# Forum on Tracking Detector Mechanics 2022

# **ATLAS Patch panel 01 (PP1) Engineering aspects and cooling system**

**INFN-LNF Tomassini Sandro; Rosatelli Filippo; Danè Emiliano 2022-06-08**







### *PP1 GLOBAL STRUCTURE*





The PP1 panel performs several functions in ATLAS ITK detector, the main ones are:

- Environmental gas sealing and Faraday cage closure of the detector volume.
- Support and housing of the detector services (i.e. High/Low voltage cables and connectors, 14k Data cables, 14 cooling pipe lines)
- Guarantee an internal dew point temperature lower than -60°C.
- Guarantee a maximum internal nitrogen overpressure of 4 mbar.
- Connection between the main plants in the pit and the subdetectors (Endcap, Outer barrel, Inner system).

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### *PP1 GLOBAL STRUCTURE*





Since the PP1 volume is in a high radiation environment, to reduce the mass budget, we decided to design the parts in Aluminium alloys (EN AW 6082-EN AW5083) made by milling technology.

- The thickness is increased only where there are threaded holes for bolted connections and gasket housings.
- The main parts are reported in the images above.
- The design allows the services integration sequence and test in the different assembly stages.

**2022/06/08**



#### *GAS SEALING-RADIATION LEVEL*

**Computed radiation dose** 

**@ 4000 fbarn^-1**



The conventional gasket materials are not eligible for gas sealing in the irradiated environment.

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#### *SEALING - O-RINGS*



**Shieldseal**®

663

#### EPDM elastomer with nuclear application life of up to 40 years

**General propertie** 

· Air

• Hot water

waste, including:

• Lower alcohols

• Ketones

· Dilute acids and alkalis

· Silicone oils and greases

· Steam

Specifically formulated to offer a

This elastomer is also suitable for

long-term contact with a wide variety

elastomer is in contact with





#### **Leachable ion conten**

Shieldseal 663 contains a very low long service life of up to 40 years in level of leachable ions such as CIapplications where ionizing radiation and SO42-, to ensure that items made of this material do not contribute to corrosion in metalwork

> Water soluble contents of sulphate and chloride are at levels below 3 ppm.

#### **Radiation/thermal resistance**

Third-party tests carried out in accordance with international standards show that Shieldseal 663 has good generic radiation resistance up to a dose of 1600 kGy in radiation conditions that include elevated temperatures up to 70°C. Thermal pre-ageing of the samples

did not significantly alter the end-of-life ageing characteristics.

#### **Third-party testing by AMEC**

Samples of initially un-aged and thermally pre-aged Shieldseal 663 from two different batches were irradiated at a dose rate of 1 kGy.h-1 up to 1000 kGy in a Co-60 irradiation facility. A number of samples were also irradiated at 70°C to assess synergistic effects.

Samples were then exposed to a further 600 kGy, at room temperature, to simulate additional radiation from a Design Basis Event (DBE) such as a loss of coolant accident (LOCA)

Levels of degradation were monitored periodically during radiation/thermal ageing by compression set measurement of button samples. Tensile test samples were aged in the same manner and tested at James Walker Technology Centre for hardness, elongation at break, and tensile strength

Mechanical testing of aged Shieldseal 663 dumbbell samples showed that, overall, the hardness, elongation at break and tensile strength for both the initially un-aged and the thermally pre-aged samples were similar for each test condition.

#### Elongation at break test results

Showed that the generally accepted end-of-life condition, defined as 50% elongation at break, was reached at a dose of 1600 kGy. (Typical elongation at break value for an un-aged, un-irradiated sample of Shieldseal 663 is 200%.)

#### **Compression set test results**

These showed that the generally accepted end-of-life value of 90% was reached at 1600 kGy. Irradiation at 70°C made



To seal the PP1 environment, the O-Rings material chosen is SHIELDSEAL® 663 (EPDM Family ) provided from James Walker and certified up to 1.6 MGy . Used in nuclear plants.

When mounted on dedicated prototypes, we can check the differences in terms of leak performance before and after irradiation.



# *O-RING QUALIFICATION UP TO 6 MGy*





We made three prototypes, to test and qualify a compressed JW O-rings, at three radiation level up to 6 MGy. O-rings will remain mounted to verify leakage at different operative conditions (rad. dose). O-R groove and thicknesses are equal to PP1 joints. Visual inspection after the test campaign. If the leak test will be passed, we can use the O-ring material to replace the metallic gasket in the PP1/IST interface. The benefits are:

- Simplify the interface integration.
- Using rubber instead of metal gasket.
- Mass budget reduction (project requirement).



## *LEAK TEST SYSTEM*





#### THE LEAK RATE OF O-RINGS WAS MEASURED WITH THE HELIUM DETECTION METHOD:

- 1. Vacuum on both lines.
- 2. Flushing Helium at p=1 bar absolute in line 2 (prototype)
- 3. Read the leak rate in time on vacuum leak detector (line1).

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#### *LEAK TEST RESULTS IRR 1*



• Send to US facility for radiation step 3 to obtain results on the most critical condition.







## *LEAK TEST RESULTS- ZERO RADIATION*



- I use this preliminary results to evaluate the leak rate of all gaskets used in PP1 connections.
- O-ring also in the cooling transfer lines.
- The total leak rate with a Safety Factor 4, is  $3.8 \times 10^{-2}$  mbar  $1/s$  He std
- The value allowed by specific is  $312 \text{ mbar}$ <sup>1</sup>/<sub>s</sub> He std. [https://edms.cern.ch/document/2019413/1]
- Data feedthroughs leak is not included. To be tested.



#### *DATA FEEDTHROUGHS*





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## *DATA FEEDTHROUGHS PROTOTYPE*





#### Main features:

- Combs to store data bundles in position.
- Three chambers to allow the filling of compounds to create the sealing.

Different prototypes to optimize the design and validate the filling process.



# *PROTOYPE AND FILLING TEST*





Prototype of Outer Feedthrough was made by Vero Clear (3D printed material):

- 1/3 of the Outer DFT length (red quote). The other two dimensions are the same.
- The combs were made separately to grant dimensional stability (3D printed).
- Grooves allow the correct positioning of combs in the DFT structure.
- Three volumes for compound injection.
- Closing flange with greasing nipples to inject resin inside chambers with greasing gun.

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## *FILLING PROCESS*





The filling process follows this step by step procedure:

- 1. Filling the chamber A with silicon compound (SC). It has high viscosity and low curing time compared to Araldite 2011.This properties are necessary to ensure a containment function. The air exits from vent hole, in opposite direction of filling. When the SC leaks out from hole, we proceed to the next injection hole.
- 2. Filling the chamber B with SC.
- 3. Now the chamber C is ready for Araldite 2011 injection. It's contained between the two full volumes (A & B) and ensure a tightness in the operative conditions of the detector, due its radiation resistance.



#### *FILLING TEST*





The pre sealing is the most critical aspect:

- Placing combs in position, we presealing them with silicon compound.
- Before placing bundles in position, we pre-seal them with silicon compound.
- Same for the closing flange to avoid spill during the filling operations.



### *FILLING TEST*



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- After 8 hours from the pre seal, we filled the chambers A and B with silicon compound. No leakage were detected.
- After 24 hours we filling the chamber C with Araldite 2011. No leakage were detected in the closing flange. Results:
- There is no differences between true twinax and fake parts in terms of leakage.
- Araldite and SC injected with greasing gun, with good results.
- No one of fillers leak out from prototype.
- Sectioned to verify the filling results.
- Preseal avoid resin leak.
- Araldite filled entirely the assigned volume.





- We use similar design for two 1:1 OS DFT prototypes, to test the resins filling process, in horizontal and vertical position. This, to simulate the integration process in PP1
- Leak test after the filling process.
- Combs will be modified to accommodate the ribbonized twinax bundles (when arrived).



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### *Outer Barrell piping: some details*



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### *Outer Barrell piping: influence of temperature*



# *Outer Barrell piping: floating manifolds*



3mm gap between the Outer Barrel Manifolds and the bulkhead is foreseen. The idea is to have the manifolds floating in order to compensate the thermal deformations.





#### *Outer Barrell Components in the exhaust line*





#### *Study of different shapes for the Exhaust connection to the flex lines*



In order to find the best position for the Electric Brake we are considering different shapes, in order to minimize the forces. Model will be more detailed in the following.



#### *Study of the forces on the ceramic component*



#### *PP1 Mockup*

#### Purpose:

- Simulate assembly procedure
- Rooming for pipes and cables
- Handling with the tools

Material: ASA, ABS

Populate with data cables in the

future.





# BACKUP SLIDES



#### *FILLING TEST*





#### Result of sealing:

- Sectioned to verify the filling results.
- Preseal avoid resin leak.
- Araldite filled entirely the assigned volume.



#### Update after Diego Input





#### Final Configuration Update after meeting with Bart Nov-5th-21



#### *Constraints:*

Lines length as short as possible: OB-CG7, OB-CG5, EC-L4 *Conditions:*

OBH2 should be rotated  $(\pi)$  respect global Z axis to obtain OBH1 OB A-side should be rotated  $(\pi)$  respect global Y axis to obtain C-side EC should be symmetric respect global XY plane to obtain C-side

- All constraints complied (OB-CG7, OB-CG5, EC-L4)
- OB lines not satisfy the OB conditions:
	- Side AH1: OB-CG1 <-> OB-CG3
	- Side CH1: OB-CG1 <-> OB-CG3
	- $Side CH2: OB-CG4 \leq DBCG2$
- EC not symetric for ECL2 and ECL3 side A- side C

#### Inner System and Endcap Manifolds





# 4. welding feasibility

- Questions:
- Do the capillaries have to be workable (for repairing) once installed?
- Can we weld and test half capillaries (the below ones) before and then the others?
- Can we eventually dismount from the BH and repair? • Vista dall'alto per far capire che lo spazio ci sarebbe
- Can we assembly them after they are tested?







