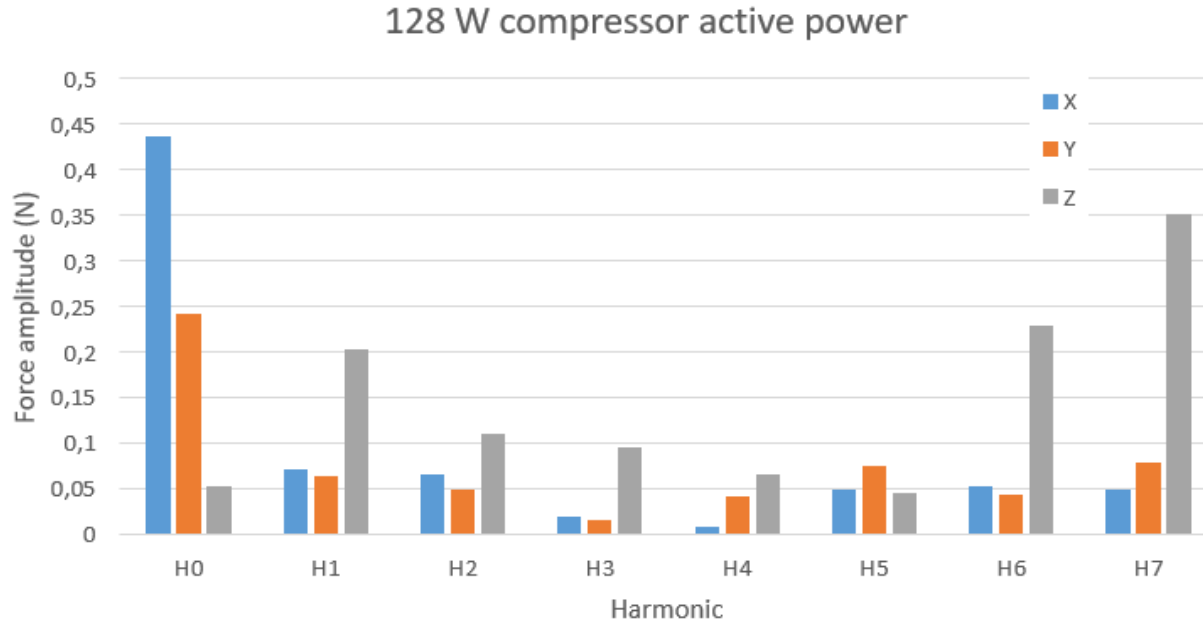
- Air Liquide handles the cooler technical specification, including electronics
 - ❖ Detailed design of the cold finger, buffer, inertance, split pipe
 - ❖ Flow down of specifications to compressor and electronics, follow up of development
 - ❖ Verification of all performances at system level
- Qualification of such cooler generally relies on several models
 - ❖ Engineering models, required for performance verification
 - ❖ Qualification models
 - ❖ Lifetime tests on some of these models
- Qualifications needed for some components, not initially space qualified
 - ❖ Temperature sensors, hermetic connectors, C-seals, vibration sensors

Typical cryogenic performances

Cooling power x CT temperature (warm end 0°C)



Typical microvibrations performances



MTG instruments



Artist's impression of MTG satellites

MTG satellites come in two brands : MTG-I and MTG-S

MTG-I : 4 satellites

IR imager called FCI (Flexible Combined Imager), instrument prime TAS-F

MTG-S : 2 satellites

IR sounder called IRS (InfraRed Sounder), instrument prime OHB Munich

One single Cryocooler requirement for both instruments

MTG instruments

FCI characteristics

Mass : 470 kg

Power : 530 W

Full earth imagery every 10 minutes

16 spectral channels

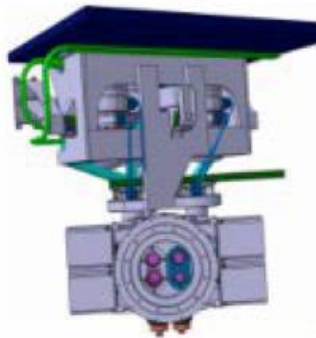
Resolution 1-2 km

Detailed imagery every 2.5 minutes

4 spectral channels

Resolution 0.5-1 km

Detector at 60K



MTG instruments

IRS characteristics

Mass : 470 kg

Power : 750 W

Michelson interferometer

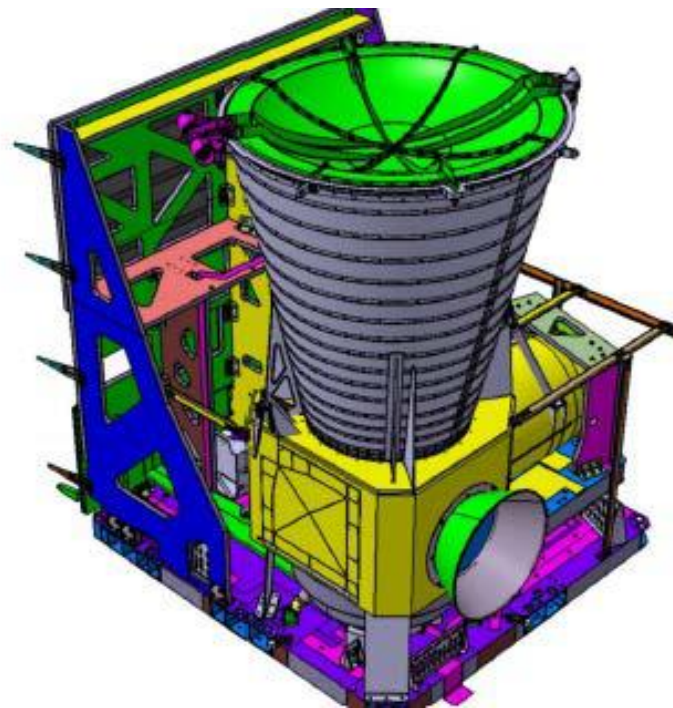
Full earth imagery every 10 minutes

2 spectral channels (LWIR & MWIR)

Resolution 4 km

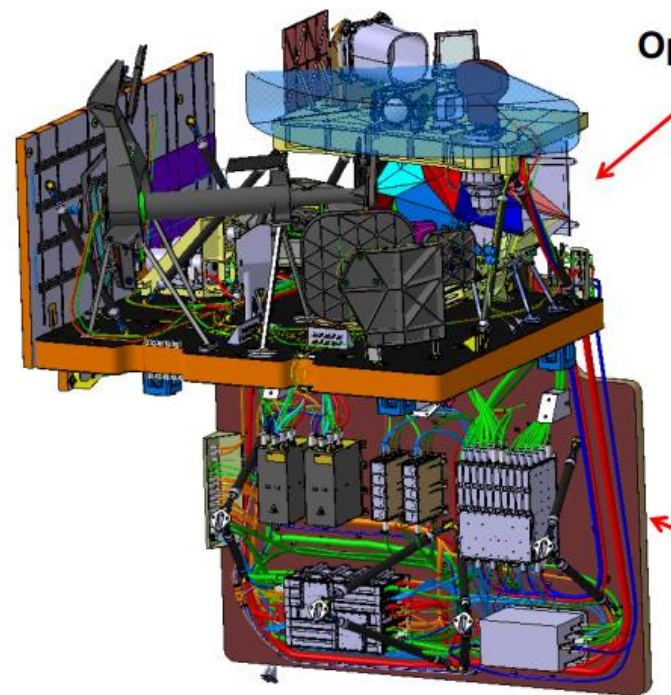
Provides atmospheric motion vectors

Detector at 55K



IASI-NG instrument (Source : WMO Oscar, CNES)

Acronym	IASI-NG			
Full name	Infrared Atmospheric Sounder Interferometer - New Generation			
Purpose	Temperature/humidity sounding, ozone profile and total-column or profiles of green-house gases (C2H2, C2H4, C2H6, CFC-11, CFC-12, CH3OH, CH4, CO, H2CO2, HCN, HNO2, HNO3, N2O, NH3, PAN, SO2)			
Short description	16,921 channels, range 645-2760 cm ⁻¹ (3.62-15.50 μm) split in 12 bands [see detailed characteristics below]. Spectral resolution 0.125 cm ⁻¹ (unapodised)			
Background	Evolution of IASI on MetOp A, MetOp-B, MetOp-C			
Scanning Technique	Step-and dwell cross-track: 28 Earth's FOV's plus two views, one of cold space and one of a blackbody. Swath 2000 km scanned in 15.6 s			
Resolution	Each FOV contains 4×4 IFOV's of 12 km size at s.s.p.			
Coverage / Cycle	Near-global coverage twice/day			
Mass	360 kg	Power	500 W	Data Rate 6 Mbps



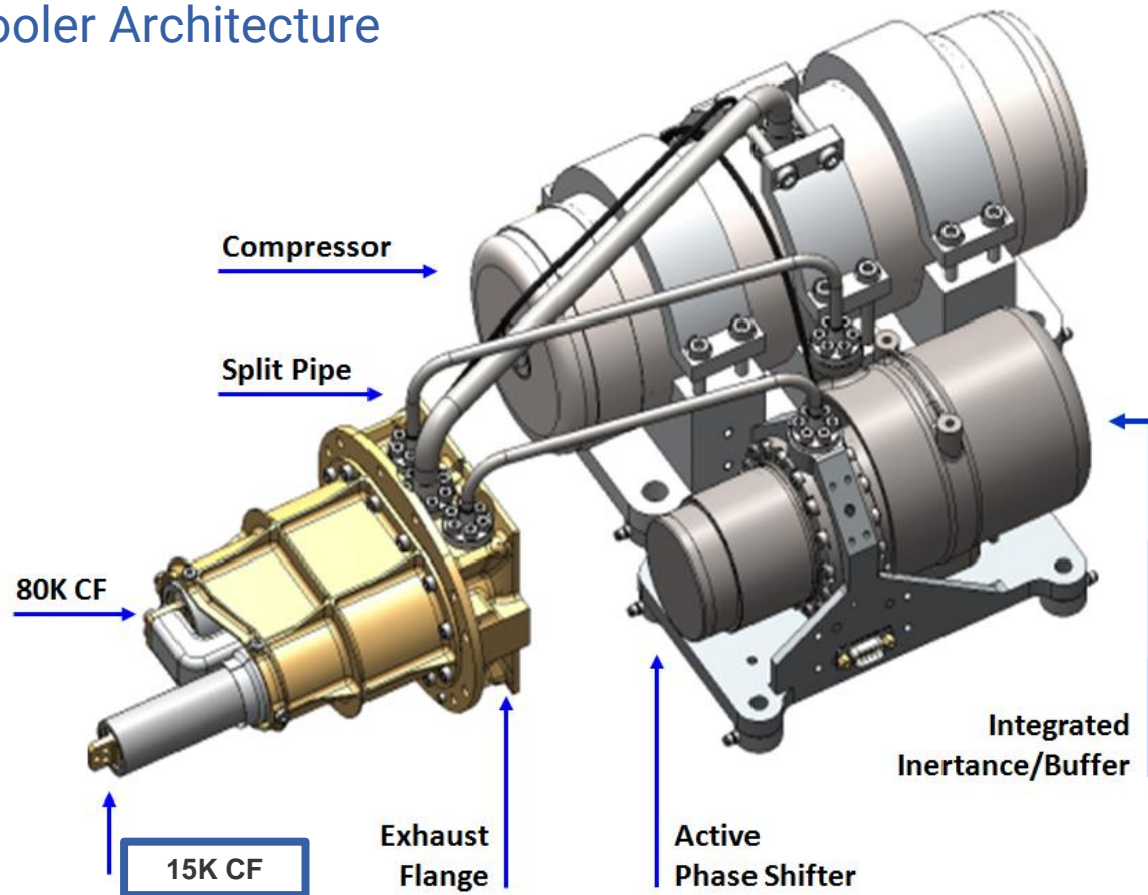
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Space Coolers for Science projects

The ATHENA mission

- Advanced Telescope for High Energy Astrophysics
- X-IFU instrument presentation
 - ❖ <https://www.youtube.com/watch?v=mOf6WIDmi30>
- **Detectors : micro calorimeters, need to be cooled down at 0.05 K**
 - ❖ Complex cooling chain : Pulse Tube coolers, Joule Thomson coolers, Adiabatic Disimination cooler
 - ❖ 15K stage, 3-4K stage, 50mK stage
 - ❖ Combination to coolers allowing redundancy

Hi-PTC Cryocooler Architecture

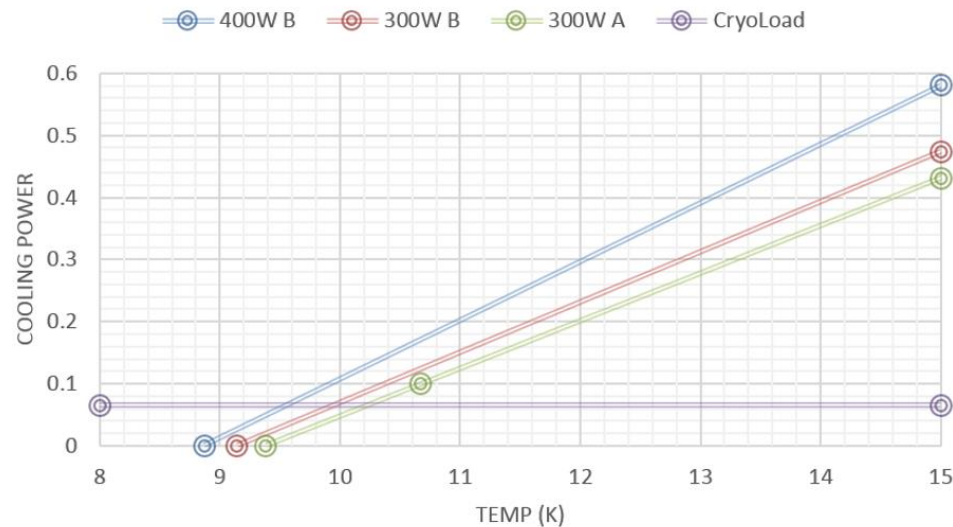


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Hi-PTC Cooling Curve



THALES Cryogenics

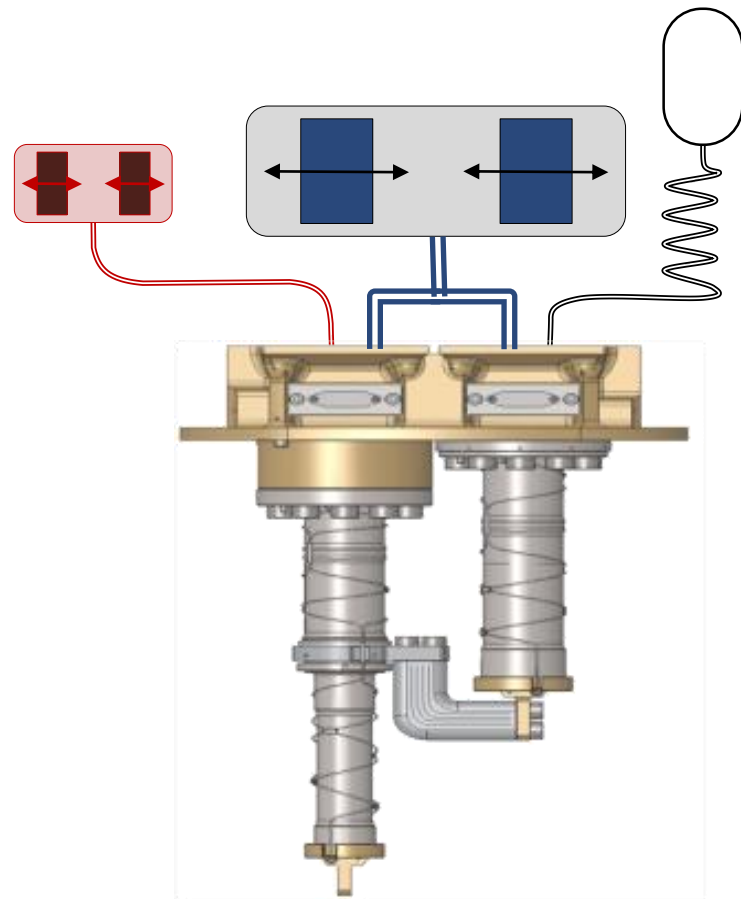
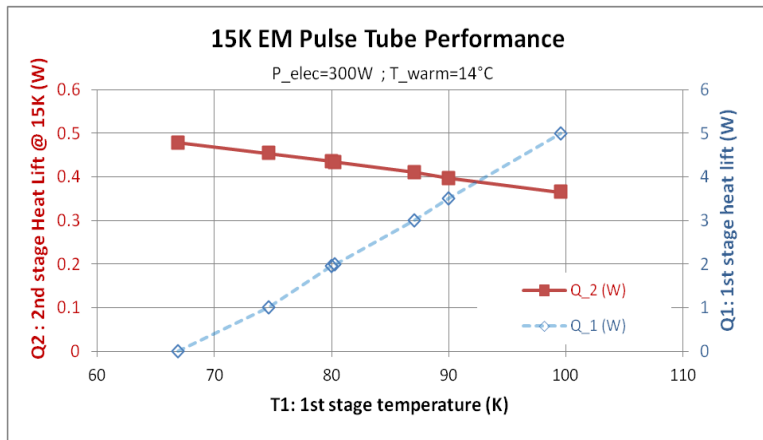


Cooling Power @ 10K: 80mW
Cooling Power @ 80K: 1,9W

HiPTC

- **EM for ATHENA preparation**

- Performance target is met
- Verification of environment for mechanical cooler
- Development electronics



3

Engineering Challenges

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AIR LIQUIDE, THE WORLD LEADER IN GASES, TECHNOLOGIES AND SERVICES FOR INDUSTRY AND HEALTH

21

3rd October 2019

WIERTZ Thierry

Air Liquide Space Department

EASISchool - Grenoble

Cryocooler technology

- Technology and design in space cryogenics is always a question of compromise:

- ❖ Mechanical strength / stiffness vs Thermal insulation / performance
- ❖ Efficiency vs reliability, lifetime

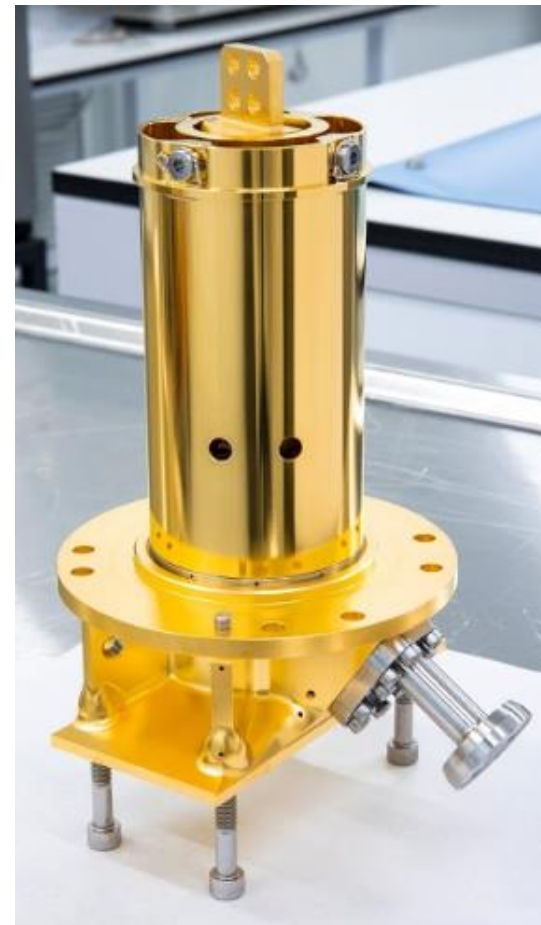
- Pulse Tube technology is chosen:

- ❖ Because it provides a good compromise between efficiency and reliability compared for example to Stirling with displacer
- ❖ Good micro vibration performance
- ❖ It is a suitable technology for 10K – 200K cooling
- ❖ It is adapted to the required cold power

Cold finger challenges

●Efficiency

- ❖ Most configurations are cold redundancy = two coolers connected to the same load, one active, one « spare » in case of failure of the first.
- ❖ This induces a thermal loss through the second cold finger = parasitic heat loss
- ❖ Design of the cold finger needs to be optimized for reducing the thermal loss
- ❖ However, launch loads need to be sustained.



Compressor challenges

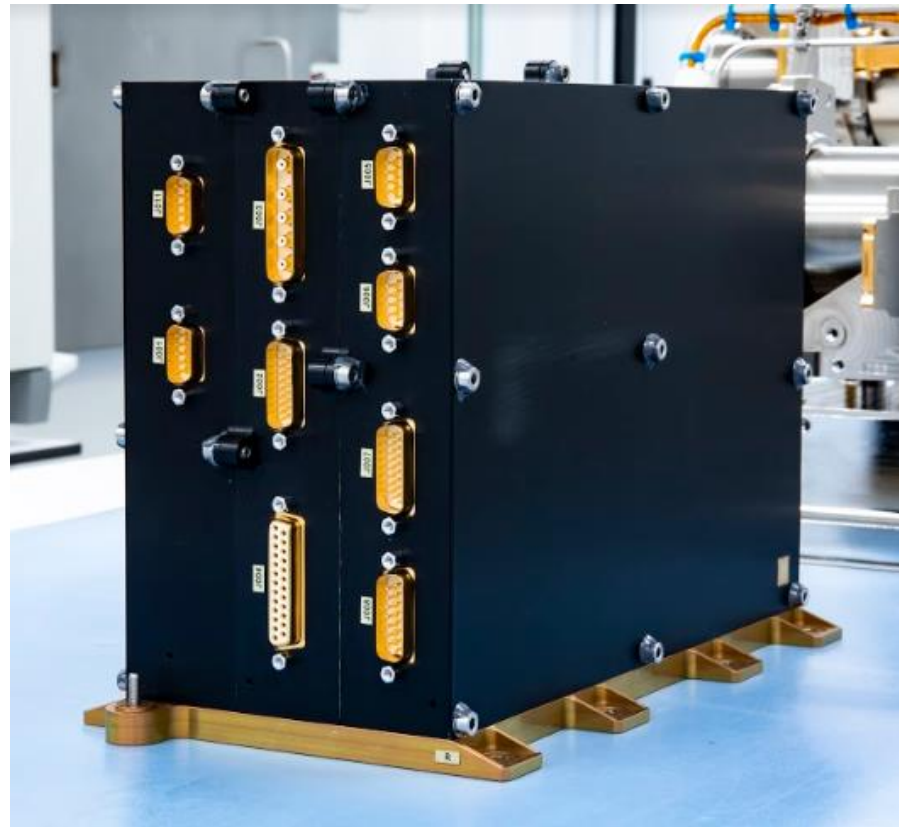


- Efficiency, EMC, micro vibrations

- ❖ Efficiency of the compressor is key to global cooler efficiency, also, compressor needs to be correctly matched to the cold finger
- ❖ Compressor technology is linear electro magnetic motors, motor architecture needs to be designed to avoid excessive radiated magnetic fields
- ❖ Opposed motors and pistons need to be perfectly balanced in order to reduce exported micro vibrations.

Electronics challenges

- Multi disciplinary product
 - ❖ Power conversion for reacting load
 - ❖ Power generation
 - ❖ Sensitive analog circuits (temperature measurement, force measurement)
 - ❖ Control logic for temperature regulation and exported vibration reduction



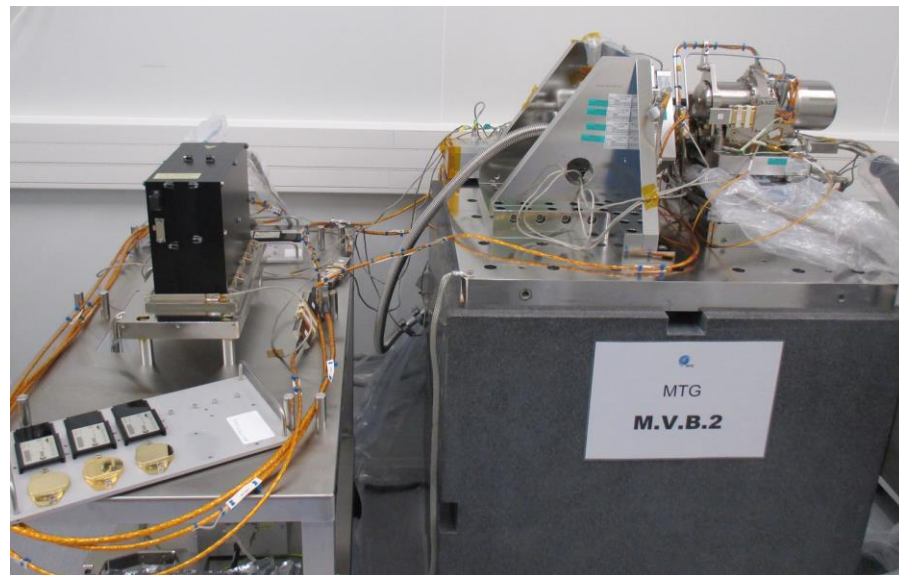
Manufacturing and Test

- 110m² of ISO 5 Clean Room for Manufacturing and Test



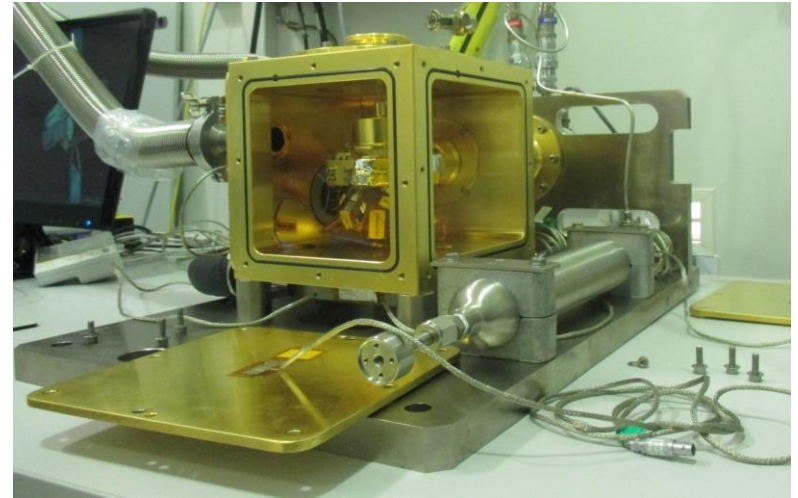
Manufacturing and Test

- Capability to run 4 CryoCooler test campaigns in parallel
- Test bench development for exported μ Vibration measurement
 - ❖ Force Accuracy lower than 10mN up to 200Hz (and 25mN up to 400Hz)
 - ❖ Resolution (incl. noise) lower than 2mN
 - ❖ Torques Accuracy and resolution lower than resp. 1mN.m and 0.1mN.m



Manufacturing and Test

- Thermal Vacuum test bench for Cryogenic performances and thermal cycling of complete systems, Cold finger parasitic losses test bench



5

Technology and Science

Conclusions and Remarks

- Space cryocoolers are complex objects
- Design and qualification requires lots of skills
- Engineering and technology is not sufficient to solve the challenges posed by the development of such high performance products
 - ❖ Efficiency optimization : CFD, thermodynamics, complex modeling
 - ❖ Micro vibrations : electrodynamics, CFD, acoustics, multi-physics modeling
 - ❖ Sensitive sensors
 - ❖ Advanced materials
- We need science people to tackle all those challenges