

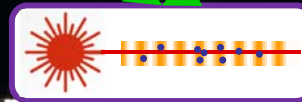
Mission I-SOC:

A world-wide frequency & time metrology network

Coordinator: S. Schiller (Univ. Düsseldorf)

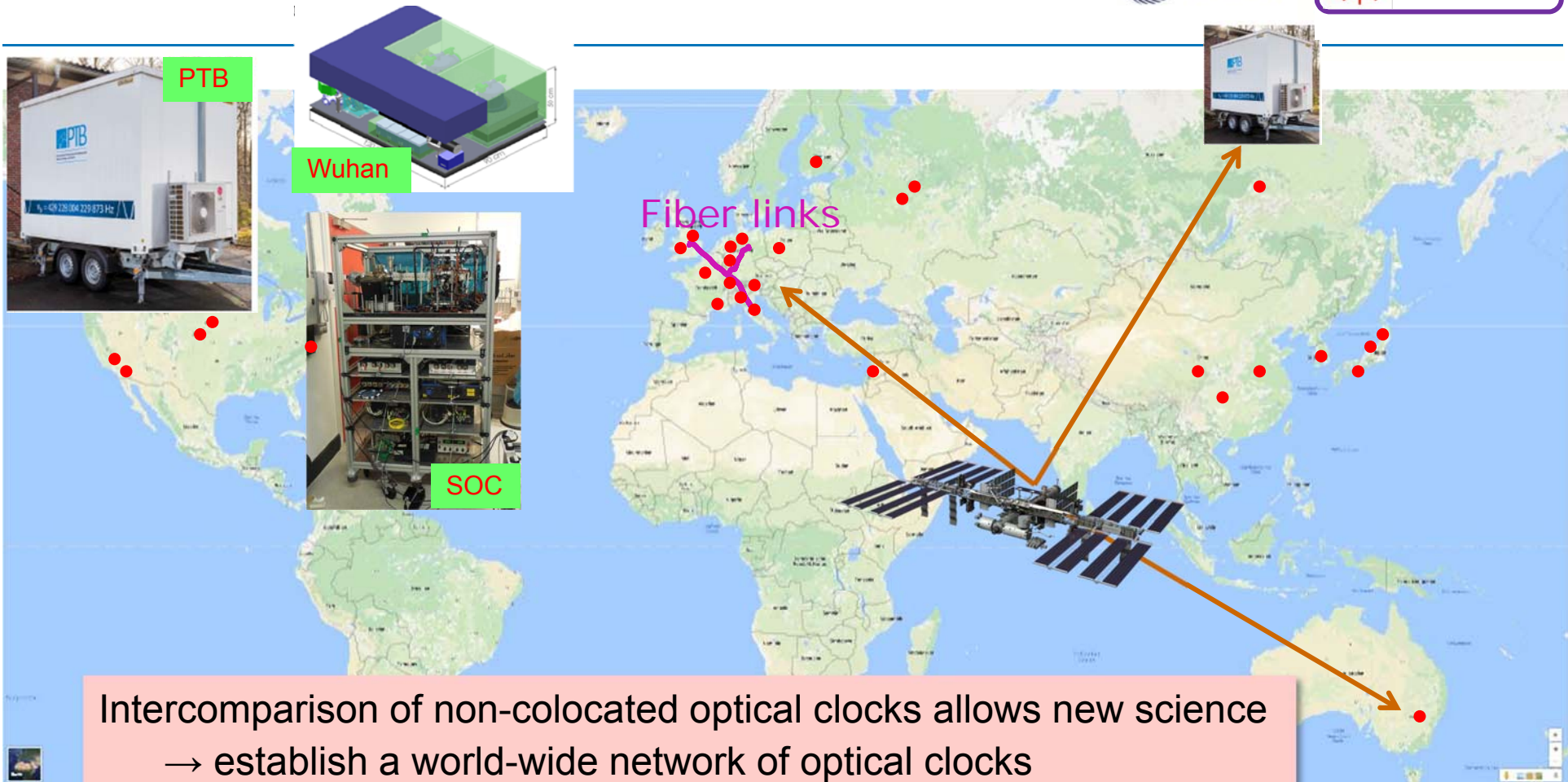
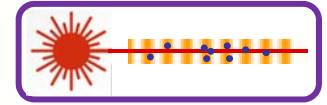


U. Sterr
Ch. Lisdat
R. Le Targat
J. Lodewyck
Y. Singh
K. Bongs
N. Poli
G.M. Tino
F. Levi
I. Prochazka
C. Salomon



- I-SOC: Mission goals and methods
- I-SOC: Elegant breadboard demonstrator
- I-SOC: Preliminary space instrument design
- Perspectives

Transportable optical clocks : world-wide

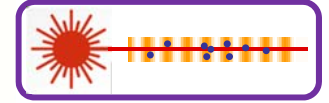


Ca: NRC
USA: NIST, JILA, USNO,
JPL, AOSense
Bra: U. Sao Carlos

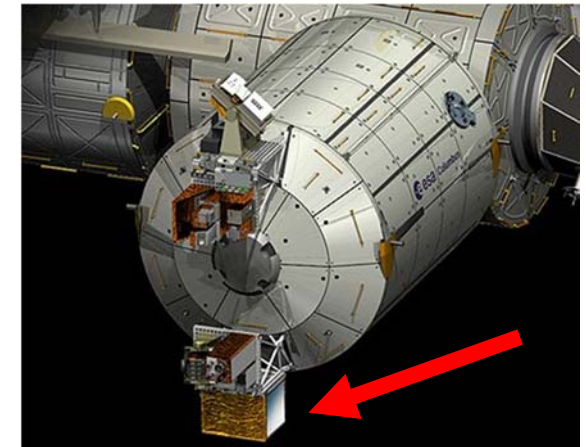
I: INRIM, LENS
F: Obs. Paris, U. Marseille
CH: METAS
D: PTB, U. Düsseldorf, U. Hannover
UK: U. Birmingham, NPL
PL: FAMO
FIN: MIKES
A: Innsbruck. NL: V.U. Amsterdam

I: WIS
Rus: VNIITRIL, FIAN, ILP
J: NIMJ, RIKEN, U. Tokyo
K: KRISS
Ch: Wuhan IPM, NTSC, NIM....
Aus: U. West. Aust.

Clocks and space missions

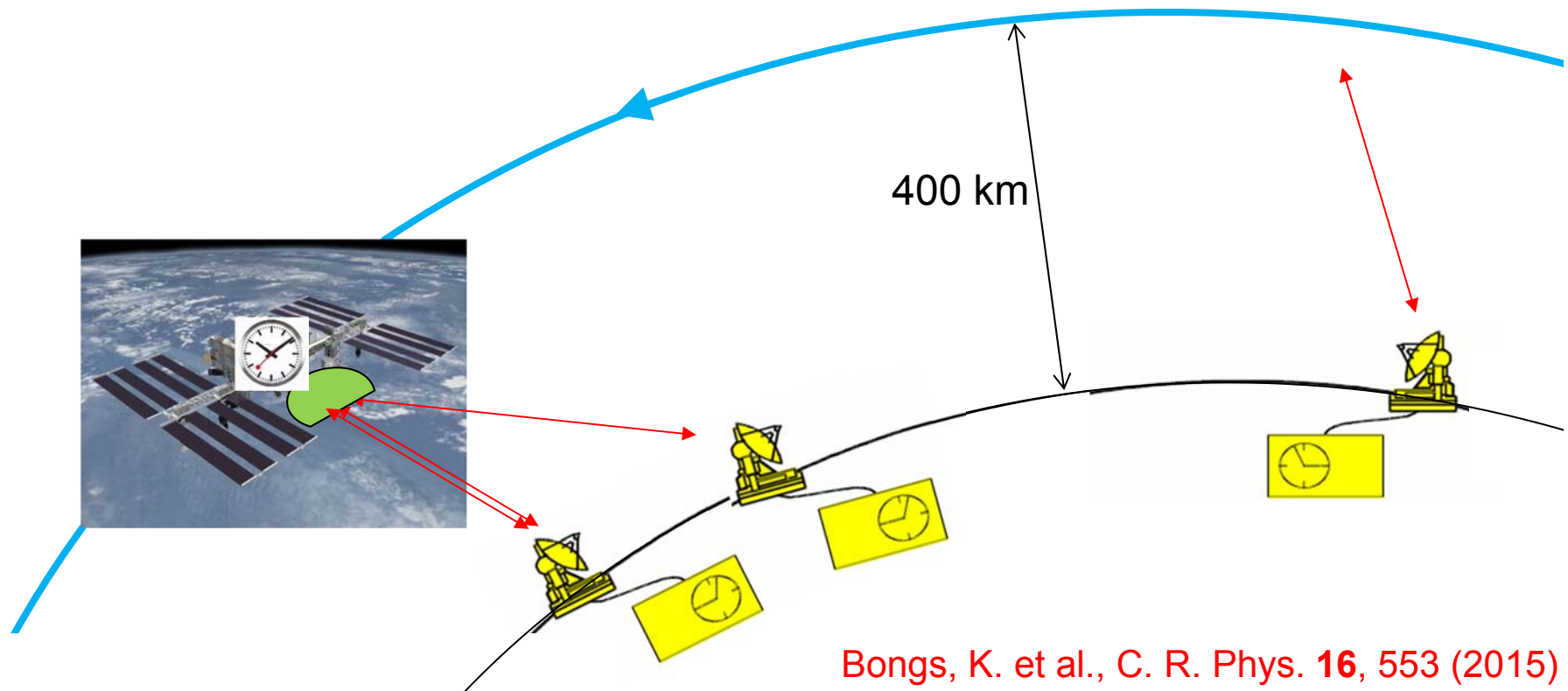


- In the upcoming ACES mission (~ 2020), optical ground clocks, combs and fiber links are a key mission component
- However, the impressive progress of ground clock performance calls for a post-ACES means of comparing them
- Improvements in ground and space technology (revolutionary & evolutionary) allow improvement by 10 – 100 in science output compared to ACES
- Can leverage on ACES heritage, BUT:
Need to develop the I-SOC within a reasonable time of ACES, in order to maintain its know-how and heritage (technology, operations, data analysis, ...)
- I-SOC can be the first and most realistic step towards future precision optical metrology space missions
- Proposed in 2004; part of the ISS utilization programme since 2006

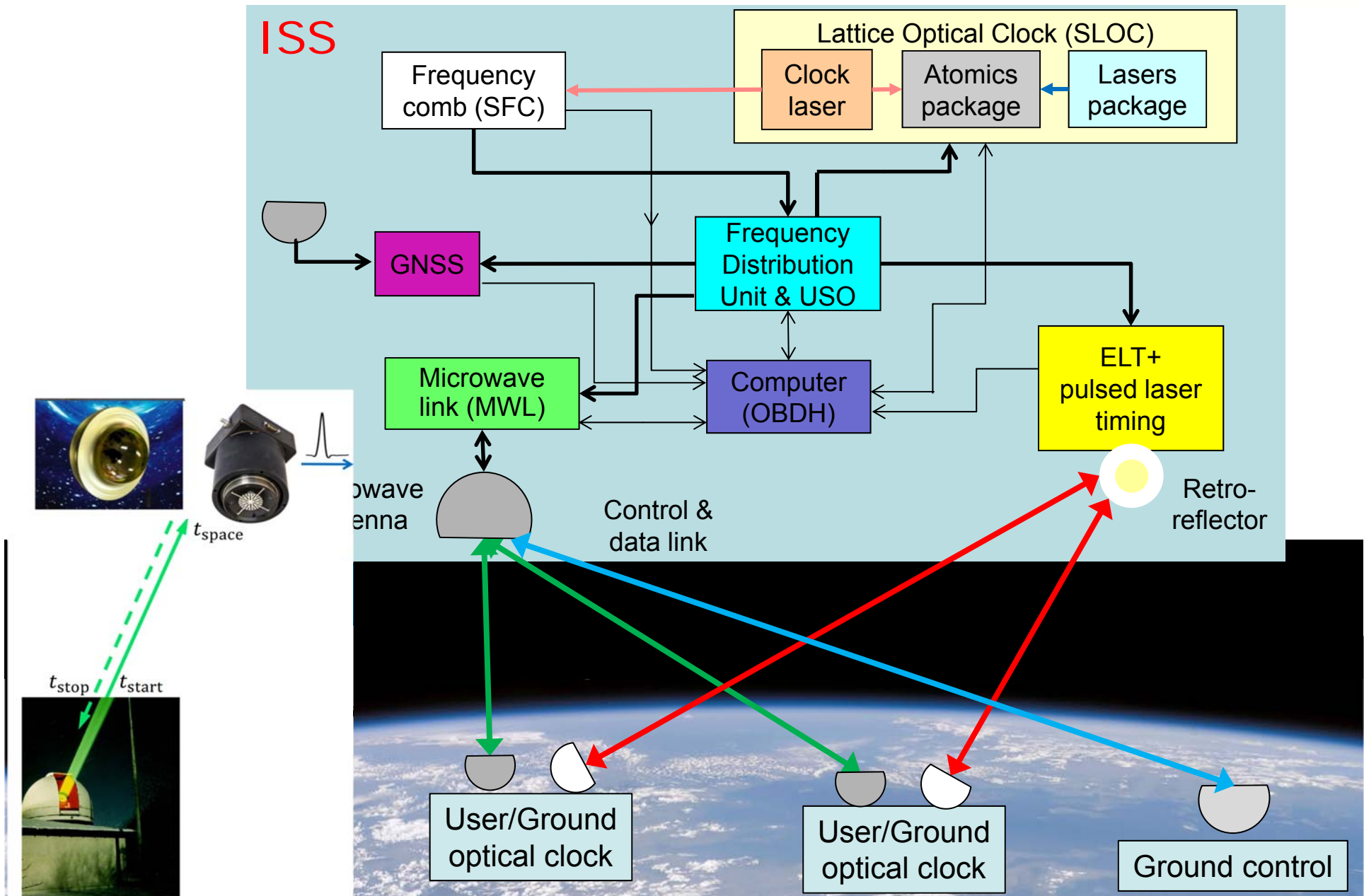


How does it work?

- An on-board optical clock allows to transfer time/frequency between locations that are not in common-view of the ISS



A diagram illustrating the wave nature of light. On the left, a red sun-like symbol emits a horizontal red line representing a ray of light. This ray passes through a series of vertical yellow bars representing slits. As the ray passes through the slits, it diffracts and interferes, creating a pattern of alternating light and dark regions (fringes) on the right side of the diagram.



- ACES estimated performance vs. I-SOC requirements

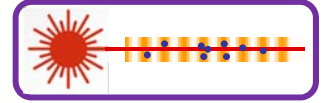
	ACES	I-SOC *	Improvem.
Clock instability	$1 \times 10^{-13} / \tau^{1/2}$	$8 \times 10^{-16} / \tau^{1/2}$ (up to 2×10^6 s)	x 100
Clock inaccuracy	1×10^{-16}	1×10^{-17}	x 10
MWL / MWL+ TDEV	$1.5 \text{ ps} \times (\tau / 10\,000 \text{ s})^{1/2}$	$0.03 \text{ ps}^{**} \quad \tau > 1000 \text{ s}$	x 150 @ 1 day
ELT / ELT+ TDEV	$8 \text{ ps} @ 10^6 \text{ s}$	$1 \text{ ps} @ 10^6 \text{ s}$	x 8
Phase coherence	yes	yes, minimum 12 h	

- I-SOC clock signal shall be phase-coherent → requirement to comb and SLOC
- ELT+ supports reaching 1×10^{-18} ground clock comparisons (or ground-to-space)
- I-SOC performance can be tested fully on the ground (*trapped* atoms)

* from I-SOC ESR document

** ground-to-space

Science goals I: Comparing clocks in an Earth-scale network



- Most accurate direct measurements of Einstein's time dilation & Special Relativity

measure Earth's gravitational time dilation at 2×10^{-7} uncertainty level

measure Sun's time dilation at 1×10^{-6} level

measure Moon's time dilation at 2×10^{-4} level

Kennedy-Thorndike test at 2×10^{-18} level

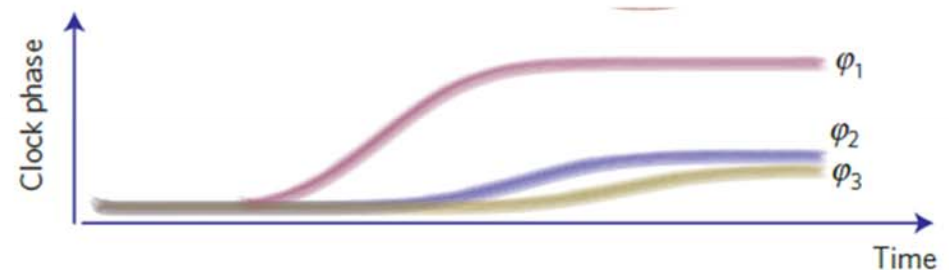
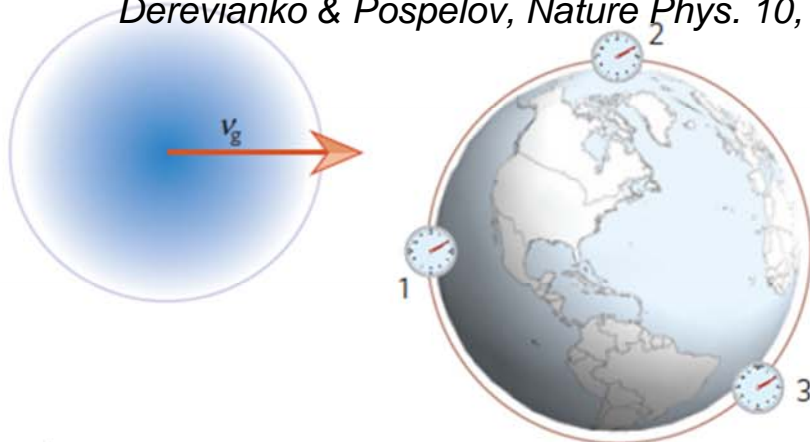
- Deliver time and frequency to users everywhere (universities, research centers, observatories)

world-wide atomic time distribution synchronized at few 10 picoseconds uncertainty

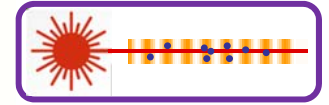
world-wide clock comparisons with 10^{-18} fractional frequency uncertainty

- Enable “new SI second”;
Support development of clocks and precision-optical measurements world-wide
- Enable fundamental physics studies: e.g. time-variability of fundamental constants
- Permit search for dark matter topological defects crossing the Earth

Derevianko & Pospelov, Nature Phys. 10, 933 (2014)



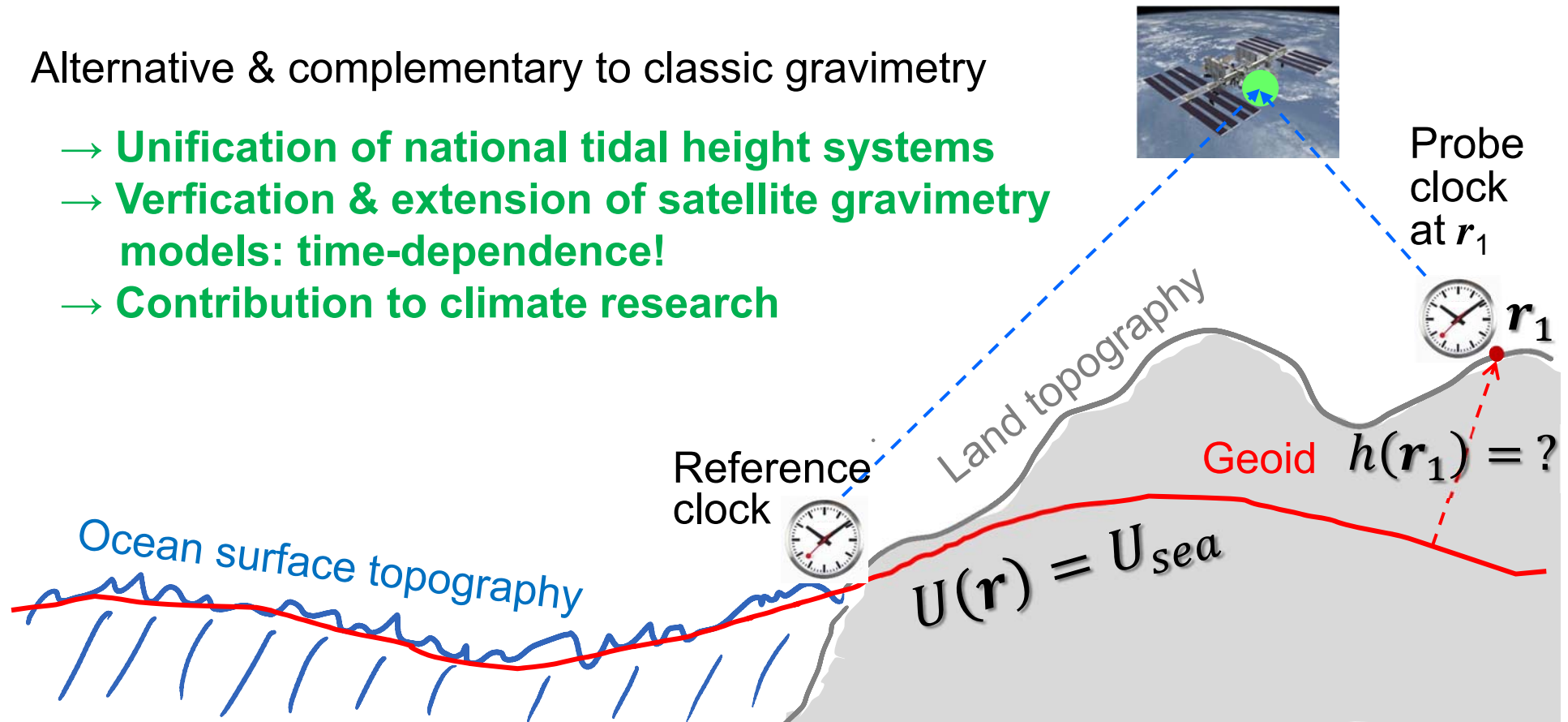
Science goals II: Geodesy via Relativity



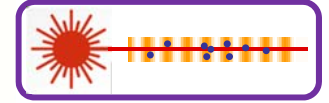
- Topography: determined using GPS
- Geoid surface: determined via transportable clock comparisons
- Near-real time & local determination
- 1 cm accuracy

Alternative & complementary to classic gravimetry

- **Unification of national tidal height systems**
- **Verification & extension of satellite gravimetry models: time-dependence!**
- **Contribution to climate research**

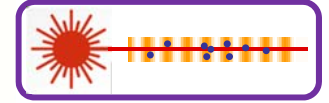


Contents



- I-SOC: mission goals and methods
- **I-SOC: elegant breadboard demonstrator**
- I-SOC: preliminary space instrument design
- Perspectives

From ACES to I-SOC



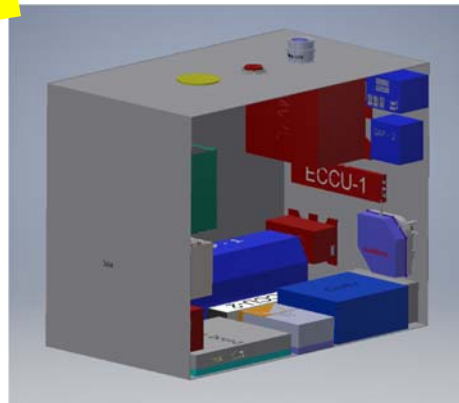
ACES
(being integrated)

Laser-cooled Cs clock ($f = 9.2$ GHz)

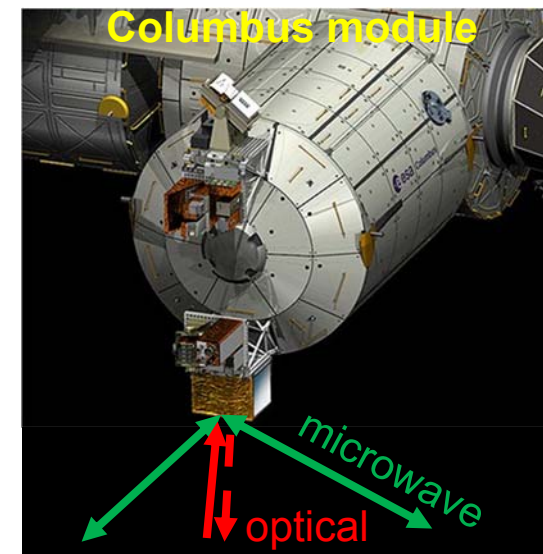
→ Strontium optical lattice clock ($f = 429$ THz)

Hydrogen maser

→ optical reference cavity + frequency comb

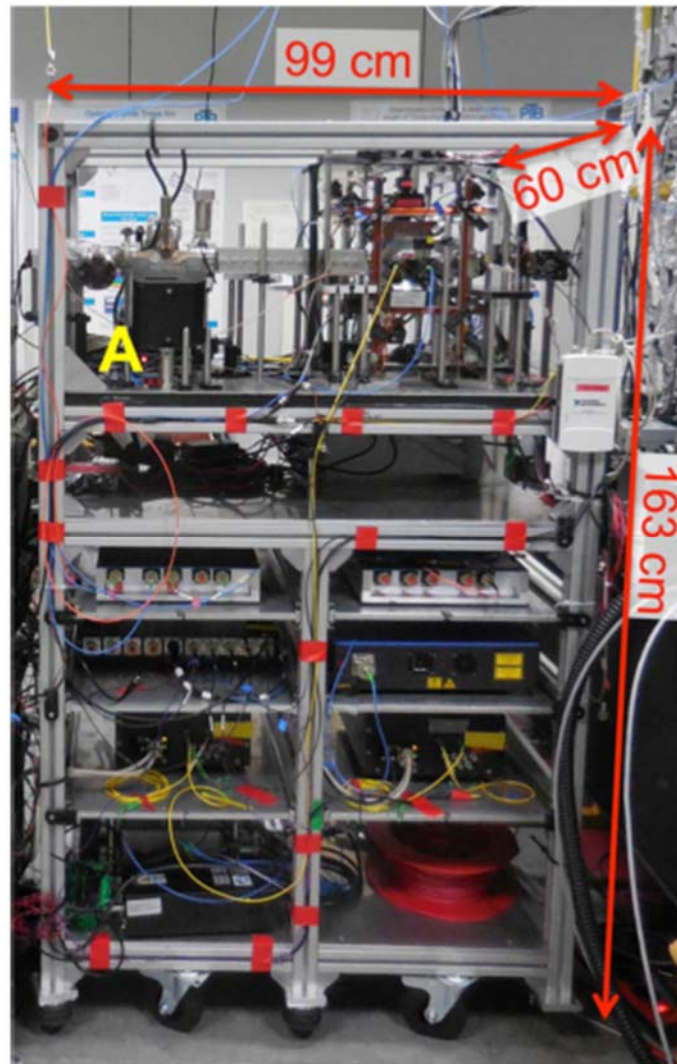


I-SOC
(prelim. design)



I-SOC elegant breadboard

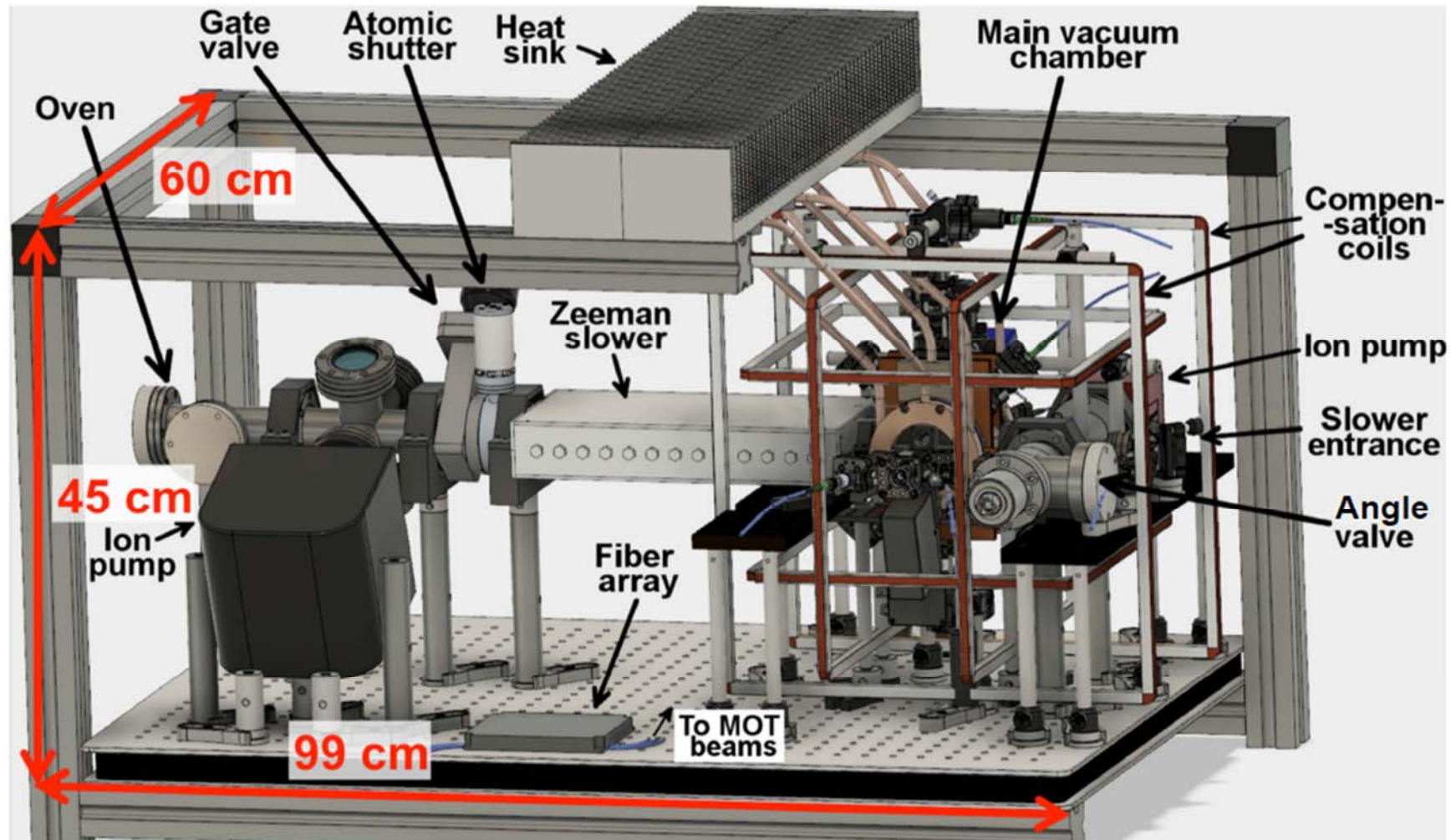
Goal: demonstrate a compact, robust system with goal performance, accessible technology, and modest mass, power, size



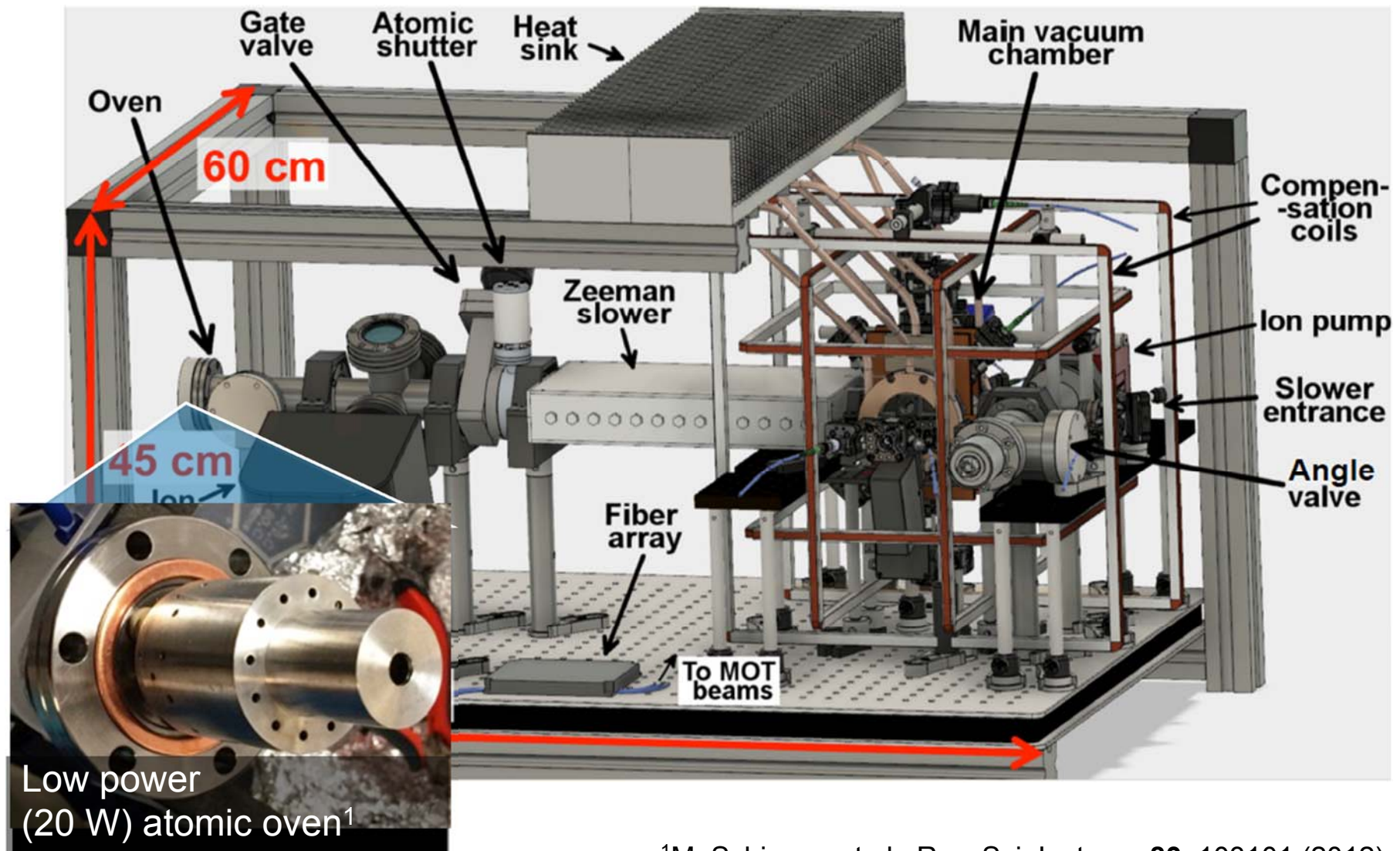
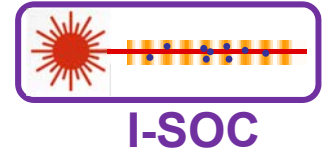
← Atomics package

← Laser package

Atoms Package

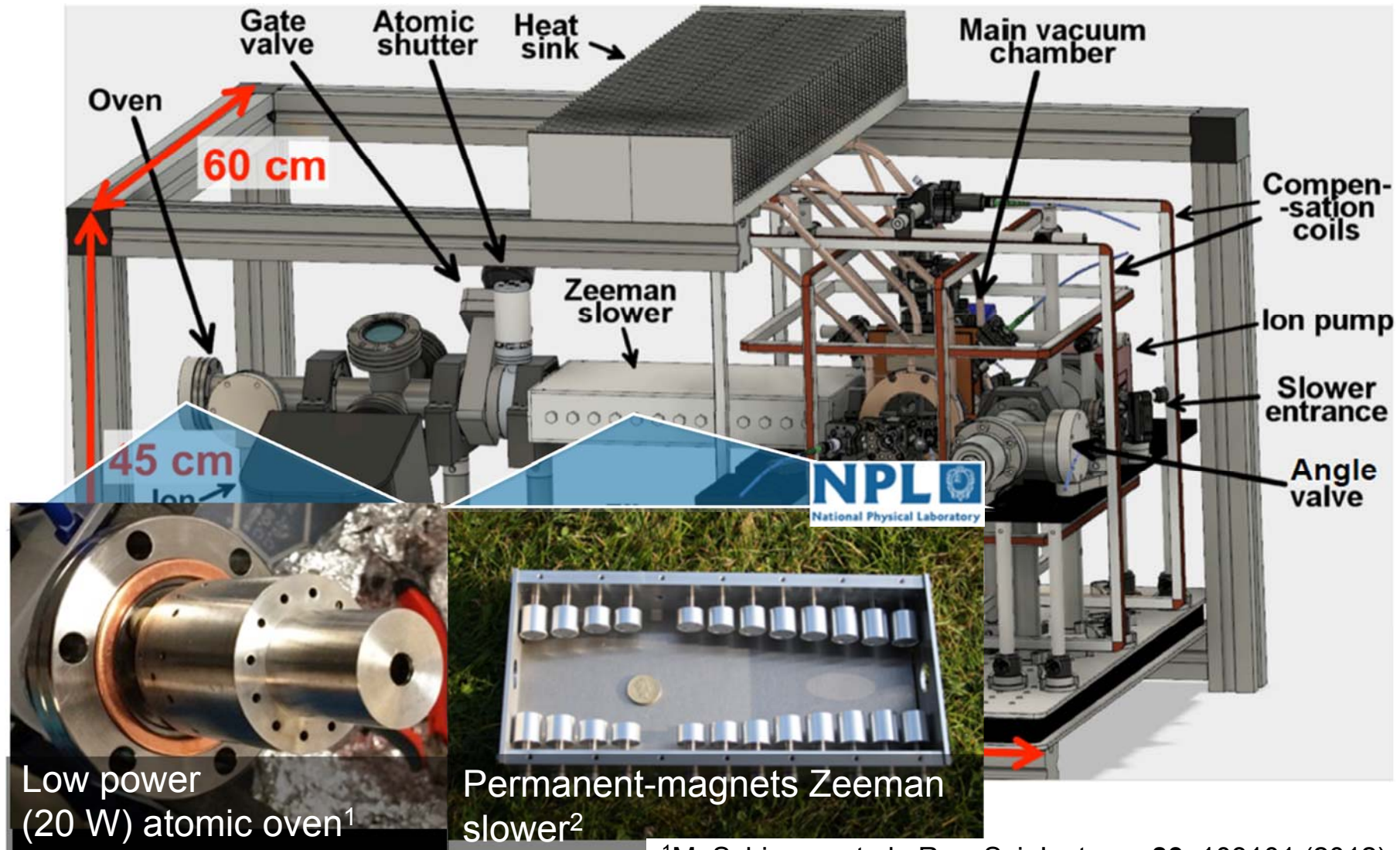
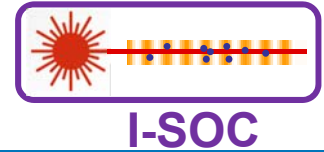


Atoms Package



¹M. Schioppo et al., Rev. Sci. Instrum. **83**, 103101 (2012)

Atoms Package



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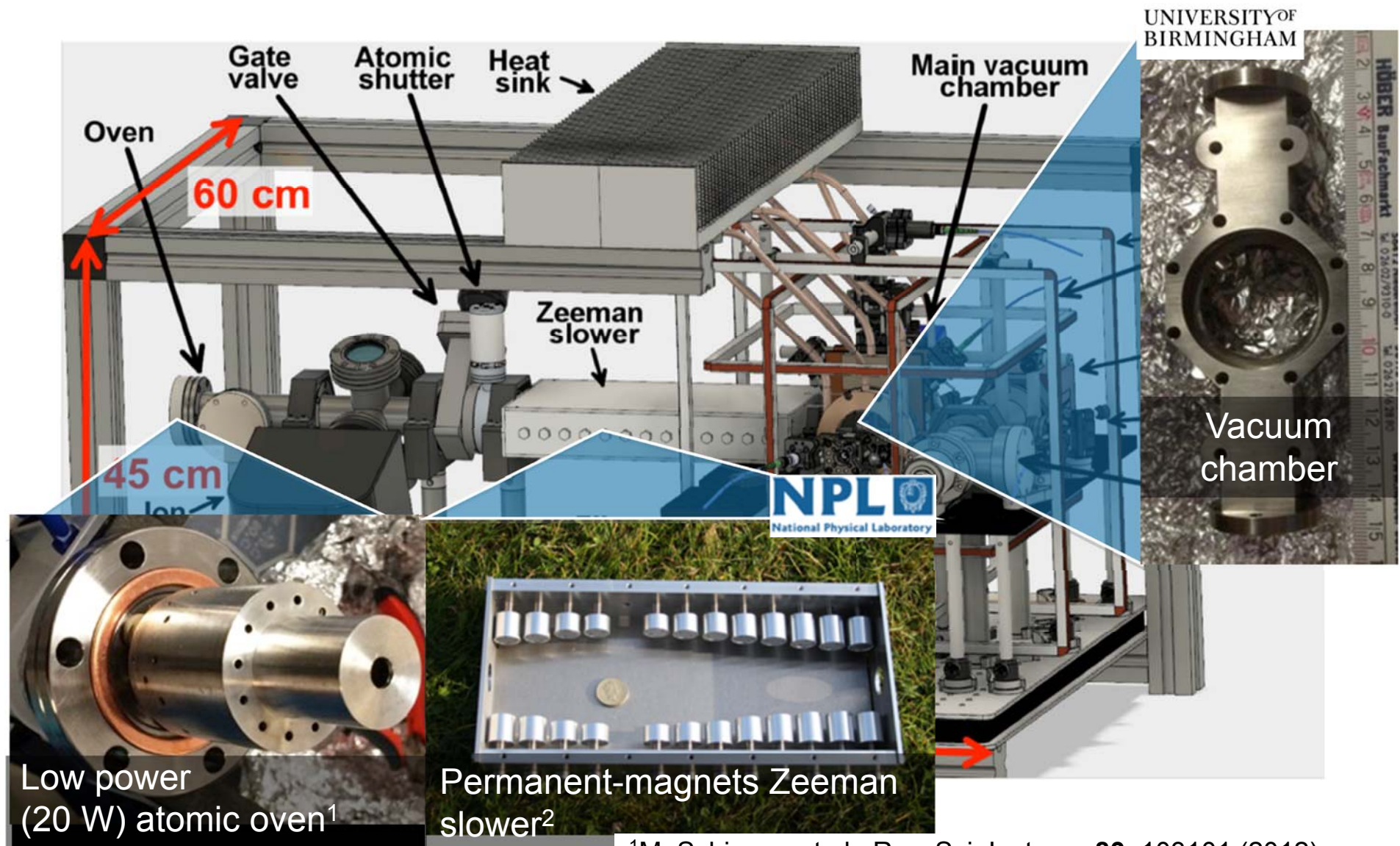
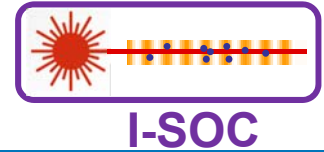


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¹M. Schioppo et al., Rev. Sci. Instrum. **83**, 103101 (2012)

²I. R. Hill et al., J. Phys. B **47**, 075006 (2014)

Atoms Package



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BIRMINGHAM

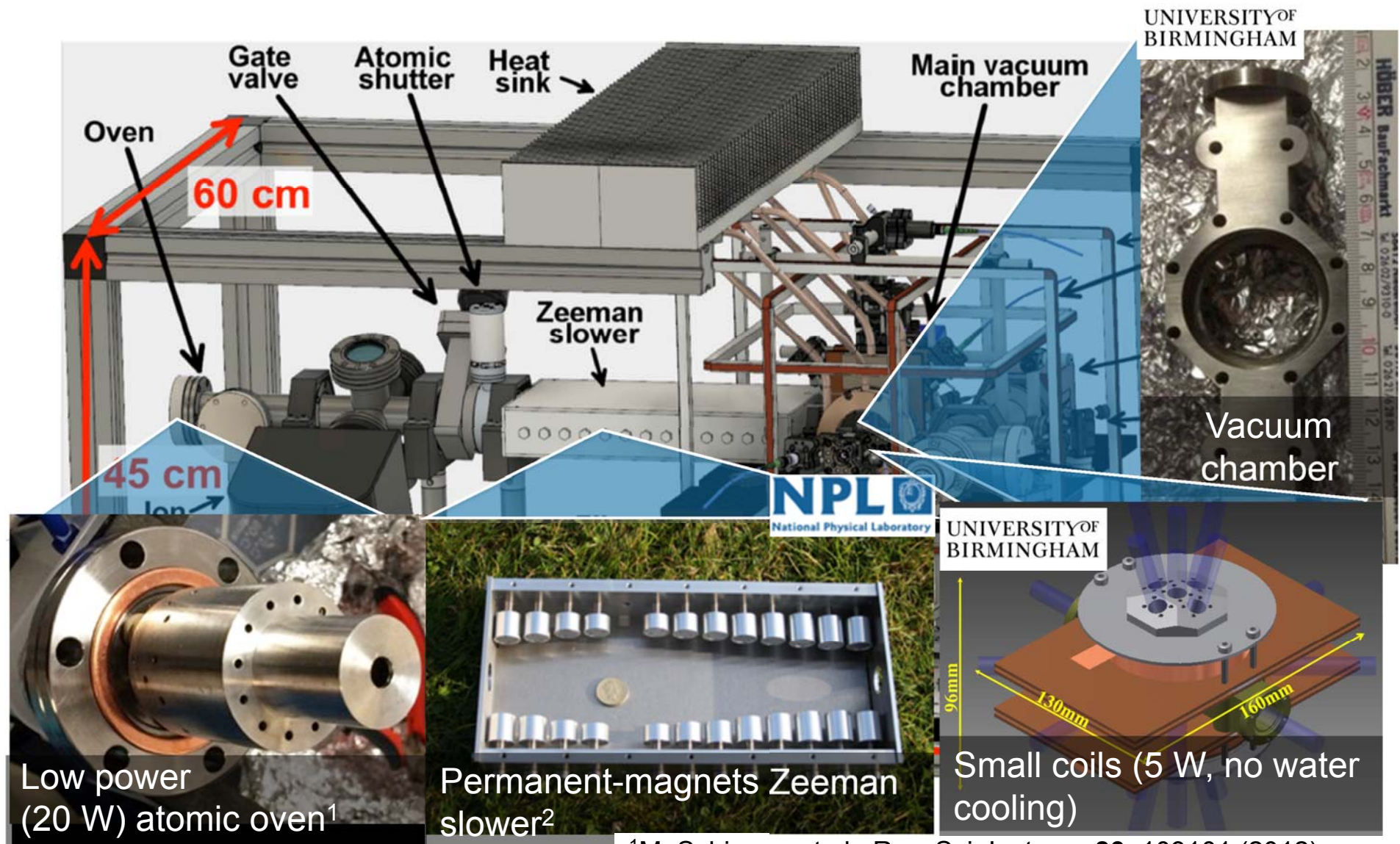
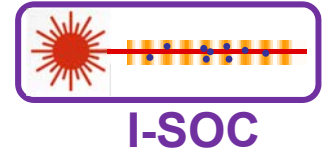


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¹M. Schioppo et al., Rev. Sci. Instrum. **83**, 103101 (2012)

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Atoms Package



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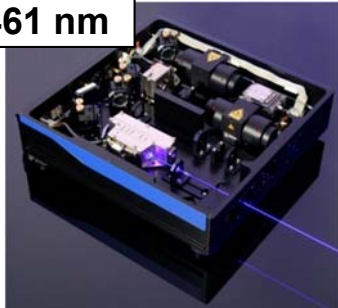
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FIRENZE

¹M. Schioppo et al., Rev. Sci. Instrum. **83**, 103101 (2012)

²I. R. Hill et al., J. Phys. B **47**, 075006 (2014)

I-SOC clock demonstrator: modular laser system

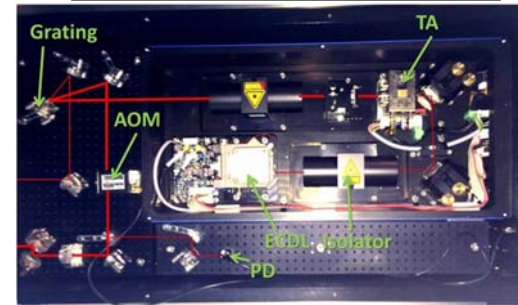
461 nm



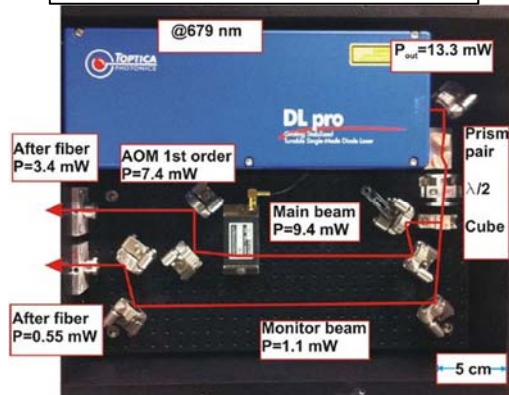
461 nm distribution



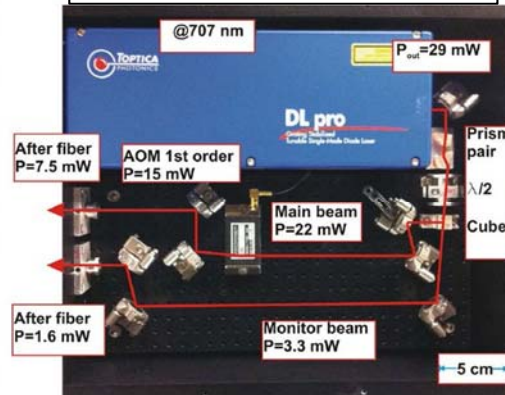
813 nm Lattice



Repumper 679 nm



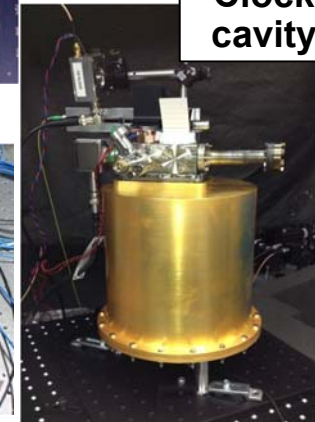
Repumper 707 nm



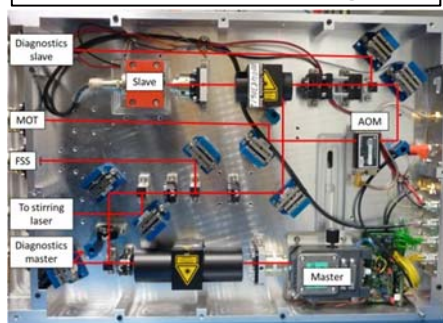
698 nm clock



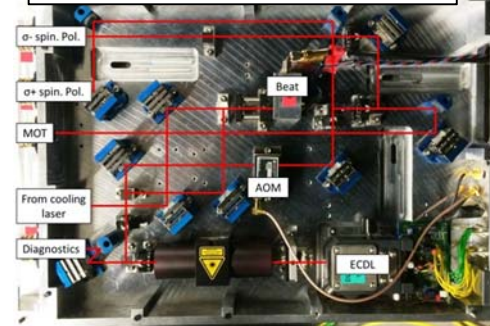
Clock cavity



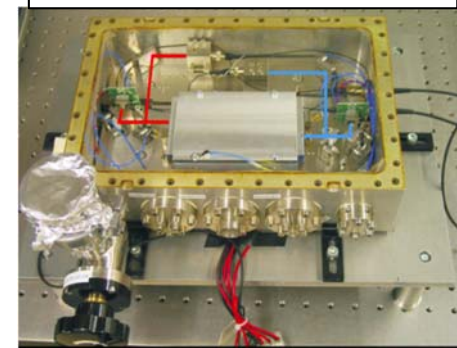
689 nm cooling



689 nm stirring



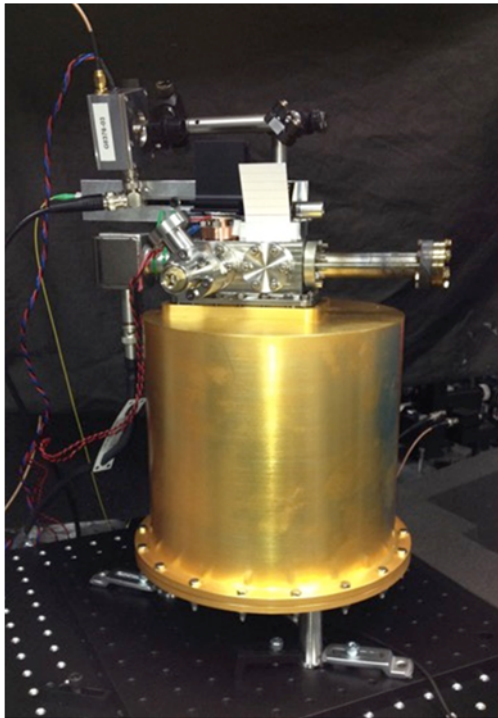
461, 689, 813 nm stabilization unit



Bongs, K. et al., C. R. Phys. **16**, 553 (2015)
 Nevsky, A. et al., Opt. Lett. **38**, 4903 (2013)
 Świerad, D. et al., Sci. Rep. **6**, 33973 (2016)



Robustness of breadboard components



Swierad et al. *Sci. Rep.* 6, 33973 (2016)

Ultra-high reflectivity mirrors
are **radiation** resitant:

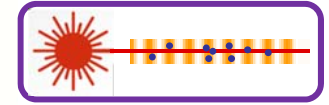
Chen et al., *Appl. Phys. B* 116, 385 (2014)



June 2015: **Birmingham, UK** →
Braunschweig, D

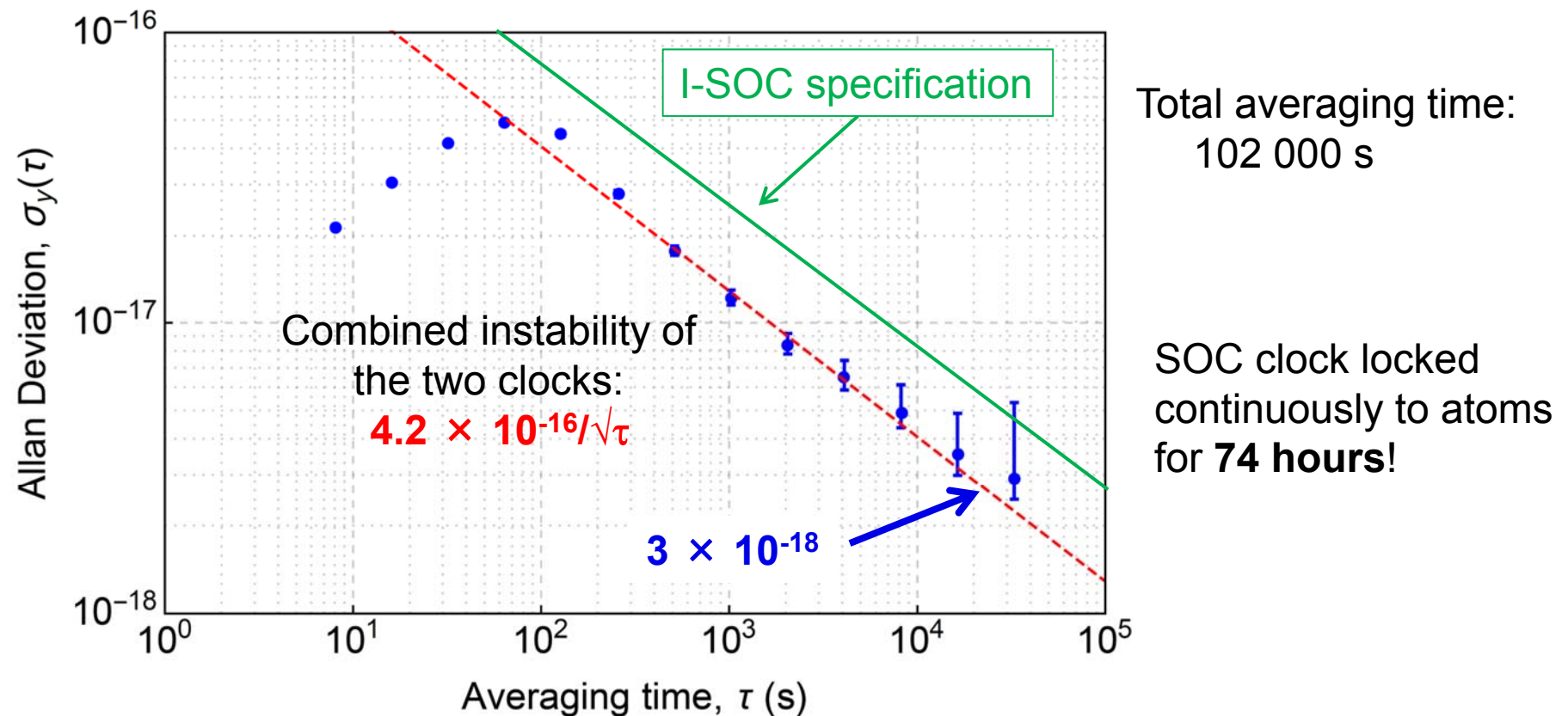
- 1st stage MOT obtained within 2 days of arrival
- Atoms trapped in lattice within 3 weeks of arrival

Breadboard clock instability



Origlia et al., *Phys. Rev. A*, 98, 053443 (2018)

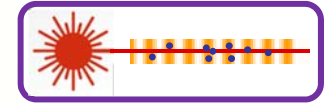
Instability determined by comparison with ^{87}Sr clock at PTB^{1,2}



¹S. Falke et al., *New J. Phys.* 16, 073023 (2014)

²A. Al-Masoudi et al., *Phys. Rev. A* 92, 063814 (2015)

Breadboard clock inaccuracy



Origlia et al., Phys. Rev. A, 98, 053443 (2018)

	I-SOC breadboard		⁸⁸ Sr clock		⁸⁷ Sr clock		PTB stationary clock
Effect	$\Delta\nu/\nu$	u/ν	$\Delta\nu/\nu$	u/ν	$\Delta\nu/\nu$	u/ν	
BBR shift	-523.4	0.8	-492.2	1.5			
BBR oven shift	0	0	-0.9	0.9			
Lattice shift ($\Delta\nu_L$)	-1.8	1.1	-0.9	0.4			
Probe light shift ($\Delta\nu_P$)	-96.1	1.3	0.0	0.0			
Cold collision shift ($\Delta\nu_{LP}$)	-0.6	0.3	0.0	0.2			
Second-order Zeeman shift ($\Delta\nu_B$)	-209.7	0.5	-3.4	0.1			
Tunneling	0	0	0.0	0.3			
Background gas collision shift	-0.13	0.13	-0.8	0.8			
dc-stark shift	-0.2	0.2	-0.2	0.1			
Gravitational shift	+5.1	0.1	0.0				
Total shift	-826.9	2.0	-498.4	2.0			

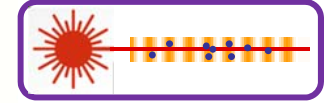
All numbers: $\times 10^{-17}$

The I-SOC breadboard demonstrator is as accurate
as the most accurate European lattice clocks

Lattice laser:

Both Ti:Sapphire and diode laser give similar performance

Comparison SOC(⁸⁸Sr) and PTB(⁸⁷Sr)



Origlia et al., Phys. Rev. A 98, 053443 (2018)

I-SOC breadboard

PTB stationary clock

$$f(^{88}\text{Sr}) - f(^{87}\text{Sr}) = 62\,188\,134.027(12) \text{ Hz}$$

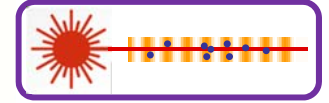
$$f(^{88}\text{Sr})/f(^{87}\text{Sr}) = 1.000\,000\,144\,883\,682\,831\,(28)$$

At the time (2018), probably the most accurate measurement of a property of nature

In agreement with an independent measurement in Japan

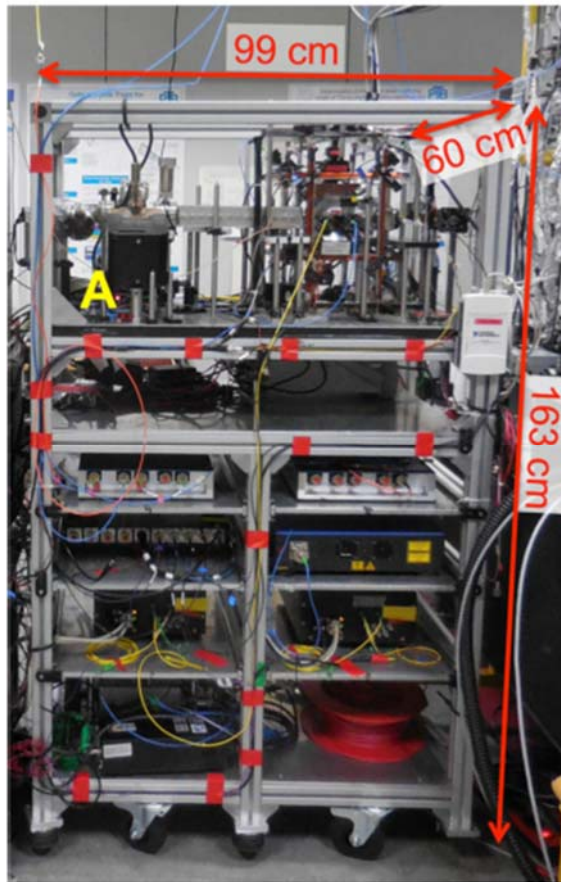
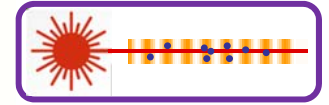
Takano et al., Applied Physics Express 10, 072801 (2017)

Contents

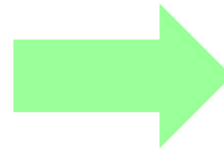


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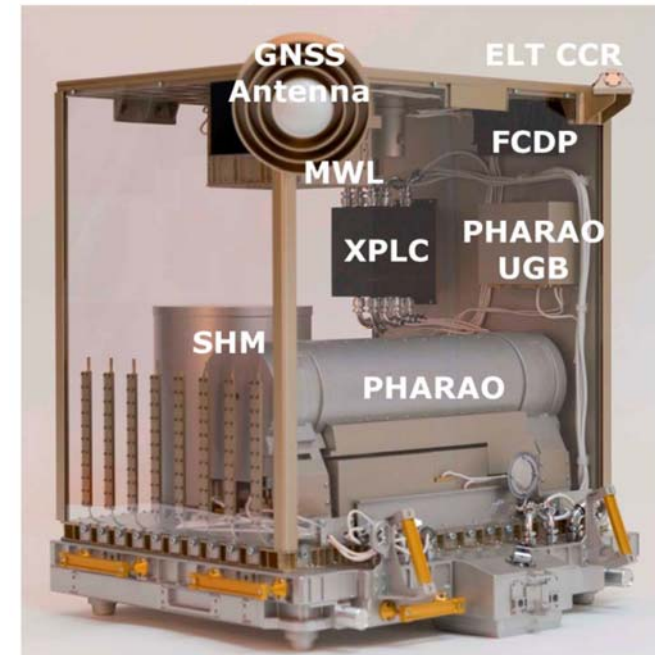
From elegant breadboard to space instrument



?



ACES



Accommodation requirements of complete space instrument:

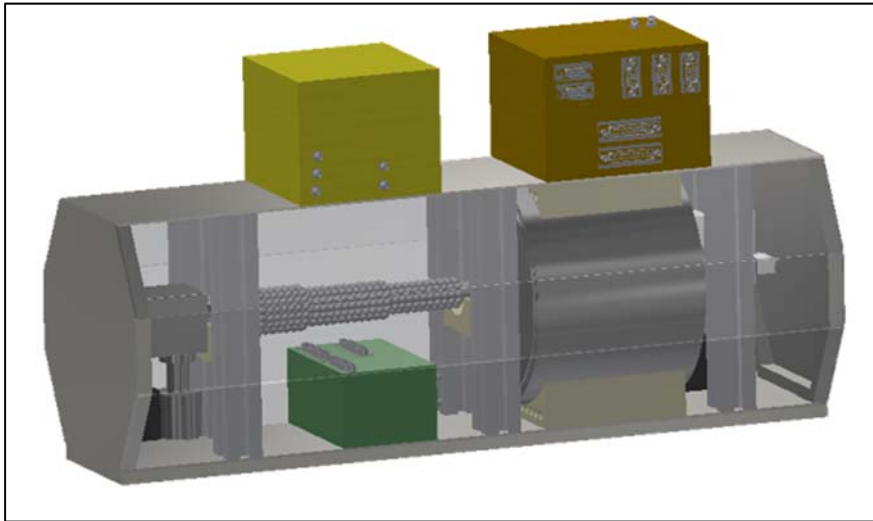
< 100 kg

< 250 W

< 480 liter (< 1 m x 0.6 m x 0.8 m)

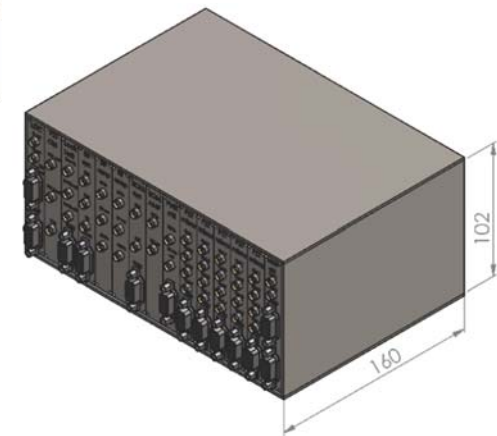
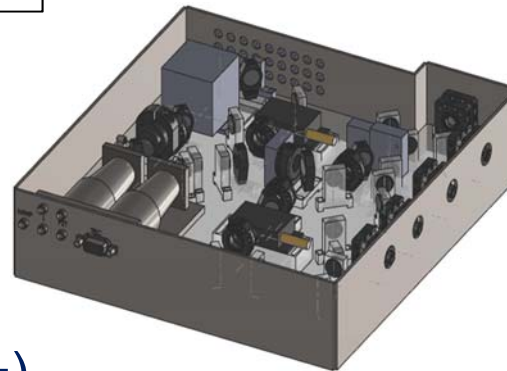
(Complete payload: 1.17 m³, 400 W, 226 kg)

Preliminary design of sub-systems: examples

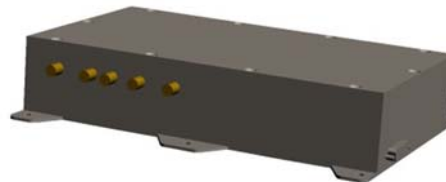
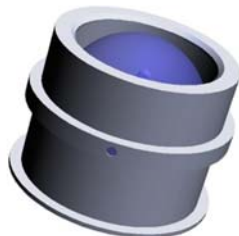


Atomics package

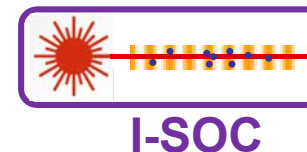
2nd-stage cooling laser (689 nm)



Single-photon time transfer (ELT+)



Physical parameters (preliminary)



Subsystem	Mass (kg)	Volume (liter)	Elec. Power (W)
Atomics package (SAP) with optics	31.3	28	29.8
Blue cooling laser (BCL) + distribution module (DBL)	7.2 + 2.2	6.2 + 2.6	10 + 1.4 avg. 2.2 pk
Red cooling and stirring laser (RCL, 689 nm)	7.9	5.7	10.3
Repumpers (RLS, 707 and 679 nm)	4.0	2.9	4.8
Lattice laser (LLS, 813 nm)	6.1	4.4	6.2
Frequency stabilization system (ECCU)	6.3	4.7	9.3
Subtotal atomics unit (with electronics)	65	57.1	71.8
CLS-1: Clock laser cavity (OSRC specifications)	10.5	10	15
CLS-2: Clock laser vibration isolation unit	n/a	n/a	n/a
CLS-3: Clock laser breadboard + SHG	10.3	10	14.3
Clock monitoring unit (CMU)	5	7.3	15 avg. 25 pk.
Total clock (SLOC) (with electronics)	90.8	84.4	116.1
Frequency comb (SFC)	10 (est.)	10 (est.)	35 (est.)
Frequency distribution package (FDP)	2	2	17.6
Microwave link (MWL)	10.4	14.4	31
Laser timing (ELT+)	3.1	4.5	15
GNSS receiver (GNSS)	4.5 (est.)	2 (est.)	15 (est.)
Total I-SOC instrument	120.8	117.3	229.7 avg.

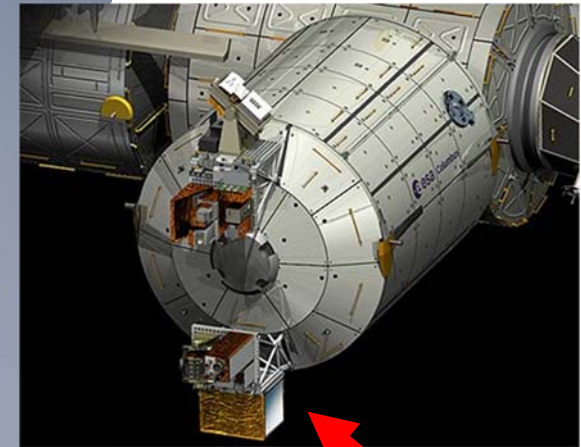
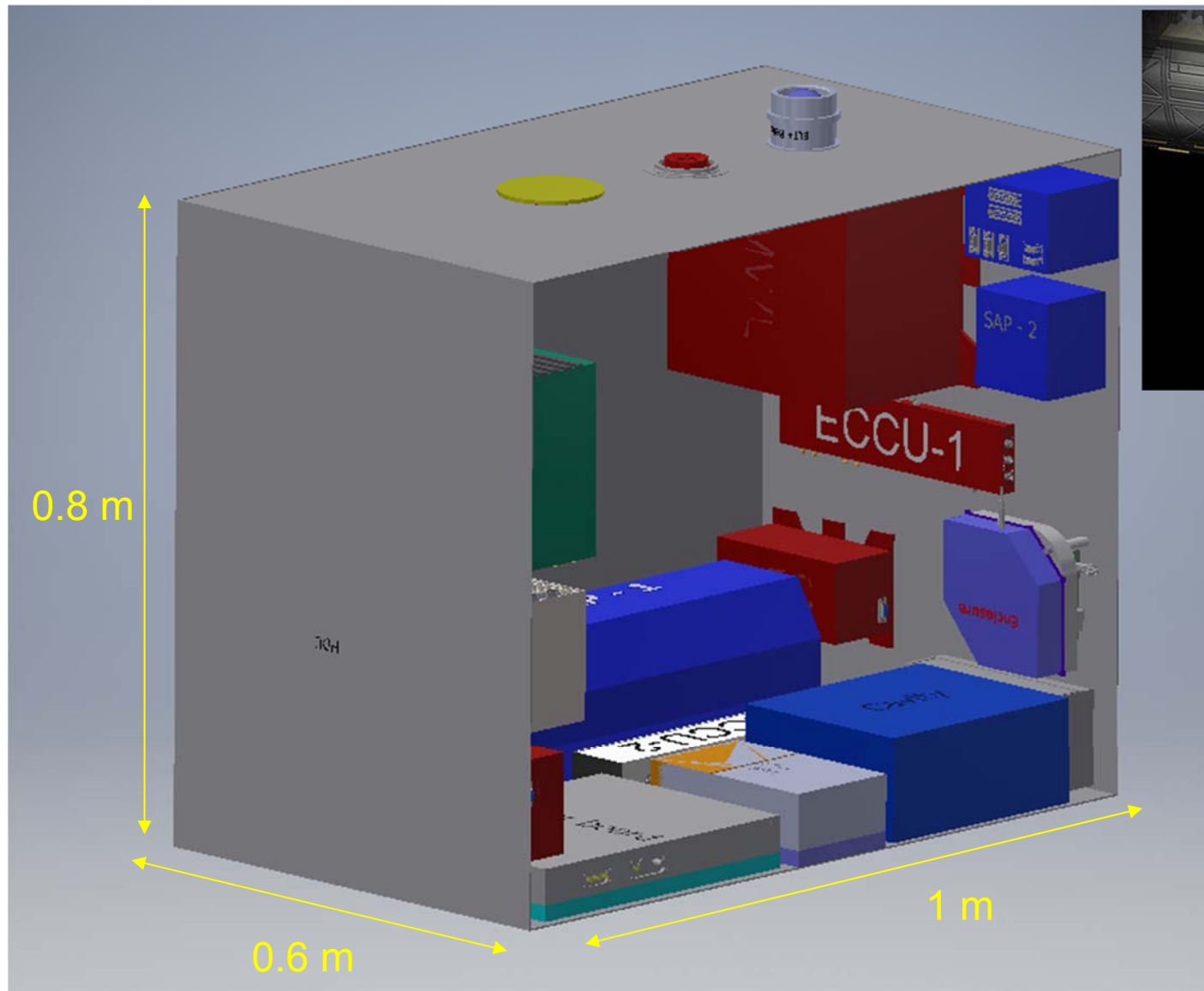
**ISS accomodation
Physical Requirements**

< 100 kg
(no margin considered)

< 480 liter
plus margin
($< 1 \times 0.6 \times 0.8 \text{ m}^3$)

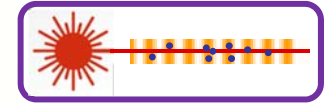
< 250 W
(no margin considered)

Accommodation

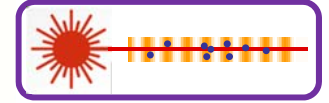


Accommodation
is possible
(with large margin)

I-SOC Phase A1: Documentation



I-SOC: Status



Delivered:

- | | |
|---------------------------------------|---------------------------------|
| 1) Experiment Science Document | 2/2017 [<i>Science team</i>] |
| 2) Phase A1: preliminary clock design | 3/2018 [<i>Science team</i>] |
| 3) Elegant Breadboard | 11/2018 [<i>Science team</i>] |

Ongoing:

- | | |
|---|--|
| 4) Further tests and enhancements on breadboard | |
| 5) ESA-funded technology developments | 2016-2019: |
| - 462 nm laser, 689 nm laser | [Fraunhofer CAP, TopGaN, CNR, <i>HHU</i>] |
| - CCU: laser frequency stabilization system | [NPL, <i>PTB</i>] |
| - 813 nm lattice laser | [Fraunhofer CAP, <i>SYRTE</i>] |
| - OSRC: reference cavity | [ADS, NPL, SpaceTech, ..., <i>PTB</i>] |
| - Two-way microwave link „HERO“ | [TAS, Wiser, DLR, <i>SYRTE</i>] |

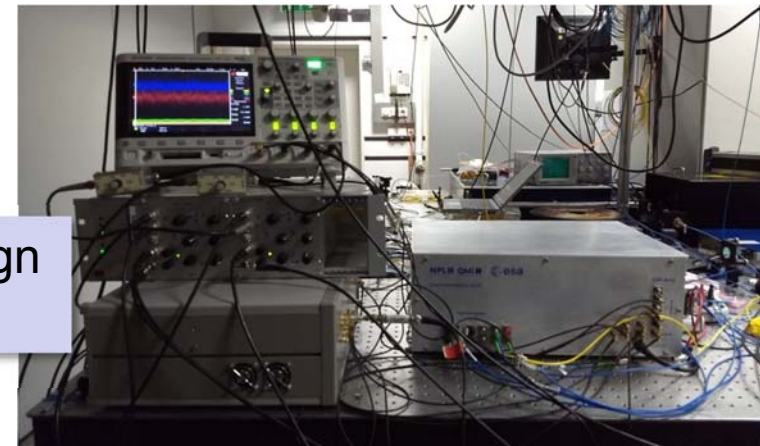
New approach by ESA:

Science team is involved

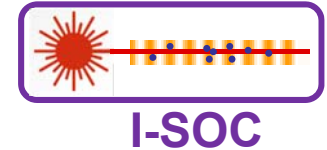
in all phases of mission development

Project is ready to enter next phase: detailed design and building of atomics and other subsystems

CCU @ *PTB*



Cost estimate



Cost after Completion (CAC) = **A** + **B** + **C**

A: subsystems with high TRL → only FM cost included

B: other subsystems

engineering model (EM) cost = breadboard model (BB) cost × 2.5
flight model (FM) cost = BB cost × 5

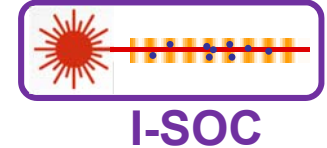
C: other (ground terminals upgrade, assembly, integration, testing;
product assurance; ground segment; flight operations;
assumed personnel costs: 0.2 M€ per person-year)

Total CAC (1 EM + 1 FM + Part C), incl. 20% margin:



Note: BB cost not included, since already built

Options for cost control / cost reduction



- **#1:** Some subsystems are national contributions, i.e. agency-furnished items.
- **#2:**
 - Involvement of research institutes in the technology development and testing activities (→ experience, lower labor costs, instrumentation)
 - Implement technology development phase with joint work of research institutes and industries
 - Involvement of scientists and their expertise for the evaluation of proposals submitted in response to ESA announcement of opportunities.

Capitalize on the increasing quantum technology know-how

- Less complex, reduced science reach; earlier launch: 2024-5
- **Scientific goal: common-view comparison of ground clocks, time-transfer, relativistic geodesy**

4 options;

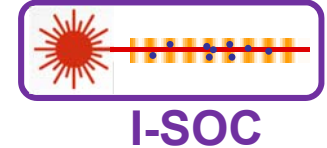
- MWL(extension of ACES) 9 M€
- **OR:** MWL+ (= HERO, improved performance: 10^{-18} level) 12.6 M€
- **ADDITIONALLY:** ELT-plus (improved performance: 10^{-18} level @ long integration time) 8.5 M€
- **ADDITIONALLY:** Technology demonstration + Test of Lorentz Invariance: high-performance cavity-stabilized laser + frequency comb 17 M€

- ESA member state ministers will decide on ESA budget in November 2019;
- The ISS utilization programme SpaceSciE will receive a fraction of that budget
- If total allocated ESA budget is large enough, there will be funds to implement new experiments, such as I-SOC pathfinder

It is important that as many national agencies as possible support strong funding of the SpaceSciE programme!

To get them to do so, the scientists have to express their support of experiments on the ISS, incl. I-SOC/I-SOC pathfinder

I-SOC: Key points



- I-SOC will enable continued and improved science results in fundamental physics, and the establishment of a new geodetic frame (via time)
- ... will provide opportunities to test key subsystems in a realistic environment
- ... will capitalize on developments done so far
- ... will keep momentum on time & frequency metrology technology also after ACES
 - not leaving the field only to Chinese
- ... will be an important (essential?) step towards future far-reaching missions on satellites requiring optical clock technology

If you wish to participate in I-SOC or help support it, please contact me:

step.schiller@hhu.de

Strategic considerations: opinion

If a mission such as AEDGE is ever to fly during our lifetime

- the interested scientific community needs to develop a **roadmap**
- Devote a **fraction** of their resources (time, funds) to cooperative efforts towards implementing it
- Proceed by **incremental** steps, FOKUS, ACES, CAL, ... I-SOC (Strontium!), gaining experience with space missions, technology development, and science
- Support such experiments by **actively** seeking support from national agencies
- **Involve** as many member states of ESA, also smaller countries
- Find a **compromise** between efforts towards ground-based devices (duplications!) and towards technology development for space
- Invent ways of cooperating as in CERN!

