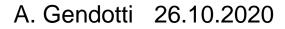
# SFGD Box Update



ETH

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- FEA General •
  - - First Static Study
    - Second Static Study
    - Third Dynamyc Study PSD Response  $\rightarrow$  Slide 8-12 •
- FEA Conclusion
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## SUMMARY OF THE FEA

## **Model Simplification**

- All the materials properties are tuned according to the INR tests and includes holes
- Carbon Fiber is modeled as a stackup of 8 Layers according to CompositeDesign prototipe (2,3mm thickness of 8 Layers at different angles)
- Bolts conections between Panels-Panels and Bracket-Panels are not implemented (Bonded contact is used)
- Pads for the PCBs simplified by taking the thinner thickness of the Panel (coservative)
- Weight of the cubes is simulated as a Force in order to be conservative.
- For the Vibration Study attempt to caracterize the cubes

#### FEA Studies

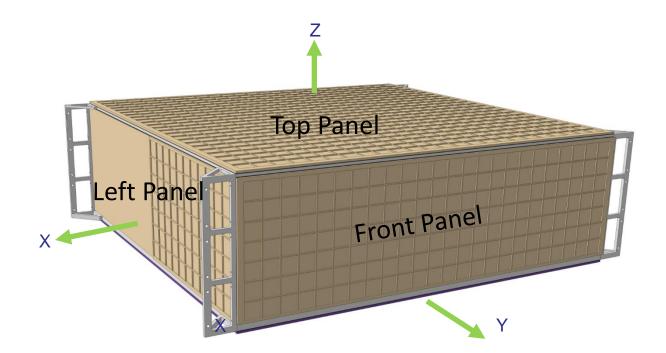
- First Static study is considered as the normal operation load: Earth Gravity at the Box and Cube weight.
- Second Static Study is cosidering Earth Gravity and 0.65g aceleration at each direction. This in order to spot critical «situation» at the level of stresses and deformations
- Third dinamic study is considering the PSD spectrum provided by KEK and simulate the PSD response at each direction. In this study the cubes have to be implemented. Since is very difficult to caracterize the Cubes behaviour, a simpification was needed. Anyway the study, to my point of you, can give a good idea of the situation. In PSD study all connection are bonded (linear)



## SUMMARY OF THE FEA

### Sandwich Materials and Thickness

- Accroding to the design at the current status the Material used for the Box are below listed:
  - Bottom Panel
  - Top Panel:
  - Front and Back Panels:
  - Left and Righ Panels:
- → Carbon Fiber Acrylic Carbon Fiber
- $\rightarrow$  Carbon Fiber Acrylic Carbon Fiber
- → Carbon Fiber Divinycel H250 Carbon Fiber
- → Carbon Fiber Divinycell H250 Carbon Fiber
- Thickness 33.6 mm Thickness 16.6mm Thickness 19.6mm Thickness 24.6mm





# SUMMARY OF THE FEA

Glass Epoxy Readout Panel

Soft Foam

- Readout panels and hard panels for Cubes Installation (blue) are G10 (glass epoxy)
- Soft Foam Panels (yellow) consists in Polipropilene Foam (used only when cubes are implemented)
- Brackets material is Aluminum

Glass Epoxy for Cube Installation

Glass Epoxy Readout Panel



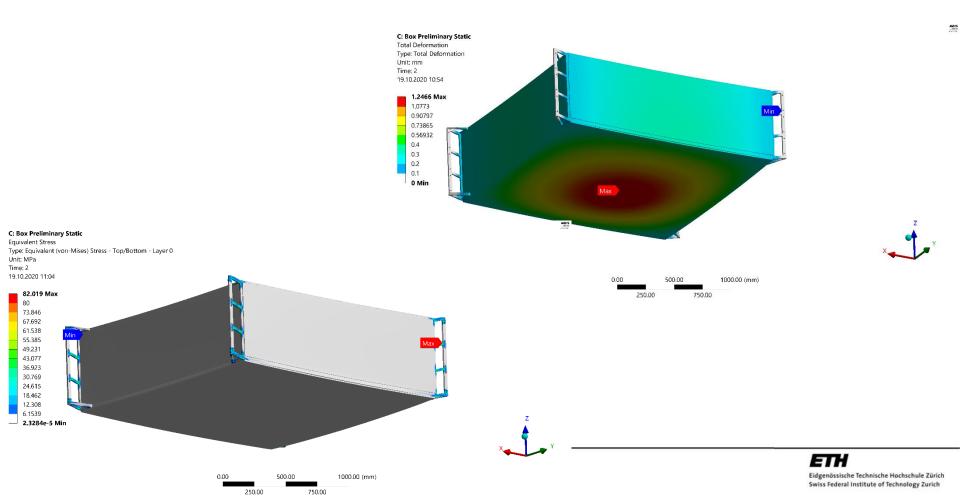
## FIRST STATIC STUDY

## Gravity on z axis (Weight of the cubes with force 2.22 Tons)

#### Z- direction:

Maximum deformation at the Bottom Panel: Maximum deformation at the Top Panel: Max Stress at the Brackets: Max Stress at Alu Frames (Bracket connect.): Max Stress at Acrylic Bottom panel Failure Safety Factor at Carbon Fiber

1.5 mm
0.85 mm
82 Mpa
35.5 Mpa
~1.26 Mpa
SF: 4



# SECOND STATIC STUDY

# <u>Gravity on z axis + 0.65g in every direction (Weight of the cubes with 0.65g $\rightarrow$ 3.7 Tons)</u>

## Z- direction:

Maximum deformation at the Bottom Panel: Maximum deformation at the Top Panel: Max Stress at the Brackets: Max Stress at Alu Frames (Bracket connect.): Max Stress at Acrylic Bottom panel Failure Safety Factor at Carbon Fiber

#### 2.03 mm ~1.4 mm 135.11 Mpa ~50Mpa ~2Mpa SF: 3.8

Directional Deformation Type: Directional Deformation(Z Axis)

> -0.41559 0.56221 -0.70882

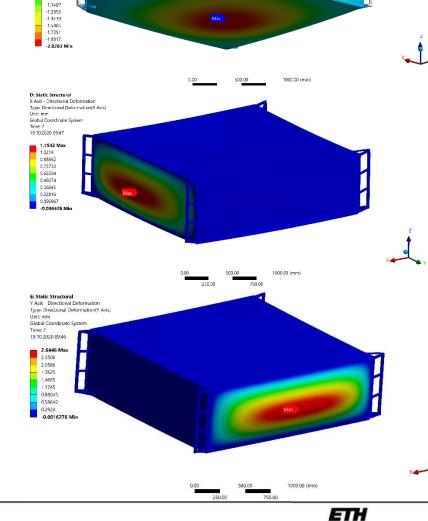
-0.85543 -1.002

Unit: mm Global Coordinate System

Time: 3 19.10.2020 09:49 0.024244 N -0.12237 -0.26898

#### X- direction:

Maximum deformation at the Left/Right Panel: Max Stress at the Brackets: Max Stress at Alu Frame Max Stress at Foam core 1.16 mm 86.26 Mpa <10Mpa <10Mpa



#### Y- direction:

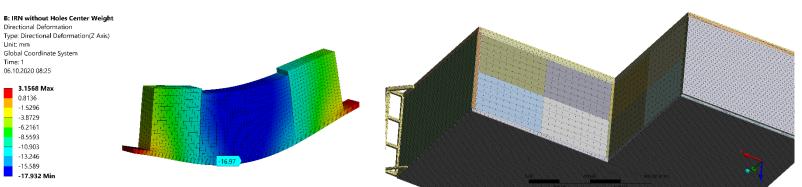
Maximum deformation at the Front/Back Panel: Max Stress at the Brackets: Max Stress at Alu Frame Max Stress at Foam core Failure Safety Factor at Carbon Fiber 2.65 mm 85.8 Mpa 46 Mpa 10Mpa SF: 4

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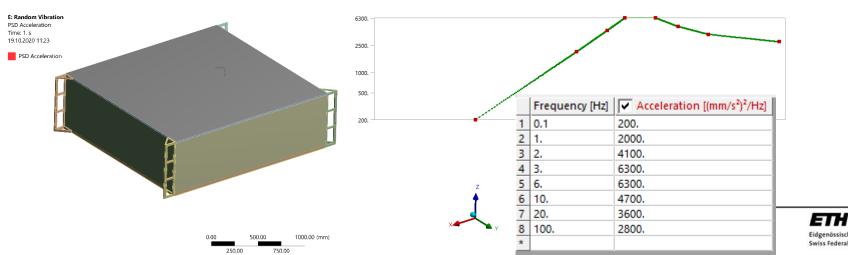
44172

## THIRD DINAMIC STUDY – PSD Response

 Cubes caracterized as a solid (best approximation up to now). Cubes Envelope divided in 32 Solid and Young's modulus lowered (jelly kind solid)



- Static calculation (non linear to verify if results are close to the first study)
- Modal Analysis → automatically considered as linear (all the contacts are bonded)
- PSD response from acceleration spectrum given by KEK
   → Random Vibration applied at the Bracket fixed supports and calculated at all 3 directions with preload from static



THIRD	STUDY	– PSD	Response
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#### Maximum deformation at the Bottom Panel: Maximum deformation at the Top Panel: Max Stress at the Brackets: Max Stress at Alu Frames (Bracket connect.): Max Stress at Acrylic Bottom panel Failure Safety Factor at Carbon Fiber

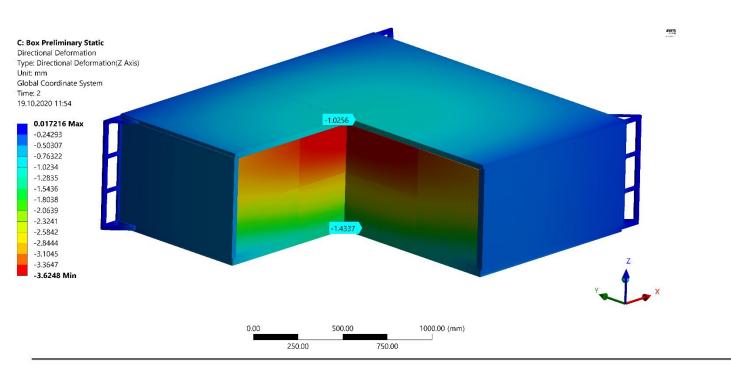
1.44 mm
1 mm
82 Mpa
27 Mpa
~1.26 Mpa
SF: ~3.8

# Static Calculation verification

# Very similar to the first static Study

Maximum deformation at the Bottom Panel:	1.5 mm
Maximum deformation at the Top Panel:	0.85 mm
Max Stress at the Brackets:	82 Mpa
Max Stress at Alu Frames (Bracket connect.):	35.5 Mpa
Max Stress at Acrylic Bottom panel	~1.26 Mpa
Failure Safety Factor at Carbon Fiber	SF 4

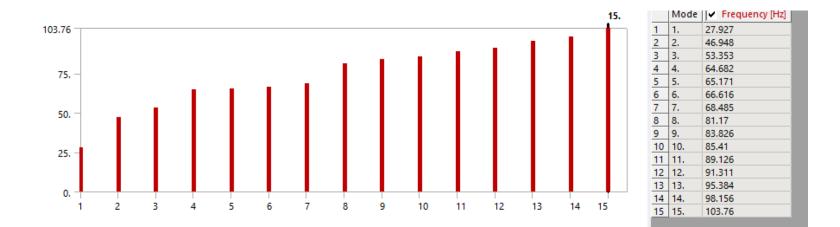
Cubes behave almost as pressure due to low Young's module



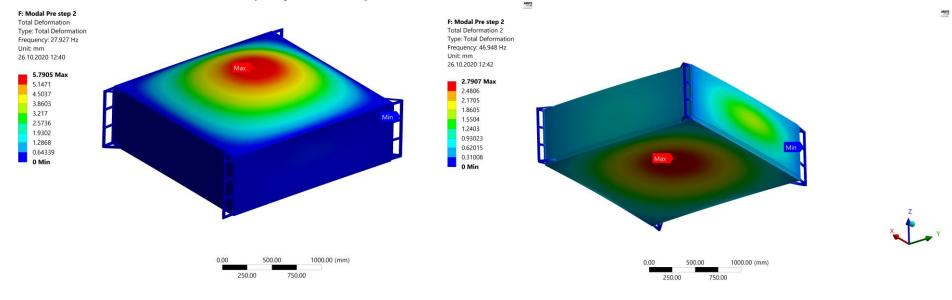


## Modal without cubes implemented

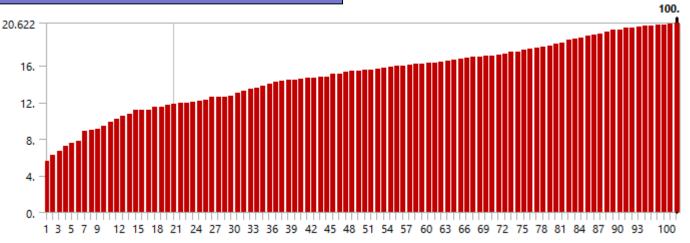
Second mode at ~47 Hz (Bottom Panel)



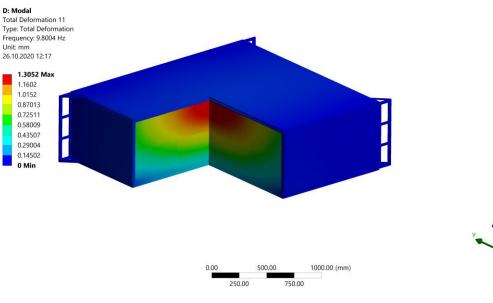
## First mode at ~28 Hz (Top Panel)



## Modal with cubes implemented



Modal Analysis at preloaded static model with cubes implemented: first 100 modes are at the cubes.



4.8419e-9

0.1

0.25

0.5

1.

2.5

5.

10.

Random Vibration in Vertical Direction 9.7979 [mm<sup>2</sup>/Hz] 1190.4 50. 1. **Top Panel** 1.e-2 (re-calculating right now) 1.e-4 1.e-6 1.0681e-8 0.25 0.5 2.5 50. 0.1 1. 5. 10. 25. 100. AKSYS E: Random Vibration **Directional Deformation** Type: Directional Deformation(Z Axis) Scale Factor Value: 3 Sigma Probability: 99.73 % Unit: mm Solution Coordina Time: 0 26.10.2020 12:22 2.7372 Max 2.4331 2.1289 1.8248 1.5207 1.2165 0.9124 0.60827 0.71224 0.30413 0 Min 500.00 1000.00 (mm) 250.00 750.00 9.6998 [mm<sup>2</sup>/Hz] 1275. 10. 0.1 **Bottom Panel** 1.e-3 1.e-5 1.e-7

!Checked all the responses in all directions and the critical case is in the vertical direction at the Top Panel!

Scale Factor Probability: Max Def Top Panel Max Def. Bottom Max Strees:

f (Hz)

f (Hz)

50.

100.

25.

3 sigma 99,73 % 0.25mm 0.72mm 11Mpa (@Bracket)



## Conclusion

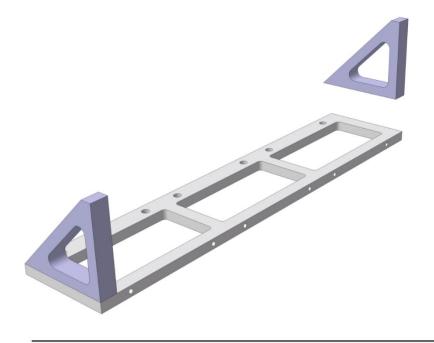
- Good caracterization of the Cubes behaviour is almost impossible.
- Deformations and stresses are considered mostly in order to get an idea of the Box behaviour under Random vibrations.
- Stresses and Deformations at the panels are always very low.
- Considering the real case of having all the cubes with frictional contact (to eachother and to the Panels) it would reduce significantly the deformation at the panels due to vibrations. The single cubes would damp a lot the vibrations. → !!My Opinion!!
- Top Panel would be actually not connected to the Cubes and free to vibrate. Increasing the thickness of the Top panel would anyway increase the first mode vibration and reduce the deformation.



• **Increase** top Panel thickness same as the Bottom Panel and use Divinycell H250 as a core instead of Acrylic.

 $\rightarrow$ This would give less deformation in case of an Earthquake and would slightly increase the Resonance frequency.

- $\rightarrow$  Reduce the possibility to hit the cubes in case of shaking
- Use 316L (non magnetic stainless steel) for the Bracket → Giving an extra safety. In case of using Aluminum, it would be necessary to use special/reinforced alloy.



Divide the Bracket in parts bolted or welded togheter. The shape could be lasered (company I'm in contact can laser up to 30mm Stainless Steel)

## EMDE Laser Company

Carbon steel Stahl bis 25mm

- Stainless steel Edelstahl bis 30 mm
  - Alu 
    Aluminium bis 25 mm
  - Copper 

    Kupfer bis 10 mm
  - Brass Messing bis 10 mm



- Accroding to the design at the current status the Material used for the Box are below listed:
  - Bottom Panel → Carbon Fiber Acrylic Carbon Fiber Thickness 33.6 mm→ waiting
     Waiting for composite design to know the exact core dimension between 29mm and 30mm
- → Carbon Fiber Acrylic Carbon Fiber Top Panel: Thickness 16.6mm → Change ٠ Increase the thickness at 33.6mm (check with Franck if there is enough clearance) and Divinycell Core Front and Back Panels: → Carbon Fiber – Divinycel H250 – Carbon Fiber Thickness 19.6mm ٠ Left and Righ Panels: → Carbon Fiber – Divinycell H250 – Carbon Fiber Thickness 24.6mm ٠ Carbon Fiber Layer is always 2.3mm Glue is 0.1mm Top Panel Left Panel **Front Panel** Y

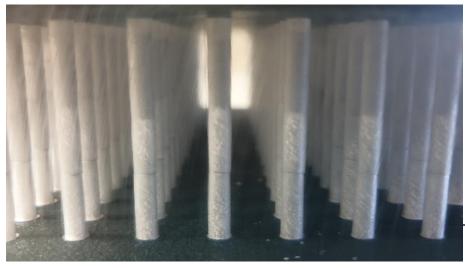


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# ACRYLIC PROTOTYPE









- 200x200mm prototype received from CompositeDesign
- 3mm Holes drilled through by CIMFORM company
- Visual Check doens't show any crack in the acrylic and glued parts looks good.
- Contacted CERN to perform stress test to verify bending etc..
- Purchasing New Prototype for INR:
  - Black G10
  - Black Acrylic
  - Size of 2000mmx150mm

